

Poznań University of Technology

DOCTORAL THESIS

Noise analysis of NASA R67 axial compressor blade with use of CFD tools

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A thesis submitted in fulfilment of the requirements for the degree of Doctor of Philosophy. Engineer.

in the

Faculty of Work Machines and Transportation Chair of Thermal Engineering

Declaration of Authorship

I, MSc. Eng. Jędrzej Mosiężny, declare that this thesis titled, 'Noise analysis of NASA R67 axial compressor blade with use of CFD tools' and the work presented in it are my own. I confirm that:

- This work was done wholly or mainly while in candidature for a research degree at this University.
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- Where I have consulted the published work of others, this is always clearly attributed.
- Where I have quoted from the work of others, the source is always given. With the exception of such quotations, this thesis is entirely my own work.
- I have acknowledged all main sources of help.
- Where the thesis is based on work done by myself jointly with others, I have made clear exactly what was done by others and what I have contributed myself.

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Abstract

This thesis proposes a simplified method of assessing flow generated noise in transonic flows in any point of computational domain without use of Ffowcs-Williams Hawkings and other aeroacoustic analogies.

First a steady state Reynolds Averaged Navier-Stokes analysis of NASA R67 transonic axial compressor is performed as a validation study of the mesh and numerical setup. The result of the steady state analysis is then used as an initialization for transient DDES analysis performed on high quality, 11 million cells mesh. The transient analysis cover 0.05s of flow time, which corresponds to c.a 800 revolutions of the rotor. Both steady state and transient simulations are performed on PL-Grid HPC infrastructure.

Transient results are analyzed with in-house build script written in Python 3.6.x, with use of Dask, Pandas and Numpy libraries. Noise analysis script uses information about transient static pressure and transient particle velocity and vorticity gathered from CFD analysis to generate information on noise pressure levels, noise frequency, effective noise power. The advantage of the script is that the only limitation of therein are: the quality of the input data and computational power of workstation running the script

Acknowledgements

In this place I would like to thank the Chair of Thermal Engineering of Poznań University of Technology, with special recognition to MSc. Eng. Bartosz Ziegler and PhD Eng. Przemysław Grzymisławski for thorough scientific and personal support.

I would also like to thank my beloved wife, Inez Mosiężna, for limitless patience and countless support during this project.

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Abbreviations

LAH List Abbreviations Here

Physical Constants

Speed of Light $c = 2.997 924 58 \times 10^8 \text{ ms}^{-8} \text{ (exact)}$

Symbols

a distance m

P power W (Js⁻¹)

 ω angular frequency rads⁻¹

To my wife. For limitless patience...

Introduction

1.1 Main Section 1

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1.2 Main Section 2

Background

2.1 Main Section 1

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2.2 Main Section 2

Approach

3.1 Main Section 1

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3.2 Main Section 2

CFD Analysis of NASA R67 rotor

4.1 Main Section 1

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4.2 Main Section 2

Flowfield noise analysis

5.1 Main Section 1

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5.1.2 Subsection 2

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5.2 Main Section 2

Results of flowfield noise analysis

6.1 Main Section 1

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6.2 Main Section 2

Conclusions & Further work

7.1 Main Section 1

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7.1.2 Subsection 2

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7.2 Main Section 2

Appendix A

Noise analysis code

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