Linear State Feedback Control - Inverted Pendulum

Group 15: Leonardo Pedroso (89691), Miguel Graça (90142), Tomás Bessa (90200), Inês Ferreira (90395)
Professor Pedro Batista and Professor João Miranda Lemos

Abstract — The aim of this project is to design a control solution for an inverted pendulum, which is a nonlinear system. It is modeled by a linear time invariant (LTI) system about the equilibrium point corresponding to the upright position. The approach followed consists of the design of a state feedback controller and a state observer to maintain the pendulum in the upright position and track piece-wise constant signals with the angle of the horizontal bar. The proposed control solution is validated via extensive simulations including nonlinear effects and other sources of nonideal dynamics.

I. INTRODUCTION

The project report is organized as follows. In Section II the plant is characterized and analyzed in terms of of its: i) dynamic modes, their origin, and effects on the control solution design; ii) observability; iii) controllability; and iv) frequency response. In Section III, the regulation problem is addressed assuming full-state feedback is available. In Section IV, a state observer is designed to provide state estimates to the previously designed controller, which are then cascaded making use of the Separation Theorem. In Section V, nonlinear effects are studied and their effects are taken into account in the design of both the controller and the observer. In Section VI, the control solution is extended to the tracking problem. Finally, Section VII presents the main conclusions of this project.

II. PLANT CHARACTERIZATION

The plant consists of an inverted pendulum. The pendulum is attached at the base to a horizontal bar that can be rotated in the horizontal plane by a DC motor. There are also two sensors that measure the angles of both bars with respect to a reference direction. The plant is depicted in Fig. 1.



Fig. 1: QUANSER rotary inverted pendulum [2].

The control strategy synthesis approach followed in this project is for systems described by linear models. Although the plant being studied is nonlinear, its dynamics can be linearized about an equilibrium point, which in this case is the upright position of the pendulum, which describes the behavior of the plant well for small deviations in relation to the equilibrium. The linearized model is, thus, given by

$$\begin{cases} \dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}u \\ \mathbf{y} = \mathbf{C}\mathbf{x} \end{cases},$$

where $\mathbf{x} := [\alpha \ \dot{\alpha} \ \beta \ \dot{\beta} \ i]^T$, α is the angle of the horizontal bar, β is the angle of the pendulum in relation to the upright position, i is the current in the DC motor, $u \in \mathbb{R}$ is the motor input signal, and \mathbf{A}, \mathbf{B} , and \mathbf{C} are matrices or appropriate dimensions.

Before proceeding with the controller implementation, it is important to characterize the system in terms of its: i) dynamic modes, their origin, and effects on the controller design; ii) observability; iii) controllability; and iv) frequency response. These aspects are addressed in this section, making use of the LTI system that is obtained linearizing the nonlinear dynamics of the plant about the upright position of the pendulum.

A. Dynamic modes

First, the dynamic modes of the open-loop system are analyzed, from which conclusions can be made regarding its stability. It is well known that the poles of the system are given by the eigenvalues of matrix **A**. These are presented in Table I. It is now of interest to analyze each mode and comment on its origin, and aspects to consider in the controller design.

TABLE I: Poles of open-loop system.

Pole	Damping	ω (rad/s)	Time constant (s)
0.00e+00	-1.00e+00	0.00e+00	$+\infty$
-7.37e+02	1.00e+00	7.37e+02	1.36e-03
-1.94e+01	1.00e+00	1.94e+01	5.16e-02
-5.76e+00	1.00e+00	5.76e+00	1.74e-01
7.00e+00	-1.00e+00	7.00e+00	_

First, the fastest pole (p=-737 rad/s) is a real pole associated with the dynamics of the DC motor. Thus, the motor dynamics are modeled as a first order low pass system with a characteristic time of 1.36 ms. Given that this mode is very stable and fast, when compared with the pendulum dynamics, it does not require much attention in the controller or observer design stages. Second, the second fastest pole (p=-19.4 rad/s) is associated with the relation between the current in the DC motor and the angular velocity of the horizontal bar. Its effect is noticeable, but it is still almost one order of magnitude faster than the pendulum dynamics. Third, there is a pole at the origin, which corresponds to integrator dynamics. It is associated with

the angle α of the horizontal bar, whose dynamics are an integrator of the angular velocity of the bar. As it is discussed in Sections V and VI, the existence of this pole on the open-loop dynamics of the plant allows for several beneficial properties such as rejection of some disturbances and robustness to model parameter identification uncertainty. Finally, the pair of poles (p=-5.76 rad/s and p=7.00 rad/s) are associated with the inverted pendulum dynamics. It is possible to prove this association upon writing the differential equation that models an inverted pendulum with friction for small angles β in relation to the equilibrium position

$$I\ddot{\beta} = mgl_{cm}\beta - \gamma_f \dot{\beta} \,, \tag{1}$$

where I is the moment of inertia about the pivot, m is the mass, l_{cm} is the distance from the pivot to the center of mass, and γ_f is a coefficient of friction. The poles of the second order system described by (1), obtained for reasonable values of the parameters and low coefficient of friction are two real poles, one stable and another unstable, whose absolute value is similar, which is consistent with the poles obtained. From an intuitive perspective, the existence of an unstable pole is expected as the inverted pendulum is naturally unstable in the upright position. One of the goals of the control strategy devised in this project is, thus, to stabilize the dynamics of the inverted pendulum.

B. Controllability

It is well known that a continuous-time system is completely controllable iff the controllability matrix

$$\mathcal{C} := \begin{bmatrix} \mathbf{B} & \mathbf{A}\mathbf{B} & \dots & \mathbf{A}^{n-1}\mathbf{B} \end{bmatrix}$$
,

where n is the order of the system, is full rank. After computing the controllability matrix and its rank, it is confirmed that it is full rank, thus the system is completely controllable. This result is, by no means, obvious, since there is only one input and the state has five components, two of which are two mechanical degrees of freedom of the physical system. It is well known that this result guarantees that there is a state feedback gain such that the closed-loop poles are wherever one desires. In Section II-D, with the analysis of the frequency response of the system, an intuitive explanation on controllability is also provided.

