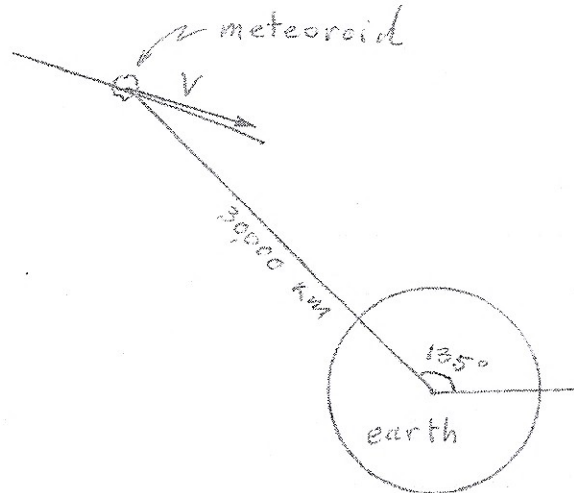


\* Only original handwritten notes and homeworks are allowed. Photocopied notes and homework solution sheets are not permitted. Except for a hand calculator, no cell phone or electronic equipment of any kind is allowed.

Show all work and give units in final answers.

- [50] 1. A satellite in earth orbit has a specific angular momentum of  $70,000 \text{ km}^2/\text{sec}$  and specific energy of  $-15 \text{ km}^2/\text{sec}^2$ .
- Calculate the parameter  $p$ , the semi-major axis, and the eccentricity of the orbit. [15 points]
  - Calculate the period of the orbit. [5 points]
  - Calculate the apogee and perigee altitudes. [10 points]
  - Calculate the maximum and minimum velocities. [10 points]
  - Calculate the value(s) of the true anomaly where the altitude is midway between the apogee and perigee altitudes. [10 points]

- [50] 2. A meteoroid is first observed approaching the earth when it is  $30,000 \text{ km}$  from the center of the earth with a true anomaly of  $135^\circ$ , as shown in the figure below. If the geocentric speed of the meteoroid  $V$  is  $5.154932912592882 \text{ km/sec}$  when it is first observed, determine:
- Whether the trajectory is elliptic, parabolic or hyperbolic. [10 points]
  - Whether the meteoroid flies by or strikes the earth. [10 points]
  - The time until the meteoroid reaches its closest approach to the earth or strikes the earth. [20 points]
  - Draw the trajectory of the meteoroid on the hodographic plane and label the point where the meteoroid is first observed and the point referred to in part (c). [10 points]



Physical constants

The Earth

Mean Radius =  $6368 \text{ km}$

$$\mu_{\text{earth}} = 3.986 \times 10^5 \text{ km}^3/\text{sec}^2$$

1)  $h = 70,000 \text{ km}^2/\text{sec}$      $e = -15 \text{ km}^2/\text{sec}^2$

a)  $p = \frac{h^2}{\mu} = \frac{(70,000)^2}{3.986 \times 10^5} = \underline{\underline{12,293 \text{ km}}} \quad [5]$

$e = -\frac{\mu}{2a} \Rightarrow a = -\frac{\mu}{2e} = -\frac{3.986 \times 10^5}{2(-15)} = \underline{\underline{13,287 \text{ km}}} \quad [5]$

$p = a(1-e^2) \Rightarrow e = \sqrt{1 - \frac{p}{a}} = \sqrt{1 - \frac{12,293}{13,287}} = \underline{\underline{0.273514}} \quad [5]$

b)  $T = 2\pi \sqrt{\frac{a^3}{\mu}} = 2\pi \sqrt{\frac{(13,287)^3}{3.986 \times 10^5}} = \underline{\underline{15,242 \text{ sec}}} = \underline{\underline{4.23 \text{ hrs}}} \quad [5]$

c)  $r = \frac{p}{1+e \cos \theta}$

$r_p = r|_{\theta=0} = \frac{p}{1+e} = \frac{12,293}{1+0.273514} = 9653 \text{ km}$

