# Summary file for stable flow of downward-sloping stream

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#### **Abstract**

Under V2.0 and V3.0 we discuss some basic assumptions about granular flow, while some of them may be wrong. Thus Here we try to distinguish some mistakes and do a summary file for stable flow.

Ref: Equ(From Shijie Z) AND V2.0 AND V3.0

## Basic equ.

#### Lame coefficients and derivations

First come back to the coordinates we have used in describing this system.

$$h_1 = 1$$
  
 $h_2 = a \cos \alpha + b \cos \alpha$   
 $h_3 = 1$ 

For the time derivative of a function in this system, we have

$$\frac{D}{Dt} = \frac{\partial}{\partial t} + \vec{u} \cdot \nabla$$

We seperate the second term by its components

$$\begin{split} \vec{u} \cdot \nabla &= (u_1 \vec{e}_1 + u_2 \vec{e}_2 + u_3 \vec{e}_3) \cdot (\partial_1 \vec{e}_1 + \partial_2 \vec{e}_2 + \partial_3 \vec{e}_3) \\ &= \frac{1}{h_1} u_1 \partial_1 + \frac{1}{h_2} u_2 \partial_2 + \frac{1}{h_3} u_3 \partial_3 \\ &= u_a \partial_a + \frac{1}{h_2} u_\theta \partial_\theta + u_b \partial_b \end{split}$$

For the divergence of a function (velocity) in this system, we have

$$\begin{split} \nabla \cdot \vec{u} &= (\partial_1 \vec{e}_1 + \partial_2 \vec{e}_2 + \partial_3 \vec{e}_3) \cdot (u_1 \vec{e}_1 + u_2 \vec{e}_2 + u_3 \vec{e}_3) \\ &= \frac{1}{h_1 h_2 h_3} \partial_1 \left( \frac{h_1 h_2 h_3}{h_1} u_1 \right) + \frac{1}{h_1 h_2 h_3} \partial_2 \left( \frac{h_1 h_2 h_3}{h_2} u_2 \right) + \frac{1}{h_1 h_2 h_3} \partial_3 \left( \frac{h_1 h_2 h_3}{h_3} u_3 \right) \\ &= \frac{1}{h_2} \partial_a \left( (a \cos \alpha + b \sin \alpha) u_a \right) + \frac{1}{h_2} \partial_\theta u_\theta + \frac{1}{h_2} \partial_b \left( (a \cos \alpha + b \sin \alpha) u_b \right) \\ &= \frac{1}{h_2} u_a \cos \alpha + \partial_a u_a + \frac{1}{h_2} \partial_\theta u_\theta + \frac{1}{h_2} \partial_b \left( (a \cos \alpha + b \sin \alpha) u_b \right) \end{split}$$

For the stress term in momentum conservation equation, there may be different ways to describe its mathematical form. Here we focus on the simple form like:

$$egin{aligned} T_{ij} &= -p\delta_{ij} + au_{ij} \ 
abla \cdot \overleftrightarrow{T} &= -
abla p + \mu 
abla^2 ec{u} \end{aligned}$$

For the  $\operatorname{gradient}$  of  $\operatorname{pressure} p$  , it is quite clear that

Here one thing needs to be noticed that the pressure term p is only related to b or (b,h) when considering the stable flow. In other words, the disturbance case need to be reconsider the pressure term from the momentum conservation of b direction.

$$abla p = \partial_a p \ ec{e}_a + rac{1}{h_2} \partial_ heta p \ ec{e}_ heta + \partial_b p \ ec{e}_b$$

For the laplacian of velocity, it may take some time to verify the concrete form.

