

Reading Notes Chapter 4 3/14/25

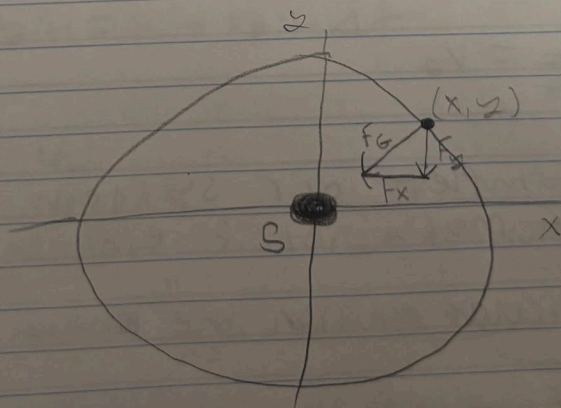
4.1 Kepler's Laws

- According to Newton's law of gravitation the magnitude of this force is:

$$F_g = \frac{GM_S M_E}{r^2}$$

- where M_S and M_E are the masses of the Sun and Earth, r is the distance between them, and G is the grav. constant.

- Our goal is to calculate the pos. of Earth as a function of time.



$$\frac{d^2 x}{dt^2} = \frac{F_{gx}}{M_E} \quad \frac{d^2 y}{dt^2} = \frac{F_{gy}}{M_E}$$

From fig 1. we have

$$F_{gx} = -\frac{GM_s M_E \cos \theta}{r^2} = -\frac{GM_s M_E x}{r^3}$$

- We now follow usual approach and write each of the second-order diff. eq:

$$\frac{dv_x}{dt} = -\frac{GM_s x}{r^3}$$

$$\frac{dx}{dt} = v_x$$

$$\frac{dv_y}{dt} = -\frac{GM_s y}{r^3}$$

$$\frac{dy}{dt} = v_y$$

- To complete our system, we need units of mass.

- For circular motion we know force equals $\frac{MEV^2}{r}$.

$$\frac{MEV^2}{r} = F_g = \frac{GM_s M_E}{r^2}$$

- V = velocity of Earth and rearranging we get:

$$GM_s = v^2 r = 4\pi^2 AU^3 / yr^2 :$$

- We convert equations of motions into diff. eq.

$$v_{x,i+1} = v_{x,i} - \frac{4\pi^2 x_i \Delta t}{r_i^3}$$

$$x_{i+1} = x_i + v_{x,i+1} \Delta t$$

$$v_{z,i+1} = v_{z,i} - \frac{4\pi^2 z_i \Delta t}{r_i^3}$$

$$z_{i+1} = z_i + v_{z,i+1} \Delta t$$

Ex. 4.1

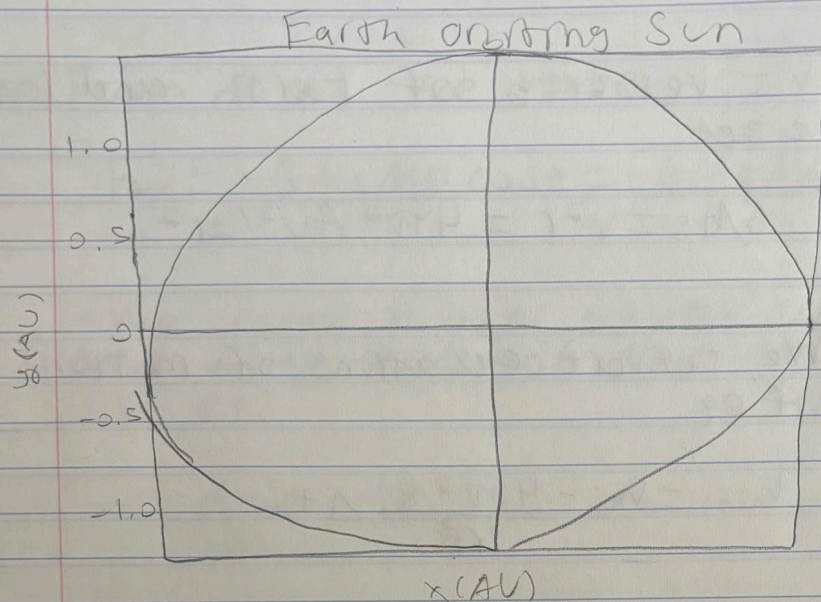
- Calculate distance r_i from Sun: $r_i = (x_i^2 + z_i^2)^{1/2}$

- Compute $v_{x,i+1} = v_{x,i} - \frac{4\pi^2 x_i \Delta t}{r_i^3}$ and $v_{z,i+1} =$

- Euler-cromer: calc. x_{i+1}/z_{i+1} using $v_{x,i+1}/v_{z,i+1}$

- Record new pos. or Plot as available

- Repeat for desired number



4.2 Inverse Square Law & Stability

- We consider a two-body system in which the interaction force depends on separation r .
- Relative motion can be studied as if it were one system.
- The moving body has a mass equal to the reduced mass:

$$\mu \equiv \frac{m_1 m_2}{m_1 + m_2}$$

- The position is given by the relative displacement $\vec{r} \equiv \vec{r}_2 + \vec{r}_1$

- The orbital trajectory for a body of reduced mass is given in polar coords.

$$\frac{d^2}{d\theta^2} \left(\frac{1}{r} \right) + \frac{1}{r} = -\frac{Mr^2 F(r)}{L^2}$$

- $L = Mr^2\dot{\theta}$: angular momentum

- $F(r)$: force acting on the body

$$\rightarrow F(r) = -\frac{GM_S M_P}{r^2}$$

- Since $F(r)$ has the inverse square form $F(r) \propto 1/r^2$ can be solved.

