# Asteroid Tidal Torque Progress Report

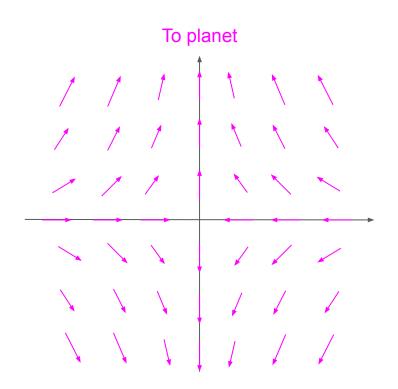
August 2, 2021

**Jack Dinsmore** 

equations of motion

Introduction &

## Goal: extract asteroid shape and density from flyby



Tidal force map

- Gravity induces not just tidal forces, but tidal torques
- Tidal torques are applied to all parts of the asteroid, not just the surface
- Light curve data can tell us the shape of the outside, so tidal torque will tell us the density on the inside

#### Task list

- Derive the asteroid equations of motion
- Simulate the asteroid flyby
- Fit an asteroid model to flyby rotation data
- Extract a density distribution from light curve and rotation data
- Analyze the strengths and weaknesses of the model and fit method

- Done
- In progress
- Not yet done

## The asteroid dynamics are defined by torque and MOI

#### Ingredients

Linear equation of motion

$$\dot{\mathbf{D}} = rac{\mu}{D^2}\hat{\mathbf{D}}$$

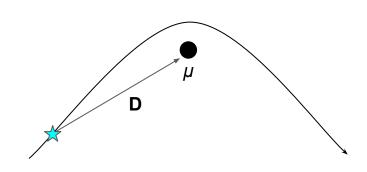
Angular equation of motion

$$\dot{\omega} = I^{-1}( au - \omega imes (I\omega))$$

- Torque
- Moment of inertia

#### **Initial Conditions**

- Asteroid starts on hyperbolic trajectory
- Initially spinning around low energy principal axis



# Torque and MOI can be expressed via $K_{lm}$

Decays like  $1/D^{l+l'}$ 

Torque is given by

$$egin{aligned} au &= -rac{1}{2}\sum_{lm}(-1)^l\mathcal{J}_{lm}\sum_{l'm'}S_{l+l',m+m'}(\mathbf{D})\ &igg[(i\hat{\mathbf{x}}+\hat{\mathbf{y}})(l'-m'+1)\mathcal{K}^*_{l',m'-1}\ &+(i\hat{\mathbf{x}}-\hat{\mathbf{y}})(l'+m'+1)\mathcal{K}^*_{l',m'+1}+2im'\hat{\mathbf{z}}\mathcal{K}^*_{l'm'}igg]. \end{aligned}$$

$$\mathcal{J}_{lm} \propto \int d^3r 
ho_{
m Planet}({f r}) R_{lm}({f r})$$

MOI is diagonal

$$egin{align} I_{xx} &= rac{2}{3}\mathcal{K}_{20} - 2\mathcal{K}_{2,-2} - 2\mathcal{K}_{22} + rac{2}{5}\mathcal{K}_{00} \ I_{yy} &= rac{2}{3}\mathcal{K}_{20} + 2\mathcal{K}_{2,-2} + 2\mathcal{K}_{22} + rac{2}{5}\mathcal{K}_{00} \ I_{zz} &= -rac{4}{3}\mathcal{K}_{20} + rac{2}{5}\mathcal{K}_{00} \ \end{array}$$

$$\mathcal{K}_{lm} \propto \int d^3r 
ho_{
m Asteroid}(\mathbf{r}) R_{lm}(\mathbf{r})$$

# Simulation

# Physical conditions restrict $K_{lm}$

MOI must be diagonal in the principal axis frame

$$\mathfrak{IK}_{22}=\mathcal{K}_{21}=0$$

The center of mass must be zero in the principal axis frame

$$\mathcal{K}_{1m}=0$$

There are only 3 low-order parameters:

$$\mathcal{K}_{00}, \quad \mathfrak{R}\mathcal{K}_{22}, \quad \mathcal{K}_{20}$$
 (Mass)

