



Balloon Design Lab

Group 3

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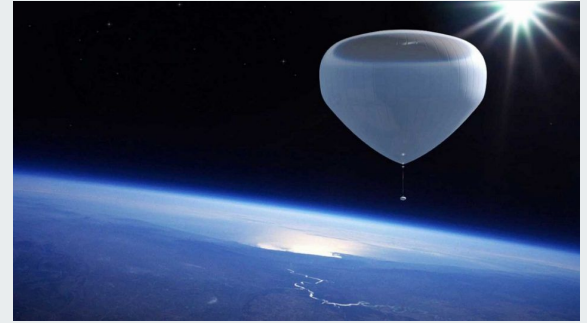




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Design Requirements

Requirements

- Remain at 25km (\pm 1km)
- Minimum of 24 hrs
- 500 kg payload (research instruments)
- Factor of safety reasonable and consistent with other projects: 1.45



Brief Design Overview

Design overview

- Final balloon size: 15.68 m radius
- Lifting gas: Helium
- Material: Thermoplastic Polyurethane Film
- Total weight: 614.54kg
- Balloon thickness: 0.00284mm
- Balloon Coating: Z93 white paint produced by AZ Technology



Design Research



Polyurethane shell

- Balloon is made of Thermoplastic Polyurethane Film
- “The most promising polymeric materials were found to be polyolefin, polyurethane, ethylene propylene diene monomer (EPDM) rubber, and silicone rubber.” - American Institute of Aeronautics and Astronautics



Reflectivity

- Z-93 white paint (reflects ~13% of solar radiation)
- Caused change in density of balloon shell:
 ρ of paint = $2,200 \text{ kg/m}^3$
- Increased mass of shell by ~10kg
- Increase mass of the system ~ 14kg
- Applied in vertical stripes to minimize effects of cracking
- \$460/ quart



Helium Propellant

- Less dense than air, relatively inexpensive and not highly flammable
- Well tested in many experiments
- Scale Model



Design Specifications



Code

Possible increase of up to 80°K
Altitude restriction of \mp 1km
Factor of Safety: 1.45
Maximum increase in temp of Helium 37.6° K
Planned increase in temperature of helium: 26°K.
Minimum reflectivity of balloon: 67.5%

Under normal (night-time) operating conditions the following values apply:

Radius: 15.49 m
Volume: 15571.51 m³
Mass of Shell: 29.61 kg
Mass of Helium: 84.93 kg
Mass of Payload: 500 kg
Total Payload: 614.54 kg
Temperature of Helium: 221.65 K

Under maximum (daytime) operating conditions these are the new Values:

Radius: 15.68 m
Volume: 16133.27 m³
Mass of Shell: 29.61 kg
Mass of Helium: 84.93 kg
Mass of Payload: 500 kg
Total Payload: 614.54 kg
Temperature of Helium: 247.65 K



Scale Model vs. Full-Scale Ballon

| | Radius | Shell Mass | Payload | Volume | Expanded Volume |
|-------------|----------|------------|---------|-----------------------|-----------------------|
| Scale Model | 17.15 cm | 9.553 g | 1.54 g | 12879 cm ³ | 19353 cm ³ |
| Full-Scale | 15.30 m | 10.12 kg | 500 kg | 14998 m ³ | 16758 m ³ |

Zero Pressure Balloons vs. Super Pressure Balloons

- Ballast system to keep Zero Pressure Balloon at specified altitude (fluctuates)
- Super Pressure don't allow change in volume to stay at altitude (level)



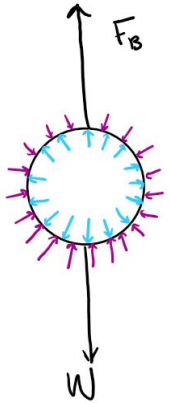


Assumptions

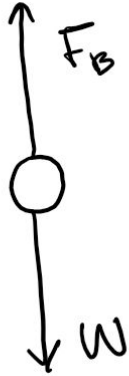
- Perfectly Spherically Shaped
- No Forces due to launching/Reaching Elevation of 25 km
- Only Forces are Weight Force and Buoyant Force
- The 500 kg payload includes any tethering mechanisms
- The gas can't escape the balloon shell (no leaks)

Design Analysis

Free Body Diagram



Local FBD



Global FBD

P_{external}
 P_{internal}
 $F_B = \text{buoyant force}$
 $W = \text{weight force}$
 \Downarrow
 $W = (m_{\text{shell}} + m_{\text{payload}} + m_{\text{gas}})g$

$$\sum F_y = ma$$

$$a = 0 \text{ (not moving)}$$

$$0 = F_B - W \Rightarrow F_B = W$$

$$\rho_{\text{air}} V = (m_{\text{shell}} + m_{\text{gas}} + m_{\text{payload}})g$$

$$\rho_{\text{air}} = \frac{m_{\text{shell}} + m_{\text{gas}} + m_{\text{PL}}}{\frac{4}{3} \pi r^3}$$



