



WICHITA STATE
UNIVERSITY

AE-707: MODULE 1, LECTURE 1: INTRODUCTION

ATRI DUTTA

AEROSPACE ENGINEERING

COURSE OVERVIEW

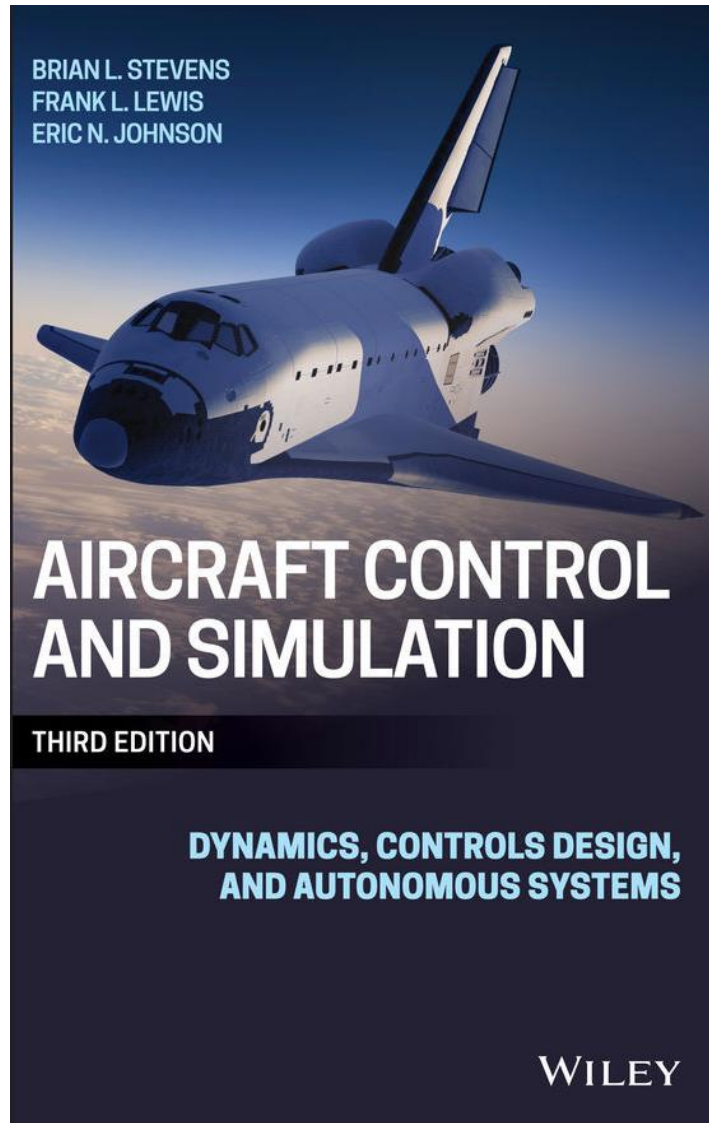
ABOUT THE COURSE

- Required course for the Flight Dynamics and Control area of the AE MS/PhD program
- Topics covered
 - State-space model
 - Techniques: eigenvalues, eigenstructures
 - Robust control: accounting for uncertainties
 - Nonlinear control: feedback linearization
 - Applications: Mostly aircraft, a bit of spacecraft
- Pre-requisites

ABOUT THE INSTRUCTOR

- Atri Dutta (pronounced more like “Au”tri D“au”tt“o”)
- Background
- Office: WH 200-B
- Walk-in office hours: MW: 3 – 4 PM, T: 4 – 5 PM
- Email: atri.dutta@wichita.edu
- Phone: 316-978-5208

