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**055738 – STRUCTURAL DYNAMICS
AND AEROELASTICITY**

07 Structural Dynamics: Modal Analysis

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Material

Preumont: Chapter 3 (up to section 2.2) and section 5.5

Masarati: Sections 5.3 and D.1



Eigenvalues and eigenvectors

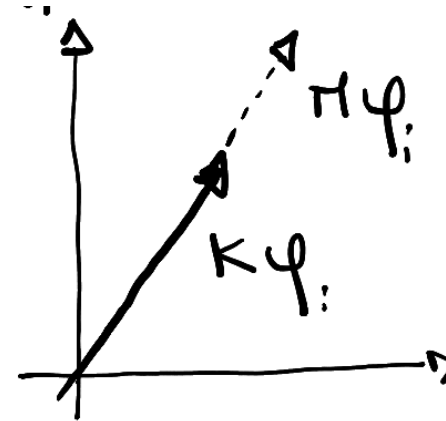
$$\mathbf{M}\ddot{\mathbf{q}} + \mathbf{K}\mathbf{q} = \mathbf{0}, \quad \mathbf{q} = \boldsymbol{\varphi}_i e^{j\omega_i t}$$

$$(-\mathbf{M}\omega_i^2 + \mathbf{K}) \boldsymbol{\varphi}_i = \mathbf{0}$$

$$\mathbf{K}\boldsymbol{\varphi}_i = \omega_i^2 \mathbf{M}\boldsymbol{\varphi}_i$$

$$\mathbf{K}\boldsymbol{\Phi} = \boldsymbol{\Lambda}\mathbf{M}\boldsymbol{\Phi}$$

$$\boldsymbol{\Phi} = [\boldsymbol{\varphi}_1, \boldsymbol{\varphi}_2, \dots, \boldsymbol{\varphi}_N]$$



The eigenmode is a vector that multiplied by \mathbf{K} is equal to itself multiplied by matrix \mathbf{M} and a constant $\lambda = \omega_i^2$

The eigenvalue is the value that allows to scale $\mathbf{M}\boldsymbol{\phi}$ to $\mathbf{K}\boldsymbol{\phi}$

$$\boldsymbol{\Lambda} = \begin{bmatrix} \ddots & & \\ & \omega_i^2 & \\ & & \ddots \end{bmatrix}$$



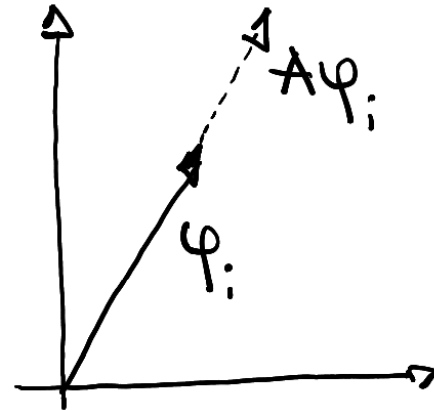
Eigenvalues and eigenvectors (state-space approach)

$$\dot{\mathbf{q}} = \mathbf{A}\mathbf{q}, \quad \mathbf{q} = \boldsymbol{\varphi} e^{\lambda t}$$

$$(\mathbf{A} - \lambda \mathbf{I}) \boldsymbol{\varphi} = \mathbf{0}$$

$$\mathbf{A}\boldsymbol{\varphi} = \lambda \boldsymbol{\varphi}$$

$$\boldsymbol{\Phi} = [\boldsymbol{\varphi}_1, \boldsymbol{\varphi}_2, \dots, \boldsymbol{\varphi}_N]$$



$$\boldsymbol{\Lambda} = \begin{bmatrix} \ddots & & \\ & \lambda_i & \\ & & \ddots \end{bmatrix}$$

$$\mathbf{A}\boldsymbol{\Phi} = \boldsymbol{\Phi}\boldsymbol{\Lambda}, \quad \boldsymbol{\Lambda} = \boldsymbol{\Phi}^{-1} \mathbf{A}\boldsymbol{\Phi}$$

The eigenmode is a vector that multiplied by \mathbf{A} is equal to itself multiplied by a constant λ



Normalization of eigenvectors or modal shapes

Modal shapes are defined unless for a constant. Several normalization are often used

- 1) Maximum displacement $\max(\phi_i) = 1.0$
- 2) Modal mass $\phi_i^T M \phi_i = \mu_i = 1$



