
ECS 323: Control Systems

Planar VTOL System Design Study

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1. Design Study Description

In this design study we need to design a control system for the given planar VTOL system with the parameters:

- $M_c = 2 \text{ kg}$
- $J_c = 0.009 \text{ kg m}^2$
- $m_l = 0.3 \text{ kg}$
- $m_r = 0.3 \text{ kg}$
- $d = 0.28 \text{ m}$
- $\mu = 0.21 \text{ kg s}^{-1}$

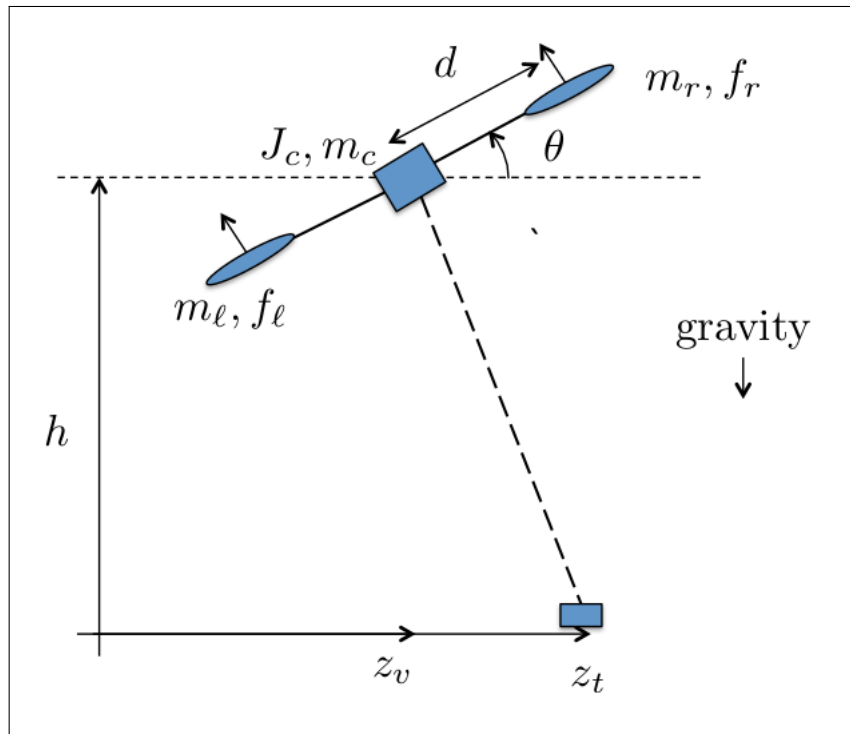


Figure 1: Planar VTOL System

2. Kinetic Energy

The postions of the various components of the VTOL are given by:

$$\begin{aligned}\mathbf{p}_c &= (z_v, h) \\ \mathbf{p}_l &= (z_v - d \cos \theta, h - d \sin \theta) \\ \mathbf{p}_r &= (z_v + d \cos \theta, h + d \sin \theta)\end{aligned}$$

So, the velocities can be written as:

$$\begin{aligned}\mathbf{v}_c &= (\dot{z}_v, \dot{h}) \\ \mathbf{v}_l &= (\dot{z}_v + d\dot{\theta} \sin \theta, \dot{h} - d\dot{\theta} \cos \theta) \\ \mathbf{v}_r &= (\dot{z}_v - d\dot{\theta} \sin \theta, \dot{h} + d\dot{\theta} \cos \theta)\end{aligned}$$

Kinetic energy of the centerpod is given by:

$$K_{pod} = \frac{1}{2}m_c \mathbf{v}_c^T \mathbf{v}_c + \frac{1}{2}\boldsymbol{\omega}_c^T J_c \boldsymbol{\omega}_c = \frac{1}{2}m_c(\dot{z}_v^2 + \dot{h}^2) + \frac{1}{2}J_c \dot{\theta}^2 \quad (1)$$

Kinetic energy of the left and right rotors is given by:

$$\begin{aligned}K_{rotors} &= \frac{1}{2}m_l \mathbf{v}_l^T \mathbf{v}_l + \frac{1}{2}m_r \mathbf{v}_r^T \mathbf{v}_r \\ &= \frac{1}{2}m_l(\dot{z}_v + d\dot{\theta} \sin \theta)^2 + \frac{1}{2}m_l(\dot{h} - d\dot{\theta} \cos \theta)^2 \\ &\quad + \frac{1}{2}m_r(\dot{z}_v - d\dot{\theta} \sin \theta)^2 + \frac{1}{2}m_r(\dot{h} + d\dot{\theta} \cos \theta)^2 \\ &= \frac{1}{2}(m_l + m_r)(\dot{z}_v^2 + \dot{h}^2) + \frac{1}{2}(m_l + m_r)d^2\dot{\theta}^2 \\ &\quad + (m_l - m_r)(\dot{z}_v \sin \theta - \dot{h} \cos \theta)d\dot{\theta}\end{aligned} \quad (2)$$

Now, the total kinetic energy of the VTOL will be given by the sum of 1 and 2:

$$K_V = K_{pod} + K_{rotors} \quad (3)$$