TEXTBOOK

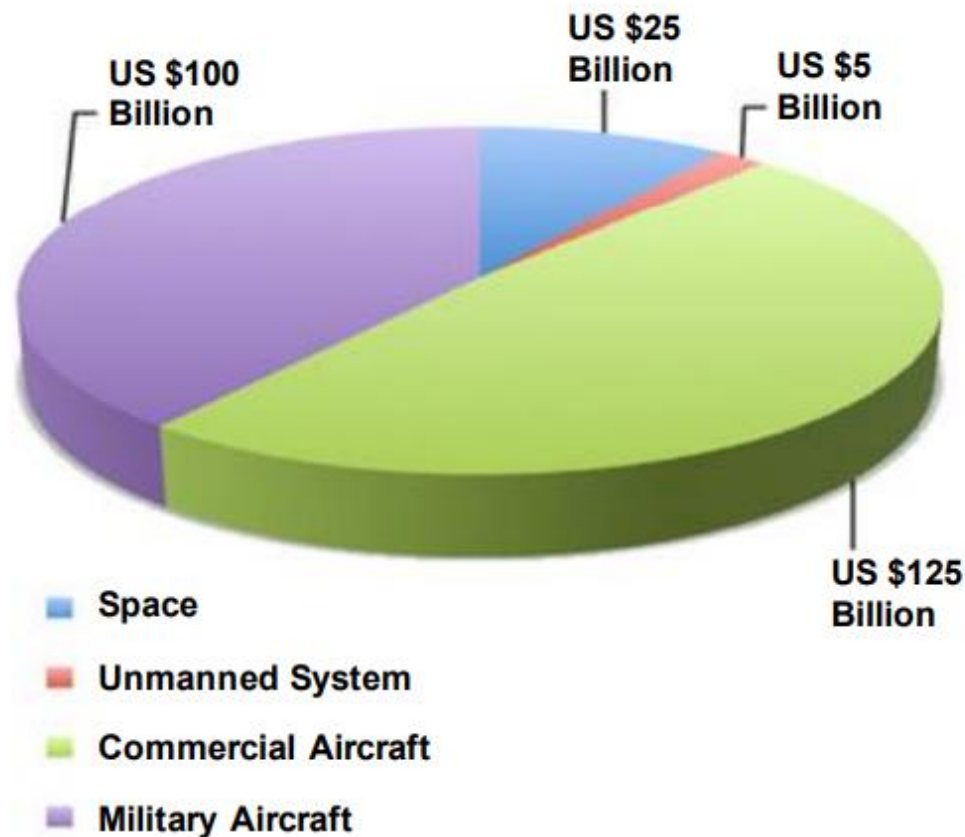


PERFORMANCE EVALUATION

- Online quiz or assignment almost each week
- Three examinations: In person, closed-book, closed-notes, bring your own equation sheet and calculator (dates already posted in schedule)
- Exam questions designed to test concepts taught
 - Familiar problem statement
 - Problems solved in class or some variation
 - Practice problems
 - Unfamiliar problem statement

AEROSPACE CONTROL SYSTEMS

AEROSPACE MARKETSIZE (2009 – 2019)

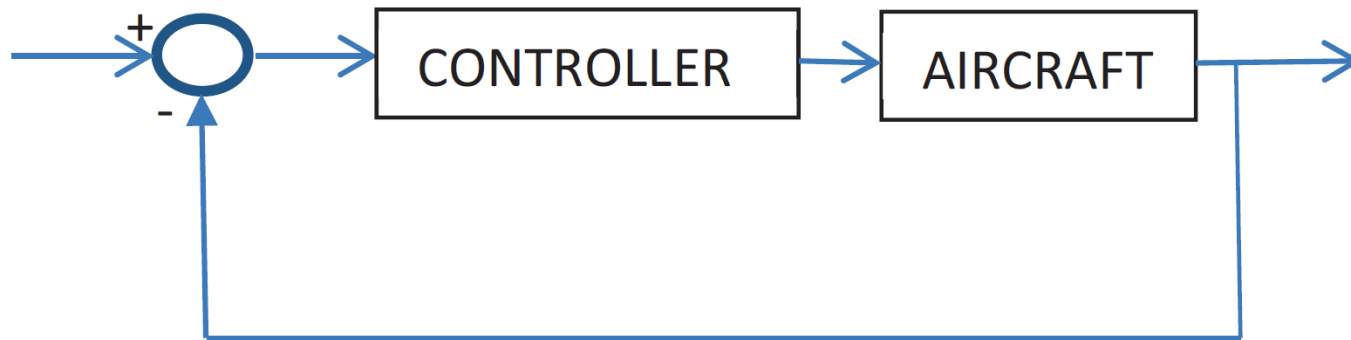


AEROSPACE CONTROL MARKETSIZE

- Aircraft demand will average around 1300 per year (Airbus and Boeing forecast 2009 – 2028); market value = \$3 trillion
- UAVs spending double over the next decade, from \$4.4 billion annually to \$8.7 billion annually (Teal Group, 2009), market value = \$62 billion
- Cost of GNC in an aircraft = 12% of total cost
- Cost of GNC in a satellite = 8 – 15% of the total cost
- The market for control technology is estimated at about \$225 billion for aviation, \$25 billion for space applications, and \$5 billion for UAVs.

