

Poznań University of Technology

DOCTORAL THESIS

Noise analysis of NASA R67 axial compressor blade with use of CFD tools

Author:

MSc. Eng. Jędrzej Mosiężny

Supervisor:

Prof. DSc. Eng. Michał CIAŁKOWSKI

A thesis submitted in fulfilment of the requirements for the degree of Doctor of Philosophy. Engineer.

in the

Faculty of Work Machines and Transportation Chair of Thermal Engineering

Declaration of Authorship

I, MSc. Eng. Jędrzej Mosiężny, declare that this thesis titled, 'Noise analysis of NASA R67 axial compressor blade with use of CFD tools' and the work presented in it are my own. I confirm that:

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- I have acknowledged all main sources of help.
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Abstract

This thesis proposes a method of assessing flow generated noise in transonic flows by direct formulation.

First a steady state Reynolds Averaged Navier-Stokes analysis of NASA R67 transonic axial compressor is performed as a validation study of the mesh and numerical setup. The result of the steady state analysis is then used as an initialization for transient DDES analysis performed on high quality, 11 million cells hexagonal mesh. The transient analysis covers 0.05s of physical flow time, which corresponds to about 800 revolutions of the rotor. Both steady state and transient simulations are performed on PL-Grid HPC infrastructure.

Transient results are analyzed with an in-house build program. The program uses information about static pressure, transient particle velocity and vorticity from each timestep. This data is then postprocessed into sound pressure levels, sound frequency and effective sound power level.

Information on generation of sound phenomena occurring in the blade passage are gathered from direct formulation and may be used as a validation case for FW-H or other computational aeroacoustic analogies dealing with flows in transonic regimes in rotating machinery.

Acknowledgements

In this place I would like to thank the Chair of Thermal Engineering of Poznań University of Technology, with special recognition to MSc. Eng. Bartosz Ziegler and PhD Eng. Przemysław Grzymisławski for thorough scientific and personal support during this project.

A big recognition goes to the owners and maintainers of the PLGRID - Polish HPC infrastructure, especially team in HPC Cyfronet center in AGH University of Science and Technology in Kraków. Being able to use the state of the art HPC clusters for analyses made this project possible.

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Abbreviations

CAA Computional Aero Acoustics

CFD Computional Fluid Dynamics

DDES Delayed Detached Eddy Simulation

DES Detached Eddy Simulation

HPC Hight Power Computing

LES Large Eddy Simulation

N-S Navier Stokes

SRS Scale Rresolving Simulation

Physical Constants

Speed of Light $c = 2.997 924 58 \times 10^8 \text{ ms}^{-8} \text{ (exact)}$

Symbols

a distance m

P power W (Js⁻¹)

 ω angular frequency rads⁻¹

To my wife. For limitless patience...

Introduction

1.1 Main Section 1

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1.2 Main Section 2

Background

2.1 Main Section 1

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2.2 Main Section 2

Approach

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3.2 Main Section 2

CFD Analysis of NASA R67 rotor

4.1 Main Section 1

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4.2 Main Section 2

Flowfield noise analysis

5.1 Main Section 1

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5.2 Main Section 2

Results of flowfield noise analysis

6.1 Main Section 1

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6.2 Main Section 2

Conclusions & Further work

7.1 Main Section 1

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7.2 Main Section 2

Appendix A

Noise analysis code

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