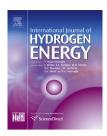


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## **Full Length Article**

# Robust fault diagnosis of proton exchange membrane fuel cells using a Takagi-Sugeno interval observer approach



Damiano Rotondo <sup>a,\*</sup>, Rosa M. Fernandez-Canti <sup>a,1</sup>, Sebastian Tornil-Sin <sup>a,b,1</sup>, Joaquim Blesa <sup>a,b,1</sup>, Vicenç Puig <sup>a,b,1</sup>

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#### ABSTRACT

In this paper, the problem of robust fault diagnosis of proton exchange membrane (PEM) fuel cells is addressed by introducing the Takagi-Sugeno (TS) interval observers that consider uncertainty in a bounded context, adapting TS observers to the so-called interval approach. Design conditions for the TS interval observer based on regional pole placement are also introduced to guarantee the fault detection and isolation (FDI) performance. The fault detection test is based on checking the consistency between the measurements and the output estimations provided by the TS observers. In presence of bounded uncertainty, this check relies on determining if all the measurements lie inside their corresponding estimated interval bounds. When a fault is detected, the measurements that are inconsistent with their corresponding estimations are annotated and a fault isolation procedure is triggered. By using the theoretical fault signature matrix (FSM), which summarizes the effects of the different faults on the available residuals, the fault is isolated by means of a logic reasoning that takes into account the bounded uncertainty, and if the number of candidate faults is more than one, a correlation analysis is used to obtain the most likely fault candidate. Finally, the proposed approach is tested using a PEM fuel cell case study proposed in the literature.

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## Introduction

In the last years, major efforts to reduce greenhouse effects and pollution have increased the demand for

environmentally friendlier energy sources [1]. Fuel cells have been pointed out as a promising alternative way for an energy source in the future, not only because they are considered a zero-emission power source, but also because

<sup>&</sup>lt;sup>a</sup> Department of Automatic Control (ESAII), Technical University of Catalonia (UPC), Rambla de Sant Nebridi 10, 08222 Terrassa, Spain

<sup>&</sup>lt;sup>b</sup> Institut de Robòtica i Informàtica Industrial, CSIC-UPC, Carrer de Llorens i Artigas, 4-6, 08028 Barcelona, Spain

<sup>\*</sup> Corresponding author. Tel.: +34 93 739 89 73. E-mail address: damiano.rotondo@yahoo.it (D. Rotondo).

<sup>&</sup>lt;sup>1</sup> Tel.: +34 93 739 89 73.

they are very quite, and reduce noise pollution. Their energy density makes them specially suitable for embedded generation systems in transport applications. Proton exchange membrane fuel cells (PEMFC) are electrochemical devices in which the energy of a reaction between a fuel, the hydrogen, and an oxidant, the oxygen, is directly and continuously converted into electrical energy, obtaining water as a subproduct [2]. Many control strategies for PEMFC have been proposed in the recent literature, e.g. optimal control [3], model predictive control [4] and sliding mode control [5].

However, fuel cell technology is still too expensive to be accessible to a mass market, and there are still considerable difficulties to overcome. One of the main drawbacks of fuel cell systems is related to the stack lifetime and reliability when working at strongly changing charge conditions, like those that could be usually found in transport applications. Several papers have analyzed the PEMFC durability taking into account the phenomena that could affect the PEMFC lifetime [6-8], e.g. catalyst degradation, fuel starvation or inadequate water management [9-11]. To overcome these drawbacks, adequate fault detection and isolation (FDI) and fault-tolerant control (FTC) strategies are needed to avoid failure modes which could compromise the stack reliability and integrity. Real-time fault diagnosis (detection and isolation) can provide valuable information either for monitoring PEMFC operation, allowing to reconfigure/accommodate the control loop whenever any malfunction is reported, or to suggest preventive maintenance actions, to extend the lifetime of the system and thereby avoiding future damage in the equipment.

In Ref. [12], fault diagnosis of PEMFC is reviewed with a special emphasis on model-based methods. Fuel cells are complex and strongly non-linear systems in which continuously varying parameters are difficult to identify, even for static operation modes. Different authors have chosen different approaches for fuel cell system modeling. Electric equivalent models [13], state space models [14,15] and bond graph representation models [16] are the most representative approaches. Even if these models can represent fuel cell behavior at some operating conditions, their application to fault diagnosis is not so straightforward and, moreover, most of them are related just to the stack and not to the overall system.

Recently, the complex and non-linear dynamics of the power generation systems based on fuel cell technology, described in detail in Ref. [14], led to the use of linear models that include parameters varying with the operating point (known as linear parameter varying or LPV models) not only for advanced control [17] but also for model-based fault diagnosis [18] purposes. The use of Takagi-Sugeno (TS) models [19] is an alternative to LPV models as a linear-like modeling framework that allows extending control [20] and FDI methods to non-linear systems [21]. Thus, this paper will consider TS models and observers for FDI in PEM fuel cells. In the literature, the use of the TS has been widely considered for FDI (see, e.g. Refs. [21,22]) and fault estimation/FTC (see e.g. Refs. [23,24]). However, due to the presence of parametric uncertainties in the TS model, the design of a conventional observer that converges to the exact value of the state is complicated [25]. In this situation, an interval estimation is still possible, and an observer that evaluates the set of admissible values for the state at each time instant can be designed [25–29].

The main contribution of this paper is to address the problem of robust fault diagnosis of PEM fuel cells by introducing the TS interval observers that consider uncertainty in a bounded context, adapting TS observers to the socalled interval approach. Design conditions for the TS interval observer based on regional pole placement are also introduced to guarantee the FDI performance. A bank of observers is designed by applying the structural analysis [30] to the PEMFC model considering the set of available sensors. The fault detection test is based on checking the full consistency between the measurements and the output estimations provided by the TS interval observers. In presence of bounded uncertainty, this check relies on determining if all the measurements lie inside their corresponding estimated interval bounds. In the case a fault is detected, the measurements that are inconsistent with their corresponding estimations are annotated and a fault isolation procedure is triggered. By using the theoretical fault signature matrix (FSM), the fault is isolated by means of a logic reasoning that takes into account the presence of bounded uncertainty, and a posterior correlation analysis provides the most likely fault candidate. Finally, the proposed approach is tested using a well known PEMFC case study proposed in the literature in Ref. [14] by deriving an interval TS model and observer.

## Modeling of the PEMFC

#### Description

The fault diagnosis approach proposed in this paper is tested on a typical PEMFC system (see Fig. 1) based on the model provided in Ref. [2].

The fuel cell, in response to a current demand by an electrical load, consumes oxygen and hydrogen and generates water and heat. The hydrogen is provided by the hydrogen supply system, whose main components are a pressurized hydrogen tank and a supply servo valve that allows controlling the hydrogen flow or pressure. The air supply system, consisting mainly of a compressor driven by an electric motor, provides the air flow. The control of the hydrogen and air supply systems aims at maintaining the required partial pressures of the hydrogen and air entering into the anode and the cathode of the stack, respectively. Additionally, since the compressor provides air at a high temperature due to the increased pressure, the cooling system is used to reduce the temperature of the air entering into the stack in order to prevent the fuel cell membrane from damaging. Finally, a humidifier acts on the cathode path to prevent membrane damage due to dehydration.

The main variables involved in the PEMFC system operation are listed in Table 1.

Following other works in the literature, e.g. Ref. [31], six variables are assumed to be measured:

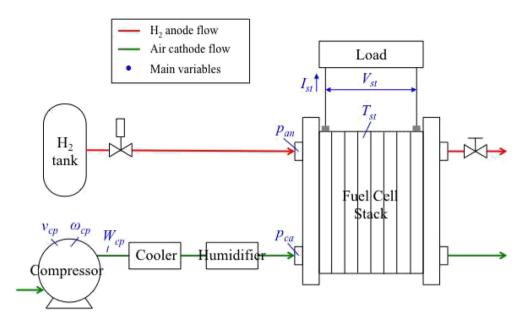


Fig. 1 – PEM fuel cell system.

