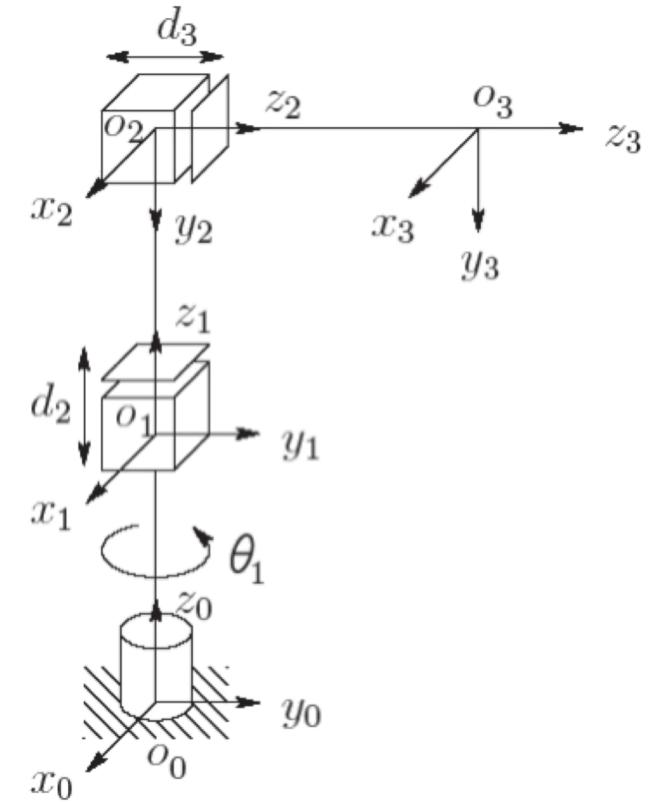


Robot Dynamics

Example: three link cylindrical robot

- Up to this point, we have developed a systematic method to determine the forward and inverse kinematics and the Jacobian for any arbitrary serial manipulator
 - Forward kinematics: mapping from joint variables to position and orientation of the end effector
 - Inverse kinematics: finding joint variables that satisfy a given position and orientation of the end effector
 - Jacobian: mapping from the joint velocities to the end effector linear and angular velocities
- Example: three link cylindrical robot



General system overview

- Ex: position control...

Dynamics Overview

- We want to come up with equations of motion for any n DOF system
 - In general, this will consist of n coupled second order ODEs
- These systems may be:
 - Linear or nonlinear
 - Conservative or nonconservative
- We want to develop an expression of the form: $\dot{q} = f(q, t)$
- Once we have this, we can use it to choose an appropriate controller that will put our dynamical system in a desired state (configuration)

Euler-Lagrange Equations

- We can derive the equations of motion for any n DOF system by using energy methods
 - All we need to know are the conservative (kinetic and potential) and non-conservative (dissipative) terms
- This is a shortcut to describing the motion of each particle in a rigid body along with the constraints that form rigid motions
- For this, we need to first use virtual displacements subject to holonomic constraints, then use the principle of virtual work, then finally use D'Alembert's Principle to derive the Euler-Lagrange equations of motion
- But first, an example...

Ex: 1DOF system

- To illustrate, we derive the equations of motion for a 1DOF system
 - Consider a particle of mass m
 - Using Newton's second law:

$$m\ddot{y} = f - mg$$

- Now define the kinetic and potential energies:

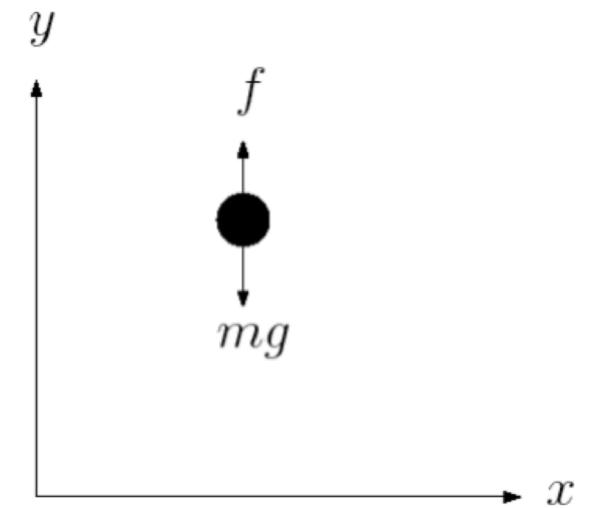
$$\underline{K = \frac{1}{2}m\dot{y}^2} \quad \underline{P = mgy}$$