$$\begin{split} \nabla^2 \vec{u} &= \left(\partial_1 \vec{e}_1 + \partial_2 \vec{e}_2 + \partial_3 \vec{e}_3\right) \cdot \left(\partial_1 \vec{e}_1 + \partial_2 \vec{e}_2 + \partial_3 \vec{e}_3\right) (u_1 \vec{e}_1 + u_2 \vec{e}_2 + u_3 \vec{e}_3) \\ &= \frac{1}{h_1 h_2 h_3} \begin{pmatrix} \partial_1 h_2 h_3 \\ \partial_2 h_1 h_3 \\ \partial_3 h_1 h_2 \end{pmatrix}^T \begin{pmatrix} \frac{1}{h_1} \partial_1 u_1 & \frac{1}{h_1} \partial_1 u_2 & \frac{1}{h_2} \partial_2 u_3 \\ \frac{1}{h_2} \partial_2 u_1 & \frac{1}{h_2} \partial_2 u_2 & \frac{1}{h_2} \partial_2 u_3 \\ \frac{1}{h_3} \partial_3 u_1 & \frac{1}{h_3} \partial_3 u_2 & \frac{1}{h_3} \partial_3 u_3 \end{pmatrix} \\ &= \frac{1}{h_1 h_2 h_3} \left[ \partial_1 \left( \frac{h_2 h_3}{h_1} \partial_1 u_1 \right) + \partial_2 \left( \frac{h_1 h_3}{h_2} \partial_2 u_1 \right) + \partial_3 \left( \frac{h_1 h_2}{h_3} \partial_3 u_1 \right) \right] \vec{e}_1 \\ &+ \frac{1}{h_1 h_2 h_3} \left[ \partial_1 \left( \frac{h_2 h_3}{h_1} \partial_1 u_2 \right) + \partial_2 \left( \frac{h_1 h_3}{h_2} \partial_2 u_2 \right) + \partial_3 \left( \frac{h_1 h_2}{h_3} \partial_3 u_2 \right) \right] \vec{e}_2 \\ &+ \frac{1}{h_1 h_2 h_3} \left[ \partial_1 \left( \frac{h_2 h_3}{h_1} \partial_1 u_3 \right) + \partial_2 \left( \frac{h_1 h_3}{h_2} \partial_2 u_3 \right) + \partial_3 \left( \frac{h_1 h_2}{h_3} \partial_3 u_3 \right) \right] \vec{e}_3 \end{split}$$

In our case, it becomes:

$$egin{aligned} (
abla^2ec{u})_i &= rac{1}{h_2} \left[ \partial_a (h_2 \partial_a u_i) + \partial_ heta \left( rac{1}{h_2} \partial_ heta u_i 
ight) + \partial_b (h_2 \partial_b u_i) 
ight] \ &= rac{\coslpha}{h_2} \partial_a u_i + \partial_{aa} u_i + rac{1}{(h_2)^2} \partial_{ heta heta} u_i + rac{\sinlpha}{h_2} \partial_b u_i + \partial_{bb} u_i \end{aligned}$$

#### Stable flow and integration

The core idea is to integrate the function along *b* direction, in this way we can use the similarity properties in thin-film flow analysis (relations about distributions between real and mean quantities)

Stable means every quantity is time-independent and only related to parameter a and b, which means:

$$\partial_t = 0$$
 ;  $\partial_\theta = 0$ 

In our case the main quantity is velocity along a direction when considering the stable flow. We define its mean as:

$$\bar{u} \equiv \frac{1}{h} \int_0^h u db \tag{1.1}$$

For the stable flow, the continuity becomes:

$$egin{aligned} 0 &= 
abla \cdot ec{u} \ &= rac{1}{h_2} u_a \cos lpha + \partial_a u_a + rac{1}{h_2} \partial_ heta u_ heta + rac{1}{h_2} \partial_b \left( (a \cos lpha + b \sin lpha) u_b 
ight) \end{aligned}$$

Do the integration on both sides and we can get:

$$0=\partial_a ar{u}+rac{\coslpha}{h}\int_0^hrac{1}{h_2}udb+rac{1}{h}\int_0^hrac{1}{h_2}\partial_b\left((a\coslpha+b\sinlpha)u_b
ight)db$$

For term III by using the boundary conditions we can obtain:(here use the approximation that  $\frac{1}{h_2}\sim\frac{1}{a\cos\alpha}$ , the provement and discussion see the **Term3** and **Appendix for**  $\frac{1}{h_2}$ )

$$egin{aligned} \operatorname{Term3} &= rac{1}{h} \int_0^h rac{1}{h_2} \partial_b \left( (a\coslpha + b\sinlpha) u_b 
ight) db \ &\simeq rac{1}{h} rac{1}{a\coslpha} (a\coslpha + b\sinlpha) u_b igg|_{b=0}^{b=h} \ &= rac{1}{h} \left[ \left( 1 + rac{h}{a} anlpha 
ight) u_b igg|_h - (1+0) u_b igg|_0^2 
ight] \ &= \left( 1 + rac{h}{a} anlpha 
ight) rac{1}{h} rac{D}{Dt} h \end{aligned}$$

Here using the lame derivation I have shown above we can get

$$\frac{D}{Dt}h = (\partial_t + \mathbf{\bar{u}} \cdot \nabla) h$$
$$= \bar{u}\partial_a h$$

Here one thing needs to mention that the orange part means we use the b-mean(which means  $u_a$  takes average along b direction) velocity  $\bar{u}$  to 'observe' the change of b. The reason of choosing this value is because it is the mean velocity of  $u_a$  along b direction, which will be shown in **velocity** similarity and velocity profile part below.

