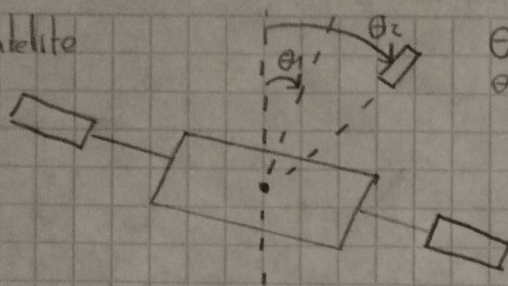
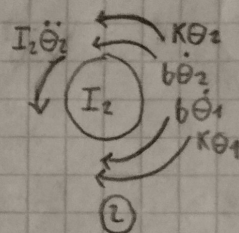
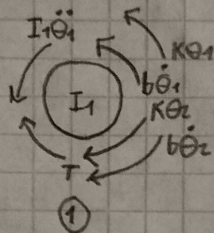
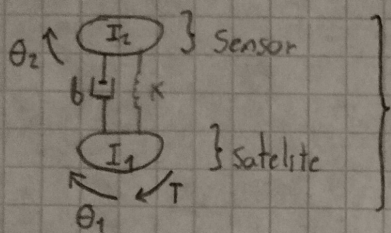


## Problema Satélite



$\theta_2$ : Con respecto al sensor  
 $\theta_1$ : Con respecto a la masa inercial

## Diagrama del problema



## Ecuaciones

$$(1) \quad T = I_1 \ddot{\theta}_1 + K\theta_1 + b\dot{\theta}_1 - K\theta_2 - b\dot{\theta}_2 \quad \ddot{\theta}_1 = \frac{1}{I_1} (T - K\theta_1 - b\dot{\theta}_1 + K\theta_2 + b\dot{\theta}_2)$$

$$(2) \quad I_2 \ddot{\theta}_2 + b\dot{\theta}_2 + K\theta_2 - b\dot{\theta}_1 - K\theta_1 = 0 \quad \ddot{\theta}_2 = \frac{1}{I_2} (b\dot{\theta}_1 + K\theta_1 - K\theta_2 - b\dot{\theta}_2)$$

Tenemos  $q_1 = \theta_1$     $q_2 = \dot{\theta}_1 = \dot{q}_1$     $q_3 = \theta_2$     $q_4 = \dot{\theta}_2 = \dot{q}_3$     $y_1 = \theta_1 = q_1$   
 $\dot{q}_2 = \ddot{\theta}_1$     $\dot{q}_4 = \ddot{\theta}_2$     $y_2 = \theta_2 = q_3$

## Reemplazando

$$(1) \quad \dot{q}_2 = \frac{1}{I_1} (T - Kq_1 - bq_2 + Kq_3 + bq_4) \quad (2) \quad \dot{q}_4 = \frac{1}{I_2} (bq_2 + Kq_1 - Kq_3 - bq_4)$$

## Representación en espacio de estados

$$\begin{bmatrix} \dot{q}_1 \\ \dot{q}_2 \\ \dot{q}_3 \\ \dot{q}_4 \end{bmatrix} = \begin{bmatrix} 0 & 1 & 0 & 0 \\ -K/I_1 & -b/I_1 & K/I_1 & b/I_1 \\ 0 & 0 & 0 & 1 \\ K/I_2 & b/I_2 & -K/I_2 & -b/I_2 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \\ q_3 \\ q_4 \end{bmatrix} + \begin{bmatrix} 0 \\ 1/I_1 \\ 0 \\ 0 \end{bmatrix} T$$

$$\begin{bmatrix} y_1 \\ y_2 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} q_1 \\ q_2 \\ q_3 \\ q_4 \end{bmatrix}$$