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- 1) Two satellites are in coplanar circular orbits around the earth. Their orbit radii are r_1 , and r_2 . How long is it before they are separated by 90° if their radius vectors are initially coincident?
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- 2) A communications satellite launched from Cape Canaveral is placed into a 160 nautical mile high circular parking orbit inclined at 28.5° to the equator.
 - a) As the vehicle crosses the equator, the upper stage is reignited to place the satellite into a highly elliptic 19,432 x 160 nautical mile transfer orbit. Determine the velocity increment needed for this maneuver.
 - b) After a final checkout of the satellite, the apogee kick motor is fired on the third apogee to place the satellite into geosynchronous orbit. Determine the velocity increment needed for this maneuver.
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- 3) Prussing and Conway, problem 7.11
 - a) Determine the cost ($\Delta V/V_0$) of a bi-elliptic transfer between circular orbits of equal radius in which the final orbit is inclined by 50° to the initial orbit and in which the radius at which the intermediate impulse is performed is twice that of the initial circular orbit.
 - b) Determine the flight time in units of the period of the initial circular orbit.
 - c) Determine the cost ($\Delta V/V_0$) of the same orbit plane change accomplished by a single impulsive maneuver.
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- 4) The General Mission Analysis Tool (GMAT) is a powerful software developed by NASA and private industry for simulating spacecraft orbits and trajectories. A free version of the software is available which you can download from <https://sourceforge.net/projects/gmat/> and install on your home computer. The software is also available on all computers in the ST-213 and ST-226 Computer Labs.

Start the software and on the Welcome page, as a start, view the video tutorial “Training:Part 3 Tutorial 1” to help you get familiarized with the software. A written tutorial “Simulating an Orbit” can be found at <https://documentation.help/GMAT/> Subheadings: Tutorials > Simulating an Orbit.

Use GMAT to determine the ground track of a satellite in a geosynchronous orbit inclined at 30° with respect to the equator. Run the animation for a 24 hour period and observe the ground track.