C. Observability

It is well known that a continuous-time system is completely observable iff the observability matrix

$$\mathcal{O} := \left[egin{array}{c} \mathbf{C} \ dots \ \mathbf{C} \mathbf{A}^{n-1} \end{array}
ight] \, ,$$

where n is the order of the system, is full rank. This test is applied to two distinct cases: i) the output is only the pendulum angle β , *i.e.*

$$\mathbf{C} := \begin{bmatrix} 0 & 0 & 1 & 0 & 0 \end{bmatrix} ;$$

and ii) the output is both angles of the bars, α and β , *i.e.*,

$$\mathbf{C} := \begin{bmatrix} 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 \end{bmatrix} . \tag{2}$$

After computing the observability matrix for both these cases, in the first case, it is not full rank, but in the second case it is.

Therefore, the system is not completely observable if the output is only the pendulum angle β . It is necessary to measure the angle of the horizontal bar α as well to infer the evolution of the state variables. In Section II-D, with the analysis of the frequency response of the system, the observability of the system is also discussed, and the reason for the lack of complete observability in the first case is explained.

D. Frequency Response

Finally, to analyze the frequency response of the system, the transfer functions (t.f.) of α and β in relation to the input u are obtained, which are given by

$$\frac{\alpha(s)}{u(s)} = \frac{1.92 \text{e}4s^2 + 92.1s - 8.26 \text{e}5}{s(s^4 + 755s^3 + 1.33 \text{e}4s^2 - 4.82 \text{e}4s - 5.76 \text{e}5)} \;, \ \, (3)$$

$$\frac{\beta(s)}{\alpha(s)} = \frac{1.92e4s^2}{1.92e4s^2 + 92.1s - 8.26e5},$$
 (4)

thus,

$$\frac{\beta(s)}{u(s)} = \frac{1.92e4s^2}{s(s^4 + 755s^3 + 1.33e4s^2 - 4.82e4s - 5.76e5)} \ . \tag{5}$$

It is convenient to analyze these t.f. together with the corresponding bode plots and pole-zero maps, which are depicted, respectively, in Figs. 2 and 3. Note that in Fig. 3 the fastest pole, corresponding to the DC motor dynamics, is omitted, and also that the poles of the t.f. (3) and (5) are superimposed.

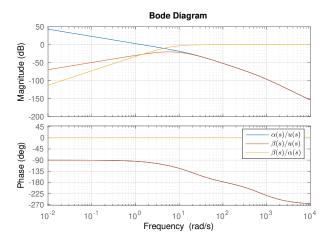


Fig. 2: Bode diagrams of the t.f. (3),(4), and (5).

On one hand, consider the t.f. $\alpha(s)/u(s)$. It is noticeable that the gain tends to infinity as $\omega \to 0$, which is consistent with the presence of an integrator pole on the dynamics of α as put forward in Section II-A. Also, note on Fig. 3 that the two zeros of the t.f. are very close to the pair of real poles associated with the dynamics of the inverted pendulum, given that α is not very affected by the dynamics of the inverted pendulum, but rather on the integrator and pole that models the angular velocity of the DC motor shaft. This analysis is consistent with the claim made on Section II-A about the correspondence between the poles and the dynamics of the physical system. For these reasons, the effect of the inverted pendulum poles is barely visible in the bode plot of the gain. The gain drops asymptotically at -40 dB/dec after p = -19.3 rad/s and at -60 dB/dec after the pole of the DC motor dynamics, attenuating high input

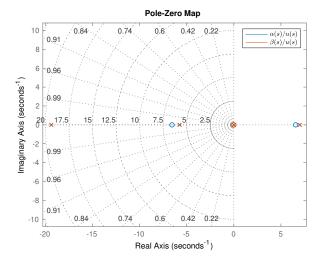


Fig. 3: Pole-zero maps of the t.f. (3) and (5).

frequencies. Thus, the angle α can only be controlled effectively from the input u on low frequencies limited by the frequency of the pole that models the DC motor shaft angular speed, which makes sense in a physical intuitive perspective. Furthermore, as seen in Section VI the existence of a nonminimum phase zero leads to the manifestation of a whiplash effect on the time response of the angle α .

On the other hand, consider the t.f. $\beta(s)/u(s)$ $(\beta(s)/\alpha(s))(\alpha(s)/u(s))$. Note that it has two zeros at the origin, one of which cancels with the integrator pole associated with the dynamics of α , which is consistent with the asymptotic evolution of +20 dB/dec at low frequencies. After the frequency of the inverted pendulum poles and the pole that models the DC motor shaft angular speed, the gain drops asymptotically at -40dB/dec and at -60dB/dec after the pole of the DC motor dynamics, attenuating high input frequencies. It is interesting to point out that, since $\alpha(s)/u(s)$ has low-pass dynamics and $\beta(s)/\alpha(s)$ has high-pass dynamics, the t.f. $\beta(s)/u(s)$ has band-pass dynamics. Thus, the angle β can only be controlled effectively from the input u on a limited band of frequencies, centered, approximately, at the characteristic frequency of the inverted pendulum, which makes sense in a physical intuitive perspective.

It is possible to interpret the frequency response of the system to comment on the controllability and observability of the system. First, as it was shown in Section II-B, the plant is controllable, despite the fact that the state has five components, two of which are two mechanical degrees of freedom of the physical system. The way the controllability of the system is achieved lies on its frequency response. First, note, for convenience, that for continuous-time systems, controllability is equivalent to the ability to drive the system from any initial position to the origin in finite time [5, Definition 9.1]. In fact, for the same signal u, different frequencies have different effects on α and β . A component with low frequency only affects α and with a frequency in the aforementioned band affects both α and β . Thus, it is possible to drive both α and β , and their derivatives, to zero in finite time. This state can be maintained with u=0 and this equilibrium i can, thus, be driven to the origin.

Second, it is pointed out in Section II-C that if the output of the system is only the angle β , then the system is not observable. Analyzing the t.f. (5) there is a pole-zero cancellation at the origin. It is possible to conclude that there is a loss either of controllability, which is already known that it is not the case, or observability. Therefore, in this case, the system is not observable. It is easily seen in this case from (4) that the evolution of β does not depend on α or its first derivative, only on the second derivative. It is, thus, evident that there is a loss of information of the evolution of α measuring only β , which is the reason behind the loss of observability. For instance, the measurement of β cannot distinguish if α is stationary or moving with constant velocity, or even if α is constant and equal to α_0 or α is constant and equal to $\alpha_1 \neq \alpha_0$.

