

Q3.1 (i) proof:

$$\frac{1}{c^2} \left(\frac{\partial^2}{\partial t^2} - \nabla^2 \right) p' = \frac{A}{c^2} e^{-\epsilon x_1^2} \delta(x_2) \delta(x_3) e^{i(\omega t - \alpha x_1)} = q(x, t)$$

$$= \int_V q(y, t) \delta(x - y) d^3y \quad \left(\begin{array}{l} \text{decompose } q(x, t) \text{ into a} \\ \text{superposition of point sources} \end{array} \right)$$

for inhomogeneous wave equation:

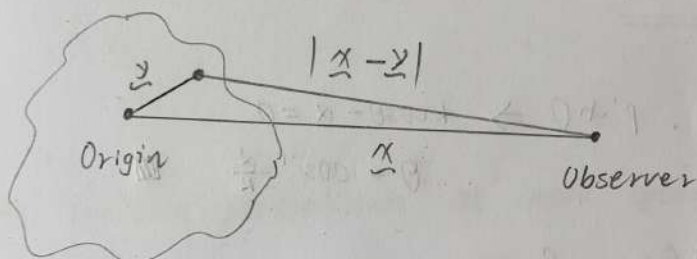
$$\left(\frac{1}{c^2} \frac{\partial^2}{\partial t^2} - \nabla_x^2 \right) p' = Q(y, t) \delta(x - y)$$

solution $p' = \frac{Q(y, t - |x - y|/c)}{4\pi |x - y|}$

$$\Rightarrow \left(\frac{1}{c^2} \frac{\partial^2}{\partial t^2} - \nabla_x^2 \right) \frac{q(y, t - |x - y|/c)}{4\pi |x - y|} = q(y, t) \delta(x - y)$$

$$\left(\frac{1}{c^2} \frac{\partial^2}{\partial t^2} - \nabla_x^2 \right) \int_V \frac{q(y, t - |x - y|/c)}{4\pi |x - y|} d^3y = \int_V q(y, t) \delta(x - y) d^3y = q(x, t)$$

where $p'(x, t) = \int_V \frac{q(y, t - |x - y|/c)}{4\pi |x - y|} d^3y$



from hint:

$$|x - y| = x \left(1 - \frac{x \cdot y}{x^2} \right) \quad \text{for } x \gg y$$

$$\frac{1}{|x - y|} = \frac{1}{x} + O\left(\frac{|y|}{x^2}\right)$$

$$\Rightarrow p'(x, t) = \frac{1}{4\pi x} \int_V q(y, t - \frac{x}{c} + \frac{x \cdot y}{xc}) d^3y$$

$$= \frac{1}{4\pi x} \int_{-\infty}^{\infty} \frac{A}{c^2} e^{-\epsilon y^2} e^{i\omega(t - \frac{x}{c} + \frac{x \cdot y}{xc})} e^{i(\alpha y)} dy$$

$x \cdot y = |x| \cdot |y| \cdot \cos \theta = xy \cos \theta$

$$= \frac{A}{c^2 4\pi x} \int_{-\infty}^{\infty} e^{-\epsilon y^2} e^{i(\omega t - kx)} e^{i \cdot ky \cos \theta} e^{i(-\alpha)y} dy$$

$$= \frac{A}{c^2 4\pi x} e^{i(\omega t - kx)} \int_{-\infty}^{\infty} e^{-\epsilon y^2} e^{i(k \cos \theta - \alpha)y} dy$$

from hint: $\int_{-\infty}^{\infty} e^{-\epsilon y^2} e^{i(k \cos \theta - \alpha)y} dy = \sqrt{\frac{\pi}{\epsilon}} e^{-(k \cos \theta - \alpha)^2 / 4\epsilon}$

$$= \frac{A}{c^2 4\pi x} e^{i(\omega t - kx)} \cdot \sqrt{\frac{\pi}{\epsilon}} e^{-(k \cos \theta - \alpha)^2 / 4\epsilon}$$

$$= \frac{A}{4c^2 x \sqrt{\pi \epsilon}} e^{i(\omega t - kx)} e^{-(k \cos \theta - \alpha)^2 / 4\epsilon}$$

