

# A Stable Galerkin Reduced Order Model (ROM) for Compressible Flow

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# Outline

## 1 Motivation

## 2 Overview of the POD/Galerkin Method for Model Reduction

- Step 1: Constructing the POD Modes
- Step 2: Galerkin Projection

## 3 A Stable Galerkin ROM for Compressible Flow

- Stability Definitions
- Equations for Compressible Flow
- Stability-Preserving “Symmetry” Inner Product for Compressible Flow

## 4 Numerical Examples

- Numerical Implementation
- Test Case 1: Purely Random Basis
- Test Case 2: 1D Acoustic Pressure Pulse
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# Why Develop a Fluid Reduced Order Model (ROM)?

CFD modeling of unsteady  
3D flows is expensive!



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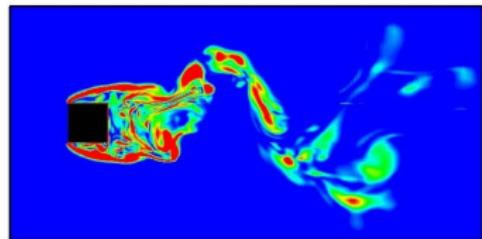
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## Applications in Fluid Dynamics:

- Predictive modeling across a parameter space (e.g., aeroelastic flutter analysis).
- System modeling for active flow control.
- Long-time unsteady flow analysis, e.g., fatigue of a wind turbine blade under variable wind conditions.



# Motivation for Numerical Analysis of ROMs

Use of ROMs in predictive applications raises questions about their stability & convergence.

- Projection ROM approach is an alternative discretization of the governing PDEs.
- Desired numerical properties of a ROM discretization:
  - **Consistency** (with continuous PDEs):
  - **Stability:**
  - **Convergence:**



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This talk focuses on how to construct a Galerkin ROM that is **stable *a priori***



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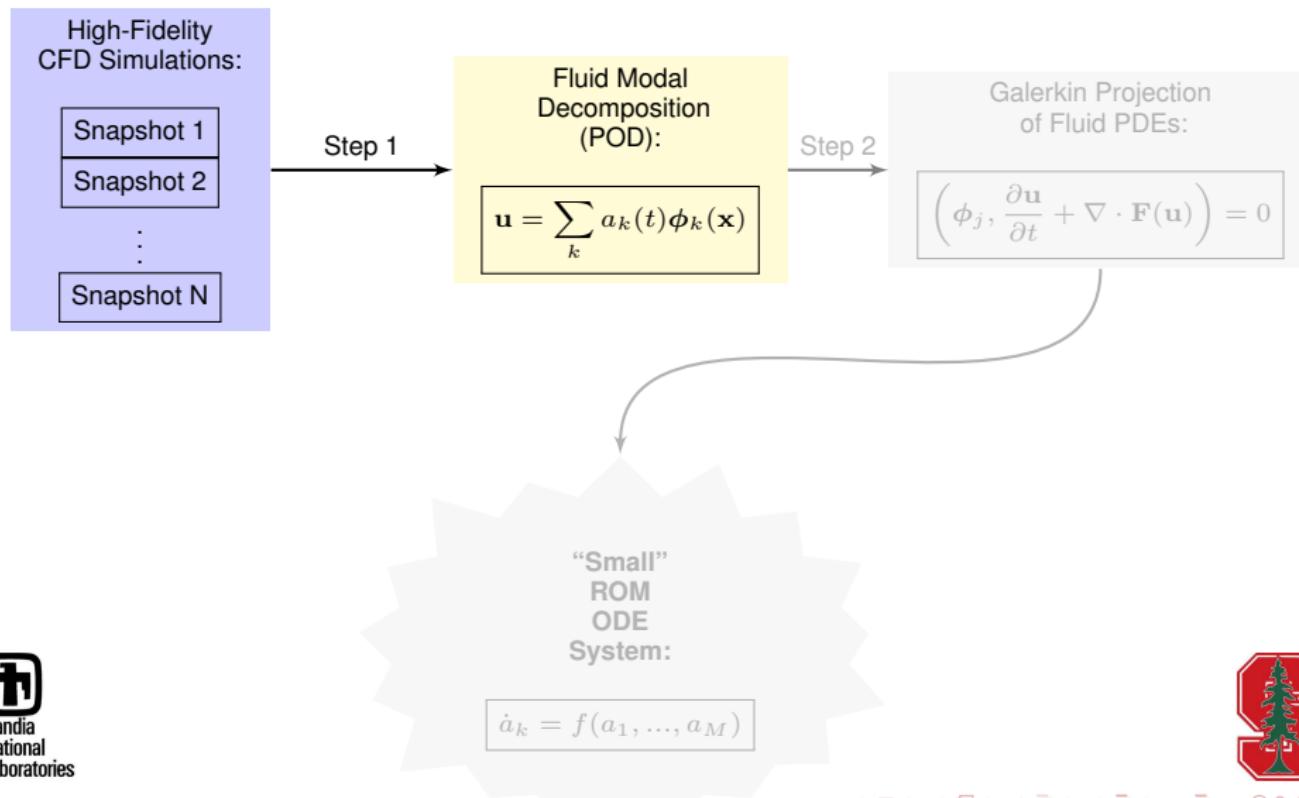
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# Step 1: Constructing the Modes



# Proper Orthogonal Decomposition (POD), a.k.a. “Method of Snapshots”

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$$\max_{\phi \in H(\Omega)} \frac{\langle (\mathbf{u}, \phi)^2 \rangle}{\|\phi\|^2} \quad (1)$$

where

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$$\mathbf{R} \equiv \langle \mathbf{u}^k(\mathbf{u}^k, \phi) \rangle$$

Solution to (1) is the set of  $M$  eigenfunctions  $\{\phi_i\}_{i=1}^M$  corresponding to the  $M$  largest eigenvalues  $\lambda_1 \geq \dots \geq \lambda_M$  of  $\mathbf{R}$

# Properties of the POD Basis

- POD basis  $\{\phi_i\}_{i=1}^M$  is **orthonormal**:  $(\phi_i, \phi_j) = \delta_{ij}$ .
- Average **energy** of projection of the snapshot ensemble onto the  $i^{th}$  mode is given by:

$$\langle (\mathbf{u}^k, \phi_i)^2 \rangle = \lambda_i$$

$$\Rightarrow \text{energy of set } \{\phi_i\}_{i=1}^M = \sum_{j=1}^M \lambda_j$$