III. STATE FEEDBACK CONTROLLER

The control strategy synthesis approach followed in this project follows two steps: i) the state feedback controller synthesis; and ii) the state observer synthesis. It is well known that due to the Separation Theorem [5, Chapter 15], one can design the state feedback controller assuming that full state feedback is available. In fact, the poles of the closed-loop estimated state feedback system are the reunion of the closed loop poles of the controller and the closed loop poles of the observer. This property is exemplified in Section IV.

In Section II-B, it was concluded that the LTI system that describes the dynamics of the plant about the upright position is completely controllable. Thus, one can find a gain ${\bf K}$ of the state feedback control action

$$u = -\mathbf{K}\mathbf{x}$$
,

such that the closed-loop system poles are wherever one desires. Nevertheless, the approach followed in this project is to select the gain that minimizes the performance index

$$J := \int_0^\infty (\mathbf{x}^T \mathbf{Q} \mathbf{x} + u^T R u) dt, \qquad (6)$$

where $\mathbf{Q} \succeq \mathbf{0}$ and $R \succ 0$ are matrices of appropriate dimensions that weigh, respectively, the state and the input of the system. It is well-known that the computation of the optimal gain is equivalent to solving an Algebraic Ricatti Equation (ARE), which was carried out numerically with appropriate software. This design method is guaranteed to stabilize the closed-loop dynamics [1, Section 3.2] of the LTI model for any choice of $\mathbf{Q} \succeq \mathbf{0}$ and $R \succ 0$.

It is necessary to properly select the weighting matrices. First, note that the result of the minimization of the performance index (6) remains unchanged if both weighting matrices are multiplied by the same scalar. Thus, without loss of generality, the input weighting matrix is selected to be given by R=1. Second, it was pointed out in Section II-A that the only modes that are not stable are associated with the angle of the horizontal bar, *i.e.*, the dynamics of α , and the inverted pendulum dynamics, *i.e.*, the dynamics of β . Thus, it was decided to weigh only the state variables α and β . The regulation of α and β ensures the regulation of their derivatives, so the states $\dot{\alpha}$ and $\dot{\beta}$ are not weighted. Third, no coupled weights between states were considered, so \mathbf{Q} is diagonal. Also, the diagonal entries were selected such that the performance index J is dimensionless.

The matrix \mathbf{Q} is of the form

$$\mathbf{Q} \!=\! \gamma \, \operatorname{diag} \!\! \left(1 / \! \left(k \frac{10 \deg}{180 \deg} \pi \right)^{\! 2} \!, 0, 1 / \! \left(\frac{10 \deg}{180 \deg} \pi \right)^{\! 2} \!, 0, 0 \right),$$

where γ is a scaling factor and k is a weighting factor between β and α .

Before proceeding with the selection of γ and k, it is important to point out a few characteristics of the approximations and assumptions made up to this point, that have a significant impact on the controller synthesis guidelines. First, it is important to stress that the model used so far for the gain synthesis is only valid for a neighborhood around the upright position of the pendulum. Second, the model parameters are nominal, but there is uncertainty in the identification of the physical and electric parameters of the plant. Third, the LTI model does not take into account the several nonlinear and non-ideal effects that are verified in the physical plant, which are explored extensively in Section V. For these reasons, while it is possible to design, with the LTI model, an infinitely fast controller, which relies excessively on the model dynamics, it will not stabilize the real plant dynamics. It is, thus, of paramount importance to design a robust controller, i.e, it is necessary to find a compromise between regulation speed and actuation smoothness, which contributes to the robustness of the controller.

Having these guidelines in mind, the next step is to select γ and k. For that purpose, a simulation of T=10 s, with $\mathbf{x}_0=[50\deg\ 0\ 10\deg\ 0\ 0]^T$, an impulsive disturbance on the plant input with an amplitude of $A_p=4\mathrm{V}$ and duration of $\Delta t_p=0.2$ s at t=5 s, is considered. The characteristics of the simulations were devised to be representative of the normal operation of the plant. Fig. 4 depicts the evolution of α and β for different values of k, with $\gamma=10$. First, as expected, the greater k is, the lesser the weight given to α and it takes longer for α to converge to zero. Second, while for k=0.1 and k=1, there is significant overshoot of β , that does not happen for k=10 or k=100. For these reasons, the value that leads to the fastest convergence of α and that does not lead to significant overshoot of β is selected, i.e., k=10.

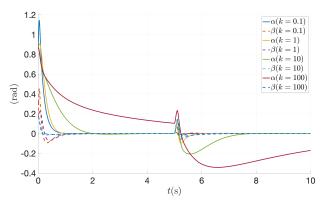


Fig. 4: Time response of α and β for different values of k, with $\gamma=10$.

Figs. 5 and 6 depict the evolution of α and β and the plant input, respectively, for different values of γ , with k=10. First, note that, as expected, the greater γ is, the greater weight is given to the state variables in relation to the weight of the

input, thus it takes longer to drive the state to zero. Conversely, if γ is lowered, the control action becomes more aggressive and the regulation is achieved faster. Second, in this particular simulation, the actuation input is greater, in absolute value, than 5V, which is as put forward in Section V the saturation of the actuation, for $\gamma=100$ and $\gamma=1000$. For these reasons, the value that leads to the fastest regulation and that does not lead to actuation input above 5V is selected, *i.e.*, $\gamma=10$.

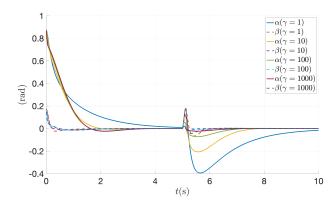


Fig. 5: Time response of α and β for different values of γ , with k=10.

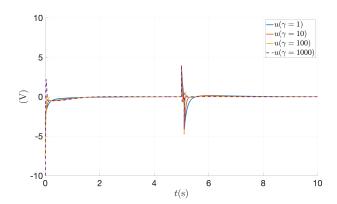


Fig. 6: Plant input for different values of γ , with k=10.