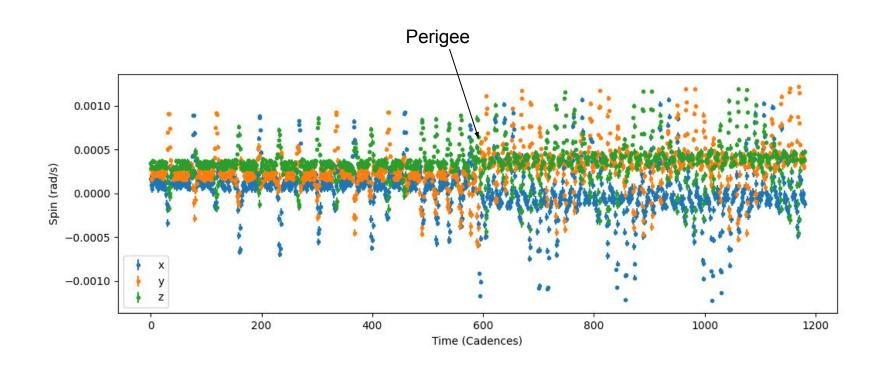
## Data collection is done by simulation

We integrate the previous equations of motion numerically (SuperCloud)

Adjustable parameters:

- $\bullet$   $K_{lr}$
- ullet  $J_{lm}$
- Orbit perigee and eccentricity
- Initial asteroid rotation speed and direction
- Initial orientation of the asteroid (one number, not three)
- Cadence of observation (e.g., 1 hour, 2 minutes)

## We are able to simulate the flyby for any parameters



# Fit method & results

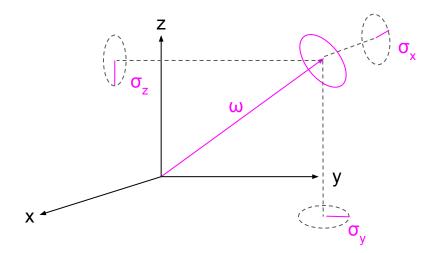
## Our fit method uses a Markov Chain Monte Carlo (MCMC)

- Want to extract original parameters from sample rotation-vector data.
- The fit is complicated and highly sensitive to parameters, so simple fit methods won't work
- We use an MCMC (Markov Chain Monte Carlo) to extract full posterior probability distributions from flat priors and a likelihood function

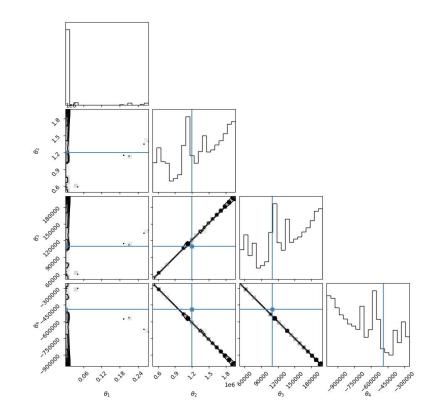
$$\ln \mathcal{L} \propto \sum_{ ext{data}} rac{(y_i - y_i^*)^2}{\sigma_i^2}$$

#### We use a semi-realistic error model

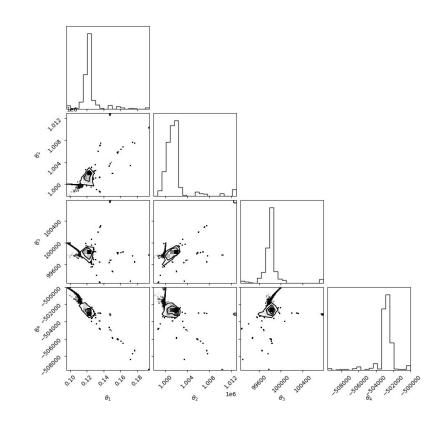
- Each rotational velocity component is shifted by a draw from a normal distribution
  - Its error is the standard deviation of that distribution.
- The distribution comes from a small change in direction to the spin vector.