- Rewrite the above differential equation
 - Left side:

$$m\ddot{y} = \frac{d}{dt}(m\dot{y}) = \frac{d}{dt} \frac{\partial}{\partial \dot{y}} \left(\frac{1}{2}m\dot{y}^2 \right) = \frac{d}{dt} \frac{\partial K}{\partial \dot{y}}$$

- Right side:

$$mg = \frac{\partial}{\partial y}(mgy) = \frac{\partial P}{\partial y}$$



$$\frac{\frac{\partial}{\partial \dot{y}} \left(\frac{1}{2}m\dot{y}^2 \right)}{\frac{\partial}{\partial y}(mgy)} = \frac{\ddot{y}}{\dot{y}} = \frac{m\ddot{y}}{m\dot{y}} =$$

Ex: 1DOF system

- Thus we can rewrite the initial equation:

$$\frac{d}{dt} \frac{\partial K}{\partial \dot{y}} = f - \frac{\partial P}{\partial y}$$

- Now we make the following definition:

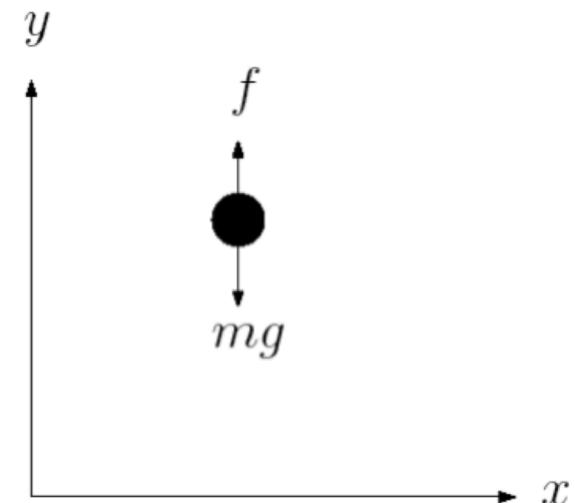
$$\underline{L = K - P}$$

- L is called the *Lagrangian*

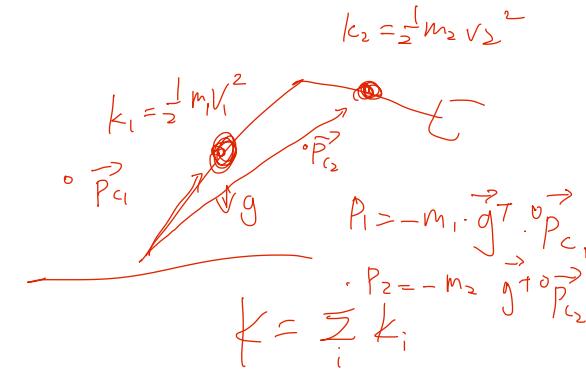
- We can rewrite our equation of motion again:

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{y}} - \frac{\partial L}{\partial y} = f$$

- Thus, to define the equation of motion for this system, all we need is a description of the potential and kinetic energies



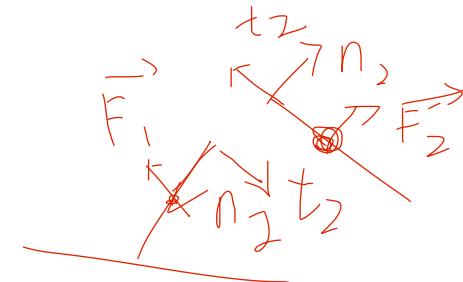
Euler-Lagrange Equations



- If we represent the variables of the system as generalized coordinates, then we can write the equations of motion for an n DOF system as:

$$\frac{d}{dt} \frac{\partial L}{\partial \dot{q}_i} - \frac{\partial L}{\partial q_i} = \tau_i$$

