

Chapter 1

Introduction

In this thesis, we focus on a specific aspect of theoretical high-energy physics and collider physics: the resummation of QCD large logarithms for the Thrust event-shape distribution in electron-positron (e^+e^-) collisions.

High energy physics, often synonymous of particle physics, and collider physics are crucial because they explore the most basic constituents of matter and help us understand three of the four fundamental forces of nature: the strong, weak, and electromagnetic forces. Collider experiments test the predictions of the Standard Model of particle physics, the most successful theory of particle interactions to date. Through high precision theoretical prediction and experimental measurements, it is possible to test the limits of the Standard Model, search for new physics beyond it and gain a better understanding of the fundamental forces.

One key parameter of the Standard Model is the strong coupling constant α_s , which measures the strength of the strong force. Precise measurements of α_s are crucial for accurate predictions in Quantum Chromodynamics (QCD), the theory of the strong force, as all calculations in QCD depend on it.

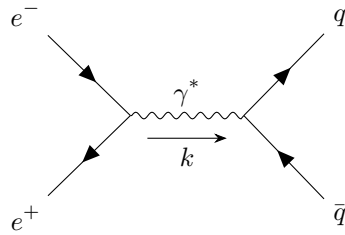


Figure 1: Tree-level Feynman diagram of electron-positron annihilation producing a virtual photon that decays into a quark-antiquark pair.

One example of a process studied in high-energy physics is electron-positron annihilation, as shown in fig. 1. Electron-positron annihilation has been extensively studied, particularly during the operation of the Large Electron-Positron Collider (LEP) at CERN from 1989 to 2000. LEP provided an optimal environment for precision studies in high-energy physics. Unlike hadron colliders, which are complicated by strongly interacting initial states, LEP enabled extremely accurate measurements of Standard Model quantities such as the Z-boson mass. These results tightly constrain beyond-the-Standard Model physics. Precision data from LEP is also used in Quantum Chromodynamics (QCD) studies to determine the strong coupling constant, α_s .

When high-energy particles collide, quarks and gluons are produced in the interactions. Due to a phenomenon called *color confinement*, these quarks and gluons cannot exist freely and thus hadronize, forming jets of particles. A jet is a collimated stream of hadrons (such as protons, pions, and kaons) that originates from the hadronization of a single quark or gluon.

In electron-positron annihilation, the electron and positron interact electromagnetically through the exchange of a virtual photon, which mediates the electromagnetic interaction, virtual refers to the fact that the photon is off-shell, meaning it does not satisfy the mass-shell condition $E = pc$ for a real photon with mass $m = 0$. This virtual photon can then decay into a quark-antiquark pair in another electromagnetic process, conserving the electrical and color charge of the initial state, as well as the four-momentum.

The quark-antiquark pair produced interacts through both the strong force and the electromagnetic force, as they possess both electric charge and color charge. They can radiate gluons via the strong force and photons via the electromagnetic force. This radiation process continues, creating a cascade of particles known as a *parton shower*. The parton shower eventually hadronizes into jets when the particles are no longer energetic enough to radiate further, and the final state particles can be revealed by the detectors, with their momenta measured to study the underlying processes.

As is typical in physics, the equations governing these interactions are highly complex so that finding exact solutions is nearly impossible. Therefore, functions of interest are often expanded perturbatively, meaning they are expressed as a power series in a small parameter.

For the electromagnetic interaction, this small parameter is the fine structure constant (or electromagnetic coupling constant) $\alpha_{em} \sim \frac{1}{137}$.

For interactions involving the strong force, it is natural to use the aforementioned strong coupling constant, α_s . This key parameter becomes small at high energies (or equivalently, short distances) due to the phenomenon known as *asymptotic freedom* of QCD.

Extraction of α_s can be achieved from comparing precise QCD prediction such as the thrust event-shape distribution against experimental data.

1.1 Thrust variable

Thrust T is defined as:

$$T = \max_{\vec{n}} \frac{\sum_i |\vec{p}_i \cdot \vec{n}|}{\sum_i |\vec{p}_i|} \stackrel{\text{def}}{=} 1 - \tau, \quad (1)$$

where the sum is over all final state particles and \vec{n} is a unit vector that points in the direction which maximizes the magnitude of T .