Table II depicts the poles of the closed loop system with full state feedback, with the gain synthesized with k=10 and $\gamma=10$. First, as guaranteed by this synthesis technique, the system became stable in closed loop, since all the poles are on the left half complex plane (LHCP). Second, the pole associated with the DC motor dynamics remains almost unchanged. Nevertheless, the remaining four poles were mapped, due to the state feedback, to two pairs of complex conjugate poles, with similar damping coefficients, the slowest of which has a time constant of 0.6s.

TABLE II: Poles of closed-loop system with full state feedback.

Pole	Damping	ω (rad/s)	Time constant (s)
-7.37e+02	1.00e+00	7.37e+02	1.36e-03
-1.89e + 01 + 1.15e + 01i	8.55e-01	2.22e+01	5.28e-02
-1.89e + 01 - 1.15e + 01i	8.55e-01	2.22e+01	5.28e-02
-1.68e + 00 + 1.09e + 00i	8.38e-01	2.00e+00	5.96e-01
-1.68e + 00 - 1.09e + 00i	8.38e-01	2.00e+00	5.96e-01

IV. STATE OBSERVER

In this section, the second and final step of the control strategy synthesis approach followed in this project is addressed, synthesizing an optimal state observer. Such optimal state observer is also known as the Kalman filter. The optimal observer and controller synthesis are dual problems, so, in this project, they are treated in a very similar manner. Consider the LTI model of the plant with zero-mean Gaussian process and sensor noise

$$\begin{cases} \dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}u + \mathbf{w} \\ \mathbf{y} = \mathbf{C}\mathbf{x} + \mathbf{v} \end{cases},$$

where $\mathbf{w} \sim \mathcal{N}(\mathbf{0}, \mathbf{Q_e})$ is the process noise, and $\mathbf{v} \sim \mathcal{N}(\mathbf{0}, \mathbf{R_e})$ is the sensor noise. Also, consider a continuous-time Luenberger observer

$$\dot{\hat{\mathbf{x}}} = \mathbf{A}\hat{\mathbf{x}} + \mathbf{B}u + \mathbf{L}(\mathbf{y} - \mathbf{C}\hat{\mathbf{x}}), \tag{7}$$

where L is the observer gain, and C is given by (2), such that the system is completely observable. The Kalman gain is computed such that the trace of the covariance of the estimation error is minimum. Analogously to the controller synthesis problem, it is guaranteed that the Kalman gain exists, since the LTI system at hand is observable.

It is necessary to properly select the covariance matrices to synthesize the observer gain. Even thought the matrices $\mathbf{Q_e}$ and $\mathbf{R_e}$ have a physical meaning, given that it is very difficult to estimate the process covariance in practice, they are often selected empirically, which is the approach followed in this project. First, analogously to the controller synthesis problem, the gain that achieves the minimum trace of the estimation error covariance matrix remains unchanged if both $\mathbf{Q_e}$ and $\mathbf{R_e}$ are multiplied by the same scalar. Thus, without loss of generality, the $\mathbf{R_e}$ is selected to be

$$\mathbf{R_e} = \left(\frac{1\deg}{180\deg}\pi\right)^2 \mathbf{I} \,,$$

since it is assumed that both angle sensors have similar additive noise. Second, process noise couplings between state variables is neglected for the observer's synthesis, which, although it is not a good approximation in this case, is a common practice for designing state observers. The process noise covariance matrix is, therefore, chosen to be a diagonal matrix. Third, the entries of matrix $\mathbf{Q_e}$ are selected such that the dimensions and orders of magnitude are consistent. As a result, $\mathbf{Q_e}$ was chosen to be of the form

$$\begin{aligned} \mathbf{Q_e} = \phi \, \mathrm{diag} & \Big(\frac{1 \, \mathrm{deg}}{180 \, \mathrm{deg}} \pi \Big)^2 \!\!\!, \left(\omega_d^* \frac{1 \, \mathrm{deg}}{180 \, \mathrm{deg}} \pi \right)^2 \!\!\!, \\ & \left(\frac{1 \, \mathrm{deg}}{180 \, \mathrm{deg}} \pi \right)^2 \!\!\!, \left(\omega_d^* \frac{1 \, \mathrm{deg}}{180 \, \mathrm{deg}} \pi \right)^2 \!\!\!, (0.025 \mathrm{A})^2 \!\!\!), \end{aligned}$$

where ϕ is a scaling factor, and ω_d^* is a characteristic frequency of the time response of α and β . Fourth, given that the dynamics of both α and β are governed by both pairs of complex conjugate closed loop poles presented in Table II, the characteristic frequency ω_d^* was chosen to be the damped frequency of the fastest pair of complex conjugate poles. It is, thus, given by $\omega_d^* = \omega_n \sqrt{1-\xi^2} \approx 10$ rad/s, where ω_n is the natural frequency of the poles and ξ their damping coefficient. Fifth, the larger ϕ is, the lower is the reliance on the model,

which makes the observer rely more on the measurements taken, which, in turn, leads to faster observer poles. It is also very important to remark that, since the controller and observer are cascaded, the observer should be, at least, four times faster than the controller [4]. Table III presents the value of the ratio between the time constants of the slowest poles of the controller and observer. If one selects too high a value for ϕ , one is relying too much on the sensor's measurements, whose noise leads to a jittery control action. For that reason, one selects the lowest value of ϕ in Table III that verifies the condition on the speed ratio, *i.e.*, $\phi = 100$.

TABLE III: Ratio between the time constants of the slowest poles of the controller and observer, for various values of ϕ .

φ	T_c/T_o
10	2.28
100	6.05
1000	9.87

Fig. 7 depicts the evolution of the estimation error, for the designed observer gain and Table IV presents the poles of the observer, which, by (7), are given by the eigenvalues of $\mathbf{A} - \mathbf{LC}$. As expected, the estimation error converges to zero and the observer poles are on the LHCP.

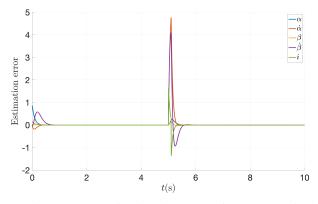


Fig. 7: Time response of estimation error of the state variables, with $\phi = 100$.

TABLE IV: Poles of the observer, with $\phi = 100$.