Eigenmodes to decouple

$$\omega_i^2 \varphi_j^T \mathbf{M} \varphi_i = \varphi_j^T \mathbf{K} \varphi_i$$

$$\omega_j^2 \varphi_i^T \mathbf{M} \varphi_j = \varphi_i^T \mathbf{K} \varphi_j$$

Since $\varphi_j^T \mathbf{M} \varphi_i$ is a scalar so is equal to its transpose. Same for $\varphi_j^T \mathbf{K} \varphi_i$

$$\omega_j^2 \varphi_j^T \mathbf{M} \varphi_i = \varphi_j^T \mathbf{K} \varphi_i$$

$$(\omega_i^2 - \omega_j^2) \varphi_j^T \mathbf{M} \varphi_i = 0$$

1. Orthogonality with respect to mass matrix

$$\varphi_i^T \mathbf{M} \varphi_j = 0 \quad \text{if } i \neq j$$

2. Orthogonality with respect to stiffness matrix

$$\varphi_i^T \mathbf{K} \varphi_j = 0 \quad \text{if } i \neq j \quad \leftarrow \quad 2\varphi_j^T \mathbf{K} \varphi_i = (\omega_i^2 + \omega_j^2) \varphi_j^T \mathbf{M} \varphi_i$$

WARNING!! In general

$$\varphi_i^T \varphi_j \neq 0 \quad \forall i, j$$



Eigenmodes to decouple

$$\mathbf{q} = \Phi \mathbf{z} \quad \text{with } \Phi = [\varphi_1, \varphi_1, \dots, \varphi_N]$$

$$\mathbf{M}\Phi\ddot{\mathbf{z}} + \mathbf{K}\Phi\mathbf{z} = \mathbf{0} \rightarrow \Phi^T (\mathbf{M}\Phi\ddot{\mathbf{z}} + \mathbf{K}\Phi\mathbf{z}) = \mathbf{0}$$

$$\Phi^T \mathbf{M} \Phi = \begin{bmatrix} \ddots & & \\ & \mu_i & \\ & & \ddots \end{bmatrix}, \quad \Phi^T \mathbf{K} \Phi = \begin{bmatrix} \ddots & & \\ & \mu_i \omega_i^2 & \\ & & \ddots \end{bmatrix}$$

$$\mu_i (\ddot{z}_i + \omega_i^2 z_i) = 0 \quad \forall i$$

The N -dimensional second order differential system is decomposed in N independent scalar linear second order differential equations.



Transformation to first order state-space form

Define a new state variable as the first derivative of states

$$\dot{\mathbf{q}} = \mathbf{q}_d$$

The equation becomes

$$\mathbf{M}\dot{\mathbf{q}}_d + \mathbf{K}\mathbf{q} = \mathbf{0}$$

$$\begin{bmatrix} \mathbf{I} & \mathbf{0} \\ \mathbf{0} & \mathbf{M} \end{bmatrix} \begin{Bmatrix} \dot{\mathbf{q}} \\ \dot{\mathbf{q}}_d \end{Bmatrix} = \begin{bmatrix} \mathbf{0} & \mathbf{I} \\ -\mathbf{K} & \mathbf{0} \end{bmatrix} \begin{Bmatrix} \mathbf{q} \\ \mathbf{q}_d \end{Bmatrix}$$

$$\check{\mathbf{M}}\dot{\check{\mathbf{q}}} = \check{\mathbf{K}}\check{\mathbf{q}}$$

if $\check{\mathbf{M}} > 0$ i.e., is positive definite, then

$$\dot{\check{\mathbf{q}}} = \check{\mathbf{M}}^{-1}\check{\mathbf{K}}\check{\mathbf{q}}, \rightarrow \dot{\check{\mathbf{q}}} = \check{\mathbf{A}}\check{\mathbf{q}}$$



Eigenmodes to decouple (state space approach)

$$\dot{\mathbf{q}} = \mathbf{A}\mathbf{q}$$

$$\begin{aligned}\boldsymbol{\varphi}_j^T \mathbf{A} \boldsymbol{\varphi}_i &= \lambda_i \boldsymbol{\varphi}_j^T \boldsymbol{\varphi}_i \\ \boldsymbol{\varphi}_i^T \mathbf{A} \boldsymbol{\varphi}_j &= \lambda_j \boldsymbol{\varphi}_i^T \boldsymbol{\varphi}_j\end{aligned}\quad (\lambda_i - \lambda_j) \boldsymbol{\varphi}_i^T \boldsymbol{\varphi}_j = 0$$

