

# Linear stability analysis of galactic outflows in the cold dark matter halo

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Collaborators

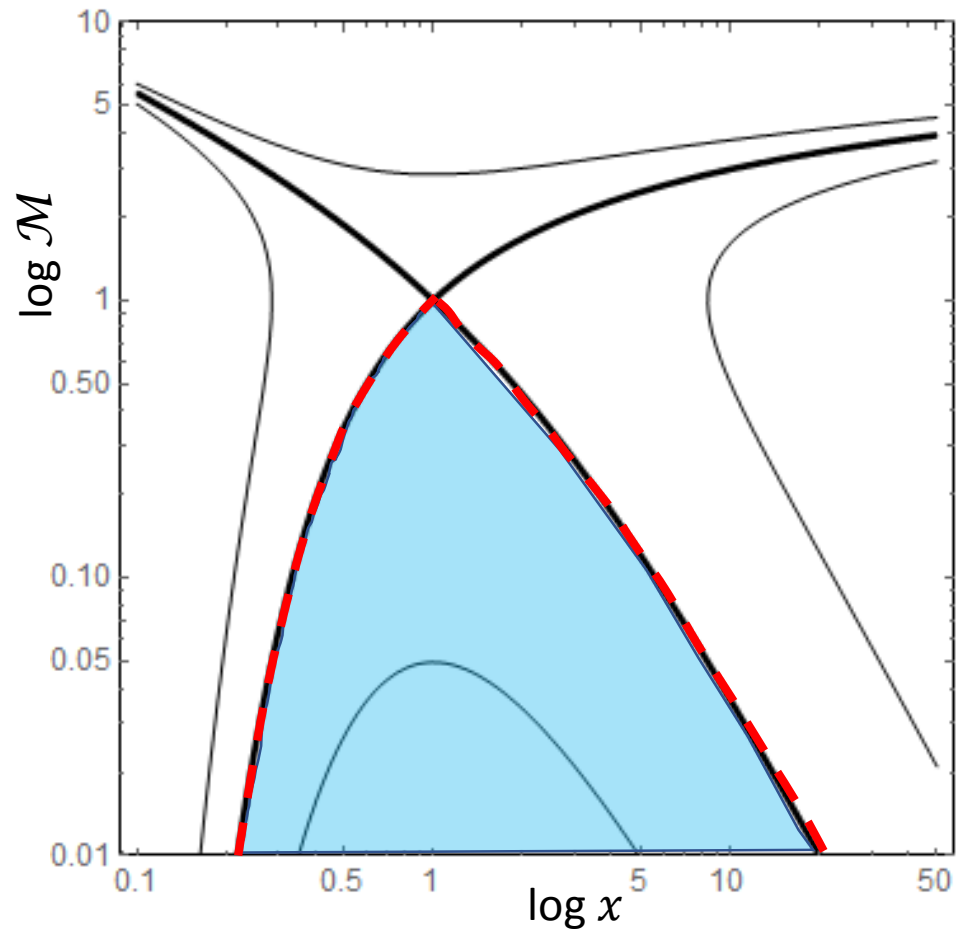
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# Introduction – Solution curves

- Transonic flow (bold line)
  - Flow continuously accelerates from subsonic to supersonic through a sonic point.
  - Transonic flow may form a terminal shock.
- Breeze (blue region)
  - Flow is always subsonic.
- Critical breeze (red dashed curve)
  - The flow accelerates until  $\mathcal{M}=1$  along the transonic solution and transitions to another transonic solution at the sonic points.



# Purpose of this study

- We investigate the linear stability analysis of galactic breezes to explore the fundamental physical properties of the galactic outflows in a cold dark matter halo.
- We want to show the importance of the transonic outflows and the relevance between the density profile of dark matter halo (DMH) and the stability of the galactic winds.

