

Problem 1

Figure 1 illustrates (in cross-section) a spherical cryogenic experiment that is placed at the center of a cubical enclosure (a box).

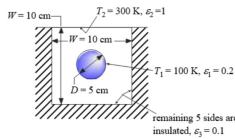


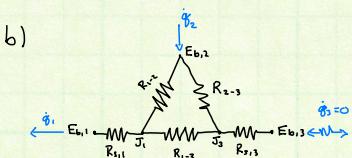
Figure 1: Spherical cryogenic experiment in a cubical enclosure

The spherical experiment (surface 1) has diameter $D = 5.0$ cm and emissivity $\epsilon_1 = 0.2$. The temperature of the experiment is maintained at $T_1 = 100$ K. Each face of the cubical enclosure is $W=10\text{cm} \times W=10\text{cm}$. The top surface of the enclosure (surface 2) is maintained at $T_2 = 300$ K, and is black. The other five sides (surface 3) are insulated externally and have emissivity $\epsilon_3 = 0.1$.

- a) If the view factor between the experiment and the top surface $F_{1+2} = 1/6$, what is the view factor between the top surface and the experiment (F_{2+1})?
 - b) Draw and clearly label a resistance network that represents the radiation heat transfer problem.
 - c) Calculate the values of all of the resistances in your diagram from (b).
 - d) What is the net rate of heat transfer to the cryogenic experiment?
 - e) How would your answer to (d) change if the emissivity of the experiment were reduced (would the heat transfer to the cryogenic experiment increase, decrease, or stay the same)?
 - f) How would your answer to (d) change if the emissivity of the insulated sides were reduced (would the heat transfer to the cryogenic experiment increase, decrease, or stay the same)?
 - g) use the Integral Equation to repeat part d) of calculating the net rate of heat transfer to the cryogenic experiment.

a) For the enclosed spherical experiment, $F_{i-2} + F_{i-3} = 1.0 \Rightarrow F_{i-3} = 1 - F_{i-2} = \frac{5}{6}$

$$F_{2-1} = F_{1-2} \frac{A_1}{A_2} = \frac{1}{6} \cdot \frac{0.007854}{0.21} = 0.131$$



c) View Factors:

$$F_{1-1} = 0 = F_{2-2}, \quad F_{1-2} = \frac{1}{6}, \quad F_{1-3} = \frac{5}{6}$$

$$F_{3-1} = F_{1-3} = \frac{A_1}{A_3} = \frac{5}{6} \cdot \frac{0.007854}{0.05} = 0.131 = F_{2-1}$$

$$F_{2-3} = 1 - F_{2-1} = 1 - 0.131 = 0.869$$

$$F_{3-2} = F_{2-3} \frac{A_2}{A_3} = \frac{0.869}{5} = 0.1738$$

$$F_{3-3} = 1 - F_{3-2} - F_{3-1} = 1 - 0.1738 - 0.131 = 0.6952$$

F	1	2	3	j
1	0	$\frac{1}{6}$	$\frac{5}{6}$	
2	0.131	0	0.869	
3	0.131	0.1738	0.6952	
i				

$$\text{Resistances: } R_{s,i} = \frac{1 - \epsilon_i}{A_i \epsilon_i}, \quad R_{i,j} = \frac{1}{A_i F_{i-j}} = \frac{1}{A_j F_{j-i}}$$

$$R_{s,1} = \frac{1 - 0.2}{0.007854 \text{ m}^2 \cdot 0.2} = 509.3 \text{ /m}^2$$

$$R_{s,3} = \frac{1 - 0.1}{0.05 \text{ m}^2 \cdot 0.1} = 180 \text{ } 1/\text{m}^2$$

$$R_{1-3} = \frac{1}{0.05m^2 \cdot 0.131} = 152.67 / m^2$$

$$R_{1-2} = \frac{1}{0.01m^2 \cdot 0.131} = 763.36/m^2$$

$$R_{2-3} = \frac{1}{0.01m^2 \cdot 0.8Gg} = 115.1 / m$$

$$d) \quad E_{bh} = \sigma T^4 \quad , \quad \sigma = 5.67 \times 10^{-8} \text{ W/m}^2 \text{ K}^4$$

$$E_{b,1} = 5.67 \times 10^{-8} \text{ W/m}^2 \text{ K}^4 \quad (100 \text{ K})^4 = 5.67 \text{ W/m}^2$$

$$E_{b,2} = 5.67 \times 10^{-8} \text{ W/m}^2 \text{ K}^4 \quad (300 \text{ K})^4 = 459.27 \text{ W/m}^2$$

$$\frac{E_{b,2} - E_{b,3}}{R_{2-3} + R_{5,3}} = \frac{E_{b,3} - E_{b,1}}{R_{5,3} + R_{4-3} + R_{3,1}}$$

