

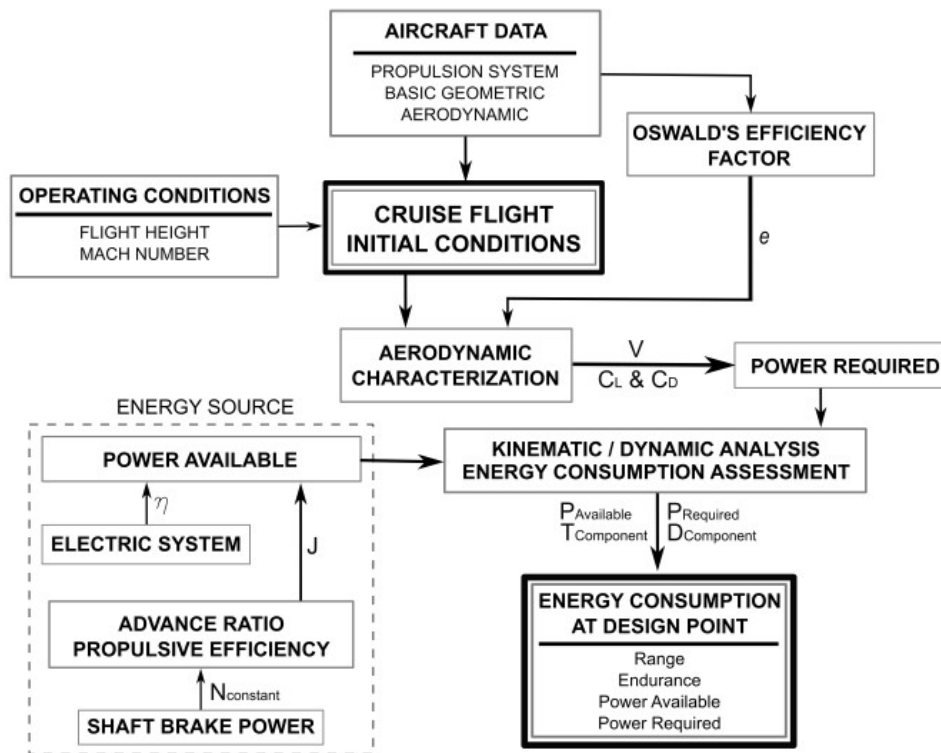
## Design and Fabrication of a Long Endurance DEP Aircraft using Battery Power Source / Fuel Cell Tech

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Target: Long endurance: 3 - 4 hours of flight using batteries/ Fuel Cell

Use: Smart agriculture, aerial surveillance

Process flow chart of the general drone aerostructural assessment



## Conceptual Design Phase

This is the early stage where broad design choices are made.

### A. Mission Requirements

- Define purpose: passenger transport, cargo, military, surveillance, etc.
- Payload, range, speed, altitude, field performance.
- Operational environment (e.g., civil airports, carrier operations, rough fields).

## B. Initial Configuration

- Select configuration: wing placement, tail type, fuselage shape.
- Choose engine type: turbofan, turboprop, piston, electric, etc.
- Estimate weight and size.
- Use empirical formulas or software (e.g., NASA's OpenVSP) to get initial shapes.

## C. Performance Estimation

- Use handbook methods or MATLAB/Python tools to estimate:
  - Lift-to-drag ratio (L/D)
  - Thrust-to-weight ratio (T/W)
  - Wing loading (W/S)
- Constraint analysis: Draw matching graphs to determine design point.

## D. Feasibility Studies

- Trade studies for design alternatives.
- Cost vs. performance vs. manufacturability.
- Initial compliance with aviation regulations (FAA/EASA).

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# 2. Preliminary Design Phase

Detailed analysis of selected configuration.

## A. Aerodynamic Design

- Airfoil selection for wings/tails.
- 3D wing design (span, sweep, dihedral, twist).
- High-lift devices: flaps, slats.
- Computational Fluid Dynamics (CFD) simulations.