(ii) phase speed $v_p = \frac{dx}{dt}$ when phase $\omega t - \beta x = \text{constant}$

far-field sound pressure: $p' = \frac{A}{4c^2 x \sqrt{\pi \epsilon}} e^{i(\omega t - kx)} e^{-(k \cos \theta - \alpha)^2 / 4\epsilon}$

from $(\frac{\partial^2}{\partial t^2} - c^2 \nabla^2) p' = A e^{-\epsilon x_1^2} \delta(x_2) \delta(x_3) e^{i(\omega t - \alpha x_1)}$

phase $= \omega t - \alpha x_1 = \text{constant}$

$\downarrow \frac{d}{dt}$
 $\omega - \alpha \frac{dx_1}{dt} = 0$

$\frac{dx_1}{dt} = \frac{\omega}{\alpha} = \text{phase speed } v_p$ VII

4'

(iii) $\epsilon \ll 1$

$p' = \frac{A}{4c^2 x \sqrt{\pi \epsilon}} e^{i(\omega t - kx)} e^{-(k \cos \theta - \alpha)^2 / 4\epsilon}$

if we want the sound to be radiated. $p' \neq 0 \Rightarrow k \cos \theta - \alpha = 0$

$\theta = \cos^{-1} \frac{\alpha}{k}$ VIII

phase speed $v_p = \frac{\omega}{\alpha} = \frac{\omega}{k \cos \theta} = \frac{k \cdot c}{k \cos \theta} = \frac{c}{\cos \theta}$ supersonic IX

6'

Q3.2

Proof:

the pressure perturbation at the neck opening: $p_t'(t)$

mass flow rate into the bulb: Q kg/s

in the bulb volume V : $V \cdot \frac{\partial p_b'}{\partial t} = Q$ ①

$$p_b' = c^2 p_b' \quad ②$$

assume harmonic disturbances with $\omega = \frac{\partial U}{\partial t}$ (characteristic frequency)

$$① \Rightarrow p_b' = \frac{Q}{i\omega V}$$

$$② \Rightarrow p_b' = \frac{c^2 Q}{i\omega V}$$

momentum balance in the neck: $p_t' - p_b' = \rho_0 l \frac{\partial u_e}{\partial t}$ ③

$$\text{where } u_e = \frac{Q}{\rho_0 S}$$

$$\Rightarrow p_t' = p_b' + \rho_0 l \frac{\partial u_e}{\partial t} = \frac{c^2 Q}{i\omega V} + \frac{i\omega l Q}{S}$$

$$= \left(\omega^2 - \frac{c^2 S}{Vl} \right) \frac{i l Q}{\omega S}$$

non-dimensional number

rms pressure perturbation at neck opening: $p_{t,rms}' = \frac{1}{2} \rho_0 U^2$

$$\left(\frac{\partial}{\partial t} \rightarrow i\omega, \text{ time scale is } \frac{D}{\partial U} \right)$$

$$\sim \left(\omega^2 - \frac{c^2 S}{Vl} \right) \frac{l Q}{\omega S}$$

$$\Rightarrow Q \sim \frac{\rho_0 U^2 \omega S}{l} \left(\omega^2 - \frac{c^2 S}{Vl} \right)^{-1}$$

$$p'(t) = \frac{\dot{Q}(t-r/c)}{4\pi r} \text{ for monopole source} \sim \frac{\rho_0 S D U}{l \omega} \left(1 - \frac{S D^2}{\omega^2 M^2 V l} \right)^{-1}$$

$$\text{Sound power radiated from the neck: } \text{Power} = \frac{\overline{p'^2}}{\rho_0 c} \cdot S = \frac{4\pi r^2}{\rho_0 c} \cdot \left[\frac{\dot{Q}(t-r/c)}{4\pi r} \right]^2$$

$$= \frac{1}{\rho_0 c} \cdot \frac{1}{4\pi} \cdot (\overline{i\omega Q})^2 \sim \frac{1}{\rho_0 c} \cdot \omega^2 Q^2$$

$$= \frac{1}{\rho_0 c} \cdot \frac{\omega^2 U^2}{D^2} \cdot \frac{\rho_0^2 S^2 D^2 U^2}{l^2 \omega^2} \cdot \left(1 - \frac{S D^2}{\omega^2 M^2 V l} \right)^{-1}$$

$$= \frac{M U^3 S^2 \rho_0}{l^2} \left(1 - \frac{S D^2}{\omega^2 M^2 V l} \right)^{-1}$$

20'