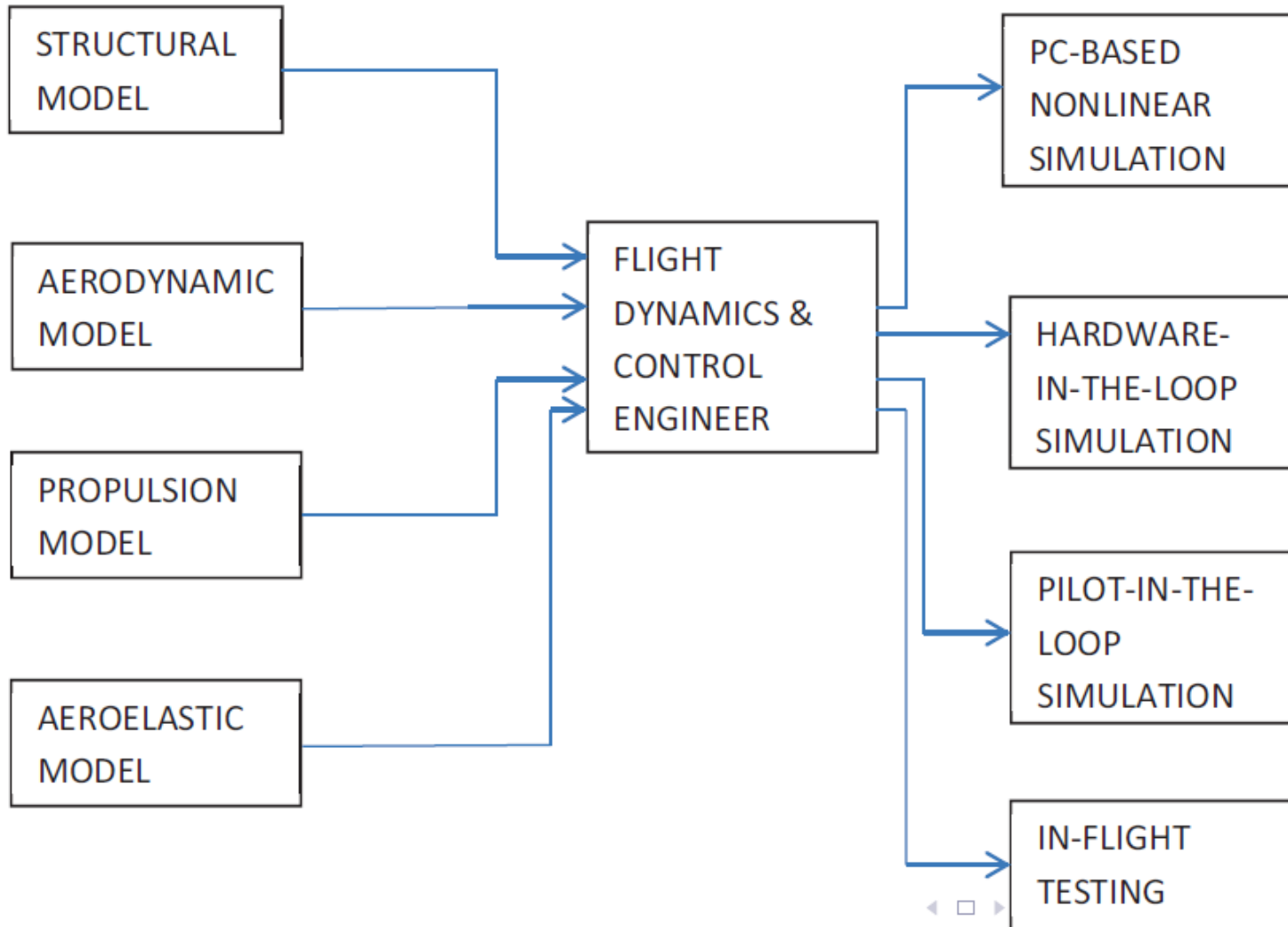
WHAT IS CONTROL SYSTEM? (1 OF 2)

WHAT IS A CONTROL SYSTEM? (2 OF 2)

