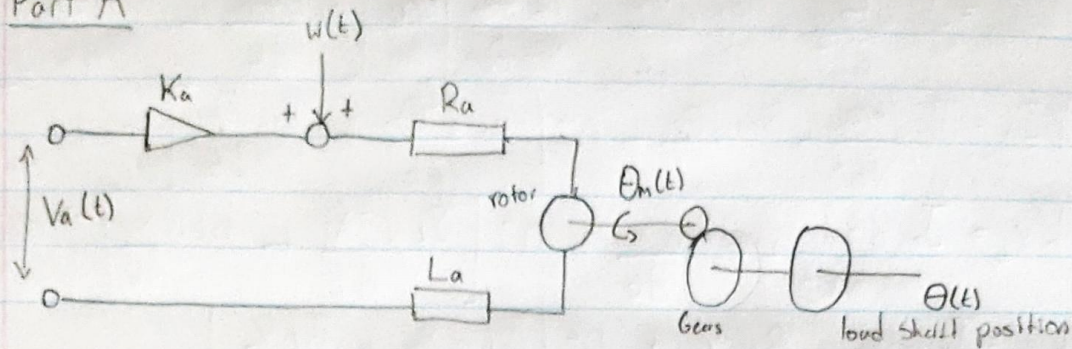


Lab 3: Preparatory Labwork

Part A



x_1 = Angular position of the motor shaft

x_2 = " velocity of the motor shaft

x_3 = Armature current (I_a)

$u = V_a$

y = Angular position of the load shaft

Apply KVL to the equation,

$$L_a \frac{dI_a}{dt} + R_a I_a + E_b = E_a \rightarrow \text{We know } E_a \text{ and } E_b$$

$$L_a \frac{dI_a}{dt} + R_a I_a + \Theta_m K_b = V_a(t) \cdot K_a \rightarrow \text{put it in terms of } x_3$$

$$L_a \frac{dx_3}{dt} + R_a x_3 + x_2 K_b = U \cdot K_a$$

$$\frac{dx_3}{dt} = -\frac{R_a x_3}{L_a} - \frac{x_2 K_b}{L_a} + \frac{U \cdot K_a}{L_a} \rightarrow \text{solved in terms of } x_3$$

By apply newtons law of motion,

$$T_m = J_m \frac{d^2 \Theta_m}{dt^2} + B_m \frac{d\Theta_m}{dt} + T_l \rightarrow \text{we know } T_2 = \frac{\Theta_m}{\Theta_l} T_l$$

$$T_m = J_m \frac{d^2 \Theta_m}{dt^2} + B_m \frac{d\Theta_m}{dt} + \frac{\Theta_l}{\Theta_m} T_2 \rightarrow \text{we know } T_2$$

$$T_m = J_m \frac{d^2 \Theta_m}{dt^2} + B_m \frac{d\Theta_m}{dt} + \frac{\Theta_l}{\Theta_m} \left[J_l \frac{d^2 \Theta_l}{dt^2} + B_l \frac{d\Theta_l}{dt} \right] \rightarrow \text{We know } T_m \text{ and } \Theta_l$$

$$KI_a = J_m \frac{d^2 \theta_m}{dt^2} + B_m \frac{d\theta_m}{dt} + \frac{n \theta_m}{\theta_m} \left[n \left[J_e \frac{d^2 \theta_m}{dt^2} + B_e \frac{d\theta_m}{dt} \right] \right]$$

$$KI_a = J_m \frac{d^2 \theta_m}{dt^2} + B_m \frac{d\theta_m}{dt} + n^2 \left[J_e \frac{d^2 \theta_m}{dt^2} + B_e \frac{d\theta_m}{dt} \right] \rightarrow \text{put in terms of states}$$

$$KX_3 = J_m \frac{d^2 x_1}{dt^2} + B_m \frac{dx_1}{dt} + n^2 \left[J_e \frac{d^2 x_1}{dt^2} + B_e \frac{dx_1}{dt} \right] \rightarrow \frac{dx_1}{dt} = x_2$$