Q3.3

(i) an unbounded fluid.

piston vibrating surface displacement : $\xi e^{i\omega t}$

$$\sim \text{velocity : } \frac{\partial(\xi e^{i\omega t})}{\partial t} = i\omega \xi \cdot e^{i\omega t} = u_0 e^{i\omega t}$$

linearised momentum equation on the surface:

$$\rho_0 \frac{\partial u_s}{\partial t} + \frac{\partial p'}{\partial x} = 0$$

$$\rho_0 \frac{\partial u_s}{\partial t} = \rho_0 \cdot i\omega \cdot u_0 e^{i\omega t} = - \frac{\partial p'}{\partial x}$$

An impenetrable compact body changing in volume:

$$p'(\underline{x}, t) = \frac{\rho_0 \ddot{V}(t - |\underline{x}|/c)}{4\pi |\underline{x}|}$$

$$\dot{V}(t) = \int_S \underline{n} \cdot \underline{u} \, ds = l^2 \cdot u_0 e^{i\omega t}$$

$$\ddot{V}(t) = l^2 \cdot u_0 i\omega e^{i\omega t}$$

$$\Rightarrow p'(\underline{x}, t) = \frac{\rho_0 l^2 u_0 i\omega e^{i\omega(t - |\underline{x}|/c)}}{4\pi |\underline{x}|}$$

Since $kr \gg 1$, far field acoustic pressure:

$$p'(r, t) = \frac{\rho_0 l^2 u_0 i\omega e^{i\omega(t - r/c)}}{4\pi r}$$

where $u_0 = i\omega \xi$

$$= - \frac{\rho_0 l^2 \omega^2 \xi e^{i\omega(t - r/c)}}{4\pi r}$$

(ii) Cube in long tube

inside the tube: $p'(x,t) = I e^{i(\omega t - kx)} + R e^{i(\omega t + kx)}$

At $x=0$, surface displacement $= \epsilon e^{i\omega t}$

$$u_s(t) = i\omega \epsilon e^{i\omega t} = \frac{1}{\rho_0 c} \cdot (I - R) e^{i\omega t}$$

At $x=l$, $p' = I e^{i(\omega t - kl)} + R e^{i(\omega t + kl)} = (I e^{-ikl} + R e^{ikl}) e^{i\omega t}$

Linearised momentum equation: $\rho_0 \frac{\partial u}{\partial t} + \frac{\partial p'}{\partial x} = 0$

$$\frac{\partial u}{\partial t} = -\frac{1}{\rho_0} \frac{\partial p'}{\partial x}$$

$$u = \frac{-1}{\rho_0 i\omega} \frac{\partial p'}{\partial x}$$

$$= \frac{-1}{\rho_0 i\omega} \cdot I e^{i\omega t} \cdot e^{-ikx} \cdot \left(-\frac{ik}{c}\right) + R e^{i\omega t} \cdot e^{ikx} \cdot \left(\frac{ik}{c}\right)$$

$$= \frac{1}{\rho_0 c} (I e^{-ikx} - R e^{ikx}) e^{i\omega t}$$

At the open end of the tube, $p' = 0 = (I e^{-ikx_{\infty}} + R e^{ikx_{\infty}}) e^{i\omega t}$

$\Rightarrow R = 0, I = i\omega \epsilon \cdot \rho_0 c$ (infinitely long tube)

$$p' = i\omega \epsilon \rho_0 c e^{i(\omega t - kt)}$$

Power 1 = $\frac{1}{\rho_0 c} \cdot \overline{p'^2} \cdot 4\pi r^2 = \frac{1}{\rho_0 c} \cdot \frac{\rho_0^2 L^4 \omega^2 \epsilon^2 \cdot \omega^2 \cdot \frac{1}{2}}{16\pi^2 \cdot L^2} \cdot 4\pi r^2$

10'

$$= \frac{\rho_0 L^4 \omega^2 \epsilon^2}{8\pi c}$$

~~$\frac{4 \cdot 8\pi \cdot \rho_0 \cdot \omega^2 \epsilon^2 \cdot c^2}{2 \cdot \rho_0 L^4 \omega^2 \epsilon^2 \cdot \omega \cdot L^2}$~~

