

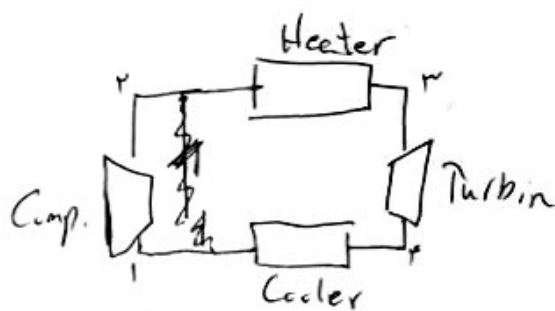
$$\frac{T_r - T_r}{T_r - T_{rs}} \Rightarrow T_r = T_r - \eta_T (T_r - T_{rs}) = T_{max} (1 - \eta_T (1 - \frac{1}{r}))$$

$$T_d = T_{max} (1 - \eta_T (1 - \frac{1}{r})) \quad \text{III}$$

ایجاد نوری معاد ① و ③ و ⑤ داریم:

$$\eta = \frac{W_T - W_C}{Q_{in}} = \frac{\eta_T \frac{T_{max}}{T_{min}} (1 - \frac{1}{r}) - \frac{r}{\eta_c} (r - 1)}{\frac{T_{max}}{T_{min}} + \eta_T (1 - \frac{1}{r}) \frac{T_{max}}{T_{min}} - (1 - \eta_T) (\frac{r-1}{\eta_c} + 1)} - \frac{T_{max}}{T_{min}} \eta_T (1 - \eta_T (1 - \frac{1}{r}))$$

ج) جگیزاری ϵ, r, η



② الف) $1800^\circ F = 1288.9 K$, $1000^\circ F = 594.9 K$

$$W_T = C_p (T_3 - T_4) = C_p \eta_T (T_3 - T_{rs})$$

$$= C_p \eta_T T_3 (1 - \frac{T_{rs}}{T_3}) = 294.7 \frac{kJ}{kg}$$

$$\frac{1}{1.0 \Delta} = \frac{1}{\Delta^{r/r}}$$

$Q_{in} = \frac{k-1}{12} = 1.17$

$C_p = 1.005 \frac{kJ}{kg \cdot K}$

$2000^\circ F$ ~~$\frac{k-1}{12} = 1.17$~~

~~$C_p = 1.179 \frac{kJ}{kg \cdot K}$~~

$$W_C = C_p (T_2 - T_1) = \frac{C_p (T_{rs} - T_1)}{\eta_c} = 248.8 \frac{kJ}{kg}$$

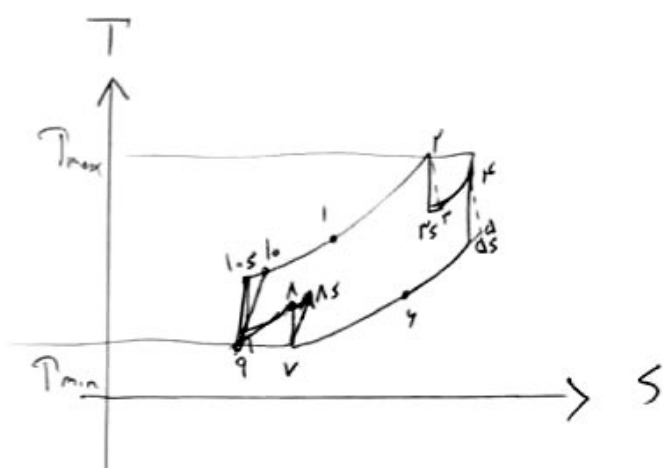
$$= \frac{T_1 C_p}{\eta_c} (\frac{T_{rs}}{T_1} - 1) = 248.8 \frac{kJ}{kg}$$

$$W_{net} = W_T - W_C = 118.1 \frac{kJ}{kg}$$

$$\Rightarrow \dot{W}_{net} = 118.1 \frac{kJ}{kg} \times \dot{m} \Rightarrow \dot{m} = \frac{248.8 \frac{kJ}{kg}}{118.1 \frac{kJ}{kg}} = 2.1 \frac{kg}{s}$$

