

CONTROL SYSTEMS - UNIT-1

- ① System → means of transfer / changing a signal. It is the func. of one or more no. of independent variables.
- ② automatic control → the existing value of a quantity / condⁿ systems is measured & compared with the desired value & the diff of these 2 values is used to initiate the action for reducing the diff.

CLASSIFICATION.

① Linear and Non-linear:

Linear → both homogeneity & superposition

Superposition: $r_1(t) \rightarrow c_1(t)$ & $r_2(t) \rightarrow c_2(t)$
then $[r_1(t) + r_2(t)] \rightarrow [c_1(t) + c_2(t)]$

homogeneity: $k r_1(t) \rightarrow k c_1(t)$

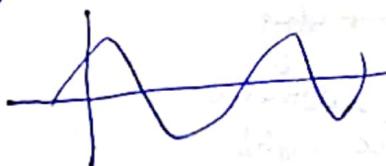
Non-linear: non homogeneous.

② Time Invariant & Time varying system:

Parameters are
independent of time
& not varying with
time.

vary with time

③ Continuous Time and Discrete Time System



→ defined for every instant of time.
→ func. of continuous time variable



defined at only specific instants of time.

(4) Dynamic and Static System:

If o/p of sys is a func of time even when i/p is constant

if i/p does not change with time then o/p will also not change with time

(5) Deterministic & stochastic systems

o/p could be exactly determined for a given i/p

For a given sys i/p, the output is uncertain, only probability can be predicted.

OPEN LOOP CONTROL SYSTEM:

- control system without feedback or non-feedback control system
- control action is independent of the desired output



The components are :

1. controller
2. controlled Process

Advantages

- simple, economical, less maintenance, not difficult

Disadvantages:

- inaccurate
- not reliable
- slow
- optimization not possible

Eg : Automatic washing machine, automatic traffic light.

CLOSED LOOP CONTROL SYSTEM:

- feedback control systems
- the control action is dependent on the desired output.
- Here, the output is compared with the reference input. An error signal is produced.

Advantages:

- more reliable
- faster
- optimization is possible

Disadvantages

- expensive
- difficult maintenance
- complicated installation.

Eg: Air conditioner → temp regulation

DIFFERENCE

Open loop

- not reliable
- easier to build
- slow
- more stable
- optimization not possible
- not that accurate

Closed loop

- reliable
- difficult to build
- fast
- less stable
- optimization is possible.
- accurate

TRANSFER FUNCTION

- Defined as the ratio of Laplace transform of the output to the Laplace transform of input with all initial conditions as zero.



$$G(s) = \frac{C(s)}{R(s)} = \frac{L(C(t))}{L(R(t))} \left\{ \begin{array}{l} \text{Output} \\ \text{Input} \end{array} \right\}$$

ADVANTAGES OF TRANSFER FUNC:

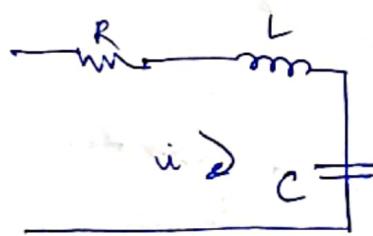
- Transfer func is a mathematical model of all system components & hence of overall system & therefore it gives the gain of the system.
- helps in study of stability of system
- does not depend on input to system.

DISADVANTAGES OF TRANSFER FUNC:

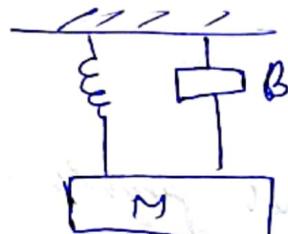
- $G(s)$ cannot be defined for non-linear system.
- initial conditions lose their importance.
- Physical structure of system cannot be determined.

ANALOGOUS SYSTEMS

- Two diff. physical systems can be described by same mathematical model. The concept is very useful in study of complex systems.



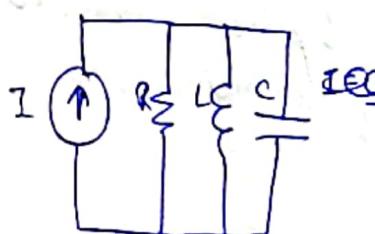
$$V_i = \frac{Rdq}{dt} + L \frac{d^2q}{dt^2} + \frac{q}{C}$$



$$F = \frac{Mdx}{dt^2} + B \frac{dx}{dt} + Kx$$

F-V Analogy

F	V
B	R
M	L
$1/K$	C



$$I = \frac{1}{R} \frac{dp}{dt} + \frac{\phi}{L} + \frac{c}{C} \frac{d^2\phi}{dt^2}$$

F	I
M	C
B	$1/R$
K	$1/L$

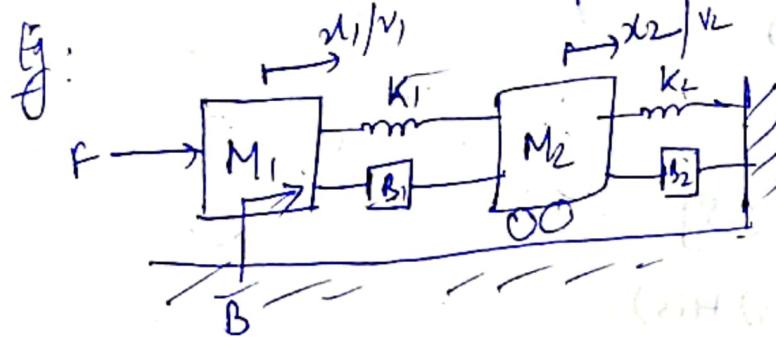
For F-V

- (*) elements having same velocity = are in series
- (*) each node (mass) = closed loop in elec.
- (*) no of meshes in elec = no. of masses in mech.
- (*) elec connected b/w 2 masses in mech is represented as a common elec b/w two meshes in elec.