In practice, the sum may be carried over the detected particles only. The thrust distribution represents the probability of observing a given value of T in e^+e^- annihilation, *i.e* the probability of observing a configuration of momenta in the final state whose thrust calculated with eq. (1) is T .

It can be seen from this definition that the thrust is an infrared and collinear safe quantity, that is, it is insensitive to the emission of zero momentum particles and to the splitting of one particle into two collinear ones.

In fact, contribution from soft particles with $\vec{p}_i \rightarrow 0$ drop out, and collinear splitting of a parton with momenta \vec{p} into two partons of momenta $(1 - \lambda)\vec{p}$ and $\lambda\vec{p}$ does not change the thrust:

$$\begin{aligned} |(1 - \lambda)\vec{p}_i \cdot \vec{n}| + |\lambda\vec{p}_i \cdot \vec{n}| &= (1 - \lambda)|\vec{p}_i \cdot \vec{n}| + \lambda|\vec{p}_i \cdot \vec{n}| = |\vec{p}_i \cdot \vec{n}|, \\ |(1 - \lambda)\vec{p}_i| + |\lambda\vec{p}_i| &= (1 - \lambda)|\vec{p}_i| + \lambda|\vec{p}_i| = |\vec{p}_i|. \end{aligned}$$

Formally, infrared-safe observables are the one which do not distinguish between (n+1)-partons and n-partons in the soft/collinear limit, *i.e*, are insensitive to what happens at long-distance (non-perturbative) scales.

Infrared safe observables are important in the context of perturbative QCD, because they allow for a meaningful comparison between theory and experiment.

A significant challenge in achieving precise theoretical predictions from QCD lies in the complexity of the relevant fixed-order calculations. While the next-to-leading-order (NLO) results for event shapes have been known since 1980 **Ellis:1980wv**, the relevant next-to-next-leading order (NNLO) calculations were completed only in 2007 **Gehrmann-DeRidder:2007nzq**.

From fig. 2, it is evident that the NNLO prediction agrees well with the data, except in the region near $T = 0.5$ (spherical final state) and $T = 1$ (pencil-like final state).

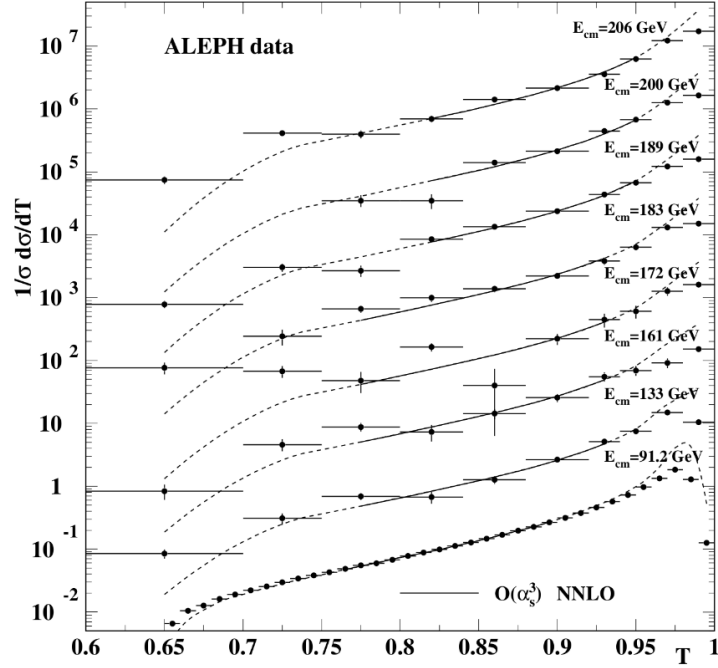


Figure 2: Distributions measured by ALEPH, after correction for backgrounds and detector effects of thrust at energies between 91.2 and 206 GeV together with NNLO QCD predictions. The error bars correspond to statistical uncertainties. The plotted distributions are scaled by arbitrary factors for presentation. Image taken from **Dissertori**’2008.

This is because thrust distribution for $T \simeq 1$ is dominated by two-jet configurations, *i.e.* the final state particles consist of two partons emitted back-to-back like fig. 1. In contrast, the tail of the distribution near $T = 0.5$ is dominated by multijet final states.

To improve the agreement between theory and experiment, we’ll need higher fixed order calculations (challenging task), but this will only improve the agreement in the tail region, to also improve the agreement in the dijet region we need to use resummation techniques.