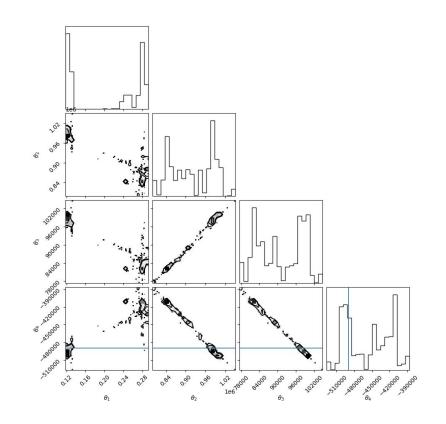
- 17 fits run for initial roll  $(\theta_1)$ ,  $K_{00}(\theta_2)$ ,  $K_{22}(\theta_3)$ ,  $K_{20}(\theta_4)$
- Converged after 200 cycles or so
- Results can be accurate



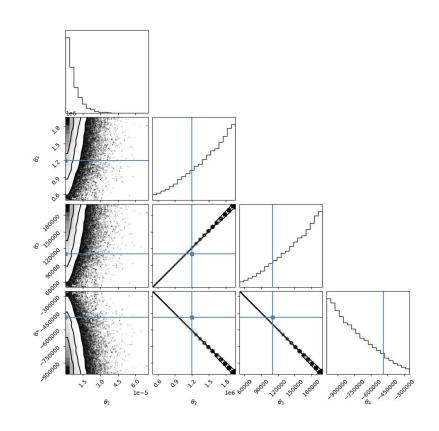
- 17 fits run for initial roll  $(\theta_1)$ ,  $K_{00}(\theta_2)$ ,  $K_{22}(\theta_3)$ ,  $K_{20}(\theta_4)$
- Converged after 200 cycles or so
- Results can be accurate
- Results can be inaccurate



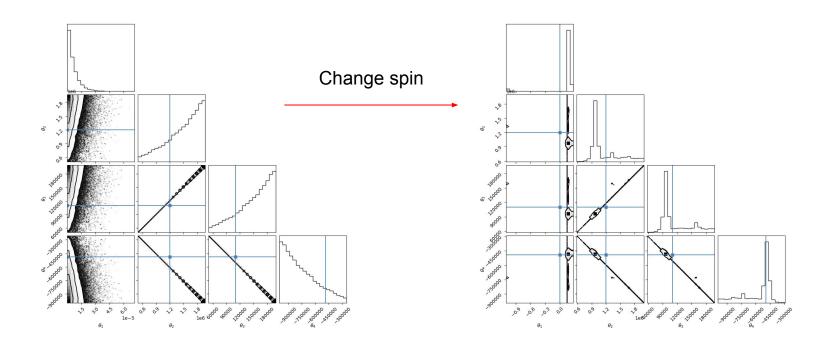
- 17 fits run for initial roll  $(\theta_1)$ ,  $K_{00}(\theta_2)$ ,  $K_{22}(\theta_3)$ ,  $K_{20}(\theta_4)$
- Converged after 200 cycles or so
- Results can be accurate
- Results can be inaccurate
- Degeneracy can be semi-broken



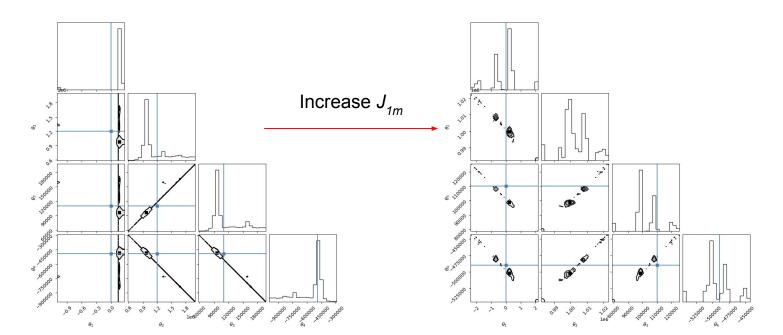
- 17 fits run for initial roll  $(\theta_1)$ ,  $K_{00}(\theta_2)$ ,  $K_{22}(\theta_3)$ ,  $K_{20}(\theta_4)$
- Converged after 200 cycles or so
- Results can be accurate
- Results can be inaccurate
- Degeneracy can be semi-broken
- Degeneracy can obscure true values



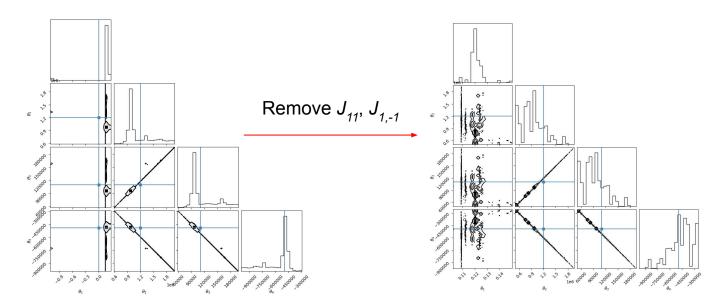
Degeneracy seems to be controlled by the choice of spin vector