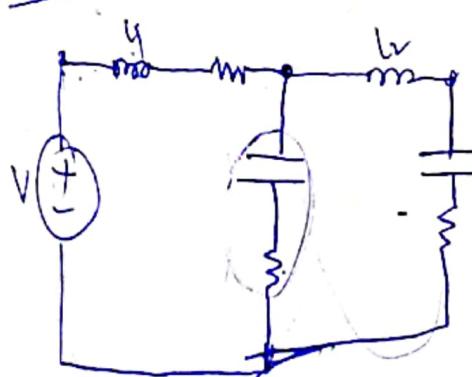
For F-I :

- (*) In mech, the elements in parallel will have same force similarly, in elec, elec in parallel have same voltage.
- (*) each node (mass) in mech corresponds to a node in elec.
- (*) no. of nodes in elec = no. of nodes in elec mech.
- (*) elec connected b/w 2 nodes in mech is represented as common elec b/w nodes in electrical system. (in parallel)

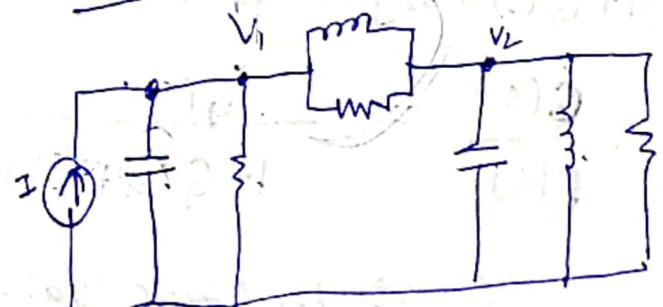
|| Elements taken in parallel.



F-V



F-I



BLOCK DIAGRAM

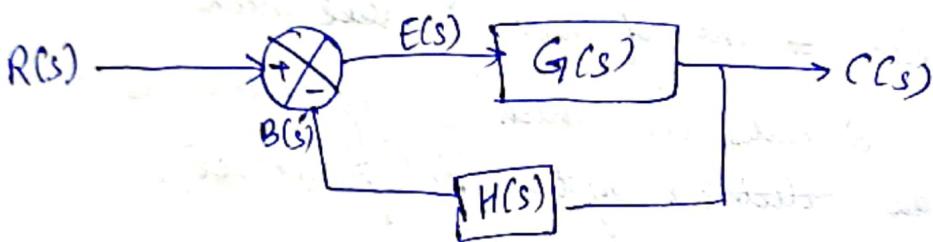
single line diagram that represents flow of system variables from one block to another.
Each block represents a single component or a group of components which is characterized as transfer fun.

Q) How to draw?

1. Write diff eqn

2. Take Laplace

3. Find expression for $I(s)$



$$E(s) = R(s) - B(s)$$

$$= R(s) - C(s), H(s)$$

$$C(s) = G(s) \cdot E(s)$$

$$C(s) = G(s) [R(s) - C(s)H(s)]$$

$$C(s) = G(s)R(s) - C(s)G(s)H(s)$$

$$C(s)[1 + G(s)H(s)] = G(s)R(s)$$

$$\frac{G(s)}{R(s)} = \frac{G(s)}{1 + G(s)H(s)}$$

(-ve feedback gain.)

$$\text{+ve feedback gain} = \frac{C(s)}{R(s)} = \frac{G(s)}{1 - G(s)H(s)}$$

BLOCK DIAGRAM REDUCTION RULES:

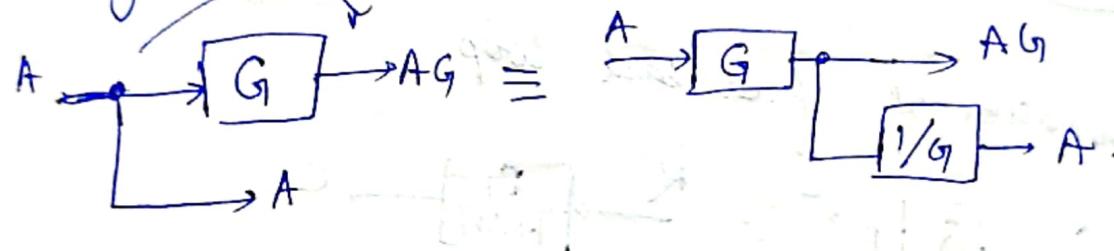
(1) combining 2 blocks in cascade/series:



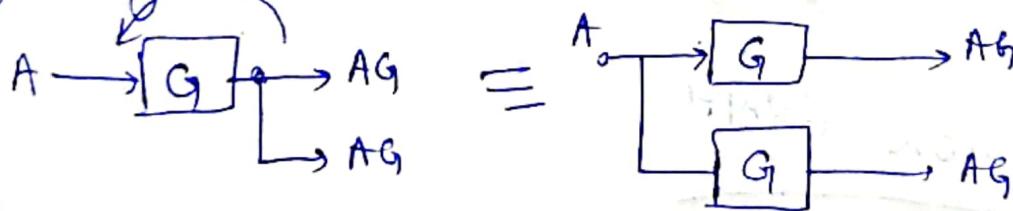
(2) combining blocks in parallel



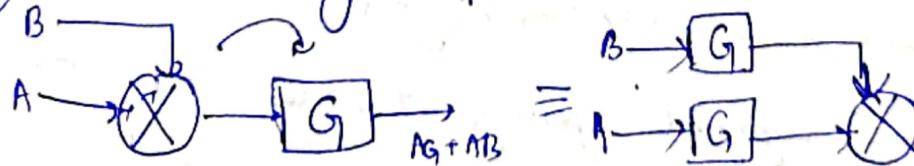
(3) Moving take off pt ahead of block



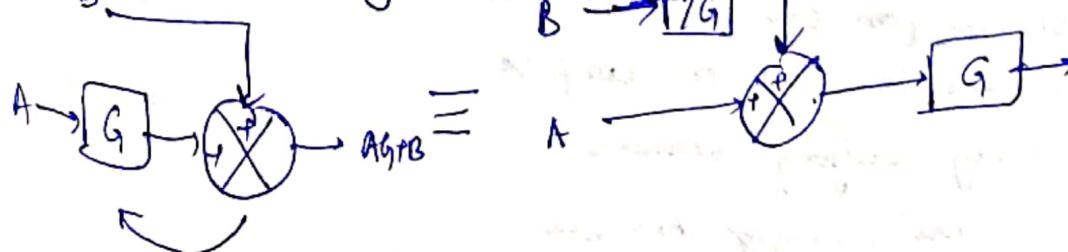
(4) Moving take-off point before block



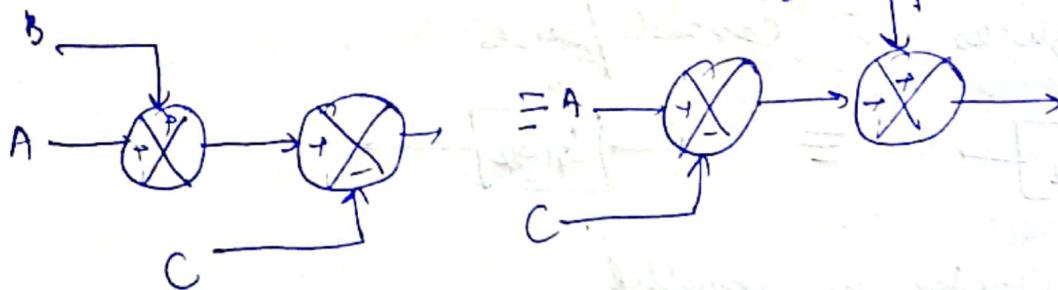
(5) Moving summing point ahead of block



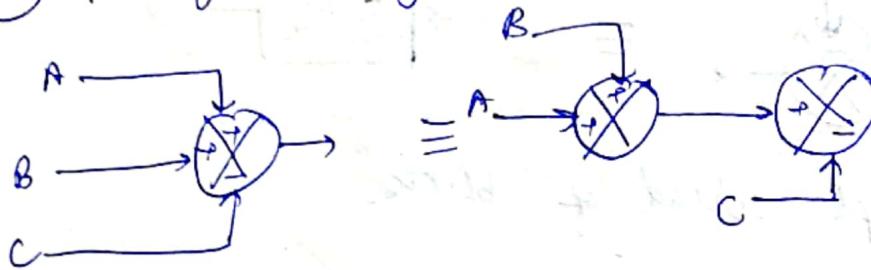
(6) Moving summing point before block



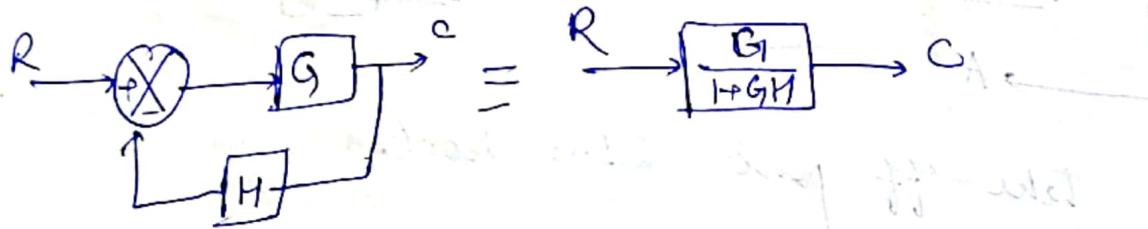
⑦ Interchanging summing point



⑧ Splitting summing points



⑨ Elimination of feedback loop



SIGNAL FLOW: GRAPH
graphical representation

① Node → point representing a variable / signal

② Branch - directed line segment joining 2 nodes. The arrow on branch indicates signal flow.

③ Transmittance → The gain acquired by the signal when it travels from one node to another.
→ It can be real or complex.

④ input node → only outgoing branches
(source)

⑤ output node → only incoming branches
(sink)

① Path: traversal of connected branches in the direction of arrows. The path should not cross a node more than once.

more than one

(i) open path (ii), closed path (starts & ends at same node)

④ Forward path → the path from i/p to o/p

- ① forward path gain → combination of transmittances in forward path.
- ② forward path gain → combination of transmittances in forward path.

- ① Individual loop \rightarrow start & end at same node without visiting a node more than once.

④ Loop gain \rightarrow product of gains of a loop

① Non-touching loop → if the loops don't have common node.

MASON'S GAIN FORMULA:

$$T = \frac{C}{R} = \frac{\sum p_i \Delta i}{\Delta}$$

where P_i = gain of i^{th} path

$$\Delta = 1 - (\text{sum of individual loop gains}) + (\text{sum of gain products of } d \text{ non-touching loops})$$

$$D_i = 1 - \left(\frac{\text{total gain}}{\text{loop gain}} \right) \cdot \text{which are not touching to } i^{\text{th}} \text{ part of path} \quad (1 - \text{that part of } S \text{ not common with } i^{\text{th}} \text{ part})$$

Step 1: Write all forward paths gain

Step 2: Write all loop gains. (start & end node of loop same)

Step 3: find non touching loops

PROPERTIES OF SIGNAL FLOW GRAPH

- applicable to linear time-invariant system
 - signal flow only along direction of arrows
 - value of variable at each node is equal to algebraic sum of all signals entering at that node.
 - signal flow graph is not unique property of system

