

Lecture 06 – Rotation

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Outline



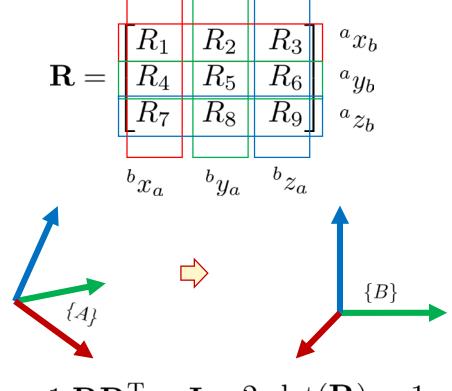
- About rotation
 - Lie group SO(3)
 - Lie algebra so(3)
 - Quaternion
 - Euler angles
 - Parameter perturbations (Update the variable with a small value)



Rotation matrix



 The rotation matrix consists of three directions of axes transformed into the target frame.



$$1.\mathbf{R}\mathbf{R}^{\mathrm{T}} = \mathbf{I} \quad 2.\det(\mathbf{R}) = 1$$



Rotation matrix



$$SO(n) = {\mathbf{M} \in \mathbb{R}^{n \times n} | \mathbf{M} \mathbf{M}^{\mathrm{T}} = \mathbf{I}, \det(\mathbf{M}) = 1}$$

A 3x3 rotation matrix is in the SO(3) group:

$$\mathbf{R} \in SO(3)$$

What is a group?

Group



- A group G is a set with a binary operation \circ defined on the elements of G, if it satisfies :
 - Closure :

$$g_1 \circ g_2 \in G$$

Identity:

$$e \circ g = g \circ e = g$$

Inverse:

$$g \circ g^{-1} = g^{-1} \circ g = e$$

Associativity (结合律)

$$g_1 \circ (g_2 \circ g_3) = (g_1 \circ g_2) \circ g_3$$

Group



- Examples of a group
 - Real number + Multiplication
 - Identity element : 1
 - Inverse : $x \leftrightarrow 1/x$
 - Associativity : $(x \times y) \times z = x \times (y \times z)$
 - Real number + Addition
 - Identity element: 0
 - Inverse: $x \leftrightarrow -x$
 - Associativity : (x+y)+z=x+(y+z)
 - Matrices + Matrix multiplication
 - Identity element : I
 - Inverse : $\mathbf{A} \leftrightarrow \mathbf{A}^{-1}$
 - Associativity : (AB)C = A(BC)

Lie group - SO(3)

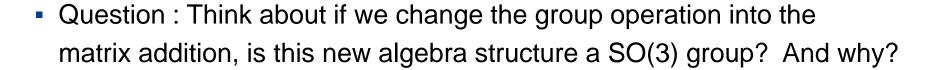
 The elements of SO(3) are the rotation matrices. The group operation of SO(3) is the matrix multiplication.

$$SO(3) = {\mathbf{R} | \mathbf{R} \mathbf{R}^{\mathrm{T}} = \mathbf{I} \det(\mathbf{R}) = 1}$$

- We can verify the following axioms:
 - Closure : $(\mathbf{R}_1\mathbf{R}_2)(\mathbf{R}_1\mathbf{R}_2)^{\mathrm{T}} = \mathbf{I}$ $\det(\mathbf{R}_1\mathbf{R}_2) = \det(\mathbf{R}_1)\det(\mathbf{R}_2) = 1$ $\mathbf{R}_1\mathbf{R}_2 \in SO(3)$
 - Identity: $I \cdot R = R \cdot I = R$
 - Inverse : $\mathbf{R} \cdot \mathbf{R}^{-1} = \mathbf{R}^{-1} \cdot \mathbf{R} = \mathbf{I}$
 - Associativity: $(\mathbf{R}_1\mathbf{R}_2)\mathbf{R}_3 = \mathbf{R}_1(\mathbf{R}_2\mathbf{R}_3)$



Lie group - SO(3)



$$AB \rightarrow A + B$$

• About 'special': $det(\mathbf{R}) = 1$

$$\rightarrow \det(\mathbf{R}) = \mathbf{r}_1^{\mathrm{T}}(\mathbf{r}_2 \times \mathbf{r}_3) = 1$$

