

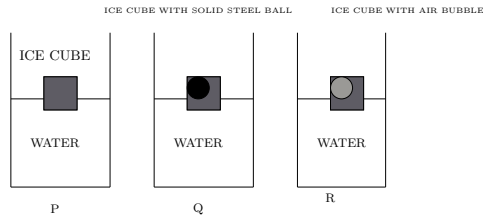
2023 February 1 Shift 2

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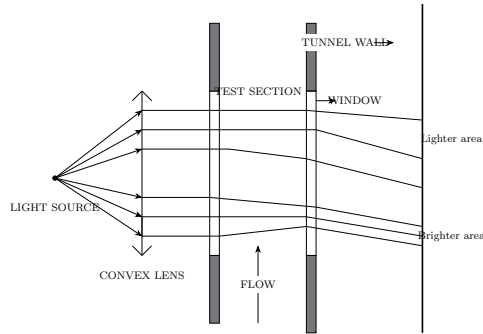
AI24BTECH11004-Bheri Sai Likith Reddy

1 SECTION-A

- 1) The figure shows three glasses P , Q and R with water and floating ice cube. Glass P has a solid ice cube, glass Q has an air bubble. After the ice cube melts, the level of water in glasses P , Q and R , respectively:



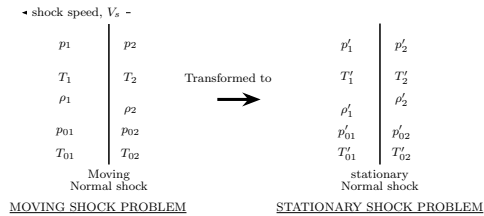
- a) remains same, increases, and decreases
b) increases, decreases, and increases
c) remains same, decreases, and decreases
d) remains same, decreases, and increases
- 2) To estimate aerodynamic loads on an aircraft flying at 100km/h at standard sea-level conditions, a one-fifth scale model is tested in a variable-density wind tunnel ensuring similarity of inertial and viscous forces. The pressure used in the wind tunnel is 10 times the atmospheric pressure. Assuming ideal gas law to hold and the same temperature conditions in model and prototype, the velocity needed in the wind tunnel test-section is _____
- a) 25km/h
b) 50km/h
c) 100km/h
d) 20km/h
- 3) The figure shows schematic of a set-up for visualization of non-uniform density field in the test section of a supersonic wind tunnel. This technique of visualization of high speed flows is known as:
- a) schlieren
b) interferometry
c) shadowgraph
d) holography
- 4) For a conventional fixed-wing aircraft in a 360° inverted vertical loop maneuver, what is the load factor (n) at the topmost point of the loop? Assume the flight to be steady at the topmost point.



- a) $n = 1$ b) $n < 1$ c) $n = -1$ d) $n > -1$

The next 5 question sare multiple select queestions and carry TWO mark each

- 5) Which of the following statement(s) is/are true about the function defined as $f(x) = e^{-x} |\cos x|$ for $x > 0$?
- Differentiable at $x = \frac{\pi}{2}$
 - Differentiable at $x = \pi$
 - Differentiable at $x = \frac{3\pi}{2}$
 - Continuous at $x = 2\pi$
- 6) A two degree of freedom spring-mass system undergoing free vibration with generalized coordinates x_1 and x_2 has natural frequencies $\omega_1 = 233.9 \text{ rad/s}$ and $\omega_2 = 324.5 \text{ rad/s}$, respectively. The corresponding mode shapes are $\phi_1 = \begin{pmatrix} 1 \\ -3.16 \end{pmatrix}$ and $\phi_2 = \text{myvec}13.16$. If the system is disturbed with certain deflection sand zero initial velocities, then which of the following statement(s) is/are true?
- An initial deflection of $x_1(0) = 6.32 \text{ cm}$ and $x_2(0) = -3.16 \text{ cm}$ would make the system oscillate with only the second natural frequency.
 - An initial deflection of $x_1(0) = 2 \text{ cm}$ and $x_2(0) = -6.32 \text{ cm}$ would make the system oscillate with only the first natural frequency.
 - An initial deflection of $x_1(0) = 62 \text{ cm}$ and $x_2(0) = -2 \text{ cm}$ would make the system oscillate with a linear combination of first and second natural frequencies.
 - An initial deflection of $x_1(0) = 1 \text{ cm}$ and $x_2(0) = -6.32 \text{ cm}$ would make the system oscillate with only the first natural frequency.
- 7) A shock moving into a stationary gas can be transformedd to a stationary shock by a change in reference frame, as shown in the figure. Which of the following is/are true relation the flow properties in the two reference frames?



- a) $T'_1 > T_1, T'_{01} > T_{01}, p'_{01} > p_{01}, \rho'_2 > \rho'_1$
 b) $T'_1 = T_1, T'_2 < T_{01}, p'_{01} > p_{01}, \rho'_2 = \rho_2$
 c) $T'_1 < T_1, p'_1 > p_1, p'_{01} > p_{01}, \rho'_2 > \rho_1$
 d) $T'_1 = T_1, p_2 > p_{01}, T'_{01} > T_{01}, p'_{01} > p_{10}$
- 8) For a conventional fixed-wing aircraft, which of the following statements are true?
- a) Making C_{m_α} more negative leads to an increase in the frequency of its short-period mode.
 b) Making C_{m_q} more negative leads to a decreased damping of the short-period mode.
 c) The primary contribution towards C_{l_p} is from the aircraft wing.
 d) Increase the size of the vertical fin leads to a higher yaw damping.
- 9) Which of the following statement(s) is/are true?
- a) Service ceiling is higher than absolute ceiling for a piston-propeller aircraft.
 b) For a given aircraft, the stall speed increases with increase in altitude.
 c) Everything else remaining the same, a tailwind increases the range of an aircraft.
 d) For a jet aircraft constrained to fly at constant altitude, there exists an altitude where its range is maximum.
- 10) A conventional fixed-wing aircraft, with a horizontal tail and vertical fin, in steady and level flight is subjected to small perturbations. Which of the following statement(s) is/are true?
- a) Vertical fin has a stabilizing effect on the lateral stability of the aircraft.
 b) Vertical fin has a destabilizing effect on the directional stability of the aircraft.
 c) Presence of wing anhedral increases the lateral stability of the aircraft.
 d) Horizontal tail has a stabilizing effect on the longitudinal static stability of the aircraft.

The next 19 questions are Numerical answer type (NAT), carry TWO marks each (no negative marks)

- 11) The ratio of the product of eigenvalues to the sum of the eigenvalues of the given matrix

$$\begin{pmatrix} 3 & 1 & 2 \\ 2 & -3 & -1 \\ 1 & 2 & 1 \end{pmatrix}$$

is _____ (round off to nearest integer)

- 12) The definite integral $\int_1^5 x^2 dx$ is evaluated using four equal intervals by two methods: first by the trapezoidal rule and then by the Simpson's one-third rule. The absolute value of the difference between the two calculations is _____ (round off to two

decimal places).

- 13) The deflection y of a certain beam of length l and uniform weight per unit length w , is given as $y = \frac{w}{48EI} (2x^4 - 3lx^3 + l^3x)$, where x is the distance from the point of support and EI is a constant. The non-dimensional location $\frac{x}{l}$, where the deflection of the beam is maximum, is _____ (round off to two decimal places).