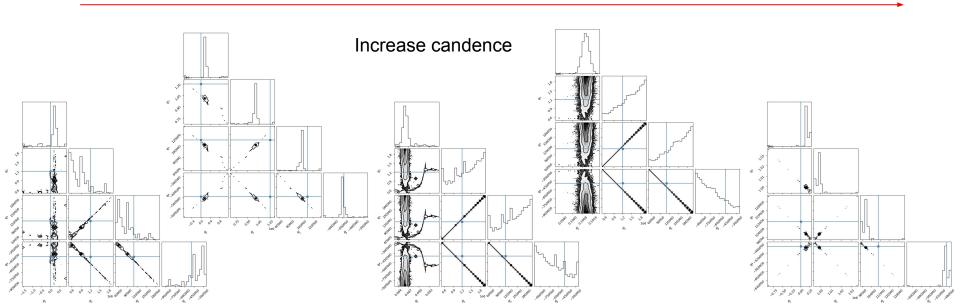
- Degeneracy seems to be controlled by the choice of spin vector
- Higher  $J_{1m}$  seems to increase accuracy



- Degeneracy seems to be controlled by the choice of spin vector
- Higher  $J_{1m}$  seems to increase accuracy
- Off diagonal  $J_{1m}$  may not help much



- Higher  $J_{1m}$  seems to increase accuracy
- Off diagonal  $J_{1m}$  may not help much
- Higher cadence doesn't necessarily help



- Off diagonal  $J_{1m}$  may not help much
- Higher cadence doesn't necessarily help
- b

# Density distribution

#### The density distribution model must yield physical results

#### Cannot

- predict negative density
- be spherically symmetric
- assume some radial density profile

#### Must

- Reproduce K<sub>Im</sub>
- Reproduce light curve data

- Assume that light curve data gives us a surface shape, and flyby spin data gives us K<sub>Im</sub>.
- What is the density distribution?

## Our density model consists of chunks of uniform density

- 1. Separate the asteroid into N chunks (N is the number of  $K_{lm}$ s known)
- 2. Calculate,  $K_{lmn}$ , the  $K_{lm}$  values for each chunk
- 3. Set up as matrix equation

known 
$$omega_{lm} = [\mathcal{K}_{lmn}] 
ho_n^{ ext{unknown}}$$

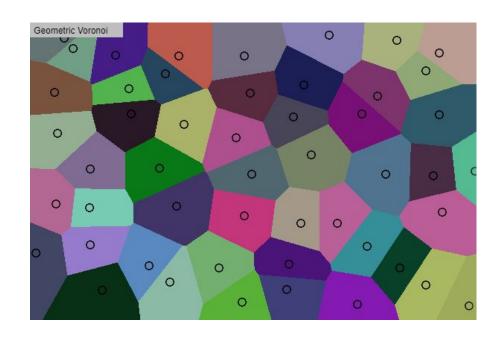
4. Solve matrix equation

$$ho_n = [\mathcal{K}_{lmn}]^{-1} \mathcal{K}_{lm}$$

#### How do you make the chunks? Voronoi cells

How do you choose the N sections?

- 1. Find a sphere that encloses the asteroid
- 2. Fill the sphere with N points
- 3. Repopulate all points outside the asteroid model until they're all inside
- Form Voronoi cells based on the points



#### Edges of Voronoi cells can be redrawn

Do we really want hard edges in random locations in our model?

- We can redo the density calculation with different Voronoi cells
- Then average together the density distributions because  $K_{lm}$  is linear in density

$$\mathcal{K}_{lm} \propto \int d^3r 
ho_{
m Asteroid}(\mathbf{r}) R_{lm}(\mathbf{r})$$