$$E_{b,3} \left(\frac{1}{R_{s,3} + R_{l-3} + R_{s,1}} + \frac{1}{R_{2-3} + R_{s,3}} \right) = \frac{E_{b,2}}{R_{2-3} + R_{s,3}} + \frac{E_{b,1}}{R_{s,3} + R_{l-3} + R_{s,1}}$$

$$E_{b,3} = 341.55 \text{ W/m}^2$$

$$\dot{Q}_1 = \frac{E_{b,2} - E_{b,1}}{R_{s,1} + R_{l-2}} + \frac{E_{b,3} - E_{b,1}}{R_{s,1} + R_{l-1} + R_{s,3}} = \frac{459.27 \text{ W/m}^2 - 5.67 \text{ W/m}^2}{509.3 \text{ m}^2} + \frac{341.55 \text{ W/m}^2 - 5.67 \text{ W/m}^2}{509.3 \text{ m}^2 + 152.67 \text{ m}^2 + 180 \text{ m}^2}$$

$$\dot{B}_1 = 0.755 \text{ W}$$

e) Try $\epsilon_{i,new} = \epsilon_{i/2} = 0.1$

$$R_{s,1} = \frac{1-0.1}{0.007854m^2 \cdot 0.1} = 1145.91 \text{ W/m}^2 \quad , F_1, R_{i-2}, \text{ and } R_{i-3} \text{ are unchanged}$$

$$E_{b,3} = 383.8 \text{ W/m}^2$$

$$\dot{q}_i = 0.493 \text{ W}$$

decreasing the emissivity of the experiment decreases the heat transferred to it

f) Try $\epsilon_{i,new} = \epsilon_{i/2} = 0.05$

$$R_{s,1} = 380 \text{ W/m}^2$$

$$E_{b,3} = 290.9 \text{ W/m}^2$$

$$\dot{q}_i = 0.63 \text{ W}$$

decreasing the emissivity of the insulated walls also decreases the heat transferred to the experiment

$$g) \frac{\dot{q}_i''}{\epsilon_i} - \sum_{j=1}^N \left(\frac{1}{\epsilon_j} - 1 \right) F_{ij} \dot{q}_j'' = E_{bi} - \sum_{j=1}^N F_{ij} E_{bj}$$

$$\dot{q}_1'' = E_{b1} - (E_{b2} F_{1-2} + E_{b3} F_{1-3})$$

$$\dot{q}_2'' = E_{b2} - (E_{b1} F_{2-1} + E_{b3} F_{2-3})$$

$$\dot{q}_3'' = E_{b3} - (E_{b1} F_{3-1} + E_{b2} F_{3-2}) \Rightarrow E_{b3} = 0 + (5.67 \cdot 0.131 + 459.27 \cdot 0.1738) \text{ W/m}^2 = 80.56 \text{ W/m}^2$$

$$\dot{q}_1'' = 5.67 \text{ W/m}^2 - (459.27 \cdot \frac{1}{6} + 80.56 \cdot \frac{5}{6}) \text{ W/m}^2 = -138 \text{ W/m}^2$$

$$\dot{q}_i = \dot{q}_i'' A_i = -138 \text{ W/m}^2 \cdot 0.007854 \text{ m}^2 = -1.08 \text{ W}$$

$$\dot{q}_i = 1.08 \text{ W}$$

Slightly larger than the result from the resistance diagram

Problem 2

A satellite orbits the earth at a height (above the surface of the earth) of $H_{\text{orbit}} = 3.5 \times 10^5 \text{ m}$. The diameter of the earth is $D_{\text{earth}} = 1.29 \times 10^7 \text{ m}$ and the temperature of the earth is $T_{\text{earth}} = 300 \text{ K}$. The distance between the earth and the sun is approximately $R = 1.497 \times 10^11 \text{ m}$. The sun has diameter $D_{\text{sun}} = 1.39 \times 10^9 \text{ m}$ and the surface temperature of the sun is approximately $T_{\text{sun}} = 5780 \text{ K}$. The earth and the sun can be considered black. The satellite is spherical with diameter $D_{\text{sat}} = 1 \text{ m}$. The emissivity of the satellite surface is $\epsilon_{\text{sat}} = 0.5$. There is $q_{\text{sat}} = 100 \text{ W}$ of power dissipation within the satellite that must be rejected from its surface through radiation.

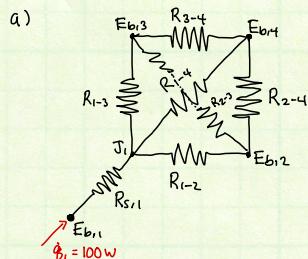
Note: Using the dimensions above, the view factor between the satellite (surface 1) and the earth (surface 2) is $F_{1-2} = 0.3417$; the view factor between the satellite (surface 1) and the sun (surface 3) is $F_{1-3} = 0.000005389$. **The outer space has temperature of 3 K. Your interest is the radiation exchange between the satellite and the others only. → assuming no radiation exchange w/ space**

a) Draw the resistance network that represents this problem and estimate the steady state temperature of the satellite when it is on the day-side of the earth.

b) Estimate the steady state temperature of the satellite when it is on the night-side of the earth (no solar radiation).

c) Plot the steady-state day-side temperature as a function of the satellite emissivity. Explain the shape of your plot.

d) Overlay on your plot from (c) the steady-state day-side temperature as a function of the satellite emissivity if the satellite power dissipation were $q_{\text{sat}} = 0 \text{ W}$. Explain the shape of your plot.