Pole	Damping	ω (rad/s)	Time constant (s)
-7.37e+02	1.00e+00	7.37e+02	1.36e-03
-1.06e+01 + 5.76e+00i	8.79e-01	1.21e+01	9.40e-02
-1.06e+01 - 5.76e+00i	8.79e-01	1.21e+01	9.40e-02
-1.02e+01	1.00e+00	1.02e+01	9.85e-02
-1.99e+01	1.00e+00	1.99e+01	5.04e-02

Having designed the controller and the observer separately, they can be cascaded together, *i.e.*, using estimated state feedback $u = -\mathbf{K}\hat{\mathbf{x}}$. This can be done without loss of stability, for the LTI system, due to the Separation Theorem mentioned beforehand. In fact, writing the estimation error dynamics, one obtains $\dot{\mathbf{e}} = (\mathbf{A} - \mathbf{LC})\mathbf{e}$, from (7), with $\mathbf{e} := \hat{\mathbf{x}} - \mathbf{x}$. And from the estimated state feedback control action, $\dot{\mathbf{x}} = (\mathbf{A} - \mathbf{BK})\mathbf{x} - \mathbf{BK}\mathbf{e}$. Considering the augmented system of state variables and estimation error, it is easily concluded that the poles of the closed-loop estimated state feedback system are the reunion of the poles of the controller and of the observer.

Thus, the stability and modes of the individually synthesized components remain unchanged. Nevertheless, it is important to point out that the transient time responses are inevitably going to change, given that there is no guarantee about the position of the zeros of the closed-loop estimated state feedback system.

V. SIMULATION RESULTS

Given the impossibility to simulate the controller and observers synthesized in the previous sections on the real system, the plant simulation has to be complemented to include nonideal effects to allow for a realistic simulation.

A. Nonlinear simulation model

In Sections III and IV, for the LTI model of the plant, it appears there is no limit on the performance of the controller and observer, which does not portray accurately the dynamics of the plant. The inclusion of nonlinear effects in the plant simulation allows to understand the physical limitations of the plant and to guide the synthesis of a control solution that is robust to such effects. The nonlinear effects considered are: i) nonlinear kinematics; ii) actuator saturation; iii) actuator dead zone; and iv) process and sensor noise. These are presented sequentially, as a means of understanding their impact on the plant response and on the gain synthesis. For that purpose, a simulation of T=10 s, with $\mathbf{x}_0=[50\deg\ 0\ 10\deg\ 0\ 0]^T$, an impulsive disturbance on the plant input with an amplitude of $A_p = 4V$ and duration of $\Delta t_p = 0.2$ s at t = 5 s, $\gamma = 10$, k=10 and $\phi=10^2$, is considered, which corresponds to the same simulation and control solution synthesis put forward in Sections III and IV.

- 1) Nonlinear kinematics: First, instead of the LTI system, the simulation was performed with the nonlinear kinematics of the plant, modeled by equation (2.19) in [2], which is shown and compared with the LTI simulation in Fig. 8. It is visible that, for small angles of β the linear model is quite accurate, which was expected since the LTI model was obtained from a linearization around $\beta=0$ deg. However, as β reaches higher values, the time responses begin to differ. In fact, while for the LTI system, one had guarantees about the stability of the controller and observer, that is no longer the case for the nonlinear system. In fact, although the closed-loop system dynamics linearized around $\beta=0$ are stable, which means that it is an asymptotically stable equilibrium of the nonlinear system, a high enough disturbance or tracking reference on α can easily destabilize the plant.
- 2) Actuator saturation: Second, a saturation of $u_{sat}=5$ V, which corresponds to the actual saturation of the physical plant, was included. Figs. 9 and 10 show the evolution of α and β and the actuator input, respectively, and their comparison with the evolution that would be obtained without saturation. It is concluded that, although the actuator is saturated for a very short duration, the effect on the time response is quite significant, leading to a significant increase of the settling time. This effect penalizes aggressive controllers and imposes a limitation on the regulation performance.
- 3) Dead zone: Third, a very common and important nonlinearity of DC motors is their dead-zone. If the interval of the dead-zone is known accurately, it can be easily inverted, which attenuates its effect. However, since it is not easy to

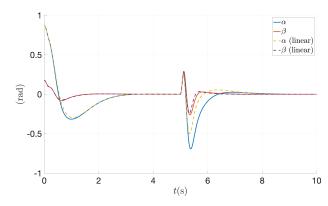


Fig. 8: Time response of α and β for the nonlinear kinematics.

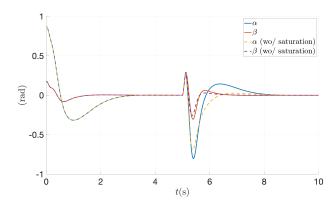


Fig. 9: Time response of α and β with input saturation.

estimate it accurately, and given that it varies with time, it was assumed that the dead-zone interval is unknown and equal to $u_{dz} = 0.2[-1;1]$ V. Figs. 11 and 12 show the evolution of α and β and the actuator input, respectively, and their comparison with the evolution that would be obtained without actuator deadzone. The introduction of the dead-zone inhibits the effect of small input voltages to the DC motor. Thus, at the equilibrium point $\beta = 0$, the controller has no authority, and the linearized dynamics correspond to the dynamics of the open loop plant, which are known to be unstable. Thus, with the inclusion of the actuator dead-zone the equilibrium point $\beta = 0$ of the nonlinear system is no longer stable, even with a state feedback controller. Nevertheless, once the actuator gets out of the dead-zone, it acts to return the plant to $\beta = 0$. For this reason, either the dead-zone is too large and the controller is unable to return the pendulum to the upright position with a given saturation, or it is able to do so resulting in an oscillatory time response, which can be seen in Fig. 11. Therefore, more aggressive controllers allow to exit the dead-band for a smaller perturbation on β , and thus achieve better regulation.

4) Process and sensor noise: Lastly, process and sensor noise were modeled as zero-mean Gaussian distributions with covariance matrices

 $\mathbf{Q_p} = 10^2 \mathrm{diag}(0.018^2, (0.18)^2, (0.018)^2, (0.18)^2, 0.025^2)$ and $\mathbf{Q_s} = 0.036^2 \mathbf{I}$, respectively. Note that this corresponds to a standard deviation of $\sigma_s \approx 1$ deg on the sensors that measure the angle of the bars. Figs. 13 and 14 show the evolution of α and β and the actuator input, respectively, and their comparison

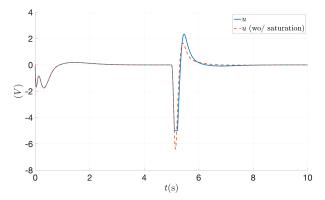


Fig. 10: Plant input with input saturation.

