# Biomechanical Analysis of Chicken Ulna and Radius: A Comparative Study of Three-Point and Four-Point Bending

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## I. Introduction

This experiment delves into the mechanical properties of chicken bones through three-point and four-point bending tests. Bones, with their intricate internal and external structures, exhibit unique bending behaviors, particularly in response to lever-like forces. The objective is to characterize how these bones respond to bending loads, offering insights into the mechanical interactions of their cortical and cancellous components. The hypothesis posits that geometric details, such as cortical thickness and cancellous distribution, intricately influence the bones' bending behavior. By connecting the experiment to bone anatomy and biomechanics, this study seeks to unravel the nuanced relationship between bone structure and mechanical response.

#### II. MATERIALS AND METHODS

Two chicken bones, the radius and ulna, underwent three-point (ulna) and four-point (radius) bending tests using the CellScale UniVert compression tester following the lab manual [1]. The machine was set up with a template, and the bones were positioned symmetrically between supports. Load cell zeroing and image documentation preceded test execution. Moment of inertia (MOI) was calculated using a caliper and ellipse approximation. In addition to the collected data, image analysis in SolidWorks and data analysis from previous Sawbone samples informed the generation of force-displacement plots, identification of linear regions, and the calculation of Young's modulus and ultimate stress for both bending configurations.

## III. RESULTS

Sample calculations for Sawbone test result values can be found in Appendix A.

TABLE I SAWBONE SAMPLE RESULTS

	3-Point Bending	4-Point Bending
Young's Modulus (E) Ultimate Stress $(\sigma_u)$ MOI	$\begin{array}{c} 1.4325 \text{ MPa} \\ 145.5435 \text{ kPa} \\ 7.2115 \times 10^{-9} \text{ m}^4 \end{array}$	$\begin{array}{c} 11.4963 \text{ MPa} \\ 338.5217 \text{ kPa} \\ 4.9426 \times 10^{-9} \text{ m}^4 \end{array}$
% Difference MOI	37.3346 %	

Sample calculations for radius and ulna test result values can be found in Appendix B.

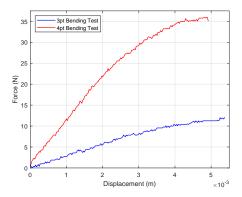


Fig. 1. Plot of force vs. displacement for both Sawbone bending tests.

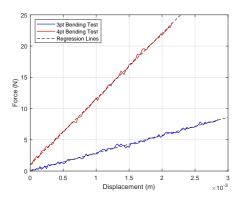


Fig. 2. Plot of the linear regions of force vs. displacement for both Sawbone bending tests.

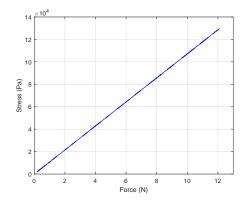


Fig. 3. Plot of stress vs. load for 3-point Sawbone bending test.

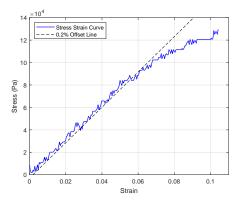


Fig. 4. Plot of stress vs. strain for 3-point Sawbone bending test.

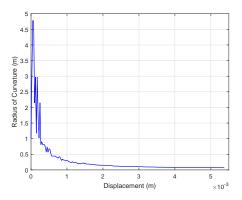


Fig. 5. Plot of radius of curvature vs. displacement for 3-point Sawbone bending test.

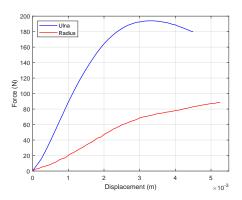


Fig. 6. Plot of force vs. displacement for both chicken bone bending tests.