Consequently,

$$\boldsymbol{\varphi}_i^T \boldsymbol{\varphi}_j = 0 \quad \text{if } i \neq j \rightarrow \boldsymbol{\Phi}^T \boldsymbol{\Phi} = \mathbf{I} \Rightarrow \boldsymbol{\Phi}^T = \boldsymbol{\Phi}^{-1}$$

The matrix of eigenvectors is an orthogonal matrix

$$\mathbf{A}\boldsymbol{\Phi} = \boldsymbol{\Phi}\boldsymbol{\Lambda} \Rightarrow \boldsymbol{\Lambda} = \boldsymbol{\Phi}^T \mathbf{A} \boldsymbol{\Phi}$$

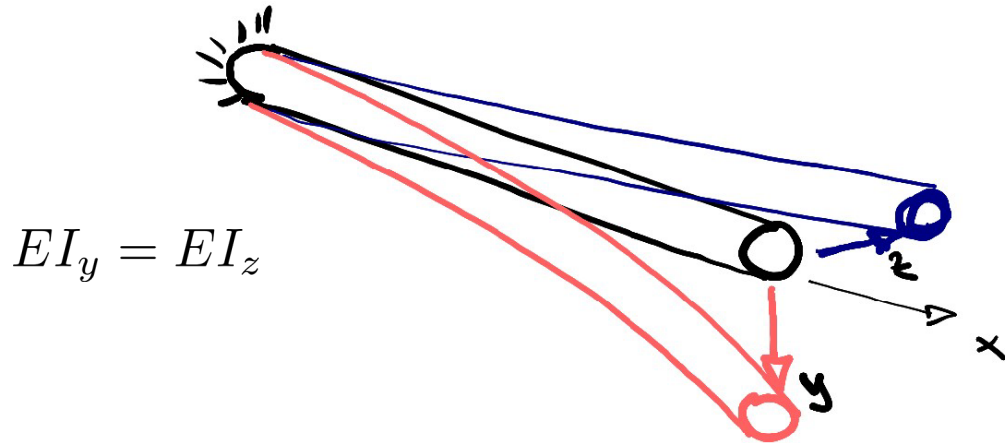
$$\text{If } \mathbf{q} = \boldsymbol{\Phi}\mathbf{z} \Rightarrow \dot{\mathbf{z}} = \boldsymbol{\Phi}^T \mathbf{A} \boldsymbol{\Phi} \mathbf{z} \rightarrow \dot{z}_i = \lambda z_i \quad \forall i$$

The N -dimensional differential system is decomposed in N independent scalar linear differential equations.



Coincident eigenvalues

They typically appear when there are symmetries in the structures under analysis



$$E_v = \{ \mathbf{v}_i : (\mathbf{A} - \lambda_i \mathbf{I}) \mathbf{v}_i = 0 \}$$

Algebraic multiplicity: the number of times an eigenvalue is repeated

Geometric multiplicity: is the dimension of the eigenspace E_v associated with an eigenvalue λ_i



Coincident eigenvalues

If the algebraic multiplicity is = geometric multiplicity the matrix is still diagonalizable because all eigenvectors associated to the same eigenvalue are linearly independent

$$\mathbf{\Lambda} = \mathbf{\Phi}^T \mathbf{A} \mathbf{\Phi}$$

or

$$\mathbf{\Lambda} \mathbf{\Phi}^T \mathbf{K} \mathbf{\Phi} = \mathbf{\Phi}^T \mathbf{M} \mathbf{\Phi}$$

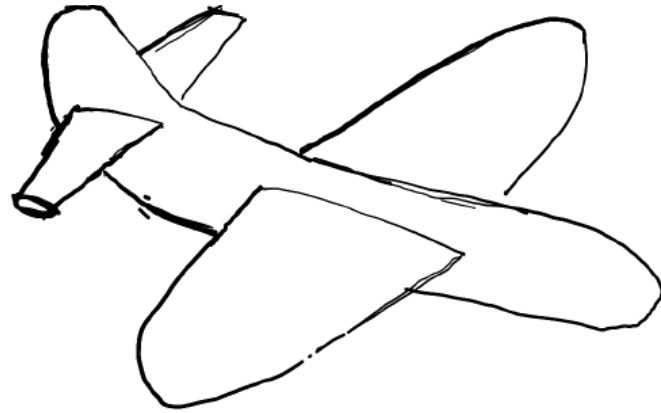


Rigid body modes

Rigid modes

$$\omega_i = 0$$

$$\omega_i^2 \mathbf{M} \boldsymbol{\varphi}_i = \mathbf{K} \boldsymbol{\varphi}_i \Rightarrow \mathbf{K} \boldsymbol{\varphi}_i = \mathbf{0}$$



It follows that the strain energy is $\boldsymbol{\varphi}_i^T \mathbf{K} \boldsymbol{\varphi}_i = 0$ that means that NO DEFORMATION is generated.

