## I will be using a Newton Raphson Solver for this homework

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M1 = 1.55;% Mach Number
t = 5; % Theta in degrees
t = deg2rad(t);
g = 1.4; % gamma, the gas constant for air
R = 287; % ideal gas constant for air
cp = -(R*g)/(1-g);
f = @(b)((2*\cot(b)*((M1^2)*(\sin(b))^2 - 1))/((M1^2)*(g + \cos(2*b)) + 2) - \tan(t)); \% b is Beta
iter = 0;
syms bsym
df = matlabFunction(diff(f(bsym)));
bm_func = @(f,b)(b-f(b)/df(b));
err = 1;
tol = 0.001;
maxiter = 50;
start = 1;
bm = start;
bmold1 = bm;
bmold2 = bm+0.01;
while err>tol && iter<maxiter</pre>
   iter = iter +1;
    bm = bm_func(f,bmold1);
    err = abs((bm-bmold1)/bm)*100;
    bmold2 = bmold1;
    bmold1 = bm;
end
if t > 0
    b = rad2deg(bm); % b is beta in degrees
elseif t == 0
    b = 90;
    bm = pi/2;
M1n = M1*sin(bm); % bm is beta in radians
M2n = sqrt((1 + ((g-1)/2)*(M1n^2))/(g*(M1n^2) - ((g-1)/2)));
M2 = (M2n)/\sin(bm-t);
p21 = 1 + (2*g*(M1n^2 - 1))/(g+1); % static pressure ratio
dels = cp*log(p21*(2 + (g-1)*M1n^2)/((g+1)*M1n^2)) - R*log(p21); % change in entropy
po21 = exp(-dels/R); % stagnation pressure ratio
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