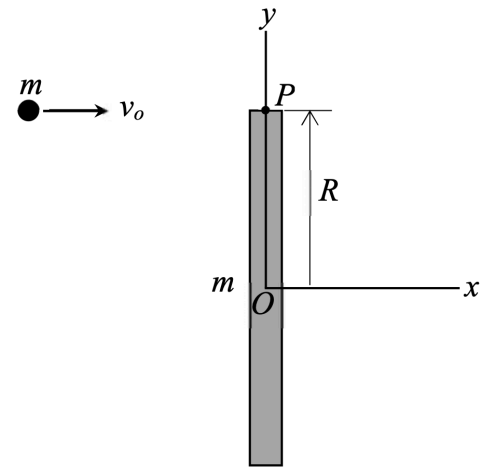
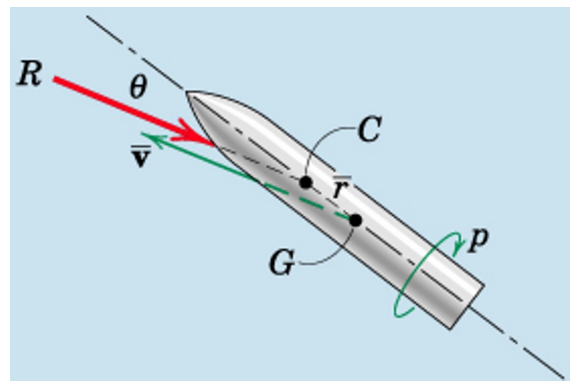


ESO209A: Dynamics: Tutorial 12
(Week: 27 Oct. - 2 Nov. Based on L15-L20)

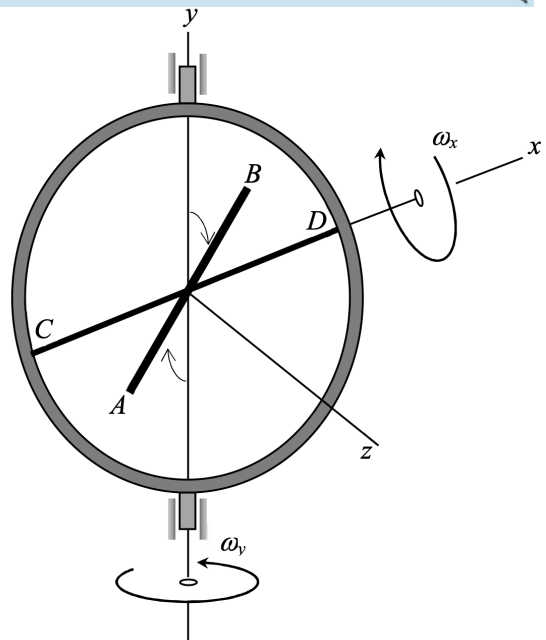
- (1) A thin uniform circular disc-type satellite of mass m and radius R is spinning at a constant angular rate of ω_0 in space about its body-fixed x -axis, as shown. A space debris of equal mass but of a point size hits the satellite normal to its plane with the velocity v_0 as shown. Subsequently, the debris gets embedded into it at point P . Find the angular velocity vector immediately after the collision. What impact model is being used here? The figure shows the side view of the satellite.



- (2) The velocity vector \mathbf{v} of a rocket, moving through the earth's atmosphere, makes an angle θ with respect to its symmetry axis as shown. The net aerodynamic drag acts parallel to \mathbf{v} but in opposite direction such that its line of action cuts the symmetry axis at C , slightly away from G , as shown. The moment of inertia of the rocket about its symmetry axis is I and that through G and perpendicular to the plane of the paper is I_0 . How much spin p about the symmetry axis should be provided to stabilize the rocket such that $\dot{\theta} = 0$?



- (3) A thin uniform rod AB of mass m and length l is rotating about the shaft CD at a constant angular velocity ω_x as shown. Note that the rod AB is normal to the shaft CD . The shaft CD is attached to a circular gimbal which rotates about the fixed y -axis at a constant angular rate of ω_y , as shown. Find the components of the moments acting on the rod AB .



- (4) The sketch shows the velocity of a small disk of mass m at the instant before its collision with a rigid body consisting of the same disks connected by a rigid massless bar. Determine the rebound velocity of each disk in terms of the coefficient of restitution ϵ .

