空气动力学 HW9 解 P= = 1+ 2r (M2-1) $\Rightarrow P_2 = \left[1 + \frac{2.3}{2.4}(2.5^2 - 1)\right] \times 1 = 7.72 \text{ atm}$ $\frac{T_2}{T_1} = \frac{I_2}{I_1} \cdot \frac{I_1}{I_2} = \left[1 + \frac{2r}{r+1} (M_1^2 I)\right] \cdot \frac{2 + (r-1)M_1^2}{(r+1)M_1^2}$ $\Rightarrow \overline{1}_2 = \left[1 + \frac{2 \cdot 8}{2 \cdot 4} \times (2 \cdot 6^2 - 1)\right] \cdot \left(\frac{2 + 0.4 \times 2 \cdot 6^2}{2 \cdot 4 \times 2 \cdot 6^2}\right) \times 288$ = 644.64 K (ANS) $\frac{\ell_z}{\varrho_i} = \frac{(\gamma+i) M_i^2}{2+(\gamma-i) M_i^2}$ P. = P.RT. $\Rightarrow \rho_{2} = \frac{P_{1}}{RT_{1}} \cdot \frac{(\gamma + 1)M_{1}^{2}}{2 + (\gamma - 1)M_{2}^{2}} = \frac{1.01 \times 10^{5}}{287 \times 288} \times \frac{2.4 \times 2.6^{2}}{2 + 0.4 \times 2.6^{2}}$ = 4.214 kg/m3 [ANS] $M_2 = \sqrt{\frac{1 + \frac{Y^{-1}}{2} M_1^2}{\frac{Y^{-1}}{2} M_2^2 - \frac{Y^{-1}}{2}}} = \sqrt{\frac{1 + \frac{0.4}{2} \times 2.6^2}{1.4 \times 2.6^2 - \frac{0.4}{2}}} = 0.5039$ $\frac{T_{0.2}}{T_{0.2}} = 1 + \frac{\gamma - 1}{2} M_{0.2}^2$ \Rightarrow $702 = (1 + \frac{0.4}{2} \times 0.505 \hat{q}) \times 644.64 = 677.37 K [ANS]$ $\frac{|P_{0.2}|}{R} = \left(\frac{T_{0.2}}{T_0}\right)^{\frac{r}{r-1}}$ $\Rightarrow P_{0.2} = 7.12 \times \left(1 + \frac{0.4}{2} \times 0.5089^{\circ}\right)^{\frac{1.4}{0.4}} = 9.181$ Otm S2-51 = Cpln T2 - Rln P2 $=\frac{287\times14}{14-1}\times\ln\frac{644.64}{228}-287\times\ln\frac{7.72}{1}$

= 222.78 (J/kg·K)

Sz-S, = Cpln Tz - Rln Pz 0 $\left[-\frac{\dot{T}_2}{T_1} = \left[1 + \frac{2r}{r+1} \left(M_1^2 - I \right) \right] \cdot \left[\frac{2 + (r-I)M_1^2}{(r+I)M_2^2} \right]$ $= \left[1 + \frac{2.9}{3.4} \left(M_1^2 - 1\right)\right] \cdot \left(\frac{2 + 0.4 M_0^2}{2.4 M_1^2}\right)$ $= \left(\frac{7}{6}M_1^2 - \frac{1}{6}\right) \cdot \left(\frac{5 + M_1^2}{4M_1^2}\right)$ $= \frac{17}{18} + \frac{7}{26} M_1^* - \frac{5}{36 M_2^2}$ $\frac{P_2}{P_1} = 1 + \frac{2\gamma}{1+1} (M_1^2 - 1)$ $= _{1} + \frac{2 \cdot 8}{2 \cdot 4} \left(M_{1}^{2} - 1 \right) = \frac{7}{6} M_{1}^{2} - \frac{1}{6}$: Cpln Tz - Rln Pz $= l_{n} \left(\frac{\overline{l_{2}}}{\overline{l_{1}}} \right)^{C_{p}} - l_{n} \left(\frac{p_{2}}{p_{1}} \right)^{R}$ $= l_{n} \frac{\left(\frac{p_{2}}{\gamma_{1}} \cdot \frac{\rho_{1}}{\rho_{2}} \right)^{\frac{1}{P+1}}}{\left(\frac{p_{2}}{\rho_{2}} \right)^{R}} = l_{n} \left(\frac{\rho_{1}}{\ell_{2}} \right)^{\frac{1}{P+1}} \cdot \left(\frac{p_{2}}{p_{1}} \right)^{\frac{1}{P+1}}$ $= \ln \left[\left(\frac{2+0.4M_1^2}{2.4M_1^2} \right)^{\frac{1.4\times287}{2.4}} \cdot \left(\frac{7}{6}M_1^2 - \frac{1}{6} \right)^{\frac{-287}{2.4}} \right]$ 0 = Cp To _ Rln Poz &