$$\frac{1}{r} = \left(\frac{MG M_S M_P}{L^2} \right) [1 - e \cos(\theta + \theta_0)]$$

or choosing $\theta_0 = 0$

$$r = \left(\frac{L^2}{MG M_S M_P} \right) \frac{1}{1 - e \cos \theta}$$

- Since angular momentum $L = \sqrt{MG M_S M_P a(1 - e^2)}$ is conserved, we may set this equal to $M_{\min} v_{\max}$ and to $M_{\max} v_{\min}$.

$$V_{\max} = \sqrt{GM_S \left[\frac{(1+e)}{a(1-e)} \left(1 + \frac{M_P}{M_S} \right) \right]}$$

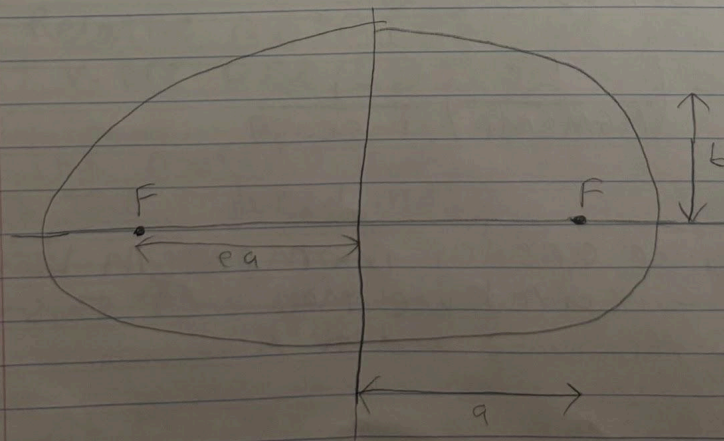
$$V_{\min} = \sqrt{GM_S \left[\frac{(1-e)}{a(1+e)} \left(1 + \frac{M_P}{M_S} \right) \right]}$$

- Orbital Period T can be obtained by dividing the area enclosed by elliptical orbit by constant rate area is swept out

- Kepler's Third Law: $T^2 \propto a^3$

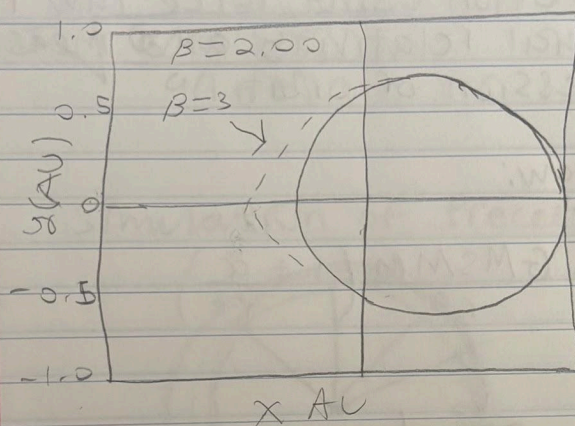
$$T^2 / a^3 = \frac{4\pi^2}{G(M_S + M_P)}$$

$$= \frac{4\pi^2}{GM_S}$$



- Figure 4.3: Hypothetical elliptical orbit.

- As B is closer by 2 orbits.



4.3 Precession of the Perihelion (Mercury)

- We have noted that most planets have orbits that are very nearly circular.

- The planets whose orbits deviate from circular are Mercury and Pluto.

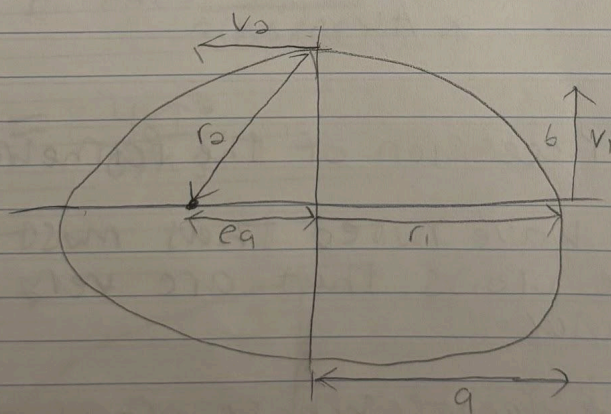
- The Precession of Perihelion (Point nearest to the Sun in orbit); the mag. is approx. 566 arcseconds per century.

- The precession due to general relativity can be calculated analytically, although it is complicated.

- All we have to do is simulate the orbital motion using force law predicted by general relativity and measure rate of precession of orbit.

- Force law:

$$F_g = \frac{G M_s M_m}{r^2} \left(1 + \frac{\alpha}{r^2} \right)$$



- We need to know initial conditions and then we get:

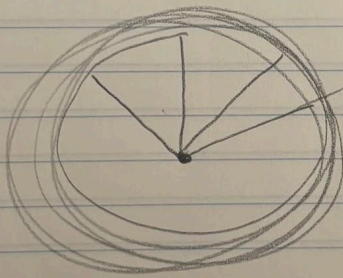
$$-\frac{G M_s M_m}{r_1} + \frac{1}{2} M m v_1^2 = -\frac{G M_s M_m}{r_2} + \frac{1}{2} M m v_2^2$$

$$-r_1 v_1 = b v_\infty$$

$$-v_1 = \sqrt{2GM_s \left[\frac{b^2}{a^2(1+e)^2 - b^2} \right] \left[\frac{1}{\sqrt{e^2 a^2 + b^2}} - \frac{1}{a+ea} \right]}$$

$$= \sqrt{\frac{GM_s(1-e)}{a(1+e)}}$$

Simulation of Precession



4.4 Three-body Problem

- Midterm Project, what we did in class.

