

- Basic Rocketry

A) Some history

B) Rocket Thrust

C) Basic Rocket Equation

When did it all begin

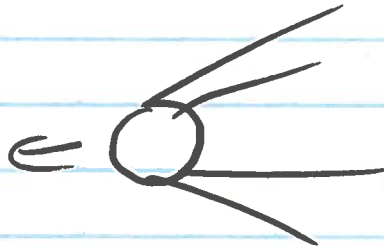
china against monguls ~1232

"Fire Arrow" 

Konstantin Tsiolkovsky 1857 

Robert Goddard - 1882

1957



→

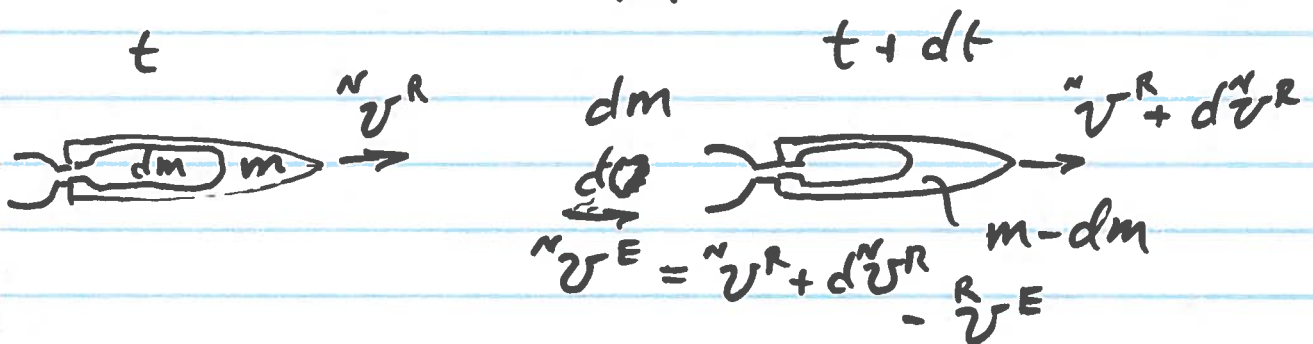
Appollo Saturn-V

$$\Sigma \vec{F} = m \vec{a} \Rightarrow \Sigma \vec{F} = \frac{d}{dt} (m \vec{v})$$

$$\Sigma F = F_g + F_D + \vec{F}_{\text{thrust}} = \cancel{\frac{d}{dt}} m(t) \vec{a}^R$$

gravity \uparrow
Drag

$$\begin{aligned} \Sigma \vec{F} &= \frac{d}{dt} (m \vec{v}) = \underbrace{\vec{v}^R \left[\frac{dm}{dt} \right]}_{\dot{m}} + m \underbrace{\frac{d \vec{v}^R}{dt}}_{\vec{a}^R} \\ &= \underbrace{\vec{v}^R \dot{m}}_{\text{Thrust}} + m_R \vec{a}^R \end{aligned}$$



$$dm \vec{v} =$$

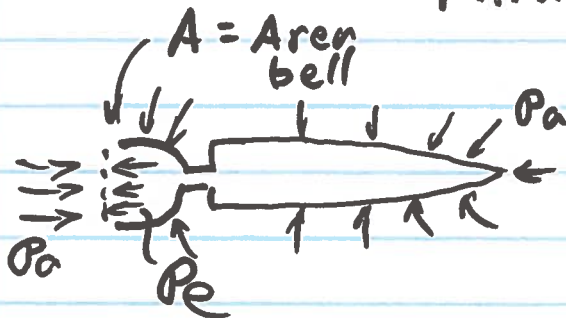
$$\begin{aligned} & \left[(m - dm) (\vec{v}^R + d\vec{v}^R) + dm (\vec{v}^R + d\vec{v}^R - \vec{v}^E) \right] \\ & - [m \vec{v}^R] \end{aligned}$$

$$\begin{aligned} &= [m \vec{v}^R + m d\vec{v}^R - dm \vec{v}^R - dm d\vec{v}^R + dm \vec{v}^R \\ &+ dm (\vec{v}^R - \vec{v}^E) - m \vec{v}^R] \\ &= m d\vec{v}^R - (dm) \vec{v}^E \end{aligned}$$