1.1.1 Thrust distribution

The cross section is defined as the probability of observing a final state with a given thrust value τ and Thrust distribution is expressed in three ways: as a differential cross section, as a cumulative distribution and as its Laplace transform.

$$\sigma(\tau) = \frac{1}{\sigma_0} \frac{d\sigma}{d\tau}, \quad (2)$$

$$R_T(\tau) = \int_0^\tau d\tau' \frac{1}{\sigma_0} \frac{d\sigma}{d\tau'}, \quad (3)$$

$$\tilde{\sigma}(\nu) = \int_0^\infty d\tau e^{-\nu\tau} \frac{d\sigma}{d\tau}. \quad (4)$$

It can be seen that a two-particle final state has fixed $T = 1$ because of momentum conservation, in fact at the zeroth order of the fixed order this corresponds to a delta distribution for the differential cross section, as shown in eq. (7), consequently the thrust distribution receives its first non-trivial contribution from three-particle final states.

The upper limit of τ or lower limit on T depends on the number of final-state partons. Knowing the lower limit of T is important because it allows us to normalize correctly the fixed-order cross section. Neglecting masses, for three particles, $\tau_{max} = 1/3$, corresponding to a symmetric trigonal-planar configuration (Figure 3). For four particles the minimum thrust corresponds to final-state momenta forming the vertices of a regular tetrahedron (Figure 4), each making an angle $\cos^{-1}(1/\sqrt{3})$ with respect to the thrust axis. Thus $\tau_{max} = 1 - \frac{1}{\sqrt{3}} \approx 0.42265$ in this case. For more than four particles, as far as we know there's no known value of τ_{max} in literature, it's only known that τ_{max} approaches $1/2$ from below as the number of particles increases. For $N = 5$ the maximum value of τ was previously estimated to be $\tau_{max} = 0.4275$ by checking at what bin the cross sections is 0 for NNLO Monte Carlo (MC) calculation of the fixed-order cross sections **Monni:2011gb**. We also have another estimate of τ_{max} for $N = 5$ from the tables in **Weinzierl'2009** of NNLO thrust cross section whose bins end at $\tau = 0.445$. The numerical NNLO calculation of the thrust distribution can only give an estimate to the upper limit of τ , limited by the number of events used in the MC calculation, the smallness of the cross section in the kinematical limit and some small, but non zero, resolution parameters which are always present in numerical calculation at higher-orders

We have used a stochastic optimization algorithm to find the lower limit of T for $N = 5$ and found a configuration which gives $\tau_{max} = 0.4539536$ for $N = 5$ (Figure 5).

Finding the lower limit of T is non trivial as it involves a double optimization problem. Given a configuration of N parton momenta one needs to find the direction of the thrust axis which maximizes the value of eq. (1) and then find the minimum value of T by varying the momenta of the partons and repeating the process of finding the thrust axis.

Minimum kinematical configuration of Thrust for 3 partons $T_{\min} = 0.6666667$ $\tau_{\max} = 0.3333333$

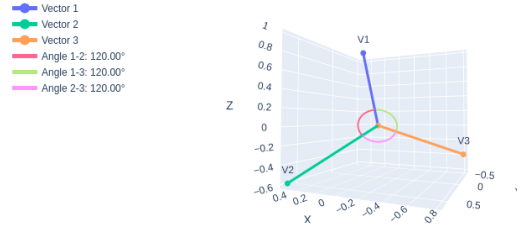


Figure 3: Configuration of final state momenta of three partons that maximizes τ , forming a trigonal-planar configuration.

Minimum kinematical configuration of Thrust for 4 partons $T_{\min} = 0.5773516$ $\tau_{\max} = 0.4226484$

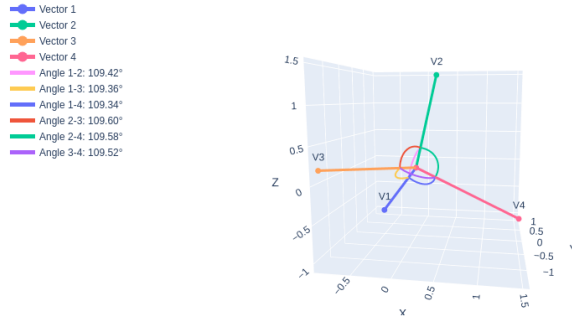


Figure 4: Configuration of final state momenta of four partons that maximizes τ , forming a tetrahedral configuration.