When we consider the flow in the thin film, we can assume that the velocity  $u_a$  (now denoted by u) has the similarity with its free-surface value (  $u|_{b=h}\equiv u_s$ ). The form can be expressed as:

$$\frac{u}{u_s} = f(\eta) \quad \text{where } \eta = \frac{b}{h}$$
 (1.2)

And by researches of earlier authors, we can express the distribution function as:

$$f(\eta) = A\eta + B\eta^2 \tag{1.2}$$

For term II it can be written as:

$$\operatorname{Term2} = \frac{\cos \alpha}{h} \int_0^h \frac{1}{h_2} u db$$

$$= \frac{\cos \alpha}{h} \int_0^h \frac{\bar{u}}{a \cos \alpha + b \sin \alpha} \left( A \frac{b}{h} + B \frac{b^2}{h^2} \right) db$$

$$= \frac{\bar{u} \cos \alpha}{h} \int_0^h \frac{1}{a \cos \alpha + b \sin \alpha} \left( A \frac{b}{h} + B \frac{b^2}{h^2} \right) db$$

The integration is quite complex, we can only focus on  $o\left(\frac{h}{a}\right)\sim o(1)$  terms (in the final form)

$$RHS = \frac{u_s \cos \alpha}{h} \frac{1}{a \cos \alpha} \int_0^h \left( 1 + \frac{b}{a} \tan \alpha \right)^{-1} \left( A \frac{b}{h} + B \frac{b^2}{h^2} \right) db$$

$$= \frac{u_s}{ah} \int_0^h \left( 1 - \frac{b}{a} \tan \alpha + o \left( \frac{b^2}{a^2} \tan^2 \alpha \right) \right) \left( A \frac{b}{h} + B \frac{b^2}{h^2} \right) db$$

$$= \frac{u_s}{a} \left( \frac{1}{2} A - \frac{1}{3} \frac{h \tan \alpha}{a} A + \frac{1}{3} B - \frac{1}{4} \frac{h \tan \alpha}{a} B \right)$$

Now we come back to see our velocity profile assumption in  $\left(1.1\right)$  and get

$$egin{aligned} ar{u} &\equiv rac{1}{h} \int_0^h u db \ &= u_s \int_0^1 (A \eta + B \eta^2) d \eta \ &= \left(rac{1}{2} A + rac{1}{3} B
ight) u_s \end{aligned}$$

Thus the  $u_s$  can be rewritten by the form of  $\overline{u}$  as below:

$$\operatorname{Term2} = \left(\frac{1}{2}A + \frac{1}{3}B\right) \frac{u_s}{a} \left(1 - \frac{h}{a} \tan \alpha \frac{\frac{1}{3}A + \frac{1}{4}B}{\frac{1}{2}A + \frac{1}{3}B}\right)$$
$$= \frac{\bar{u}}{a} \left(1 - \frac{h}{a} \frac{4A + 3B}{6A + 4B} \tan \alpha\right)$$

Thus the final form of continuum is:

$$\partial_a \bar{u} + \frac{\bar{u}}{a} \left( 1 - \frac{h}{a} \frac{4A + 3B}{6A + 4B} \tan \alpha \right) + \left( 1 + \frac{h}{a} \tan \alpha \right) \frac{1}{h} \bar{u} \partial_a h + \frac{1}{b} \int_0^h \frac{1}{h_2} \partial_\theta u_\theta db = 0$$
 (1.3)

Now we correct it to the non-zero first order form, the continuum of stable form becomes:

$$\partial_a(\bar{u}h) + \frac{\bar{u}h}{a} = 0 \tag{1.3*}$$

Appendix for  $\frac{1}{h_2}$ : from (1.3) to  $(1.3^*)$  we know that in most integration in our model,  $\left(\frac{1}{h_2},u\right)\sim\left(\frac{1}{a\cos\alpha},\overline{u}\right)$  is a well-done zeroth-ordered approximation.

