Worksheet

2. Simplified aircraft model. Transfer function =

num = den =

Controller transfer function =

$$k = D =$$

Phase margin =

Amount of extra time delay which can be tolerated =

2.1. PIO. Period of oscillation (observed) =

Period of oscillation (theoretical) =

2.2. Sinusoidal disturbances.

Maximum stabilising gain =

0 0

Phase at 0.66 Hz =

Open loop T.F. $(y \rightarrow d) =$

Closed loop T.F. $(y \rightarrow d) =$

2.3. Fastest pole. T =

Gain at 0.66 Hz =

3. Autopilot. Proportional gain $K_c =$

Period of oscillation $T_c =$

3.1 Transfer function of PID controller =

PID constants: $K_p = T_i = T_d =$

Final value of $T_d =$