- We will come back to this, but it is important to recognize the form of the above equation:
 - The left side contains the conservative terms
 - The right side contains the non-conservative terms
- This formulation leads to a set of n coupled 2nd order differential equations



Force and Torque relation

Thus we have

$$\mathcal{F} \cdot \delta \mathcal{X} = \tau \cdot \delta \Theta, \quad (5.91)$$

where \mathcal{F} is a 6×1 Cartesian force-moment vector acting at the end-effector, $\delta \mathcal{X}$ is a 6×1 infinitesimal Cartesian displacement of the end-effector, τ is a 6×1 vector of torques at the joints, and $\delta \Theta$ is a 6×1 vector of infinitesimal joint displacements. Expression (5.91) can also be written

$$\mathcal{F}^T \delta \mathcal{X} = \tau^T \delta \Theta. \quad (5.92)$$

The definition of the Jacobian is

$$\delta \mathcal{X} = J \delta \Theta, \quad (5.93)$$

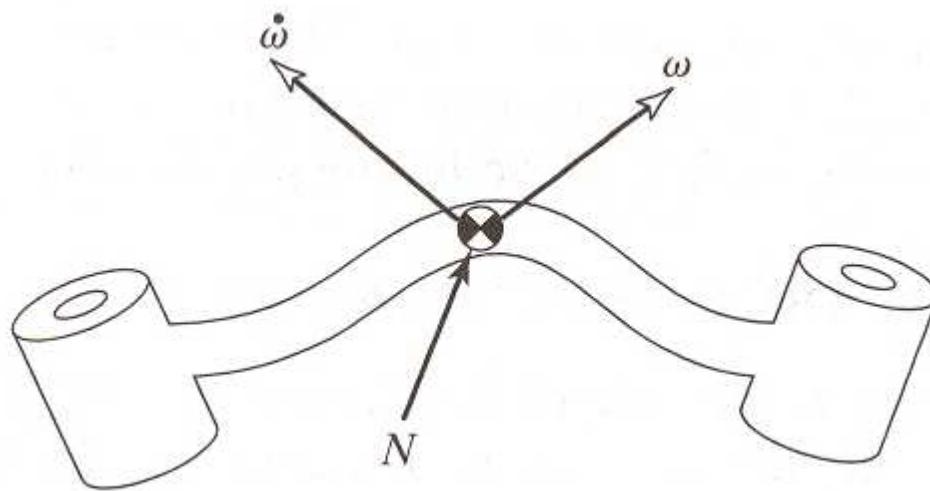
and so we may write

$$\mathcal{F}^T J \delta \Theta = \tau^T \delta \Theta. \quad (5.94)$$

which must hold for all $\delta \Theta$, and so we have

$$\mathcal{F}^T J = \tau^T. \quad (5.95)$$

Motion of the center of mass



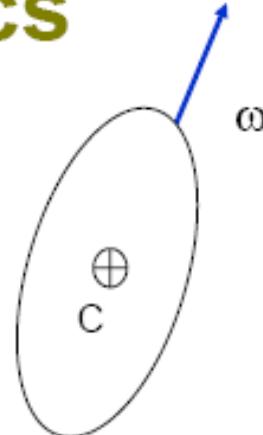
Newton-Euler Equations

$$\vec{P} = m \vec{v}$$
$$\vec{F} = m \vec{a}$$
$$K = \frac{1}{2} m v^2$$

Rigid Body Dynamics

Newton's law: $\sum F = m \mathbf{a}_c$

Euler's Equation: $\sum \mathbf{M}_c = \dot{\mathbf{H}}_c$



The rate of change of the angular momentum is given by

$$\dot{\mathbf{H}}_c = \boldsymbol{\omega} \times \mathbf{I}_c \boldsymbol{\omega} + \mathbf{I}_c \dot{\boldsymbol{\omega}}$$



Velocities and accelerations

Linear acceleration

$$A\vec{V}_Q = \underline{\underline{B}}_R^A \vec{V}_Q + \underline{\underline{B}}_R^A \underline{\Omega}_B Q$$

Rotation
W angle ν

We start by restating (5.12), an important result from Chapter 5, which describe the velocity of a vector ${}^B Q$ as seen from frame $\{A\}$ when the origins are coincident

$${}^A V_Q = {}_B R^A V_Q + {}^A \Omega_B \times {}_B R^A Q. \quad (6.5)$$

The left-hand side of this equation describes how ${}^A Q$ is changing in time. So, because origins are coincident, we could rewrite (6.5) as

$$\frac{d}{dt}({}_B R^A Q) = {}_B R^A V_Q + {}^A \Omega_B \times {}_B R^A Q. \quad (6.6)$$

This form of the equation will be useful when deriving the corresponding acceleration equation.