Minimum kinematical configuration of Thrust for 5 partons $T_{\min} = 0.5460464$ $\tau_{\max} = 0.4539536$

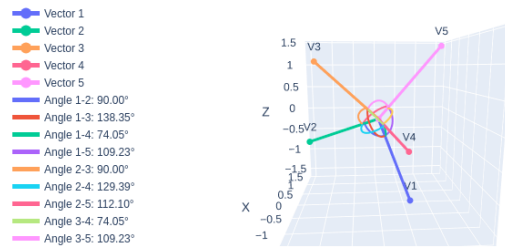


Figure 5: Configuration of final state momenta of five partons that maximizes τ .

Table 1: The maximum kinematically allowed value of τ as a function of the number of partons N in the final state. They should be interpreted as lower limits to the maximum value of τ since a stochastic algorithm is not guaranteed to find the absolute minimum.

N	3	4	5	6	7	8
τ_{max}	$\frac{1}{3}$	$1 - \frac{1}{\sqrt{3}}$	0.4539536	0.4616661	0.4716270	0.4737373
N	9	10	11	12	13	14
τ_{max}	0.4782286	0.4802864	0.4816904	0.4840608	0.4836307	0.4842613

A quick way to calculate the thrust axis \vec{n} for small number N of partons in the final state is described in Appendix A of **Weinzierl'2009**, where the optimization problem of finding the unit vector which maximizes a certain configuration of final state partons is reduced to a simple loop over $2^{N-1} - 1$ configurations of signs $\{s_1, \dots, s_N\}$ which satisfies the constraint

$$\vec{p}_j \cdot \vec{n} = s_j |\vec{p}_j \cdot \vec{n}|, \quad (5)$$

where \vec{p}_j is the momenta of the j -th parton.

To see why the Thrust axis can be found by this method, suppose the constraint eq. (5) is satisfied. Then the thrust can be written as:

$$T = \max_{\vec{n}} \frac{\sum_j |\vec{p}_j \cdot \vec{n}|}{\sum_j |\vec{p}_j|} = \max_{\vec{n}} \frac{\sum_j s_j \vec{p}_j \cdot \vec{n}}{\sum_j |\vec{p}_j|} = \max_{\vec{n}} \frac{\vec{P} \cdot \vec{n}}{\sum_i |\vec{p}_i|}, \quad (6)$$

where $\vec{P} = \sum_j s_j \vec{p}_j$. It is clear that the thrust axis is the direction of the vector $\vec{n} = \frac{\vec{P}}{|\vec{P}|}$ and can be found by checking which of the $2^{N-1} - 1$ configurations of signs $\{s_1, \dots, s_N\}$ satisfies the constraint eq. (5) for all $j = 1, \dots, N$ and among them choose the one which maximizes the value of eq. (1). There are in total 2^N configurations of signs s_1, \dots, s_N , but we have to exclude the one where all the signs are equal because it violates momentum conservation, and also thrust is invariant under the transformation $\vec{n} \rightarrow -\vec{n}$.

We have implemented this algorithm in C++ to calculate the value of Thrust eq. (1) given a set of N momenta. We then used metaheuristic optimization algorithms (Genetic Algorithm (GA) and Particle Swarm Optimization (PSO)), to find the maximum value of τ (or minimum of T) for a given number N of partons by perturbing repeatedly the initially randomly generated momenta of the partons, always respecting the constraint of momentum conservation, and calculating the thrust for each configuration. The results for the first few N are shown in table 1, then the problem becomes computationally expensive

because the evaluation of the thrust scales as $O(2^N)$.

