

1) Kepler's 2nd Law $\frac{dA}{dt} = \frac{h}{2}$

2) Eccentricity \vec{e}

3) Actual Anomaly Θ

4) Position & Velocity

$$r(\Theta) = \frac{h^2}{\mu(1+e\cos\Theta)}$$

$$v_{\Theta}(\Theta) = \frac{\mu}{h} (1+e\cos\Theta)$$

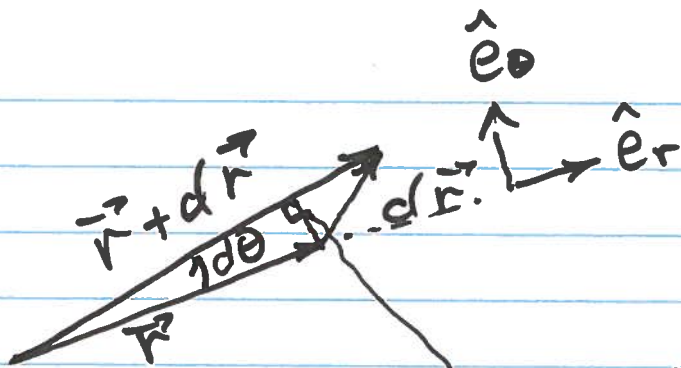
$$v_r(\Theta) = \frac{\mu}{h} e \sin\Theta$$

4) circular orbits $e=0$

5) Parabolic trajectory $-1 < e < 1$

6) Hyperbolic trajectories $e > 1$





$$\vec{r} = r \hat{e}_r$$

$$\vec{v} = \dot{r} \hat{e}_r + r \dot{\theta} \hat{e}_\theta$$

$$\vec{h} = \vec{r} \times \vec{v}$$

$$r \hat{e}_r \times (\dot{r} \hat{e}_r + r \dot{\theta} \hat{e}_\theta)$$

$$\underbrace{r^2 \dot{\theta}}_h \hat{k}$$

$$dA = \frac{1}{2} (r + dr) (r d\theta)$$

$$= \frac{1}{2} (r^2 d\theta + \cancel{r dr d\theta})$$

$$\frac{dA}{dt} = \frac{1}{2} r^2 \frac{d\theta}{dt} = \frac{1}{2} \underbrace{r^2 \dot{\theta}}_h$$

Eccentricity \vec{e}

$$\vec{F}_g = m_a \vec{a}$$

$$\vec{f}_g = \vec{a}$$

$$-\frac{\mu \vec{r}}{r^3} = \ddot{\vec{r}}$$

$$\left\{ \begin{aligned} -\frac{\mu \vec{r}}{r^3} \times \vec{h} &= \ddot{\vec{r}} \times \vec{h} \end{aligned} \right.$$

$$\frac{d}{dt}(\dot{\vec{r}} \times \vec{h}) = \ddot{\vec{r}} \times \vec{h} + \dot{\vec{r}} \times \frac{d}{dt} \vec{h}$$

$$\vec{r} \times \vec{h} = \vec{r} \times (\vec{r} \times \dot{\vec{r}}) = \vec{r}(\underbrace{\vec{r} \cdot \dot{\vec{r}}}_{r\dot{r}}) - \underbrace{\dot{\vec{r}}(\vec{r} \cdot \vec{r})}_{r^2}$$

$$\vec{A} \times (\vec{B} \times \vec{C}) = \vec{B}(\vec{A} \cdot \vec{C}) - \vec{C}(\vec{A} \cdot \vec{B})$$

$$\vec{r} \cdot \dot{\vec{r}} = r \hat{e}_r \cdot (\dot{r} \hat{e}_r + r \dot{\theta} \hat{e}_\theta) = r \dot{r}$$

$$\Rightarrow \frac{\vec{r} \times \vec{h}}{r^3} = \frac{\vec{r} \dot{r} - \dot{\vec{r}} r}{r^2} = -\frac{d}{dt} \left(\frac{\vec{r}}{r} \right)$$

$$\Rightarrow \frac{d}{dt} \left(\frac{\vec{r}}{r} \right) = \frac{d}{dt} (\dot{\vec{r}} \times \vec{h})$$

$$\frac{d}{dt} \left[\underbrace{\dot{\vec{r}} \times \vec{h}}_{=\vec{c}} - \mu \left(\frac{\vec{r}}{r} \right) \right] = 0$$

$$\dot{\vec{r}} \times \vec{h} - \mu \left(\frac{\vec{r}}{r} \right) = \vec{C}$$

$$\Rightarrow \frac{\vec{v} \times \vec{h}}{\mu} - \left(\frac{\vec{r}}{r} \right) = \frac{\vec{C}}{\mu} = \vec{e}$$