$$KX_3 = J_m \frac{dx_2}{dt} + B_m x_2 + n^2 \left[J_e \frac{dx_2}{dt} + B_e x_2 \right]$$

$$KX_3 = J_m \frac{dx_2}{dt} + B_m x_2 + n^2 J_e \frac{dx_2}{dt} + n^2 B_e x_2$$

$$\frac{dx_2}{dt} = \frac{K}{J_{eq}} x_3 - \frac{B_{eq}}{J_{eq}} x_2 \rightarrow \text{Simplified equation}$$

Since the derivative of velocity is position, we know that

$$\frac{dx_1}{dt} = x_2$$

\therefore From the equations derived, we can put the values into state system form

\rightarrow

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \end{bmatrix} = \underbrace{\begin{bmatrix} 0 & 1 & 0 \\ 0 & -B_q/J_q & K/J_q \\ 0 & -W/L & -R/L \end{bmatrix}}_A \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} + \underbrace{\begin{bmatrix} 0 \\ 0 \\ 1/L \end{bmatrix}}_B u$$

$$y = \underbrace{\begin{bmatrix} 0 & 1 & 0 \end{bmatrix}}_C \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}$$

2.

Using a sampling frequency of 150 Hz.

$\text{sysc} = \text{ss}(A, B, C, D) \rightarrow$ file attached

$\text{sysd} = \text{c2d}(\text{sysc}, 1/150) \leftarrow$ sampling period

By solving for the equivalent discrete-time ss model, the result is:

$$x_{k+1} = \Phi x_k + \Gamma u_k$$

$$y_k = C x_k + D u_k$$

$$\Phi = \begin{bmatrix} 1 & 0.005875 & 0.005595 \\ 0 & 0.7706 & 0.7402 \\ 0 & -0.001567 & -0.001506 \end{bmatrix}$$

$$\Gamma = \begin{bmatrix} 0.1288 \\ 37.3 \\ 0.2581 \end{bmatrix}$$

\rightarrow

To check if the system is stable, we need to find the poles of the system:

System poles = eigenvalues of Φ

Using MATLAB;

$$\text{eig } \Phi = \begin{Bmatrix} 1.00 \\ 0.7691 \\ 0 \end{Bmatrix}$$

Not all the poles are inside the unit circle therefore the system is not stable.

To check if the system is controllable, we need to solve for the controllability matrix. The controllability matrix is:

$$W_c = [\Gamma \quad \Phi\Gamma \quad \Phi^2\Gamma]$$

In this case, $\text{rank}(W_c) = 3$, but we can verify if the system is controllable if the $\det(W_c) \neq 0$. With MATLAB, $\det(W_c) = -2.4190$ \therefore the system is controllable.

To check if the system is observable, we need to solve for the observability matrix. The observability matrix is:

$$W_o = \begin{bmatrix} C \\ C\Phi \\ C\Phi^2 \end{bmatrix}$$

\rightarrow

In this case, the $\text{rank}(W_0) = 3$ and the $\det(W_0) = 2.17 \times 10^{-7}$
 \therefore Since the $\det(W_0) \neq 0$, the system is therefore observable.

PART B

Discrete Time Model:

$$\begin{aligned} X_{k+1} &= \Phi X_k + \Gamma U_k & \text{Where } X_k &= [X_{1k}, X_{2k}, X_{3k}]^T \\ Y_k &= C X_k + D U_k \end{aligned}$$

Design a feedback controller:

$$U_k = \bar{N} r_k - K x_k$$

W/ $PO \leq 15\%$ and Settling time: $T_s(1\%) \leq 0.25s$.