By differentiating (6.5), we can derive expressions for the acceleration of ${}^B Q$ as viewed from $\{A\}$ when the origins of $\{A\}$ and $\{B\}$ coincide:

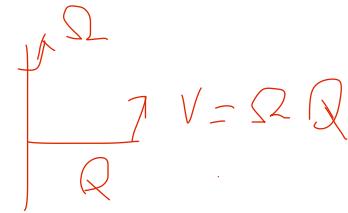
$${}^A \dot{V}_Q = \frac{d}{dt}({}_B R^A V_Q) + {}^A \dot{\Omega}_B \times {}_B R^A Q + {}^A \Omega_B \times \frac{d}{dt}({}_B R^A Q). \quad (6.7)$$

Now we apply (6.6) twice—once to the first term, and once to the last term. The right-hand side of equation (6.7) becomes

$$\begin{aligned} {}_B R^A \dot{V}_Q &+ {}^A \Omega_B \times {}_B R^A V_Q + {}^A \dot{\Omega}_B \times {}_B R^A Q \\ &+ {}^A \Omega_B \times ({}_B R^A V_Q + {}^A \Omega_B \times {}_B R^A Q). \end{aligned} \quad (6.8)$$

Combining two terms, we get

$${}_B R^A \dot{V}_Q + 2 {}^A \Omega_B \times {}_B R^A V_Q + {}^A \dot{\Omega}_B \times {}_B R^A Q + {}^A \Omega_B \times ({}^A \Omega_B \times {}_B R^A Q). \quad (6.9)$$



Finally, to generalize to the case in which the origins are not coincident, we add one term which gives the linear acceleration of the origin of $\{B\}$, resulting in the final general formula:

$$\begin{aligned} {}^A \dot{V}_{BORG} &+ {}_B^A R {}^B \dot{V}_Q + 2{}^A \Omega_B \times {}_B^A R {}^B V_Q + {}^A \dot{\Omega}_B \times {}_B^A R {}^B Q \\ &+ {}^A \Omega_B \times ({}^A \Omega_B \times {}_B^A R {}^B Q). \end{aligned} \quad (6.10)$$

A particular case that is worth pointing out is when ${}^B Q$ is constant, or

$${}^B V_Q = {}^B \dot{V}_Q = 0. \quad (6.11)$$

In this case, (6.10) simplifies to

$${}^A \dot{V}_Q = {}^A \dot{V}_{BORG} + {}^A \Omega_B \times ({}^A \Omega_B \times {}_B^A R {}^B Q) + {}^A \dot{\Omega}_B \times {}_B^A R {}^B Q. \quad (6.12)$$

We will use this result in calculating the linear acceleration of the links of a manipulator with rotational joints. When a prismatic joint is present, the more general form of (6.10) will be used.

$${}^A\Omega_B + {}^A\tilde{R} {}^B\Omega_C = {}^A\Omega_C$$

Angular acceleration

Consider the case in which $\{B\}$ is rotating relative to $\{A\}$ with ${}^A\Omega_B$ and $\{C\}$ is rotating relative to $\{B\}$ with ${}^B\Omega_C$. To calculate ${}^A\Omega_C$, we sum the vectors in frame $\{A\}$:

$${}^A\Omega_C = {}^A\Omega_B + {}^A_R {}^B\Omega_C. \quad (6.13)$$

By differentiating, we obtain

$${}^A\dot{\Omega}_C = {}^A\dot{\Omega}_B + \frac{d}{dt}({}^A_R {}^B\Omega_C). \quad (6.14)$$

Now, applying (6.6) to the last term of (6.14), we get

$${}^A\dot{\Omega}_C = {}^A\dot{\Omega}_B + {}^A_R {}^B\dot{\Omega}_C + {}^A\Omega_B \times {}^A_R {}^B\Omega_C. \quad (6.15)$$

We will use this result to calculate the angular acceleration of the links of a manipulator.