Power 2 = $\frac{1}{\rho_0 c} \overline{p'^2} \cdot L^2 = \frac{1}{\rho_0 c} \cdot \frac{1}{2} \omega^2 \epsilon^2 \rho_0^2 c^2 \cdot L^2$

$$= \frac{1}{2} \rho_0 \omega^2 \epsilon^2 c L^2$$

12'

$$\frac{\text{Power 2}}{\text{Power 1}} = \frac{\rho_0 \omega^2 \epsilon^2 c L^2 \cdot 4\pi c}{\rho_0 L^4 \omega^2 \epsilon^2 L^2 \omega^2} = 4\pi \cdot L^{-2} \cdot k^{-2} = 4\pi (kl)^{-2}$$

□

Q3.4

rigid surface 1

i) Proof:

flame with the rate of heat addition per unit volume $w(x, t)$ → for the acoustic source: $p = p(p, s)$

$$dp = \left. \frac{\partial p}{\partial p} \right|_s dp + \left. \frac{\partial p}{\partial s} \right|_p ds$$

$$= \frac{1}{c^2} dp + \left. \frac{\partial p}{\partial s} \right|_p ds$$

for perfect gas: $s - s_0 = C_v \ln \frac{p}{p_0} - C_p \ln \frac{p}{p_0}$

$$\Rightarrow \left. \frac{\partial p}{\partial s} \right|_p = -\frac{p}{C_p}$$

$$\Rightarrow dp = \frac{1}{c^2} dp - \frac{p}{C_p} ds$$

$$\frac{dp}{dt} = \frac{1}{c^2} \frac{dp}{dt} - \frac{p}{C_p} \frac{ds}{dt} \stackrel{\uparrow}{=} \frac{1}{c^2} \frac{dp}{dt} - \frac{\gamma-1}{c^2} w$$

$$(\rho T \frac{ds}{dt} = w)$$

$$(\frac{1}{C_p T} = \frac{\gamma-1}{c^2})$$

$$\text{linearising: } \frac{1}{c^2} \frac{\partial p'}{\partial t} - \frac{\partial p'}{\partial t} = \frac{\gamma-1}{c^2} w$$

$$\left. \begin{aligned} \frac{\partial}{\partial t} \rightarrow \frac{\partial p'}{\partial t} + \rho_0 \nabla \cdot \underline{u} &= 0 \\ \nabla \cdot \rightarrow \rho_0 \frac{\partial \underline{u}}{\partial t} + \nabla p' &= 0 \end{aligned} \right\} \Rightarrow \frac{\partial^2 p'}{\partial t^2} - \nabla^2 p' = 0$$

$$\Rightarrow \frac{1}{c^2} \frac{\partial^2 p'}{\partial t^2} - \nabla^2 p' = \frac{\gamma-1}{c^2} \frac{\partial w}{\partial t} \quad (\text{wave equation})$$

$$\text{where } p'(x, t) = \frac{\gamma-1}{4\pi c^2} \frac{\partial}{\partial t} \int_V \frac{w(y, t - |x-y|/c)}{|x-y|} d^3y$$

$$\text{Given: } \frac{\partial^2 p'}{\partial t^2} - c^2 \nabla^2 p' = \frac{\gamma-1}{c^2} \frac{\partial w}{\partial t}$$

$$\Rightarrow p'(x, t) = \frac{\gamma-1}{4\pi c^4} \frac{\partial}{\partial t} \int_V \frac{w(y, t - |x-y|/c)}{|x-y|} d^3y \quad \checkmark$$

Since the flame source distribution is compact, far-field: $|x| \gg |y|$

$$p'(x, t) = \frac{\gamma-1}{4\pi c^4} \frac{\partial}{\partial t} \int_V \frac{w(y, t - |x|/c)}{|x|} d^3y$$

$$= \frac{\rho_0}{4\pi c^2 |x|} \frac{\partial^2}{\partial t^2} \int_V \beta(y, t - |x|/c) d^3y \quad (\text{from hint})$$

$$\frac{\partial}{\partial t} \sim \frac{u}{L}, \quad \int_V d^3y \sim L^3, \quad \beta \sim 1$$

$$\Rightarrow p' \sim \frac{\rho_0}{c^2 |x|} \cdot \frac{u^2}{L^2} \cdot L^3 = \frac{\rho_0 L}{|x|} \cdot \left(\frac{u}{c}\right)^2 = \frac{\rho_0 L}{|x|} \cdot M^2$$