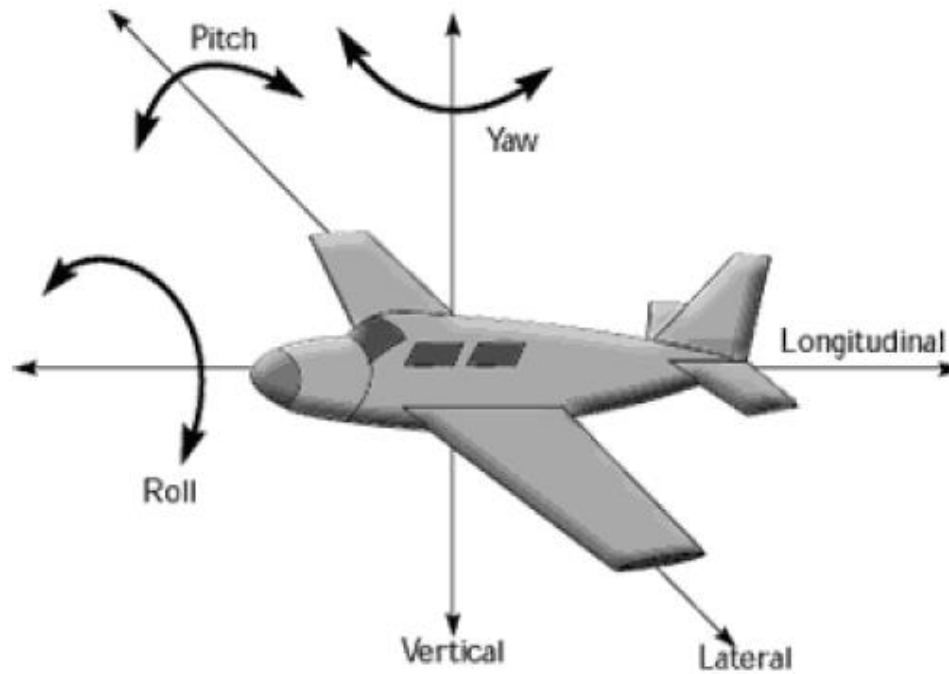


- Dynamic system that can be modified
- Desired behavior
- Actual behavior
- Corrections to the behavior of the dynamic system