For a free flying aircraft 3 rigid translation and 3 rigid rotations are possible, so 6 rigid body modes with null frequency

\mathbf{K} matrix is six times singular for a free flying aircraft

$$(\omega_i^2 - \omega_j^2) \boldsymbol{\varphi}_i^T \mathbf{M} \boldsymbol{\varphi}_j = 0$$

$\omega_i = 0$ for $i = 1, \dots, 6$ so

$$\Rightarrow \boldsymbol{\varphi}_i^T \mathbf{M} \boldsymbol{\varphi}_j \neq 0 \quad \text{when } i, j = 1, \dots, 6$$

The coincidence between algebraic and geometric multiplicity is ensured

$$E_\varphi = \{\boldsymbol{\varphi}_i : \mathbf{K} \boldsymbol{\varphi}_i = \mathbf{0}\} \equiv \mathbb{R}^6$$



Rigid body modes

If the initially computed rigid body modes are not linearly independent, it is possible to compute those who diagonalize the matrices through the GRAM-SCHMIDT orthotgonalization

If $\varphi_j^T \mathbf{M} \varphi_i \neq 0$ define a modified j mode shape

$$\tilde{\varphi}_j = \varphi_j - \alpha \varphi_i$$

so that

$$\tilde{\varphi}_j^T \mathbf{M} \varphi_i = 0$$

$$\varphi_j^T \mathbf{M} \varphi_i - \alpha \varphi_i^T \mathbf{M} \varphi_i = 0$$

This allows to compute α as

$$\alpha = \frac{\varphi_j^T \mathbf{M} \varphi_i}{\varphi_i^T \mathbf{M} \varphi_i} \Rightarrow \tilde{\varphi}_j = \varphi_j - \frac{\varphi_j^T \mathbf{M} \varphi_i}{\varphi_i^T \mathbf{M} \varphi_i} \varphi_i$$



Computation of the solution to initial conditions

$$\mathbf{M}\ddot{\mathbf{q}} + \mathbf{K}\mathbf{q} = \mathbf{0}, \quad \mathbf{q} \in \mathbb{R}^n, \quad \mathbf{q}(0) = \underline{\underline{q}}, \quad \dot{\mathbf{q}}(0) = \underline{\underline{\dot{q}}}$$

The general solution is obtained as superimposition of $\varphi_i e^{j\omega_i t}$ and $\varphi_i e^{-j\omega_i t}$ for all i

$$\mathbf{q} = \sum_{i=1}^n \varphi_i (A_i \cos \omega_i t + B_i \sin \omega_i t)$$

Using $t = 0$

It is easy to verify that

$$\varphi_k^T \mathbf{M}\mathbf{q} = \mu_k (A_k \cos \omega_k t + B_k \sin \omega_k t)$$

$$\varphi_k^T \mathbf{M}\dot{\mathbf{q}} = \omega_k \mu_k (-A_k \sin \omega_k t + B_k \cos \omega_k t)$$

$$A_k = \frac{\varphi_k^T \mathbf{M}\underline{\underline{q}}}{\mu_k},$$

$$B_k = \frac{\varphi_k^T \mathbf{M}\underline{\underline{\dot{q}}}}{\omega_k \mu_k}$$

$$\mathbf{q} = \sum_{i=1}^n \varphi_i \left(\frac{\varphi_k^T \mathbf{M}\underline{\underline{q}}}{\mu_k} \cos \omega_i t + \frac{\varphi_k^T \mathbf{M}\underline{\underline{\dot{q}}}}{\omega_k \mu_k} \sin \omega_i t \right)$$



Computation of the solution to initial conditions

$$\mathbf{q} = \sum_{i=1}^n \varphi_i \left(\frac{\varphi_k^T \mathbf{M} \mathbf{q}}{\mu_k} \cos \omega_i t + \frac{\varphi_k^T \mathbf{M} \dot{\mathbf{q}}}{\omega_k \mu_k} \sin \omega_i t \right)$$