Before shack wave: $0 = C_{1} \frac{T_{0.1}}{T_{1}} - R \ln \frac{P_{0.1}}{P_{1}}$ After $0 = C_{1} \frac{T_{0.2}}{T_{2}} - R \ln \frac{P_{0.2}}{P_{2}}$ Combine $0 \oplus 3$: $S_{2} - S_{1} = C_{1} \ln \frac{T_{0.2}}{T_{0.1}} - R \ln \frac{P_{0.2}}{P_{0.1}}$ $= -R \ln \frac{P_{0.2}}{P_{0.1}}$ $\Rightarrow \frac{P_{0.2}}{P_{0.1}} = e^{-\frac{(S_{1} - S_{1})}{P_{0.1}}} = e^{-\frac{199.5}{287}}$ = 0.499

From Appendix B: $M_1 = 2.5$ at $\frac{R_{0.1}}{P_{0.1}} = 0.499$

$$\frac{P_{0.1}}{P_{1}} = \frac{1.555}{1} = 1.555 < 1.893$$

$$\frac{P_{0.1}}{P_{1}} = \left(\frac{T_{0.1}}{T_{1}}\right)^{\frac{V}{V-1}} = \left(1 + \frac{V-1}{2}M_{1}^{2}\right)^{\frac{V}{V-1}}$$

$$\Rightarrow \left(1 + \frac{0.4}{2}M_{1}^{2}\right)^{\frac{1.4}{0.4}} = 1.555$$

$$M_{1} = 0.8199$$

$$V_{1} = Q_{1}M_{1} = \sqrt{YRT_{1}} \cdot M_{1}$$

$$= \sqrt{1.4\times287\times288} \times 0.8199$$

$$= 278.9 \quad m/s$$

8.14 解: for supersonic flow

$$\begin{array}{c} M_{i}>1 \\ P_{i} \oplus \\ C \oplus$$

shock wave

由②→②:
$$\frac{P_2}{P_1} = 1 + \frac{2r}{r+1} (M_1^2 - 1)$$

由②→③: $\frac{P_{0.2}}{P_2} = (\frac{T_{0.2}}{T_2})^{\frac{V}{r-1}} = (1 + \frac{v-1}{2} M_2)^{\frac{r}{r-1}}$

$$\frac{P_{02}}{P_1} = \frac{P_{02}}{P_2} \cdot \frac{P_2}{P_1}$$

$$= \left(1 + \frac{\gamma - 1}{2} M_2^2\right)^{\frac{\gamma}{\gamma - 1}} \cdot \left[1 + \frac{2\gamma}{\gamma + 1} (M_1^2 - 1)\right]$$

$$M_2^2 = \frac{1 + \frac{\gamma - 1}{2} M_1^2}{\gamma M_1^2 - \frac{3\gamma - 1}{2}}$$

$$\frac{P_{0.2}}{P_{1}} = \left(1 + \frac{\gamma - l}{2} \cdot \frac{1 + \frac{\gamma - l}{2} M_{1}^{2}}{\gamma M_{1}^{2} - \frac{\gamma - l}{2}}\right)^{\frac{\gamma}{\gamma - l}} \cdot \left[1 + \frac{2\gamma}{\gamma + l} (M_{1}^{2} - l)\right]$$

$$= \left[\frac{(\gamma - l)\left[2 + (\gamma - l)M_{1}^{2}\right] + 4\gamma M_{1}^{2} - 2\gamma + 2}{4\gamma M_{1}^{2} - 2(\gamma - l)}\right]^{\frac{\gamma}{\gamma - l}} \cdot \left(\frac{1 - \gamma + 2\gamma M_{1}^{2}}{\gamma + l}\right)$$

$$= \left[\frac{(\gamma + l)^{2} M_{1}^{2}}{4\gamma M_{1}^{2} - 2(\gamma - l)}\right]^{\frac{\gamma}{\gamma - l}} \cdot \left(\frac{1 - \gamma + 2\gamma M_{1}^{2}}{\gamma + l}\right)$$

8.16

$$\frac{P_{0.2}}{P_1} = \frac{1.13}{0.1} = 11.3 > 1.893$$

$$\frac{P_{0.2}}{P_1} = 11.3 = \left[\frac{(1.4+1)^2 M_1^2}{5.1 M_1^2 - 0.8} \right]^{\frac{1.4}{0.4}} \times \left(\frac{2.8 M_1^2 - 0.4}{2.4} \right)$$

$$\approx \left(\frac{2.4^2 M_1^2}{5.6 M_1^2} \right)^{\frac{1.4}{0.4}} \times \left(\frac{2.9 M_1^2 - 0.4}{2.4} \right)$$

$$\Rightarrow M_1 = 3.0 \quad (2.14)$$
And from Appendix B 1

With $\frac{P_{0.2}}{P_1} = 11.3$, we get $M_1 = 2.9$

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