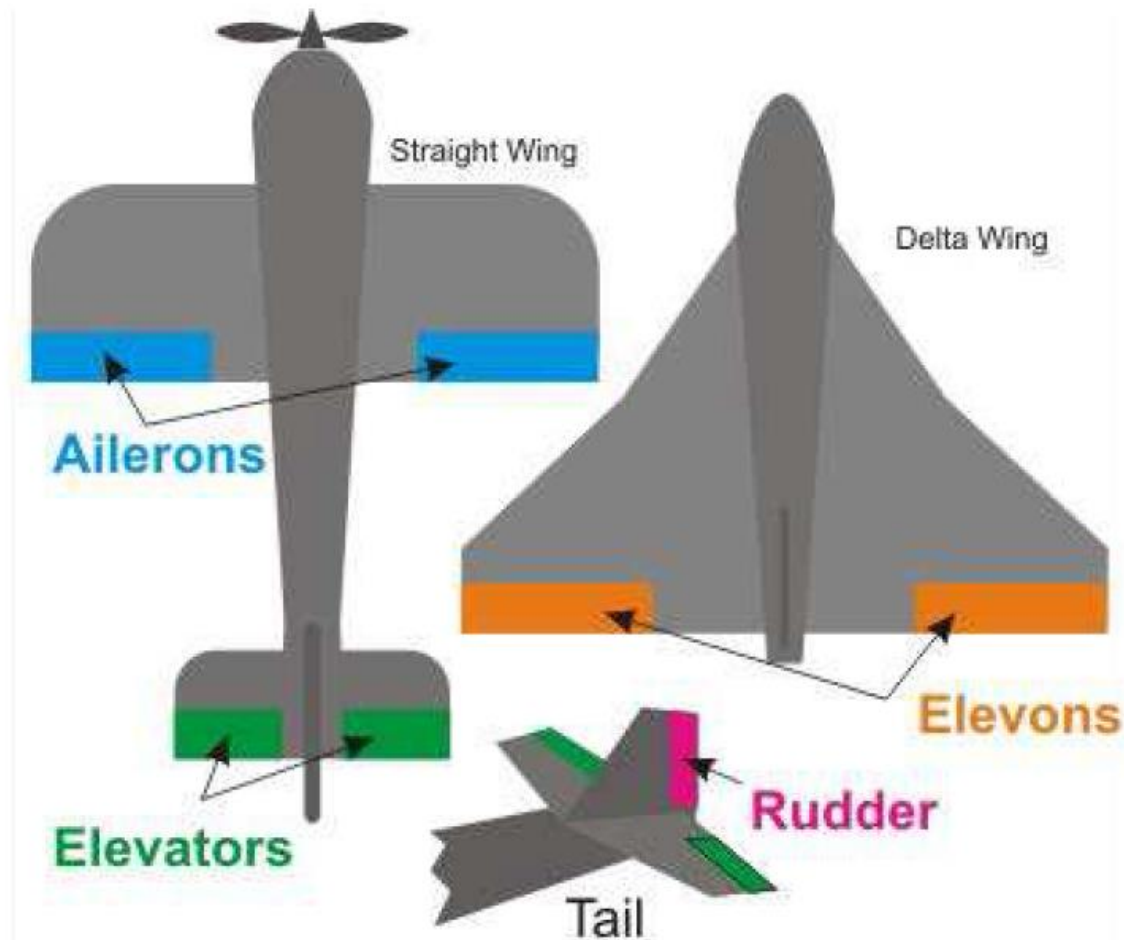
ROLE OF A CONTROL ENGINEER



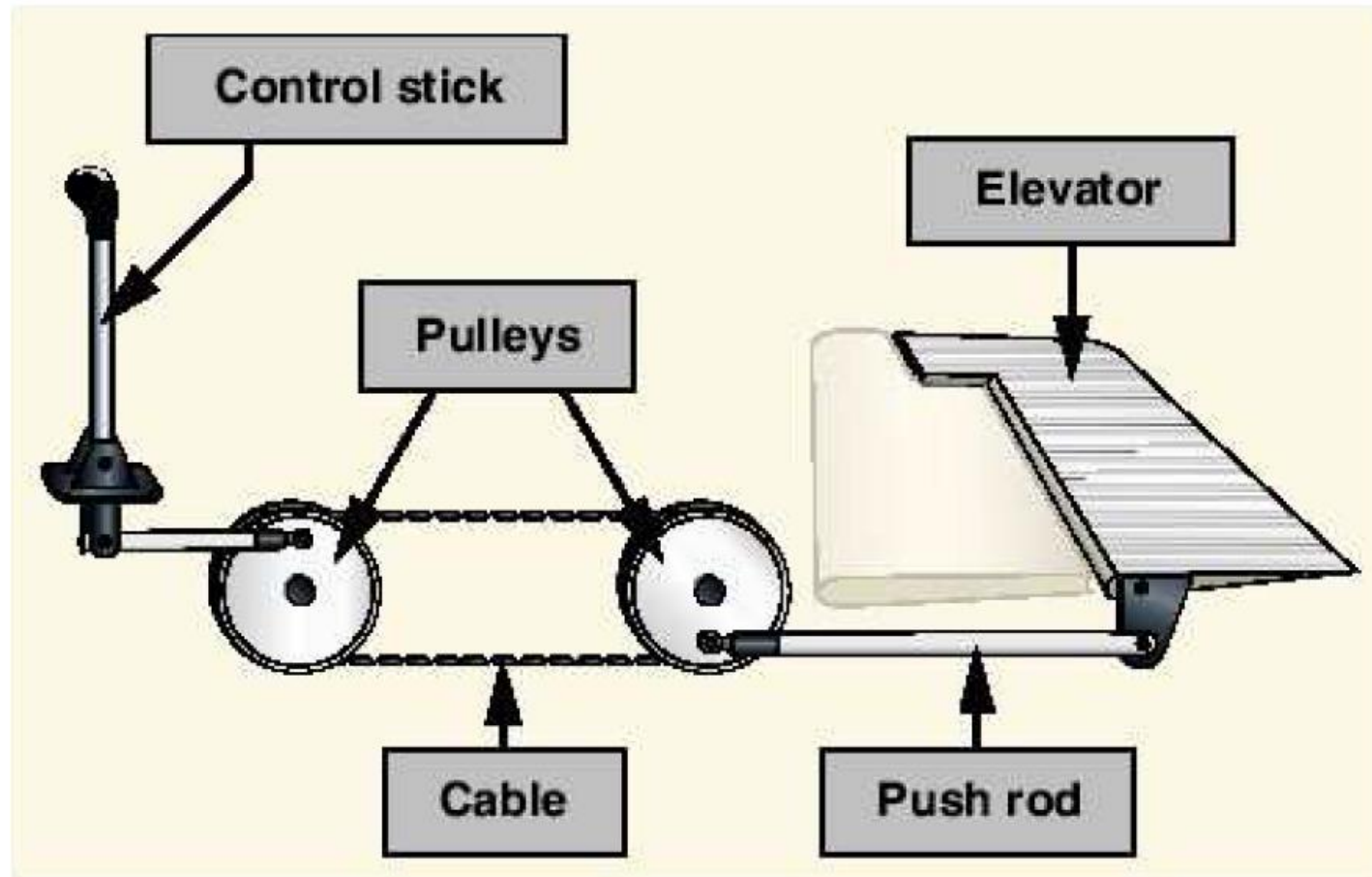
AIRCRAFT ATTITUDE ANGLES



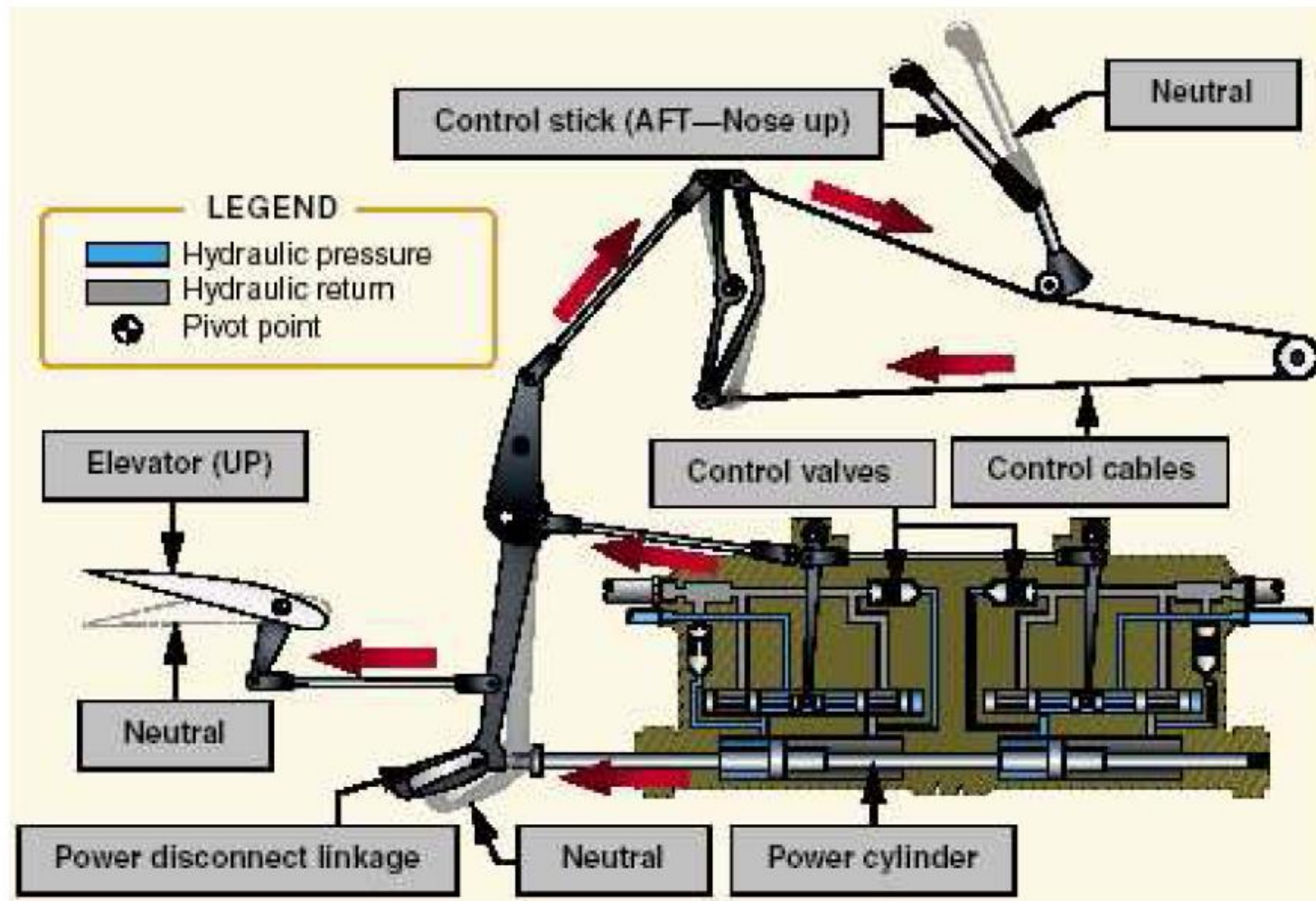
AIRCRAFT CONTROL SURFACES



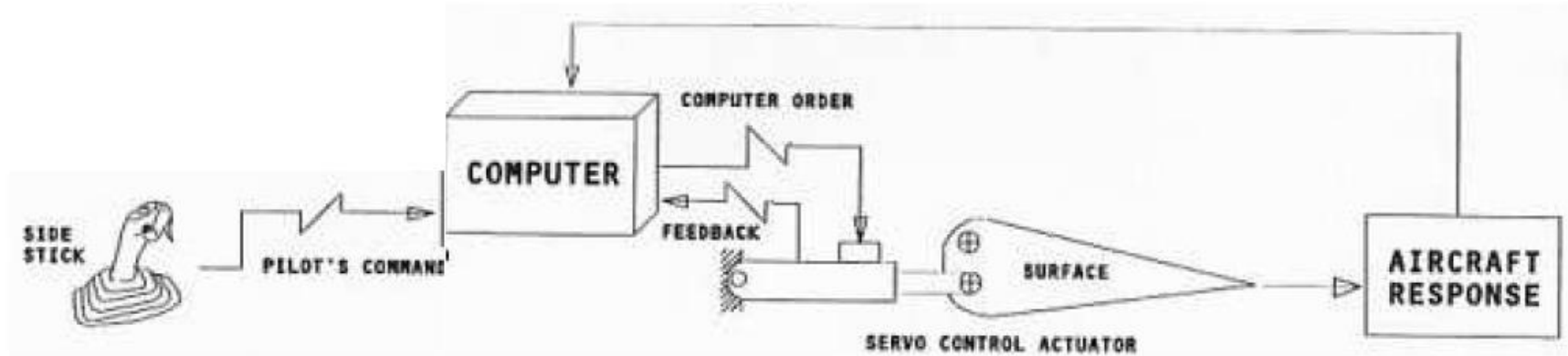
MECHANICAL FLIGHT CONTROL SYSTEM



HYDRO-MECHANICAL FCS



FLY-BY-WIRE FCS



- Advantage: weight savings due to removal of mechanical linkages; allows use of different controller in different flight conditions
- Apollo program served as the catalyst for using fly-by-wire technology in aerospace applications!

HISTORY OF FBW USE IN AIRCRAFT