1

(الف)



$$\begin{aligned}
 W_T &= m \cdot h_T (h_r - h_{rs}) + m \cdot h_T (h_f - h_{fs}) \\
 &= m \cdot h_T C_p (T_r - T_{rs}) + m \cdot h_T C_p (T_f - T_{fs}) = m \cdot h_T C_p T_r \left(1 - \frac{T_{rs}}{T_r}\right) \\
 &+ m \cdot h_T C_p T_f \left(1 - \frac{T_{fs}}{T_f}\right) = m \cdot h_T C_p T_r \left(1 - \left(\frac{P_{rs}}{P_r}\right)^{\frac{k-1}{k}}\right)^{\frac{1}{\gamma}} \\
 &+ m \cdot h_T C_p T_f \left(1 - \left(\frac{P_{fs}}{P_f}\right)^{\frac{k-1}{k}}\right)^{\frac{1}{\gamma}} = 2 m \cdot h_T C_p \left(1 - \frac{1}{\gamma}\right) T_{max}
 \end{aligned}$$

مثال ١٠

$$W_c = \frac{m \cdot C_p T_q}{h_c} \left(\frac{T_{hs}}{T_v} - 1\right) + \frac{m \cdot C_p T_q}{h_c} \left(\frac{T_{hs}}{T_a} - 1\right) = 2 \times \frac{m \cdot C_p T_{min}}{h_c} (\gamma - 1)$$

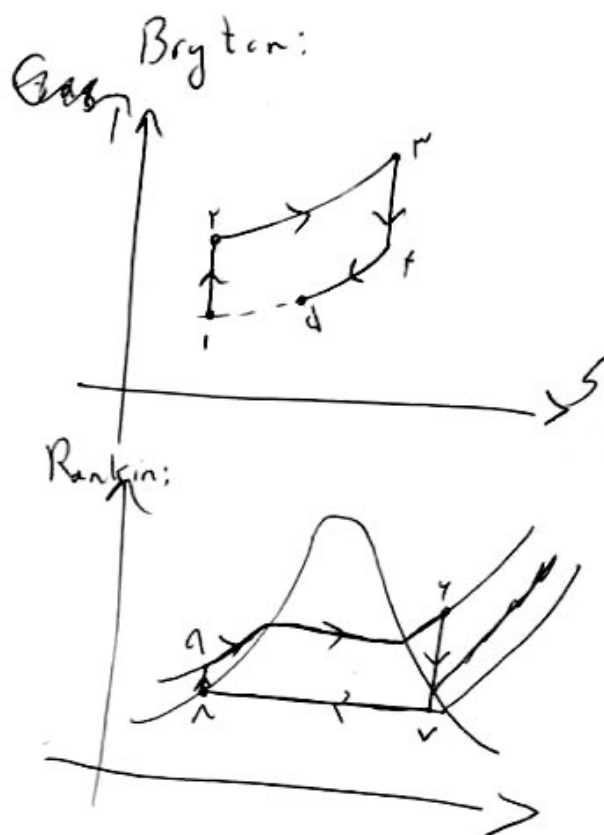
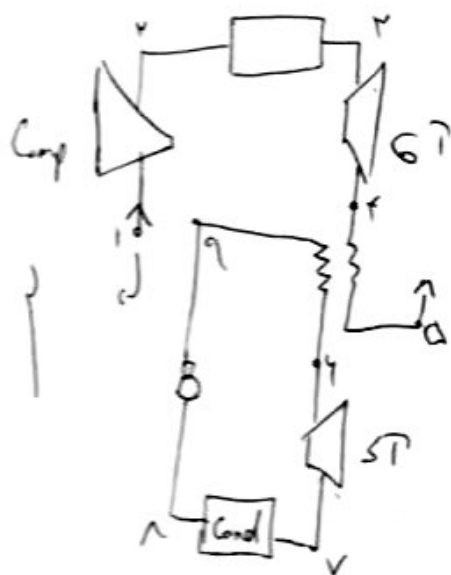
$$\Rightarrow Q_{in} = m \cdot C_p (\gamma T_{max} - T_{hs})$$