- Truncated POD basis  $\{\phi_i\}_{i=1}^M$  describes more energy (on average) of the ensemble than any other linear basis of the same dimension.
- Given  $M \ll N$  modes, ROM solution can be represented as a linear combination of these modes:

$$\underbrace{\mathbf{u}_M(\mathbf{x}, t)}_{\text{ROM solution}} = \sum_{i=1}^M a_i(t) \phi_i(\mathbf{x}) \quad (2)$$

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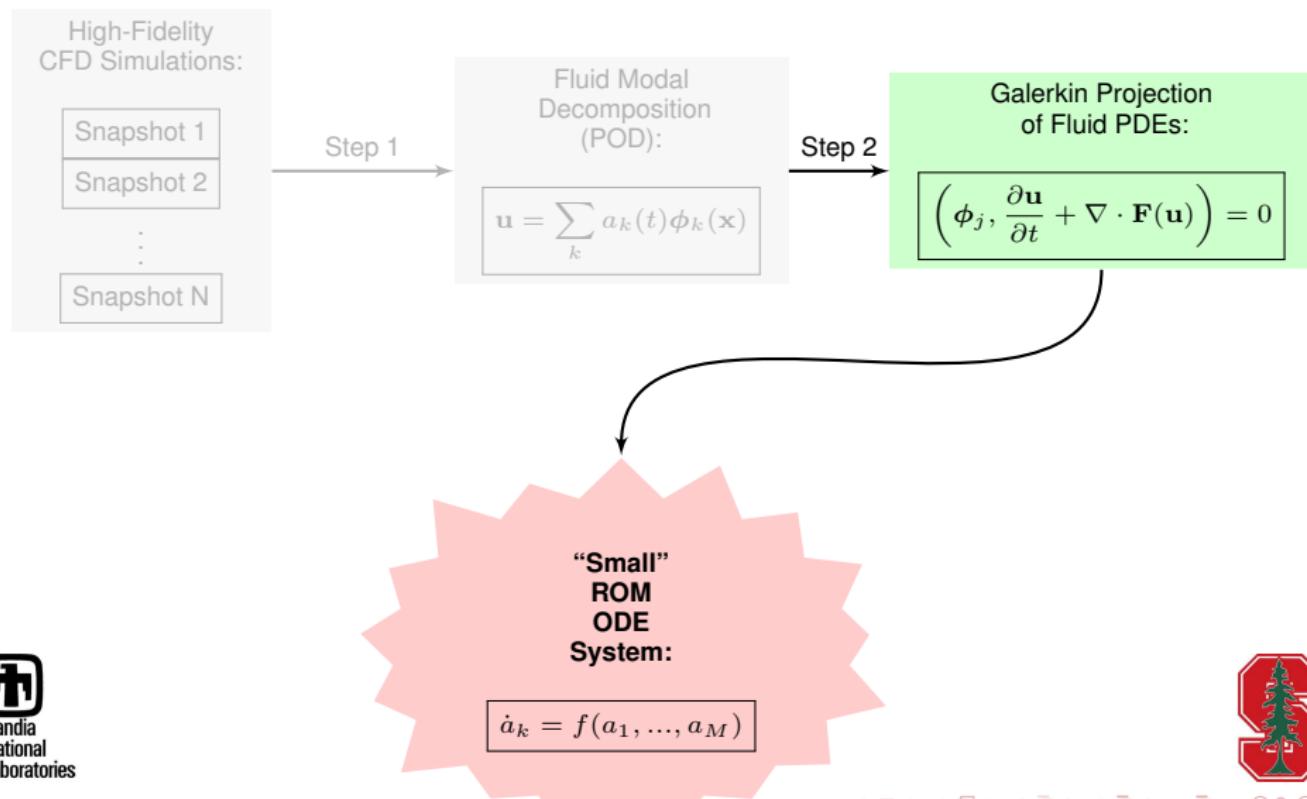
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$$\underbrace{\mathbf{u}_M(\mathbf{x}, t)}_{\text{ROM solution}} = \sum_{i=1}^M \underbrace{a_i(t)}_{\text{unknown ROM dofs}} \underbrace{\phi_i(\mathbf{x})}_{\text{}} \quad (2)$$

# Step 2: Galerkin Projection



# Galerkin Projection of (*Continuous!*) Equations

Governing System of PDEs:

$$\frac{\partial \mathbf{u}}{\partial t} = \mathcal{L}\mathbf{u} + \mathcal{N}_2(\mathbf{u}, \mathbf{u}) + \mathcal{N}_3(\mathbf{u}, \mathbf{u}, \mathbf{u}) \quad (3)$$

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- **Step 2.2:** Substitute the modal decomposition  $\mathbf{u}_M = \sum_{k=1}^M a_k(t) \phi_k(\mathbf{x})$   
→  $\mathbf{u}$  into (4)

$$\begin{aligned} \dot{a}_k(t) &= \sum_{l=1}^M a_l(\phi_k, \mathcal{L}(\phi_l)) + \sum_{l,m=1}^M a_l a_m (\phi_k, \mathcal{N}_2(\phi_l, \phi_m)) \\ &\quad + \sum_{l,m,n=1}^M a_l a_m a_n (\phi_k, \mathcal{N}_3(\phi_l, \phi_m, \phi_n)) \end{aligned} \quad (5)$$

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# Continuous vs. Discrete Projection Approach

## DISCRETE APPROACH

Governing Equations

$$\mathbf{u}_t = \mathcal{L}\mathbf{u}$$



CFD Model

$$\dot{\mathbf{u}}_N = \mathbf{A}_N \mathbf{u}_N$$



Discrete Modal Basis  $\Phi$



Projection of CFD Model  
(Matrix Operation)



ROM

$$\dot{\mathbf{a}} = \Phi^T \mathbf{A}_N \Phi \mathbf{a}$$

## CONTINUOUS APPROACH

Governing Equations

$$\mathbf{u}_t = \mathcal{L}\mathbf{u}$$



CFD Model

$$\dot{\mathbf{u}}_N = \mathbf{A}_N \mathbf{u}_N$$



Continuous Modal Basis\*  $\phi_j(\mathbf{x})$



Projection of Governing Equations  
(Numerical Integration)



ROM

$$\dot{a}_j = (\phi_j, \mathcal{L} \phi_k) a_k$$



\* Continuous functions space is defined using finite elements.

