

Homework 1

A spacecraft is modeled as a rigid body B + damper \mathcal{D} + 3 wheels $\mathcal{W}_1, \mathcal{W}_2, \mathcal{W}_3$

Each wheel is a flat disk with mass $m_{W1} = m_{W2} = m_{W3} = 5 \text{ kg}$ and radius $R = 1 \text{ m}$

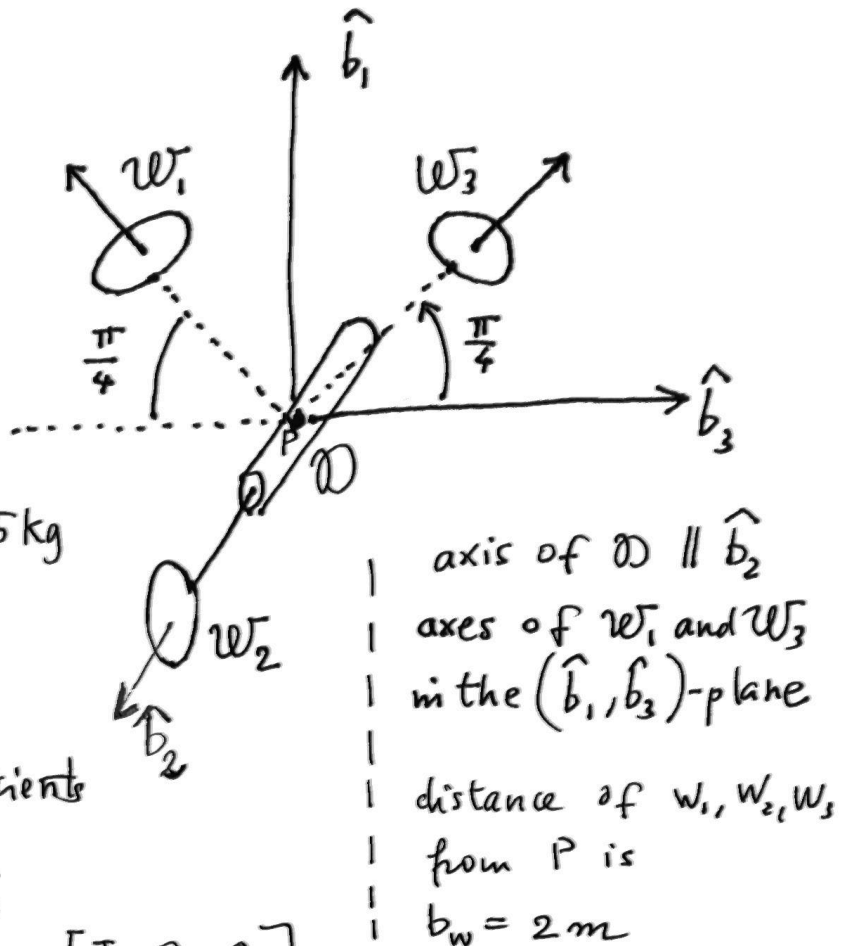
\mathcal{D} has rest point at P and mass $m_D = 10 \text{ kg}$ and coefficients

$$K_d = 5.5 \frac{\text{kg}}{\text{sec}^2} \quad C_d = 30 \frac{\text{kg}}{\text{sec}}$$

B has inertia matrix and mass $m_B = 100 \text{ kg}$

$$J_{B,P}^{(B)} = \begin{bmatrix} J_1 & 0 & 0 \\ 0 & J_2 & 0 \\ 0 & 0 & J_3 \end{bmatrix}$$

$$J_1 = 350 \text{ kg m}^2 \quad J_2 = 300 \text{ kg m}^2 \quad J_3 = 400 \text{ kg m}^2$$



The 3 wheels are either

- rotating with constant angular velocity ω_{si} (i.e. the initial value)
- subject to viscous damping, with $K_w = 0.1 \frac{\text{kg m}^2}{\text{sec}}$

2 cases are considered, with propagation time of 30 minutes

- ω_1 and ω_3 set to (a), ω_2 set to (b)
- $\omega_1, \omega_2, \omega_3$ set to (b)

Initial conditions for (1) and (2) are

$$\underline{\omega}^T(0) = [36 \quad 3 \quad 3]^T \frac{\text{deg}}{\text{sec}} \quad \omega_{s1}(0) = \omega_{s3}(0) = 720 \frac{\text{deg}}{\text{sec}}; \quad \omega_{s2}(0) = 60 \frac{\text{deg}}{\text{sec}}$$

$$\xi(0) = 0 \quad \dot{\xi}(0) = 0 \quad \underline{v}_P^T(0) = [0 \quad 0 \quad 0]^T$$

Portray the time histories (in $[0, 30] \text{ min}$) of

- components of $\underline{\omega}(t)$
- $\omega_{si}(t), i=1,2,3$
- $\xi(t)$ and $\dot{\xi}(t)$
- Mechanical energy for cases (1) and (2) and comment on them