For the stable flow, we also need to determine the momentum relations in different directions. Due to Shijie Zhong's form, we can get:

$$d_t = \partial_t + \vec{u} \cdot \nabla$$

Here we introduce the time derivative as

$$u_a = \partial_t + u_a \partial_a + rac{1}{h_2} u_ heta \partial_ heta + u_b \partial_b \partial_ heta$$

and the gravity as  $ec{g}=g\sinlpha~\hat{e}_a-g\coslpha~\hat{e}_b$ 

$$\begin{aligned} a - \text{direction:} \quad & \rho \phi \left( d_t u_a - \frac{1}{h_2} u_\theta^2 \cos \alpha \right) = (\nabla \cdot \overrightarrow{T})_a + \rho \phi g \sin \alpha \\ \theta - \text{direction:} \quad & \rho \phi \left( d_t u_\theta + \frac{1}{h_2} u_\theta (u_a \cos \alpha + u_b \sin \alpha) \right) = (\nabla \cdot \overrightarrow{T})_\theta \end{aligned}$$

$$b- ext{direction:} \quad 
ho\phi\left(d_tu_b-rac{1}{h_2}u_{ heta}^2\sinlpha
ight)=(
abla\cdot\stackrel{\leftrightarrow}{T})_b-
ho\phi g\coslpha$$

Something needs to be discussed that here we can use different models to describe the stress term  $(\nabla \cdot \overleftrightarrow{T})$ . To verify our model, we choose the basic constitutive relation applied here:

$$T_{ij} = -p\delta_{ij} + au_{ij} \qquad ext{where } au_i = (\mu 
abla^2 ec{u})_i$$

In stable flow case, things become quite easy that everything relates to  $\theta$  and  $\partial_t$  will be zero. We first apply the Lame form we have calculated above to show the form of the shear stress term:

$$(\mu
abla^2ec{u})_i = \mu\left[rac{\coslpha}{h_2}\partial_a u_i + \partial_{aa}u_i + rac{1}{(h_2)^2}\partial_{ heta heta}u_i + rac{\sinlpha}{h_2}\partial_b u_i + \partial_{bb}u_i
ight]$$

While in the thin-film, terms along the wall direction can be considered as small quantities compared with those terms perpendicular to the wall, thus the actual relations of momentum are:

a – direction: 
$$\rho\phi\left(u_a\partial_au_a + u_b\partial_bu_a\right) = -\partial_ap + \mu\frac{\sin\alpha}{h_2}\partial_bu_a + \mu\partial_{bb}u_a + \rho\phi g\sin\alpha$$
b – direction: 
$$\rho\phi\left(u_a\partial_au_b + u_b\partial_bu_b\right) = -\partial_bp + (\mu\nabla^2u_b) - \rho\phi g\cos\alpha$$

Here we use the assumption that the velocity along the b direction is much smaller than the velocity along the  $a \& \theta$  direction (and other partial-derivative terms), thus we can simplify those functions above as:

$$a - \text{direction:} \quad \rho\phi\left(u_a\partial_a u_a + u_b\partial_b u_a\right) = -\partial_a p + \mu\frac{\sin\alpha}{h_2}\partial_b u_a + \mu\partial_{bb}u_a + \rho\phi g\sin\alpha \quad (1.4)$$

$$b - \text{direction:} \quad \rho\phi\left(u_a\partial_a u_b + u_b\partial_b u_a\right) = -\partial_b p + (\mu\nabla^2 u_b) - \rho\phi g\cos\alpha \quad (1.5)$$

Thus from the (1.5) we can directly get the pressure distribution:

$$p = -\rho \phi q \cos \alpha \ b + C$$

Apply the B.C. on the free-surface:  $pert_{b=h}=p_0$ , it becomes:

$$C = p_0 + \rho \phi g \cos \alpha \ h$$

Here we use the additional pressure  $p-p_0$  to express p, it becomes:

$$p = \rho \phi g \cos \alpha (h - b) \tag{1.6}$$

For (1.4) we integrate it along b direction and apply (1.6) in it and obtain:

$$rac{1}{h}\int_0^h u\partial_a udb + g\coslpha\partial_a h = g\sinlpha + rac{\mu}{
ho\phi}rac{\sinlpha}{h}\int_0^h rac{1}{h_2}\partial_b udb + rac{\mu}{
ho\phi}rac{1}{h}\int_0^h \partial_{bb} udb$$