CONSTRUCTION OF SPG

$$E: y_2 = t_{21} y_1 + t_{23} y_3$$

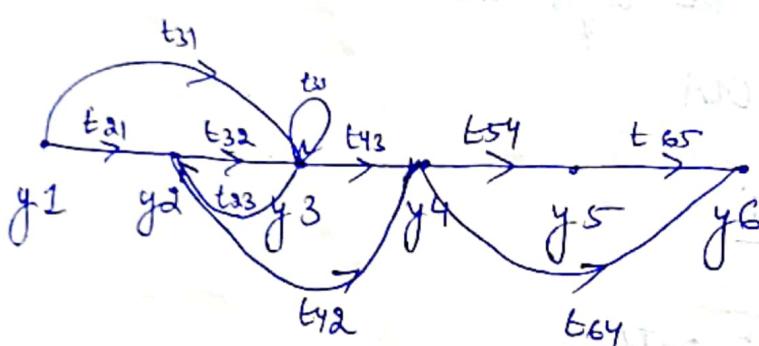
$$y_3 = t_{32} y_2 + t_{33} y_3 + t_{31} y_1$$

$$y_4 = t_{43} y_3 + t_{42} y_2$$

$$y_5 = t_{54} y_4$$

$$y_6 = t_{65} y_5 + t_{64} y_4$$

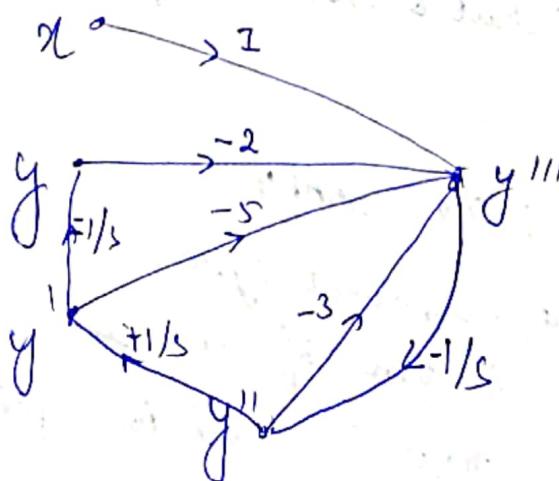
From
eqns



* From Differential eqn.

$$y''' + 3y'' + 5y' + 2y = x$$

$$y''' = x - 3y'' - 5y' - 2y$$



Reverse sign of

- ① { y''' to y''
- y'' to y'

- ② { y'' to y'
- y' to y

DIFFERENCE BTW BLOCK DIAGRAM & SFG

(*) Block diag

- ① applicable to linear time invariant system only
- ② each element represented by block
- ③ summing pt & take off pts are separate.
- ④ self loop does not exist
- ⑤ time consuming method
- ⑥ block diag is reqd at each and every step.
- ⑦ transfer func of element is shown inside the corresponding block.
- ⑧ feedback path is present.

(*) Signal flow graph:

- ① applicable to LTI systems
- ② each variable is represented by a node
- ③ summing & take off points are absent
- ④ self loop can exist
- ⑤ require less time by using Mason gain formula.
- ⑥ At each step, it is not necessary to draw SFG
- ⑦ transfer func is shown along the branches connecting the nodes
- ⑧ feedback loops are used.



CONSTRUCTION OF SFG from BLOCK DIAG:

Rules: (C & R) 

- ① All variables, summing pts & take off points are represented with nodes.
- ② If summing pt is placed before a take off point in the direction of signal flow, then represent summing pt & take off pt by a single node.
- ③ If summing pt is placed after a take off point in the direction of signal flow, then represent summing pt & take off pt by separate nodes connected by a branch having transmittance unity.

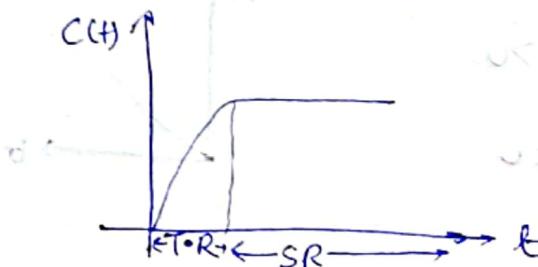
ST \rightarrow single pt.

CONTROL SYSTEMS - UNIT-2

TIME DOMAIN ANALYSIS

→ variation of o/p of system with time

- ① Transient Response → that part of response varying with time.
- ② Steady State Response → that part of total response after transient has died.



TEST SIGNALS

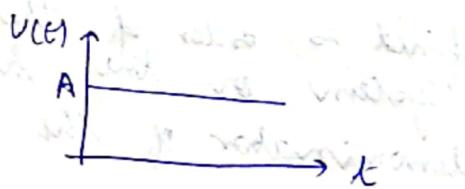
1) Step signal (displacement function)

$$V(t) = \begin{cases} A & t \geq 0 \\ 0 & t < 0 \end{cases}$$

When $A=1 \rightarrow$ unit step

$$L[V(t)] = \frac{1}{s} \quad (\text{unit step signal})$$

$$R(s) = \frac{1}{s}$$



2) Ramp function

$$\delta(t) = \begin{cases} kt & t \geq 0 \\ 0 & t < 0 \end{cases}$$

where $k = \text{slope for line}$

for $k > 0 \rightarrow$ positive slope, $k < 0 \rightarrow$ negative slope, downwards.

$$L[\delta(t)] = \frac{k}{s^2}$$

$$\text{for unit ramp} = \frac{1}{s^2}$$



3) Impulse signal:

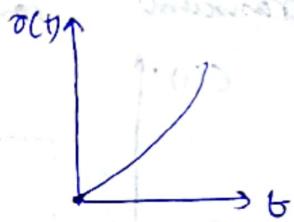
$$\delta(t) = \begin{cases} 1 & t=0 \\ 0 & t \neq 0 \end{cases}$$

$$L[\delta(t)] = \infty$$

$$L[\delta(t)] = 1$$

4) Parabolic signal

$$x(t) = \begin{cases} \frac{Kt^2}{2}, & t \geq 0 \\ 0, & t < 0 \end{cases}$$



$$L[x(t)] = \frac{K}{s^3}$$

ORDER OF THE SYSTEM:

→ defined as order of the differential eqⁿ representing the system or the highest power of s^n in the denominator of the transfer function.