$$\xi = \frac{-\ln(PO/100)}{\sqrt{\pi^2 + \ln^2(PO/100)}} = \frac{-\ln(15/100)}{\sqrt{\pi^2 + \ln^2(15/100)}} = 0.5169$$

$$T_s(1\%) \leq 0.25 \rightarrow \frac{4.6}{\xi \omega_n} = 0.25 \rightarrow \frac{4.6}{(0.5169)\omega_n} = 0.25 \rightarrow \omega_n = 35.60$$

\therefore The dominant CL poles are:

$$s_d = -\xi \omega_n \pm j \omega_n \sqrt{1 - \xi^2} = 18.4 \pm j 30.465$$

\rightarrow

However, we can use the Bessel transfer function table to solve for the pole locations

$\phi_{cl}(s)$ for 3rd order function is:

$$\phi_{cl}(s) = (s + 0.9420)(s + 0.7455 \pm 0.7112j) \rightarrow \text{for } \omega_n = 1$$

\hookrightarrow Put in terms of $\omega_n = 35.6$

$$\begin{aligned} S_d &= \omega_n \times (s + 0.9420)(s + 0.7455 \pm 0.7112j) \\ &= 35.6 \times (s + 0.9420)(s + 0.7455 \pm 0.7112j) \\ &= (s + 33.5352)(s + 26.5398 \pm 25.3187j) \end{aligned}$$

$$z_d = e^{s_d T} = (0.7997 + 0j)(0.8259 \pm 0.1407j)$$

$$\phi_{cl}^d(z) = [1.00 \quad -2.4515 \quad 2.0229 \quad -0.5613]$$

$$K_n = [0.0401 \quad -0.0033 \quad -2.1844]$$

$$\bar{N} = 0.0401$$

\rightarrow

Part C

The servo will have a non-zero steady state error due to the presence of a dc offset disturbance, w_k .

$$\therefore U_k = \bar{N}r_k - K\hat{x}_k - \hat{w}_k$$

→ Since the disturbance is a dc offset, assume the disturbance to be constant. In this case, the OL system becomes the plant + integration and LCL poles are required.

Therefore, the Bessel poles in Part B and an extra pole will be used at $-4\%W_n \rightarrow s_c = -52.9306 \rightarrow z_c = 0.7027$

(OL) System model = Plant + Integrator

$$\begin{bmatrix} x_{1,k+1} \\ x_{k+1} \end{bmatrix} = \begin{bmatrix} 1 & -C \\ 0 & \Phi \end{bmatrix} \begin{bmatrix} x_{1,k} \\ x_k \end{bmatrix} + \begin{bmatrix} 0 \\ \Gamma \end{bmatrix} U_k + \begin{bmatrix} 0 \\ \Gamma \end{bmatrix} w_k + \begin{bmatrix} 1 \\ 0 \end{bmatrix} r_k$$

$$y_k = \begin{bmatrix} 0 & C \end{bmatrix} \begin{bmatrix} x_{1,k} \\ x_k \end{bmatrix}$$

$$\therefore \begin{aligned} X_{k+1}^A &= \Phi_k^A X_k^A + \Gamma_k^A U_k + \Gamma_k^A w_k + \Gamma_k^B r_k \\ y_k &= C_k^A X_k^A \end{aligned}$$

Controller:

$$U_k = \bar{N}r_k - K_A X_k^A$$

$$K^A = \begin{bmatrix} K_I & K \end{bmatrix}$$

\uparrow \uparrow
 1×1 1×3

$$\begin{aligned} K^A &= [0 \ 0 \ 0 \ 1] (W_c^A)^{-1} \phi_{cl}^d(\Phi^A) \\ K^A &= -0.015 \quad 0.2022 \quad 0.006 \quad -1.3349 \end{aligned}$$

→

There are a total of 3 possibilities for \bar{N} :

a) $\bar{N} = N_u + KN_x = 0.2022$

b) $\bar{N} = \emptyset$

c) $\bar{N} = \frac{K_I}{Z_c - 1} = 0.04101$