

$$\frac{dP}{\rho} + \frac{dU}{u} + \frac{dA}{A} = 0$$

$$\frac{dP}{\rho} + u du = - \frac{\tau_w c dA}{\rho A} + \eta \delta \dot{w}$$

$$C_p dT + u du = \delta q + \delta \dot{w}$$

$$\frac{dP}{\rho} = \frac{dP}{\rho} + \frac{dT}{T}$$

$$\frac{dP}{\rho} \Rightarrow \frac{P_i - P_{i-1}}{\rho_i} + \frac{u_i - u_{i-1}}{u_i} + \frac{A_i - A_{i-1}}{A_i} = 0$$

$$\frac{P_i - P_{i-1}}{\rho_i} + u_i (u_i - u_{i-1}) = \frac{-\tau_w c (A_i - A_{i-1})}{\rho_i A_i} + \eta \delta \dot{w}$$

$$C_p (T_i - T_{i-1}) + u_i (u_i - u_{i-1}) = \delta q + \delta \dot{w}$$

$$\frac{P_i - P_{i-1}}{\rho_i} = \frac{P_i - P_{i-1}}{\rho_i} + \frac{T_i - T_{i-1}}{T_i}$$

$$\begin{bmatrix} 1 - P_{i-1} & 1 - u_{i-1} & 0 & 0 \end{bmatrix}$$

$$\begin{bmatrix} P_i \\ u_i \\ P_i \\ T_i \end{bmatrix} \quad \frac{-(A_i - A_{i-1})}{A_i}$$

$$① \quad [J]^k \{x\}^k = \{f\}^k$$

$$② \quad \bar{x}^k = \underline{A} \bar{x}^k + \bar{x}^k$$

Solve ① with Gauss-Seidel

$$\frac{A_1}{A^*} = \frac{A_2}{A^*}$$

$$\frac{A_2}{A_1} = \frac{A_1}{A^*}$$

$$.85 \times 1.09$$

$$1) \frac{A_2}{A_1} = 5.32 \Rightarrow M_2 = 0.99$$

$$2) M_2 = 0.55696$$

$$3) M_2 = 0.35467$$

$$302.4 \text{ K} \quad 5)$$

$$\frac{T_{te}}{T_{ti}} = 4.29$$

$$T_{te} = 1297 \text{ K}$$

$$T_{ti} = 334.656 \text{ K}$$

$$\frac{T_{te}}{T_{ti}} = .5537$$

$$T_{te} = 185.328 \text{ K}$$

$$P_{ti} = 171371 \text{ Pa}$$

$$\frac{P_{te}}{P_{ti}} = .126385$$

$$7) \frac{df}{f} + \frac{du}{u} + \frac{dA}{A} = 0$$

$$\frac{dp}{f} + u du = - \frac{\tau_w \epsilon dx}{f A} + \eta \delta w$$

$$c_p dT + u du = f g + \delta w$$

$$f q_{conv} = C_p C_f (T_w - T_f) \frac{dx}{D}$$

$$\frac{dP}{P} = \frac{df}{f} + \frac{dT}{T}$$

$$\frac{df}{f} + \frac{du}{u} + \frac{dA}{A} = 0$$

$$\frac{dp}{f} + u du = 0$$

$$c_p dT + u du = 0$$

$$\frac{dp}{p} = \frac{df}{f} + \frac{dT}{T} \quad \frac{G(M_i^2)}{G(M_e^2)} \frac{A_e}{A_i} \Rightarrow M_e$$

$$G(M_i^2) = 1.09437$$

$$M_e = .099995$$

$$T_{+i} = \left(1 + \frac{\gamma-1}{2} M_i^2\right) T_{+i} = 316.224 \text{ K}$$

$$P_{+i} = \left(1 + \frac{\gamma-1}{2} M_i^2\right) P_i = 140548.043 \text{ Pa}$$

$$P_e = \frac{P_{+e}}{\left(1 + \frac{\gamma-1}{2} M_e^2\right)^{\gamma/(\gamma-1)}} = 139568.7153 \text{ Pa}$$

$$T_e = \frac{T_{+e}}{1 + \frac{\gamma-1}{2} M_e^2} = 315.593$$

$$u_e = M_e \sqrt{\gamma R T_e} = 35.608 \text{ m/s}$$

$$\begin{aligned} \text{Thrust} &= \left[\dot{m} (u_e - u_i) + P_e A_e - P_i A_i \right] \\ &= \left[29.1699 (35.608 - 238.1219) + 139568.7153 \times .532 \right. \\ &\quad \left. - 101325 \times 1.1 \right] \end{aligned}$$