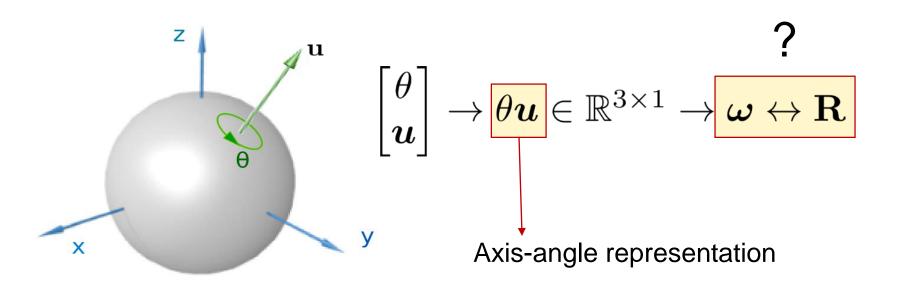
$$\rightarrow$$
 $r_1 = r_2 \times r_3$

→ Right-handed coordinate frame

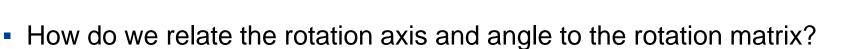


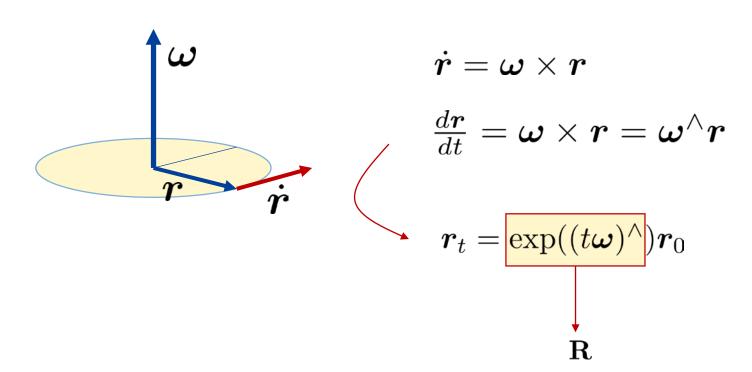
Lie algebra - so(3)

• Intuitively, a rotation can be described as a spinning action about an axis $\mathbf{u}, (\|\mathbf{u}\| = 1)$ with some amount of angle θ .



Exponential map





Here $m{x}^\wedge = [m{x}]_ imes$ is a 3x3 skew symmetric matrix, $(m{x}^\wedge)^\mathrm{T} = m{x}^\wedge$



Exponential map



• We can use an exponential map to map a skew symmetric matrix x^\wedge to a rotation matrix :

$$\mathbf{R} = \exp(\boldsymbol{x}^{\wedge})$$

• Here $\exp: so(3) \rightarrow SO(3)$

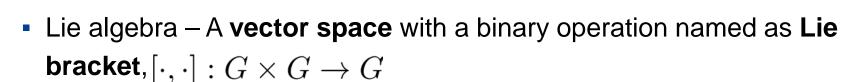
$$oldsymbol{x} \stackrel{(\cdot)^{\wedge}}{-\!\!\!\!-\!\!\!\!-} [oldsymbol{x}]_{ imes} \stackrel{\exp(\cdot)}{-\!\!\!\!\!-} {f R}$$

• Note that $x^{\wedge} \in so(3)$ - lie algebra for 3x3 skew symmetric matrices.

$$so(n) = {\mathbf{S} \in \mathbb{R}^{n \times n} | \mathbf{S}^{\mathrm{T}} = -\mathbf{S}}$$



Lie algebra – so(3)



• Bilinearity : given two scalars $\,a,b\,$

$$[a\mathbf{x} + b\mathbf{y}, \mathbf{z}] = a[\mathbf{x}, \mathbf{z}] + b[\mathbf{y}, \mathbf{z}]$$
$$[\mathbf{z}, a\mathbf{x} + b\mathbf{y}] = a[\mathbf{z}, \mathbf{x}] + b[\mathbf{z}, \mathbf{y}]$$

Alternating:

$$[\boldsymbol{x}, \boldsymbol{x}] = 0$$

Jacobi Identity

$$[x, [y, z]] + [z, [x, y]] + [y, [z, x]] = 0$$

Anti-commutativity

$$[oldsymbol{x},oldsymbol{y}]=-[oldsymbol{y},oldsymbol{x}]$$

Lie algebra – so(3)



$$[oldsymbol{x},oldsymbol{y}]=oldsymbol{x} imesoldsymbol{y}$$

ullet Example 2. 3x3 skew-symmetric matrix : $\,G = \{\Phi = {m x}^\wedge \}\,$

$$(oldsymbol{x}^\wedge)^{\mathrm{T}} = oldsymbol{x}^\wedge$$

$$m{x}^\wedge = [m{x}]_ imes = egin{bmatrix} x_1 \ x_2 \ x_3 \end{bmatrix}^\wedge = egin{bmatrix} 0 & -x_3 & x_2 \ x_3 & 0 & -x_1 \ -x_2 & x_1 & 0 \end{bmatrix}$$

with the bracket operation defined as

$$[\Phi_1, \Phi_2] = \Phi_1 \Phi_2 - \Phi_2 \Phi_1 \quad (\Phi_1 = \phi_1^{\wedge}, \Phi_2 = \phi_2^{\wedge})$$

