MER

AE 5535 Homework 3

Assigned: 2/22/2021 Due: 2/26/2021

## Variable Area Turbojet Problem

Consider the performance of an ideal non-afterburning turbojet with flow at station 4 (turbine entrance) and station 8 (nozzle throat) choked. A4 is fixed and A8 is varied in order to maintain constant compressor total pressure ratio ( $\pi_C$ ).

On-design conditions are as follows:  $\pi_C = 15$ 

$$M_0 = 2.0$$

$$\tau_{2} = 7.0$$

Find the required ratio of nozzle throat area (off-design) to nozzle throat area (on-design) for the engine operating at the same flight Mach number (2.0) but at the off-design condition such that  $\tau_{\lambda} = 6.0$ .

Homeword 3 VATjet (TC = constant) TC = TCR SO TC = TCR (R = ON-design) POR= TC=1+ nm (Tx) (1-14) Solve for 7+: 7+=1-(TOR-1) TE MNOW that  $T_{+}^{\frac{1}{2}} = AB$   $T_{+}^{\frac{1}{2}} = AB$   $AB = T_{+}^{\frac{1}{2}} = T_{+}^{\frac{1}{2}}$   $ABR = T_{+}^{\frac{1}{2}} = T_{+}^{\frac{1}{2}}$   $T_{+}^{\frac{1}{2}} = T_{+}^{\frac{1}{2}} = T_{+}^{\frac{1}{2}}$ MARINIMANS  $T_{-}^{\frac{1}{2}} = T_{-}^{\frac{1}{2}} = T_{-}^{\frac{1}{2}}$ hae STER= TER = 1-(Ter-1) TOR , THR= THR.

TH= 1-(TOR-1) TO , TH= TH For this problem: TT= 15, 8=1.4, Mar = 2.0, Tar=1 { Mo= 2.0, T1 = 6.0 } off-derign ; [find AB] Trr= 1.8 = T TCR = Tra = 2-168 Ttr= 0.7; Ttr=, 2865; Tt=.6496; TT=, 22  $\int \frac{AB}{ABR} = 1.25$