Volume and Buoyancy

$$\rho_{atm} = \frac{m_{payload} + m_{shell} + m_{He}}{\frac{4}{3}\pi r^3}$$

$$\rho_{shell} = \frac{m}{V} = \frac{m}{SA \cdot t} = \frac{m}{4\pi r^2 \cdot t} \quad \rho_{shell} = \frac{m}{4\pi r^2 \cdot t}$$

$$m_{payload} = 500\text{kg} \quad m_{shell} = \rho_{shell} \cdot t \cdot 4\pi r^2$$

$$m_{He} = \rho_{He} \cdot \frac{4}{3}\pi r^3$$

Volume and Buoyancy

$$PV = nRT \quad P = \rho_{He} R_{specific} T \quad \rho_{He} = \frac{P}{R_{spec} T}$$

$$t = \frac{P_g r}{2\sigma} \quad \sigma = \frac{Y.S.}{F.S.} \quad t = \frac{P_g \cdot r \cdot F.S.}{2 \cdot Y.S.}$$

$$m_{shell} = \rho_{shell} \left(\frac{2\pi \cdot P_g \cdot F.S.}{Y.S.} \right)$$

$$\rho_{atm} = \frac{m_{payload} + m_{shell} + m_{He}}{\frac{4}{3}\pi r^3}$$

$$r = \sqrt[3]{\frac{m_{payload}}{\frac{4}{3}\pi(\rho_{atm} - \rho_{He}) - \rho_{shell} \left(\frac{2\pi \cdot P_g \cdot F.S.}{Y.S.} \right)}}$$

$$V = \frac{4}{3}\pi r^3$$



Final Mass Budget

Our Design

- The Balloon does not have a ballast system or venting gas.
- With the use of Z93, the amount of energy entering the balloon system is lowered
- This keeps our balloon within the 25 ± 1 km altitude
- Mass = 614.54 kg

Scale Model

- The scale model had no release of gas or ballast system
- Final mass equaled Initial mass
- Mass = 12.641 g

Performance Subject to Radiation



Design Decisions

- After using radiation equations a given absorption keeps balloon in altitude range
- Researched many highly reflective materials
 - Silvered teflon
 - BoPET
 - Aluminized Kapton
- We choose Z93 because it had a high reflectivity of 87% while still being manufactured
 - Paint could be easier to work with



Assumptions, Equations, Processes

- Z93 coating would offer 87 percent reflectivity
- The Balloon is launched at sunset on winter solstice
- Constant Mass (Z93 degradation)
- Z93 adheres to polyurethane shell without need for primers

Processes:

- A spray painting gun is used to apply the Z93 to our specified thickness across a fully inflated balloon.

Design Deliverables

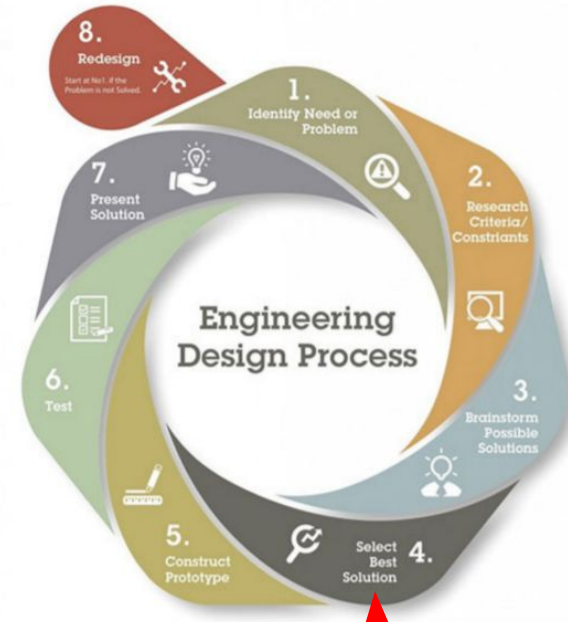


Further Research

- Weather conditions
- Recovery system
- Increasing time aloft
- Material Design

Findings

- Our design is theoretical and would need scale model testing before it could be cleared.
- Not bound by cost. We designed our balloon to surpass the design requirements rather than being a cost effective solution.
 - Cost estimate:
- Our beginning research was not adequate to have an understanding of the entire project.
- Decisions were made and changed multiple times.
- Final Decisions
 - Thermoplastic Polyurethane Film (TPU), Z93, and Helium to fulfill the design requirements



Our progress!



Sources

RTP Company Website -

<https://www.rtpcompany.com/manufacturing/film/>

AZ Technology Company Website -

<http://www.aztechnology.com/>

Material Challenges for Lighter-Than-Air Systems in High Altitude Applications-American Institute of Aeronautics and Astronautics

http://www.tcomlp.com/wp-content/uploads/2014/09/2005_7488.pdf



Images

Super Pressure and Zero Pressure -

https://sites.wff.nasa.gov/code820/spb_differences_between_zpandspb.html

Title Slide Image -

<https://www.optimistdaily.com/2020/07/nasa-will-fly-a-stadium-sized-high-altitude-balloon-to-watch-newborn-stars/>