RESPONSE OF FIRST ORDER SYSTEM WITH UNIT STEP I/P.

$$\frac{C(s)}{R(s)} = \frac{1}{s\tau + 1} \Rightarrow C(s) = \frac{1}{s\tau + 1} R(s)$$

$$\Rightarrow C(s) = \frac{1}{s} \left(\frac{1}{s\tau + 1} \right) = \frac{1}{s} - \frac{1}{s\tau + 1}$$

⇒ inverse laplace gives:

$$C(t) = 1 - e^{-t/\tau}$$

When $t = \tau$,

$$C(t) = 1 - e^{-1} = 0.632 = 63.2\%$$

τ = time constant

= time reqd for signal to attain 63.2% of final or steady state value.

= how fast system reaches final value.

④ Smaller the τ , faster is the system response

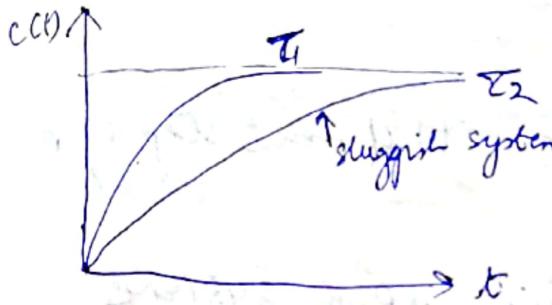
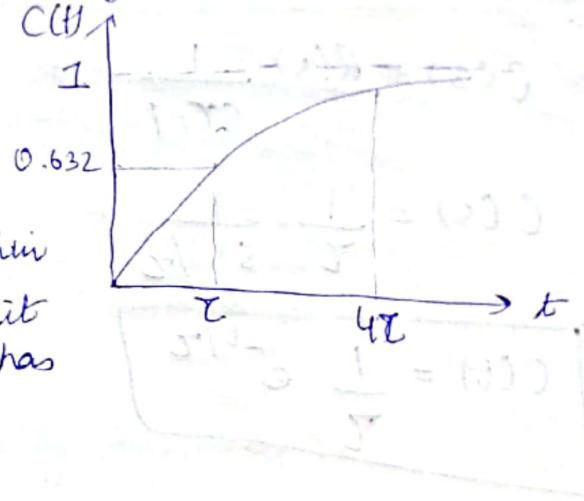
$$\tau = 63.2\%$$

$$2\tau = 86.4\%$$

$$4\tau = 98\%$$

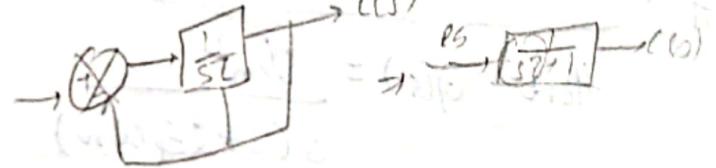
Settling time

(When actual op reaches within 2% of the desired output, it is said that steady state has reached)



$$\tau_1 < \tau_2$$

$$\begin{aligned} \text{Error } e(t) &= r(t) - c(t) \\ &= 1 - (1 - e^{-t/\tau}) \\ &= e^{-t/\tau} \end{aligned}$$



RESPONSE OF FIRST ORDER SYSTEM WITH UNIT RAMP

$$R(s) = \frac{1}{s^2}$$

$$\begin{aligned} C(s) &= \frac{1}{s^2} \left(\frac{1}{s\tau + 1} \right) \Rightarrow C(s) = \frac{A}{\tau s + 1} + \frac{B s + C}{s^2} \\ &= \frac{\tau^2}{\tau s + 1} + \frac{1 - \tau s}{s^2} \end{aligned}$$

$$\Rightarrow C(s) = \frac{1}{s^2} - \frac{\tau}{s} + \frac{\tau}{s + 1/\tau}$$

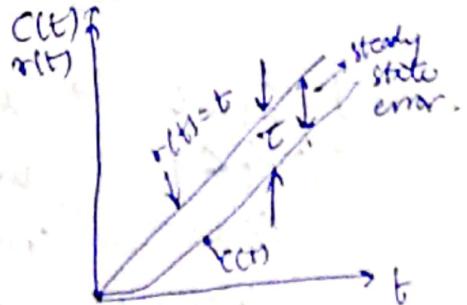
$$\Rightarrow C(t) = t - \tau(1) + \tau e^{-t/\tau}$$

For signal, $e(t) = r(t) - c(t)$

$$= t - [t - \tau + \tau e^{-t/\tau}]$$

$$= \tau(1 - e^{-t/\tau})$$

④ Less τ , more speed.

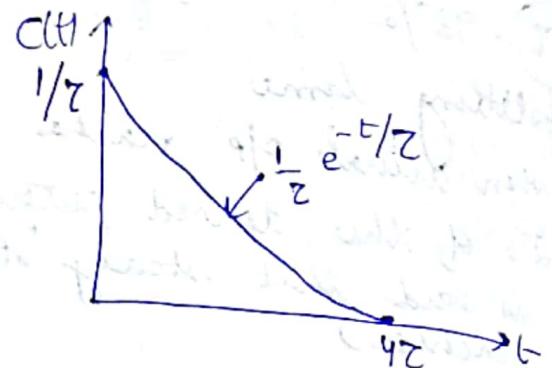


TIME RESPONSE OF UNIT IMPULSE FUNCTION (FIRST ORDER)

$$C(s) = R(s) \frac{1}{sT+1} = 1 \cdot \frac{1}{sT+1}$$

$$C(s) = \frac{1}{T} \cdot \frac{1}{s+1/T}$$

$$C(t) = \frac{1}{T} e^{-t/T}$$



For LTI systems,

$$\frac{d}{dt} C(t) \text{ of unit ramp} = C(t) \text{ of unit step } \downarrow P.$$

