

$$\eta_{c_p} = .85$$

$$M_0 = 8$$

$$\pi_{c_1} = 15$$

$$M_2 = .5$$

Altitude = sea level ($T_0 = 288K$, $P_0 = 101325 \text{ N/m}^2$) A_1 (inlet face area) = 0.00318 m^2 $\pi_s = 1.0$ $\eta_c = .90$ RPM_{design} = 60,000 RPM $T_{14} = 1500K$ $\pi_d = .9904$

$$\tau_r = \frac{T_{12}}{T_0}$$

$$A_1 = A_0 \rightarrow \text{design}$$

$$T_{12} = T_0 \left(1 + \frac{\gamma-1}{2} M_0^2\right)$$

$$P_{12} = P_0 \left(1 + \frac{\gamma-1}{2} M_0^2\right)^{\frac{\gamma}{\gamma-1}} P_0 = P_{t1} \quad P_{t2} = P_{t1} \pi_d$$

$$\dot{m} = \rho u A \quad \tau_c = 1 + \frac{\frac{\gamma-1}{2} M^2}{\eta_c}$$

$$P_{t3} = P_{t4}$$

$$\tau_4 = 1 - \frac{T_{12}}{T_{14}} (\tau_c - 1)$$

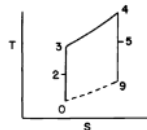


Fig. 5.20 Temperature-entropy diagram.

$$M_1 = 1.0 \Rightarrow T_2 = \dots$$

Mach number=	1	Mach angle=	90	P-M angle=	0
p/p ₀ =	0.52828178	rho/rho ₀ =	0.63393814	T/T ₀ =	0.83333333
p/p* =	1	rho/rho* =	1.00000000	T/T* =	1

$$\pi$$

$$N_{L2}$$

$$\sqrt{0.2}$$

$$T_{+2} = T_{+4} \tau_c$$

$$M_4 = M_8 = 1 \Rightarrow T_8 \Rightarrow P_8$$

Velocities \rightarrow temps. \rightarrow machs \rightarrow

$$\rho_0 u_0 A_0$$

$$\frac{\rho_0}{R T_0} M_0 \sqrt{\gamma R T_0} A_0 = \frac{101325}{287 \cdot 288} \cdot 8 \sqrt{1.4 (287) 288} (0.00318) = \dot{m} = 1.0609 \text{ kg/s}$$

$$\tau_r = 1 + \frac{\gamma-1}{2} M_0^2 = 1 + \frac{1}{2} (8)^2 = 1.128$$

$$T_{t3} = \tau_c T_{t2} = 2.374 (324.864) = 771.227K$$

$$T_{t0} = \tau_r T_0 = 324.864 K = T_{t2}$$

$$P_{t0} = P_0 \tau_r^{\frac{\gamma}{\gamma-1}} = 154453.252 \text{ Pa}$$

$$P_{t2} = \pi_d P_{t0} = 152970.996 \text{ Pa}$$

$$\dot{m}_{corr2} = \frac{\dot{m} \sqrt{T_{t2}/T_{t1p}}}{P_{t2}/P_{t1p}} = 746 \text{ kg/s}$$

$$\tau_c = 1 + \frac{\pi_c \frac{\gamma-1}{2} - 1}{\eta_c}, \quad \eta_c = .85 \Rightarrow \tau_c = 1 + \frac{15 \frac{1.4}{2} - 1}{.85} = 2.374$$

$$T_{14} = 1500$$

$$\frac{T_{t4}}{T_{t2}} = \frac{1500}{324.864} = 4.617$$

$$\tau_t = 1 - \frac{T_{t2}}{T_{t4}} (\tau_c - 1) = 1 - \frac{1}{4.617} (2.374 - 1) = .702$$

$$\pi_t = \left(1 - \frac{1 - \tau_t}{\eta_t}\right)^{\gamma/(\gamma-1)} = \left[1 - \frac{1 - .702}{.9}\right]^{1.4/1.4} = .2448$$

$$T_{t8} = \tau_t T_{t4} = 1053 \text{ K}$$

$$P_{t8} = \pi_t P_{t4} = 561769.446 \text{ Pa}$$

$$P_{t4} = P_{t3} = \pi_c P_{t2} = 2294564.533 \text{ Pa}$$

$$M_8 = 1 \Rightarrow \frac{T_{t8}}{T_8} = 1 + \frac{\gamma-1}{2} M_8^2 \Rightarrow \frac{T_{t8}}{1.2} = T_8 = 2775$$

$$u_8 = M_8 \sqrt{\gamma R T_8} = 593.784 \text{ m/s}$$

$$u_o = M_o \sqrt{\gamma R T_o} = 272.139 \text{ m/s}$$

$$T = \dot{m}(1+F)(u_8 - u_o) + (P_8 - P_o)A_8$$

$$f = \frac{\tau_c (T_{t4}/T_{t2} - \tau_c)}{\frac{h}{c_p T_o}} = .01644 \Rightarrow \dot{m}_f = .0174$$