If there are rigid modes for which $\omega_i = 0$ it results

$$\begin{aligned} \cos \omega_i t &\rightarrow 1 \\ \frac{\sin \omega_i t}{\omega_i} &\rightarrow t \end{aligned}$$

$$\mathbf{q} = \sum_{k=1}^r \varphi_k (A_k + B_k t) + \sum_{i=r+1}^n \varphi_i \left(\frac{\varphi_k^T \mathbf{M} \mathbf{q}}{\mu_k} \cos \omega_i t + \frac{\varphi_k^T \mathbf{M} \dot{\mathbf{q}}}{\omega_k \mu_k} \sin \omega_i t \right)$$

Polynomial
response for rigid
body modes



Rayleigh quotient

Consider a generic mode shape \mathbf{u}_i compatible with kinematic BC

$$R(\mathbf{u}_i) = \frac{\mathbf{u}_i^T \mathbf{K} \mathbf{u}_i}{\mathbf{u}_i^T \mathbf{M} \mathbf{u}_i} \rightarrow \mathbf{u}_i = \sum_{k=1}^{\infty} \alpha_k \boldsymbol{\varphi}_k = \boldsymbol{\Phi} \boldsymbol{\alpha}$$

$$\omega_i^2 = \frac{\int E J_2 (w_i'')^2 dx_1}{\int m w_i^2 dx_1} \rightarrow \omega_i^2 = \frac{\boldsymbol{\varphi}_i^T \mathbf{K} \boldsymbol{\varphi}_i}{\boldsymbol{\varphi}_i^T \mathbf{M} \boldsymbol{\varphi}_i}$$

$$R(\mathbf{u}_i) = \frac{\boldsymbol{\alpha}^T \boldsymbol{\Phi}^T \mathbf{K} \boldsymbol{\Phi} \boldsymbol{\alpha}}{\boldsymbol{\alpha}^T \boldsymbol{\Phi}^T \mathbf{M} \boldsymbol{\Phi} \boldsymbol{\alpha}} = \frac{\sum_{k=1}^n \alpha_k^2 \omega_k^2 \mu_k}{\sum_{k=1}^n \alpha_k^2 \mu_k}$$

Through the use of the Rayleigh quotient, it is possible to obtain an approximation of an eigenvalue

The closer \mathbf{u} is to an eigenvector the closer will be the approximation of the eigenvalue

The higher is the separation between the eigenvalues the faster is the convergence

if $\omega_1 > \omega_2 > \dots > \omega_n$ then

$$R(\mathbf{u}_i) = \omega_1 \frac{1 + \sum_{k=2}^n \frac{\alpha_k^2}{\alpha_1^2} \frac{\omega_k^2}{\omega_1^2} \frac{\mu_k}{\mu_1}}{1 + \sum_{k=2}^n \frac{\alpha_k^2}{\alpha_1^2} \frac{\mu_k}{\mu_1}}$$



Rayleigh quotient: error on the eigenvalue

$$R(\mathbf{u}) = \frac{\mathbf{u}^T \mathbf{K} \mathbf{u}}{\mathbf{u}^T \mathbf{M} \mathbf{u}}, \quad \mathbf{u} = \varphi_i + \delta \mathbf{u}$$

Since

$$R(\mathbf{u}) = \frac{\varphi_i^T \mathbf{K} \varphi_i + 2\varphi_i^T \mathbf{K} \delta \mathbf{u} + \delta \mathbf{u}^T \mathbf{K} \delta \mathbf{u}}{\varphi_i^T \mathbf{M} \varphi_i + 2\varphi_i^T \mathbf{M} \delta \mathbf{u} + \delta \mathbf{u}^T \mathbf{M} \delta \mathbf{u}} \quad \delta \mathbf{u} = \sum_j \alpha_j \varphi_j = \Phi \alpha \text{ with } \alpha_i = 0$$

Consequently,

$$\begin{aligned} \varphi_i^T \mathbf{K} \delta \mathbf{u} &= \varphi_i^T \mathbf{K} \Phi \alpha = 0 \\ \varphi_i^T \mathbf{M} \delta \mathbf{u} &= \varphi_i^T \mathbf{M} \Phi \alpha = 0 \end{aligned}$$

$$R(\mathbf{u}) = \frac{\varphi_i^T \mathbf{K} \varphi_i}{\varphi_i^T \mathbf{M} \varphi_i} + O(\|\delta \mathbf{u}\|^2)$$