## B. Weight Estimation

- Detailed breakdown:
  - Structure
  - Propulsion
  - Avionics
  - Payload
  - Fuel
- Center of Gravity (CG) range analysis.

## C. Propulsion Integration

- Engine performance modeling (e.g., thrust lapse rate).
- Inlet and nozzle design.
- Fuel system planning.

## D. Structural Design

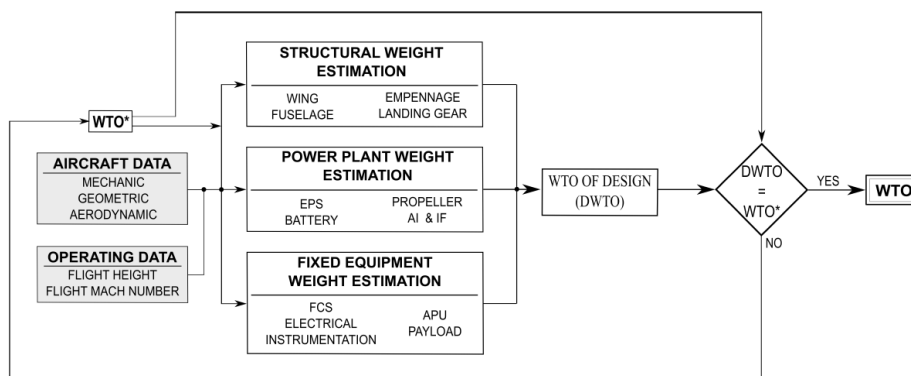
- Material selection (aluminum, composites, titanium).
- Load paths and stress analysis (e.g., using Finite Element Method - FEM).
- Design of fuselage frames, stringers, wing spars, and ribs.

## E. Stability and Control

- Static and dynamic stability.
- Control surface sizing.
- Longitudinal, lateral, and directional handling qualities.

## F. Systems Layout

- Avionics placement.
- Electrical and hydraulic systems.
- Landing gear configuration and retraction mechanism.



# 3. Detailed Design Phase

Transition from concept to manufacturable parts.

## A. CAD Modeling

- Full 3D solid modeling (e.g., CATIA, Siemens NX, SolidWorks).
- Detailed geometry for all parts and assemblies.

- Interference and tolerance checks.

## **B. System Integration**

- Wire harness routing.
- Integration of sensors, actuators, and software.
- Redundancy and fault tolerance analysis.

## **C. Manufacturing Planning**

- Material procurement.
- Tooling and jig design.
- Assembly procedures.

## **D. Documentation**

- Engineering drawings.
  - Bill of Materials (BOM).
  - Maintenance and operation manuals.
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# **4. Testing and Validation**

## **A. Ground Testing**

- Structural tests: static, fatigue, crashworthiness.
- Systems checks: hydraulics, avionics, landing gear.
- Engine run-up tests.

## **B. Wind Tunnel Testing**

- Low- and high-speed tests of scale models.
- Validation of CFD results.

## **C. Flight Testing**

- Envelope expansion: handling at various speeds/altitudes.
  - Performance and stability verification.
  - Certification testing (FAA Part 23/25, MIL-SPECs, etc.).
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# **5. Certification & Production**

## **A. Certification**

- Regulatory compliance (e.g., FARs, CS-25).
- Design data approval.
- Environmental certification (noise, emissions).

## **B. Production Readiness**

- Supply chain alignment.
- Quality assurance setup.
- Final assembly line layout.

## **C. Support Planning**

- Training materials for pilots and maintenance.
- Spare parts logistics.
- In-service support documentation.



## **6. Post-Production & Upgrades**

- Operational feedback and incident analysis.
- Mid-life upgrades: avionics, engines, interiors.
- Variant design: stretched versions, cargo models, etc.