## Pros and Cons of this density model

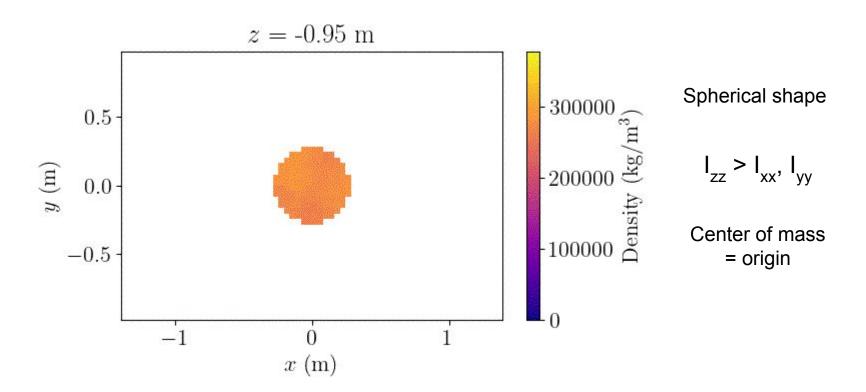
#### Pros:

- Massively parallelizable
- Works for any shape provided by light curve data
- Density calculation is nondegenerate
- Uncomplicated

#### Cons:

- A fully known shape model is needed
- Cannot yet account for uncertainties in K<sub>lm</sub> or shape
- Voronoi cell calculation is degenerate
- Average does not necessarily converge

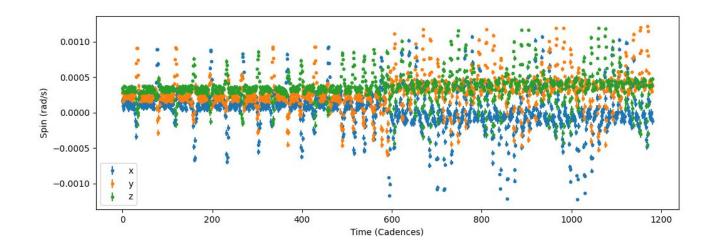
# We can generate a density model given $K_{lm}$ and a shape



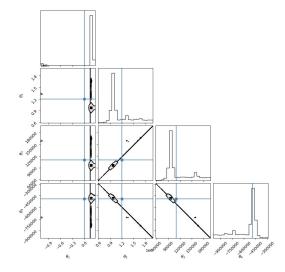
We have derived the equations of motion for an asteroid flyby

$$au = -rac{1}{2} \sum_{lm} (-1)^l \mathcal{J}_{lm} \sum_{l'm'} S_{l+l',m+m'}(\mathbf{D}) \qquad \qquad I_{xx} = rac{2}{3} \mathcal{K}_{20} - 2 \mathcal{K}_{2,-2} - 2 \mathcal{K}_{22} + rac{2}{5} \mathcal{K}_{00} \ \left[ (i\hat{\mathbf{x}} + \hat{\mathbf{y}})(l' - m' + 1) \mathcal{K}^*_{l',m'-1} \qquad \qquad I_{yy} = rac{2}{3} \mathcal{K}_{20} + 2 \mathcal{K}_{2,-2} + 2 \mathcal{K}_{22} + rac{2}{5} \mathcal{K}_{00} \ + (i\hat{\mathbf{x}} - \hat{\mathbf{y}})(l' + m' + 1) \mathcal{K}^*_{l',m'+1} + 2im'\hat{\mathbf{z}} \mathcal{K}^*_{l'm'} 
ight]. \qquad I_{zz} = -rac{4}{3} \mathcal{K}_{20} + rac{2}{5} \mathcal{K}_{00}$$

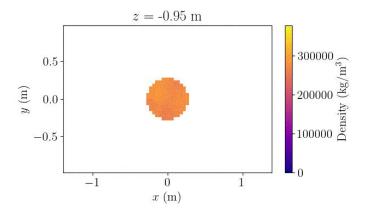
- We have derived the equations of motion for an asteroid flyby
- We have simulated the asteroid over the flyby



- We have derived the equations of motion for an asteroid flyby
- We have simulated the asteroid over the flyby
- We have fitted parameters to sample flyby data



- We have derived the equations of motion for an asteroid flyby
- We have simulated the asteroid over the flyby
- We have fitted parameters to sample flyby data
- We have extracted a density distribution from asteroid parameters



#### To do

- Play with fit parameters to analyze
  - What causes fits to be degenerate?
  - What causes fits to find incorrect values?
- Assess the success of fits in different scenarios
  - Precision on higher orders of  $K_{lm}$
  - $\circ$  Effect of increased cadence, different  $J_{lm}$ , higher perigee, etc.