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Analyzed with the **Energy Method**:  
uses an equation for the evolution of numerical solution  
“energy” to determine stability

$$\|u_N(\mathbf{x}, t)\|_E \equiv \left\{ \begin{array}{l} \text{energy of } u_N \text{ in norm } \|\cdot\|_E \\ \text{induced by inner product } (\cdot, \cdot)_E \end{array} \right\}$$

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$$\|u_N(\mathbf{x}, t)\|_E \stackrel{?}{\leq} e^{\beta t} \|u_N(\mathbf{x}, 0)\|_E, \quad \beta \in \mathbb{R}$$

# 3D Linearized Compressible Euler Equations

- Useful for aero-elasticity, aero-acoustics, flow instability analysis.

## Linearization of Full Compressible Euler Equations

$$\mathbf{q}^T(\mathbf{x}, t) \equiv \begin{pmatrix} u_1 & u_2 & u_3 & \zeta & p \end{pmatrix} \equiv \underbrace{\bar{\mathbf{q}}^T(\mathbf{x})}_{\text{mean}} + \underbrace{\mathbf{q}'^T(\mathbf{x}, t)}_{\text{fluctuation}} \in \mathbb{R}^5$$

$$\Rightarrow \boxed{\frac{\partial \mathbf{q}'}{\partial t} + \mathbf{A}_i \frac{\partial \mathbf{q}'}{\partial x_i} + \mathbf{C} \mathbf{q}' = \mathbf{0}} \quad (6)$$

where

$$\mathbf{A}_1 = \begin{pmatrix} \bar{u}_1 & 0 & 0 & 0 & \bar{\zeta} \\ 0 & \bar{u}_1 & 0 & 0 & 0 \\ 0 & 0 & \bar{u}_1 & 0 & 0 \\ -\bar{\zeta} & 0 & 0 & \bar{u}_1 & 0 \\ \gamma \bar{p} & 0 & 0 & 0 & \bar{u}_1 \end{pmatrix}, \quad \mathbf{A}_2 = \begin{pmatrix} \bar{u}_2 & 0 & 0 & 0 & 0 \\ 0 & \bar{u}_2 & 0 & 0 & \bar{\zeta} \\ 0 & 0 & \bar{u}_2 & 0 & 0 \\ 0 & -\bar{\zeta} & 0 & \bar{u}_2 & 0 \\ 0 & \gamma \bar{p} & 0 & 0 & \bar{u}_2 \end{pmatrix}$$

$$\mathbf{A}_3 = \begin{pmatrix} \bar{u}_3 & 0 & 0 & 0 & 0 \\ 0 & \bar{u}_3 & 0 & 0 & 0 \\ 0 & 0 & \bar{u}_3 & 0 & \bar{\zeta} \\ 0 & 0 & -\bar{\zeta} & \bar{u}_3 & 0 \\ 0 & 0 & \gamma \bar{p} & 0 & \bar{u}_3 \end{pmatrix}, \quad \mathbf{C} = \begin{pmatrix} \frac{\partial \bar{u}_1}{\partial x_1} & \frac{\partial \bar{u}_1}{\partial x_2} & \frac{\partial \bar{u}_1}{\partial x_3} & \frac{\partial \bar{p}}{\partial x_1} & 0 \\ \frac{\partial \bar{u}_2}{\partial x_1} & \frac{\partial \bar{u}_2}{\partial x_2} & \frac{\partial \bar{u}_2}{\partial x_3} & \frac{\partial \bar{p}}{\partial x_2} & 0 \\ \frac{\partial \bar{u}_3}{\partial x_1} & \frac{\partial \bar{u}_3}{\partial x_2} & \frac{\partial \bar{u}_3}{\partial x_3} & \frac{\partial \bar{p}}{\partial x_3} & 0 \\ \frac{\partial \bar{\zeta}}{\partial x_1} & \frac{\partial \bar{\zeta}}{\partial x_2} & \frac{\partial \bar{\zeta}}{\partial x_3} & -\nabla \cdot \bar{\mathbf{u}} & 0 \\ \frac{\partial \bar{p}}{\partial x_1} & \frac{\partial \bar{p}}{\partial x_2} & \frac{\partial \bar{p}}{\partial x_3} & 0 & \gamma \nabla \cdot \bar{\mathbf{u}} \end{pmatrix}$$

# Symmetrized Compressible Euler Equations & Symmetry Inner Product

Energy stability can be proven following “symmetrization” of the linearized compressible Euler equations.

- Linearized hyperbolic compressible Euler system is “symmetrizable”.
- Pre-multiply equations by symmetric positive definite matrix:

$$\mathbf{H} = \begin{pmatrix} \bar{\rho} & 0 & 0 & 0 & 0 \\ 0 & \bar{\rho} & 0 & 0 & 0 \\ 0 & 0 & \bar{\rho} & 0 & 0 \\ 0 & 0 & 0 & \alpha^2 \gamma \bar{\rho} \bar{p} & \bar{\rho} \alpha^2 \\ 0 & 0 & 0 & \bar{\rho} \alpha^2 & \frac{1+\alpha^2}{\gamma \bar{p}} \end{pmatrix} \Rightarrow \boxed{\mathbf{H} \frac{\partial \mathbf{q}'}{\partial t} + \boxed{\mathbf{H} \mathbf{A}_i} \frac{\partial \mathbf{q}'}{\partial x_i} + \mathbf{H} \mathbf{C} \mathbf{q}' = \mathbf{0}}$$

- $\mathbf{H}$  is called the “symmetrizer” of the system:  $\mathbf{H} \mathbf{A}_i$  are all symmetric.
- Define the “symmetry” inner product and “symmetry” norm:

$$(\mathbf{q}'^{(1)}, \mathbf{q}'^{(2)})_{(\mathbf{H}, \Omega)} \equiv \int_{\Omega} [\mathbf{q}'^{(1)}]^T \mathbf{H} \mathbf{q}'^{(2)} d\Omega, \quad \|\mathbf{q}'\|_{(\mathbf{H}, \Omega)} \equiv (\mathbf{q}', \mathbf{q}')_{(\mathbf{H}, \Omega)}$$
(7)

