

$$n_m = 1440 \text{ rpm} < n_{sync} \Rightarrow n_{sync} = \frac{120 \times 50}{p} > 1440 \Rightarrow p = 4$$

$$n_{sync} = 1500 \text{ rpm}$$

$$s = \frac{n_{sync} - n_m}{n_{sync}} \times 100 \Rightarrow s = \frac{1500 - 1440}{1500} \times 100 = 4\%$$

$$f_{re} = s f_{sc} \Rightarrow f_{re} = 0.04 \times 25 = 1 \text{ Hz}$$

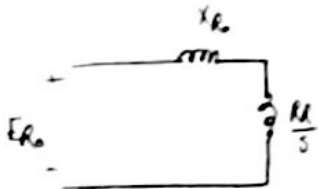
سرعت میدان استاتور نسبت به استاتور: $n_{sync} = \frac{120 \times 50}{4} = 1500 \text{ rpm}$

سرعت میدان استاتور نسبت به روتور: $n_{slip} = 1500 - 1440 = 60 \text{ rpm}$

سرعت میدان روتور نسبت به روتور: $1440 - 1500 = -60 \text{ rpm}$

استاتور " " " " : 1440 rpm

استاتور " " " " : صفر



$$n_{sync} = \frac{120 f_c}{p} = \frac{120 \times 50}{4} = 1500 \text{ rpm}$$

$$\omega_{sync} = \frac{4 \pi f_c}{p} = 157$$

$$s_{fl} = \frac{n_{sync} - n_m}{n_{sync}} = \frac{1500 - 1440}{1500} = 0.04$$

در حالت راه اندازی $s=1 \Rightarrow I_A = \frac{E_{\phi}}{jX_d + R} = 117.7 \angle -76.7^\circ$

بنابراین $\cos \phi = \cos(\angle E_{\phi} - \angle I_A) = \cos(76.7^\circ) \Rightarrow PF = 0.196$

$$T_{ind} = \frac{P_{AG}}{\omega_s} = \frac{3 E_{\phi} I_A \cos \phi}{\omega_{sync}} = \frac{3 \times 120 \times 117.7 \times 0.196}{157} = 52.9 \text{ (Nm)}$$

at full load: $s = 0.04 \quad I_A = \frac{E_{\phi}}{jX_d + \frac{R}{s}} = \frac{120 \angle 0^\circ}{j + \frac{0.2}{0.04}} = 23.5 \angle -11.3^\circ$

بنابراین $\cos \phi = \cos(\angle E_{\phi} - \angle I_A) = \cos(11.3^\circ) \Rightarrow PF = 0.981$

$$T_{ind} = \frac{P_{AG}}{\omega_{sync}} = \frac{3 E_{\phi} I_A \cos \phi}{\omega_{sync}} = 52.9 \text{ (Nm)}$$

تفاوت بین سرعت میدان روتور و سرعت میدان استاتور در موتور القای است. در موتور القای، سرعت میدان روتور کمتر از سرعت میدان استاتور است. این تفاوت باعث ایجاد گشتاور می‌شود. در موتور القای، سرعت میدان روتور را می‌توان با تغییر تعداد قطب‌ها یا تغییر فرکانس تغذیه تغییر داد.



2020-21

at no load: $S = 1$ $I_R = \frac{E_R}{jX_L + \frac{R_L}{S}} = \frac{120 \angle 0}{j16 + \frac{0.2}{1}} = 76.0 \angle -39.8^\circ$

$\cos \phi = \cos(45^\circ - 39.8^\circ) = \cos(5.2^\circ) \Rightarrow PF = 0.996$

$T_{ind} = \frac{P_{AG}}{w_s} = \frac{3 \times 76 \times 0.996}{w_s} = 135 \text{ Nm}$

at full load: $S = 0.04$ $I_R = \frac{E_R}{jX_L + \frac{R_L}{S}} = \frac{120 \angle 0}{j16 + \frac{0.2}{0.04}} = 19.7 \angle -9.4^\circ$

$\cos \phi = \cos(9.4^\circ) \Rightarrow PF = 0.987$ $T_{ind} = \frac{P_{AG}}{w_s} = 44.6 \text{ (Nm)}$

با این روش می توانیم توان تلفات را در بارهای مختلف و ضریب توان را در بارهای مختلف و ...