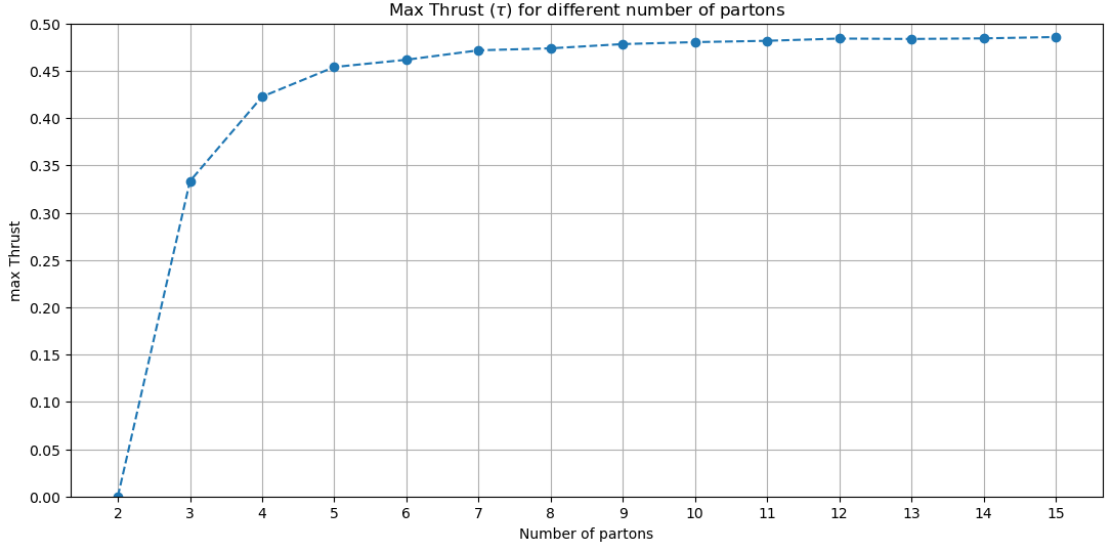


Figure 6: The maximum kinematically allowed value of τ as a function of the number of partons N in the final state. The value of τ_{max} approaches $1/2$ from below as the number of partons increases.

At small values of τ , higher order terms enhanced by powers of $\ln(\tau)$ dominates the thrust distribution. In this kinematical region the real expansion parameter is the large effective coupling $\alpha_s \ln^2(\tau)$ and therefore any finite-order perturbative calculation cannot give an accurate evaluation of the cross section.

For example, at leading order in perturbation theory the thrust distribution has the form:

$$\frac{1}{\sigma_0} \frac{d\sigma}{d\tau} = \delta(\tau) + \frac{2\alpha_s}{3\pi} \left[\frac{-4 \ln \tau - 3}{\tau} + \dots \right]_+, \quad (7)$$

where σ_0 is the born cross section, the ellipsis denotes terms that are regular as $\tau \rightarrow 0$ and the subscript $+$ denotes the plus distribution, which is defined as:

$$\left[\frac{-4 \ln \tau - 3}{\tau} \right]_+ = \lim_{\epsilon \rightarrow 0} \left[\left(\frac{-4 \ln \tau - 3}{\tau} \right) \theta(\tau - \epsilon) + \delta(\tau - \epsilon) \int_1^\tau d\tau' \left(\frac{-4 \ln \tau' - 3}{\tau'} \right) \right]. \quad (8)$$

Upon integration over τ , we obtain the cumulative distribution:

$$\begin{aligned}
R_T(\tau) &= \int_0^\tau d\tau' \frac{1}{\sigma_0} \frac{d\sigma}{d\tau'} \\
&= 1 + \frac{2\alpha_s}{3\pi} + \lim_{\epsilon \rightarrow 0} \left[\int_\epsilon^\tau d\tau' \left(\frac{-4 \ln \tau' - 3}{\tau'} + \dots \right) + \int_1^\epsilon d\tau' \left(\frac{-4 \ln \tau' - 3}{\tau'} + \dots \right) \right] \\
&= 1 + \frac{2\alpha_s}{3\pi} \left[-2 \ln^2 \tau - 3 \ln \tau + \dots \right].
\end{aligned} \tag{9}$$

Double logarithmic terms of the form $\alpha_s^n \ln^{2n} \tau$ plagues the fixed order expansion in the strong coupling. In the dijet region, higher order terms are as important as lower order ones, necessitating resummation to achieve reliable predictions.