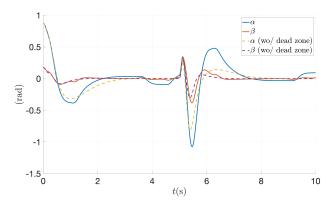


Fig. 11: Time response of α and β with actuator dead-zone.

with the evolution that would be obtained without noise. It is visible that there was a loss of regulation performance as well as an increase on jitteriness of the actuator input.

B. Performance evaluation

To objectively evaluate the regulation performance of the controller on the plant with nonlinear effects, a cost function of the sum of the squared error on the angles α and β , i.e.,

$$\bar{J} = \int_0^T \alpha(t)^2 + \beta(t)^2 ,$$

is considered. From the analysis conducted in the previous subsection it was possible to conclude that the nonlinear effects introduce performance limitations. In fact, on one hand, the saturation on the plant input imposes the use of a not too aggressive controller to keep the desired inputs outside the saturation region. On the other hand, the more aggressive the controller is, the less effect the dead-zone has and the quicker the state variables are regulated. In this sense, one has to find a compromise between these two different effects. For that reason, \bar{J} was computed for various combinations of controller and observer gains and it is shown in Fig. 15.

First, it is visible that there are combinations of controller and observer gains that destabilize the closed loop system, which correspond to the green and yellow *plateaus* in Fig. 15. As discussed in the previous subsection, the nonlinear effects can easily destabilize the closed loop dynamics if the gains are not synthesized robustly and with awareness of the effects. Second, as put forward in Section IV, it is necessary that the observer is

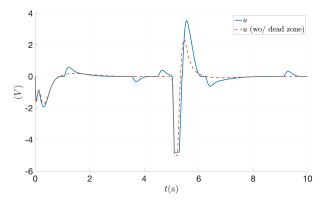


Fig. 12: Plant input with with actuator dead-zone.

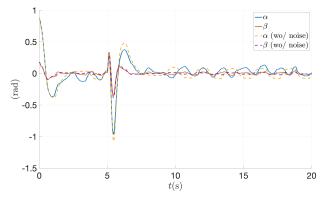


Fig. 13: Time response of α and β with process and sensor noise.

at least four times faster than the controller. The combinations of controller and observer gains which do not follow this constraint are represented in the orange region of the surface of Fig. 15. It is visible that, in this region, the system does not become, generally, unstable, but the regulation performance degrades significantly. Third, one can easily find the couple of controller and observer gains that minimizes the cost function, which is represented by the red star in Fig. 15. The minimum corresponds to $\gamma = 316.2$ and $\phi = 100.0$. Figs. 13 and 14 show the evolution of α and β and the actuator input for the optimal solution, respectively, and their comparison with the evolution that was obtained in the previous section. It is visible that there was a clear improvement on the perturbation disturbance properties of the control solution. It is also noticeable that the optimal solution is quite aggressive, which allows to attenuate the effect of the dead zone. Table V shows the poles of the optimal closed loop solution. It is possible to conclude that there are two complex conjugate pairs of poles and the real pole of the DC motor dynamics, which is very fast, remains close to its open loop position.

TABLE V: Eigenvalues of the optimal closed loop system.

Pole	Damping	ω (rad/s)	Time constant
-7.37e + 02	1.00e+00	7.37e+02	1.36e-03
-3.75e + 01 + 3.44e + 01i	7.37e-01	5.08e+01	2.67e-02
-3.75e + 01 - 3.44e + 01i	7.37e-01	5.08e+01	2.67e-02
-1.54e + 00 + 1.38e + 00i	7.45e-01	2.07e+00	6.49e-01
-1.54e + 00 - 1.38e + 00i	7.45e-01	2.07e+00	6.49e-01

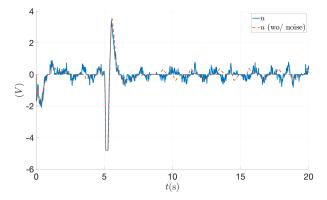


Fig. 14: Plant input with process and sensor noise.

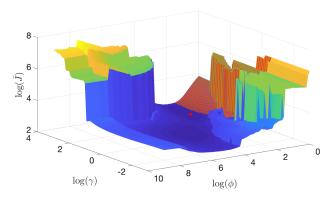


Fig. 15: Regulation cost function for various controllers and observers.

C. Disturbances and integral action

Apart from the nonideal effects considered in Section V-A, it also important to consider an additional two nonideal effects: i) the model identification uncertainty; and ii) constant disturbances.

1) Model uncertainty: The model that describes the physical plant depends on a set of physical and electrical parameters that are identified performing a set of experiments on the plant. Although this project does not cover such procedure, it is still important to take into account the uncertainty of the parameters. The uncertainty of the parameters of the plant under study is presented in [2]. To study the effects of this uncertainty the optimal solution obtained in Section V-B is simulated with uncertainty in the physical and electrical parameters of the network. Each parameter was obtained from a Gaussian distribution with mean of the nominal value of the parameter and standard deviation presented in [2]. The evolution of the angles for this simulation is shown is Fig. 18. It is visible that the identified parameters are very accurate, because there is no visible impact on the time evolution of α or β . Nevertheless, as aforementioned, the plant dynamics have an integrator, which guarantees that even in the presence of substantial model uncertainty, as long as the system remains stable, α is regulated to zero, or, considering the tracking problem, there is no steadystate error because of this effect. All the simulations presented from this point include parameter uncertainty.

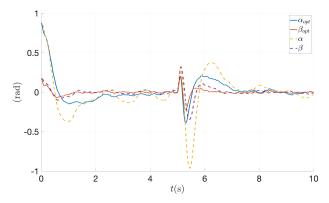


Fig. 16: Time response of α and β of the optimal solution.