- Compressor speed ( $\omega_{cp}$ ).
- Pressure in the cathode supply manifold  $(p_{sm})$ .
- Mass of hydrogen in the anode  $(m_{H_2})$ .
- Pressure in the anode  $(p_{an})$ .
- Pressure in the cathode  $(p_{ca})$ .
- Stack voltage (vst).

The feedback control and safe operation of a PEMFC can be compromised by sensor faults. Hence, it is important to develop fault diagnosis methodologies that are able to detect and isolate such faults.

In this paper, faults affecting the previously listed sensors are considered. These faults will be identified as  $f_1$ ,  $f_2$ , ...,  $f_6$ , in accordance with the order provided in the list. The extension to other types of faults is straightforward by applying the proposed methodology to the new set of faults.

#### Linear-like representation of the PEMFC model

The model used in this paper is derived from the non-linear equations presented in Ref. [14], that are valid under the following assumptions: the stack temperature  $T_{\rm st}$  is constant (it changes slowly with respect to the other system

Table 1 $-$ PEMFC variables.	
Symbol	Description
$v_{cp}$	Compressor input voltage
$\omega_{cp}$	Compressor speed
$W_{cp}$	Compressor air flow
$p_{ca}$	Air pressure in the cathode
p <sub>an</sub>	Hydrogen pressure in the anode
$T_{ m st}$	Stack temperature
$I_{st}$	Stack current
$v_{\rm st}$	Stack voltage

variables); the temperature and humidity of the inlet reactant flows are perfectly controlled; the cathode and anode volumes of the multiple fuel cells are lumped as a single stack cathode and anode volumes; all the reactants behave as ideal gases.

According to [14], the stack current  $I_{st}$  can be considered as the system input and nine state variables can be used to describe the system dynamics in a non-linear state space representation: mass of oxygen in the cathode  $(m_{O_2})$ , mass of hydrogen in the anode  $(m_{H_2})$ , mass of nitrogen in the cathode  $(m_{N_2})$ , compressor speed  $(\omega_{cp})$ , air pressure in the supply manifold  $(p_{sm})$ , air mass in the supply manifold  $(m_{sm})$ , mass of water in the anode  $(m_{w,an})$ , mass of water in the cathode  $(m_{w,ca})$ , and pressure in the return manifold  $(p_{rm})$ .

The non-linear model of the PEMFC can be reshaped into a linear-like representation, obtaining a model with the following structure:

$$\begin{pmatrix} \dot{m}_{O_2} \\ \dot{m}_{H_2} \\ \dot{m}_{N_2} \\ \dot{m}_{sm} \\ \dot{p}_{sm} \\ \dot{p}_{rm} \\ \dot{\omega}_{cp} \\ \dot{m}_{w,ca} \end{pmatrix} = \begin{bmatrix} \widetilde{\alpha}_{11} & 0 & \widetilde{\alpha}_{13} & 0 & \alpha_{15} & \widetilde{\alpha}_{16} & 0 & 0 & 0 \\ 0 & \widetilde{\alpha}_{22} & 0 & 0 & \widetilde{\alpha}_{25} & 0 & 0 & \widetilde{\alpha}_{28} & 0 \\ \widetilde{\alpha}_{31} & 0 & \widetilde{\alpha}_{33} & 0 & \alpha_{35} & \widetilde{\alpha}_{36} & 0 & 0 & 0 \\ \widetilde{\alpha}_{51} & 0 & \widetilde{\alpha}_{53} & 0 & \widetilde{\alpha}_{55} & 0 & 0 & 0 & 0 \\ \widetilde{\alpha}_{51} & 0 & \widetilde{\alpha}_{53} & 0 & \widetilde{\alpha}_{55} & 0 & 0 & 0 & 0 \\ \widetilde{\alpha}_{61} & 0 & \alpha_{63} & 0 & 0 & \widetilde{\alpha}_{66} & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & \widetilde{\alpha}_{77} & 0 & 0 \\ 0 & \widetilde{\alpha}_{82} & 0 & 0 & \widetilde{\alpha}_{85} & 0 & 0 & \widetilde{\alpha}_{88} & 0 \\ \widetilde{\alpha}_{91} & 0 & \widetilde{\alpha}_{93} & 0 & \widetilde{\alpha}_{95} & \widetilde{\alpha}_{96} & 0 & 0 & 0 \end{bmatrix} \begin{pmatrix} m_{O_2} \\ m_{H_2} \\ m_{N_2} \\ m_{sm} \\ p_{sm} \\ p_{rm} \\ \omega_{cp} \\ m_{w,an} \\ m_{w,ca} \end{pmatrix} + (\widetilde{\beta}_1 & \widetilde{\beta}_2 & \widetilde{\gamma}_3 & \widetilde{\gamma}_4 & \widetilde{\gamma}_5 & \gamma_6 & \widetilde{\gamma}_7 & \widetilde{\gamma}_8 & \widetilde{\gamma}_9 )^T I_{st}$$

$$(1)$$

where the coefficients with tilde, e.g.  $\tilde{\alpha}_{11}$ , are functions of the scheduling parameter  $I_{st}$  and the remaining are constant. All the coefficients depend on constants (physical and chemical constants, physical coefficients related with the PEMFC construction and operating conditions) whose values are taken from Ref. [14].

The measured variables  $\omega_{cp}$ ,  $p_{sm}$  and  $m_{H_2}$  correspond directly to state variables in Ref. [1]. On the other hand,  $p_{an}$ ,  $p_{ca}$  and  $v_{st}$  are related to some of the state variables according to the following general expressions:

$$p_{ca} = f(m_{O_2}, m_{N_2})$$
 (2)

$$p_{an} = f(m_{H_2}, m_{w,an})$$
 (3)

$$v_{st} = f(m_{H_2}, m_{O_2}, p_{ca})$$
 (4)

## Takagi-Sugeno interval observers

#### System representation

TS systems, as proposed in Ref. [19], are described by local models merged together using IF-THEN rules [32]. In particular, in this paper, we will consider an uncertain discrete-time TS system, as follows:

$$\begin{array}{l} \text{IF } \vartheta_{1}(k) \text{ is } M_{i1} \text{ AND } ... \text{ AND } \vartheta_{p}(k) \text{ is } M_{ip} \\ \text{THEN} \left\{ \begin{array}{l} x_{i}(k+1) = (A_{0i} + \Delta A_{i})x(k) + B_{i}u(k) + d(k) \\ y_{i}(k) = Cx(k) + v(k) & i = 1, ..., N \end{array} \right. \end{array} \tag{5}$$

where  $\mathbf{x}(k) \in \mathbb{R}^{n_x}$  is the state vector,  $u(k) \in \mathbb{R}^{n_u}$  is the input vector,  $y(k) \in \mathbb{R}^{n_y}$  is the output vector available for measurements,  $d(k) \in \mathbb{R}^{n_x}$  is the exogenous disturbance and  $v(k) \in \mathbb{R}^{n_y}$  is the measurement noise.  $M_{ij}$  denote the fuzzy sets and N is the number of model rules.  $\vartheta_1(k), ..., \vartheta_p(k)$  are the premise variables that can be functions of endogenous and/or exogenous variables. Each linear consequent equation is called a subsystem.

