## Development of h-p Adjoint-based error estimation for LES of reactive flows

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**Doctoral Examination Committee** Meeting I University of Toronto, Institute for Aerospace Studies

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### Outline:

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3 Methodology

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- b Overview of error
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Computing power

combustion experimenta Turbulent combustion

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### Introduction

### Cost of experiment vs numerical simulation

- Computational Fluid Dynamics (CFD) developed to reduce the time and cost of prototypes in fluid flow experiments.
- Complexities may be expensive to set up in experimental modeling
- CFD methods and models have been developed to capture this phenomenon to varying extents of accuracy

Moore's law: Computing power  $\approx$  doubles every 2 years

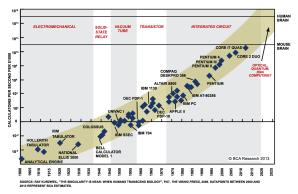


Figure 1: Moore's Law over the years [BCA Blog] [1]]



## Turbulent combustion - experimental

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#### Turbulent combustion experimental

Turbulent combustion simulation

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Turbulent combustion: real fluid flows almost always involve turbulence. Large eddy simulation (LES) - technique to achieve higher accuracy than Reynolds' averaged Navier Stokes (RANS) at lower computational cost (time, resources) than direct numerical simulation (DNS).

Lifted turbulent Ethylene ( $C_2H_4$ ) jet flame issuing into a concentric co-flow of air. Zone between flame-base and nozzle may have partial premixing. Fuel and air temperature, pressure near standard [Köhler 2006] [2]

- Dimensions: Nozzle diameter = 2.0 mm; Co-flow air annulus diameter = 140 mm
- Exit Reynolds number: 10000
- Air mass flow: 320 g/min
- Mean fuel jet velocity: 44 m/s

[Köhler 2006] [3]



### Turbulent combustion - simulation example

Computing power
Turbulent

Experimental
Turbulent
combustion simulation

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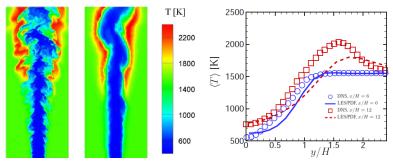
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Considering that DNS resolves all the scales, LES models sub-filter scales (SFS) while resolving the larger scales, and that RANS models all the scales, then we can expect the most accurate to be DNS, then LES, then RANS. Computational results that Yang, Pope and Chen obtained are:



(a) temperature in x-y plane: DNS (I), LES/PDF (r)

(b) Mean temperature :DNS and LES

Figure 2: DNS and LES results for a turbulent Ethylene jet flame in hot co-flow, [Yang et al, 2013 [4]]



### Turbulent combustion - simulation example cont'd

Computing

Turbulent combustion

Turbulent combustion -

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### Their numerical setup:

- DNS
  - Grid points =  $1.3 \times 10^9$ .
  - Computational cost  $=\approx 14 \times 10^6$  CPU hours.
  - Computational domain = 3D cuboid  $L_x \times L_y \times L_z = 15H \times 20H \times 3H$  in the streamwise x-, transverse y-, and spanwise z-directions, where H = 2 mm is the jet width. Boundary conditions (BCs) are inflow/outflow in x and y, while periodic in z.
- LES
  - Grid points  $\approx 8.3 \times 10^3$ .
  - Computational cost = not specified expected to be several orders of magnitude *lower*.
  - Computational domain = 3D cuboid  $L_X \times L_Y \times L_Z = 15H \times 30H \times 3H$ . (larger y to move the transverse boundary away from the central turbulent jet, which can avoid the artifact of the Dirichlet boundary condition on entrainment near the jet.)

The results they obtained for mean temperature reveal good agreement between LES and DNS at x/H=6, with lower-than-predicted values at x/H=12. They anticipate mean temperature prediction to improve with finer mesh resolution in the LES grid.



## Scope of research

Introduction

Scope

this will be a subsection this will be another subsection this will be th last subsection

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Reducing numerical error

■ High Order ... CENO

Explicit filtering

Adjoint based error estimation

Using h and p adapatation

(Chris Ngigi - DEC I) 6/25



# Methodology

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Favre Averaged Governing Equations

Large Eddy Simulation:

Explicit Filtering

Some LES errors: Aliasing, Commutation

Sub-filter scale (SFS) modeling

 High-order finite volume methods: CENO technique - benefits of higher accuracy on a coarse mesh

AMR

■ Block-based AMR: speed and parallelization

 Anisotropic vs Isotropic: how cell count (computational cost) can be reduced

Now the non-uniform vs the uniform block modification

Mesh geometry: CFFC can deal with cartesian or curvilinear coordinates - is this via using mapping functions for reference elements?