# Model – Basic equations

- Steady, spherically symmetric, isothermal outflows

➤ Basic equations

$$\frac{\partial}{\partial x}(\rho v x^2) = 0 \quad v \frac{\partial v}{\partial x} = -\frac{1}{\rho} \frac{\partial p}{\partial x} - \frac{\partial \Phi}{\partial x} \quad p = c_s^2 \rho$$



➤ Equation of base flow

$$\left( \mathcal{M} - \frac{1}{\mathcal{M}} \right) \frac{d\mathcal{M}}{dx} = \frac{2}{x} - \frac{1}{c_s^2} \frac{d\Phi(x)}{dx}$$



$x = r/r_s$   
 $r_s$  : scale radius of DMH  
 $c_s$  : sound velocity (constant)  
 $\Phi$  : gravitational potential  
 $x_0$  : start point of flows  
 $\mathcal{M}_0$  : Mach number at start point

$$\mathcal{M} = \begin{cases} \sqrt{-W_0 \left( -\frac{\mathcal{M}_0^2 x_0^4}{x^4} \exp \left( -\mathcal{M}_0^2 + 2(\Phi(x) - \Phi(x_0)) \right) \right)} & \text{(for subsonic)} \\ \sqrt{-W_{-1} \left( -\frac{\mathcal{M}_0^2 x_0^4}{x^4} \exp \left( -\mathcal{M}_0^2 + 2(\Phi(x) - \Phi(x_0)) \right) \right)} & \text{(for supersonic)} \end{cases}$$

➤ Lambert W function  
 defined the inverse function of  $y = x \exp(x)$

# Model – Non-perturbed state

RHS of equation of base flow

$$\frac{2}{x} - \frac{1}{c_s^2} \frac{d\Phi(x)}{dx} = \frac{2}{x} - K_{\text{DMH}} f(x)$$

$f(x)$  : function obtained by  
integrating DMH density profile

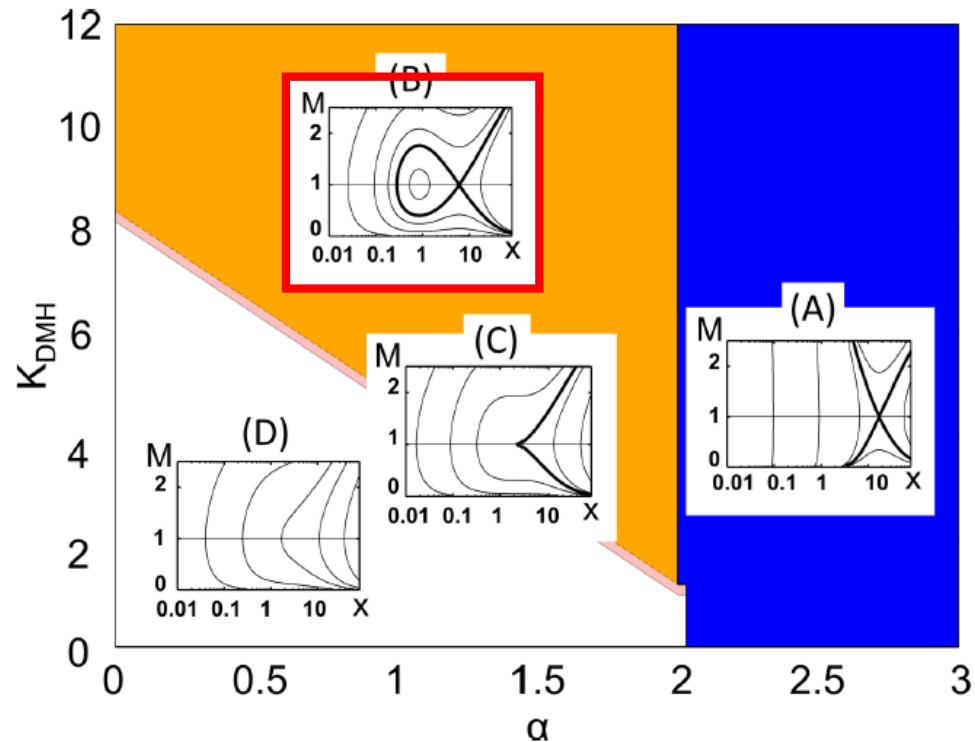
➤ DMH density profile

$$\rho(x) = \begin{cases} \frac{\rho_d}{x(x+1)^2} & \text{(NFW profile)} \\ \frac{\rho_d}{(x+1)^3} & \text{(core profile)} \end{cases}$$

➤ Parameter K

$$K_{\text{DMH}} = \frac{2\pi\rho_d r_d^2 G}{c_s^2} \approx \frac{\text{gravitational potential}}{\text{thermal energy}}$$

Using  $K=10.0$ , to consider transonic solution and compare two type DM profile.