Rodrigues' formula



• Given an axis-angle vector ω , we have its rotation matrix being

$$\mathbf{R} = \exp(\boldsymbol{\omega}^{\wedge}) = \exp(\theta[\boldsymbol{u}]_{\times})$$

By Taylor expansion, we have

$$\exp(\theta[\boldsymbol{u}]_{\times}) = \mathbf{I} + \theta[\boldsymbol{u}]_{\times} + \frac{\theta^{2}[\boldsymbol{u}]_{\times}^{2}}{2!} + \frac{\theta^{3}[\boldsymbol{u}]_{\times}^{3}}{3!} + \cdots$$

$$egin{aligned} [oldsymbol{u}]_ imes^2 &= oldsymbol{u} oldsymbol{u}^\mathrm{T} - oldsymbol{\mathrm{I}} \ [oldsymbol{u}]_ imes^3 &= - [oldsymbol{u}]_ imes \end{aligned}$$

$$[\boldsymbol{u}]_{ imes}^3 = -[\boldsymbol{u}]_{ imes}$$

$$\left[\mathbf{u}\right]_{\times}^{4} = -\left[\mathbf{u}\right]_{\times}^{2} \qquad \left[\mathbf{u}\right]_{\times}^{5} = \left[\mathbf{u}\right]_{\times}^{2} \qquad \left[\mathbf{u}\right]_{\times}^{6} = \left[\mathbf{u}\right]_{\times}^{2} \qquad \left[\mathbf{u}\right]_{\times}^{7} = -\left[\mathbf{u}\right]_{\times} \quad \cdots$$



Rodrigues' formula



$$\exp(\theta[\boldsymbol{u}]_{\times}) = \mathbf{I} + \frac{(\theta - \frac{\theta^3}{3!} + \frac{\theta^5}{5!} - \cdots)}{\sin(\theta)} [\boldsymbol{u}]_{\times} + \frac{(\frac{\theta^2}{2!} - \frac{\theta^4}{4!} + \frac{\theta^6}{6!} + \cdots)}{1 - \cos(\theta)} [\boldsymbol{u}]_{\times}^2$$

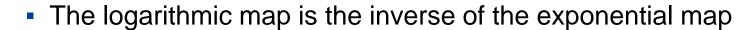
Finally we have the Rodrigues' formula

$$\mathbf{R} = \mathbf{I} + \sin(\theta)[\mathbf{u}]_{\times} + (1 - \cos(\theta))[\mathbf{u}]_{\times}^{2}$$

or

$$\mathbf{R} = \cos(\theta)\mathbf{I} + \sin(\theta)[\mathbf{u}]_{\times} + (1 - \cos(\theta))\mathbf{u}\mathbf{u}^{\mathrm{T}} \quad ([\mathbf{u}]_{\times}^{2} = \mathbf{u}\mathbf{u}^{\mathrm{T}} - \mathbf{I})$$

Logarithmic maps



$$\log : SO(3) \to so(3) \quad \log(\mathbf{R}) \to [\boldsymbol{u}\theta]_{\times}$$
$$\phi = \arccos\left(\frac{\operatorname{trace}(\mathbf{R}) - 1}{2}\right)$$
$$\mathbf{u} = \frac{(\mathbf{R} - \mathbf{R}^{\top})^{\vee}}{2\sin\phi},$$

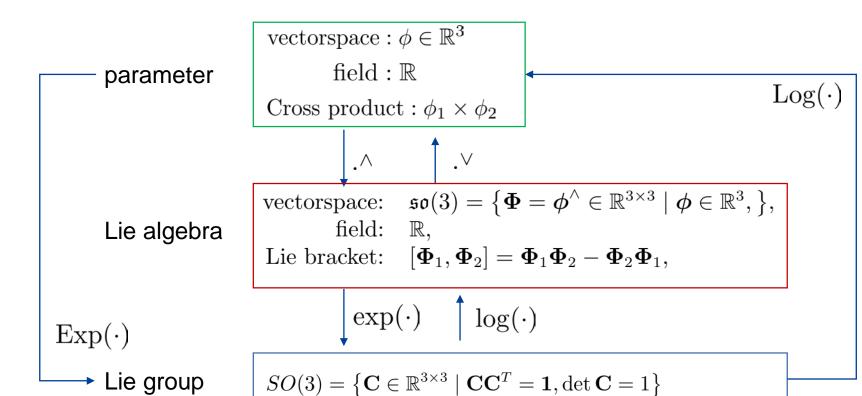
• Here, $(\cdot)^{\vee}$ is the inverse of $(\cdot)^{\wedge}$, namely $([x]_{\times})^{\vee} = x$

$$oldsymbol{x} \leftarrow \stackrel{(\cdot)^ee}{-} [oldsymbol{x}]_ imes \leftarrow \stackrel{\log(\cdot)}{-} \mathbf{R}$$