$$(ii) \quad p' \propto \rho_0 \left(\frac{L}{\lambda} \right) M^2 \quad \Rightarrow \quad \overline{p'^2} \propto \rho_0^2 \frac{L^2}{\lambda^2} M^4$$

far-field acoustic power

$$p' = c^2 \cdot \rho' \quad \text{or} \quad \rho' = \frac{p'}{c^2}$$

$$\text{Power} = \int_S \overline{p' u} dS = 4\pi \lambda^2 \cdot \frac{\overline{p'^2}}{\rho_0 c}$$

$$p' = \frac{r-1}{4\pi \lambda c^2} \cdot \frac{\partial^2}{\partial t^2} \int_V w(\underline{y}, t - \lambda/c) d^3 y$$

$$4\pi \lambda^2 \cdot \frac{(c^2 \rho')^2}{\rho_0 c} = 4\pi \lambda^2 \cdot \frac{c^4 \cdot \overline{p'^2}}{\rho_0 c}$$

$$= \frac{\rho_0}{4\pi \lambda} \frac{\partial^2}{\partial t^2} \int_V \beta(\underline{y}, t - \lambda/c) d^3 y \propto \frac{\rho_0}{\lambda} \cdot \frac{u^2}{L^2} \cdot L^3 = \frac{\rho_0 u^2 L}{\lambda}$$

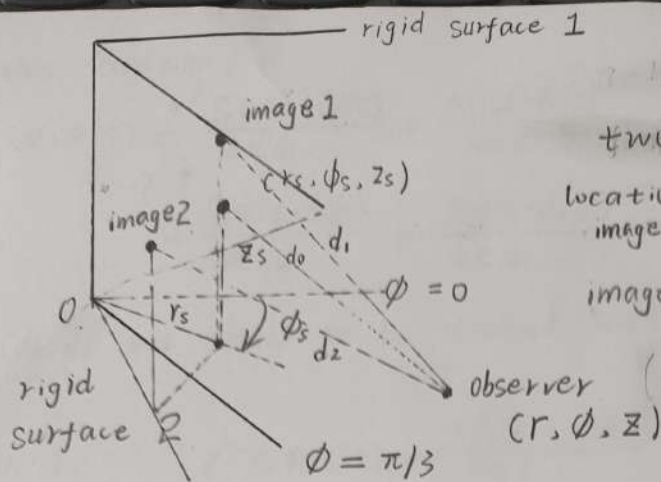
$$\Rightarrow \text{Power} \propto \lambda^2 \cdot \frac{1}{\rho_0 c} \cdot \frac{\rho_0^2 u^4 L^2}{\lambda^2} = \rho_0 L^3 u^3 M^4$$

$$= c^2 \rho_0 L^2 M^4$$

$$= u^3 \rho_0 L^2 M$$

Q3.5

Solution:



basic

two necessary images of the loudspeaker

locations:

image 1: $(r_s, -\phi_s, z_s)$

image 2: $(r_s, \frac{2}{3}\pi - \phi_s, z_s)$

small loudspeaker with a rate of volume outflow $Q(t)$

$p_0 \dot{Q}(t)$

image source is symmetric about the rigid surface

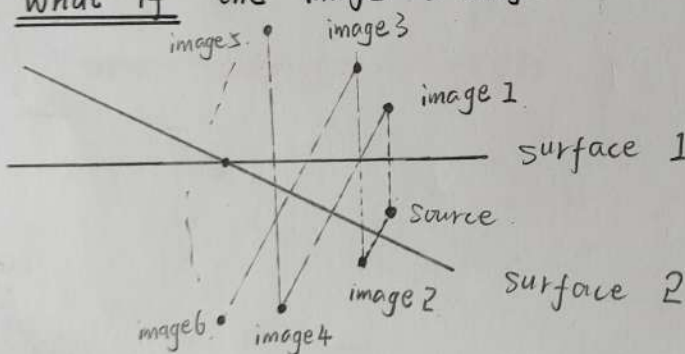
$$p'(r, \phi, z, t) = \frac{p_0 \dot{Q}(t - d_0/c)}{4\pi d_0} + \frac{p_0 \dot{Q}(t - d_1/c)}{4\pi d_1} + \frac{p_0 \dot{Q}(t - d_2/c)}{4\pi d_2} \quad \square$$