Igarashi et al. (2013)

# Model - Perturbation

Applying the solar wind model given by Velli (2001),

➤ Perturbation  $y^\pm = \widehat{\mathcal{M}} \pm \hat{p} = y^\pm(x) \exp\{(-i\omega + \gamma)t\}$

$y^\pm$  are the conserved quantities (Riemann invariants) along characteristics.

➤ Perturbation equation

$$(\mathcal{M} \pm 1)y^{\pm'} + (-i\omega + \gamma)y^\pm + \frac{1}{2}(y^{\mp} + y^\pm) \frac{\mathcal{M}'}{\mathcal{M}} (\mathcal{M} \mp 1) = 0$$

Assumption

① : long wavelength approximation ( $\omega = 0$ )

② :  $y^+(x_0) = y^-(x_0) \neq 0$

③ :  $|y^+(x_1)|, |y^-(x_1)| \rightarrow 0 \quad (x_1 \gg 1)$

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0 : start point of flows

1 : very large distance from galactic center

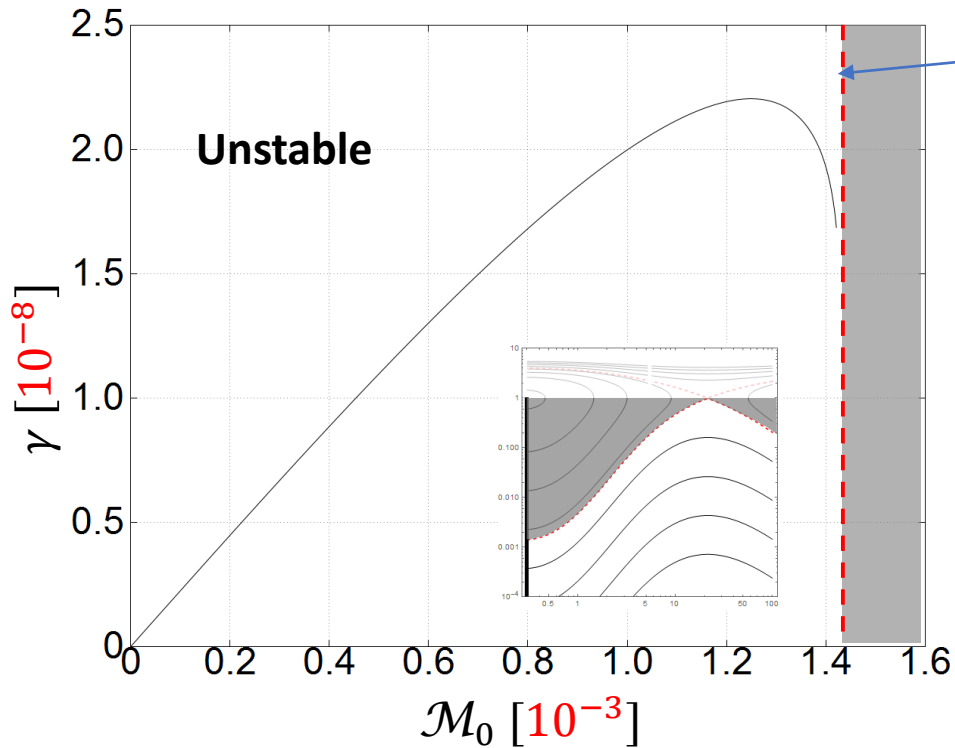
➤ Growth rate of the instability

$$\gamma = \frac{2(|y^+|_{x_0}^2 - |y^+|_{x_1}^2)}{\int_{x_0}^{x_1} \mathcal{M}^{-1} [(\mathcal{M} + 1)|y^+|^2 - (\mathcal{M} - 1)|y^-|^2] dx}$$