# 1. Conceptual Design Phase – Common Problems

Area	Problem	Explanation
Requirements	<b>Unclear or changing requirements</b>	Stakeholders (military, airline, etc.) may modify specs mid-process, impacting the entire design.
Configuration	<b>Overly ambitious design</b>	Choosing novel or unproven layouts (e.g., blended wing body) can increase complexity and risk.
Sizing	<b>Incorrect weight or thrust estimates</b>	Early estimations based on assumptions can lead to underpowered or overweight aircraft.
Trade Studies	<b>Inadequate trade-off analysis</b>	Failing to explore enough alternatives can lock the project into a suboptimal path.
Regulations	<b>Neglecting early regulatory input</b>	Not accounting for certification requirements (e.g., FAR 25) can lead to major redesigns later.

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## 2. Preliminary Design Phase – Common Problems

Area	Problem	Explanation
Aerodynamics	<b>CFD results that contradict wind tunnel data</b>	CFD is sensitive to mesh and turbulence models, which can lead to misleading results.
Structural Design	<b>Weight growth</b>	Structural reinforcements add weight, which snowballs across the design (e.g., more lift → larger wings → more weight).
Propulsion	<b>Engine-airframe integration issues</b>	Engine placement can affect aerodynamics, noise, and maintenance access.
Stability & Control	<b>Poor dynamic stability</b>	Improper tail sizing or CG location can result in oscillatory or unstable flight characteristics.
Systems Layout	<b>Space conflicts</b>	Subsystems may compete for space or interfere with structural members, especially in tightly packed designs.

Area	Problem	Explanation
Human Factors	<b>Neglecting cockpit ergonomics</b>	Early designs sometimes miss pilot visibility, reachability, or workload issues.

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### 3. Detailed Design Phase – Common Problems

Area	Problem	Explanation
CAD Models	<b>Integration mismatches</b>	Subsystems designed by different teams might not fit together due to tolerance or miscommunication.
Systems Integration	<b>Wiring &amp; plumbing clashes</b>	Routing electrical, hydraulic, and fuel lines through confined spaces is prone to errors.
Manufacturing	<b>Designs that are hard to build</b>	Sharp corners, complex curves, or non-standard materials increase tooling complexity and cost.
Documentation	<b>Incomplete or inconsistent drawings</b>	Errors in BOMs or drawings lead to rework and delays on the shop floor.
Supply Chain	<b>Unavailable or long-lead parts</b>	Relying on rare materials or components not in production can cause critical path delays.

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### 4. Testing and Validation – Common Problems

Area	Problem	Explanation
Ground Tests	<b>Test article failures</b>	Components may fail under static or fatigue loads due to overlooked stress concentrations.
Wind Tunnel	<b>Scale model issues</b>	Scaling laws (Reynolds number, Mach number) may not represent full-scale behavior well.
Flight Test	<b>Unexpected behavior</b>	Aircraft may display flutter, control reversal, or handling problems not predicted in simulations.
Certification	<b>Noncompliance with regulations</b>	Overlooking specific certification rules can require costly retrofits or testing redo.

Area	Problem	Explanation
Safety	<b>Inadequate fail-safes</b>	Failure to simulate critical failures (engine-out, fire) can reveal last-minute safety gaps.

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## 5. Certification & Production – Common Problems

Area	Problem	Explanation
Certification	<b>Lack of documentation or traceability</b>	Authorities require detailed compliance documentation and testing traceability.
Quality Control	<b>Manufacturing defects</b>	Inconsistent workmanship or inadequate quality checks can lead to rejected parts or field failures.
Production Readiness	<b>Inadequate tooling or automation</b>	Production bottlenecks arise if manufacturing was not considered early in design.
Cost Overruns	<b>Design creep</b>	Added features, systems, or performance targets inflate costs late in the process.

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## 6. Post-Production – Common Problems

Area	Problem	Explanation
In-Service Support	<b>Poor maintainability</b>	Difficult access to components, unclear maintenance manuals increase operating costs.
Upgrades	<b>Integration of new tech</b>	Adding newer avionics or engines might not be compatible with original design.
Feedback Loop	<b>Slow response to field issues</b>	Delayed or ineffective product support damages reputation and trust.

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