

Jan Kämper & Florian Börgel

Universität Oldenburg
Semester 2016
11.10.2016

Table of Contents

- 1 Selection of Main Parameters
- 2 Rotor design, BEM
- 3 Control and characteristic curves
- 4 Tower design, modal analysis
- 5 External conditions
- 6 Fatigue load analysis
- 7 Extreme load analysis

Table of Contents

- 1 Selection of Main Parameters
- 2 Rotor design, BEM
- 3 Control and characteristic curves
- 4 Tower design, modal analysis
- 5 External conditions
- 6 Fatigue load analysis
- 7 Extreme load analysis

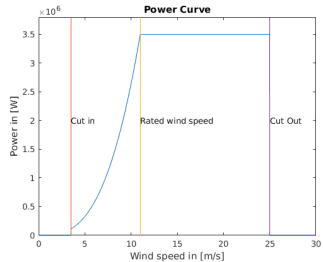
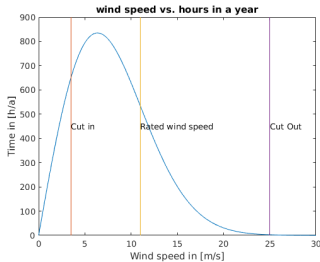
Formulas

$$\begin{aligned}
 & \text{CP}_{\text{ref}} \times \text{Mech Eff} \times \text{Elec Eff} = \text{Tot Eff} \\
 & \text{Rated Pow} \div \text{Tot Eff} = \text{Tot Wind Power} \\
 & \left(\frac{2 * \text{Rated Pow}}{\rho * R_{\text{max}}^2 * \pi * \text{Tot Eff}} \right)^{1/3} = v_{\text{rated}} \\
 & \left(\frac{2 * \text{Rated Pow}}{\rho * \pi * v_{\text{rated}}^3 * \text{Tot Eff}} \right)^{1/2} = R \\
 & R - R_{\text{Hub}} = \text{Blade length} \\
 & \text{Rated Pow} \div R \Pi^2 = \text{Spec Rating} \\
 & \Omega \times R \div v_{\text{rated}} = \lambda \\
 & \Omega \cdot 60 \div R \cdot 2\pi = \text{rrs} \\
 & \text{Blade length} \div 8 = \text{Blade element len}
 \end{aligned}$$

Results

Main parameters	Unit	Value
Calculate total conversion efficiency	m	0.4704
Total wind power that needs to be extracted	kW	7439.258
Rated wind speed (rounded up)	m/s	11
Rotor radius (rounded up)	m	54
Blade length (without hub)	m	52.75
Rotor area (rounded radius)	m^2	9160.884
Specific rating (design)	W/m^2	382.051
λD Design tip speed ratio	-	7.454
Rotor rated speed	rpm	14.5
Blade element length (8 elements, same length)	m	6.593

Estimation of AEP



$$AEP = \sum_v n_v \cdot p_v$$

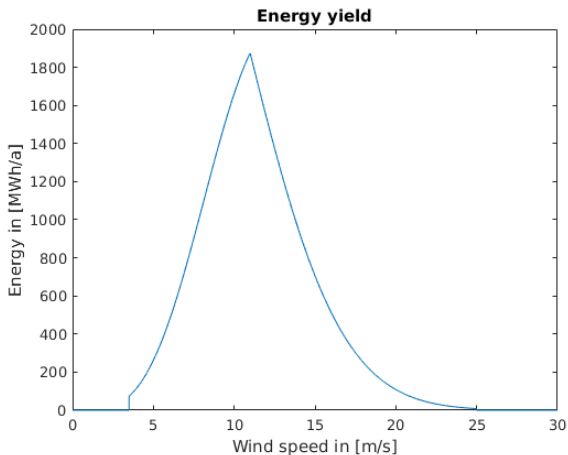
with:

n = number of hours

p = power curve

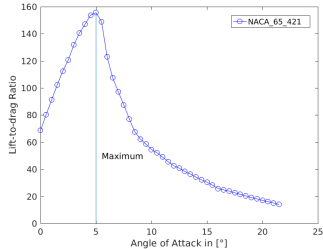
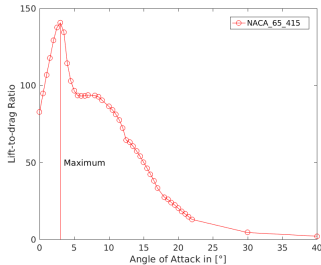
v = wind speed