$$\vec{e} = \frac{\vec{v} \times \vec{h}}{\mu} - \frac{\vec{r}}{r}$$

$$e = |\vec{e}|$$

$$\vec{h} = \vec{r} \times \vec{v}$$

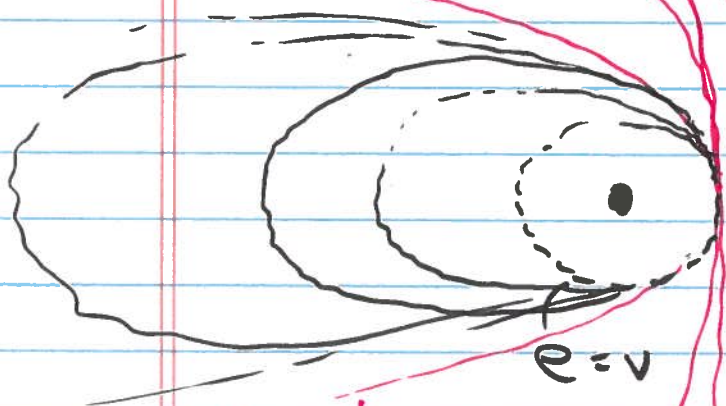
$$e \begin{cases} 0 < e < 1 \\ e = 1 \\ e > 1 \end{cases}$$

Ellipse

$e = 0$ circular

parabola

hyperbola



$e = 1$

$e > 1$

produce predictive relation

$$\frac{\vec{v} \times \vec{h}}{\mu} - \frac{\vec{r}}{r} = \vec{e}$$

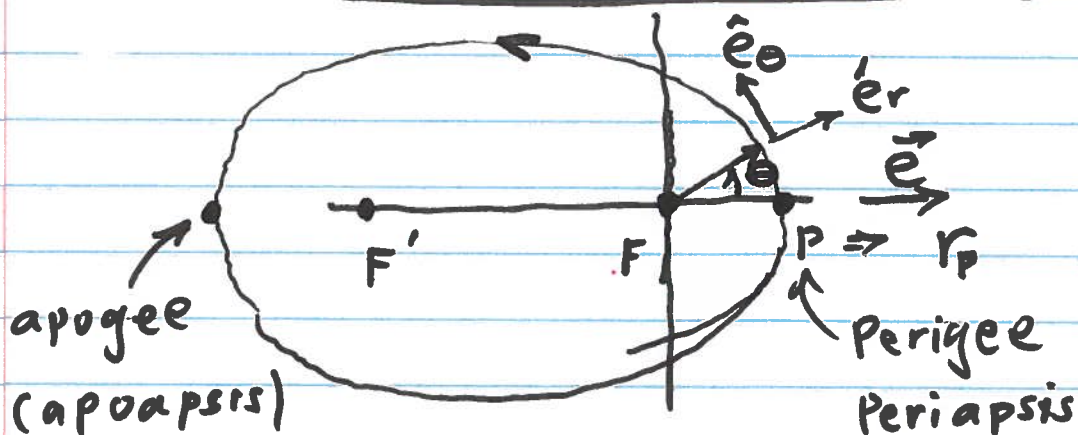
$$\frac{\vec{r} \cdot (\vec{v} \times \vec{h})}{\mu} - \underbrace{\frac{\vec{r} \cdot \vec{r}}{r}}_r = \underbrace{\vec{r} \cdot \vec{e}}_{r \cos \theta} \quad \swarrow \text{Actual}$$

$$\underbrace{\vec{r} \cdot (\vec{v} \times \vec{h})}_{h^2} = \vec{v} \cdot (\vec{h} \times \vec{r}) = \vec{h} \cdot (\underbrace{\vec{r} \times \vec{v}}_{\vec{h}}) = h^2$$

$$\vec{A} \cdot (\vec{B} \times \vec{C}) = \vec{B} \cdot (\vec{C} \times \vec{A}) = \vec{C} \cdot (\vec{A} \times \vec{B})$$

$$\Rightarrow \frac{h^2}{\mu} - r = r \cos \theta$$

$$\Rightarrow r(\theta) = \frac{h^2}{\mu(1 + e \cos \theta)}$$



$$v_r = \dot{r}$$

$$r > 0$$

$$\text{outbound } \dot{r} > 0$$

$$\vec{v} \cdot \vec{r} > 0 \Rightarrow \text{outbound}$$

$$\text{inbound } \dot{r} < 0$$

$$\Rightarrow \vec{v} \cdot \vec{r} < 0 \Rightarrow \text{Arccos wrong}$$

velocity

$$\vec{r} = r \hat{e}_r$$

$$r(\theta) = \frac{h^2}{\mu(1 + e \cos \theta)}$$

$$\vec{v} = \underbrace{\dot{r} \hat{e}_r}_{v_r} + \underbrace{r \dot{\theta} \hat{e}_\theta}_{v_\theta \text{ } (v_\perp)}$$