TABLE II
ULNA AND RADIUS RESULTS USING THE ELLIPSE APPROXIMATION
METHOD AND IMAGE ANALYSIS METHOD

	Ulna	Radius
MOI <sub>image</sub>	$71.23~\mathrm{mm}^4$	16.26 mm <sup>4</sup>
MOI <sub>ellipse</sub>	$24.553~\mathrm{mm}^4$	$7.2942 \text{ mm}^4$
% Difference MOI	97.464~%	76.1293~%
Young's Modulus (E) <sub>image</sub>	5.0289 GPa	8.9242 GPa
Young's Modulus (E) <sub>ellipse</sub>	14.5891 GPa	19.8935 GPa
% Difference E	97.464 %	76.1293%
Ultimate Stress $(\sigma_u)_{image}$	108.6298 MPa	109.4894 MPa
Ultimate Stress $(\sigma_u)_{\text{ellipse}}$	315.1427 MPa	244.0707 MPa
% Difference $\sigma_u$	97.464%	76.1293%

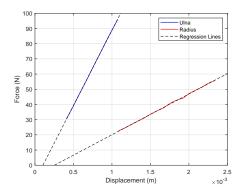


Fig. 7. Plot of the linear regions of force vs. displacement for both chicken bone bending tests.

#### IV. DISCUSSION

Rotating the Sawbone specimen by 90° in a three-point bending experiment alters geometric parameters, affecting bending stiffness and MOI, while leaving material-specific properties like Young's modulus and ultimate stress unaffected. Applying the 0.2% offset method in the three-point bending test predicted the Sawbone specimen's yield stress for a tensile test, yielding 85.1897 kPa. This method correlates with tension in a uniaxial test. The flexure formula calculates the constant radius of curvature (R) across the beam length under linear elastic bending, observed in Fig. 5. In four-point bending, the UniVert machine's limitations influenced support distance choices, impacting load application and machine load capacity.

Three-point bending induces uniform stress along the beam, resulting in a single peak bending moment, while four-point bending introduces complexity, illustrated in Fig. 5 of [2], potentially causing two-point fractures. Our experiment showed single-point fractures, but varied results exist. Three-point bending suits initial material characterization, while four-point bending delves into material behavior intricacies and potential weak points.

Bones exhibit varying strengths in tension and compression, crucial for resisting bending loads and fractures [3]. Cortical bone provides compression strength, while cancellous bone absorbs energy and deforms during loading [4]. Comparing image analysis and ellipse approximation for moment of inertia (MOI) calculations involves accuracy versus practicality trade-offs. Image analysis is meticulous but requires specialized software, while the ellipse method simplifies irregular bone shapes quickly.

Experimental results in Table I and Table II reveal material differences. The ulna's higher MOI suggests greater bending resistance, aligning with its forearm stability role [5]. The radius's higher Young's Modulus signifies increased stiffness, ideal for rotational movements like pronation and supination [6]. These adaptations align with distinct biomechanical roles of the ulna and radius in the forearm.

#### REFERENCES

- Lab 3 Three and Four Point Bending, Department of Systems Design Engineering, 2023.
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- [5] Y. Lu, G. Thiagarajan, D. P. Nicolella, and M. Johnson, "Load/strain distribution between ulna and radius in the mouse forearm compression loading model," *Medical Engineering & Physics*, vol. 34, pp. 350–356, 2012.
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# APPENDIX A SAMPLE CALCULATIONS FOR SAWBONES

A. MOI — 3pt

Moment of inertia about x-axis [7]:

$$I_x = \frac{1}{12}ba^3$$

$$= \frac{1}{12} (0.00751 \text{ m}) (0.01243 \text{ m})^3$$

$$= 7.2115 \times 10^{-9} \text{ m}^4$$

# B. Young's Modulus — 3pt

The maximum deflection equation for three-point bending [1] is as follows:

$$\delta_{\max} = \frac{PL^3}{48EI_x} \implies \frac{P}{\delta_{\max}} = \frac{48EI_x}{L^3}$$
 (1)

The slope of the linear region of the force vs. displacement curve  $\frac{P}{\delta_{\max}}$  was found using the polyfit function in MATLAR:

$$\frac{P}{\delta_{\rm max}} = 2853.9519 \text{ N} = 2.8540 \text{ kN}$$

Plugging in known values into (1):

$$(2853.9519 \text{ N}) = \frac{48E(7.2115 \times 10^{-9} \text{ m}^4)}{(0.0558 \text{ m})^3}$$
$$E = 1432467.1787 \text{ Pa} = 1.4325 \text{ MPa}$$