Conversion





Summary

- Group
- Lie group SO(3): $SO(3) = {\mathbf{R} \in \mathbb{R}^{3 \times 3} | \mathbf{R} \mathbf{R}^T = \mathbf{I}, \det(\mathbf{R}) = 1}$
- Lie algebra so(3) : $so(3) = \{ \Phi \in \mathbb{R}^{3 \times 3} \mid \Phi^{T} = -\Phi \}$
- Exponential map : $exp : so(3) \rightarrow SO(3)$
- Rodrigues formula : $\mathbf{R} = \cos(\theta)\mathbf{I} + \sin(\theta)[\mathbf{u}]_{\times} + (1 \cos(\theta))\mathbf{u}\mathbf{u}^{\mathrm{T}}$
- Logarithmic map : $\log : SO(3) \rightarrow so(3)$

$$egin{aligned} oldsymbol{x} & \xrightarrow{(\cdot)^{\wedge}} [oldsymbol{x}]_{ imes} & & \exp(\cdot) \\ oldsymbol{x} & \xrightarrow{(\cdot)^{ee}} [oldsymbol{x}]_{ imes} & & \log(\cdot) \\ oldsymbol{\mathbb{R}}^3 & & so(3) & & SO(3) \end{aligned}$$





- What is a quaternion?
 - A quaternion is a vector with four components:
 - A composite of a scalar and three different imaginary parts
- Mathematically, $\mathbf{q}=q_w+q_x\mathbf{i}+q_y\mathbf{j}+q_z\mathbf{k}$



Imaginary part

$$\mathbf{i}^2 = \mathbf{j}^2 = \mathbf{k}^2 = \mathbf{i}\mathbf{j}\mathbf{k} = -1$$

 $\mathbf{i}\mathbf{j} = \mathbf{k}, \mathbf{j}\mathbf{k} = \mathbf{i}, \mathbf{k}\mathbf{i} = \mathbf{j}$





- Quaternion an extension of a complex number
 - What we learn from a complex number also applies to a quaternion

• General quaternions :
$$\mathbf{q}=q_w+\mathbf{q}_v=egin{bmatrix}q_w\\\mathbf{q}_v\end{bmatrix}$$

• Real quaternions :
$$q_w = \begin{bmatrix} q_w \\ \mathbf{0}_v \end{bmatrix}$$

• Pure quaternions (3D vectors):
$$\mathbf{q}_v = \begin{bmatrix} 0 \\ \mathbf{q}_v \end{bmatrix}$$





The sum of two quaternion

$$\mathbf{p} + \mathbf{q} = \begin{bmatrix} p_w \\ p_x \\ p_y \\ p_z \end{bmatrix} + \begin{bmatrix} q_w \\ q_x \\ q_y \\ q_z \end{bmatrix} = \begin{bmatrix} p_w + q_w \\ p_x + q_x \\ p_y + q_y \\ p_z + q_z \end{bmatrix}$$

It is commutative and associative

$$\mathbf{p} + \mathbf{q} = \mathbf{q} + \mathbf{p}$$

 $\mathbf{p} + (\mathbf{q} + \mathbf{r}) = (\mathbf{p} + \mathbf{q}) + \mathbf{r}$



• Identity:
$$\mathbf{q_1} = 1 = \begin{bmatrix} 1 \\ \mathbf{0}_v \end{bmatrix}$$

• Conjugate :
$$\mathbf{q}* = \begin{bmatrix} q_w \\ -\mathbf{q}_v \end{bmatrix} \quad (\mathbf{p} \otimes \mathbf{q})^* = \mathbf{q}^* \otimes \mathbf{p}^*$$

• Norm:
$$\|\mathbf{q}\| \triangleq \sqrt{\mathbf{q} \otimes \mathbf{q}^*} = \sqrt{\mathbf{q}^* \otimes \mathbf{q}} = \sqrt{q_w^2 + q_x^2 + q_y^2 + q_z^2} \in \mathbb{R}$$

• Inverse:
$$\mathbf{q}^{-1} = \mathbf{q}^* / \|\mathbf{q}\|^2$$
 $\mathbf{q} \otimes \mathbf{q}^{-1} = \mathbf{q}^{-1} \otimes \mathbf{q} = \mathbf{q}_1$