Newton-Euler Quations

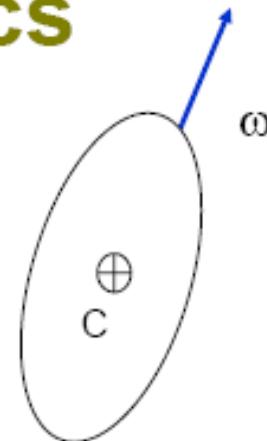
Rigid Body Dynamics

Newton's law:

$$\sum F = m\mathbf{a}_c$$

Euler's Equation:

$$\sum \mathbf{M}_c = \dot{\mathbf{H}}_c$$



The rate of change of the angular momentum is given by

$$\dot{\mathbf{H}}_c = \boldsymbol{\omega} \times \mathbf{I}_c \boldsymbol{\omega} + \mathbf{I}_c \dot{\boldsymbol{\omega}}$$

change direction

change magr

Angular Momentum

Angular Momentum

Definition: Angular momentum is the total moment of the momentum of a rigid body's constituting particles:

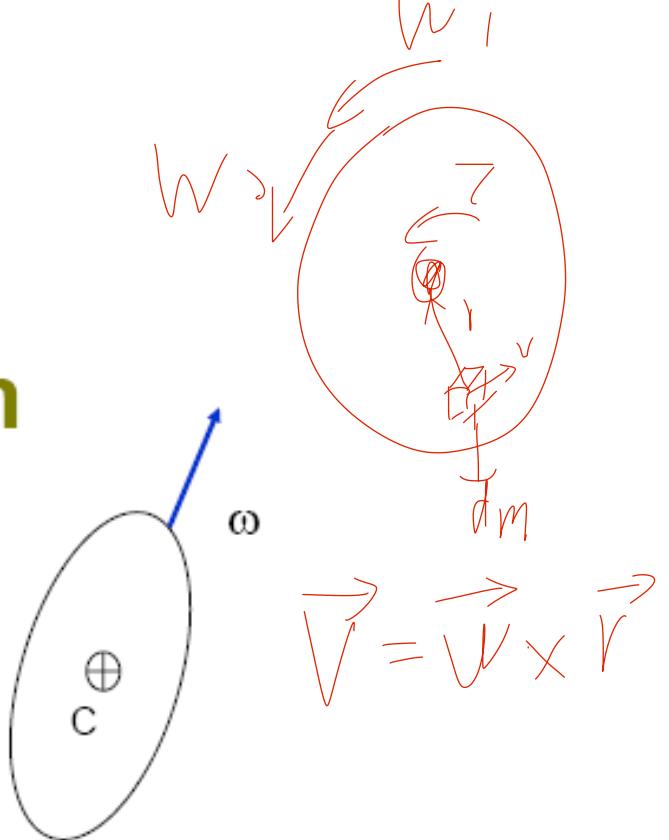
$$\mathbf{H}_c = \int_{\mathcal{B}} \mathbf{r} \times \mathbf{v} dm = \mathbf{I}_c \boldsymbol{\omega}$$

And \mathbf{I}_c is the intertia matrix given by $\mathbf{I}_c = \int \mathbf{S}(\mathbf{r})^T \mathbf{S}(\mathbf{r}) dm$

$$\mathbf{I}_c = \begin{bmatrix} I_{xx} & -I_{xy} & -I_{xz} \\ -I_{xy} & I_{yy} & -I_{yz} \\ -I_{xz} & -I_{yz} & I_{zz} \end{bmatrix}$$

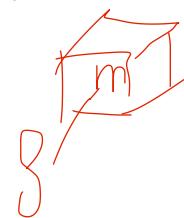
$$\vec{r} \times \vec{v} = \vec{r} \times \vec{w} \times \vec{r}$$

$$I_{xx} = \int_{\mathcal{B}} (y^2 + z^2) dm \quad I_{xy} = \int_{\mathcal{B}} xy dm$$
$$I_{yy} = \int_{\mathcal{B}} (x^2 + z^2) dm \quad I_{xz} = \int_{\mathcal{B}} xz dm$$
$$I_{zz} = \int_{\mathcal{B}} (x^2 + y^2) dm \quad I_{yz} = \int_{\mathcal{B}} yz dm$$