[3]

$P_{in} = P_{out} + P_{scL} + P_{core} + P_{rot} + P_{ACL}$

(الف)

از اول: $I_1 = \frac{I_L}{\sqrt{3}} = \frac{5}{\sqrt{3}} = 4.62 \text{ A} \Rightarrow P_{scL} = 3 I_1^2 R_1 = 76.8 \text{ W}$

$660 = 76.8 + P_{core} + 0 + 420 + 0 \Rightarrow P_{core} = 163.2 \text{ W}$

at full load: $P_{out} = 10 \text{ kW}$

$I_1 = \frac{I_L}{\sqrt{3}} = \frac{16}{\sqrt{3}} = 10.39 \text{ A} \Rightarrow P_{ACL} = 3 R_1 I_1^2 = 344.8 \text{ W}$

$P_{in} = P_{scL} + P_{core} + P_{rotor loss} + P_{out} \Rightarrow 11.2 \text{ kW} = 366.8 + 163.2 + P_{rotor} + 10 \text{ kW} \Rightarrow P_{rotor} = 646 \text{ W}$

$P_{rotor loss} = P_{rot} + P_{ACL} \Rightarrow 646 = P_{ACL} + 420 \Rightarrow P_{ACL} = 226 \text{ W}$

(2)

$P_{out} = \frac{1-S}{1} P_{ACL} \approx 10 \text{ kW} = \frac{1-1}{1} \cdot 226 \Rightarrow S = 2.2 \%$

(3)

$T_{ind} = \frac{P_{core}}{w_m} = \frac{P_{rot} + P_{ACL}}{w_m} = 7.11 \text{ (Nm)}$ $T_{load} = \frac{P_{out}}{w_p} = 6.8 \text{ (Nm)}$

(4)

$\eta = \frac{P_{out}}{P_{in}} \times 100 = 89.3 \%$

$$S_{max} = \frac{R_2'}{\sqrt{R_{th}^2 + (X_{th} + X_2)^2}} = \frac{R_2'}{X_{th} + X_2} \Rightarrow X_{th} + X_2 = 10 R_2'$$

4

(1)

$$\tau = \frac{3 V_{th}^2 \frac{R_2'}{s}}{\omega_s \sqrt{(R_{th} + \frac{R_2'}{s})^2 + (X_{th} + X_2)^2}} \xrightarrow{R_{th}=0} \frac{\tau_b}{\tau_{max}} = \frac{s_{max}}{s_b} \times \frac{(\frac{R_2'}{s_{max}})^2 + (X_{th} + X_2)^2}{(\frac{R_2'}{s_b})^2 + (X_{th} + X_2)^2}$$

$$\Rightarrow \frac{1}{2} = \frac{0.1 \left[\left(\frac{R_2'}{0.1} \right)^2 + (10 R_2')^2 \right]}{s_b \left[\left(\frac{R_2'}{s_b} \right)^2 + (10 R_2')^2 \right]} \Rightarrow s_b = 2.36 \%$$

$$\omega_{sync} = \frac{120 f_e}{p} = 1500 \text{ rpm} \quad n_m = (1 - s_b) n_{sync} = 464.6 \text{ (rpm)}$$

$$\left. \begin{aligned} P_{cu2} &= s_b P_{AG} \\ P_{rot=0} &\Rightarrow P_{out} = (1 - s_b) P_{AG} \end{aligned} \right\} \Rightarrow P_{mech} = \frac{s_b}{1 - s_b} P_{out} = \frac{0.0236}{1 - 0.0236} \times 10 \text{ kW} = 241.7 \text{ (W)} \quad (2)$$

$$\tau = \frac{3 V_{th}^2 \frac{R_2'}{s}}{\omega_{sync} [(R_{th} + \frac{R_2'}{s})^2 + (X_{th} + X_2)^2]} \xrightarrow{R_{th}=0} \frac{\tau_{start}}{\tau_b} = \frac{s_b \left[\left(\frac{R_2'}{s_b} \right)^2 + (X_{th} + X_2)^2 \right]}{[R_2']^2 + (X_{th} + X_2)^2} \quad (3)$$

$$= \frac{0.0236 \left[\left(\frac{R_2'}{0.0236} \right)^2 + (10 R_2')^2 \right]}{[R_2']^2 + (10 R_2')^2} = 0.442$$

$$\tau_a = \frac{P_{mech}}{\omega_m} = \frac{P_{out} + P_{rot}}{\omega_m} = \frac{10 \text{ kW}}{\frac{1964.6}{60} \times 2\pi} = 65.2 \text{ (Nm)}$$

$$\tau_{start} = 28.7 \text{ (Nm)}$$

$$I_s = I_R' = \frac{V_{th}}{\sqrt{(R_{th} + \frac{R_2'}{s})^2 + (X_{th} + X_2)^2}} \quad (R_{th}=0)$$