Given x(k), u(k), d(k) and v(k), the state of the TS system at time k+1 and the output at time k can be easily inferred as:

$$x(k+1) = \sum_{i=1}^{N} \rho_i(\vartheta(k))[(A_{0i} + \Delta A_i)x(k) + B_iu(k) + d(k)]$$
 (6)

$$y(k) = Cx(k) + v(k) \tag{7}$$

where  $\vartheta(k) = [\vartheta_1(k), ..., \vartheta_p(k)]^T$  is the vector containing the premise variables, and  $\rho_i(\vartheta(k))$  is defined as follows:

$$\rho_{i}(\vartheta(k)) = \frac{\prod_{j=1}^{p} M_{ij}(\vartheta_{j}(k))}{\sum_{i=1}^{N} \prod_{j=1}^{p} M_{ij}(\vartheta_{j}(k))}$$
(8)

where  $M_{ij}(\vartheta_j(k))$  is the grade of membership of  $\vartheta_j(k)$  in  $M_{ij}$  and  $\rho_i(\vartheta(k))$  is such that:

$$\begin{cases}
\sum_{i=1}^{N} \rho_i(\vartheta(k)) = 1 \\
\rho_i(\vartheta(k)) > 0, \quad i = 1, \dots, N
\end{cases}$$
(9)

The matrices  $A_{0i} \in \mathbb{R}^{n_x \times n_x}$ ,  $B_i \in \mathbb{R}^{n_x \times n_u}$  and  $C \in \mathbb{R}^{n_y \times n_x}$  are known, whereas the matrices  $\Delta A_i \in \mathbb{R}^{n_x \times n_x}$  represent the uncertainty, and are not known. Also, the exact values of d(k) and v(k) are not available. However, it is assumed that some bounds are known, as follows:

$$\underline{d}(k) \le d(k) \le \overline{d}(k) \tag{10}$$

$$|v(k)| < V(k) \tag{11}$$

$$\Delta A_i \le \Delta A_i \le \overline{\Delta A_i} \tag{12}$$

for some known functions  $\underline{d}(k)$ ,  $\overline{d}(k)$ , V(k) and for some known matrices  $\underline{\Delta A_i}$ ,  $\overline{\Delta A_i} \in \mathbb{R}^{n_x \times n_x}$ . The relations  $\leq$  are understood elementwise.

The fault diagnosis test is based on checking whether the measurements are consistent with the system model or not. However, due to the presence of uncertainty, unknown disturbances and noise, an exact estimation  $\widehat{x}(k)$  of the state x(k), to be used for comparison with the measurements, cannot be obtained. However, taking into account (10)–(12), it is possible to design an observer that provides an interval estimation of x(k), i.e. some lower and upper estimates  $\underline{x}(k)$ ,  $\overline{x}(k)$  of x(k), such that:

$$\underline{x}(k) \le x(k) \le \overline{x}(k) \quad \forall k \in \mathbb{Z}$$
 (13)

This observer will be denoted as TS interval observer.

#### The TS interval observer

The TS interval observer for the system (5) has the following form, obtained as a slight modification of the one presented in Ref. [25]:

$$THEN \begin{cases} x_{i}(k+1) = \left[A_{0i} - \underline{L_{i}}C\right]\underline{x}(k) + B_{i}u(k) \\ + \underline{\Delta A_{i}}^{+}\underline{x}^{+}(k) - \overline{\Delta A_{i}}^{+}\underline{x}^{-}(k) \\ - \underline{\Delta A_{i}}^{-}\overline{x}^{+}(k) + \overline{\Delta A_{i}}^{-}\overline{x}^{-}(k) \\ + \underline{L_{i}}y(k) - \left|\underline{L_{i}}\right|V(k)E_{n_{y}} + \underline{d}(k) \end{cases} \tag{14} \\ \overline{x_{i}}(k+1) = \left[A_{0i} - \overline{L_{i}}C\right]\overline{x}(k) + B_{i}u(k) \\ + \overline{\Delta A_{i}}^{+}\overline{x}^{+}(k) - \underline{\Delta A_{i}}^{+}\overline{x}^{-}(k) \\ - \overline{\Delta A_{i}}^{-}\underline{x}^{+}(k) + \underline{\Delta A_{i}}^{-}\underline{x}^{-}(k) \\ + \overline{L_{i}}y(k) + \left|\overline{L_{i}}\right|V(k)E_{n_{y}} + \overline{d}(k) \end{cases}$$

where  $\underline{L_i} \in \mathbb{R}^{n_x \times n_y}$  and  $\overline{L_i} \in \mathbb{R}^{n_x \times n_y}$  are the observer gains to be designed,  $\overline{\Delta A_i}^+ = \max\{0, \overline{\Delta A_i}\}, \quad \overline{\Delta A_i}^- = \overline{\Delta A_i}^+ - \overline{\Delta A_i},$   $\underline{\Delta A_i}^- = \max\{0, \underline{\Delta A_i}\}, \quad \underline{\Delta A_i}^- = \underline{\Delta A_i}^+ - \underline{\Delta A_i}, \quad \overline{x}^+ = \max\{0, \overline{x}\},$   $\overline{x}^- = \overline{x}^+ - \overline{x}, \, \underline{x}^+ = \max\{0, \underline{x}\}, \, \underline{x}^- = \underline{x}^+ - \underline{x} \text{ and, finally, } E_{n_y} \in \mathbb{R}^{n_y \times 1}$  denotes the column vector with all elements equal to 1.

Then, the lower and upper estimates of y(k) can be obtained as:

$$y(k) = C^{+}\underline{x}(k) - C^{-}\overline{x}(k)$$
(15)

$$\overline{y}(k) = C^+ \overline{x}(k) - C^- \underline{x}(k)$$
 (16) where  $C^+ = \max\{0, C\}$  and  $C^- = C^+ - C$ .

#### Design conditions

In order to guarantee a proper dynamics regarding fault detection, the interval estimation of x(k) with constraints on the closed-loop poles location of the interval observer is desired. Although from the mathematical point of view the concept of poles refers to linear time invariant (LTI) systems,

[33] has shown that when pole placement in linear matrix inequality (LMI) regions [34] is considered, some transient properties related to the poles location, e.g. decay ratio, would also hold for the case of fuzzy TS systems. As discussed in Ref. [35], the observer poles play an important role in the observer fault sensitivity. The following theorem, based on the reasoning made by Ref. [25], provides some conditions to accomplish this goal under the assumption that:

$$\underline{x}(0) \le x(0) \le \overline{x}(0) \tag{17}$$

**Theorem 1**. Given an LMI region, defined as [34]:

$$\mathscr{D} = \{ z \in \mathbb{C} : f_{\mathscr{D}}(z) < 0 \} \tag{18}$$

where the characteristic function  $f_{\varnothing}(z)$  is defined as:

$$f_{\mathscr{D}}(z) = \alpha + z\beta + z^*\beta^{T} = \{\alpha_{kl} + \beta_{kl}z + \beta_{lk}z^*\}_{k,l \in [1,m]}$$
(19)

with  $\alpha=\alpha^T\in\mathbb{R}^{m\times m}$  and  $\beta\in\mathbb{R}^{m\times m}$ , if there exist a diagonal matrix  $P\in\mathbb{R}^{2n_x\times 2n_x}$ , a symmetric matrix  $Q=Q^T\in\mathbb{R}^{2n_x\times 2n_x}$ , block diagonal matrices  $W_i\in\mathbb{R}^{2n_x\times 2n_u}$ , i=1,...,N with the following structure:

$$W_{i} = \begin{pmatrix} \frac{W_{i}}{0} \in \mathbb{R}^{n_{x} \times n_{u}} & 0\\ 0 & \overline{W_{i}} \in \mathbb{R}^{n_{x} \times n_{u}} \end{pmatrix}$$
 (20)

and constants  $\varepsilon_1 > 0$ ,  $\varepsilon_2 > 0$ ,  $\gamma > 0$  such that:

$$P > 0$$
 (21)

$$Q > 0 \tag{22}$$

and, for i=1, ...,N:

$$\begin{pmatrix} \frac{P}{1+\epsilon_{1}} & PD_{i} - W_{i}\Upsilon & \frac{P}{1+\epsilon_{1}} \\ (PD_{i} - W_{i}\Upsilon)^{T} & P - Q - \gamma\eta^{2}I_{2n_{x}} & 0 \\ \frac{P}{1+\epsilon_{1}} & 0 & \gamma I_{2n_{x}} - \epsilon P \end{pmatrix} \geq 0$$
 (23)

$$P\begin{bmatrix} A_{0,i} & 0\\ 0 & A_{0,i} \end{bmatrix} - W_i \Upsilon \ge 0 \tag{24}$$

$$\begin{cases} \alpha_{kl} P + \beta_{kl} \begin{pmatrix} A_{0,i}^T & 0 \\ 0 & A_{0,i}^T \end{pmatrix} P - \Upsilon^T W_i^T \end{pmatrix} + \beta_{lk} \begin{pmatrix} P \begin{pmatrix} A_{0,i} & 0 \\ 0 & A_{0,i} \end{pmatrix} - W_i \Upsilon \end{pmatrix}_{k,l \in [1,m]} < 0$$
(25)

with:

$$D_{i} = \begin{pmatrix} A_{0,i} + \underline{\Delta A_{i}}^{+} & 0\\ 0 & A_{0,i} + \overline{\Delta A_{i}}^{+} \end{pmatrix}$$
 (26)

$$\Upsilon = \begin{pmatrix} C & 0 \\ 0 & C \end{pmatrix} \tag{27}$$

$$\eta = 2 \max_{i=1,\dots,N} \left( \left\| \underline{\Delta A_i}^+ - \overline{\Delta A_i}^+ \right\|_2 + \left\| \underline{\Delta A_i}^- \right\|_2 + \left| \left| \overline{\Delta A_i}^- \right| \right|_2 \right) \tag{28}$$

$$\varepsilon = 1 + \varepsilon_2 + (1 + \varepsilon_1)^{-1} \tag{29}$$

then, the TS interval observer (14) with gains calculated as:

$$L_i = P^{-1}W_i \tag{30}$$

$$\overline{L_i} = P^{-1}\overline{W_i} \tag{31}$$

ensures the interval estimation of x(k) given by (13), provided that (10)–(12) and (17) hold, and that:

$$\exists ! i \in \{1, ..., N\} | \rho_i(\vartheta(k)) = 1, \rho_i(\vartheta(k)) = 0 \ \forall j \neq i, \forall k \in \mathbb{Z}$$
 (32)

Moreover, for each subsystem, the closed-loop poles of the TS interval observer are in  $\mathscr{D}$ .

Proof: See Appendix A. ■

The overall system of LMIs (21)—(24) and (25) can be solved efficiently using available solvers, e.g. the SeDuMi [36] in the YALMIP toolbox [37].

It is worth remarking that the proposed TS interval observer operates under the assumption that the premise variables are known. Cases in which such assumption does not hold could be dealt with using results about TS observers with unmeasurable premise variables, see e.g. Ref. [38], but this extension goes beyond the goal of this paper.

## Fault diagnosis

#### Overview of the proposed fault diagnosis methodology

The proposed fault diagnosis methodology uses a bank of TS interval observers, designed as described in previous section, to take into account the uncertainty, and can be considered an alternative to the single observer methodology presented in Refs. [18,39]. By means of structural analysis, each observer can be made sensitive to a subset of faults, thus avoiding the need to perform a complex sensitive analysis, like the one in Ref. [39].

The whole procedure can be divided into off-line and on-line stages, and it can be summarized as follows.

#### Off-line stage

The linear-like representation of the PEMFC model provided in (1) is used as a starting point to obtain a set of analytical redundancy relations (ARRs) by means of structural analysis [30]. ARRs are equations relating only measured variables (so they can be evaluated on-line), that are satisfied in absence of faults. Each ARR is converted into a TS model by choosing the premise variables, and by considering a set of operating points that cover the whole range of system operation. Then, the model uncertainty associated to the TS models is adjusted in such a way that the outputs of the TS interval observers designed using these models "cover", with minimal overbounding, all the measurements obtained in a complete fault-free scenario. This assures that no false alarms will occur when these TS interval observers are used for fault detection purposes.

#### On-line stage

At each time-instant, the TS observers provide estimations of the bounds of the measured variables, which would be compatible with the fault-free operation. The differences between the measurements and these bounds are called *residuals*, and are used for fault detection and isolation, as detailed in the following.

#### Fault detection

Fault detection is based on generating residuals, i.e. on comparing the measurements of the physical variables of the process y(k) with the estimations  $\hat{y}(k)$  provided by the associated system model:

$$r(k) = y(k) - \widehat{y}(k) \tag{33}$$

where  $r(k) \in \mathbb{R}^{n_y}$  is the residual vector.

When considering model uncertainty, the residual generated by (33) deviates from zero even in a non-faulty scenario. To cope with the parameter uncertainty effect, a passive robust approach based on adaptive thresholding can be used [40]. Thus, using this passive approach, the effect of the parameter uncertainty on the residual r(k) associated to each system output y(k) is bounded by the interval that will include the zero value in absence of faults. Hence, a fault detection test can be formulated as:

$$0 \in [\underline{r}(k), \overline{r}(k)] \tag{34}$$

where:

$$\underline{r}(k) = y(k) - \overline{y}(k) \tag{35}$$

$$\overline{r}(k) = y(k) - y(k) \tag{36}$$

being y(k) the output coming from the sensors,  $\underline{y}(k)$  and  $\overline{y}(k)$  the bounds of the predicted output given by (15)—(16).

The fault detection test (34) can be alternatively formulated as:

$$y(k) \in \left[\underline{y}(k), \overline{y}(k)\right]$$
 (37)

#### Fault isolation

Fault isolation consists in identifying the faults affecting the system. It is carried out on the basis of fault signatures, generated by the fault detection module, and their relation with all the considered faults. Robust residual evaluation presented in (34) allows obtaining a set of fault signatures  $\phi(k) = [\phi_1(k), \phi_2(k), ..., \phi_{n_y}(k)]$ , where each fault indicator is given by:

$$\phi_{i}(k) = \begin{cases} 0 & \text{if} \quad 0 \in [\underline{r}_{i}(k), \overline{r}_{i}(k)] \\ 1 & \text{if} \quad 0 \notin [\underline{r}_{i}(k), \overline{r}_{i}(k)] \end{cases}$$
(38)

The values of the fault signatures are compared with the theoretical FSM, denoted as M. An element  $m_{i,j}$  (i indicates rows, j indicates columns) of M is equal to 1 if the fault  $f^j$  affects the computation of the residual  $r_i$ ; otherwise, the element  $m_{i,j}$  is zero-valued.

Due to the uncertainty and to the different dynamics of the residuals, in presence of a given fault, the different fault indicators  $\phi_i(k)$  that are sensitive to this particular fault can become active for non-overlapping short periods of time. In this case, proper fault isolation cannot be achieved. In order to minimize this problem and achieve correct fault isolation, a memory component  $\phi_i^{max}(k)$  that stores the maximum value of  $\phi_i(k)$  in a prefixed waiting time  $T_w$  is used in Algorithm 1, as proposed in Ref. [41].

Additionally, to take into account the particular case where, due to the presence of uncertainty, a given fault does not

trigger some of the fault indicators that are potentially affected by it according to the theoretical FSM, the strategy proposed in Ref. [42], based on considering only the components of the observed fault signature that are equal to 1, is used.