$$\Sigma F_x dt = dP = m d\vec{v}^R - dm \vec{v}^E$$

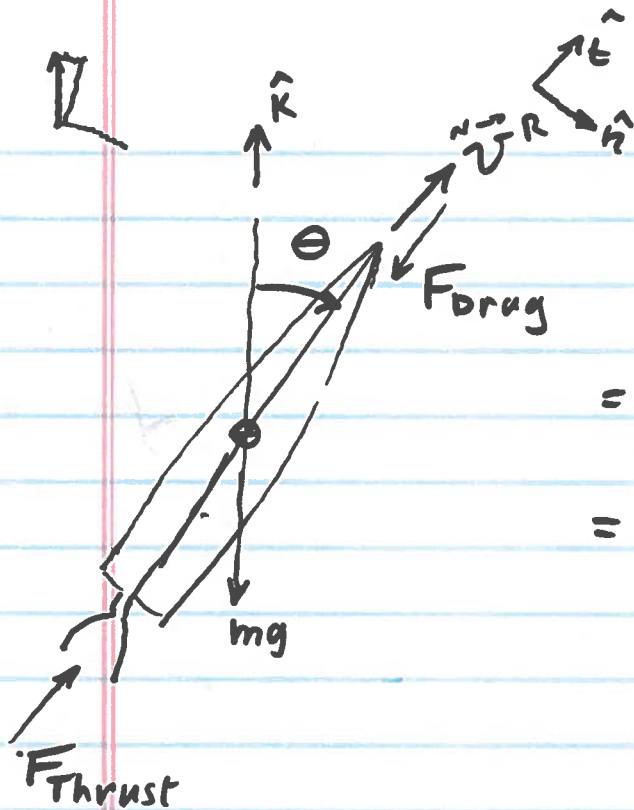
$$\Sigma F_{\text{ext}} = m \left(\frac{d\vec{v}^R}{dt} \right) - \left(\frac{dm}{dt} \right) \vec{v}^R E$$

$$\Rightarrow \Sigma F_{\text{ext}} + \underbrace{\left(\frac{dm}{dt} \right) \vec{v}^R E}_{\dot{m}_b \text{ Thrust}} = m \underbrace{\left(\frac{d\vec{v}^R}{dt} \right)}_{\vec{a}^R}$$



$$\Rightarrow (p_e - p_a) A$$

$$\Rightarrow \Sigma F_{\text{ext}} = F_g + F_D + \underbrace{[(p_e - p_a) A + \dot{m} \vec{v}^R E]}_{\text{static Thrust}} = m \vec{a}^R$$



$$\hat{k} = \hat{c} \hat{t} - \hat{s} \hat{n}$$

$$\begin{aligned} \Sigma \vec{F}_{ext} &= m(t) \vec{\ddot{a}}^R \\ &= F_g \hat{k} - \frac{1}{2} \rho C_D v^2 A_s \hat{t} + F_T \hat{t} \\ &= \frac{-m\mu}{(R_E + h)^2} [\hat{c} \hat{t} - \hat{s} \hat{n}] \\ &\quad - \frac{1}{2} \rho C_D |v| v A_s \hat{t} \\ &\quad + (\dot{m} \hat{v}^E) \hat{t} = m \vec{\ddot{a}}^R \\ &= m(\dot{v} \hat{t} + \underbrace{\left\{ \frac{v^2}{\rho} \right\}}_{\hat{n} \left(\frac{v \dot{\theta}}{\rho} \right)}) \end{aligned}$$

$$\Rightarrow \rho \dot{\theta} = v$$

$$\hat{n} \hat{t}: \frac{m\mu s \theta}{(R_E + h)^2} = m v \dot{\theta}$$

$$\begin{aligned} \hat{t}: \quad & \frac{-m\mu c \theta}{(R_E + h)^2} - \frac{1}{2} \rho C_D |v| v A_s + \underbrace{(\dot{m} \hat{v}^E)}_{\text{Thrust}} \\ &= m \dot{v} \end{aligned}$$

$$\dot{\theta} = \frac{\mu s \theta}{v (R_E + h)^2} + (\dot{m} \hat{v}^E)$$

$$\dot{v} = \frac{1}{m} \left\{ \frac{-m\mu c \theta}{(R_E + h)^2} - \frac{1}{2} \rho C_D |v| v A_s \right\}$$

$$\dot{h} = v \cos \theta$$

$$\dot{x} = v \sin \theta$$