# C. Young's Modulus — 4pt

The displacement equation along the shaft in four-point bending [1] is as follows:

$$\begin{split} &\delta(x) \\ &= \frac{P(L-a)}{6LEI_x} \left[ \frac{L}{L-a} (x-a)^3 - x^3 + (L^2 - (L-a)^2) x \right] \\ &\quad + \frac{Pa}{6LEI_x} \left[ \frac{L}{a} (x - (L-a))^3 - x^3 + (L^2 - a^2) x \right] \end{split}$$

Plugging in L = 0.05946 m and a = 0.0201 m:

$$\delta(a) = \frac{P(0.11033)}{EI_x} \left( 3.1804 \times 10^{-5} \right) + \frac{P(0.05634)}{EI_x} \left( 3.3687 \times 10^{-5} \right)$$
$$\delta(a) = \frac{P}{EI_x} \left( 5.4067 \times 10^{-6} \right)$$

The equation can then be rearranged as follows:

$$\frac{P}{\delta(x)} = \frac{EI_x}{5.4067 \times 10^{-6}} \tag{2}$$

The slope of the linear region of the force vs. displacement curve  $\frac{P}{\delta(x)}$  was found using the polyfit function in MAT-IAR.

$$\frac{P}{\delta(x)} = 10509.418 \text{ N} = 10.5194 \text{ kN}$$

Plugging in known values into (2):

$$(10509.418 \text{ N}) = \frac{E(4.9426 \times 10^{-9} \text{ m}^4)}{5.4067 \times 10^{-6}}$$

$$E = 11496261.382 \text{ Pa} = 11.4963 \text{ MPa}$$

# D. Ultimate Stress — 3pt

Peak bending moment is observed at maximum load and at the farthest point from the neutral axis, thus  $c=\frac{a}{2}=0.006215$  m. The ultimate stress can then be found using the bending stress equation [7].

$$\begin{split} |\sigma_u| &= \frac{Mc}{I_x} \\ &= \frac{(0.16888 \text{ N} \cdot \text{m})(0.006215 \text{ m})}{(7.2115 \times 10^{-9} \text{ m}^4)} \\ &= 145543.4559 \text{ Pa} = 145.5435 \text{ kPa} \end{split}$$

# APPENDIX B

SAMPLE CALCULATIONS FOR CHICKEN BONES

# A. MOI Using Ellipse Approximation — Ulna

As seen in Fig. 8, measurements were taken in the orientation where bending occurs about the x-axis.

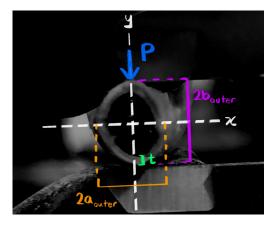


Fig. 8. Cross-section photo of ulna, with measurements taken labeled.

 $a_{outer}=0.0036\,$  m,  $b_{outer}=0.00286\,$  m, and thickness  $t=0.00066\,$  m measured were used to calculate values needed to calculate MOI:

$$a_{inner}=a_{outer}-\frac{t}{2}=0.00327~\mathrm{m}$$
 
$$b_{inner}=b_{outer}-\frac{t}{2}=0.00253~\mathrm{m}$$

Using the equation of moment of inertia of an ellipse [1], the inner and outer MOI was calculated to find the MOI of the hollow ellipse shape of the bone.

$$(I_x)_{outer} = \frac{\pi(a_{outer})(b_{outer})^3}{4} = 6.6144 \times 10^{-11} \text{ m}^4$$

$$(I_x)_{inner} = \frac{\pi(a_{inner})(b_{inner})^3}{4} = 4.1591 \times 10^{-11} \text{ m}^4$$

$$(MOI)_{ellipse} = (I_x)_{outer} - (I_x)_{inner} = 2.4553 \times 10^{-11} \text{ m}^4$$