As a result of the previous strategies, the procedure could attribute the observed fault signature to more than one fault candidate, although guaranteeing that the real fault present in the system is among the provided candidates [42]. Due to this fact, a final decision based on taking into account the correlation of the nominal residual  $r^0(k)$  with the steady state fault sensitivities  $s_j$  of the different fault candidates is implemented to obtain the most likely fault (see Ref. [41] for more details).

The proposed fault detection and isolation procedure is summarized in Algorithm 1.

```
Algorithm 1 Fault diagnosis
 1: fault \leftarrow FALSE
 2: isolation \leftarrow FALSE
 3: \mathcal{FC} \leftarrow \{f^1, f^2, \dots, f^{n_f}\}
 5: while isolation \neq TRUE do
        k \leftarrow k + 1
        Obtain input-output data \{u(k), y(k)\} at time instant k
        Obtain [r_i(k), \overline{r_i}(k)], i = 1, \dots, n_y using Eqs. (35)-(36)
 9.
        for i = 1 to n_y do
10:
           if 0 \notin [r_i(k), \overline{r_i}(k)] then
              \phi_i(k) = 1
11:
               if fault \neq TRUE then
12:
13:
                  k_f = k
               end if
14:
               fault \leftarrow \text{TRUE}
15:
16:
            else
              \phi_i(k) = 0
17:
18:
            end if
19:
           \phi_i^{max}(k) = \max \quad \phi_i(j)
                            j \in \{k-T_W,...,k\}
        end for
20:
21:
        if fault == TRUE then
22:
            for i = 1 to n_y do
              if \phi_i^{max}(k) = 1 then
23.
                  for j = 1 to n_f do
24:
                    if m_{i,j} == 0 then \mathcal{FC} \leftarrow \mathcal{FC} - f^j
25:
26:
                     end if
27:
28:
                  end for
29:
              end if
30:
            end for
31:
            if k == k_f + T_W then
              isolation \leftarrow TRUE
32.
            end if
33:
34:
        end if
35: end while
36: Fault candidate set \mathcal{FC}
37: compute \mathbf{r}^0(k) = \left(\frac{\underline{r}_1(k) + \overline{r}_1(k)}{2}\right).
38: for j = 1 to |\mathcal{FC}| do
        \pi_j(k) = \frac{\mathbf{r}^{\cdot}(k) - \mathbf{s}_j}{\|\mathbf{r}^0(k)\| \, \|\mathbf{s}_j\|}
40: end for
41: Candidate fault is: \arg \max_{j=1,\cdots,|FC|} \pi_j(k)
```

#### **Application results**

## Off-line design

As it can be noticed by looking at the model structure in (1), the states are highly coupled. However, the information about the variables that are measured can be used for decoupling the

system in sub-models, using structural analysis [30]. In this way, four decoupled sub-models can be obtained from Equations (1)—(4), as detailed in Appendix B.

The decoupled sub-models allow obtaining the following residuals:

$$r_1 \triangleq \omega_{cp} - \widehat{\omega}_{cp}$$
 (39)

$$r_2 \triangleq p_{sm} - \hat{p}_{sm} \tag{40}$$

$$r_3 \triangleq \widehat{m}_{0_2,1} - \widehat{m}_{0_2,2}$$
 (41)

$$r_4 \triangleq m_{\rm H_2} - \widehat{m}_{\rm H_2} \tag{42}$$

where the indicates an estimation while the absence of indicates a measurement.

It must be noticed that the mass of oxygen  $m_{O_2}$  is not a measured variable according to the list provided in the PEMFC modelling Section. Hence, a second estimation for it (the first one is provided by the cathode plus return manifold observer) is necessary to obtain analytical redundancy. This estimation can be computed by inverting numerically the expression that relates  $m_{O_2}$  with the measured stack voltage  $v_{\rm st}$  [31].

The structural analysis of the system also allows determining the FSM associated to the set of faults considered in the PEMFC modelling Section and the set of residuals (39)–(42). The obtained FSM is presented in Table 2. Since all the columns in the FSM are different, all the faults are potentially isolable.

It is worth highlighting that, by looking at the FSM (Table 2) and at the equations of the decoupled sub-models (Appendix B), the importance of the availability of the measurements listed in the PEMFC modelling Section becomes clear. In fact, with the exception of the compressor speed  $\omega_{cp}$  sensor (which is needed for feedback control purposes), the lack of one of the considered sensors would lead either to non-detectability/non-isolability issues (e.g. if  $p_{sm}$  is not available, both the  $r_2$  row and the  $f_2$  column in the FSM should be removed, and it is straightforward to see that the faults  $f_5$  and  $f_6$  would not be isolable) or to the appearance of unknown inputs in the decoupled sub-models, which would impede the correct operation of the corresponding TS interval observer.

Then, a bank of TS interval observers is designed for estimating lower and upper bounds of  $\omega_{cp}$ ,  $p_{sm}$ ,  $m_{O_2}$  and  $m_{H_2}$ . In order to take into account the variation of the model parameters with the value of the stack current  $I_{st}$ , the interval of possible values of  $I_{st} \in [100 \text{ mA}, 300 \text{ mA}]$  has been gridded using a step of 10 mA, obtaining a set of 21 possible operating points and continuous-time TS representations of the ARRs. Then, in order to obtain discrete-time TS representations suitable for applying the interval observer theory introduced

Table 2	– Fault s	signature	matrix.			
ARR	$f_1$	$f_2$	$f_3$	$f_4$	$f_5$	$f_6$
r <sub>1</sub>	1	0	0	0	0	0
$r_2$	0	1	0	0	1	0
r <sub>3</sub>	0	1	1	0	1	1
r <sub>4</sub>	0	1	1	1	0	0

in the TS interval observer Section, an Euler discretization with sampling time  $T_s = 0.01$  s has been performed.

For each TS representation, a TS interval observer has been designed using the design conditions described in the TS interval observer Section. In particular, disks of center (0.01,0) and radius 0.01 have been chosen as LMI regions, in order to obtain TS interval observers whose behavior recalls the one of a predictor. Notice that in order to design the interval observer, the uncertainty must be estimated. Here, an adaptation to the TS interval observers of the approach proposed in Ref. [43] for interval observers is used. This approach relies on adjusting the parameter uncertainty to guarantee that no false alarms appear using data from non-faulty scenarios. Also, it has been assumed that the data coming from the sensors is affected by bounded noise with a bound of 1% of the measurement. On the other hand, a bound of 2% of the estimation has been used for the mass of oxygen  $m_{O_2}$  estimated from the stack voltage  $v_{st}$ .

It is worth remarking that, from a theoretical point of view, the design conditions described in the TS interval observer Section guarantee the properties of the TS interval observer only at the design points. However, from a practical point of view, due to the use of a common Lyapunov matrix P, it is reasonable to assume that, if the gridding of the parameter space is dense enough, then the desired properties of interval bounded estimation and closed-loop pole clustering would also hold at operating points different from the design ones. A deep theoretical study of this fact is possible using the results developed by Ref. [44].

## On-line simulation results

The proposed methodology has been applied to the case study with an extensive performance assessment phase, where simulation experiments were repeated for different faults and different magnitudes. In particular, faults where the value provided by the sensor corresponds to the measured value plus a percentage due to the fault are considered (e.g.  $f_1 = 30\%$  means that the compressor sensor  $\omega_{cp}$  returns 1.3 times the value that would have returned in the non-faulty case).