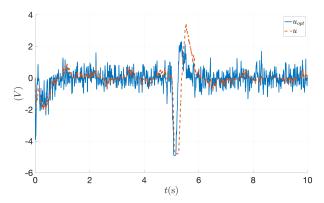


Fig. 17: Plant input with of the optimal solution.

2) Disturbances: Another important aspect to take into account are disturbances. Impulsive disturbances were studied in Section V-A, and the controller successfully attenuates such perturbations. Furthermore, there can also be constant (or slowly time-varying) disturbances either i) at the output of the plant, or ii) at the input. For this plant in particular, a constant disturbance on α at the output of the plant is successfully rejected because of the presence of the integrator in the plant dynamics. Nevertheless, there can also be a constant disturbance at the input of the plant, caused, for instance, by a bias on the DC motor amplifier. Consider that the motor amplifier has a bias of $u_{bias} = 0.2$ V. Fig. 19 shows the evolution of α and β in these conditions and it is compared with the evolution without DC motor amplifier bias. It is evident that the controller is unable to reject this constant disturbance and the angle α is not regulated to close to zero.

The poor performance in the presence of such disturbance suggests the need to include integral action. Consider a new state variable x_I defined by the differential equation $\dot{x_I} = \mathbf{C}_{\alpha}\mathbf{x} - 0$, with $\mathbf{C}_{\alpha} = \begin{bmatrix} 1 & 0 & 0 & 0 \end{bmatrix}$, *i.e.*, an integrator of the regulation error, and a new control action $u = -\mathbf{K}\mathbf{x} - K_Ix_I$. The augmented dynamics are, thus, given by

$$\begin{bmatrix} \dot{x} \\ x_I \end{bmatrix} = \mathcal{A} \begin{bmatrix} x \\ x_I \end{bmatrix} + \mathcal{B}u , \qquad (8)$$

where

$$\mathcal{A} := egin{bmatrix} \mathbf{A} & 0 \\ \mathbf{C}_{lpha} & 0 \end{bmatrix} \quad \text{and} \quad \mathcal{B} := egin{bmatrix} \mathbf{B} \\ 0 \end{bmatrix} \ .$$

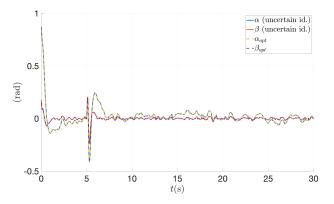


Fig. 18: Time response of α and β with identification uncertainty.

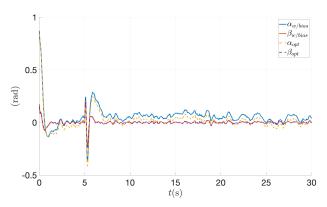


Fig. 19: Time response of α and β with a bias on the DC motor amplifier.

The same controller synthesis procedure may be used to synthesize a gain for the augmented system (8) including the integral action. For this purpose, an augmented matrix $\mathcal Q$ is chosen

$$\mathcal{Q} = \mathrm{diag}(\mathbf{Q}^{opt}, \mathbf{Q}_{11}^{opt}/100) ,$$

where \mathbf{Q}^{opt} is the state weighting matrix found to be optimal in Section V-B. Note that the weight of the integral state was chosen to be two orders of magnitude below the weight of α , such that the previously designed poles are not very affected, but rejection of constant (or slowly time-varying) disturbances is achieved. The poles of the closed-loop system obtained with this this design approach are shown in Table VI. Comparing Tables V and VI, it is possible to conclude that, as desired, the closed loop poles remain very close to those obtained before the addiction of integral action. Fig. 20 shows the evolution of α and β with integral action and amplifier bias and it is compared with the evolution without DC motor amplifier bias. The controller is now able to regulate the angle α , and a performance similar to the optimal synthesis is achieved.

It is important to note that the transient response after the pulse became significantly worse with the addition of integral action. Although in this case, for a relatively small disturbance, the system remains stable, for more significant disturbance pulses, or significant changes in the tracking angle of the tracking problem, that is not true. A well-known solution to this problem is the use of an integrator anti-windup technique,

TABLE VI: Eigenvalues of the closed loop system with integral action.

Pole	Damping	ω (rad/s)	Time constant (s)
-7.37e + 02	1.00e+00	7.37e+02	1.36e-03
-3.75e + 01 + 3.44e + 01i	7.37e-01	5.08e+01	2.67e-02
-3.75e + 01 - 3.44e + 01i	7.37e-01	5.08e+01	2.67e-02
-1.53e + 00 + 1.39e + 00i	7.41e-01	2.07e+00	6.52e-01
-1.53e + 00 - 1.39e + 00i	7.41e-01	2.07e+00	6.52e-01
-3.16e - 01	1.00e+00	3.16e-01	3.16e+00

described, for instance, in [3]. The block diagram of the proposed solution is presented in Fig. 21, in which the integral state variable is limited to ± 0.3 rad s.

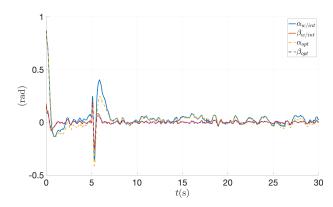


Fig. 20: Time response of α and β with a bias on the DC motor amplifier and integral action.

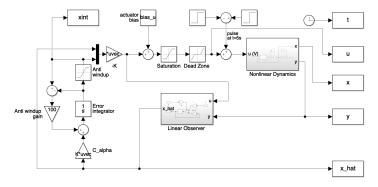


Fig. 21: Block diagram of the controller with integral action.

VI. REFERENCE TRACKING

The tracking problem is now considered, in which the goal is to follow piece-wise constant reference signals with α . To extend the results and synthesis procedure of Section V-C, a suitable feedforward control action is devised to achieve null steady state error without the presence of the bias on the DC motor amplifier, and the integral state is redefined.