$H_p = r_p - r_e = 9653 - 6368 = \underline{\underline{3285 \text{ km}}} \quad [5]$

$r_a = r|_{\theta=\pi} = \frac{p}{1-e} = \frac{12,293}{1-0.273514} = 16,921 \text{ km}$

$H_a = r_a - r_e = 16,921 - 6368 = \underline{\underline{10,553 \text{ km}}} \quad [5]$

d)  $h = r_p V_p = r_a V_a$

$V_p = \frac{h}{r_p} = \frac{70,000}{9653} = \underline{\underline{7.25163 \text{ km/sec}}} \quad [5]$

$V_a = \frac{h}{r_a} = \frac{70,000}{16,921} = \underline{\underline{4.13687 \text{ km/sec}}} \quad [5]$

$$e) \text{ For } H = \frac{H_p + H_a}{2} = \frac{3285 + 10,553}{2} = 6919 \text{ km}$$

$$r = H + r_e = 6919 + 6368 = 13,287 \text{ km} \quad [4]$$

$$r = \frac{P}{1 + e \cos \theta} \Rightarrow \cos \theta = \frac{\frac{P}{r} - 1}{e} = \frac{\frac{12,293}{13,287} - 1}{0.273514} = -0.273514$$

$$\theta = 106^\circ, 254^\circ$$

[3] [3]

$$2) \ a) \ \epsilon = \frac{V^2}{2} - \frac{\mu}{r} = \frac{(5.154932912592882)^2}{2} - \frac{3.986 \times 10^5}{30,000} = \underline{\underline{0}} \quad [5]$$

$\therefore$  Trajectory is parabolic [5]

b) For parabolic trajectory

$$r = \frac{P}{1 + \cos \theta} \Rightarrow p = r(1 + \cos \theta) = 30,000(1 + \cos 135^\circ) = 8786.8 \text{ km}$$

At perigee  $\theta = 0$

$$r_p = \frac{P}{2} = \frac{8786.8}{2} = 4393.4 \text{ km} < r_e = 6368 \text{ km} \quad [5]$$

$\therefore$  Meteoroid will collide [5]

$$c) \quad r = \frac{P}{1 + \cos \theta}$$

At impact

$$\cos \theta_2 = \frac{P}{r_2} - 1 = \frac{8786.8}{6368} - 1 = 0.379837$$

$$\theta_2 = 1.18118 \text{ rad} = 67.6764^\circ \quad [5]$$

$$D_1 = \sqrt{P} \tan \frac{\theta_1}{2} = \sqrt{8786.8} \tan \left( \frac{135^\circ}{2} \right) = 226.303 \text{ km}^{1/2} \quad [3]$$

$$D_2 = \sqrt{P} \tan \frac{\theta_2}{2} = \sqrt{8786.8} \tan \left( \frac{67.6764^\circ}{2} \right) = 62.8429 \text{ km}^{1/2} \quad [3]$$

$$t_1 = \frac{1}{2\sqrt{\mu}} \left( P D_1 + \frac{D_1^3}{3} \right) = \frac{1}{2\sqrt{3.986 \times 10^5}} \left[ (8786.8)(226.303) + \frac{(226.303)^3}{3} \right]$$

$$= 4634.29 \text{ sec} \quad [3]$$

$$t_2 = \frac{1}{2\sqrt{\mu}} \left( P D_2 + \frac{D_2^3}{3} \right) = \frac{1}{2\sqrt{3.986 \times 10^5}} \left[ (8786.8)(62.8429) + \frac{(62.8429)^3}{3} \right]$$

$$= 502.79 \text{ sec} \quad [3]$$

$$t_1 - t_2 = 4634 - 503 = \underline{\underline{4131 \text{ sec}}} = \underline{\underline{1.148 \text{ hr}}} \quad [3]$$