• Dot product :
$$\mathbf{p}^T \mathbf{q} = p_w q_w + p_x q_x + p_y q_y + p_z q_z$$

= $\mathbf{p} \otimes \mathbf{q}^*$ or = $\mathbf{p}^* \otimes \mathbf{q}$



• Product of two quaternions $\mathbf{p} = p_w + p_x \mathbf{i} + p_y \mathbf{j} + p_z \mathbf{k}$ $\mathbf{q} = q_w + q_x \mathbf{i} + q_y \mathbf{j} + q_z \mathbf{k}$

$$\mathbf{p} \otimes \mathbf{q} = (p_w + p_x \mathbf{i} + p_y \mathbf{j} + p_z \mathbf{k})(q_w + q_x \mathbf{i} + q_y \mathbf{j} + q_z \mathbf{k})$$

$$\mathbf{p} \otimes \mathbf{q} = \begin{bmatrix} p_w q_w - p_x q_x - p_y q_y - p_z q_z \\ p_w q_x + p_x q_w + p_y q_z - p_z q_y \\ p_w q_y - p_x q_z + p_y q_w + p_z q_x \\ p_w q_z + p_x q_y - p_y q_x + p_z q_w \end{bmatrix}$$

- Not commutative: $\mathbf{p} \otimes \mathbf{q} \neq \mathbf{q} \otimes \mathbf{p}$
- Associative
- Distributive over sum
- Bi-linear



Left/right quaternion-product

$$[\mathbf{q}]_{L} = \begin{bmatrix} q_{w} & -q_{x} & -q_{y} & -q_{z} \\ q_{x} & q_{w} & -q_{z} & q_{y} \\ q_{y} & q_{z} & q_{w} & -q_{x} \\ q_{z} & -q_{y} & q_{x} & q_{w} \end{bmatrix}, \qquad [\mathbf{q}]_{R} = \begin{bmatrix} q_{w} & -q_{x} & -q_{y} & -q_{z} \\ q_{x} & q_{w} & q_{z} & -q_{y} \\ q_{y} & -q_{z} & q_{w} & q_{x} \\ q_{z} & q_{y} & -q_{x} & q_{w} \end{bmatrix}$$

• Or briefly $[\mathbf{q}]_L = q_w \mathbf{I} + \begin{bmatrix} 0 & -\mathbf{q}_v^{\mathsf{T}} \\ \mathbf{q}_v & [\mathbf{q}_v]_{\mathsf{X}} \end{bmatrix}$, $[\mathbf{q}]_R = q_w \mathbf{I} + \begin{bmatrix} 0 & -\mathbf{q}_v^{\mathsf{T}} \\ \mathbf{q}_v & -[\mathbf{q}_v]_{\mathsf{X}} \end{bmatrix}$ where $\mathbf{q}_v = [q_x, q_y, q_z]^T$

$$\mathbf{p} \otimes \mathbf{q} = [\mathbf{p}]_L \, \mathbf{q} = [\mathbf{q}]_R \, \mathbf{p}$$





$$[\mathbf{q}]_L[\mathbf{q}]_L^T = \mathbf{I}_{4\times 4} \qquad [\mathbf{p}]_R[\mathbf{p}]_R^T = \mathbf{I}_{4\times 4}$$

Dot product is preserved :

$$\mathbf{r}^{\mathrm{T}}\mathbf{t} = ([\mathbf{q}]_{L}\mathbf{r})^{\mathrm{T}}([\mathbf{q}]_{L}\mathbf{r}) = (\mathbf{q} \otimes \mathbf{r})^{\mathrm{T}}(\mathbf{q} \otimes \mathbf{t})$$

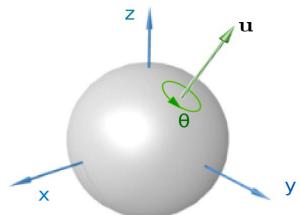
 $\mathbf{r}^{\mathrm{T}}\mathbf{t} = ([\mathbf{q}]_{R}\mathbf{r})^{\mathrm{T}}([\mathbf{q}]_{R}\mathbf{r}) = (\mathbf{r} \otimes \mathbf{q})^{\mathrm{T}}(\mathbf{t} \otimes \mathbf{q})$