$$\dot{m} = \rho_8 u_8 A_8 \Rightarrow \frac{\dot{m} R T_8}{P_8 u_8} = .0055 \text{ m}^2$$

$$T = \dot{m}(u_8 - u_o) + (P_8 - P_o)A_8$$

$$A_2 = \frac{\dot{m}}{\rho_2 u_2} = \frac{\dot{m} R T_2}{P_2 M_2 \sqrt{\gamma R T_2}}$$

$$\frac{P_{t2}}{(1 + \frac{\gamma-1}{2} M_2^2)^{\frac{\gamma}{\gamma-1}}} = P_2 = 128957.483 \text{ Pa}$$

$$\frac{T_{t2}}{1 + \frac{\gamma-1}{2} M_2^2} = T_2 = 309.394$$

$$\frac{P_{t8}}{P_8} = \left(1 + \frac{\gamma-1}{2} M_8^2\right)^{\frac{\gamma}{\gamma-1}} \Rightarrow P_8 = 296740.897$$

$$f_g = 1.178$$

$$\rho_2 = \frac{P_2}{RT_2} = 1.452 \text{ kg/m}^3$$

$$u_2 = M_2 \sqrt{\gamma R T_2} = 176.291 \text{ m/s}$$

$$u_3 = u_2$$

$$A_2 = \frac{\dot{m}}{\rho_2 u_2} = 0.00414$$

$$A_4 = \frac{\dot{m}}{\rho_4 u_4}$$

$$\frac{P_{t4}}{(1 + (\gamma - 1)/2)^{\frac{\gamma}{\gamma - 1}}} = P_4 = \frac{2294564.933}{(1.2)^{1.4/1.4}} = 1212176.865 \text{ Pa}$$

$$\frac{T_{t4}}{1.2} = T_4 = 1250 \text{ K}$$

$$\rho_4 = \frac{P_4}{RT_4} = 3.379$$

$$u_4 = \sqrt{\gamma R T_4} = 708.696 \text{ m/s}$$

$$A_4 = \frac{1.0609}{3.379 \cdot 708.696} = 4.43 \text{ E-4 m}^2$$

$$A_3 = \frac{\dot{m}}{\rho_3 u_3} \quad u_2 = u_3$$

$$T_3 = T_{t3} - T_{t2} + T_{t2} = 753.757$$

$$M_3 = \frac{u_3}{\sqrt{\gamma R T_3}} = 3.199$$

$$P_3 = \frac{P_{t3}}{(1 + M_3^2 \cdot 2)^{\frac{\gamma}{\gamma - 1}}} = \frac{2294564.933}{[1 + 2(3.199)^2]^{\frac{1.4}{1.4}}} = 2137470.78 \text{ Pa}$$

$$\rho = \frac{P_3}{RT_3} = 9.855 \text{ kg/m}^3$$

$$A = \frac{1.0609}{9.855} = 6.107 \text{ E-4 m}^2$$

$$\frac{\dot{m} \sqrt{T_{t4}/T_{t5P}}}{P_{t4} \sqrt{P_{t5P}}} = 107$$

$$\frac{N_t}{\sqrt{P_4}} = 26290.683 \quad \frac{N_c}{\sqrt{P_2}} = 56443.268$$

$$\left. \begin{aligned} \dot{m} \sqrt{P_2} \frac{1}{\pi c} \frac{N_c}{\sqrt{P_2}} &= 2809 \\ \dot{m}_{corr4} \frac{N_t}{\sqrt{P_4}} &= 2810 \end{aligned} \right\}$$

$$\dot{m}_{corr8} = \frac{\dot{m} \sqrt{T_{t8}/T_{t5T3}}}{P_{t3}/P_{t5P}} = 366$$

$$\frac{\dot{m}_{corr8}}{\dot{m}_{corr2}} = \frac{366}{221.771} = 1.65$$

$$\frac{\sqrt{T_{t4}/T_{t2}} \cdot \frac{1}{\pi c P_{t2}}}{\sqrt{T_{t4}/T_{t2}} \cdot \frac{1}{\pi c P_{t2}}} = 1.491$$

$$\rightarrow RT_3$$

$$A_3 = \frac{1.0609}{9.855 \cdot 176.291} = 6.107E-4 \text{ m}^2$$

$$A_5 = \frac{\dot{m}}{\rho_5 u_5}$$

$$u_5 = 0.8 u_4 = 566.957$$

$$\frac{.18 u_4^2}{C_p} T_4 + T_{+5} - T_{+4} = 260.6 \text{ K} \quad T_5 =$$

$$M_5 = \frac{u}{\sqrt{\gamma R T_5}} = .9641$$

$$P_5 = \frac{P_4}{(1 + \frac{\gamma-1}{2} M_4^2)^{\frac{\gamma}{\gamma-1}}} = 309258.738 \text{ Pa}$$

$$\rho_5 = \frac{P_5}{RT} = 1.252 \text{ Kg/m}^3$$

$$A_5 \approx .001494 \text{ m}^2 \Rightarrow A_8 = .001492 \text{ m}^2$$