On the meaning of the normalisation parameter G_0 and the normalised Energy Release Rates in the fiber/matrix interface problem

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Abstract.

List of acronyms

BEM Boundary Element Method

ERR Energy Release Rate (here used as synonim of SERR)

FEM Finite Element Method SERR Strain Energy Release Rate VCCT Virtual Crack Closure Technique

List of symbols

1. Introduction

The numerical analysis of the fiber/matrix interface has focused on the determination of mode mixity through the calculation of mode I and mode II energy release rates. In early papers, where the problem was solved in the complex domain by means of the Airy function and conformal transformations, it was shown that the stress field was due to oscillate in a very small region close to the crack tip. Thus, the evalution of stresses and stress intensity factors represents an ardous obstacle when to the discrete counterpart of the problem.

Due to their definition as rates of energy change, the calculation of ERRs fits better into the frameworks of discrete procedures, whether the Finite Element Method (FEM) or the Boundary Element Method (BEM). Different variations of the same principle has been derived over the years, namely the Crack Closure Technique, the Crack Closure Integral, the Virtual Crack Closure Technique and the Virtual Crack Closure Integral. The idea at the core of these methods is that, given that the crack is propagating in a linear elastic medium, the energy released by the creation of a unit area of crack's surfaces is equal to the work needed to close the new created surfaces back together.

Energy Release Rates have been so far reported in a normalized form, where a reference energy release rate G_0 is used as normalization parameter. In [?], the authors claim that the use of

$$G_0 = \frac{1 + k_m}{8\mu_m} \sigma_0^2 R_f \pi \tag{1}$$

would make the results comparable between different material systems. In equation 1, k_m is the Kolosov constant for the matrix, which is equal to $3-4\nu$ for plane strain and $\frac{3-\nu}{1+\nu}$ for plane stress conditions, μ_m is the shear modulus of the matrix, σ_0 is the applied stress at the boundary and R_f is the radius of the inclusion.

In [?], the same normalization parameter is used to analyze the effect of a neighbouring fiber on the Energy Release Rates. A similar use can be found in [?] It seems that the first apperance of G_0 can be retrieved in [?], soon followed by [?] where a parametric study of Toya's analytical results is performed and tabulated. However, in Toya the normalization is performed for the maximum? stress at the crack tip, and its actual formulation is formula from toya.

The question thus arises: what is the meaning of G_0 ? And consequently, what is the meaning of the normalized Energy Release Rates? How does the selection of this peculiar normalization value make results comparable across different material systems and ply geoemetries? In this brief note we will try to answer these questions.

2. Analysis of G_0

Consider an infinite plate made of an isotropic homogeneous material with a central crack of length 2a in its center. We assume that the material is the same as matrix constituent of a composite, and thus we will use the pedix m to identify the properties. Under an applied remote traction σ_0 perpendicular to the crack faces, the mode I Stress Intensity Factor (SIF) reads [?]

$$K_I^{CC} = \sigma_0 \sqrt{\pi a}.$$
 (2)

Across the literature on Linear Elastic Fracture Mechanics, in textbooks and handbooks, the expression in eq. 2 has assumed the role of reference for the expression of Stress Intensity Factors for all possible modes of fracture and geometries. Thus, the SIF for any other arbitrary geometrical configuration of crack and sorrounding structure can be expressed as

$$K_{mode} = \beta K_{ref}$$
 where $K_{ref} = K_I^{CC} = \sigma_0 \sqrt{\pi a}$ (3)

or, equivalently,

$$K_{mode} = \beta \sigma_0 \sqrt{\pi a} \tag{4}$$

where β is the shape factor. Thus, evaluation of the SIF for a new configuration reduces to the determination of its shape factor.

The corresponding energy release rate can be computed to be

$$G_{ref} = G_I^{CC} = \frac{\left(K_I^{CC}\right)^2}{E_m^*}.$$
 (5)

For plane strain,

$$E_m^* = \frac{E_m}{1 - \nu^2}$$
 and thus $G_{ref} = \frac{1 - \nu^2}{E_m} \sigma_0^2 \pi a$, (6)

while for plane stress,

$$E_m^* = E_m$$
 and thus $G_{ref} = \frac{\sigma_0^2 \pi a}{E_m}$. (7)

The energy release rate for the generic configuration thus reads

$$G_{mode} = \beta G_{ref}, \tag{8}$$

which particularizes to $G_{mode} = \beta \frac{1-\nu^2}{E_m} \sigma_0^2 \pi a$ for plane strain, and $G_{mode} = \beta \frac{\sigma_0^2 \pi a}{E_m}$ for plane stress.