TIME RESPONSE OF SECOND ORDER SYSTEMS

$$\text{Here } G(s) = \frac{\omega_n^2}{s(s+2\zeta\omega_n)}$$

So, generalized transfer func we get is:

$$\frac{C(s)}{R(s)} = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

where $\omega_n \rightarrow$ undamped natural freq
 $\zeta \rightarrow$ damping factor

$\omega_d \rightarrow$ freq of damped oscillations

$$\omega_d = \omega_n \sqrt{1 - \zeta^2}$$

- | | |
|--|---|
| if $\zeta = 0 \rightarrow$ undamped system | } |
| $\zeta < 1 \rightarrow$ under damped | |
| $\zeta = 1 \rightarrow$ critically damped | |
| $\zeta > 1 \rightarrow$ overdamped system | } |

Characteristic eqn:

$$s^2 + 2\zeta\omega_n s + \omega_n^2 = 0$$

$$s_1, s_2 = \frac{-2\zeta\omega_n \pm \sqrt{(2\zeta\omega_n)^2 - 4\omega_n^2}}{2}$$

$$= -\zeta\omega_n \pm \omega_n\sqrt{\zeta^2 - 1}$$

When $\zeta = 0$,

$$s_1, s_2 = \pm j\omega_n \text{ (purely imaginary)}$$

When $\zeta = 1$,

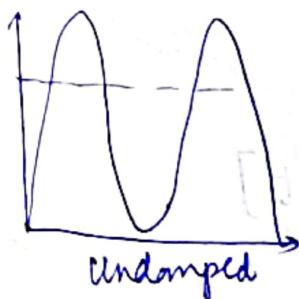
$$s_1, s_2 = -\omega_n \text{ (purely real)}$$

When $0 < \zeta < 1$,

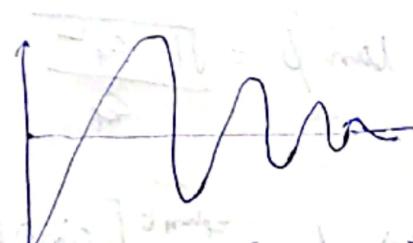
$$\begin{aligned} s_1, s_2 &= -\zeta\omega_n \pm j\omega_n\sqrt{1-\zeta^2} \quad (\text{complex conjugates}) \\ &= -\zeta\omega_n \pm j\omega_d \end{aligned}$$

When $\zeta > 1$,

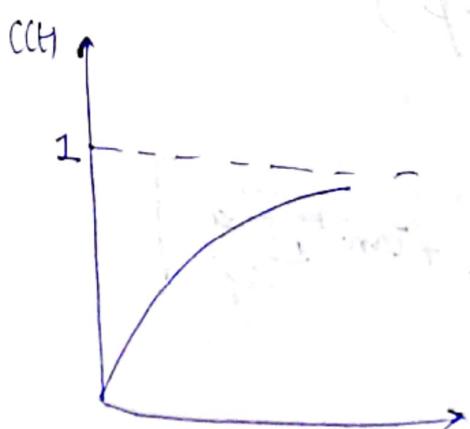
$$s_1, s_2 = -\zeta\omega_n \pm \omega_n\sqrt{\zeta^2 - 1}$$



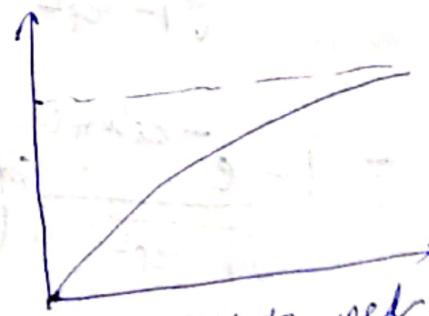
undamped



under-damped



critically damped



over-damped

RESPONSE OF 2ND ORDER SYSTEM WITH UNIT STEP I/P

$$\frac{C(s)}{R(s)} = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

$$C(s) = \frac{1}{s} \cdot \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

On simplification,

$$C(s) = \frac{1}{s} - \frac{s + \zeta\omega_n + \zeta\omega_n}{(s + \zeta\omega_n)^2 + \omega_d^2}$$

$$\Rightarrow C(t) = 1 - \left[e^{-\zeta\omega_n t} \cos \omega_d t + \frac{\zeta\omega_n}{\omega_d} e^{-\zeta\omega_n t} \sin \omega_d t \right]$$

$$= 1 - \frac{e^{-\zeta\omega_n t}}{\sqrt{1-\zeta^2}} \left[\sqrt{1-\zeta^2} \cos \omega_d t + \zeta \sin \omega_d t \right]$$

$$\begin{aligned} \text{Put } \sqrt{1-\zeta^2} &= \sin \phi \\ \therefore \cos \phi &= \zeta \\ \tan \phi &= \frac{\sqrt{1-\zeta^2}}{\zeta} \end{aligned}$$

$$\begin{aligned} \therefore C(t) &= 1 - \frac{e^{-\zeta\omega_n t}}{\sqrt{1-\zeta^2}} \left[\sin \phi \cos \omega_d t + \cos \phi \sin \omega_d t \right] \\ &= 1 - \frac{e^{-\zeta\omega_n t}}{\sqrt{1-\zeta^2}} \sin (\omega_d t + \phi) \end{aligned}$$

$$= 1 - \frac{e^{-\zeta\omega_n t}}{\sqrt{1-\zeta^2}} \sin \left[(\omega_n \sqrt{1-\zeta^2})t + \tan^{-1} \frac{\sqrt{1-\zeta^2}}{\zeta} \right]$$