1.1.2 Fixed Order Cross Section

The fixed-order thrust differential distribution has been calculated to leading order analytically and to NLO and NNLO numerically, as mentioned earlier. At a centre-of-mass energy Q and for a renormalization scale μ , the differential cross section takes the form:

$$\frac{1}{\sigma_0} \frac{d\sigma}{d\tau}(\tau, Q) = \delta(\tau) + \frac{\alpha_s(\mu)}{2\pi} \frac{dA}{d\tau}(\tau) + \left(\frac{\alpha_s(\mu)}{2\pi} \right)^2 \frac{dB}{d\tau} \left(\tau, \frac{\mu}{Q} \right) + \left(\frac{\alpha_s(\mu)}{2\pi} \right)^3 \frac{dC}{d\tau} \left(\tau, \frac{\mu}{Q} \right) + \mathcal{O}(\alpha_s^4), \tag{10}$$

where the complete leading order expression for the thrust distribution reads **Ellis:1980wv**:

$$\frac{dA}{d\tau} = C_F \left(9\tau - \frac{3}{\tau} + \left(-6 + \frac{4}{(1-\tau)\tau} \right) \ln \left(\frac{1-2\tau}{\tau} \right) + 6 \right), \tag{11}$$

with $C_F = \frac{4}{3}$ the Casimir of the fundamental representation of SU(3). As mentioned earlier, this expansion becomes unreliable near the dijet limit $\tau \rightarrow 0$ due to the presence of large logarithms. Resummation of these terms to all orders in α_s is necessary to obtain a reliable prediction.

The cumulant distribution has the following fixed-order expansion:

$$R_T(\tau) = 1 + A(\tau) \frac{\alpha_s(\mu)}{2\pi} + B(\tau, \mu) \frac{\alpha_s(\mu)^2}{2\pi} + C(\tau, \mu) \frac{\alpha_s(\mu)^3}{2\pi} + \mathcal{O}(\alpha_s^4), \tag{12}$$

the fixed-order coefficient A, B and C can be obtained by integrating the differential cross section eq. (11) to all order and imposing the normalization condition $R(\tau_{max}) = 1$, where τ_{max} is the maximum kinetically allowed value of τ . At leading order ($e^+e^- \rightarrow q\bar{q}g$) $\tau_{max} = \frac{1}{3}$, at Next-to-Leading Order is $\tau_{max} = 1 - \frac{1}{\sqrt{3}}$ and from NNLO onwards τ_{max} needs to be estimated numerically table 1.

At leading order we have:

$$\begin{aligned}
A(\tau) = \int_0^\tau d\tau' \frac{dA}{d\tau'} = C_F \Big(& \frac{9\tau^2}{2} - 2\ln^2(1-\tau) - 2\ln^2(\tau) + 6\tau(\ln(\tau) + 1) \\
& + 4\ln(1-\tau)\ln(\tau) - 3\ln(\tau) + 3(1-2\tau)\ln(1-2\tau) \\
& - 4\text{Li}_2\left(\frac{\tau}{1-\tau}\right) \Big). \tag{13}
\end{aligned}$$

For NLO and NNLO calculations, we use the numerical results from Table 1 in **Weinzierl'2009**, interpolating the data to obtain the fixed-order coefficients B and C .

We plot the fixed-order results for the thrust distribution in fig. 7 and compare it with data from OPAL **OPAL2005**, DELPHI **DELPHI:2000uri**, **hepdata.13245**, ALEPH **ALEPH:2003obs**, **hepdata.12794**, L3 **Achard'2004** experiments at LEP, as well as the SLD experiment **Abe'1995** at SLAC. The fixed-order predictions show good agreement with the data in the middle region of τ .