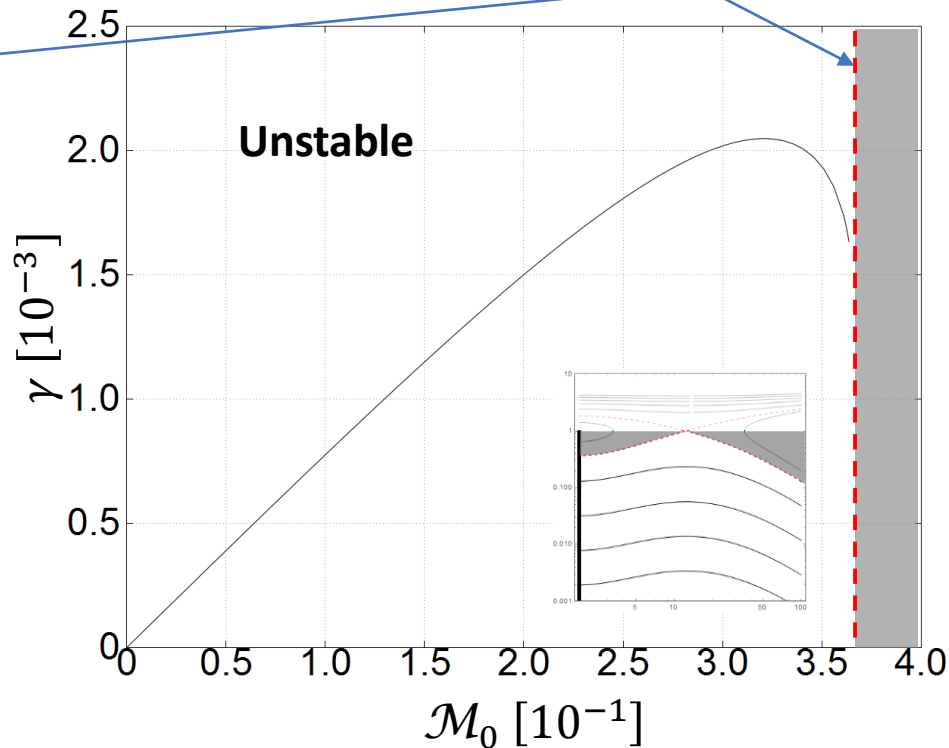
**Estimate  $\gamma$  for  $\mathcal{M}_0$ , Mach number at the start point, of each breeze.**

# Result – the growth rate $\gamma$

## NFW profile



## Core profile



Critical breeze  
/ Transonic flow

- $\gamma > 0 \rightarrow$  Unstable
- These galactic breezes are always unstable.
- $\gamma$  for NFW profile is much smaller than that for core profile.

$$\gamma \sim \begin{cases} 10^{-8} & \text{(NFW)} \\ 10^{-3} & \text{(core)} \end{cases}$$

# Discussion - Growth time of perturbation

$$y^\pm = y^\pm(x) \exp(\gamma t) \longrightarrow t = \frac{t'}{t_s} \longrightarrow t_{\text{grow}} = \frac{(r_s/c_s)}{\gamma} \quad \gamma \sim \begin{cases} 10^{-8} & (\text{NFW}) \\ 10^{-3} & (\text{core}) \end{cases}$$

Sound crossing time

$M_{\text{halo}}$	$z$	$r_s[\text{kpc}]$	Growth time $t_{\text{grow}} [\text{yr}]$	
			NFW	Core
$10^8 M_\odot$	0	2.4	$7.8 \times 10^{14}$	$7.8 \times 10^9$
	9	0.2	$7.7 \times 10^{13}$	$7.7 \times 10^8$
$10^{10} M_\odot$	0	11	$3.5 \times 10^{15}$	$3.5 \times 10^{10}$
	9	1.1	$3.5 \times 10^{14}$	$3.5 \times 10^9$
$10^{12} M_\odot$	0	51	$1.6 \times 10^{16}$	$1.6 \times 10^{11}$
	9	5.1	$1.6 \times 10^{15}$	$1.6 \times 10^{10}$

$$c_s \cong 300 \text{ km/s}$$

NFW model:

- The growth times are far longer than the age of the universe.
- So, the breeze is mathematically unstable, but it is virtually stable.
- On the other hand, very small  $\mathcal{M}_0$  ( $\sim 10^{-3} \sim 10^{-4}$ ), Mach number at the start point, indicates that the breeze hardly exists in actual galaxies.

Core model:

- In less-massive high- $z$  galaxies, the breezes are physically unstable; therefore, they could not exist during the galactic evolution.

**Transonic galactic outflows play an essential role in galactic evolution.**



# Summary

- We investigate the stability of galactic breezes.
- In NFW profile, the breeze is mathematically unstable but virtually stable. Besides, for very small  $\mathcal{M}_0$ , the breeze hardly exists in actual galaxies.
- In Core profile, the breezes in less-massive high- $z$  galaxies is physically unstable.
- Transonic outflows are essential to galactic evolution.