$$Reg: h_r = \frac{T_1 - T_b}{T_a - T_b} \Rightarrow T_1 = T_{10} + (T_a - T_{10}) \frac{h_r}{h_c} \quad \textcircled{I}$$

$$Q_{in} = m \cdot (h_r - h_1) + m \cdot (h_f - h_r) = m \cdot C_p (T_r T_1) + m \cdot C_p (T_f - T_r)$$

$$h_c = \frac{T_{hs} - T_a}{T_1 - T_a} \Rightarrow T_1 = \frac{T_{hs} - T_a}{h_c} + T_a \Rightarrow T_1 = T_{min} \left(\frac{\gamma - 1}{h_c} + 1\right) \quad \textcircled{II}$$

8-13: Combined Gas - ~~Electric~~ Steam



a) heat added per pound mass of air

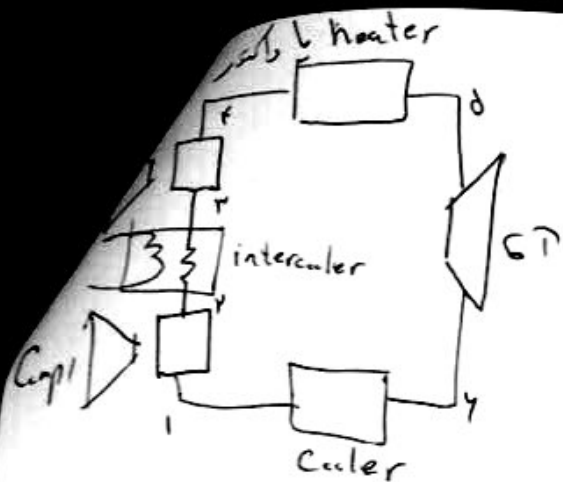
$$\textcircled{1} \begin{cases} P = 1 \text{ atm} \\ T = 540^\circ \text{R} \end{cases} \xrightarrow{\text{Table I-1}} \begin{cases} P_{r1} = 1.0149 \\ h_1 = 119.18 \frac{\text{Btu}}{\text{lbm Air}} \end{cases} \xrightarrow{\sim} P_{r2} = 4.987$$

$$\textcircled{2} \begin{cases} \text{Percent theoretical air} \\ T = 1400^\circ \text{R} \end{cases} \xrightarrow{\text{I-2}} \begin{cases} P_{r3} = 44.11 \\ h_3 = 1844.1 \frac{\text{Btu}}{\text{lbm Products}} \end{cases} \xrightarrow{\sim} P_{r4} = 17.11$$

$$\frac{1.0149 \text{ lbm Air}}{1.0149 \text{ lbm Products}}$$

$$1 \text{ lbm Air}$$

$$\text{Percent} = \frac{1.0149}{1.0149} \times 100 = 100\%$$



$$W_T = 398,14 \frac{\text{kJ}}{\text{kg}} \quad (\text{ج})$$

$$W_{C1} = W_{C2} \Rightarrow W_C = 2 \times C_p (T_r - T_1)$$

$$(\sqrt{2})^{1,4}$$

$$\frac{P_r}{P_1} \times \frac{P_r}{P_r} = d \Rightarrow \frac{P_r}{P_1} = \sqrt{2}$$

$$\eta_c = \frac{T_{r5} - T_1}{T_r - T_1} \Rightarrow T_{r5} = T_1 \left(\frac{T_{r5}}{T_1} - 1 \right) + 1 = 39,90^\circ \text{K} \Rightarrow W_C = 187,2 \frac{\text{kJ}}{\text{kg}}$$

$$T_f = T_r = 39,90^\circ \text{K}$$

$$W_{C2} = \frac{W_C}{\eta_c} = \frac{187,2}{0,8} = 234,0 \frac{\text{kJ}}{\text{kg}}$$