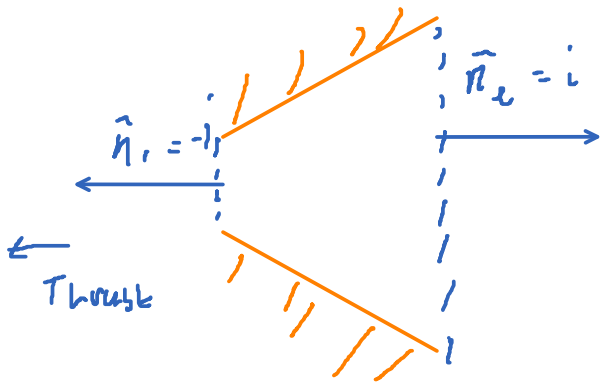
$$= -58210 \text{ N}$$

$$\delta w = 0$$

$$\delta q = 0, \text{ reversible.}$$

$$u_i = M_i \sqrt{\gamma R T_i} = 238.1219 \text{ m/s}$$

$$\dot{m} = \rho u A = 29.1699 \text{ kg/s}$$



$$\{F_x = \int_{CS} \rho \vec{u} \cdot \vec{n} dA = \int_{S_{cv}} p \hat{n} dA + \int_{S_{cv}} \tau_w dA$$

$$\text{Thrust} = - \int_{S_{wet}} p \hat{n} dA - \int_{S_{wet}} \tau_w dA$$

$$\text{Thrust} = \int_{CS} \rho \vec{u} \cdot \vec{n} dA + \int_{CS} p \hat{n} dA$$

in/out

$$\frac{df}{f} + \frac{du}{u} + \frac{dA}{A} = 0$$

$$\frac{dp}{f} + u du = - \frac{\gamma \omega \epsilon dx}{f A} + \cancel{\eta \delta \omega}$$

$$c_p dT + u du = f g + \cancel{\delta \omega}$$

$$\frac{dp}{p} = \frac{df}{f} + \frac{dT}{T}$$

$$f(M_1^2) = \left\{ \frac{1 + \frac{\gamma-1}{2} M_1^2}{(1 + \gamma M_1^2)^2} \right\} M_1^2 = .036175$$

$$T_{t,i} = T_i \left(1 + \frac{\gamma-1}{2} M_1^2 \right) = 290.304$$

$$P_{t,i} = 104190.5846 \text{ Pa}$$

$$\frac{T_{t,e}}{T_{t,i}} = 1 + \frac{\gamma_{1 \rightarrow 2}}{c_p T_{t,i}} = 1 + \frac{1026}{1004.5 \cdot 290.304} = 4.4292$$

$$f(M_e^2) = .160147 \Rightarrow M_e = .55695$$

$$T_{t,e} = 1285.824 \text{ K}$$

$$u_e = M_e \sqrt{\gamma R T_e} = 388.46228$$

$$T_e = \frac{T_{t,e}}{1 + M_e^2 \frac{\gamma-1}{2}} = 1210.709 \text{ K}$$

$$\frac{P_2}{P_1} = \frac{1 + \gamma M_1^2}{1 + \gamma M_2^2} = .7362507 \Rightarrow P_2 = 74600.60408 \text{ Pa}$$

$$P_{t,2} = P_2 \left(1 + \frac{\gamma-1}{2} \right)^{\gamma/(\gamma-1)} = 92095.48146 \text{ Pa}$$

$$\dot{m} = \rho u A = 83.3426 \text{ kg/s} \quad u_i = 68.0348 \text{ m/s}$$

$$+ thrust = - [m(u_e - u_i) + (P_e - P_i) A_e]$$

$$= [8.33426 (388.465 - 68.0348) + (74600.60408 - 101325) \times 0.1]$$

$$= -1.775638$$

N thrust

$$\frac{df}{f} + \frac{du}{u} + \frac{dA}{A} = 0$$

$$\frac{dp}{f} + u du = - \frac{\gamma \omega \epsilon dx}{f A} + \cancel{\gamma \delta \omega}$$

