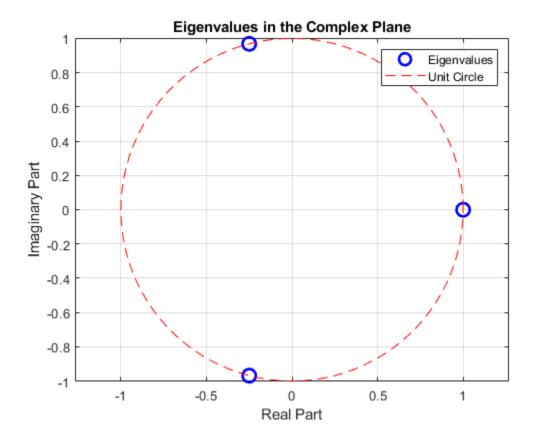
```
% ARO 4090 - Space Vehicle Dyn. & Cntrl. | Dr. Maggia | Justin Millsap |
Homework 5 %
clc; clear; close all;
```

Problem 1

```
clc; clear; close all
R = [0 \ 1/2 \ sqrt(3)/2 \ ; -1 \ 0 \ 0 \ ; \ 0 \ -sqrt(3)/2 \ 1/2];
I = [1 \ 0 \ 0 ; \ 0 \ 1 \ 0 ; \ 0 \ 0 \ 1];
% ~~~~~~~~ PART A ~~~~~~ %
% Show that R is a rotation Matrix
    % Must satisfy det(R) = 1
    % & RR^T = I
condition 1 = det(R);
condition 2 = R*R';
% ~~~~~~~ PART B ~~~~~~ %
% Find eigenvalues
eigenvalues = eig(R)
% Plot the eigenvalues on the complex plane
figure; % Opens a new figure window
plot(real(eigenvalues), imag(eigenvalues), 'bo', 'MarkerSize', 10,
'LineWidth', 2);
xlabel('Real Part');
ylabel('Imaginary Part');
title('Eigenvalues in the Complex Plane');
grid on;
axis equal;
hold on;
% Additionally, plot the unit circle for reference
th = 0:pi/50:2*pi;
xunit = cos(th);
yunit = sin(th);
plot(xunit, yunit, 'r--'); % Unit circle in red dashed line
legend('Eigenvalues', 'Unit Circle');
hold off;
% ~~~~~~ PART C ~~~~~~ %
% Find the angle of rotation (alpha)
alpha = acosd((trace(R) - 1) / 2)
% Find axis of rotation (a hat)
```

```
a = [-1 ; 1 ; -sqrt(3)];
a hat = -a/norm(a); % Normalize the vector
fprintf('The angle of rotation is alpha = %.2f degrees ',alpha)
disp(' ')
disp('The axis of rotation is a=')
disp(a hat)
% ~~~~~~~ PART D ~~~~~~ %
% Use alpha and unit vector a to retrive Matrix R
a \times = [0 - a hat(3,1) a hat(2,1); a hat(3,1) 0 - a hat(1,1); -a hat(2,1)
a hat(1,1) 0]
R prime = cosd(alpha)*I + (1 - cosd(alpha)) * a hat * a hat' -
sind(alpha) *a x
% ~~~~~~~ PART E ~~~~~~ %
% Solve for Eurler Angles by using most opitmal positions to compare between
R BI & Rotation Sequence Matrix
theta = asind(-R(1,3)) + 360;
    = atan2d(R(2,3), R(3,3));
phi
psi = atan2d(R(1,2), R(1,1));
% Check if Eurler angles work
R1 = [100 ; 0 cosd(phi) sind(phi) ; 0 -sind(phi) cosd(phi)];
R2 = [\cos d(theta) \ 0 - \sin d(theta); \ 0 \ 1 \ 0; \ \sin d(theta) \ 0 \ \cos d(theta)];
R3 = [\cos d(psi) \sin d(psi) 0 ; -\sin d(psi) \cos d(psi) 0 ; 0 0 1];
Rot Seq = R1*R2*R3;
                            % [3-2-1] Rotation Sequence
disp(' The Eurler Angles for the given R Matrix for a given Rotation
Sequence are')
fprintf('psi = %.2f\n', psi)
fprintf('theta = %.2f\n', theta)
fprintf('phi = %.2f\n', phi)
eigenvalues =
  -0.2500 + 0.9682i
  -0.2500 - 0.9682i
  1.0000 + 0.0000i
```

```
alpha =
 104.4775
The angle of rotation is alpha = 104.48 degrees
The axis of rotation is a=
   0.4472
  -0.4472
   0.7746
a x =
       0
          -0.7746 -0.4472
   0.7746 0 -0.4472
   0.4472
           0.4472
R prime =
           0.5000
                      0.8660
        0
  -1.0000
                      0.0000
            0
  -0.0000 -0.8660
                     0.5000
 The Eurler Angles for the given R Matrix for a given Rotation Sequence are
psi = 90.00
theta = 300.00
phi = 0.00
```