If the eigenvector has an error of order $O(\delta \mathbf{u})$ then the eigenvalue approximated with the Rayleigh quotient $R(\mathbf{u})$ has an error of order $O(\delta \mathbf{u}^2)$



Computation of a group of eigenvalues/eigenvectors

$$\mathbf{u} = \mathbf{u}_0 + \delta \mathbf{u} \quad \mathbf{u} = \tilde{\Phi} \boldsymbol{\alpha}$$

$$R(\mathbf{u}) = R(\mathbf{u}_0) + O(\|\delta \mathbf{u}\|^2)$$

where $\tilde{\Phi}$ is an approximation of the proper orthogonal modes matrix. If we consider a Taylor expansion it is easy to infer that

$$\frac{dR(\mathbf{u})}{d\mathbf{u}} = 0 \Rightarrow \frac{\partial R(\mathbf{u})}{\partial \boldsymbol{\alpha}} = \frac{dR(\mathbf{u})}{d\mathbf{u}} \frac{d\mathbf{u}}{d\boldsymbol{\alpha}} = \mathbf{0}$$

$$R(\mathbf{u}) = \frac{\mathbf{u}^T \mathbf{K} \mathbf{u}}{\mathbf{u}^T \mathbf{M} \mathbf{u}} = \frac{\boldsymbol{\alpha}^T \tilde{\Phi}^T \mathbf{K} \tilde{\Phi} \boldsymbol{\alpha}}{\boldsymbol{\alpha}^T \tilde{\Phi}^T \mathbf{M} \tilde{\Phi} \boldsymbol{\alpha}} = \frac{\boldsymbol{\alpha}^T \tilde{\mathbf{K}} \boldsymbol{\alpha}}{\boldsymbol{\alpha}^T \tilde{\mathbf{M}} \boldsymbol{\alpha}}$$

$$\frac{\partial R(\mathbf{u})}{\partial \boldsymbol{\alpha}} = \frac{2\tilde{\mathbf{K}}\boldsymbol{\alpha}\boldsymbol{\alpha}^T\tilde{\mathbf{M}}\boldsymbol{\alpha} - 2\boldsymbol{\alpha}^T\tilde{\mathbf{K}}\boldsymbol{\alpha}\tilde{\mathbf{M}}\boldsymbol{\alpha}}{(\boldsymbol{\alpha}^T\tilde{\mathbf{M}}\boldsymbol{\alpha})^2} = \mathbf{0}$$

$$\frac{1}{\boldsymbol{\alpha}^T\tilde{\mathbf{M}}\boldsymbol{\alpha}} \left(\tilde{\mathbf{K}}\boldsymbol{\alpha} - \frac{\boldsymbol{\alpha}^T\tilde{\mathbf{K}}\boldsymbol{\alpha}}{\boldsymbol{\alpha}^T\tilde{\mathbf{M}}\boldsymbol{\alpha}}\tilde{\mathbf{M}}\boldsymbol{\alpha} \right) = \mathbf{0}$$

Since $\boldsymbol{\alpha}^T \tilde{\mathbf{M}} \boldsymbol{\alpha} \neq 0$

$$\tilde{\mathbf{K}}\boldsymbol{\alpha} = \tilde{\lambda}_i \tilde{\mathbf{K}}\boldsymbol{\alpha}$$

Computing the eigenvalues of the reduced and approximated mass and stiffness matrix it is possible to obtain and approx. of eigenvalues



Bloch-Stodola block iteration

① Start with $\tilde{\Phi}_0$

② $k = 0, \dots, n$

$$K \tilde{\Phi}_{k+1} = M \tilde{\Phi}_k \quad (\text{power method})$$

$$\textcircled{3} \quad \bar{K} = \tilde{\Phi}_{k+1}^T K \tilde{\Phi}_{k+1} \quad \bar{M} = \tilde{\Phi}_{k+1}^T M \tilde{\Phi}_{k+1}$$

$$\textcircled{4} \quad \text{solve } \bar{K} \alpha = \lambda \bar{M} \alpha \quad \begin{matrix} \lambda_i \\ \alpha_i \end{matrix}$$

$$\textcircled{5} \quad \tilde{\Phi}_n = \tilde{\Phi}_{k+1} \alpha \quad \text{restart.}$$