Unit quaternion

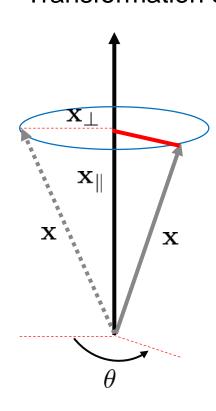
- For unit quaternion, we have $\|\mathbf{q}\| = 1$ and $\mathbf{q}^{-1} = \mathbf{q}^*$
- The unit quaternions also forms a group with respect to quaternion multiplication.
- For a rotation $\mathbf{R}=\exp([m{u} heta]^\wedge)$, its corresponding unit quaternion is given by

$$\mathbf{q} = \begin{bmatrix} \cos(\theta/2) \\ \mathbf{u}\sin(\theta/2) \end{bmatrix}$$



Vector rotation





$$\mathbf{x}_{||} = \mathbf{u} \, \mathbf{u}^{\top} \, \mathbf{x}$$

 $\mathbf{x}_{\perp} = \mathbf{x} - \mathbf{u} \, \mathbf{u}^{\top} \, \mathbf{x}$

$$\mathbf{x}'_{\perp} = \mathbf{x}_{\perp} \cos(\theta) + (\mathbf{u} \times \mathbf{x}) \sin(\theta)$$

$$\Rightarrow$$
 $\mathbf{x}' = \mathbf{x}'_{\perp} + \mathbf{x}_{\parallel}$

$$\Rightarrow$$
 $\mathbf{x}' = \mathbf{x}_{\perp} \cos(\theta) + (\mathbf{u} \times \mathbf{x}) \sin(\theta) + \mathbf{x}_{\parallel}$

$$\mathbf{x}' = (\cos(\theta)\mathbf{I} + \sin(\theta)[\mathbf{u}]_{\times} + (1 - \cos(\theta))\mathbf{u}\mathbf{u}^{\mathrm{T}})\mathbf{x}$$



Unit quaternion

Transformation of a vector

$$\mathbf{x}' = \mathbf{q} \otimes \mathbf{x} \otimes \mathbf{q}^*$$

$$\mathbf{q} = \begin{bmatrix} \cos(\theta/2) \\ \mathbf{u}\sin(\theta/2) \end{bmatrix}$$

$$\mathbf{x}' = \mathbf{x}_{\perp} \cos(\theta) + (\mathbf{u} \times \mathbf{x}) \sin(\theta) + \mathbf{x}_{\parallel}$$



Unit quaternion

Convert a unit quaternion to a rotation matrix

$$\mathbf{x}' = \mathbf{q} \otimes \mathbf{x} \otimes \mathbf{q}^*$$

$$\mathbf{x}' = \mathbf{R}\mathbf{x}$$

$$\mathbf{R}\{\mathbf{q}\} = (q_w^2 - \mathbf{q}_v^{\mathsf{T}} \mathbf{q}_v) \mathbf{I} + 2 \mathbf{q}_v \mathbf{q}_v^{\mathsf{T}} + 2 q_w [\mathbf{q}_v]_{\times}$$

Summary

- Quaternion : $\mathbf{i}^2 = \mathbf{j}^2 = \mathbf{k}^2 = \mathbf{i}\mathbf{j}\mathbf{k} = -1$
- Quaternion multiplication :

$$\mathbf{p} \otimes \mathbf{q} = (p_w + p_x \mathbf{i} + p_y \mathbf{j} + p_z \mathbf{k})(q_w + q_x \mathbf{i} + q_y \mathbf{j} + q_z \mathbf{k})$$

$$\mathbf{p} \otimes \mathbf{q} = [\mathbf{p}]_L \, \mathbf{q} = [\mathbf{q}]_R \, \mathbf{p}$$

- Unit quaternion : $\mathbf{q} = \begin{bmatrix} \cos(\theta/2) \\ \mathbf{u}\sin(\theta/2) \end{bmatrix}$
- Vector rotation : $\mathbf{x}' = \mathbf{x}_{\perp}' + \mathbf{x}_{\parallel}$
- Vector rotation by a unit quaternion : $\mathbf{x}' = \mathbf{q} \otimes \mathbf{x} \otimes \mathbf{q}^*$

$$\mathbf{x}' = \mathbf{x}_{\perp} \cos(\theta) + (\mathbf{u} \times \mathbf{x}) \sin(\theta) + \mathbf{x}_{\parallel}$$

Convert a quaternion to a rotation matrix

$$\mathbf{R}\{\mathbf{q}\} = (q_w^2 - \mathbf{q}_v^{\mathsf{T}} \mathbf{q}_v) \mathbf{I} + 2 \mathbf{q}_v \mathbf{q}_v^{\mathsf{T}} + 2 q_w [\mathbf{q}_v]_{\times}$$