Inertial Matrix

$$\begin{aligned}
 I_{xx} &= \int_0^h \int_0^l \int_0^\omega (y^2 + z^2) \rho \, dx \, dy \, dz \\
 &= \int_0^h \int_0^l (y^2 + z^2) \omega \rho dy \, dz \\
 &= \int_0^h \left(\frac{l^3}{3} + z^2 l \right) \omega \rho dz \\
 &= \left(\frac{hl^3 \omega}{3} + \frac{h^3 l \omega}{3} \right) \rho \\
 &= \underline{\underline{\frac{m}{3}(l^2 + h^2)}},
 \end{aligned}$$



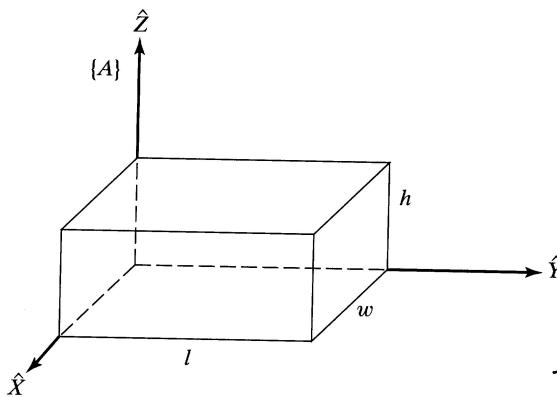
$$\begin{aligned}
 \overline{I}_{xx} &= \int (y^2 + z^2) dm \\
 m &= \rho V \\
 dm &= \rho dV = \rho dx dy dz
 \end{aligned}$$

where m is the total mass of the body. Permuting the terms, we can get I_{yy} and I_{zz} by inspection:

$$\underline{\underline{I_{yy} = \frac{m}{3}(\omega^2 + h^2)}}$$

and

$$\underline{\underline{I_{zz} = \frac{m}{3}(l^2 + \omega^2)}}.$$



$$\begin{aligned}
 I_{xy} &= \int_0^h \int_0^l \int_0^\omega xy\rho dx dy dz \\
 &= \int_0^h \int_0^l \frac{\omega^2}{2} y\rho dy dz \\
 &= \int_0^h \frac{\omega^2 l^2}{4} \rho dz \\
 &= \underline{\underline{\frac{m}{4}\omega l}}.
 \end{aligned}$$

Permuting the terms, we get

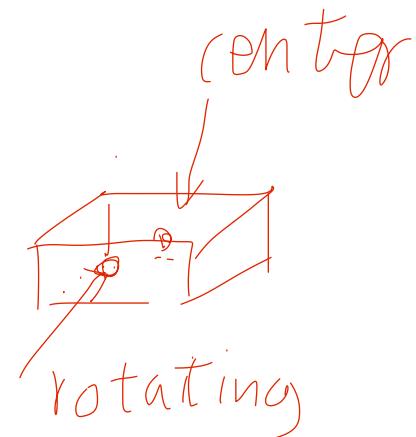
$$\underline{\underline{I_{xz} = \frac{m}{4}h\omega}}$$

and

$$\underline{\underline{I_{yz} = \frac{m}{4}hl.}}$$

Hence, the inertia tensor for this object is

$${}^A I = \begin{bmatrix} \frac{m}{3}(l^2 + h^2) & -\frac{m}{4}\omega l & -\frac{m}{4}h\omega \\ -\frac{m}{4}\omega l & \frac{m}{3}(\omega^2 + h^2) & -\frac{m}{4}hl \\ -\frac{m}{4}h\omega & -\frac{m}{4}hl & \frac{m}{3}(l^2 + \omega^2) \end{bmatrix}.$$



Find the inertia tensor for the same solid body described for Example 6.1 when it is described in a coordinate system with origin at the body's center of mass.

