HOMI BHABHA CENTRE FOR SCIENCE EDUCATION

Tata Institute of Fundamental Research

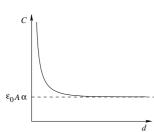
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Indian National Physics Olympiad - 2013 Solutions

PLEASE NOTE THAT ALTERNATE/EQUIVALENT SOLUTIONS MAY EXIST. Brief solutions are given below.

1. (a)
$$C = \frac{\epsilon_0 A \alpha}{1 - e^{-\alpha d}}$$





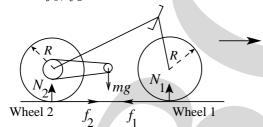
(c) Charge =
$$\frac{\epsilon_0 A \alpha V}{1 - e^{-\alpha d}}$$

(d)
$$\vec{E}(x) = \frac{\alpha V e^{-\alpha x}}{1 - e^{-\alpha d}} \hat{x}$$

2. (a)
$$\phi = \sin^{-1}\left(\frac{nh}{d\sqrt{2m_0K}}\right)$$

(b)
$$d \approx 2.4 \text{ Å}$$

3. (a) Here f_1 , f_2 are frictional forces and N_1 , N_2 are normal reactions.



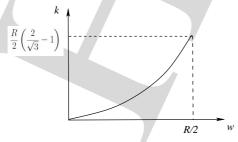
(b)
$$a = \frac{\tau}{MR^2 + 2I}R$$

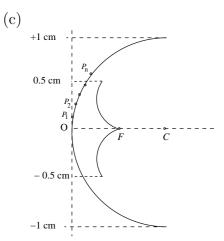
(c)
$$a \le \frac{\mu g/2}{\left(1 - \frac{\mu}{4}\right)}$$

(d)
$$a_m = 2g/3$$

4. (a)
$$k = \frac{R}{2} \left[\frac{R}{(R^2 - \omega^2)^{1/2}} - 1 \right]$$

(b)





5. (a) Torque =
$$\frac{2\mu_0 I^2(a^2 + b^2)abd\sin\phi}{\pi \left[(a^2 - b^2)^2 + 4a^2b^2\sin^2\phi \right]} \hat{z}$$

(b)
$$\phi = \alpha$$
; $-\alpha$; $\pi - \alpha$; $\pi + \alpha$
where $\alpha = \sin^{-1}[(a^2 - b^2)/2ab]$

6. (a)
$$g(r, \theta, \dot{r}, \dot{\theta}) = \frac{F(r)}{m} + r\dot{\theta}^2$$

(b) From null azimuthal component

$$2\dot{r}\dot{\theta} + r\ddot{\theta} = 0$$

$$\frac{d}{dt}(r^2\dot{\theta}) = 0$$

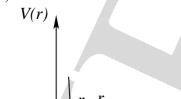
$$\frac{d}{dt}(mr^2\dot{\theta}) = 0$$

$$\frac{d}{dt}L = 0$$

 $\Rightarrow L = \text{constant}$

(c)
$$E = \frac{m(\dot{r}^2 + r^2\dot{\theta}^2)}{2} - \frac{GMm}{r}$$

(d)
$$V(r) = \frac{L^2}{2mr^2} - \frac{GMm}{r}$$
 (e)



 $V_0 \mid - V_0 \mid$ where $r_0 = \frac{L^2}{GMm^2}$, $r_c = \frac{L^2}{2GMm^2}$ and $V_0 = -\frac{G^2M^2m^3}{2L^2}$

(f) For E>0 orbit is hyperbola. For E<0 orbit is ellipse. At $r=r_o$ orbit is circular.

$$(g) r_0 = \frac{L^2}{GMm^2}$$

(h)
$$T_r = 2\pi \sqrt{\frac{r_0^3}{GM}}$$

(i)
$$n > -3$$

7. (a)
$$\Gamma_a = \frac{(1 - \gamma_a)m_a g}{\gamma_a R}$$
 or $-\frac{m_a g}{C_a}$ where $\gamma_a = 7/5$ (ratio of specific heats at constant pressure and volume) and C_a is

molar specific heat at constant pressure for air. $|\Gamma_a| \approx 0.01^{\circ} \mathrm{K \cdot m^{-1}}$

(b)
$$\Gamma_b = -\frac{m_a g T_b}{C_b T_a}$$

(c)
$$\ddot{z} = g \left(\frac{m_a T_b}{T_a m_b} - 1 \right)$$

(d)
$$z_0 = \frac{T_0}{\Gamma_a} \left[1 - \left(\frac{m_b}{m_a}\right)^{1/(\eta - 1)} \right]$$

where $\eta = C_a/C_b$.

(e) Condition:
$$C_a > C_b$$

$$\sqrt{m_*(n-1)/m_*} \sqrt{1/(\eta_*)}$$

$$\omega = g \sqrt{\frac{m_a(\eta - 1)}{C_a T_0} \left(\frac{m_a}{m_b}\right)^{1/(\eta - 1)}}$$

(f)
$$\tau \approx 95 \text{ s}$$