## Future work

- Liner stability analysis for transonic outflows and inflows with a termination shock.
- Comparing with observations.

# Appendix

# Acceleration mechanism (1)

## Accelerate mechanism : Laval nozzle

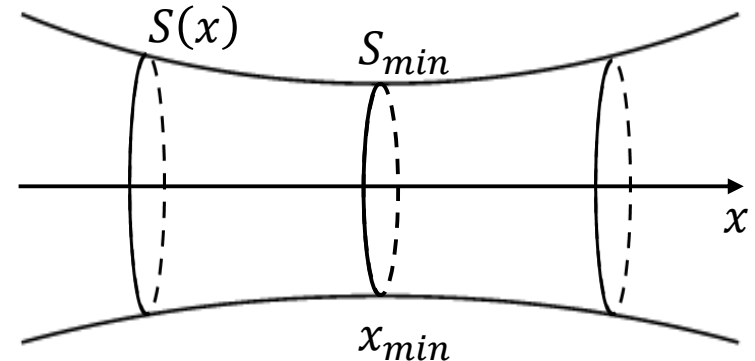
Fluid equation

$$(\mathcal{M}^2 - 1) \frac{1}{v} \frac{dv}{dx} = \frac{1}{S} \frac{dS(x)}{dx}$$

$\mathcal{M}$  : Mach number ( $= v/c_s$ )

$c_s$  : sound velocity

$S$  : cross section of nozzle



For subsonic ( $\mathcal{M} < 1$ ),

If  $dS/dx < 0$ , fluid is accelerated.

If  $dS/dx > 0$ , fluid is decelerated.

For supersonic ( $\mathcal{M} > 1$ ), change of cross section affects inverted effect for fluid.

$dS/dx = 0$  at  $x = x_{min}$ , if  $\mathcal{M} = 1$  then,  $dv/dx$  doesn't decided uniquely.

**This point,  $(x, \mathcal{M}) = (x_{min}, 1)$  called transonic point.**

# Acceleration mechanism (2)

For galactic wind,

Fluid equation

$$(\mathcal{M}^2 - 1) \frac{1}{v} \frac{dv}{dx} = \frac{1}{S} \frac{dS(x)}{dx}$$



$$\left( \mathcal{M} - \frac{1}{\mathcal{M}} \right) \frac{d\mathcal{M}}{dx} = \boxed{\frac{2}{x}} - \boxed{\frac{1}{c_s^2} \frac{d\Phi(x)}{dx}}$$

Expansion

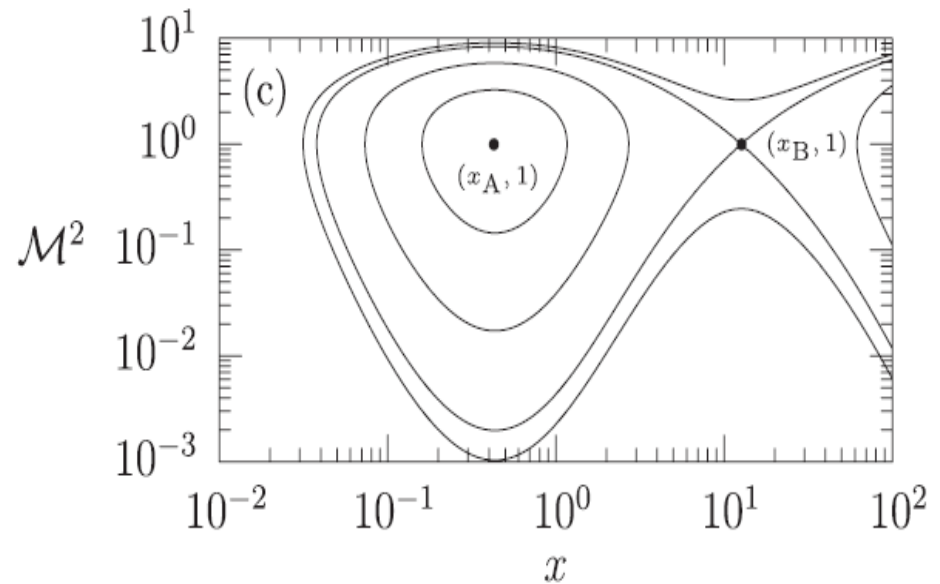
Contraction

$\Phi(x)$  : gravitational potential

Right hand side shows magnitude correlation between thermal energy and gravitational potential.