# Stability in the Symmetry Inner Product

$$\begin{aligned}
 \frac{d}{dt} \|\mathbf{q}'\|_{(\mathbf{H}, \Omega)}^2 &= - \int_{\Omega} [\mathbf{q}']^T \mathbf{H} \left[ \mathbf{A}_i \frac{\partial \mathbf{q}'}{\partial x_i} + \mathbf{C} \mathbf{q}' \right] d\Omega \\
 &= - \int_{\partial\Omega} [\mathbf{q}']^T \mathbf{H} \mathbf{A}_i \mathbf{n}_i \mathbf{q}' dS + \int_{\Omega} [\mathbf{q}']^T \left( \frac{\partial}{\partial x_i} (\mathbf{H} \mathbf{A}_i) - \mathbf{H} \mathbf{C} - \mathbf{C}^T \mathbf{H} \right) \mathbf{q}' d\Omega \\
 &= - \int_{\partial\Omega} [\mathbf{q}']^T \mathbf{H} \mathbf{A}_i \mathbf{n}_i \mathbf{q}' dS + \int_{\Omega} [\mathbf{q}']^T \mathbf{H}^{-T/2} \mathbf{B} \mathbf{H}^{T/2} \mathbf{q}' d\Omega \\
 &\leq - \int_{\partial\Omega} [\mathbf{q}']^T \mathbf{H} \mathbf{A}_i \mathbf{n}_i \mathbf{q}' dS + \beta (\mathbf{q}', \mathbf{q}')_{(\mathbf{H}, \Omega)} \\
 &\leq \beta \|\mathbf{q}'\|_{(\mathbf{H}, \Omega)}^2 \quad \text{if } \int_{\partial\Omega} [\mathbf{q}']^T \mathbf{H} \mathbf{A}_i \mathbf{n}_i \mathbf{q}' dS \geq 0 \text{ (well-posed BCs)}
 \end{aligned}$$

where  $\beta$  is an upper bound on the eigenvalues of

$$\mathbf{B} \equiv \mathbf{H}^{-T/2} \frac{\partial(\mathbf{H} \mathbf{A}_i)}{\partial x_i} \mathbf{H}^{-1/2} - \mathbf{H}^{1/2} \mathbf{C} \mathbf{H}^{-1/2} - (\mathbf{H}^{1/2} \mathbf{C} \mathbf{H}^{-1/2})^T$$

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- Exact solutions to the linearized Euler equations satisfy:

$$\|\mathbf{q}'(\mathbf{x}, t)\|_{(\mathbf{H}, \Omega)} \leq e^{\beta t} \|\mathbf{q}'(\mathbf{x}, 0)\|_{(\mathbf{H}, \Omega)}$$

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 &= - \int_{\partial\Omega} [\mathbf{q}']^T \mathbf{H} \mathbf{A}_i n_i \mathbf{q}' dS + \int_{\Omega} [\mathbf{q}']^T \mathbf{H}^{-T/2} \mathbf{B} \mathbf{H}^{T/2} \mathbf{q}' d\Omega \\
 &\leq - \int_{\partial\Omega} [\mathbf{q}']^T \mathbf{H} \mathbf{A}_i n_i \mathbf{q}' dS + \beta (\mathbf{q}', \mathbf{q}')_{(\mathbf{H}, \Omega)} \\
 &\leq \beta \|\mathbf{q}'\|_{(\mathbf{H}, \Omega)}^2 \quad \text{if } \int_{\partial\Omega} [\mathbf{q}']^T \mathbf{H} \mathbf{A}_i n_i \mathbf{q}' dS \geq 0 \text{ (well-posed BCs)}
 \end{aligned}$$

where  $\beta$  is an upper bound on the eigenvalues of

$$\mathbf{B} \equiv \mathbf{H}^{-T/2} \frac{\partial(\mathbf{H} \mathbf{A}_i)}{\partial x_i} \mathbf{H}^{-1/2} - \mathbf{H}^{1/2} \mathbf{C} \mathbf{H}^{-1/2} - (\mathbf{H}^{1/2} \mathbf{C} \mathbf{H}^{-1/2})^T$$

- Exact solutions to the linearized Euler equations satisfy:

$$\|\mathbf{q}'(\mathbf{x}, t)\|_{(\mathbf{H}, \Omega)} \leq e^{\beta t} \|\mathbf{q}'(\mathbf{x}, 0)\|_{(\mathbf{H}, \Omega)}$$

- It turns out that the Galerkin approximation  $\mathbf{q}'_M = \sum_{i=1}^M a_k(t) \phi_k(\mathbf{x})$  satisfies the same energy expression as for the continuous equations:

$$\|\mathbf{q}'_M(\mathbf{x}, t)\|_{(\mathbf{H}, \Omega)} \leq e^{\beta t} \|\mathbf{q}'_M(\mathbf{x}, 0)\|_{(\mathbf{H}, \Omega)}$$

i.e., it is stable.

# Stability in the Symmetry Inner Product

$$\begin{aligned}
 \frac{d}{dt} \|\mathbf{q}'\|_{(\mathbf{H}, \Omega)}^2 &= - \int_{\Omega} [\mathbf{q}']^T \mathbf{H} \left[ \mathbf{A}_i \frac{\partial \mathbf{q}'}{\partial x_i} + \mathbf{C} \mathbf{q}' \right] d\Omega \\
 &= - \int_{\partial\Omega} [\mathbf{q}']^T \mathbf{H} \mathbf{A}_i n_i \mathbf{q}' dS + \int_{\Omega} [\mathbf{q}']^T \left( \frac{\partial}{\partial x_i} (\mathbf{H} \mathbf{A}_i) - \mathbf{H} \mathbf{C} - \mathbf{C}^T \mathbf{H} \right) \mathbf{q}' d\Omega \\
 &= - \int_{\partial\Omega} [\mathbf{q}']^T \mathbf{H} \mathbf{A}_i n_i \mathbf{q}' dS + \int_{\Omega} [\mathbf{q}']^T \mathbf{H}^{-T/2} \mathbf{B} \mathbf{H}^{T/2} \mathbf{q}' d\Omega \\
 &\leq - \int_{\partial\Omega} [\mathbf{q}']^T \mathbf{H} \mathbf{A}_i n_i \mathbf{q}' dS + \beta (\mathbf{q}', \mathbf{q}')_{(\mathbf{H}, \Omega)} \\
 &\leq \beta \|\mathbf{q}'\|_{(\mathbf{H}, \Omega)}^2 \quad \text{if } \int_{\partial\Omega} [\mathbf{q}']^T \mathbf{H} \mathbf{A}_i n_i \mathbf{q}' dS \geq 0 \text{ (well-posed BCs)}
 \end{aligned}$$

where  $\beta$  is an upper bound on the eigenvalues of

$$\mathbf{B} \equiv \mathbf{H}^{-T/2} \frac{\partial(\mathbf{H} \mathbf{A}_i)}{\partial x_i} \mathbf{H}^{-1/2} - \mathbf{H}^{1/2} \mathbf{C} \mathbf{H}^{-1/2} - (\mathbf{H}^{1/2} \mathbf{C} \mathbf{H}^{-1/2})^T$$