# B. Young's Modulus — Ulna

Following steps outlined in Appendix A-B, the slope of the linear region of the force vs. displacement curve  $\frac{P}{\delta_{\max}}$  was found using the polyfit function in MATLAB once more:

$$\frac{P}{\delta_{\rm max}} = 98962.6626 \text{ N} = 98.9627 \text{ kN}$$

Plugging in known values into (1):

$$(98962.6626 \text{ N}) = \frac{48E(7.123 \times 10^{-11} \text{ m}^4)}{(0.0558 \text{ m})^3}$$
 
$$E = 5028862791.3324 \text{ Pa} = 5.0289 \text{ GPa}$$

# C. Young's Modulus — Radius

Following steps outlined in Appendix A-C, the slope of the linear region of the force vs. displacement curve  $\frac{P}{\delta(x)}$  was found using the polyfit function in MATLAB once more:

$$\frac{P}{\delta(x)} = 26838.1894 \text{ N} = 26.8382 \text{ kN}$$

Plugging in known values into (2):

$$(26838.1894 \text{ N}) = \frac{E(7.123 \times 10^{-11} \text{ m}^4)}{5.4067 \times 10^{-6}}$$
 
$$E = 8924179123.0254 \text{ Pa} = 8.9242 \text{ GPa}$$

# D. Ultimate Stress — Ulna

Following steps outlined in Appendix A-D, the ultimate stress is calculated using  $c=b_{outer}=0.00286$  m:

$$\begin{split} |\sigma_u| &= \frac{Mc}{\left(MOI\right)_{image}} \\ &= \frac{\left(2.7055 \text{ N} \cdot \text{m}\right)\left(0.00286 \text{ m}\right)}{\left(7.123 \times 10^{-11} \text{ m}^4\right)} \\ &= 108629846.6096 \text{ Pa} = 108.6298 \text{ MPa} \end{split}$$



Fig. 9. Cross-section photo of ulna taken with vernier caliper for scaling.

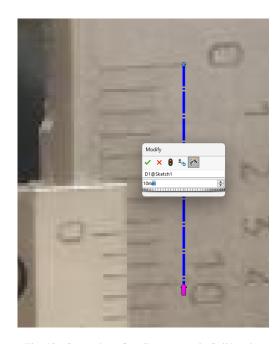


Fig. 10. Screenshot of scaling process in Solidworks.

# E. MOI Using Image Analysis — Ulna

Using Solidworks, a picture of the cross-section of ulna (Fig. 9) was imported into a sketch layer in the front plane so that the coordinate system matched that of Fig. 8. The picture was scaled to match real-life dimensions using the caliper markings, as seen in Fig. 10.

Then, using the spline tool, the cross-sectional area of the bone was outlined, as seen in Fig. 11.

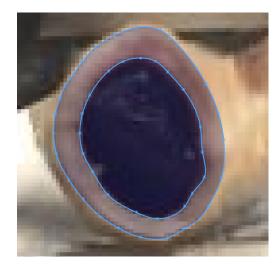


Fig. 11. Screenshot of sketching process in Solidworks.

After confirming the sketch, it was extruded into a solid body (Fig. 12).

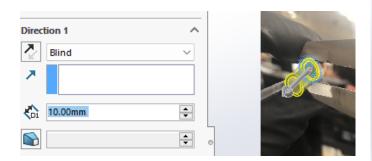


Fig. 12. Screenshot of extruding solid process in Solidworks.

Finally, the cross-sectional face of the solid body was selected and section property analysis was ran on it; the results of the analysis can be seen in Fig. 13.  $I_{xx} = 7.123 \times 10^{-11} \text{ m}^4$  was selected as the appropriate MOI for the bending setup as bending occurred about the x-axis.

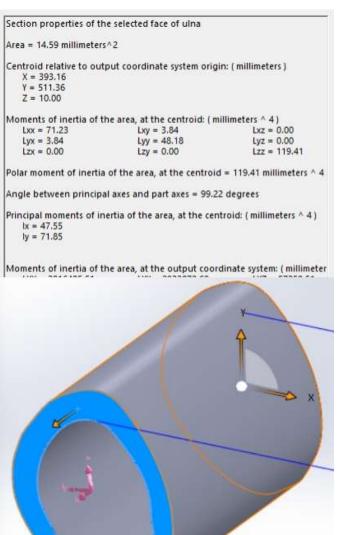


Fig. 13. Results of section property analysis on ulna.