$$\frac{Q_{in}}{m} = C_p (T_d - T_f) = 144,11 \frac{\text{kJ}}{\text{kg}}$$

$$\eta = 27,2\%$$

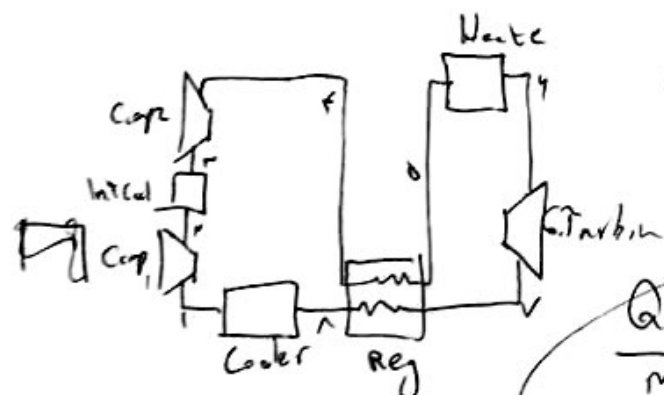
$$W_{net} = 211,9 \frac{\text{kJ}}{\text{kg}}$$

$$\Delta \text{cool} = 211,9 \text{V} \times m' \Rightarrow m' = 228,11 \frac{\text{kg}}{\text{s}}$$

$$m' = 228,11 \frac{\text{kg}}{\text{s}} \quad W_{net} = 211,9 \frac{\text{kJ}}{\text{kg}} \quad \text{ج} \Rightarrow W_{net} (m')$$

$$\begin{aligned} \text{توربین} &= 124981,2 \text{ hp} \\ \text{مضخة} &= 21492,4 \text{ hp} \end{aligned}$$

$$m' = 228,11 \frac{\text{kg}}{\text{s}} \quad W_{net} = 211,9 \frac{\text{kJ}}{\text{kg}} \quad \text{ج} \Rightarrow W_{net} (m')$$



$$\eta = 27,2\%$$

$$\epsilon_r = \frac{T_d - T_f}{T_r - T_f} = 1 \Rightarrow T_d = 144,11^\circ \text{K}$$

$$\frac{Q_{in}}{m'} = C_p (T_r - T_d) = 144,11 \frac{\text{kJ}}{\text{kg}}$$

$$W_T = C_p (T_r - T_v) \Rightarrow T_v = 144,11^\circ \text{K}$$

$$\eta = 27,2\%$$

ادامه ۲ الف

$$\frac{Q_{in}}{m} = C_p (T_r - T_r)$$

$$h_c = \frac{T_{rs} - T_1}{T_r - T_1} \Rightarrow T_r = \frac{T_{rs} - T_1}{h_c} + T_1 = 50.5, 175^{\circ}K$$

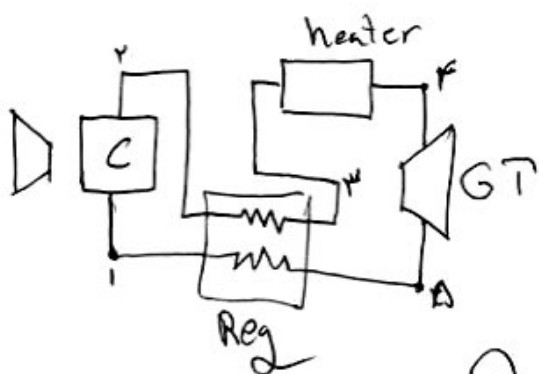
$$\frac{T_{rs}}{T_1} = 5^{rn}$$

$$\frac{Q_{in}}{m} = 1,00 f(1755, 175 - 50.5, 175) = 1752, 47 \frac{kJ}{kg}$$

$$\eta = \frac{W_T - W_C}{Q_{in}} = 25\%$$

قدرت تولید = $248, 81 \times 175, 175 = 14, 417, 8$ hp

تولید کمپرسور = $1752, 47$ hp



از دست قیل:

$$W_T = 1752, 47 \frac{kJ}{kg} \quad W_C = 1752, 47 \frac{kJ}{kg}$$

$$W_{net} = 1752, 47 \frac{kJ}{kg} \quad m' = 248, 81 \frac{kg}{s}$$

$$\frac{Q_{in}}{m} = C_p (T_r - T_r) = 1752, 47 \frac{kJ}{kg}$$

$$\epsilon_r = \frac{T_r - T_r}{T_r - T_r} = 1 \Rightarrow T_{rs} = 175, 175^{\circ}K$$