Rotation matrix vs Quaternion



	Rotation matrix, \mathbf{R}	Quaternion, q
Parameters	$3 \times 3 = 9$	1 + 3 = 4
Degrees of freedom	3	3
Constraints	6	1
Constraints	$\mathbf{R}\mathbf{R}^{\top} = \mathbf{I} , \det(\mathbf{R}) = +1$	$\mathbf{q}\otimes\mathbf{q}^*=1$
Identity	I	1
Inverse	$\mathbf{R}^{ op}$	\mathbf{q}^*
Composition	${f R}_1{f R}_2$	$\mathbf{q}_1 \otimes \mathbf{q}_2$
Rotation operator	$\mathbf{R} = \mathbf{I} + \sin \phi \left[\mathbf{u} \right]_{\times} + (1 - \cos \phi) \left[\mathbf{u} \right]_{\times}^{2}$	$\mathbf{q} = \cos \phi / 2 + \mathbf{u} \sin \phi / 2$
Rotation action	$\mathbf{R}\mathbf{x}$	$\mathbf{q} \otimes \mathbf{x} \otimes \mathbf{q}^*$



Euler angle



Rotation about the x axis

$$\mathbf{R}(\phi) = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \phi & -\sin \phi \\ 0 & \sin \phi & \cos \phi \end{bmatrix}$$

Rotation about the y axis

$$\mathbf{R}(\theta) = \begin{bmatrix} \cos \theta & 0 & \sin \theta \\ 0 & 1 & 0 \\ -\sin \theta & 0 & \cos \theta \end{bmatrix}$$

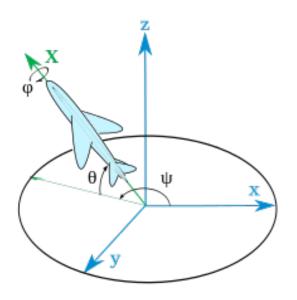
Rotation about the z axis

$$\mathbf{R}(\psi) = \begin{bmatrix} \cos \theta & -\sin \theta & 0\\ \sin \theta & \cos \theta & 0\\ 0 & 0 & 1 \end{bmatrix}$$

Euler angle



- Tait-Bryan convention (z-y-x): Yaw-pitch-roll
- Describes the pose of an aerial vehicle in the ENU coordinate system



$$\mathbf{R}_{wb} = \mathbf{R}_z(\psi)\mathbf{R}_y(\theta)\mathbf{R}_x(\phi)$$

Euler angle



A quaternion to a z-y-x Euler angle

$$\boldsymbol{q}_{wb} = \begin{bmatrix} \cos(\psi/2) \\ 0 \\ 0 \\ \sin(\psi/2) \end{bmatrix} \begin{bmatrix} \cos(\theta/2) \\ 0 \\ \sin(\theta/2) \\ 0 \end{bmatrix} \begin{bmatrix} \cos(\phi/2) \\ \sin(\phi/2) \\ 0 \\ 0 \end{bmatrix}$$

A z-y-x Euler angle to a quaternion

$$\begin{bmatrix} \phi \\ \theta \\ \psi \end{bmatrix} = \begin{bmatrix} \tan 2(2(q_w q_x + q_y q_z), 1 - 2(q_x^2 + q_y^2)) \\ \sin(2(q_w q_y - q_z q_x)) \\ \tan 2(2(q_w q_z + q_x q_y), 1 - 2(q_y^2 + q_z^2)) \end{bmatrix}$$



About parameter perturbations

 As we mentioned in the previous lecture, in an iterative solver, we need to update the variable repeatedly:

$$x \leftarrow x + \Delta x$$

 If the variable is not in a vector space, like a rotation, it cannot be updated by a simple vector addition :

$$\boldsymbol{x} \leftarrow \boldsymbol{x} + \Delta x$$

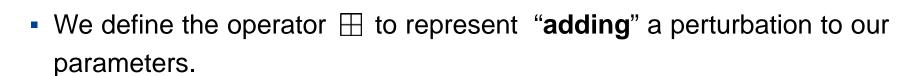
 Sometimes, the dimension of the parameter space of a rotation is not the same as the dimension of the parameters :

$$\boldsymbol{x} \in \mathbb{R}^m; \quad \Delta \boldsymbol{x} \in \mathbb{R}^n$$

$$m \neq n$$



Parameter perturbations



• Suppose the parameter space is denoted by \mathcal{X} and the perturbation space is \mathbb{R}^n . The operator is a mapping defined as :

$$\boxplus : \mathcal{X} \times \mathbb{R}^n \to \mathcal{X}$$



Parameter perturbations

• If the parameter space is a vector space, $\mathcal{X} = \mathbf{R}^n$,

$$oldsymbol{x} oxplus \Delta oldsymbol{x} = oldsymbol{x} + \Delta oldsymbol{x}$$

If the parameter is a Lie group (like rotation).