Consider a control action of the form $u = -\mathbf{K}\mathbf{x} - K_Ix_I + N\alpha_{ref}$, where N is the feedforward gain and α_{ref} is the piece-wise constant reference signal. The state and integral action feedback gains are those synthesized in Section V-C, achieving the closed-loop poles in Table VI. First, neglect the presence of the integral action feedback term. Following the usual procedure, put forward in the course lectures, the transfer

function is given by

$$\frac{\alpha(s)}{\alpha_{ref}(s)} = \mathbf{C}_{\alpha}(s\mathbf{I} - (\mathbf{A} - \mathbf{B}\mathbf{K})^{-1})\mathbf{B}N.$$

Given that the closed-loop system is stable (assuming that the non-ideal effects do not destabilize the system), then one can make use of the Finite-Value Theorem, to find the static gain $G_0 = -\mathbf{C}_{\alpha}(\mathbf{A} - \mathbf{B}\mathbf{K})^{-1}\mathbf{B}N$. To achieve unitary static gain in these conditions, then, one ought to select

$$N = -1/(\mathbf{C}_{\alpha}(\mathbf{A} - \mathbf{B}\mathbf{K})^{-1}\mathbf{B}). \tag{9}$$

Before continuing, it is important to remark that although it may seem from (9) that the value of N is not robust as far as uncertainty in the model is concerned, that is not actually the case. It is easy to verify that due to the presence of an integrator pole in the plant dynamics, the structure of matrix \mathbf{A} is such that, for variations of physical and electrical parameters in relation to the nominal values, unitary G_0 is achieved. Finally, to reject the bias on the DC motor amplifier, the integral state is redefined, using the differential equation $\dot{x}_I = \mathbf{C}_{\alpha} \mathbf{x} - \alpha_{ref}$.

Before proceeding it is important to point out that the use of a feedforward term in the control action is not necessary as far as ensuring an unitary static gain is concerned. In fact, the presence of the integral term achieves unitary static gain, if the system remains stable. Nevertheless, abrupt changes in the reference signal, which are very common in piece-wise constant signals, require significant changes in the steady-state value of the integral state x_I . This consequence is very undesirable, not only because it leads to significant overshoot, but can also destabilize the system. For that reason, with the addition of the feedforward control action, the integral state is only such that constant (or slowly time-varying perturbations) are rejected. The integral state anti-windup technique can, then, be used, as in the previous section, to limit integral control action. Fig. 22 depicts the block diagram of the proposed tracking solution.

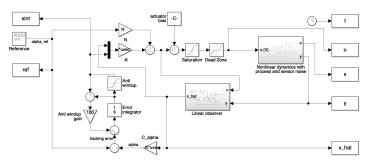


Fig. 22: Block diagram of the controller with integral action for the tracking problem.

Figs. 23 and 24 show the evolution of the angles α , β , and the reference α_{ref} , and the control action, respectively. First, it is possible to confirm that the proposed control solution successfully tracks the piece-wise constant reference signal. Second, it is also worth pointing out that the transient response has no noticeable overshoot, because of the anti-windup scheme that was implemented. Third, the presence of a nonminimum phase zero, as pointed out in Section II-A, is very evident in this time response. In fact, every time there is an abrupt change in the reference signal, there is a whiplash effect on the evolution of angle α . Fourth, it is possible to conclude that the input is

such that it does not reach saturation, but is aggressive enough to achieve good performance in the presence of a dead zone. This is only possible due to the thorough control solution synthesis performed in Sections III and IV, which was adjusted afterwards taking into account the non-ideal effects of the real plant, in Section V.

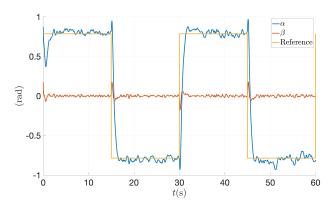


Fig. 23: Time response of α and β with process and sensor noise.

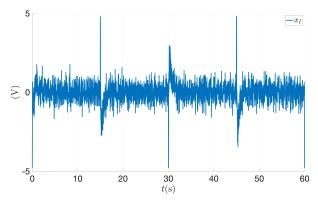


Fig. 24: Plant input with process and sensor noise.

VII. CONCLUSION

In this project, a control solution for an inverted pendulum is proposed. Although its dynamics are nonlinear, the control solution synthesis procedure followed makes use of the LTI model of the linearized pendulum about the upright position. A regulation strategy is proposed, which is then extended to the tracking problem, both yielding good performance. First, the plant was characterized in terms of its dynamic modes, where a correspondence was established between the poles of the LTI model and the physical interactions of the real plant. Second, the system was analyzed in terms of controllability and observability, having concluded that it is controllable about the upright position, and, if the angles of both bars are measured, it is observable. Third, the frequency response of the LTI model was analyzed as well. It gave insights into the reason why the system can be controlled with a single input, and why it is necessary to measure the angle of the horizontal bar for the system to be observable. Fourth, the control solution was divided into two design problems: i) the controller with full state feedback; and ii) the state observer. The Separation Theorem guarantees that, when cascaded, the estimated state feedback system remains stable. Although it is possible to design an infinitely fast controller for the LTI model, it is not realist for the application in the real plant. Thus, a compromise between a fast and robust controller is made. Fifth, the state observer is designed to regulate the state estimation error dynamics as fast as possible. Given that the sensor measurements are contaminated with noise, the observer is designed to achieve a compromise between speed and robustness to sensor noise. Sixth, several non-ideal dynamics were taken into account, and their effects were analyzed individually to assess their impact on the control solution design. Seventh, considering a cost index, the previously designed gains were recomputed to mitigate the non-ideal effects, yielding a significant improvement in regulation performance. Eighth, it was shown that the control solution is insensitive to model identification uncertainty, impulsive perturbations, and slowly time-varying disturbances on the output. Integral action was added to the control solution as a means of rejecting a constant bias on the DC motor amplifier that could not be rejected otherwise. Finally, the regulation problem was extended to track piece-wise constant signals with the angle of the horizontal bar. A control solution based on both feedback and feedforward components is proposed, yielding good tracking performance.

REFERENCES

- [1] Brian DO Anderson and John B Moore. *Optimal control: linear quadratic methods*. Courier Corporation, 1989.
- [2] Samuel Balula. "Nonlinear control of an inverted pendulum". In: *M. Sc. Dissertation. Instituto Superior Tecnico, Lisboa* (2016).
- [3] Emilio Frazzoli. *Feedback Control Systems Lecture Notes*. MIT OpenCourseWare, Fall 2010.
- [4] J. M. Lemos. Controlo no Espaço de Estados. IST Press, Jan. 2019.
- [5] Wilson J Rugh. *Linear system theory*. Prentice-Hall, Inc., 1996.