Balance of two energies decides the acceleration of wind.



(Tsuchiya et al. 2013)

# Previous work (1) – Velli (2001)

- This work investigated the instability of breeze for steady, spherically symmetric, isothermal solar wind.

## Fundamental equations of flow

Mass conservation  $\frac{\partial}{\partial r}(\rho v r^2) = 0$

Equation of motion  $v \frac{\partial v}{\partial r} = -\frac{1}{\rho} \frac{\partial p}{\partial r} - \frac{GM_{\odot}}{r^2}$

$$p = c_s^2 \rho$$

Perturbations are linear combination of Mach number  $\widehat{\mathcal{M}}$  and pressure  $\hat{p}$ .

$$y^{\pm} = \widehat{\mathcal{M}} \pm \hat{p} = y^{\pm}(r) \exp\{(-i\omega + \gamma)t\}$$

## Equations of perturbation

$$(\mathcal{M} \pm 1)y^{\pm'} + (-i\omega + \gamma)y^{\pm} + \frac{1}{2}(y^{\mp} + y^{\pm})\frac{\mathcal{M}'}{\mathcal{M}}(\mathcal{M} \mp 1) = 0$$

# Previous work (2) - Velli (2001)

Assumption

- ① : long wavelength approximation ( $\omega = 0$ )
- ② :  $y^+(0) = y^-(0) \neq 0$
- ③ :  $|y^+(r_1)|, |y^-(r_1)| \rightarrow 0$  ( $r_1 \gg 1$ )

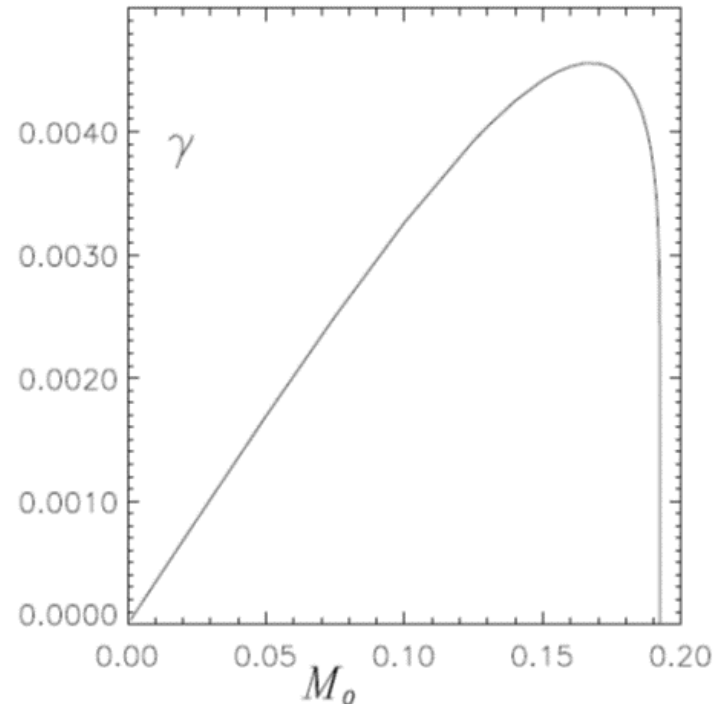
**Instability growth rate**

$$\gamma = \frac{2(|y^+|_0^2 - |y^+|_{r_1}^2)}{\int_{r_0}^{r_1} \mathcal{M}^{-1}[(\mathcal{M} + 1)|y^+|^2 - (\mathcal{M} - 1)|y^-|^2]}$$

$\gamma > 0 \rightarrow$  flow is stable.

$\gamma < 0 \rightarrow$  flow is instable.