$$v_\theta = r \dot{\theta}$$

from before $h = r^2 \dot{\theta}$

$$\dot{\theta} = \frac{h}{r^2}$$

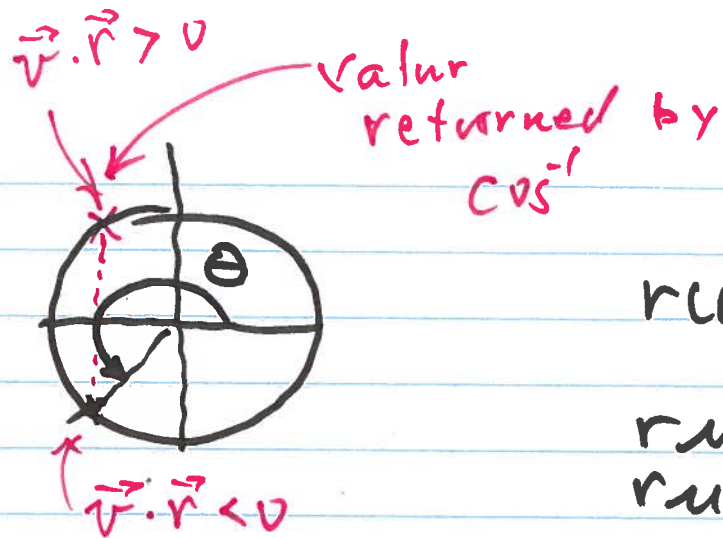
$$v_\theta = r \left(\frac{h}{r^2} \right) = \frac{h}{r}$$

$$v_\theta = \frac{\mu}{h} (1 + e \cos \theta) \quad \leftarrow \frac{h^2}{\mu(1 + e \cos \theta)}$$

$$v_r = \dot{r} = \frac{d}{dt} \left\{ \frac{h^2}{\mu(1 + e \cos \theta)} \right\}$$

$$v_r = \frac{\mu}{h} e \sin \theta$$

$$\vec{v} \cdot \vec{r} = (v_r \hat{e}_r + v_\theta \hat{e}_\theta) \cdot r \hat{e}_r = v_r r$$



$$r(\theta) = \frac{h^2}{\mu(1 + e \cos \theta)}$$

$$r\mu(1 + e \cos \theta) = h^2$$

$$r\mu + r\mu e \cos \theta = h^2$$

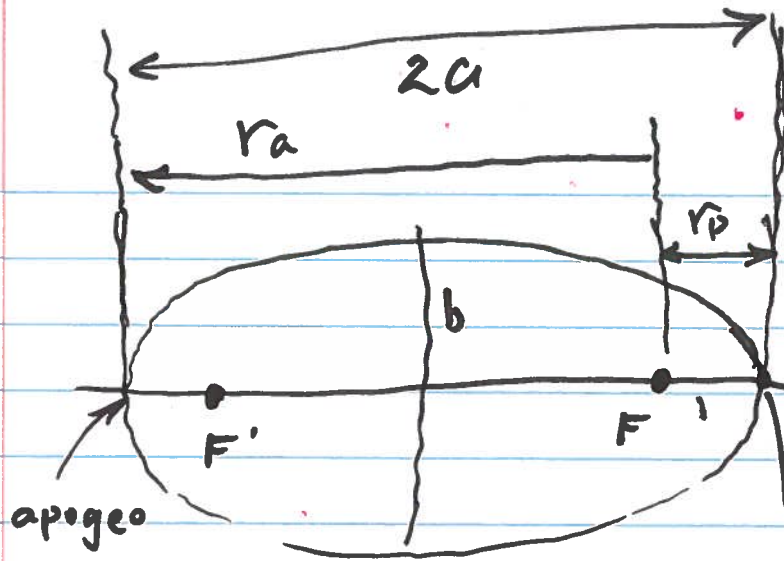
$$e = -\frac{h^2}{r\mu} \cos^{-1} \left[\frac{h^2 - r\mu}{r\mu} \right]$$



$$0 < \theta < \pi \text{ (180)}$$

wrong 50% of time.