case 1: for undamped case, $\epsilon_f = 0$

$$C(t) = 1 - \sin(\omega_n t + \frac{\pi}{2})$$

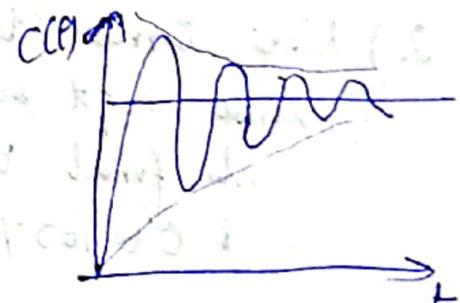
$$\boxed{C(t) = 1 - \cos \omega_n t}$$

Case 2: for underdamped system, $0 < \epsilon_f < 1$

$$C(t) = \frac{1 - e^{-\epsilon_f \omega_n t}}{\sqrt{1 - \epsilon_f^2}} \sin(\omega_d t + \phi)$$

where $\omega_d = \omega_n \sqrt{1 - \epsilon_f^2}$

$$\phi = \tan^{-1} \left(\frac{\sqrt{1 - \epsilon_f^2}}{\epsilon_f} \right)$$

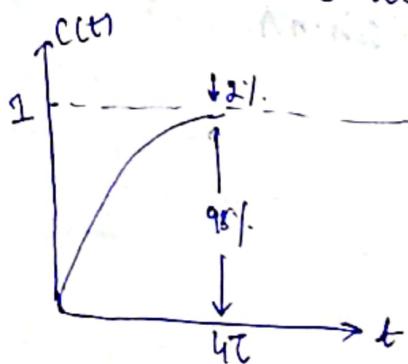


Case 3: Critically damped system, $\epsilon_f = 1$

$$C(t) = 1 - e^{-\omega_n t} (1 + \omega_n t)$$

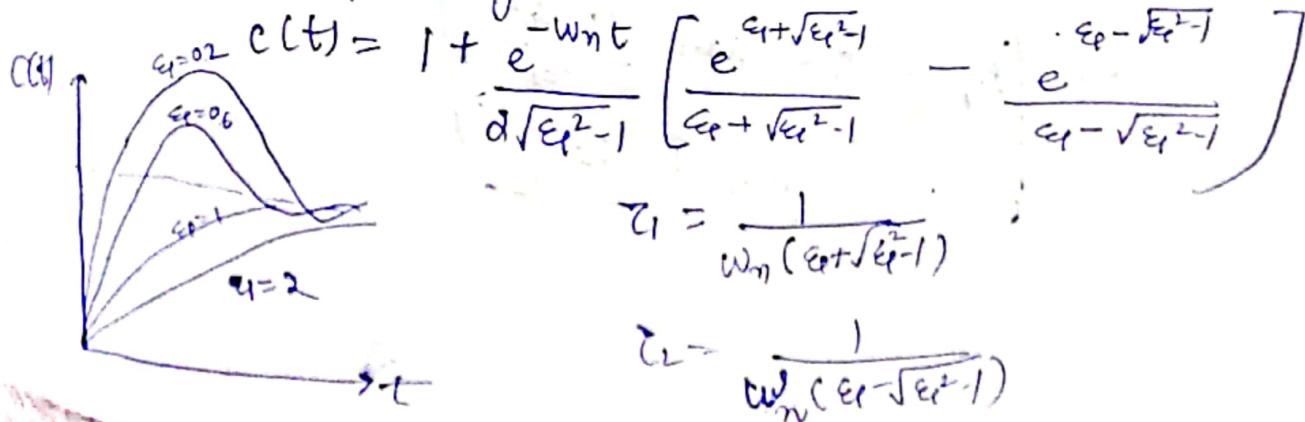
In this oscillations disappear.

Damping ratio, $\epsilon_f = \frac{\text{Actual damping}}{\text{Critical damping}} = \frac{\epsilon_f \omega_n}{\omega_n}$



time constant = $\frac{1}{\epsilon_f \omega_n}$

Case 4: Overdamped system, $\epsilon_f > 1$



TIME-DOMAIN SPECIFICATIONS

→ desired performance characteristics of control systems are specified in terms of time domain specifications.

1) Delay time (t_d)

→ time taken by response to reach 50% of final value in first time.

2) Rise time (t_r):

time reqd by response to rise from 10% to 90% of its final value for overdamped system.
0 to 100% for underdamped systems.

$$CCS = 1 - \frac{e^{-\zeta \omega_n t}}{\sqrt{1-\zeta^2}} \sin[(\omega_n \sqrt{1-\zeta^2})t + \phi]$$

$$\text{At } t = t_r, CCS = 1$$

$$\Rightarrow \frac{e^{-\zeta \omega_n t_r}}{\sqrt{1-\zeta^2}} \sin[(\omega_n \sqrt{1-\zeta^2})t_r + \phi] = 0$$

$$\Rightarrow \sin[(\omega_n \sqrt{1-\zeta^2})t_r + \phi] = 0 = \sin n\pi$$

$$\text{Put } n=1$$

$$(\omega_n \sqrt{1-\zeta^2})t_r + \phi = \pi$$

$$\Rightarrow t_r = \frac{\pi - \phi}{\omega_n \sqrt{1-\zeta^2}}$$

$$\Rightarrow \boxed{t_r = \frac{\pi - \tan^{-1} \frac{\sqrt{1-\zeta^2}}{\zeta}}{\omega_n \sqrt{1-\zeta^2}}}$$

3) Peak time (t_p): time reqd for the response to reach the first peak of the time response or first peak overshoot.