- Exact solutions to the linearized Euler equations satisfy:

$$\|\mathbf{q}'(\mathbf{x}, t)\|_{(\mathbf{H}, \Omega)} \leq e^{\beta t} \|\mathbf{q}'(\mathbf{x}, 0)\|_{(\mathbf{H}, \Omega)}$$

- It turns out that the Galerkin approximation  $\mathbf{q}'_M = \sum_{i=1}^M a_k(t) \phi_k(\mathbf{x})$  satisfies the same energy expression as for the continuous equations:

$$\|\mathbf{q}'_M(\mathbf{x}, t)\|_{(\mathbf{H}, \Omega)} \leq e^{\beta t} \|\mathbf{q}'_M(\mathbf{x}, 0)\|_{(\mathbf{H}, \Omega)}$$

i.e., it is stable.

- For uniform base flow, the Galerkin scheme satisfies the strong stability estimate:

$$\|\mathbf{q}'_M(\mathbf{x}, t)\|_{(\mathbf{H}, \Omega)} \leq \|\mathbf{q}'_M(\mathbf{x}, 0)\|_{(\mathbf{H}, \Omega)}$$

# Stability in the Symmetry Inner Product (cont'd)

- Stability analysis dictates that we use the symmetry inner product

$$\begin{aligned} (\mathbf{q}'^{(1)}, \mathbf{q}'^{(2)})_{(\mathbf{H}, \Omega)} &\equiv \int_{\Omega} [\mathbf{q}'^{(1)}]^T \mathbf{H} \mathbf{q}'^{(2)} d\Omega \\ &= \int_{\Omega} \left[ \bar{\rho} \mathbf{u}'^{(1)} \cdot \mathbf{u}'^{(2)} + \alpha^2 \gamma \bar{\rho}^2 \zeta'^{(1)} \zeta'^{(2)} \right. \\ &\quad \left. + \frac{1+\alpha^2}{\gamma \bar{\rho}} + \alpha^2 \bar{\rho} (\zeta'^{(2)} p'^{(1)} + \zeta'^{(1)} p'^{(2)}) \right] d\Omega \end{aligned}$$

to compute the POD modes and perform the Galerkin projection.

## Practical Implication of Stability Analysis

Symmetry inner product ensures that any “bad” modes will not introduce spurious non-physical numerical instabilities into the Galerkin approximation.



- Galerkin projection step is stable for *any* basis in the symmetry inner product!



# Steps to Obtain a Stable Compressible Fluid ROM

- Galerkin-project the equations in the symmetry inner product (7):

$$\left( \phi_j, \frac{\partial \mathbf{q}'_M}{\partial t} \right)_{(\mathbf{H},\Omega)} + \left( \phi_j, \mathbf{A}_i \frac{\partial \mathbf{q}'_M}{\partial x_i} \right)_{(\mathbf{H},\Omega)} + (\phi_j, \mathbf{C} \mathbf{q}'_M)_{(\mathbf{H},\Omega)} = 0 \quad (8)$$

# Steps to Obtain a Stable Compressible Fluid ROM

- Galerkin-project the equations in the symmetry inner product (7):

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- Integrate second term in (8) by parts

$$\left( \phi_j, \frac{\partial \mathbf{q}'_M}{\partial t} \right)_{(\mathbf{H}, \Omega)} = \int_{\Omega} \left[ \frac{\partial}{\partial x_i} [\phi_j^T \mathbf{H} \mathbf{A}_i] - \phi_j^T \mathbf{H} \mathbf{C} \right] \mathbf{q}'_M d\Omega - \int_{\partial\Omega} \phi_j^T \mathbf{H} \mathbf{A}_n \cdot \mathbf{q}'_M ds$$

# Steps to Obtain a Stable Compressible Fluid ROM

- Galerkin-project the equations in the symmetry inner product (7):

$$\left( \phi_j, \frac{\partial \mathbf{q}'_M}{\partial t} \right)_{(\mathbf{H}, \Omega)} + \left( \phi_j, \mathbf{A}_i \frac{\partial \mathbf{q}'_M}{\partial x_i} \right)_{(\mathbf{H}, \Omega)} + (\phi_j, \mathbf{C} \mathbf{q}'_M)_{(\mathbf{H}, \Omega)} = 0 \quad (8)$$

- Integrate second term in (8) by parts and apply boundary conditions:

$$\left( \phi_j, \frac{\partial \mathbf{q}'_M}{\partial t} \right)_{(\mathbf{H}, \Omega)} = \int_{\Omega} \left[ \frac{\partial}{\partial x_i} [\phi_j^T \mathbf{H} \mathbf{A}_i] - \phi_j^T \mathbf{H} \mathbf{C} \right] \mathbf{q}'_M d\Omega - \int_{\partial\Omega} \phi_j^T \mathbf{H} \mathbf{A}_i n_i \cdot \mathbf{q}'_M dS$$

Insert boundary conditions into boundary integrals (weak implementation)

\* Energy stability is maintained if the boundary conditions are such that

$$\int_{\partial\Omega} \phi_j^T \mathbf{H} \mathbf{A}_i n_i \mathbf{q}'_M dS \geq 0.$$

# Steps to Obtain a Stable Compressible Fluid ROM

- Galerkin-project the equations in the symmetry inner product (7):

$$\left( \phi_j, \frac{\partial \mathbf{q}'_M}{\partial t} \right)_{(\mathbf{H}, \Omega)} + \left( \phi_j, \mathbf{A}_i \frac{\partial \mathbf{q}'_M}{\partial x_i} \right)_{(\mathbf{H}, \Omega)} + (\phi_j, \mathbf{C} \mathbf{q}'_M)_{(\mathbf{H}, \Omega)} = 0 \quad (8)$$