Also use the appendix for  $\frac{1}{h_2}$ , the approach of Term2 and B.C., we can find that

$$rac{1}{2h}\int_0^h\partial_a u^2\ db+g\coslpha\partial_a h=g\sinlpha+rac{\mu}{
ho\phi}rac{c_2}{ah} anlpha\ ar{u}+rac{\mu}{
ho\phi}rac{1}{h}\left(0-\partial_b u|_{b=0}
ight)$$

Here we need two approximations to determine the first term and the stress caused by the wall, and I will prove that these two assumptions can be obtained by the velocity profile later.

$$egin{align} rac{1}{2h}\int_0^h\partial_a u^2\ db &= c_1\partial_aar u^2 \ au_\omega &= \left.\partial_b u
ight|_{b=0} = c_2rac{ar u}{h} \ \end{pmatrix} \ (1.2^*) 
onumber \ (1.2^*)$$

Thus the integration becomes:

$$c_1 \partial_a \bar{u}^2 + g \cos \alpha \, \partial_a h = g \sin \alpha + \frac{\mu}{\rho \phi} \frac{c_2}{ah} \bar{u} \tan \alpha - \frac{\mu}{\rho \phi} \frac{1}{h^2} c_2 \bar{u} \tag{1.7}$$

Here we calculate the function (1.3\*) and get its solution is

$$\bar{u}ha = c_0 \tag{1.8}$$

(Notice that here we can determine the const  $c_0$  by the definition of  $c_0=q^*\equiv rac{Q^*}{2\pi}$ )

Now we can plug (1.8) in the (1.7) to eliminate other variable  $ar{u}$ , let it be f(h,a)=0

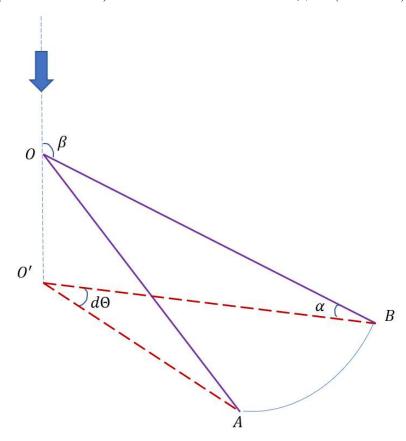
$$c_{1}\partial_{a}\left(\frac{c_{0}}{ha}\right)^{2} + g\cos\alpha \,\partial_{a}h = g\sin\alpha + \frac{\mu}{\rho\phi}\frac{c_{0}c_{2}}{h^{2}a^{2}}\tan\alpha - \frac{\mu}{\rho\phi}\frac{1}{ah^{3}}c_{2}c_{0}$$

$$-c_{1}c_{0}^{2}\frac{2}{h^{3}a^{3}}(a\partial_{a}h + h) + g\cos\alpha \,\partial_{a}h = g\sin\alpha + \frac{\mu}{\rho\phi}\frac{c_{0}c_{2}}{h^{2}a^{2}}\tan\alpha - \frac{\mu}{\rho\phi}\frac{1}{ah^{3}}c_{2}c_{0}$$

$$\left((-2c_{1}c_{0}^{2})\frac{1}{h^{3}a^{2}} + g\cos\alpha\right)\partial_{a}h = c_{1}c_{0}^{2}\frac{2}{h^{2}a^{3}} + g\sin\alpha + \frac{\mu}{\rho\phi}\frac{c_{0}c_{2}}{h^{2}a^{2}}\tan\alpha - \frac{\mu}{\rho\phi}\frac{1}{ah^{3}}c_{2}c_{0}$$

Try to simplify the form, the final result is:

$$\left(2c_1\frac{{c_0}^2}{a^2}-gh^3\cos\alpha\right)\partial_a h=-gh^3\sin\alpha-2c_1\frac{h{c_0}^2}{a^3}+c_2\frac{\mu}{\rho\phi}\frac{c_0}{a}\left(1-\tan\alpha\frac{h}{a}\right) \tag{1.9}$$