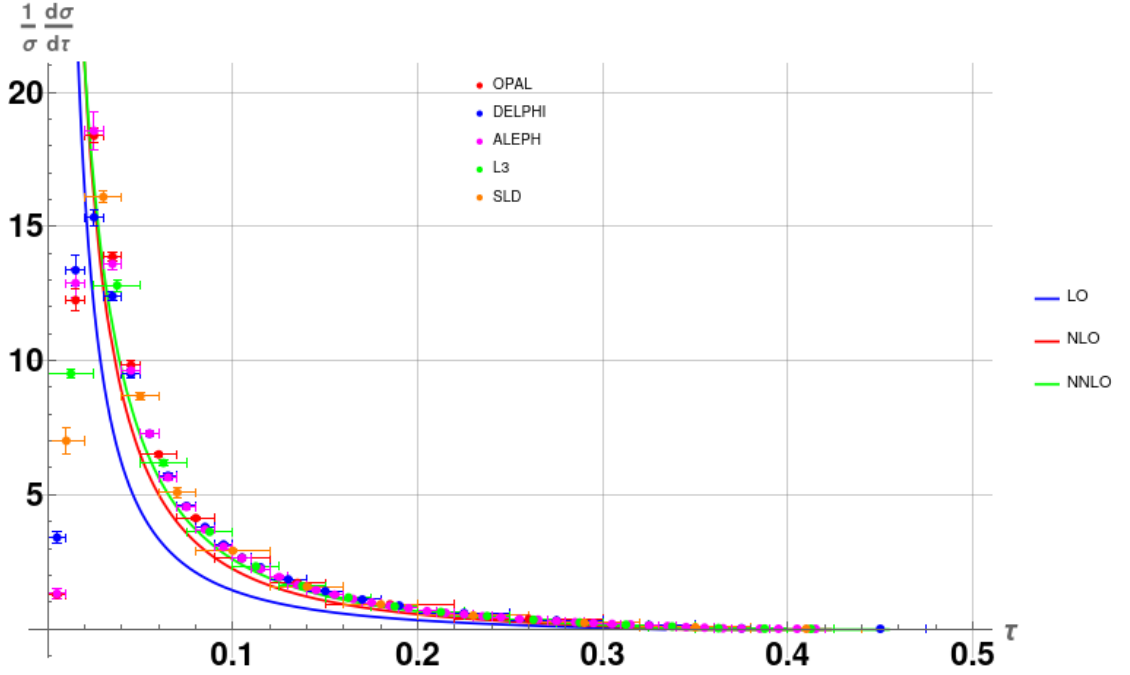


Figure 7: Fixed-order predictions for the thrust distribution at leading order (LO), next-to-leading order (NLO) and next-to-next-to-leading order (NNLO) compared with data from LEP and SLD experiments.

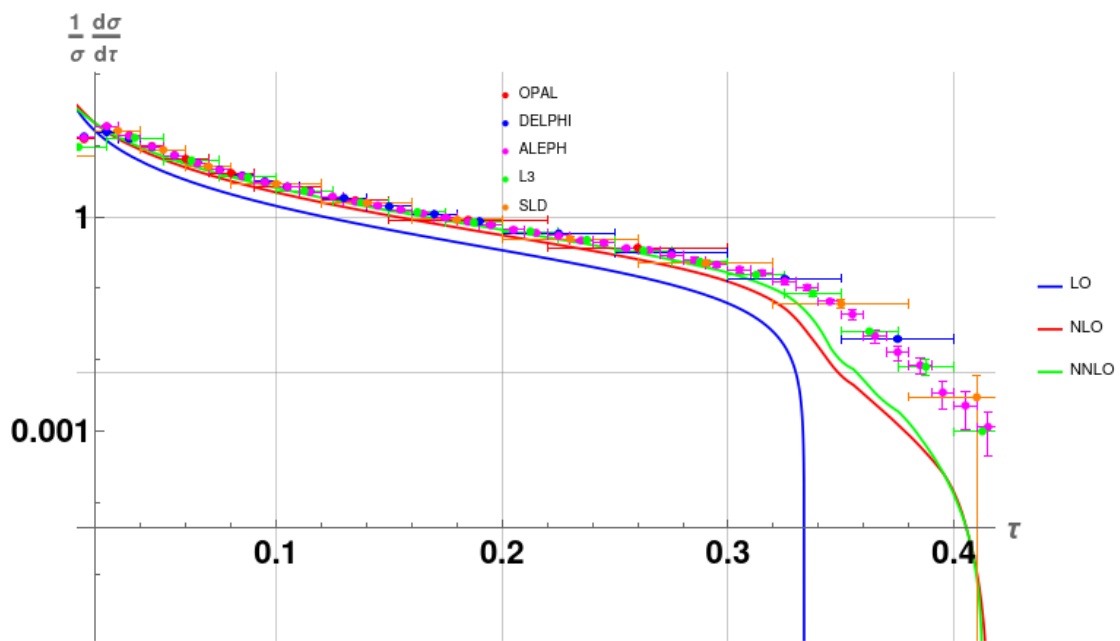


Figure 8: Log plot of the fixed-order predictions for the thrust distribution at leading order (LO), next-to-leading order (NLO) and next-to-next-to-leading order (NNLO) in the tail region.