- Integrate second term in (8) by parts and apply boundary conditions:

$$\left( \phi_j, \frac{\partial \mathbf{q}'_M}{\partial t} \right)_{(\mathbf{H}, \Omega)} = \int_{\Omega} \left[ \frac{\partial}{\partial x_i} [\phi_j^T \mathbf{H} \mathbf{A}_i] - \phi_j^T \mathbf{H} \mathbf{C} \right] \mathbf{q}'_M d\Omega - \int_{\partial\Omega} \phi_j^T \mathbf{H} \mathbf{A}_i n_i \mathbf{q}'_M dS$$

Insert boundary conditions into boundary integrals (weak implementation)

\* Energy stability is maintained if the boundary conditions are such that

$$\int_{\partial\Omega} \phi_j^T \mathbf{H} \mathbf{A}_i n_i \mathbf{q}'_M dS \geq 0.$$

- Substitute modal decomposition  $\mathbf{q}'_M = \sum_k a_k(t) \phi_k(\mathbf{x})$  to obtain an  $M \times M$  linear dynamical system of the form  $\dot{\mathbf{a}} = \mathbf{K}\mathbf{a}$

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## 5 Summary & Further Work

# Numerical Implementation of Fluid ROM

- So far, all analysis is for continuous and smooth basis functions, and exact evaluation of inner product integrals.

## Stability-Preserving Discrete Implementation:

- Define solution snapshots and POD basis functions using a piecewise smooth finite element representation:

$$\mathbf{q}_e^{h'}(\mathbf{x}) = \sum_{i=1}^{N_n} N_i(\mathbf{x}) \mathbf{q}'_i$$

- Apply Gauss quadrature rules  $\left( \int_{\Omega} f(\mathbf{x}) d\Omega = \sum_{j=1}^{n^{quad}} \omega_j f(\mathbf{x}_j) \right)$  of sufficient accuracy to exactly integrate the inner products:

$$(\mathbf{u}, \mathbf{v})_{(\mathbf{H}, \Omega^e)} = \int_{\Omega^e} [\mathbf{u}]^T \mathbf{H} \mathbf{v} d\Omega^e = [\mathbf{u}_e^h]^T \mathbf{W}^e \mathbf{v}_e^h$$

where  $\mathbf{w}_{kl}^e \mathbf{I}$  with  $\mathbf{w}_{kl}^e = \sum_{j=1}^{n^{quad}} \mathbf{H}_e^h N_k^e(\mathbf{x}_j) N_l^e(\mathbf{x}_j) \omega_j$  is the  $(k, l)^{th}$  block of  $\mathbf{W}^e$ .



# Numerical Implementation of Fluid ROM (cont'd)

- AERO-F was used to generate the CFD simulations, using unstructured tetrahedral meshes.
- Piecewise-linear finite elements were used to represent snapshot data and POD modes
- $H$  was taken to be piecewise constant over each element.
- A computer code was written that reads in the snapshot data written by AERO-F, assembles the necessary finite element representation of the snapshots, computes the numerical quadrature for evaluation of inner products, and projects the equations onto the modes.
- ROMs integrated in time using RK-4 scheme with same time step that was used in the CFD computation.



# Numerical Stability & Convergence Tests

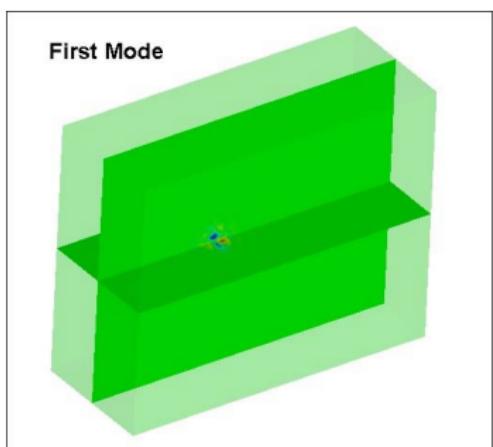
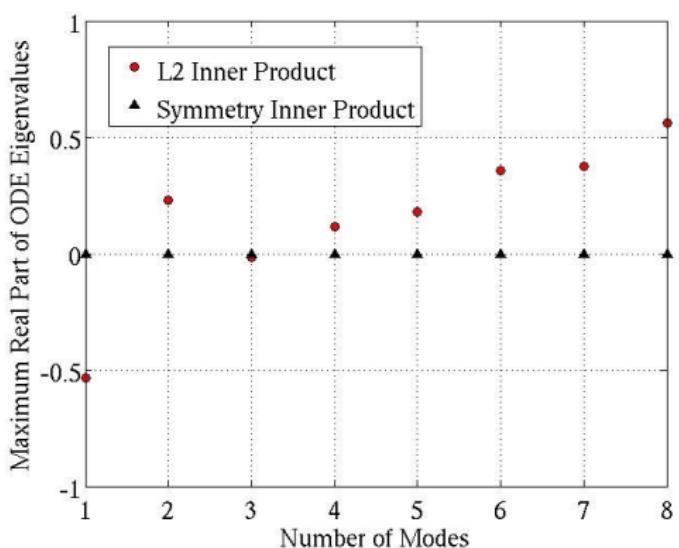
- To test *a posteriori* the **stability** of a ROM dynamical system  $\dot{\mathbf{a}} = \mathbf{K}\mathbf{a}$ , check the Lyapunov condition:

$$\max_i \mathcal{R}\{\lambda_i(\mathbf{K})\} \leq 0?$$

- To test *a posteriori* the **convergence** of a ROM solution  $\mathbf{q}'_M \rightarrow \mathbf{q}'_{CFD}$  as  $M \rightarrow \infty$ , check:

- $(\mathbf{q}'_M, \phi_j)_{(\mathbf{H}, \Omega)} = \left( \sum_{i=1}^M a_i \phi_i, \phi_j \right)_{(\mathbf{H}, \Omega)} = a_j \rightarrow (\mathbf{q}'_{CFD}, \phi_j)_{(\mathbf{H}, \Omega)}$ ?
- $\langle \|\mathbf{q}'_M - \mathbf{q}'_{exact}\|_{(\mathbf{H}, \Omega)} \rangle \rightarrow \langle \|\mathbf{q}'_{CFD} - \mathbf{q}'_{exact}\|_{(\mathbf{H}, \Omega)} \rangle$ ?