The results of the performed simulations are shown in Tables 3–8 for each of the six available sensors. For each considered sensor fault, the rate of successful fault detection and isolation, as well as the average detection and isolation times are reported. It can be seen that, for all the sensors, there are critical fault magnitudes: for magnitudes above the upper critical one (e.g.  $f_1 = 6\%$  for the compressor sensor  $\omega_{cp}$ ), the performances of the fault diagnosis approach are very

Table 3 $-$ Results of fault diagnosis for $f_1$ ( $\omega_{cp}$ sensor fault).				
$f_1$	Successful fault detection	Successful fault isolation	Ŧ <sub>d</sub>	ī <sub>i</sub>
≥6.6%	100%	100%	0.01 s	0.01 s
6.2%, 6.4%	95%	95%	0.01 s	0.01 s
5.8%, 6%	80%	80%	0.01 s	0.01 s
5.6%	55%	55%	0.01 s	0.01 s
5.4%	35%	35%	0.01 s	0.01 s
5.2%	15%	15%	0.01 s	0.01 s
≤5%	0%	0%	-	-

Table 4 –	Table 4 – Results of fault diagnosis for $f_2$ ( $p_{sm}$ sensor fault).				
$f_2$	Successful fault detection	Successful fault isolation	<del>T</del> d	<del></del> t <sub>i</sub>	
≥6%	100%	100%	0.01 s	0.01 s	
5.5%	100%	100%	0.015 s	0.015 s	
[3.5%, 5%]	100%	100%	0.02 s	0.03 s	
3%	100%	100%	0.02 s	0.031 s	
2.5%	100%	100%	0.022 s	0.034 s	
2%	100%	100%	0.033 s	0.055 s	
1.5%	100%	80%	0.103 s	0.694 s	
1%	10%	0%	0.850 s	-	
≤0.5%	0%	0%	_	=	

Table	Table 5 – Results of fault diagnosis for $f_3$ ( $v_{st}$ sensor fault).				
$f_3$	Successful fault detection	Successful fault isolation	$\overline{t}_d$	$\overline{t}_{i}$	
≥3.5%	100%	100%	0.01 s	0.01 s	
3%	100%	100%	2.361 s	2.361 s	
2.5%	100%	100%	9.597 s	9.597 s	
2%	100%	100%	11.507 s	11.507 s	
1.5%	100%	100%	11.998 s	11.998 s	
≤1%	0%	0%	_	_	

Table fault).	Table 6 – Results of fault diagnosis for $f_4$ ( $m_{H_2}$ sensor fault).			
f <sub>4</sub>	Successful fault detection	Successful fault isolation	<del>T</del> d	$\overline{t}_{i}$
≥14%	100%	100%	0.01 s	0.02 s
13%	95%	95%	0.01 s	0.02 s
12%	85%	85%	0.01 s	0.02 s
11%	70%	70%	0.01 s	0.02 s
10%	60%	60%	0.01 s	0.02 s
9%	20%	20%	0.01 s	0.02 s
8%	15%	15%	0.01 s	0.02 s
≤7%	0%	0%	0.01 s	-

Table	Table 7 – Results of fault diagnosis for $f_5$ ( $p_{ca}$ sensor fault).					
f <sub>5</sub>	Successful fault detection	Successful fault isolation	$\overline{t}_d$	$\overline{t}_{i}$		
≥25%	100%	100%	0.02 s	0.02 s		
20%	100%	100%	0.02 s	0.026 s		
15%	100%	100%	0.021 s	0.078 s		
10%	10%	0%	0.03 s	_		
5%	0%	0%	-	-		

Table	Table 8 $-$ Results of fault diagnosis for $f_6$ ( $p_{an}$ sensor fault).				
f <sub>6</sub>	Successful fault detection	Successful fault isolation	<del>T</del> d	$\overline{t}_{i}$	
≥3%	100%	100%	0.02 s	0.02 s	
2.5%	100%	100%	0.022 s	0.025 s	
2%	100%	100%	0.031 s	0.034 s	
1.5%	100%	100%	0.218 s	0.231 s	
≤1%	0%	0%	_	_	

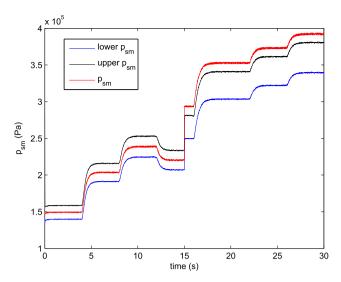


Fig. 2  $-p_{sm}$ ,  $\underline{p}_{sm}$  and  $\overline{p}_{sm}$  with  $f_2=30\%$  (fault in the supply manifold sensor  $p_{sm}$ ).

good, since correct fault detection and isolation are always achieved within a small time. On the other hand, for fault magnitudes below the lower critical one (e.g.  $f_1=5\%$  for  $\omega_{cp}$ ), fault detection is not achievable, i.e. those faults are not detectable. Between these two extremes, the behavior of the fault diagnosis approach varies depending on the considered fault. In some cases, the smaller the fault magnitude is, the smaller the rate of successful fault detection and isolation is, but with the detection/isolation time remaining unaffected. This is the case of the faults in  $\omega_{cp}$  and  $m_{H_2}$ . In the remaining cases, the effect of decreasing the fault magnitude is visible on the detection/isolation time.

Moreover, to better illustrate the proposed method, let us consider the following two fault scenarios:

• Fault scenario 1:  $f_2 = 30\%$  (fault in the supply manifold sensor  $p_{sm}$ ), starting from t = 15 s

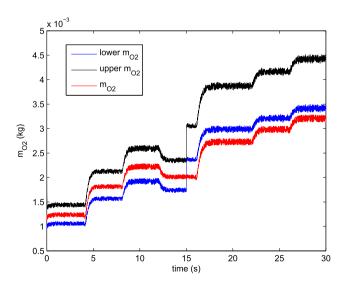


Fig.  $3 - m_{O_2}$ ,  $\underline{m}_{O_2}$  and  $\overline{m}_{O_2}$  with  $f_2 = 30\%$  (fault in the supply manifold sensor  $p_{sm}$ ).

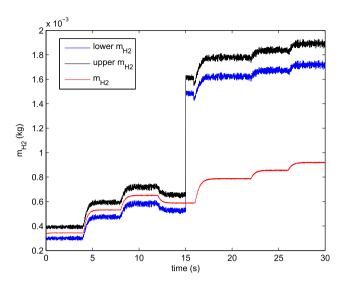


Fig.  $4 - m_{\rm H_2}$ ,  $\underline{m}_{\rm H_2}$  and  $\overline{m}_{\rm H_2}$  with  $f_2 = 30\%$  (fault in the supply manifold sensor  $p_{\rm sm}$ ).

• Fault scenario 2:  $f_4 = 10\%$  (fault in the anode pressure sensor  $p_{an}$ ), starting from t = 15 s

Figs. 2–4 show the measured value (red line (in the web version)), lower interval estimation (blue line) and upper interval estimation (black line) in the first fault scenario, for the residuals corresponding to  $p_{sm}$ ,  $m_{O_2}$  and  $m_{H_2}$  that are sensitive to this fault according to the FSM presented in Table 2. Notice that while the fault is not present, the measurement is inside the interval estimation provided by the corresponding TS interval observer. However, when the fault appears (at t = 15 s), the measured values are outside the intervals estimated by the TS interval observers, leading to the observer fault signature given by  $\phi_2$ ,  $\phi_3$  and  $\phi_4$  shown in Fig. 5. By matching this observed fault signature with the FSM presented in Table 2, the fault in the supply manifold sensor  $p_{sm}$  can be diagnosed.

On the other hand, in the second fault scenario, the fault affects only the residual corresponding to  $m_{\rm H_2}$  (see Fig. 6) according to the FSM matrix presented in Table 2. As in the case

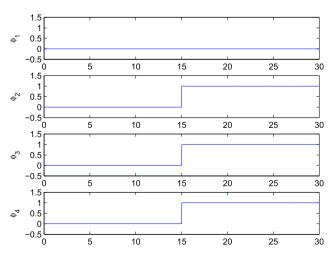


Fig. 5 – Fault signatures with  $f_2 = 30\%$  (fault in the supply manifold sensor  $p_{sm}$ ).