A homogeneous isotropic material has only 2 independent parameters that determine its elastic behavior; thus it holds

$$G_m = \frac{E_m}{2\left(1 + \nu_m\right)}. (9)$$

Let us consider once again the reference energy release rate for plane strain states and rework its expression

$$G_{ref} = \frac{1 - \nu_m^2}{E_m} \sigma_0^2 \pi a =$$

$$= \frac{(1 - \nu_m) (1 + \nu_m)}{E_m} \sigma_0^2 \pi a =$$

$$= \frac{(1 - \nu_m)}{2G_m} \sigma_0^2 \pi a =$$

$$= \frac{4}{4} \frac{(1 - \nu_m)}{2G_m} \sigma_0^2 \pi a =$$

$$= \frac{1 + 3 - 4\nu_m}{8G_m} \sigma_0^2 \pi a =$$

$$= \frac{1 + k_m}{8G_m} \sigma_0^2 \pi a$$
(10)

where $k_m=3-4\nu_m$ is the Kolosov's constant for plane strain. Similarly, for plane stress

$$G_{ref} = \frac{\sigma_0^2 \pi a}{E_m} = \frac{\sigma_0^2 \pi a}{2G_m (1 + \nu_m)} = \frac{4}{4} \frac{1}{2G_m (1 + \nu_m)} \sigma_0^2 \pi a$$

$$= \frac{1}{8G_m} \frac{4 + \nu_m - \nu_m}{1 + \nu_m} \sigma_0^2 \pi a$$

$$= \frac{1}{8G_m} \left(\frac{1 + \nu_m}{1 + \nu_m} + \frac{3 - \nu_m}{1 + \nu_m} \right) \sigma_0^2 \pi a = \frac{1 + \frac{3 - \nu_m}{1 + \nu_m}}{8G_m} \sigma_0^2 \pi a = \frac{1 + k_m}{8G_m} \sigma_0^2 \pi a$$

$$= \frac{1 + k_m}{8G_m} \sigma_0^2 \pi a$$
(11)

where $k_m = \frac{3-\nu_m}{1+\nu_m}$ is the Kolosov's constant for plane stress. Recalling now eq. 1 and comparing with eqs. 10 and 11, we can now identify a clear physical meaning: G_0 is the mode I energy release rate for a central crack of total length equal to the inclusion diameter $2R_f$ in an infinite plate made of the same material as the matrix. Furthermore, a historical purpose can be identified, as this formulation of G_0 allow for the expression of the energy release rates for the fiber/matrix interface problem as

$$G_I = \beta_I G_{ref} \qquad G_{II} = \beta_{II} G_{ref}, \tag{12}$$

where β_I and β_{II} are respectively the mode I and mode II shape parameters and can now be identified as the functions $\frac{G_I}{G_0}$ and $\frac{G_{II}}{G_0}$ that represent a main part of the results of the fiber/matrix interface problem.

3. A new formulation of G_0 for a multi-scale analysis of transverse cracking

Given the elastic properties of the transversely isotropic UD ply $E_1, E_2, nu_{12}, nu_{23}$, for a 90° ply under transverse tension the cross section along the direction of the load coincides with the plane of transversal isotropy. It is thus possible, for a system in plane strain, to define equivalent isotropic Young's modulus and Poisson's ratio as follows. The effective Young's modulus and Poisson's ratio in plane strain in the plane of isotropy are defined as

$$E^* = \frac{E_2}{1 - \nu_{21}\nu_{12}} \qquad \nu^* = \frac{\nu_{23} + \nu_{21}\nu_{12}}{1 + \nu_{23}}$$
 (13)

References

- [1] Griffiths R. 1994 Stiffness matrix of the four-node quadrilateral element in closed form Int. J. Numer. Meth. Eng. 57 (2) 109-143
- [2] Krueger R. 2004 Virtual crack closure technique: History, approach, and applications Appl. Mech. Rev. 57 (2) 109-143
- [3] ABAQUS 2016 ABAQUS 2016 Analysis User's Manual Online Documentation Help: Dassault Systémes
- [4] Rice J. R. 1968 A Path Independent Integral and the Approximate Analysis of Strain Concentration by Notches and Cracks J. Appl. Mech. 35 379-386