$$c_p dT + u du = \cancel{f g} + \cancel{\delta \omega}$$

$$\frac{dp}{p} = \frac{df}{f} + \frac{dT}{T}$$

$$f_v(M_1^2) = \frac{\gamma+1}{2} \ln \left[\frac{1 + \frac{\gamma-1}{2} M_1^2}{M_1^2} \right] - \frac{1}{M_1^2} = -21.128$$

$$\gamma C_f \frac{CL}{A} = f_{v_2} - f_{v_1}$$

$$\gamma C_f \frac{4}{D} L + f_{v_1} = 5.4327 \quad f_v(M_2^2) \Rightarrow M_2 = .3564$$

$$T_{w1} = T_{w2} = (1 + .2^3) \cdot 288 = 290.304 \text{ K}$$

$$T_2 = \frac{T_{w2}}{(1 + \frac{\gamma-1}{2} M_2^2)} = \frac{290.304}{1 + .2 \cdot .3564^2} = 283.117 \text{ K}$$

$$P_2 = P_1 \frac{M_1}{M_2} \left[\frac{1 + \frac{\gamma-1}{2} M_1^2}{1 + \frac{\gamma-1}{2} M_2^2} \right]^{1/2} = 56375.659 \text{ Pa}$$

$$P_{t2} = P_2 \left(1 + \frac{\gamma-1}{2} M_2^2 \right)^{\gamma/(\gamma-1)} = 61549.488 \text{ Pa}$$

$$u_2 = M_2 \sqrt{\gamma R T_2} = 120.2047 \text{ m/s}$$

$$u_0 = 68.0348 \text{ m/s} \quad \dot{m} = 8.3343 \text{ kg/s}$$

$$thrust = - \left[\dot{m} (u_2 - u_1) + (p_2 - p_1) A_2 \right]$$

$$= [8.3343 \{120.2047 - 68.0348\} + (56375.659 - 101325) \cdot 0.1]$$

$$= 4000.134 \text{ N}$$

$$\frac{df}{f} + \frac{du}{u} + \frac{dA}{A} = 0$$

$$\frac{dp}{f} + u du = - \frac{\tau_w \epsilon dx}{f A} + \cancel{\eta \delta w}$$

=>

$$c_p dT + u du = f g + \cancel{\delta w}$$

$$\frac{dP}{P} = \frac{df}{f} + \frac{dT}{T}$$

$$\frac{df}{f} + \frac{du}{u} + \cancel{\frac{dA}{A}} = 0$$

$$\frac{dp}{f} + u du = - \frac{\tau_w \epsilon dx}{f A}$$

$$c_p dT + u du = f g = 2 c_p c_f (T_w - T_+) \frac{dx}{D}$$

$$\frac{dP}{P} = \frac{df}{f} + \frac{dT}{T}$$

No work interaction or Area variation

$$\frac{df}{f} + \frac{du}{u} + \frac{dA}{A} = 0$$

$$\frac{dp}{f} + u du = \eta \delta w; \eta = 1$$

$$c_p dT + u du = \delta w$$

$$\frac{dp}{p} = \frac{df}{f} + \frac{dT}{T}$$

$$\frac{T_{t2}}{T_{t1}} = 1 + \frac{W_{1 \rightarrow 2}}{c_p T_{t1}}; T_{t1} = T_1 \left(1 + \frac{\gamma-1}{2} M_1^2\right) = (1 + 1.2 \cdot 0.5^2) \cdot 288 = 302.4 \text{ K}$$

$$\frac{T_{t2}}{T_{t1}} = 1 + \frac{10 \text{ kW}}{1004.5 (302.4)} = 4.2921 \Rightarrow T_{t2} = 1297.92026 \text{ K}$$

$$P_{t1} = P_1 \left(1 + \frac{\gamma-1}{2} M_1^2\right)^{\gamma/(\gamma-1)} = 120192.9955 \text{ Pa}$$

$$P_{t2} = P_{t1} \left(\frac{T_{t2}}{T_{t1}}\right)^{\gamma/(\gamma-1)} = 19688418.31 \text{ Pa}$$