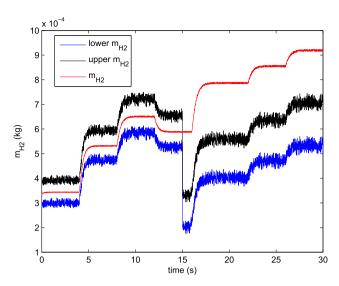


Fig. 6 –  $m_{\rm H_2}$ ,  $\underline{m}_{\rm H_2}$  and  $\overline{m}_{\rm H_2}$  with  $f_6=10\%$  (fault in the anode pressure sensor  $p_{an}$ ).

of the first fault scenario, before the fault appearance, the measurement  $m_{\rm H_2}$  is inside the interval estimation provided by the corresponding TS interval observer, as shown in Fig. 7. However, after the fault occurs, the measurement goes out of the estimation interval, allowing to detect the fault generating the observed fault signature  $\phi_4$  that, according to Table 2, allows diagnosing the fault in the anode pressure sensor  $p_{an}$ .

In order to conclude the performance analysis, the proposed approach has been compared to a fault diagnoser based on fixed thresholds. In fact, the main advantage of the proposed robust fault diagnosis using a TS interval observer approach is that it is able to generate adaptive thresholds that guarantee that no false alarms occur due to the model uncertainty. As shown in Table 9, where the minimum fault magnitudes that assure a 100% successful fault isolation rate are listed, the conservatism provided by fixed thresholds leads to worse performances than the proposed approach.

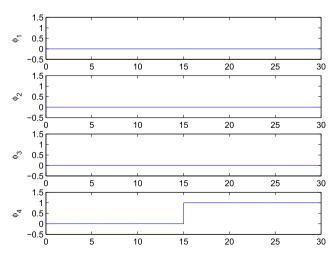


Fig. 7 – Fault signatures with  $f_6 = 10\%$  (fault in the anode pressure sensor  $p_{an}$ ).

Table 9 — Minimum fault magnitudes that assure a 100%
successful fault isolation rate.

	Adaptive thresholds	Fixed thresholds
Fault 1 $\omega_{cp}$	6.5%	9.2%
Fault 2 p <sub>sm</sub>	1.8%	3.4%
Fault 3 $v_{\rm st}$	1.4%	1.4%
Fault 4 m <sub>H2</sub>	13.4%	22%
Fault 5 p <sub>ca</sub>	11.4%	22%
Fault 6 p <sub>an</sub>	1.4%	3.3%

## **Conclusions**

In this paper, the problem of robust fault diagnosis of PEM fuel cells has been addressed by using a bank of TS interval observers that consider uncertainty in a bounded context. The design of the TS interval observers is performed using LMIs and calibrating the uncertainty in such a way that, in a nonfaulty situation, the measurements will be inside the computed intervals, thus avoiding the possibility of false alarms. Fault detection is based on checking whether all the measurements lie inside their corresponding estimated interval bounds. In the case a fault is detected, the measurements that are inconsistent with their corresponding estimations are annotated and a fault isolation procedure is triggered. By using the theoretical fault signature matrix, the fault is isolated by means of a logic reasoning that takes into account the presence of bounded uncertainty and, if the number of candidate faults is more than one, a posterior correlation analysis provides the most probable fault.

The proposed approach has been successfully tested using a well known PEM fuel case study proposed in the literature. It has been shown that as long as the fault magnitudes are above some critical value, correct fault detection and isolation is achieved. On the other hand, for each considered fault, there is a lower critical value, below which fault detection cannot be achieved, i.e. the fault is not detectable. Between these two extreme cases, the performance of the fault diagnoser increases when the fault magnitude increases.

The obtained results are satisfactory and the proposed strategy is promising. Future research includes its application to a real set-up, as well as to extend the proposed approach to other types of faults, e.g. affecting the actuators, and integrate the proposed fault diagnosis methodology with some fault tolerant control strategy in order to increase the reliability and performance of PEM fuel cells under fault occurrence.

## Acknowledgments

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## Appendix A. Proof of Theorem 1

If (32) holds, the dynamics of the interval estimation errors  $\underline{e}(k) = x(k) - \underline{x}(k)$  and  $\overline{e}(k) = \overline{x}(k) - x(k)$  are given by:

$$\underline{e}(k+1) = (A_{0i} - \underline{L}_i C)\underline{e}(k) + \sum_{j=1}^{3} \underline{w}_i^j(k)$$
(A.1)

$$\overline{e}(k+1) = \big(A_{0i} - \overline{L}_iC\big)\overline{e}(k) + \sum_{i=1}^3 \overline{w}_i^j(k) \tag{A.2} \label{eq:alpha}$$

$$\underline{w}_{i}^{1}(k) = \Delta A_{i}x(k) - \underline{\Delta A_{i}}^{+}\underline{x}^{+}(k) + \overline{\Delta A_{i}}^{+}\underline{x}^{-}(k) + \underline{\Delta A_{i}}^{-}\overline{x}^{+}(k) - \overline{\Delta A_{i}}^{-}\overline{x}^{-}(k)$$
(A.3)

$$\underline{w}_{i}^{2}(k) = \left|\underline{L_{i}}\right| V(k) E_{n_{y}} - \underline{L_{i}} v(k) \tag{A.4}$$

$$w_i^3(k) = d(k) - d(k) \tag{A.5}$$

$$\overline{w}_{i}^{1}(k) = -\Delta A_{i} x(k) + \overline{\Delta A_{i}}^{+} \overline{x}^{+}(k) - \underline{\Delta A_{i}}^{+} \overline{x}^{-}(k) \\
-\overline{\Delta A_{i}}^{-} \underline{x}^{+}(k) + \Delta A_{i}^{-} \underline{x}^{-}(k)$$
(A.6)

$$\overline{w}_{i}^{2}(\mathbf{k}) = \left| \overline{L_{i}} \right| V(\mathbf{k}) E_{n_{y}} + \overline{L_{i}} v(\mathbf{k}) \tag{A.7}$$

$$\overline{w}_i^3(k) = \overline{d}(k) - d(k) \tag{A.8}$$

Following [25], if (24) holds, then:

$$A_{0i} - L_i C, A_{0i} - \overline{L_i} \in \mathbb{R}^{n \times n}_{\perp} \tag{A.9}$$

where  $\mathbb{R}^{n\times n}_+$  denotes the set of real matrices with nonnegative elements. Hence, the dynamics for  $\underline{e}(k)$  and  $\overline{e}(k)$  is cooperative [45] and (13) holds as long as  $\underline{w}_i^j(k) \geq 0$  and  $\overline{w}_i^j(k) \geq 0$   $\forall i=1,...,N, \ \forall j=1,...,3$  and  $\forall k \geq 0$ , which is true due to the assumptions (10)–(12) [25].