(4)

$$\frac{I_{start,R}}{I_{b,R}} = \frac{I_{start,R'}}{I_{b,R'}} = \frac{\sqrt{\left(\frac{R_2'}{s} \right)^2 + (X_{th} + X_2)^2}}{\sqrt{\left(\frac{R_2'}{s_b} \right)^2 + (X_{th} + X_2)^2}} = \frac{\sqrt{\left(\frac{R_2'}{0.0236} \right)^2 + (10 R_2')^2}}{\sqrt{(R_2')^2 + (10 R_2')^2}} = 4.33$$

$$\frac{I_{max,s}}{I_{b,s}} = \frac{\sqrt{\left(\frac{R'_2}{0.0230}\right)^2 + (10R'_1)^2}}{\sqrt{\left(\frac{R'_2}{0.2}\right)^2 + (10R'_2)^2}} = 3.07 \quad (5)$$

$$S_{max} = \frac{R'_2}{x_{th} + x'_2} = \frac{3R}{10R} \Rightarrow S_{max} = 2.07 \quad (7)$$

$$\frac{R'_2 / s_b}{\left(\frac{R'_2}{s_b}\right)^2 + (x_{th} + x'_2)^2} = \frac{\frac{R'}{s_b}}{\left(\frac{R}{s_b}\right)^2 + (x_{th} + x'_2)^2} \Rightarrow \frac{\frac{3R}{s_b}}{\left(\frac{3R}{s_b}\right)^2 + (10R)^2} = \frac{\frac{R}{0.0236}}{\left(\frac{R}{0.0236}\right)^2 + (10R)^2} \quad (8)$$

$$\Rightarrow s_b = 0.0703 \Rightarrow 1/s_b = 1.037 \quad n_m = (1 - s_b) n_{sync} = 1393.8 \text{ rpm} \Rightarrow \omega_m = \frac{n_m}{60} \times 2\pi = 146 \frac{\text{rad}}{\text{s}}$$

$$\frac{T_{start}}{T_b} = \frac{s_b \left[\left(\frac{R'_2}{s_b}\right)^2 + (x_{th} + x'_2)^2 \right]}{\left[(R'_2)^2 + (x_{th} + x'_2)^2 \right]} = \frac{0.0703 \left[\left(\frac{3R}{0.0703}\right)^2 + (10R)^2 \right]}{(3R)^2 + (10R)^2} = 1.23 \quad (9)$$

$$\Rightarrow T_{start} = 1.23 \times 65.2 = 80.3 \text{ (N.m)}$$

$$\frac{I_{start,R}}{I_{b,R}} = \frac{I'_{start,R}}{I'_{b,R}} = \frac{\sqrt{\left(\frac{3R}{0.0703}\right)^2 + (10R)^2}}{\sqrt{(3R)^2 + (10R)^2}} = 4.17 \quad (10)$$

$$V_{th} = 0.982$$

$$(2\phi) \rightarrow \frac{V_{th}}{Z_{th}} = X_1 + Y_m \Rightarrow Y_m = 191.756$$

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$$\frac{(420 - \frac{X_m}{X_1 + Y_m})^2}{R_c} = 289 \Rightarrow R_c = 575.37 \Omega$$

$$I_1 = \frac{420}{(\frac{R_1}{s} + jX_1) \parallel R_c \parallel (Y_m + jX_2 + Z_2)} = 7.228 \angle -27.54^\circ \Rightarrow |I_1| = 7.228 \text{ A}$$

$$I_2 = \frac{(\frac{Y_m}{X_2 + Y_m} V_{th})}{\frac{R_2}{s} + jX_2} = 5.667 \angle -3.39^\circ \Rightarrow |I_2| = 5.667 \text{ A}$$

$$\tau = \frac{3V_{th}^2 \frac{R_c}{s}}{\omega_s [(R_{th} + \frac{R_1}{s})^2 + (X_{th} + X_1)^2]} = 5.86$$

$$S_{max} = \frac{R_2}{\sqrt{R_{th}^2 + (X_{th} + X_2)^2}} = 0.1845$$

$$N_r = (1-s) N_s \Rightarrow N_r = 407.750 \text{ (rpm)}$$

$$T_{max} = \frac{3V_{th}^2}{\omega_s (R_{th} + \sqrt{R_{th}^2 + (X_{th} + X_2)^2})} = 316.05 \text{ (N-m)}$$

$$\omega \tau = \frac{3V_{th}^2 R_c s}{\omega_s R_c^2} = \frac{3V_{th}^2}{\omega_s R_c} s$$