### Existing framework

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■ The CFFC code already includes the following required features:

Block-Based : people, year

AMR:

Deconick's research on explicit filters

High Order FVM with CENO:

Scott's work/input: Newton iterations and gmres solver

 Lucie's non-uniform approach - improves accuracy of flux evaluations and reduces computational cost for anisotropic

 PCM-FPI combustion modelling: modeled by F. Hernandez-Perez and N. Shahbazian

■ Initial adjoint analysis done by Martin for the advection equations

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### Overview of error

Error

Types of numerical error

- Truncation error
- Solution error Then explain a bit how they arise and how they can be dealt with

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## Adaptive mesh refinement (AMR)

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AMR

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- Benefits of AMR
- How the block-based technique works
- Ghost cells for intercommunication
- Current stencils
- how the high-order will affect the current stencil
- use radial sphere diags

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### High order finite volume method

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High order FVM CENO

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An explanation on h.o FVM. how the high-order works, and how it reduces numerical error. separate slide of other groups researching this: Ihme and Poinssot. show some of their results

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### **CENO**

CENO

- Lucien's work
- Ramy's work
- Marc Charest's work
- Luiz's work



### A background on gradient/physics-based refinement

Describe gradient based techniques

Gradient technique

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### Adjoint-based error estimation

Introduction

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Adioint

Gradient techniqu

#### About the adjoint

Error estimation indicators
Steady vs unsteady

Mesh adaptatio

Refinement

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explaining what the adjoint is. who was the first to use adjoint

 Cite initial work for this: Giles and Pierce, venditti and darmofal, fidkowski, jameson

continuous and discrete adjoint formulations

- continuous adjoint formulation
- discrete adjoint formulation: methods to evaluate the discrete adjoint
  - one
  - two
  - three



### Adjoint-based error estimation cont'd

#### About the adioint

description of the adjoint methods to evaluate psi

- first
- second
- third

Techniques to evaluate dR/dU

- complex step
- finite differencing
- automated differentiation
- approximate method



### Error estimation indicators

#### Error estimation indicators

All about residual weighting (flag for refinement) and a 1D cartoon example, perhaps, of restriction/prolongation

- projecting onto fine space
- restricting onto coarse space
- getting the error in the residual and using this as a flag for refinement

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## Steady vs unsteady

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About th

Error estimation

Steady vs unsteady

Benefits

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Lastly: treatment of steady vs unsteady adjoints

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### Benefits of the adjoint approach

Benefits

Expected benefits of adjoint vs gradient based methods

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## Mesh adaptation based on adjoint

Mesh adaptation

This is a separate slide on mesh adaptation as based on the adjoint. Enough diagrams from venditti and darmofal, fidkowski

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## Basis of refinement: h and p

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Adjoint

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Show or put some figures with citations. Show what other groups have done. WHO has researched or is using adjoint with AMR?

- Fidkowski and Darmofal [2011] Review of Output-Based Error Estimation and Mesh Adaptation in Computational Fluid Dynamics
- Hartmann, ERROR ESTIMATION AND ADJOINT-BASED ADAPTATION IN AERODYNAMICS, [2006]
- Nemec and Aftosmis [2007] Adjoint Error Estimation and Adaptive Refinement for Embedded-Boundary Cartesian Meshes
- Hartmann, Held and Leicht [2010] Adjoint-based error estimation and adaptive mesh refinement for the RANS and k- turbulence model equations
- Woopen, May and Schütz [2013] Adjoint-Based Error Estimation and Mesh Adaptation for Hybridized Discontinuous Galerkin Methods
- Li, Allaneau and Jameson [2011] Continuous Adjoint Approach for Adaptive Mesh Refinement
- Diskin and Yamaleev [2011] Grid Adaptation Using Adjoint-Based Error Minimization

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### How we can use this

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Framework

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Refinement

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- using it for mesh refinement how some previous groups used this
- how we can link mesh adaptation AMR to the adjoint via h
- how we can use p based refinement

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## Progress to date

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CFFC code familiarization: LES test case - on parallel clusters - SciNET.
 Job scheduling and post-processing results (tecplot)

 creating and solving linear systems in parallel implementation - trilinos and MPI

- 2D Poisson problem
- 3D Poisson problem
- Preliminary work with the discrete adjoint shockcube problem
  - give the initial states, I and r
  - how the code was modified multiblock and multiproc for uniform blocks
  - some results
  - work in progress
    - boundary conditions
    - compare with other techniques to get dR/dU



### Timeline: April 2015 - January 2016

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AMR

FVM

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Present

Put a table of what you have done till now

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## Projected milestones

Future

■ Put a table of what you will do in the next steps

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Progress

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Future

# Thank You For Your Attention!

Questions?

(Chris Ngigi - DEC I) 25/25



### References I

- BCA Research Blog, http://blog.bcaresearch.com/wp-content/uploads/2013/10/ Chart-III-8-Moores-Law-Over-199-Years-And-Going-Strong.png, Accessed 29-03-2015
- [2] Köhler, M., Boxx, I., Geigle, K. P., and Meier, W., Simultaneous planar measurements of soot structure and velocity fields in a turbulent lifted jet flame at 3-kHz", Applied Physics B 103 (2), 271-279, 2011
- Adelaide international sooting flame (ISF) workshop, http://www.adelaide.edu.au/cet/isfworkshop/data-sets/turbulent/, Accessed 29-03-2015
- [4] Yang, Y., Pope, S. B., Chen, J. H., "An LES/PDF study of a turbulent lifted ethylene jet flame in a heated coflow", 8th US National Combustion Meeting, 2013

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# Backup Slide

■ Important backup slide point.

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