$$d_0 = \sqrt{(x - x_s)^2 + (y - y_s)^2 + (z - z_s)^2} = \sqrt{(r \cos \phi - r_s \cos \phi_s)^2 + (r \sin \phi - r_s \sin \phi_s)^2 + (z - z_s)^2}$$

$$d_1 = \sqrt{(r \cos \phi - r_s \cos \phi_0)^2 + (r \sin \phi + r_s \sin \phi_s)^2 + (z - z_s)^2}$$

$$d_2 = \sqrt{[r \cos \phi - r_s \cos(\frac{2}{3}\pi - \phi_s)]^2 + [r \sin \phi - r_s \sin(\frac{2}{3}\pi - \phi_s)]^2 + (z - z_s)^2}$$

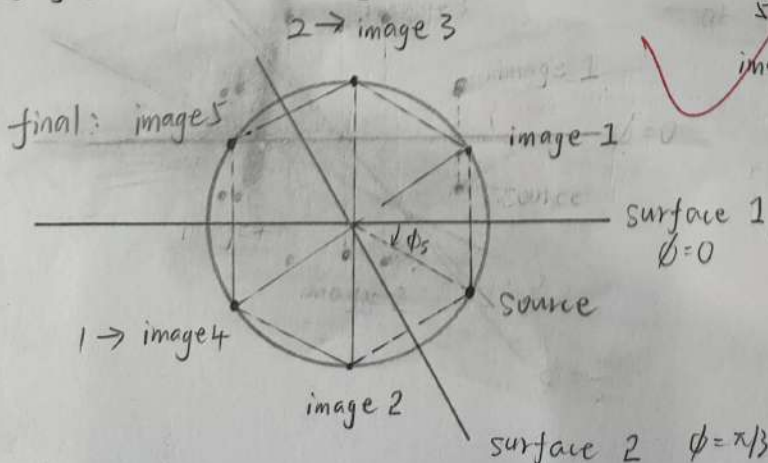
what if the image \rightarrow image in another surface.



the number of the images

is dependent of the location of source

To find the necessary images, consider:



5 images in total are necessary

image 3 $(r_s, -\frac{2}{3}\pi + \phi_s, z_s)$

4 $(r_s, \frac{2}{3}\pi + \phi_s, z_s)$

5 $(r_s, \phi_s + \pi, z_s)$

in this condition:

$$p'(r, \phi, z, t) = \frac{\rho_0 \dot{Q}(t-d_0/c)}{4\pi d_0} + \frac{\rho_0 \dot{Q}(t-d_1/c)}{4\pi d_1} + \frac{\rho_0 \dot{Q}(t-d_2/c)}{4\pi d_2} + \frac{\rho_0 \dot{Q}(t-d_3/c)}{4\pi d_3} \\ + \frac{\rho_0 \dot{Q}(t-d_4/c)}{4\pi d_4} + \frac{\rho_0 \dot{Q}(t-d_5/c)}{4\pi d_5} \quad \square$$

d_3, d_4, d_5 are similar to be obtained with d_0, d_1, d_2 .

Small r_s : $p'(r, \phi, z, t) = \frac{3\rho_0 \dot{Q}(t-r/c)}{4\pi r}$

(if image \rightarrow image: $p' = \frac{6\rho_0 \dot{Q}(t-r/c)}{4\pi r}$) ✓

unbounded: $\frac{\rho_0 \dot{Q}}{4\pi r}$

acoustic power $\propto \overline{p'^2} \Rightarrow$ a factor of 9 (of $\frac{36}{4}$) ✓



~~hemisphere~~

Because the sound source is mirrored by the rigid surfaces, creating increased sound pressure.

$p' = 6 p'_0$

$\overline{p'^2} = 36 \overline{p_0'^2}$

$\frac{\overline{p_0'^2}}{\rho_0 c} \cdot 4\pi r_0^2$

~~area: $4\pi r^2 \cdot \frac{1}{2}$ $r = \frac{1}{2} r_0$~~

$\frac{\pi}{3}$ 扇形面积

$\frac{\overline{p'^2}}{\rho_0 c} \cdot \left(4\pi r_0^2 \cdot \frac{1}{6}\right)$ ✓

$\frac{36 \overline{p_0'^2}}{\rho_0 c} \cdot \frac{1}{6} \cdot 4\pi r_0^2 \rightarrow 6 \text{ 倍 Power.}$