# Test Case 1: Purely Random Basis



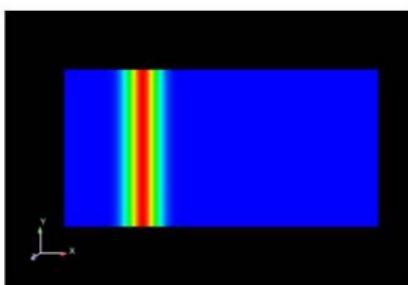
- Uniform base flow: physically stable to any linear disturbance.
- Each mode is a random disturbance field that decays to 0 at the domain boundaries.
- Model problem for modes dominated by numerical error: extreme case of “bad” modes.

# Test Case 2: 1D Acoustic Pressure Pulse

- 1D acoustic pressure pulse prescribed as the initial condition in  $\Omega = (0, 20) \times (-5, 5) \times (0, 1)$ :

$$p'|_{t=0} = -\bar{\rho}\bar{c}e^{-(x-5)^2}, \quad u'_1|_{t=0} = u'_3|_{t=0} = 0$$

CFD animation: pressure



- Uniform base flow,  $M_\infty \equiv \bar{u}/\bar{c} = 0.5$  in the  $x$ -direction (pulse propagates in  $x$ -direction with velocity  $\bar{u} + \bar{c}$ ).
- Slip wall boundary conditions applied on constant  $y$  and  $z$  boundaries.



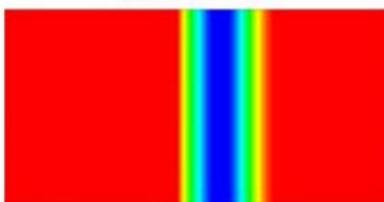
# POD Modes for 1D Acoustic Pressure Pulse Example

- CFD simulation run until  $T_{tot} = 5.25$  (non-dimensional time) using 512 time steps.
- Snapshots taken every 8 time steps ( $N = 64$  snapshots).
- $M = 4$  POD modes captured 85.5% of energy;  $M = 8$  POD modes captured 99.5% of total ensemble energy.

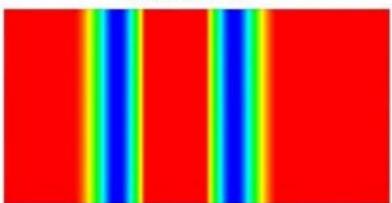
Mode 1



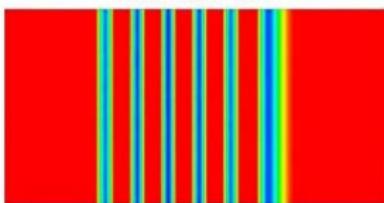
Mode 2



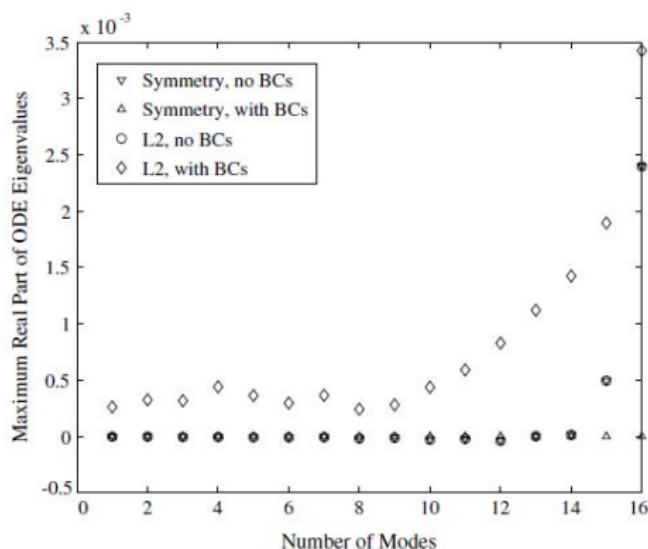
Mode 3



Mode 12



# Stability for 1D Acoustic Pressure Pulse Example

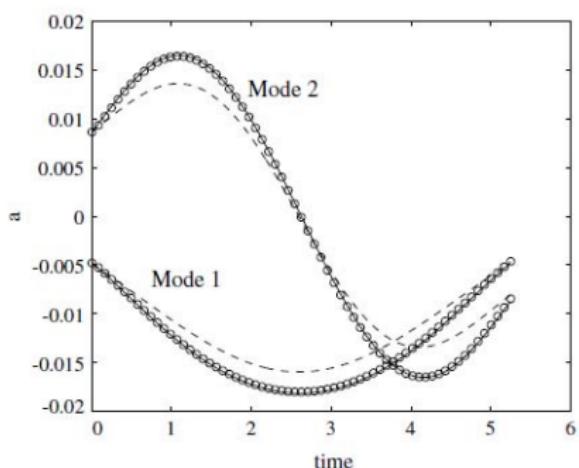


## ■ Four Galerkin schemes:

- 1 Symmetry inner product with BCs.
- 2 Symmetry inner product without BCs.
- 3  $L^2$  inner product with BCs.
- 4  $L^2$  inner product without BCs.

Only the symmetry inner product with BCs produces a stable ROM for all  $M$   
 $(\max_i \mathcal{R}\{\lambda_i(\mathbf{K})\} < 10^{-9})$

# Convergence of the ROM for the 1D Acoustic Pressure Pulse Example



Convergence check:

$$\mathbf{q}'_M = \sum_{i=1}^M a_i(t) \phi_i(\mathbf{x})$$

?

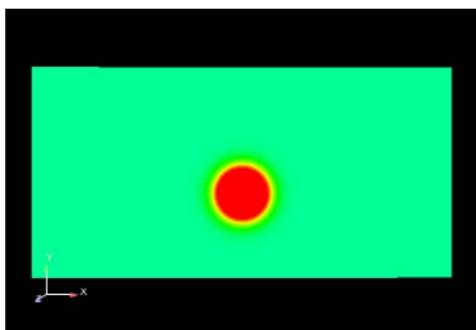
$$\Pi_M \mathbf{q}'_{CFD} = \sum_{i=1}^M (\mathbf{q}'_{CFD}, \phi_i)_{(\mathbf{H}, \Omega)} \phi_i(\mathbf{x})$$

- Figure shows symmetry ROM (with BCs) coefficients  $a_i$  vs.  $(\mathbf{q}'_{CFD}, \phi_i)_{(\mathbf{H}, \Omega)}$  [- - 4 mode ROM; – 8 mode ROM;  $\circ$  CFD solution].
- Symmetry ROM (with BCs) appears to be convergent as the number of modes increases.