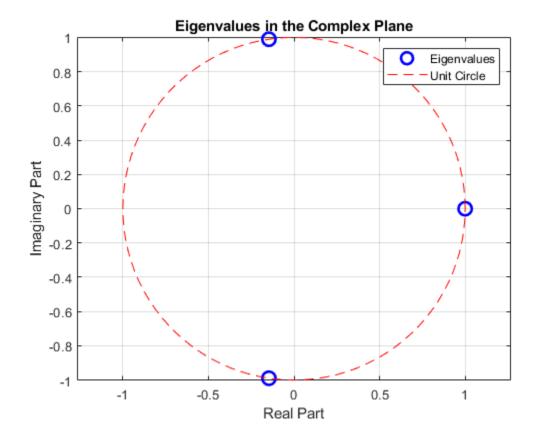
Problem 2

```
clc; clear; close all
I = [1 \ 0 \ 0 \ ; \ 0 \ 1 \ 0 \ ; \ 0 \ 0 \ 1];
% ~~~~~~~~ PART A ~~~~~~~ %
% Show that R is a rotation Matrix
   % Must satisfy det(R) = 1
   % & RR^T = I
condition 1 = det(R)
condition_2 = R*R'
% ~~~~~~~~ PART B ~~~~~~ %
% Find eigenvalues
eigenvalues = eig(R)
% Plot the eigenvalues on the complex plane
figure; % Opens a new figure window
plot(real(eigenvalues), imag(eigenvalues), 'bo', 'MarkerSize', 10,
'LineWidth', 2);
```

```
xlabel('Real Part');
ylabel('Imaginary Part');
title('Eigenvalues in the Complex Plane');
grid on;
axis equal;
hold on;
% Additionally, plot the unit circle for reference
th = 0:pi/50:2*pi;
xunit = cos(th);
yunit = sin(th);
plot(xunit, yunit, 'r--'); % Unit circle in red dashed line
legend('Eigenvalues', 'Unit Circle');
hold off;
% ~~~~~~ PART C ~~~~~~ %
% Find the angle of rotation (alpha)
alpha = acosd((trace(R) - 1) / 2)
a = [1 ; 0 ; 0]
% Find axis of rotation (a hat)
if alpha >= 0
    if alpha <= 180
       a = (1/(2*sind(alpha))) * [R(2,3) - R(3,2) ; R(3,1) - R(1,3) ;
R(1,2) - R(2,1)
    end
end
a hat = a/norm(a)
% Normalize the vector
fprintf('The angle of rotation is alpha = %.2f degrees ',alpha)
disp(' ')
disp('The axis of rotation is a=')
disp(a hat)
% ~~~~~~~ PART D ~~~~~~ %
% Use alpha and unit vector a to retrive Matrix R
a \times = [0 - a hat(3,1) a hat(2,1) ; a hat(3,1) 0 - a hat(1,1) ; -a hat(2,1)
a hat(1,1) 0]
R prime = cosd(alpha)*I + (1 - cosd(alpha)) * a hat * a hat' -
sind(alpha) *a x
% ~~~~~~~ PART E ~~~~~~ %
```

```
% Solve for Eurler Angles by using most opitmal positions to compare between
R BI & Rotation Sequence Matrix
theta = asind(-R(1,3));
phi = atan2d(R(2,3), R(3,3));
psi = atan2d(R(1,2), R(1,1));
% Check if Eurler angles work
R1 = [1 \ 0 \ 0 \ ; \ 0 \ cosd(phi) \ sind(phi) \ ; \ 0 \ -sind(phi) \ cosd(phi)];
R2 = [cosd(theta) \ 0 - sind(theta); \ 0 \ 1 \ 0; \ sind(theta) \ 0 \ cosd(theta)];
R3 = [\cos d(psi) \sin d(psi) 0 ; -\sin d(psi) \cos d(psi) 0 ; 0 0 1];
Rot Seq = R1*R2*R3; % [3-2-1] Rotation Sequence
disp(' The Eurler Angles for the given R Matrix for a given Rotation
Sequence are')
fprintf('psi = %.2f\n', psi)
fprintf('theta = %.2f\n', theta)
fprintf('phi = %.2f\n', phi)
condition 1 =
    1.0000
condition 2 =
    1.0000
                            0
              0
            1.0000
                              0
              0
                        1.0000
eigenvalues =
  -0.1464 + 0.9892i
  -0.1464 - 0.9892i
  1.0000 + 0.0000i
alpha =
   98.4211
a =
     1
     0
     0
```

```
a =
    0.3574
    0.8629
    0.3574
a hat =
    0.3574
    0.8629
    0.3574
The angle of rotation is alpha = 98.42 degrees
The axis of rotation is a=
   0.3574
    0.8629
    0.3574
a_x =
        0 -0.3574
                      0.8629
    0.3574
                     -0.3574
             0
   -0.8629 0.3574
R prime =
    0.0000
            0.7071
                     -0.7071
             0.7071
                      0.7071
        0
    1.0000
             0.0000
                       0.0000
The Eurler Angles for the given R Matrix for a given Rotation Sequence are
psi = 90.00
theta = 45.00
phi = 90.00
```



Published with MATLAB® R2023b