- 1972: F-8 crusader jet fighter
- 1988: A320 series by Airbus (passenger aircraft)
- 1995: Boeing B-777
- 2007: Dassault Falcon 7X (business jets)
- Use in rotorcraft more recent
 - Sikorsky S92, Bell Relentless 525, Eurocopter
 - V-22 Osprey, Agusta-Westland AW609 (tilt-rotors)

MATHEMATICAL MODELS FOR DYNAMIC SYSTEM

LINEAR CONTROL DESIGN

- What is a linear system?

- State-space model

$$\mathbf{A} \in \mathbb{R}^{n \times n}$$

$$\dot{\underline{\mathbf{x}}}(t) = \mathbf{A}(t)\underline{\mathbf{x}}(t) + \mathbf{B}(t)\underline{\mathbf{u}}(t)$$

$$\mathbf{B} \in \mathbb{R}^{n \times m}$$

- Feedback controller

$$\underline{\mathbf{u}}(t) = \mathbf{K}(t)\underline{\mathbf{x}}(t)$$

$$\mathbf{K} \in \mathbb{R}^{m \times n}$$

NONLINEAR CONTROL DESIGN

- What is a nonlinear system?

$$\dot{\underline{x}}(t) = \underline{f}(t, \underline{x}, \underline{u}), \quad \underline{x}(t_0) = \underline{x}_0$$

$$\underline{x} \in \mathbb{R}^n$$

$$\underline{u} \in \mathbb{R}^m$$

- Feedback control

$$\underline{u}(t) = \underline{u}(\underline{x}(t))$$

CHALLENGES IN CONTROL DESIGN

- Challenges
 - Unmodeled dynamics
 - Output may not have all state information
 - Feedback corrupted by noise