**Surfaces:**1: satellite, $T_1 = ?$, $\epsilon_1 = 0.5$, $A_1 = 3.14 \text{ m}^2$ 2: earth, $T_2 = 300 \text{ K}$, $\epsilon_2 = 1.0$, $A_2 = 5.228 \times 10^{14} \text{ m}^2$ 3: sun, $T_3 = 5780 \text{ K}$, $\epsilon_3 = 1.0$, $A_3 = 6.07 \times 10^{18} \text{ m}^2$ 4: space, $T_4 = 3 \text{ K}$, $\epsilon_4 = 1.0$ (assumed)**View Factors:**

$$F_{1-2} = 0.3417$$

$$a) F_{1-3} = 0.000005389$$

$$b) F_{1-4} = 0$$

$$F_{2-3} = 1, F_{2-4} = 0.658245, F_{3-4} = 0.65833 \rightarrow \text{not considered}$$

$$E_b = \sigma T^4, E_{b,1} = 459.27 \text{ W/m}^2, E_{b,2} = 63,284,071.5 \text{ W/m}^2$$

$$R_{i-j} = \frac{1}{A_i F_{i,j}}, R_{1-2} = 2.926 \text{ m}^2, R_{1-3} = 185,563.2 \text{ m}^2, R_{s,1} = \frac{1 - \epsilon_1}{A_1 \epsilon_1}$$

$$\dot{q}_i = \frac{E_{b,1} - E_{b,2}}{R_{s,1} + R_{1-2}} - \frac{E_{b,1} - E_{b,11}}{R_{s,1} + R_{1-3}} = E_{b,1} \left(\frac{1}{R_{s,1} + R_{1-2}} + \frac{1}{R_{s,1} + R_{1-3}} \right) - \frac{E_{b,2}}{R_{s,1} + R_{1-2}} - \frac{E_{b,11}}{R_{s,1} + R_{1-3}}$$

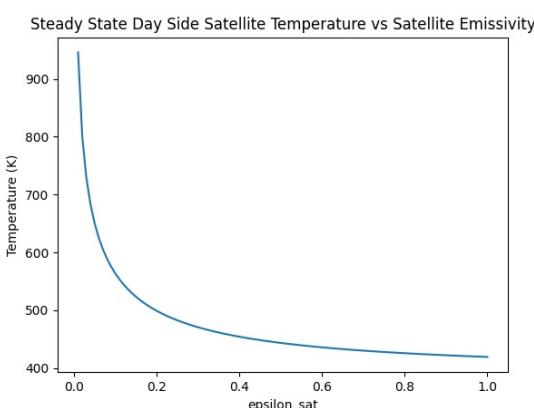
$$E_{b,11} = \left(\dot{q}_i + \frac{E_{b,2}}{R_{s,1} + R_{1-2}} + \beta \frac{E_{b,13}}{R_{s,1} + R_{1-3}} \right) / \left(\frac{1}{R_{s,1} + R_{1-2}} + \beta \frac{1}{R_{s,1} + R_{1-3}} \right), \beta = 1 \text{ if day-side}, \beta = 0 \text{ if night-side}$$

$$T_i = \sqrt[4]{E_{b,1}/\sigma}$$

$$T_{i,\text{day}} = 443.37 \text{ K}$$

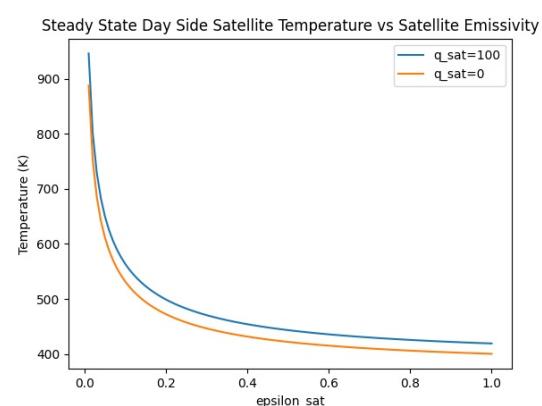
$$b) T_{i,\text{night}} = 350.11 \text{ K}$$

c)



$R_{s,1}$ is inversely proportional to ϵ
as $R_{s,1}$ increases, so does T_i steady-state

d)



the steady state temperature is lower if the satellite doesn't have to reject heat from its electronics