We can apply the parallel axis theorem, (6.25), where

$$\begin{bmatrix} x_c \\ y_c \\ z_c \end{bmatrix} = \frac{1}{2} \begin{bmatrix} u \\ l \\ h \end{bmatrix}$$

Then we find

$$\begin{aligned} {}^C I_{zz} &= \frac{m}{12} (u^2 + l^2), \\ (6.27) \end{aligned}$$

$${}^C I_{xy} = 0.$$

The other elements are found by symmetry. The resulting inertia tensor written in the frame at the center of mass is

$${}^C I = \begin{bmatrix} \frac{m}{12} (h^2 + l^2) & 0 & 0 \\ 0 & \frac{m}{12} (u^2 + h^2) & 0 \\ 0 & 0 & \frac{m}{12} (l^2 + u^2) \end{bmatrix}. \quad (6.28)$$

Since the result is diagonal, frame $\{C\}$ must represent the principal axes of this body. ■



- Inertia matrix is a positive-definite symmetric matrix
- The inertia matrix is not in general constant and is frame-dependent:
$$\mathbf{I}_c^i = R_i^T \mathbf{I}_c R_i$$
- Any rigid body has a set of principal directions with respect to which the inertia matrix is diagonal

Example

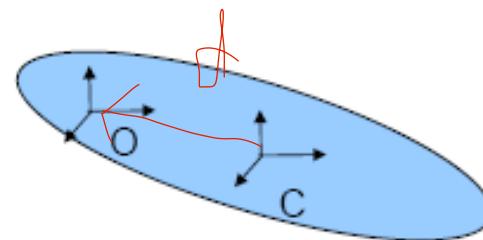
Special cases

- If xy is the plane of symmetry, then $I_{xz}=I_{yz}=0$. Similarly if xz or yz is the plane of symmetry then the corresponding products of inertia are zero.
- If body is axisymmetric (e.g., symmetric about z) then the inertia matrix is diagonal and 2 of the moments of inertia equal (e.g., $I_{xx}=I_{yy}$ if z is the axis of symmetry)



Parallel-Axis Theorem

The moment of inertia about an arbitrary point O with axes parallel to the original xyz is given by



$$\mathbf{I}_o = \mathbf{I}_c + m\mathbf{S}(\mathbf{d})^T \mathbf{S}(\mathbf{d}) = \begin{bmatrix} I_{xx} + md_x^2 & -(I_{xy} + md_x d_y) & -(I_{xz} + md_x d_z) \\ * & I_{yy} + md_y^2 & -(I_{yz} + md_y d_z) \\ * & * & I_{zz} + md_z^2 \end{bmatrix}$$

I_{xy} + md_x · d_y

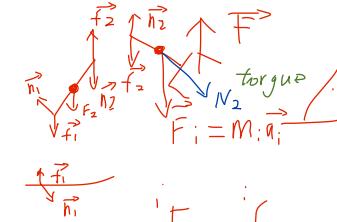
Inward Calculations

$$\left. \begin{array}{l} f_{i+1} \\ n_{i+1} \end{array} \right\} \text{large } =$$