Resulting Energy yield



$$AEP = 13,49 GWh$$

Lift-to-drag ratio



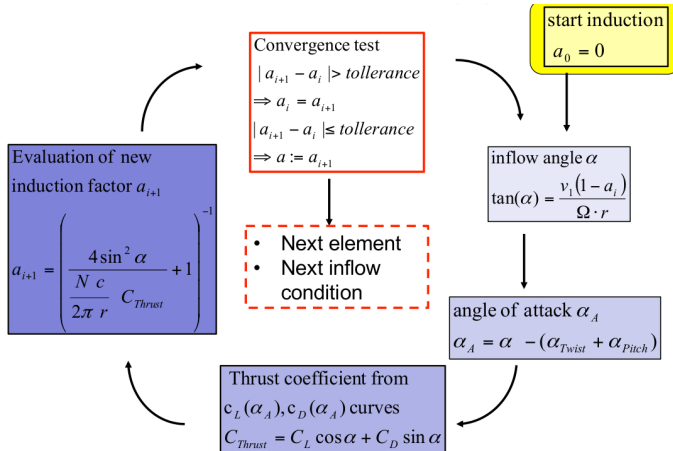
NACA 65-415	α	C_l	C_d	C_m
80% method	10	1.345	0.016	0.071
lift-to-drag method	3.0	0.710	0.005	0.088
NACA 65-421	α	C_l	C_d	C_m
80% method	11	1.255	0.026	0.055
lift-to-drag method	5.0	0.952	0.006	0.092

Table : Main aerodynamic parameters

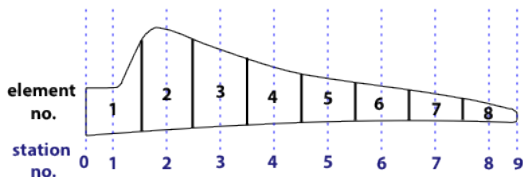
Table of Contents

- 1 Selection of Main Parameters
- 2 Rotor design, BEM
- 3 Control and characteristic curves
- 4 Tower design, modal analysis
- 5 External conditions
- 6 Fatigue load analysis
- 7 Extreme load analysis

BEM algorithm



Blade design



Station	1	2	3	4	5	6	7	8	9
	Cylinder	65-421	65-421	65-421	65-421	65-415	65-415	65-415	65-415
Blade	3.297	9.891	16.484	23.078	29.672	36.266	42.859	49.453	52.750
Chord	6,628	5,426	3,935	3,014	2,425	1,887	1,617	1,413	1,329
Twist	27,590	11,022	3,813	0,055	-2,211	-2,715	-3,783	-4,580	-4,907

Table of Contents

- 1 Selection of Main Parameters
- 2 Rotor design, BEM
- 3 Control and characteristic curves**
- 4 Tower design, modal analysis
- 5 External conditions
- 6 Fatigue load analysis
- 7 Extreme load analysis

Table of Contents

- 1 Selection of Main Parameters
- 2 Rotor design, BEM
- 3 Control and characteristic curves
- 4 Tower design, modal analysis**
- 5 External conditions
- 6 Fatigue load analysis
- 7 Extreme load analysis

Table of Contents

- 1 Selection of Main Parameters
- 2 Rotor design, BEM
- 3 Control and characteristic curves
- 4 Tower design, modal analysis
- 5 External conditions**
- 6 Fatigue load analysis
- 7 Extreme load analysis

Table of Contents

- 1 Selection of Main Parameters
- 2 Rotor design, BEM
- 3 Control and characteristic curves
- 4 Tower design, modal analysis
- 5 External conditions
- 6 Fatigue load analysis**
- 7 Extreme load analysis

Table of Contents

- 1 Selection of Main Parameters
- 2 Rotor design, BEM
- 3 Control and characteristic curves
- 4 Tower design, modal analysis
- 5 External conditions
- 6 Fatigue load analysis
- 7 Extreme load analysis**

Thanks!