$$h_r = \frac{T_r - T_r}{T_r - T_{rs}} \Rightarrow T_r = T_r - h_r (T_r - T_{rs}) = T_r - h_r T_r (1 - \frac{1}{5})$$

$$= 175, 175^{\circ}K$$

$$\eta = 4\%$$

قدرت تولید کمپرسور و موتور

$$c) W_{GP} = h_r - h_f = 44 \text{ Btu/lb} - 44 \text{ Btu/lb} = 0 \text{ Btu/lb}$$

$$W_{SP} = h_g - h_v = 1044 \text{ Btu/lb} - 1044 \text{ Btu/lb} = 0 \text{ Btu/lb}$$

$$W_c = h_p - h_1 = 44 \text{ Btu/lb} = 44 \text{ Btu/lb}$$

$$W_{net} = 44 \text{ Btu/lb} + 44 \text{ Btu/lb} - 44 \text{ Btu/lb} = 44 \text{ Btu/lb}$$

$$d) \eta_{cc} = \frac{W_{net}}{Q} = \frac{44 \text{ Btu/lb}}{44 \text{ Btu/lb}} = 100\%$$

$$e) \eta_{BC} = \frac{W_{GP} - W_c}{Q} = 100\%$$

$$= 1144 \times \frac{1}{1.411} \times 1.017 = 844.1 \frac{\text{Btu}}{\text{lbm Air}}$$

$$q = h_r - h_r = 844.1 - 194 = 650.1 \frac{\text{Btu}}{\text{lbm air}}$$

b) the steam flow

$$\textcircled{d} \rightarrow T = 1000^\circ\text{R} \rightarrow h = \frac{1099.1 \text{ Btu}}{\text{lb.mol products gas}} = 844.1 \frac{\text{Btu}}{\text{lbm Air}}$$

$$\begin{aligned} \textcircled{y} & \left. \begin{array}{l} 1 \dots \text{psi} \\ 1 \dots 1000^\circ\text{F} \end{array} \right\} \Rightarrow \left\{ \begin{array}{l} v_y = 1.1990 \frac{\text{ft}^3}{\text{lbm}} \\ h_y = 180.4 \frac{\text{Btu}}{\text{lbm}} \\ s_y = 1.487 \frac{\text{Btu}}{\text{lbm} \cdot ^\circ\text{F}} \end{array} \right. \end{aligned}$$

$$\textcircled{A} \left. \begin{array}{l} 1 \text{ psi} \\ 1000^\circ\text{F} \end{array} \right\} \Rightarrow \left\{ \begin{array}{l} v_A = 1.191 \frac{\text{ft}^3}{\text{lbm}} \\ h_A = 49.1 \frac{\text{Btu}}{\text{lbm}} \\ s_A = 1.174 \frac{\text{Btu}}{\text{lbm} \cdot ^\circ\text{F}} \end{array} \right. = h_A \sim \frac{1}{2} w_p \approx 0$$

\textcircled{V}

$$\begin{aligned} P_v = P_A = 1 \text{ psi} & \left\{ \begin{array}{l} s_f = 1.174 \\ s_{fg} = 1.1748 \end{array} \right. \Rightarrow x = 17.5\% \Rightarrow \\ s_v = s_y = 1.487 & \quad h_v = 917.8 \end{aligned}$$

$$h_f - h_A = 49.1 - 180.4 = -131.3 \frac{\text{Btu}}{\text{lbm Air}}$$

$$h_y - h_A = 180.4 - 49.1 = 131.3 \frac{\text{Btu}}{\text{lbm steam}}$$

$$m_{\text{steam}} \times (h_y - h_A) = m_{\text{Air}} \times (h_f - h_A) \Rightarrow \frac{m_{\text{steam}}}{m_{\text{Air}}} = \left| \frac{131.3}{-131.3} \right| = 1.11$$