

# SU2 5주차 보고서

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#### 해석조건

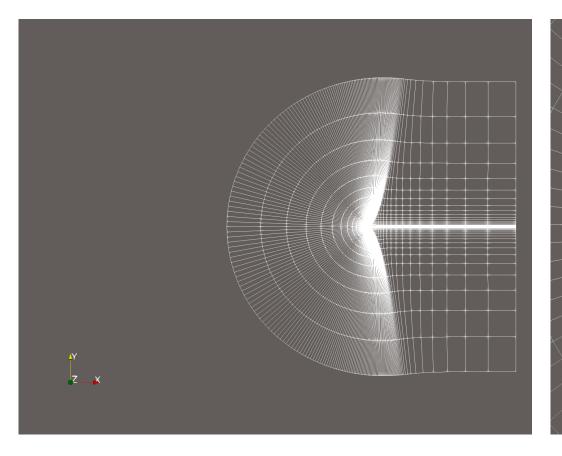


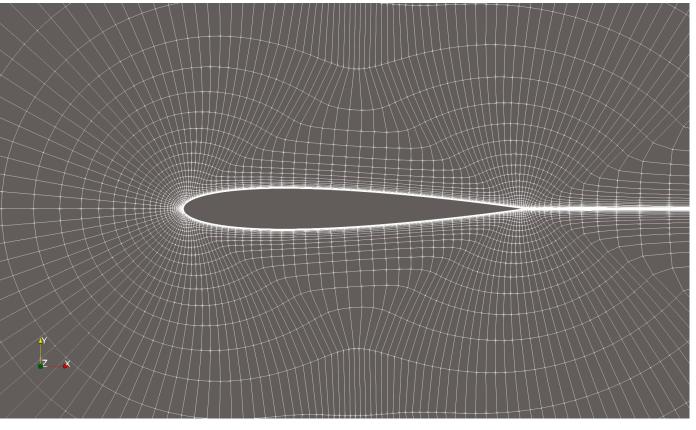
- SOLVER= RANS
- AOA= 0,2,4,6,8,10,12,14,15,16,18
- MACH\_NUMBER= 0.15
- CONV\_NUM\_METHOD\_FLOW= HLLC
- MARKER\_HEATFLUX= ( airfoil, 0.0 )
- MARKER\_FAR= (farfield)
- MARKER\_PLOTTING= ( airfoil )
- MARKER\_MONITORING= ( airfoil )

### Mesh



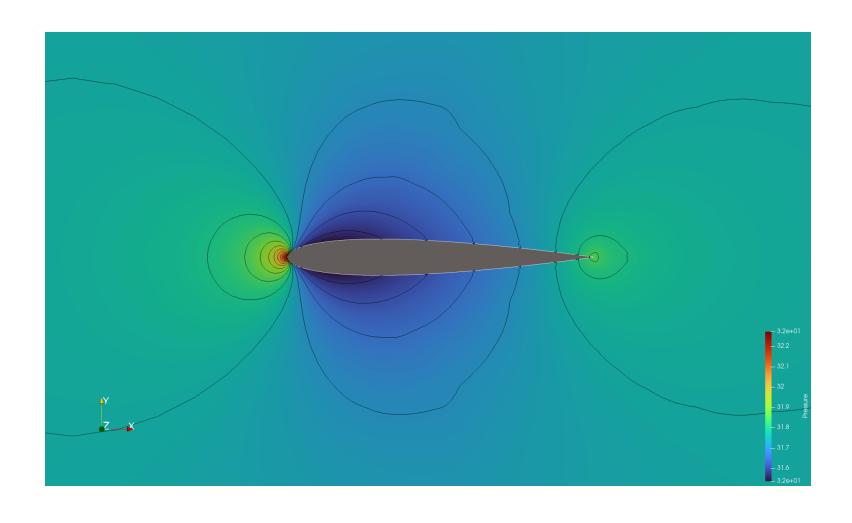
- NDIME=2
- NELEM=14336





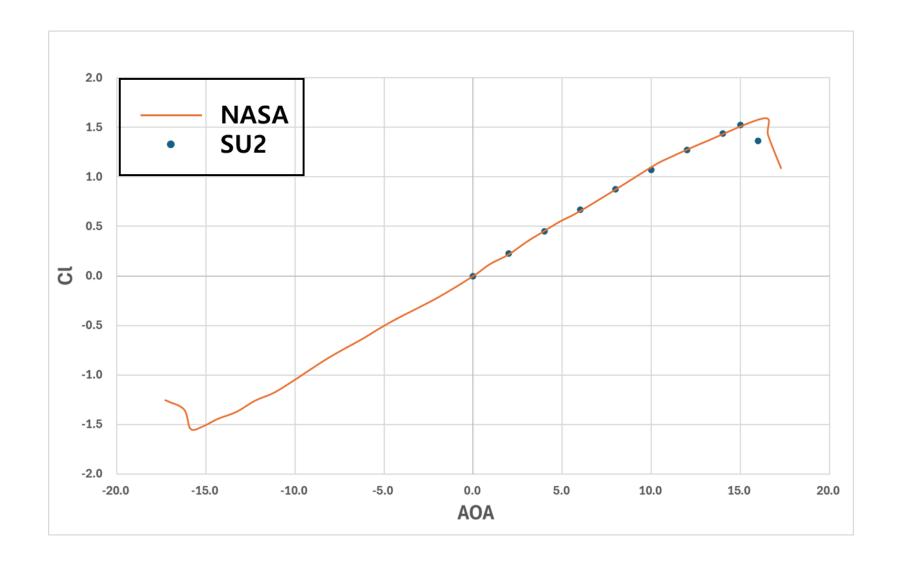
#### **Pressure Contour**





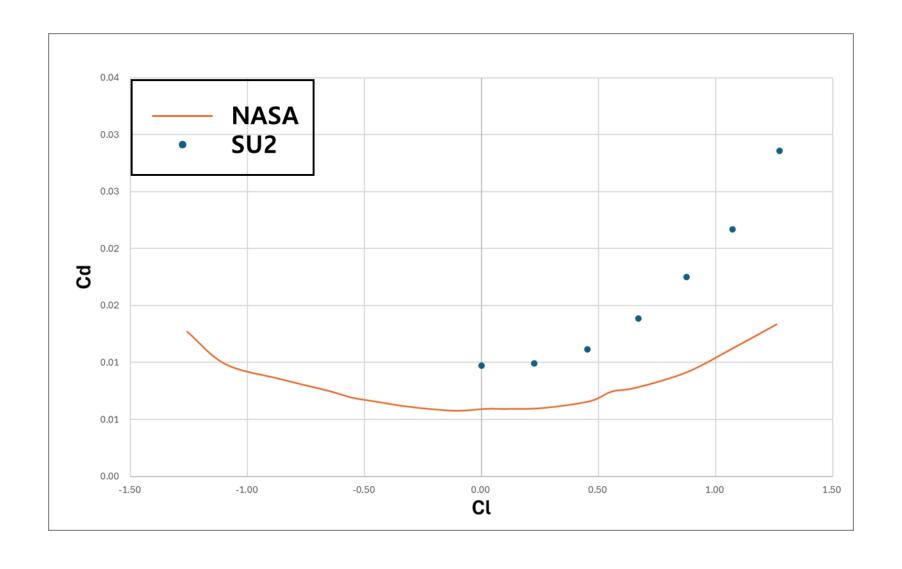
## **Comparison of Lift Coefficient**





## **Comparison of Drag Coefficient**





## **Comparison of Pressure Coefficient**



