Derivation of Four Relations Needed in Using the Patched Conic Method

Prussing a Conway, prob. 1.14 In terms of Vs, which is circular orbit speed at the surface of a planet of vadius Vs, and Vp, which is periapse radius of a hyperbolic orbit about the planet, show that

a)
$$\mu = V_5 V_5^2$$

b)
$$e = 1 + \psi$$
 where $\psi = \left(\frac{V_{ab}}{V_{s}}\right)^{2} \left(\frac{V_{P}}{V_{s}}\right)^{2}$

(c)
$$\sin\left(\frac{d}{z}\right) = \frac{1}{1+\psi}$$

a) For a circular orbit Vc = Jru

m= Ve Ve

on the surface of a planet va= Vs, Va= Vs

b) The energy integral for a hyperbolic trajectory is

$$\frac{\sqrt{2}}{Z} - \frac{\mu}{V} = \frac{\mu}{Za}$$

(Vo is called the hyperbolic excess speed) As V-> 00, V= Vao

$$\frac{V_{\infty}^2}{2} = \frac{\mu}{2a}$$

OV

$$a = \frac{\mu}{V_{\infty}^2} \qquad (2)$$

$$\alpha = V_s \left(\frac{V_s}{V_{\infty}}\right)^2$$
 (3)

Also for a hyperbola

$$V = \frac{a(e^2 - 1)}{1 + e \cos \theta}$$
 (4)

$$V_p = \frac{a(e^2 - 1)}{1 + e} = a(e - 1)$$
 (5)

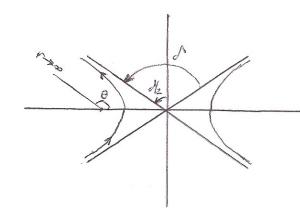
ov

$$e = 1 + \frac{v_e}{a}$$
 (6)

$$e = 1 + \frac{V_P}{V_S} \left(\frac{V_{\infty}}{V_S} \right)^2$$

or

$$e=1+\psi$$
 (7) where $\psi=\left(\frac{V_{\infty}}{V_{5}}\right)^{2}\left(\frac{V_{P}}{V_{5}}\right)$ (8)



of is called the turn angle

From the above figure, as v > 0, 0 > =+ &

From (A) as V -> 20

1+ecos0 = 0

1+ecos (=+=)=0

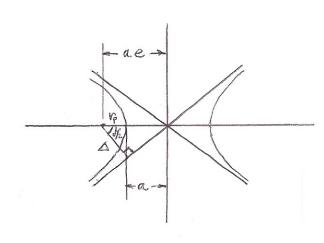
1-esin() = 0

 $Sin\left(\frac{d}{2}\right) = \frac{1}{e}$ (9)

Sub. (7) into (9)

 $\sin\left(\frac{\delta}{2}\right) = \frac{1}{1+\psi}$ (10)

d)



aiming values

From the above figure
$$\cos\left(\frac{d}{2}\right) = \frac{D}{dE}$$

$$\Delta = ae\cos\left(\frac{d}{2}\right) = ae\sqrt{1-sin^2\left(\frac{d}{2}\right)}$$
 (11)

Solve (5) for a

Sub. (10) & (12) into (11)

$$\Delta = \frac{V_{p}}{e-1} e \sqrt{1 - \frac{1}{(1+\psi)^{2}}} = V_{p} \frac{e}{e-1} \sqrt{\frac{(1+\psi)^{2}-1}{(1+\psi)^{2}}}$$

$$= V_{p} \frac{e}{e-1} \frac{1}{1+\psi} \sqrt{\frac{\psi(\psi+z)}{(1+\psi)^{2}}} \qquad (13)$$

QV