$$M_e \Rightarrow P_{t2} \sqrt{\frac{\gamma}{R} \frac{T}{T_{t2}}} M_e = \left(1 + \frac{\gamma-1}{2} M_e^2\right)^{\frac{\gamma(\gamma+1)}{2(\gamma-1)}} = \dot{m} = 20.8357 \text{ kg/s}$$

$$M_e = 1.099703$$

$$P_2 = \frac{P_{t2}}{\left(1 + \frac{\gamma-1}{2} M_e^2\right)^{\gamma/(\gamma-1)}} = 19522651.36 \text{ Pa}$$

$$T_2 = \frac{T_{t2}}{\left(1 + \frac{\gamma-1}{2} M_e^2\right)} = 1294.78848 \text{ K}$$

$$u_2 = M_e \sqrt{\gamma R T_2} = 79.319467 \text{ m/s}$$

$$u_1 = 170.087 \text{ m/s}$$

$$\text{thrust} = \dot{m}(u_e - u_i) + p_e A_e - p_i A_i$$

$$= (8.334(79.319467 - 170.87) + 19640903.66 (.005) - 101325 \cdot 1) = -86717.77616 \text{ N}$$

$$\frac{df}{f} + \frac{du}{u} + \frac{dA}{A} = 0$$

$$\frac{dp}{f} + u du = - \frac{\cancel{\gamma} \omega \cancel{e}}{\cancel{f} A} dx + \eta \delta \omega$$

$$c_p dT + u du = \cancel{\delta q} + \delta w$$

$$\frac{dp}{p} = \frac{df}{f} + \frac{dT}{T}$$

$$\frac{T_{te}}{T_{ti}} = 1 + \frac{\omega_{i \rightarrow e}}{c_p T_{ti}}$$

$$T_{ti} = T_i \left(1 + \frac{\gamma-1}{2} M_i^2\right) = 288 \cdot (1 + 0.2 \cdot 9^2) = 334.654 \text{ K}$$

$$= 1 + \frac{-1.5 \times 10^5}{1004.5 \cdot T_{ti}} = 554 \Rightarrow T_{te} = 185.328 \text{ K}$$

$$P_{ti} = P_i \left(1 + \frac{\gamma-1}{2} M_i^2\right)^{\gamma/(\gamma-1)} = 101325 (1 + 0.2 \cdot 9^2)^{1.4/0.4} = 171257 \text{ Pa}$$

$$\frac{P_{te}}{P_{ti}} = \left(\frac{T_{te}}{T_{ti}}\right)^{1.4/0.4} = 0.12639 \Rightarrow P_{te} = 21658.897 \text{ Pa}$$

$$M_e = P_{te} \sqrt{\frac{\gamma}{R} \frac{T}{T_{te}}} M_{te} = \left(1 + \frac{\gamma-1}{2} M_e^2\right)^{\frac{-(\gamma+1)}{2(\gamma-1)}} \Rightarrow M_e = 0.598$$

$$T_e = T_{te} / \left(1 + \frac{\gamma-1}{2}\right) \Rightarrow T_e = 172.957 \text{ K} \quad P_e = 17007 \text{ N/m}^2$$

$$u_e = M \sqrt{\gamma R T_e} = 157.644 \text{ m/s}$$

$$\text{Thrust} = m(u_e - u_i) + P_e A_e - P_i A_i$$

$$= [37.5042 (157.644 - 306.1567) + 17007 \cdot 0.7 - 101325 \cdot 1] = +3223.148 \text{ N}$$

drag

$$\dot{m} c_p (T_{te} - T_{ti}) =$$

$$\left(\frac{u}{\dot{m}}\right) \dot{m} = u$$

$$\Rightarrow m(p(T_{+1}-T_{-1}))$$

Q7

Tuesday, April 13, 2021 4:06 PM

$$\frac{df}{f} + \frac{du}{u} + \frac{dA}{A} = 0$$

$$\frac{dp}{f} + u du = - \frac{2\epsilon}{f A} dx + \eta \delta w$$

$$c_p dT + u du = 2c_p c_f (T_w - T_+) \frac{dx}{D} + \delta w$$

$$\frac{dp}{p} = \frac{df}{f} + \frac{dT}{T}$$

Work interaction with area variation with irreversibility with convective heat transfer, i.e., all the terms

$$c_p dT_+ = q + w$$

$$q = c_p dT_+ - w$$

$$\dot{Q} = (c_p dT_+ - w) \dot{m}$$