$\theta_{\text{correct}} = 360 - \theta_{\text{acos}}$
when on return leg.



$$r(\theta) = \frac{h^2}{\mu(1 + e \cos \theta)}$$

$$r_p = \frac{h^2}{\mu(1 + e)}$$

$$r_a = \frac{h^2}{\mu(1 - e)}$$

$$a = \frac{r_p + r_a}{2} \Rightarrow r_p = a(1 - e)$$

$$r_a = a(1 + e)$$

$$e = \frac{r_a - r_p}{r_a + r_p}$$

$$\Rightarrow h^2 = a\mu(1 - e^2)$$

$$r(\theta) = a \frac{1 - e^2}{1 + e \cos \theta} = \frac{r_p(1 + e)}{1 + e \cos \theta}$$

$$\mathcal{E} = \frac{v^2}{2} - \frac{\mu}{r}$$

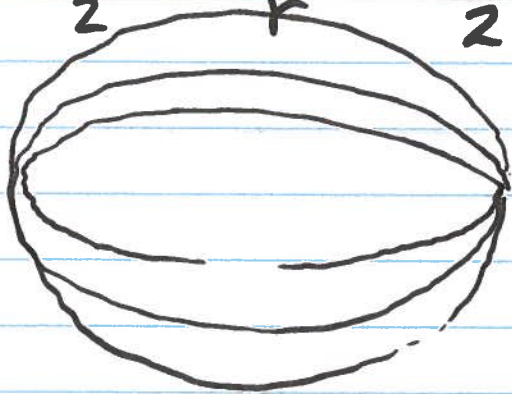
Evaluate @ r_p ($\theta = 0$)

$$v^2 = v_{\theta}^2 = \left[\frac{\mu}{h} (1 + e \cos^2 \theta) \right]^2$$

$$r(\theta) = r_p = \frac{h^2}{\mu(1 + e)}$$

& use $h^2 = a\mu(1 - e^2)$

$$\Rightarrow \epsilon = \frac{v^2}{2} - \frac{\mu}{r} = -\frac{\mu}{2a} = \begin{cases} < 0 & \text{Elliptical} \\ 0 & \text{Parabolic} \\ > 0 & \text{Hyperbolic} \end{cases}$$



$$\Rightarrow \epsilon = \frac{v^2}{2} - \frac{\mu}{r} = -\frac{\mu}{2a} \quad \left\{ \begin{array}{l} \text{Ellipse} \\ \text{or} \\ \text{parabola} \end{array} \right.$$

$$\epsilon = \frac{v^2}{2} - \frac{\mu}{r} = \frac{\mu}{2a} \quad \left\{ \begin{array}{l} \text{Hyperbola} \end{array} \right.$$

circular orbits $e=0$

$$\vec{f}_g = \vec{a}$$

$$\vec{f}_g = \frac{-\mu \vec{r}}{r^3} = -\frac{\mu \hat{e}_r}{r^2}$$

$$\vec{a} = (\ddot{r} - r\dot{\theta}^2) \hat{e}_r + (r\ddot{\theta} + 2\dot{r}\dot{\theta}) \hat{e}_\theta$$

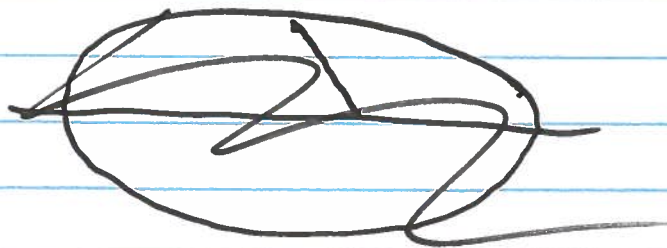
$$(\overset{0}{\ddot{r}} - r\dot{\theta}^2) \hat{e}_r + (\underbrace{r\ddot{\theta}}_{=0} + 2\overset{0}{\dot{r}}\dot{\theta}) \hat{e}_\theta = -\frac{\mu}{r^2} \hat{e}_r$$

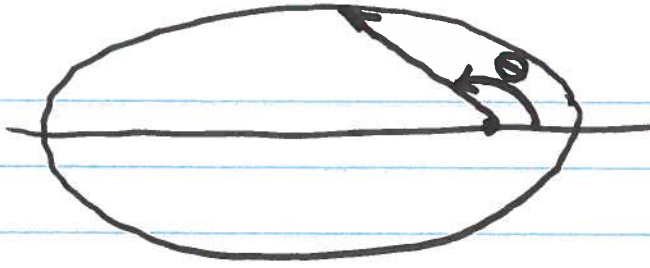
$\ddot{\theta} = 0 \Rightarrow \dot{\theta} = \text{constant}$

$$\dot{\theta}^2 r = \frac{\mu}{r^2} \quad \dot{\theta} = \sqrt{\frac{\mu}{r^3}}$$

Orbit period $T = \frac{2\pi}{\dot{\theta}}$

$$T = 2\pi \sqrt{\frac{r^3}{\mu}}$$





Elliptic Area of Ellipse = πab

$$\frac{dA}{dt} = \frac{h}{2}$$

$$dA = \frac{h}{2} dt$$

$$A = \int dA = \frac{h}{2} \int_0^T dt$$

$$A = \pi ab = \frac{h}{2} T$$

$$T_{\text{Ellip}} = \frac{2\pi ab}{h} = \frac{2\pi a^{3/2}}{\sqrt{\mu}}$$