**Breeze is instable for solar wind.**



(Velli 2001)

# Model – gravitational source

- Gravitational source : DMH only

DMH density profile

$$\rho(x; \alpha, \beta, \gamma) = \frac{\rho_d}{x^\alpha (x^\beta + 1)^\gamma}$$

$\rho_d$  : scale density  
 $\alpha, \beta, \gamma$  : power law index

In this work, density profile assumes NFW profile and core profile.

$$\rightarrow \text{NFW} : \alpha = 1, \beta = 1, \gamma = 2$$

$$\text{core} : \alpha = 0, \beta = 1, \gamma = 3$$

Gravitational potential and force for each profile

NFW profile	Core profile
$\frac{d\Phi(x)}{dx} = 4\pi\rho_d r_d^2 G \left( \ln(x+1) - \frac{x}{x+1} \right)$ $\Phi(x) = -4\pi\rho_d r_d^2 G \frac{\log(x+1)}{x}$	$\frac{d\Phi(x)}{dx} = \frac{4\pi\rho_d r_d^2 G}{x^2} \left( \ln(x+1) - \frac{x(3x+2)}{2(x+1)^2} \right)$ $\Phi(x) = -4\pi\rho_d r_d^2 G \left( \frac{\log(x+1)}{x} - \frac{1}{2(x+1)} \right)$

# Function of Mach number (1)

- To formulate Mach number of  $x$ , using Lambert W function.

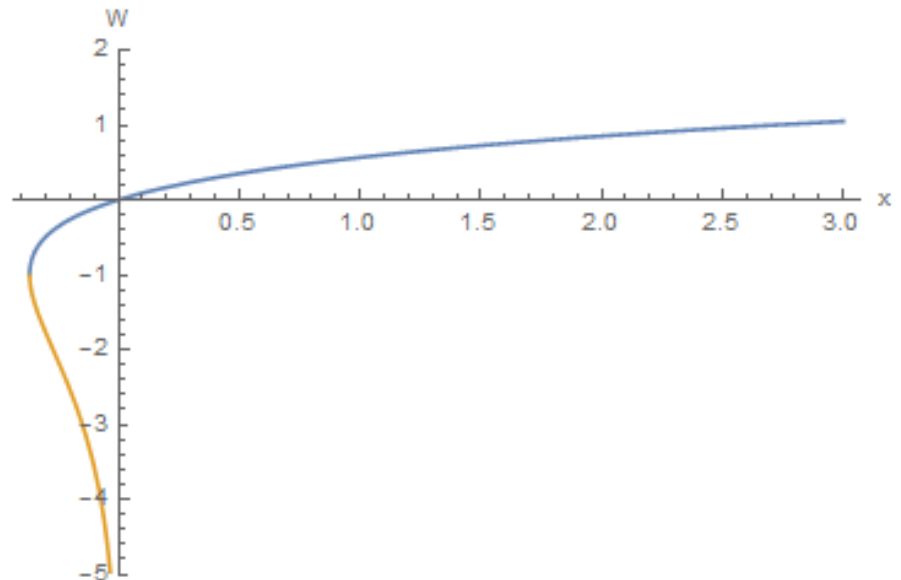
➤ Lambert W function  
defined the inverse function of  $y = x \exp(x)$

$$y = \begin{cases} W_0(x) \\ W_{-1}(x) \end{cases}$$

Each branched functions  
correspond velocities of flows.

$W_0 \Rightarrow$  subsonic flow

$W_{-1} \Rightarrow$  supersonic flow





# Function of Mach number (2)

- Mass conservation

$$\frac{\partial}{\partial x}(\rho v x^2) = 0 \rightarrow p \mathcal{M} x^2 = p_0 \mathcal{M}_0 x_0^2 \quad \text{(Subscribe 0 show quantities at start point of flows)}$$

- Equation of motion

$$v \frac{dv}{dx} = -\frac{1}{\rho} \frac{dp}{dx} - \frac{d\Phi}{dx} \rightarrow p = p_0 \exp\left(-\frac{1}{2}(\mathcal{M}^2 - \mathcal{M}_0^2) - \frac{1}{c_s^2}(\Phi(x) - \Phi(x_0))\right)$$

