

A) Two-Body Problem

$$B) \sum \vec{F}^Q = m_Q \ddot{\vec{a}}^Q$$

$\uparrow \quad \vec{F}_g^Q$

C) UNITS

D) Basis Transformation

(Direction Cosine Matrices)

DCM

$$\sum \vec{F}^Q = \underbrace{\vec{F}_g^Q}_{\vec{F}_{g_{Earth}}} + \vec{F}_{g_{Sun}}^Q + \vec{F}_{g_{Moon}}^Q + \vec{F}_{sp}^Q + \vec{F}_{mag}^Q + \vec{F}_{Aero}^Q$$

\uparrow
Solar Pressure

$$F_g = \frac{GMEm}{r^2} \Rightarrow \vec{F}_g^Q = -\frac{GMEm_Q \vec{r}^{QOQ}}{(r^{QOQ})^3}$$

$$r^{QOQ} = |\vec{r}^{QOQ}|$$

$$\Rightarrow -\frac{GMEm_Q \vec{r}^{QOQ}}{(r^{QOQ})^3} = m_Q \ddot{\vec{a}}^Q$$

$$\Rightarrow \mu \Rightarrow \boxed{-\frac{\mu \vec{r}^{QOQ}}{(r^{QOQ})^3} = \ddot{\vec{a}}^Q}$$

$$\mu \approx GM_E$$

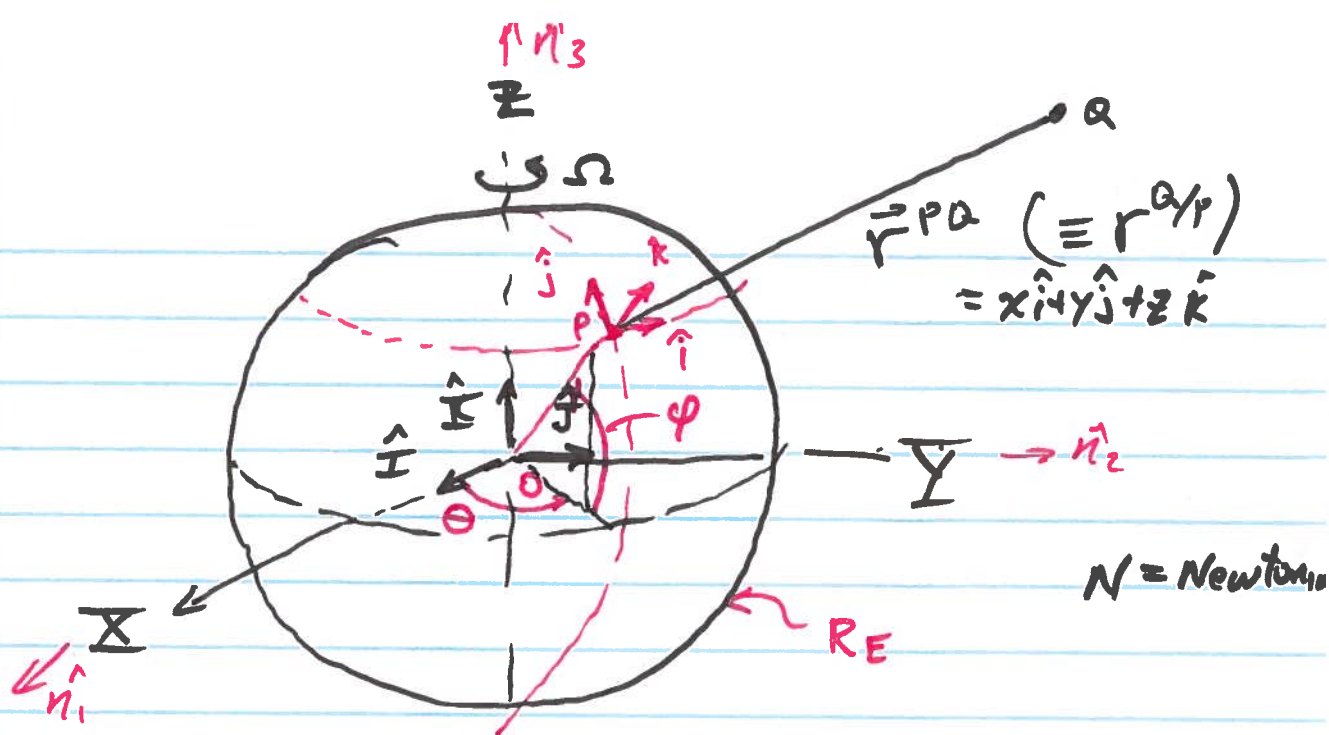
$$\mu_E = 398600 \text{ km}^3/\text{s}^2$$

$$\mu = G(M_E + m_Q)$$

$$SI: [kg] [m] [s] \\ (km)$$

$$English [lbf] [ft] [s]$$

$$\frac{F_1}{a_1} = \frac{F_2}{a_2} = \frac{F_n}{a_n} = m$$



$$\vec{F}_g^Q = m_Q \vec{a}^Q$$

$$\Rightarrow \frac{-\mu \vec{r}^{OQ}}{(r^{OQ})^3} = \vec{a}^Q$$

$$= \vec{a}^P + \vec{a}^{Q/P} + \vec{\omega}^A \times \vec{r}^{PQ} + \vec{\omega}^A \times (\vec{\omega}^A \times \vec{r}^{PQ}) + 2\vec{\omega}^A \times \vec{v}^{Q/P}$$