$$^i F_i = ^i f_i - {^i R^{i+1}} f_{i+1}.$$

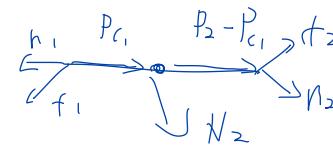
$$F_2 = f_1 - f_2$$

$$\vec{F} = (\vec{F})$$

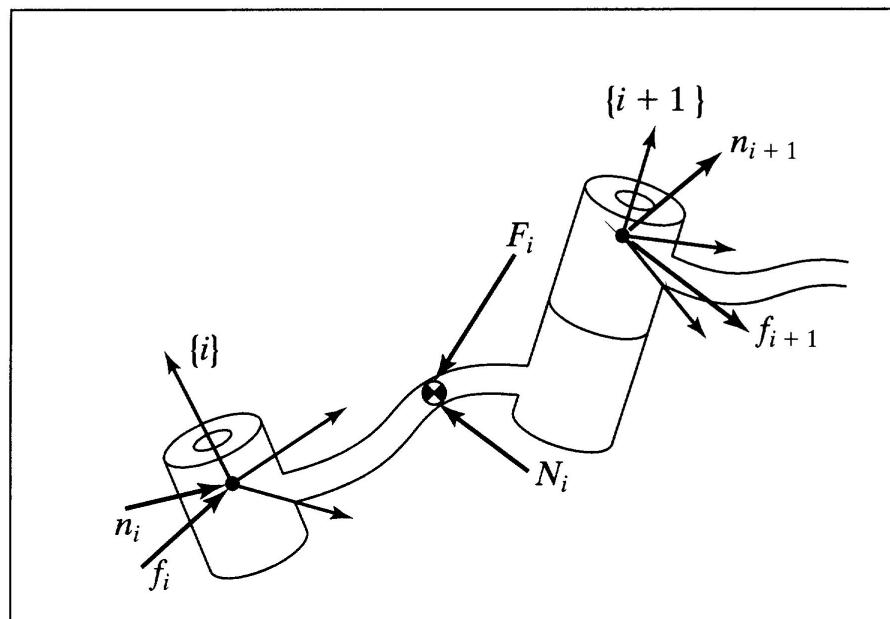


$$F_i = f_i - {^i R^{i+1}} f_{i+1}$$

$$N_2 = H_2 \quad \vec{n}_1 = \vec{n}_2 + N_2 +$$



$$^i N_i = ^i n_i - {^i n_{i+1}} + (-{^i P_{C_i}}) \times {^i f_i} - ({^i P_{i+1}} - {^i P_{C_i}}) \times {^i f_{i+1}}.$$



Iterative Closed Loop Form (Newton-Euler)

Outward iterations: $i : 0 \rightarrow 5$

cheat paper

$${}^{i+1}\omega_{i+1} = {}^i{}_i R {}^i\omega_i + \dot{\theta}_{i+1} {}^{i+1}\hat{Z}_{i+1},$$

$${}^{i+1}\dot{\omega}_{i+1} = {}^i{}_i R {}^i\dot{\omega}_i + {}^i{}_i R {}^i\omega_i \times \dot{\theta}_{i+1} {}^{i+1}\hat{Z}_{i+1} + \ddot{\theta}_{i+1} {}^{i+1}\hat{Z}_{i+1},$$

$${}^{i+1}\dot{v}_{i+1} = {}^i{}_i R ({}^i\dot{\omega}_i \times {}^i P_{i+1} + {}^i\omega_i \times ({}^i\omega_i \times {}^i P_{i+1}) + {}^i\dot{v}_i),$$

$${}^{i+1}\dot{v}_{C_{i+1}} = {}^{i+1}\dot{\omega}_{i+1} \times {}^{i+1}P_{C_{i+1}}$$

$$+ {}^{i+1}\omega_{i+1} \times ({}^{i+1}\omega_{i+1} \times {}^{i+1}P_{C_{i+1}}) + {}^{i+1}\dot{v}_{i+1},$$

$$\begin{aligned} {}^0\vec{J}_0 &= 0 & {}^0\vec{V}_0 &= 0 \\ {}^0\vec{V}_0 &\neq \cancel{{}^0\vec{W}_0} & {}^0\vec{W}_0 &= 0 \end{aligned}$$

$${}^{i+1}F_{i+1} = m_{i+1} {}^{i+1}\dot{v}_{C_{i+1}},$$

$${}^{i+1}N_{i+1} = {}^{C_{i+1}}I_{i+1} {}^{i+1}\dot{\omega}_{i+1} + {}^{i+1}\omega_{i+1} \times {}^{C_{i+1}}I_{i+1} {}^{i+1}\omega_{i+1}.$$

Inward iterations: $i : 6 \rightarrow 1$

$${}^i f_i = {}^i{}_{i+1} R {}^{i+1}f_{i+1} + {}^i F_i,$$

$$\begin{aligned} {}^i n_i &= {}^i N_i + {}^i{}_{i+1} R {}^{i+1}n_{i+1} + {}^i P_{C_i} \times {}^i F_i \\ &\quad + {}^i P_{i+1} \times {}^i{}_{i+1} R {}^{i+1}f_{i+1}, \end{aligned}$$

$$\tau_i = {}^i n_i^T {}^i \hat{Z}_i.$$

x和y的力矩会有机器人本体承受

只对旋转节有效