# Test Case 3: 2D Pressure Pulse

- Reflection of cylindrical Gaussian pressure pulse in  $\Omega = (0, 20) \times (-5, 5) \times (0, 1)$ :

$$p'|_{t=0} = e^{-(x-10)^2 - (y+1)^2}, \quad u'_1|_{t=0} = u'_2|_{t=0} = u'_3|_{t=0} = 0$$



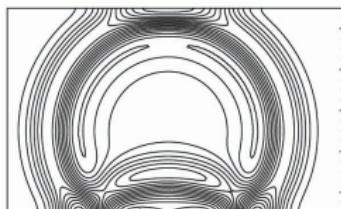
- Uniform base flow,  $M_\infty = 0.25$  in  $x$ -direction.
- Slip wall boundary conditions applied on constant  $y$  and  $z$  boundaries.



# Results for the 2D Pressure Pulse Example

- CFD simulation run until  $T_{tot} = 6.4$  (non-dimensional time) using 624 time steps.
- Snapshots taken every 4 time steps starting at time  $t = t_0 = 0.57$ .
- 6 mode basis captures 97.4% of total ensemble energy.
- Good qualitative agreement between CFD solution and 6 mode symmetry ROM (with BCs) on large scale.
- Excellent agreement between CFD solution and 14 mode symmetry ROM (with BCs).
- Symmetry ROM (with BCs) is stable – vs.  $L^2$  ROM, which experienced instability when more than 6 or 7 modes were used.

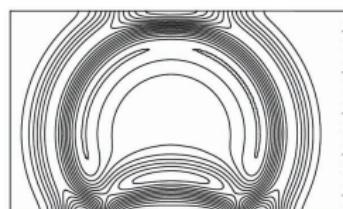
Pressure contours at  $t - t_0 = 5.0$ .



CFD



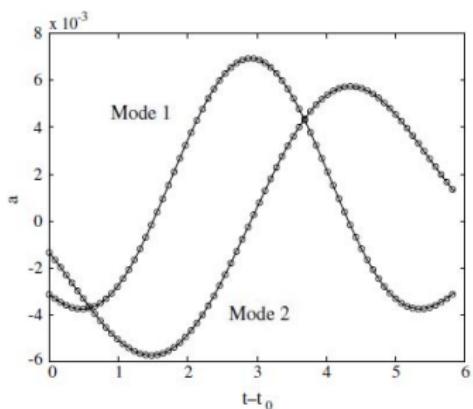
6 mode ROM



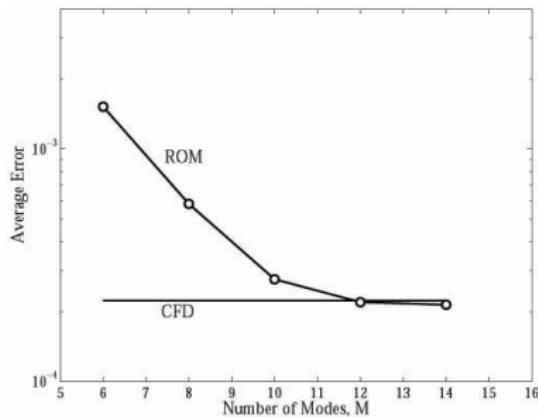
14 mode ROM



# Convergence of the ROM for the 2D Pressure Pulse Example



$a_i$  vs.  $(\mathbf{q}'_{CFD}, \phi_i)_{(\mathbf{H}, \Omega)}$  for  $M = 12$   
(- 12 mode ROM;  $\circ$  CFD solution)



Time-average error of the symmetry ROM solution as a function of  $M$ , compared with the time-average error in the CFD solution

- Tests demonstrate numerically the convergence of the symmetry ROM with BCs.
- For  $M \geq 12$ , ROM gives solution trajectory that is slightly closer to exact solution than the CFD solution.

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# Summary

- A Galerkin ROM in which the *continuous* equations are projected onto the modal basis in a *continuous* inner product is proposed.
- For this *continuous* Galerkin projection approach, the choice of inner product is crucial to stability.
- For linearized, compressible flow, Galerkin projection in the “symmetry” inner product leads to an approximation that is numerically stable for any choice of basis.
- A weak enforcement of the boundary conditions preserves stability, provided they are well-posed.
- A numerical implementation using finite elements that preserves stability is presented.
- Numerical stability of some POD/Galerkin ROMs constructed using this scheme is examined on several model problems.



# Further Work

- A structure ROM governed by the non-linear plate equations was also developed (Segalman *et al.*).
- ROM convergence was examined mathematically, and *a priori* error estimates for the ROM solution error were derived (Kalashnikova & Barone 2010 *in press*).
- Extension of symmetry inner product methods to non-linear equations using an interpolation procedure to handle efficiently the non-linear terms (e.g., “best points interpolation procedure” of Peraire, Nguyen, *et al.*).



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- [1] I. Kalashnikova, M.F. Barone. "On the Stability and Convergence of a Galerkin Reduced Order Model (ROM) of Compressible Flow with Solid Wall and Far-Field Boundary Treatment". *Int. J. Numer. Meth. Engng* (in print).
- [2] M.F. Barone, I. Kalashnikova, M.R. Brake, D.J. Segalman. "Reduced Order Modeling of Fluid/Structure Interaction". *Sandia National Laboratories Report, SAND No. 2009-7189*. Sandia National Laboratories, Albuquerque, NM (2009).
- [3] M.F. Barone, I. Kalashnikova, D.J. Segalman, H. Thornquist. "Stable Galerkin Reduced Order Models for Linearized Compressible Flow". *J. Comput. Phys.* **288** (2009) 1932–1946.
- [4] M.F. Barone, D.J. Segalman, H. Thornquist, I. Kalashnikova. "Galerkin Reduced Order Models for Compressible Flow with Structural Interaction". *AIAA Paper No. 2008–0612 , 46th AIAA Aerospace Science Meeting and Exhibit*, Reno, NV (January 2008).

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Thank You!

# References

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Questions?

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