In order to show that  $\underline{x}(k)$  and  $\overline{x}(k)$  remain bounded for all  $t \ge 0$ , the equations in (14) are rewritten as:

$$\underline{x_i}(k+1) = \left[A_{0i} - \underline{L_i}C + \underline{\Delta A_i}^+\right]\underline{x}(k) + \underline{f_i}(\underline{x}(k), \overline{x}(k)) + \underline{\delta_i}(k) \quad \text{(A.10)}$$

$$\overline{x_i}(k+1) = \left[A_{0i} - \overline{L_i}C + \overline{\Delta A_i}^+\right] \overline{x}(k) + \overline{f_i}(\underline{x}(k), \overline{x}(k)) + \overline{\delta_i}(k) \tag{A.11}$$

with:

$$\underline{f_{\underline{i}}}(\underline{x}(k), \overline{x}(k)) = \left(\underline{\Delta A_{\underline{i}}}^{+} - \overline{\Delta A_{\underline{i}}}^{+}\right) \underline{x}^{-}(k) \\
-\Delta A_{\underline{i}}^{-} \overline{x}^{+}(k) + \overline{\Delta A_{\underline{i}}}^{-} \overline{x}^{-}(k)$$
(A.12)

$$\overline{f}_{i}(\underline{x}(k), \overline{x}(k)) = \left(\overline{\Delta A_{i}}^{+} - \underline{\Delta A_{i}}^{+}\right) \overline{x}^{-}(k) \\
-\overline{\Delta A_{i}}^{-} \underline{x}^{+}(k) + \underline{\Delta A_{i}}^{-} \underline{x}^{-}(k)$$
(A.13)

$$\underline{\delta_i}(k) = \underline{L_i}y(k) - \left|\underline{L_i}\right|V(k)E_{n_y} + \underline{d}(k) \tag{A.14} \label{eq:alpha}$$

$$\overline{\delta_i}(k) = \overline{L_i} y(k) - \left| \overline{L_i} \right| V(k) E_{n_y} + \overline{d}(k) \tag{A.15}$$

Then, the boundedness of  $\underline{x}(k)$  and  $\overline{x}(k)$  is a consequence of the nonnegativity of  $A_{0i} - \underline{L}_i C + \underline{\Delta A_i}^+$  and  $A_{0i} - \overline{L}_i C + \overline{\Delta A_i}^+$ , the boundedness of the inputs  $\delta_i(k)$  and  $\overline{\delta_i}(k)$  and the property of

the functions  $\underline{f_i}$  and  $\overline{f_i}$  of being globally Lipschitz [25], and is proved by introducing the system:

$$\xi(k+1) = G_i\xi(k) + \phi_i(\xi(k)) + \delta_i(k) \tag{A.16}$$

where

$$\xi(\mathbf{k}) = \left(\frac{\mathbf{X}(\mathbf{k})}{\overline{\mathbf{X}}(\mathbf{k})}\right) \phi_i(\xi(\mathbf{k})) = \left(\frac{f_i}{f_i}(\xi(\mathbf{k}))\right) \delta_i(\mathbf{k}) = \left(\frac{\underline{\delta}_i(\mathbf{k})}{\overline{\delta}_i(\mathbf{k})}\right) \tag{A.17}$$

$$G_{i} = D_{i} - \begin{pmatrix} \underline{L}_{i} & 0\\ 0 & \overline{L}_{i} \end{pmatrix} \Upsilon$$
 (A.18)

and:

$$|\phi_i(\xi(k))| \le \eta |\xi(k)| \tag{A.19}$$

with  $\eta$  defined as in (28). In fact, using Schur complement [46], it can be shown that if (23) holds, the following is true for the increment  $\Delta \mathcal{V}(k)$  of the Lyapunov function  $\mathcal{V}(k) = \xi(k)^T P \xi(k)$  [25]:

$$\Delta \mathcal{V}(\mathbf{k}) < -\xi(\mathbf{k})^{\mathrm{T}} Q \xi(\mathbf{k}) + \left(1 + \varepsilon_{1}^{-1} + \varepsilon_{2}^{-1}\right) \delta_{i}(\mathbf{k})^{\mathrm{T}} P \delta_{i}(\mathbf{k}) \tag{A.20}$$

that proves the boundedness of  $\underline{x}(k)$  and  $\overline{x}(k)$  [47].

Finally, it must be shown that the closed-loop poles of the TS interval observer are in  $\mathscr{D}$ , defined as in Ref. [18]. This is straightforward, by taking into account that the closed-loop matrix of the TS interval observer is given by:

$$A_{cl,i} = \begin{pmatrix} A_{0,i} & 0 \\ 0 & A_{0,i} \end{pmatrix} - \begin{pmatrix} \underline{L}_i & 0 \\ 0 & \overline{L}_i \end{pmatrix} \Upsilon \tag{A.21}$$

such that [25] is obtained from the direct application of Theorem 2.2 in Ref. [34] to  $A_{clj}^T$ . This completes the proof.

#### Appendix B. Decoupled sub-models

For illustrative purpose, the first sub-model obtained by applying structural analysis [30] to the linear-like model of the PEMFC [1] is completely detailed in the following. For the remaining three sub-models, only their structure is provided.

#### Compressor

Starting from the non-linear equation for the compressor provided in Ref. [14], and by approximating intermediate variables and terms as functions of the stack current  $I_{st}$ , the following sub-model for the compressor can be obtained:

$$\dot{\omega}_{cp} = \widetilde{a}_{11}\omega_{cp} + \widetilde{c}t_1$$
 (B.1)

with:

$$\widetilde{a}_{11} = -\frac{\widetilde{z}_3 v_{cp}}{J_{cp} R_{cm}} k_v \tag{B.2}$$

$$\widetilde{c}t_{1} = \frac{\widetilde{z}_{3}}{J_{cp}} \left( \frac{v_{cp}^{2}}{R_{cm}} - \frac{C_{p}T_{amb}}{\eta_{cm}\widetilde{\eta}_{cp}} \widetilde{z}_{1}\widetilde{z}_{2} \right)$$
(B.3)

$$v_{cp} = 0.672519I_{st} + 33.554115$$
 (B.4)

$$\widetilde{\eta}_{cp} \approx 0.777217$$
 (B.5)

$$\tilde{z}_1 \approx 0.275641 \cdot 10^{-3} I_{st} - 0.340993 \cdot 10^{-3}$$
 (B.6)

$$\tilde{z}_2 \approx 0.001375I_{st} - 0.023710$$
 (B.7)

 $\tilde{z}_3 \approx -0.000426 \cdot 10^{-3} I_{st} + 0.213459 \cdot 10^{-3}$  (B.8) where  $k_v = 0.0153$  V/(rad/s) is the motor electric constant,  $J_{cp} = 5 \times 10^{-5} \text{ kg} \cdot \text{m}^2$  is the compressor and motor inertia,  $R_{cm} = 0.816 \ \Omega$  is the compressor motor circuit resistance,  $C_p = 1004 \text{ J/(kg} \cdot \text{K)}$  is the air heat capacity at constant pressure,  $T_{amb} = 298 \ \text{K}$  is the ambient temperature and  $\eta_{cm} = 0.9$  is the compressor efficiency.

## Supply manifold

The supply manifold sub-model is given by the following expression:

$$\dot{p}_{sm} = \tilde{a}_{22}p_{sm} + \tilde{b}_{25}p_{ca} + \tilde{c}t_2 \tag{B.9}$$

## Cathode plus return manifold

The cathode plus manifold sub-model is a second order state space model, where the pressure at the return manifold  $p_{rm}$  is a state variable that is not measured. It can be expressed as:

$$\begin{bmatrix}
\dot{m}_{O_{2}} \\
\dot{p}_{rm}
\end{bmatrix} = \begin{bmatrix}
\tilde{a}_{77} & \tilde{a}_{78} \\
0 & \tilde{a}_{88}
\end{bmatrix} \begin{bmatrix}
m_{O_{2}} \\
p_{rm}
\end{bmatrix} + \begin{bmatrix}
\tilde{b}_{75} \\
b_{85}
\end{bmatrix} p_{ca} 
+ \begin{bmatrix}
b_{70} \\
0
\end{bmatrix} p_{sm} + \begin{bmatrix}
b_{70} \\
0
\end{bmatrix} I_{st} + \begin{bmatrix}
\tilde{c}t_{7} \\
0
\end{bmatrix} \tag{B.10}$$

#### Anode

Finally, the anode sub-model is expressed as follows:

$$\dot{m}_{\rm H_2} = \tilde{b}_{32} p_{\rm sm} + \tilde{b}_{34} p_{an} + b_{30} I_{\rm st} + \tilde{c} t_3$$
 (B.11)

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