$$C(t) = 1 - e^{-\zeta \omega_n t} \sin \left[\omega_n \sqrt{1-\zeta^2} t_p + \tan^{-1} \frac{\sqrt{1-\zeta^2}}{\zeta} \right]$$

for Max, $\frac{dC(t_p)}{dt} = 0$

$$\Rightarrow t_p = \frac{n\pi}{\omega_n \sqrt{1-\zeta^2}}$$

For first peak put $n=1$

$$t_p = \frac{\pi}{\omega_n \sqrt{1-\zeta^2}}$$

The first minimum undershoot occurs at $n=2$

$$t_{min} = \frac{2\pi}{\omega_n \sqrt{1-\zeta^2}}$$

4) Maximum Overshoot / Peak overshoot (M_p)

→ normalized difference b/w the peak of time response & steady output.

$$\text{Maximum \% overshoot} = \frac{C(t_p) - C(\infty)}{C(\infty)} \times 100$$

Put $t=t_p = \frac{\pi}{\omega_n \sqrt{1-\zeta^2}}$ in $C(t)$, you will get:

$$C(t) = 1 - e^{-\frac{\zeta \pi}{\sqrt{1-\zeta^2}}} \sin (\pi + \phi)$$

$$= 1 + \frac{e^{-\frac{\zeta \pi}{\sqrt{1-\zeta^2}}}}{\sqrt{1-\zeta^2}} \sin \phi$$

$$C(t)_{max} = 1 + \frac{e^{-\frac{\zeta \pi}{\sqrt{1-\zeta^2}}}}{\sqrt{1-\zeta^2}}$$

(by putting $\sin \phi = \sqrt{1-\zeta^2}$)

$$M_p = C(t)_{\max} - 1$$

$$\div e^{-\frac{\pi \epsilon p}{\sqrt{1-\epsilon^2}}}$$

~~100%~~

$$\% M_p = e^{-\frac{\pi \epsilon p}{\sqrt{1-\epsilon^2}}} \times 100$$

5) settling time (t_s):

time reqd for response to reach & stay within the specified range (2 to 5%) of its final value.

$$t_s = 4Z = \frac{4}{\epsilon p \omega_n}$$

6) steady state error (e_{ss}):

difference b/w actual output & desired o/p as time 't' tends to ∞ .

$$e_{ss} = \lim_{t \rightarrow \infty} [r(t) - C(t)]$$

$$t_r = \pi - \tan^{-1} \frac{\sqrt{1-\epsilon^2}}{\epsilon p}$$

$\omega_n \sqrt{1-\epsilon^2}$

$$t_s = \frac{4}{\epsilon p \omega_n}$$

$$t_p = \frac{\pi}{\omega_n \sqrt{1-\epsilon^2}}$$

$$M_p = e^{-\frac{\pi \epsilon p}{\sqrt{1-\epsilon^2}}} \times 100$$

④ Type number of control system:

→ specified for loop transfer function $G(s) H(s)$

→ no. of poles of loop transfer function lying at the origin decides the type no. of system

poles → zeros of P

$$G(s) + H(s) = \frac{K(s+2_1)(s+2_2)}{s^n(s+p_1)(s+p_2)}$$

N = no. of poles at origin

(*) static error constants:

- When control sys. is excited with standard input signal, the steady state error may be zero, constant or ∞ .
- The value depends on type number & input signal.

Types:

$$1) \text{positional error constant } K_p = \lim_{s \rightarrow 0} G(s) H(s)$$

$$2) \text{Velocity error constant } K_v = \lim_{s \rightarrow 0} s G(s) H(s)$$

$$3) \text{acceleration error constant } K_a = \lim_{s \rightarrow 0} s^2 G(s) H(s)$$

where K_p, K_v, K_a → static error constants.

(*) steady state error when i/p is unit step signal:

$$e_{ss} = \lim_{s \rightarrow 0} s E(s)$$

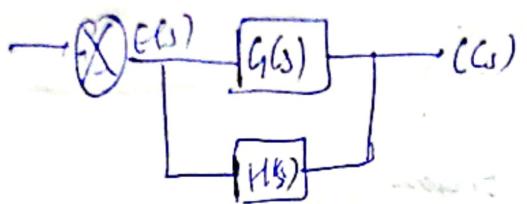
$$\text{where } E(s) = \frac{R(s)}{1+G(s)H(s)}$$

$$e_{ss} = \lim_{s \rightarrow 0} s \cdot \frac{1}{s} \cdot \frac{1}{1+G(s)H(s)} = \frac{1}{1+\left(\lim_{s \rightarrow 0} G(s)H(s)\right)}$$

$$e_{ss} = \frac{1}{1+K_p}$$

(*) Steady state error:

value of error signal when $t \rightarrow \infty$. It is measure of system accuracy. These errors arise from nature of inputs, type of system & from non-linearity of system components.



$$\begin{aligned}
 E(s) &= R(s) - C(s) \cdot H(s) \\
 &= R(s) - [E(s) G(s)] H(s) \\
 E(s)[1 + G(s) H(s)] &= R(s) \\
 E(s) &= \frac{R(s)}{1 + G(s) H(s)}
 \end{aligned}$$

$$\begin{aligned}
 e(t) &= L^{-1}[E(s)] \\
 &= L^{-1}\left[\frac{R(s)}{1 + G(s) H(s)}\right]
 \end{aligned}$$

$$e_{ss} = \lim_{t \rightarrow \infty} e(t)$$

$$\begin{aligned}
 &= e(\infty) \\
 &= \lim_{s \rightarrow 0} s \cdot E(s) \quad (\text{using final value theorem})
 \end{aligned}$$

$$e_{ss} = \lim_{s \rightarrow 0} s E(s)$$

- ① Limitations of time-domain analysis -
- ② don't give info on steady state error when i/p's are other than standard: step, ramp, parabolic.
- ③ does not give precise error value
- ④ applicable to stable systems
- ⑤ does not provide variation of error with time which may be needed for design purpose