$A \equiv E$

$$\vec{r}^{OQ} = \vec{r}^{OP} + \vec{r}^{PQ}$$

$$R_E = 6378 \text{ km}$$

$$= R_E \hat{k} + (x\hat{i} + y\hat{j} + z\hat{k})$$

$$= x\hat{i} + y\hat{j} + (R_E + z)\hat{k} \quad (x, y, z) \text{ Measured}$$

$$\vec{v}^{Q/P} = \frac{d}{dt} \vec{r}^{PQ} = \frac{d}{dt} (x\hat{i} + y\hat{j} + z\hat{k})$$

$$= \dot{x}\hat{i} + \dot{y}\hat{j} + \dot{z}\hat{k} + x \frac{d\hat{i}}{dt} + y \frac{d\hat{j}}{dt} + z \frac{d\hat{k}}{dt}$$

$$\vec{a}^{Q/P} = \frac{d}{dt} \vec{v}^{Q/P} = \ddot{x}\hat{i} + \ddot{y}\hat{j} + \ddot{z}\hat{k}$$

$$\vec{\omega}^E = \Omega \hat{K}$$

$$\Omega = \frac{2\pi}{1 \text{ day}} [\text{s}]$$

$$\Omega \sim 7.292 \times 10^{-5} \text{ rad/s}$$

$$1 \text{ day} = 23 \text{ hr } 56 \text{ min } 4 \text{ sec} \\ \sim 23.9344?$$

$$\vec{a}^Q = \vec{a}^P + \vec{a}^{Q/P} + \vec{\omega}^E \times \vec{r}^{PQ} \\ + \vec{\omega}^E \times (\vec{\omega}^E \times \vec{r}^{PQ}) \\ + 2\vec{\omega}^E \times \vec{v}^{Q/P}$$

$$\vec{a}^P = \vec{a}^O + \vec{a}^{P/O} + \vec{\omega}^E \times \vec{r}^{OP} \\ + \vec{\omega}^E \times (\vec{\omega}^E \times \vec{r}^{OP}) \\ + 2\vec{\omega}^E \times \vec{v}^{P/O} \\ = \underbrace{\vec{\omega}^E \times (\vec{\omega}^E \times \vec{r}^{OP})}_{\hat{R} \hat{E} \hat{K}}$$

\Rightarrow MUST move from $\hat{i}, \hat{j}, \hat{k}$ to $\hat{I}, \hat{J}, \hat{K}$
and back easily

$$\vec{r}^O = A_1 \hat{a}_1 + A_2 \hat{a}_2 + A_3 \hat{a}_3 \Rightarrow \begin{Bmatrix} A_1 \\ A_2 \\ A_3 \end{Bmatrix}_A \\ = X \hat{I} + Y \hat{J} + Z \hat{K} \Rightarrow \begin{Bmatrix} X \\ Y \\ Z \end{Bmatrix}_N \\ = B_1 \hat{b}_1 + B_2 \hat{b}_2 + B_3 \hat{b}_3 \Rightarrow \begin{Bmatrix} B_1 \\ B_2 \\ B_3 \end{Bmatrix}_B$$

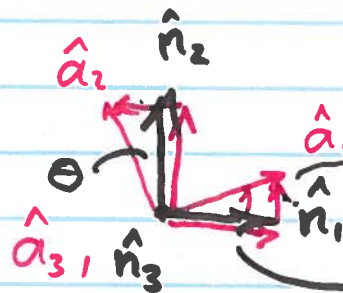
$$\begin{Bmatrix} A_1 \\ A_2 \\ A_3 \end{Bmatrix}_A \neq \begin{Bmatrix} B_1 \\ B_2 \\ B_3 \end{Bmatrix}_B \neq \begin{Bmatrix} X \\ Y \\ Z \end{Bmatrix}_N$$

$$\begin{Bmatrix} B_1 \\ B_2 \\ B_3 \end{Bmatrix}_B = \begin{bmatrix} B & A \\ C & \end{bmatrix} \begin{Bmatrix} A_1 \\ A_2 \\ A_3 \end{Bmatrix}_A$$

↑
WANT

↑
HAVE

Rot #1



$$\hat{a}_1 = \cos\theta \hat{n}_1 + \sin\theta \hat{n}_2$$

$$\hat{a}_2 = -\sin\theta \hat{n}_1 + \cos\theta \hat{n}_2$$

$$\hat{a}_3 = \hat{n}_3$$

	\hat{n}_1	\hat{n}_2	\hat{n}_3
\hat{a}_1	$\cos\theta$	$\sin\theta$	0
\hat{a}_2	$-\sin\theta$	$\cos\theta$	0
\hat{a}_3	0	0	1

$$\begin{aligned} \hat{n}_1 &= \cos\theta \hat{a}_1 - \sin\theta \hat{a}_2 \\ \hat{n}_2 &= \sin\theta \hat{a}_1 + \cos\theta \hat{a}_2 \\ \hat{n}_3 &= \hat{a}_3 \end{aligned}$$

$${}^A C^N = \begin{bmatrix} \cos\theta & \sin\theta & 0 \\ -\sin\theta & \cos\theta & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

5