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4 اگر روتور 190 rpm یعنی 0.9 برابر می باشد و V_{th} 0.9 برابر می شود و همچنین X_1 1.235 برابر می باشد و $(\frac{10}{9})^2$ برابر می باشد 1.235 برابر می باشد

$$N_{r2} = (1 - 1.235s) N_s$$

$$N_{r1} = (1 - s) N_s$$

4 روتور 190 rpm یعنی 0.9 برابر می باشد و V_{th} 0.9 برابر می باشد و همچنین X_1 1.235 برابر می باشد و $(\frac{10}{9})^2$ برابر می باشد 1.235 برابر می باشد

$$P_{cu,r} = s P_{Ag} = s \frac{R_2}{s} I_r^2 = R_2 \frac{V_{th}^2}{(\frac{R_2}{s})^2 + (X_2)^2}$$

4 در این جا V_{th} 0.9 و $(\frac{10}{9})^2$ برابر می باشد و X_2 0.9 برابر می باشد و $(\frac{10}{9})^2$ برابر می باشد 1.235 برابر می باشد

$$\text{no load: } X_1 + Y_m = \frac{\frac{400}{\sqrt{3}}}{7.5} = 30.792 \Omega$$

$$\text{locked rotor: } \begin{cases} R_1 + R_2 = \frac{\frac{150}{\sqrt{3}}}{3.5} \times 0.44 = 2.474 \\ X_1 + X_2 = \frac{150}{\sqrt{3} \times 3.5} \times 0.99 = 2.22 \Omega \\ X_1 = 2X_2 \end{cases} \Rightarrow X_1 = 1.481 \Omega, X_2 = 0.74 \Omega$$

$$X_m = 39.311 \Omega$$

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$$P_{out} = R_2 I_2^2 = R_2 \frac{V_{th}^2}{\left(\frac{R_2}{s}\right)^2 + X_2^2} = 62.472$$

$R_1 \approx R_2$ ✓ $I_1 = I_2$ ✓ X_m ✓ \Rightarrow $\frac{V_{th}}{s}$ ✓

$$960 = \frac{120 \text{ fe}}{p} \Rightarrow p = 6 \Rightarrow s = \frac{1000 - 960}{1000} = 0.04$$

$$P_{AG} = \frac{P_{out}}{s} = 1561.8 \Rightarrow P_{out} = P_{conv} = (1-s)P_{AG} = 1299.33 \text{ (w)}$$

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$$\eta = \frac{P_{out}}{w_m} = \frac{P_{conv}}{(1-s)w_{sync}} = \frac{P_{AG}}{w_{sync}} = 14.914$$

$$\eta = \frac{P_{out}}{P_{in}} \times 100 = \frac{P_{out}}{P_{out} + 2P_{out}} = 92.3\%$$

$$2R_s \parallel R_s = 0.6 \Rightarrow \frac{2R_s \times R_s}{3R_s} = 0.6 \Rightarrow R_1 = \frac{3 \times 0.6}{2} = 0.9 \Omega$$

$$\cos \theta = \frac{P_m}{3V_L I_L} = \frac{P_m}{\sqrt{3} V_L I_L} = \frac{5 \text{ kW}}{\sqrt{3} \times 210 \times 20} = 0.687$$

$$R_1 + R_2 = \frac{210}{\frac{20}{\sqrt{3}}} \cdot 0.687 = 12.5 \Rightarrow R_1 = 0.9 \Omega, R_2 = 11.6 \Omega$$

$$X_1 + X_2 = \frac{210}{\frac{20}{\sqrt{3}}} \times \sqrt{1 - (0.687)^2} = 13.2$$

$$T_{s(hi)} = \frac{3 V_{th}^2 \frac{R_2}{s}}{w_s \left[\underbrace{\left(R_{th} + \frac{R_2}{s}\right)^2}_{\approx R_1^2} + \underbrace{\left(X_{th} + X_2\right)^2}_{\approx X_1^2} \right]} = \frac{450^2 \cdot 3 \cdot 11.6}{50 \pi (12.5^2 + 13.2^2)} \approx 118.25$$

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