$$m{x} oxplus \Delta m{x} = m{x} \otimes \exp(\Delta m{x}^\wedge)$$
 (Right multiplication)
$$= \exp(\Delta m{x}^\wedge) \otimes m{x}$$
 (Left multiplication)

• Here $\boldsymbol{x} \in SO(3), \Delta \boldsymbol{x} \in \mathbb{R}^3$, we have

$$\boxplus : SO(3) \times \mathbb{R}^3 \to SO(3)$$



Rotation perturbations



$$x \boxplus \Delta x \leftrightarrow \mathbf{R} \cdot \delta \mathbf{R}$$

• Matrix exponential : $A \in \mathbb{R}^{n \times n}$

$$\exp(\mathbf{A}) = \mathbf{I} + \mathbf{A} + \frac{1}{2!}\mathbf{A}^2 + \frac{1}{3!}\mathbf{A}^3 + \dots$$

 The incremental rotation can be approximated from the perturbation vector by :

$$\delta \mathbf{R} = \exp([\Delta \boldsymbol{x}]_{\times}) = \mathbf{I} + [\Delta \boldsymbol{x}]_{\times} + [\Delta \boldsymbol{x}]_{\times} [\Delta \boldsymbol{x}]_{\times} + \dots$$

 $\approx \mathbf{I} + [\Delta \boldsymbol{x}]_{\times}$

$$oldsymbol{x} \boxplus \Delta oldsymbol{x} pprox \mathbf{R}(\mathbf{I} + [\Delta oldsymbol{x}]_{ imes})$$



Rotation perturbations

When the variable is represented by a unit quaternion:

$$\boldsymbol{x} \boxplus \Delta \boldsymbol{x} \leftrightarrow \boldsymbol{q} \otimes \delta \boldsymbol{q}$$

Definition of a unit quaternion:

$$oldsymbol{q} = egin{bmatrix} \cos(heta/2) \\ oldsymbol{u}\sin(heta/2) \end{bmatrix} \stackrel{ heta o 0}{\longrightarrow} oldsymbol{q} pprox egin{bmatrix} 1 \\ oldsymbol{u} heta/2 \end{bmatrix}$$

• Let the perturbation vector $\Delta oldsymbol{x} riangleq oldsymbol{u} heta$, we have

$$m{x} oxplus \Delta m{x} pprox m{q} \{m{x}\} \otimes egin{bmatrix} 1 \ \Delta m{x}/2 \end{bmatrix}$$



Jacobian matrix with respect to non-vector parameters

- How do we compute the Jacobian matrix if the parameter is not a vector?
- The change of the function value after a small perturbation is described by

$$\frac{\partial}{\partial \Delta x} h = \frac{h(x \boxplus \Delta x) - h(\mathbf{x})}{\Delta \mathbf{x}} \mid_{\Delta x \to 0}$$



Numerical differentiation



- The Jacobian matrix can be evaluated by numerical differentiation if the analytic approach is too complicated.
- At each time, the perturbation is only enabled in a single dimension.

$$\Delta \boldsymbol{x}^{(j)} = \left[0 \cdots \delta \cdots 0\right]^T \in \mathbb{R}^n$$

 δ is a small value: $\max(|10^{-4}x_i|, 10^{-6})$

$$\mathbf{H}(:,j) pprox rac{\mathbf{h}(\mathbf{x} \boxplus \Delta \mathbf{x}^{(j)}) - \mathbf{h}(\mathbf{x})}{\delta}$$



Summary



- Non-vector parameters, like rotations, cannot be simply added
- Perturbation operator $\boxplus : \mathcal{X} \times \mathbb{R}^n \to \mathcal{X}$.
- Rotation perturbation :

$$egin{aligned} oldsymbol{x} &\boxplus \Delta oldsymbol{x} \leftrightarrow \mathbf{R} \cdot \delta \mathbf{R} & oldsymbol{x} &\boxplus \Delta oldsymbol{x} pprox \mathbf{R} (\mathbf{I} + [\Delta oldsymbol{x}]_{ imes}) \ oldsymbol{x} &\boxplus \Delta oldsymbol{x} \leftrightarrow oldsymbol{q} \otimes \delta oldsymbol{q} & oldsymbol{x} &\boxplus \Delta oldsymbol{x} pprox oldsymbol{q} \{oldsymbol{x}\} \otimes egin{bmatrix} 1 \ \Delta oldsymbol{x}/2 \end{bmatrix} \end{aligned}$$

Jacobian matrix with respect to non-vector parameters:

$$\frac{\partial}{\partial \Delta x} h = \frac{h(x \boxplus \Delta x) - h(x)}{\Delta x} \mid_{\Delta x \to 0}$$

Numerical differentiation