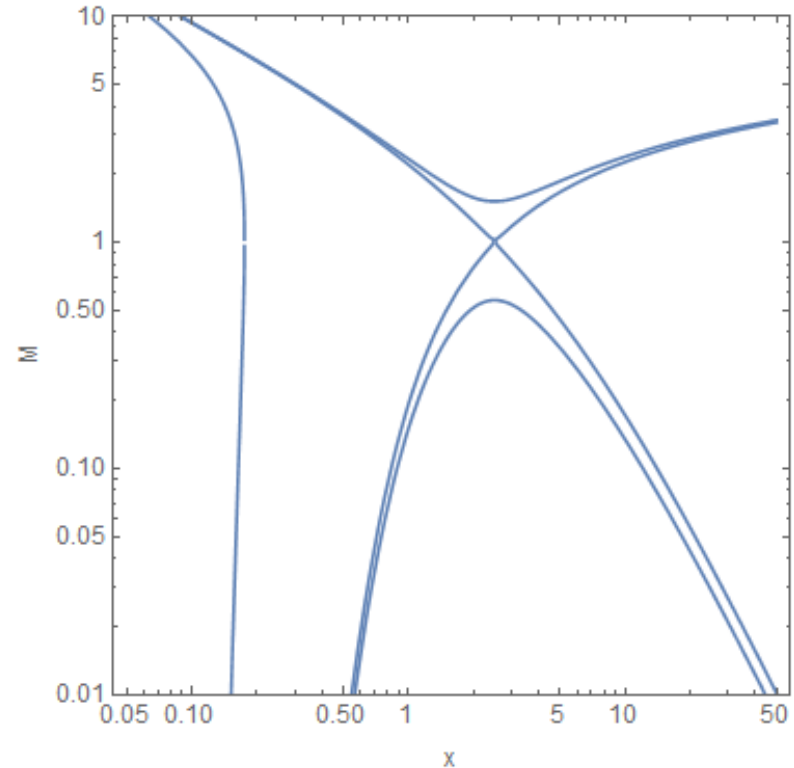
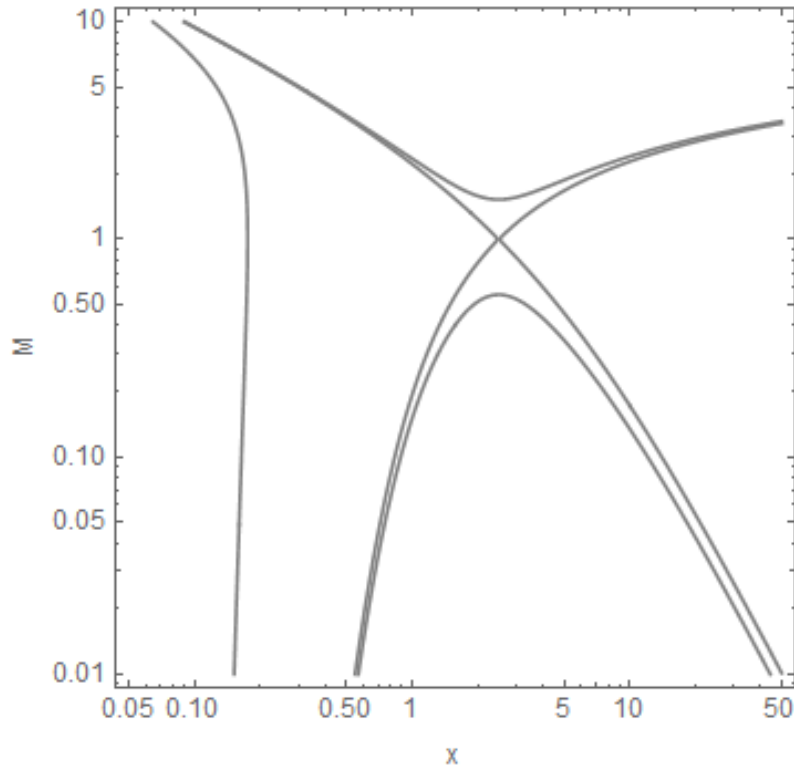
To combine above two functions, formulating Mach number function using Lambert W.

$$\mathcal{M} = \begin{cases} \sqrt{-W_0\left(-\frac{\mathcal{M}_0^2 x_0^4}{x^4} \exp\left(-\mathcal{M}_0^2 + 2(\Phi(x) - \Phi(x_0))\right)\right)} & \text{(for subsonic)} \\ \sqrt{-W_{-1}\left(-\frac{\mathcal{M}_0^2 x_0^4}{x^4} \exp\left(-\mathcal{M}_0^2 + 2(\Phi(x) - \Phi(x_0))\right)\right)} & \text{(for supersonic)} \end{cases}$$

# Function of Mach number (3)

M-x phase diagram for solar wind

(left : differential equation of Mach number, right : function using Lambert W)



**We obtain the function of Mach number for arbitrary gravitational potential.**