Compare our result with Guangzhao Zhou's  $\mathsf{result}^{[1]},$  we can find that

$$\left(\frac{54}{35} \frac{q^2}{r^2 \sin^2 \beta} - gh^3 \sin \beta\right) \frac{\mathrm{d}h}{\mathrm{d}r} 
= h^3 g \cos \beta - \frac{54}{35} \frac{hq^2}{r^3 \sin^2 \beta} + \frac{3vq}{r \sin \beta} \left(1 + \cot \beta \frac{h}{r}\right)$$
(G1)

In spherical coordinates, the q-flux can be expressed as:

$$\begin{split} Q &\equiv \iint_S \vec{u} \cdot \vec{n} dS \\ &= -2\pi r^2 \int_0^{\Psi} \sin\theta \ u \ d\psi \\ &= -2\pi r^2 \int_0^{\Psi} \sin(\beta - \psi) \ u \ d\psi \\ &= -2\pi r^2 \int_0^{\Psi} (\sin\beta \cos\psi - \sin\psi \cos\beta) \ u \ d\psi \\ &\simeq -2\pi r^2 \sin\beta \int_0^{\Psi} u \ d\psi \end{split}$$

recall the mean velocity definition, it is:

$$ar{u} = rac{\int_0^\Psi r u \ d\psi}{\int_0^\Psi r \ d\psi}$$

So it becomes:

$$egin{aligned} q &\equiv rac{Q}{2\pi} = -r^2 \sineta\Psiar{u} \ &= rhar{u}\sineta \ &= q^*\sineta \end{aligned}$$

Thus our result can correspond formally to the result of Guangzhao Zhou's result here. And it also shows that, when we use the velocity profile like below:

$$rac{u}{u_s} = rac{\psi}{\Psi} \left( 2 - rac{\psi}{\Psi} 
ight)$$

Thus we can find the corresponding coefficients:

$$(A=2,B=-1) \longrightarrow \left(c_1=rac{27}{35},c_2=3
ight)$$

To verify this relation, we need to calculate the  $(c_1, c_2)$  from (A, B) directly! The core calculation is to use (1.2) to obtain the (1.2\*), see below:

Here (A=2,B=-1) means that  $ar{u}=rac{2}{3}u_s$ 

$$egin{aligned} \partial_b u|_{b=0} &= rac{A}{h} u_s \ &= rac{2}{h} rac{3}{2} ar{u} \ &\longrightarrow c_2 = 3 \ &= 3 rac{ar{u}}{h} \end{aligned}$$

To verify the relation 1, we first assume that  $\partial_a h \sim o(1)$ 

$$\begin{split} \frac{1}{2h} \int_0^h \partial_a u^2 \ db &= \frac{1}{2} \partial_a \left( \frac{1}{h} \int_0^h u^2 \ db \right) \\ &= \frac{1}{2} \partial_a \left( \int_0^1 \left( A \eta + B \eta^2 \right)^2 u_s \ d\eta \right) \\ &= \frac{1}{2} \left( \frac{4}{3} - 1 + \frac{1}{5} \right) \frac{9}{4} \partial_a \bar{u}^2 \\ &= \frac{3}{5} \partial_a \bar{u}^2 \end{split}$$

This result can correspond to Kasimov's result. While this assumption may cause some error, if we consider the h=h(a) and take  $\partial_a h$  terms into consideration, Guangzhao Zhou's result is more reasonable.

# Supplementary instruction

### The green part intergration between (1.6) and (1.7)

$$\frac{\sin \alpha}{h} \int_0^h \frac{1}{h_2} \partial_b u db = \frac{\sin \alpha}{h} \int_0^h \frac{1}{a \cos \alpha + b \sin \alpha} \left( \frac{A}{h} + 2B \frac{b}{h^2} \right) u_s db$$

$$= \frac{\sin \alpha}{h} u_s \left( \frac{A}{h} \int_0^1 \frac{1}{\frac{a}{h} \cos \alpha + \eta \sin \alpha} d\eta \right)$$

$$+ \frac{2B}{h} \int_0^1 \frac{\eta}{\frac{a}{h} \cos \alpha + \eta \sin \alpha} d\eta$$

For term 1, there is:

$$\frac{A}{h} \int_0^1 \frac{1}{\frac{a}{h}\cos\alpha + \eta\sin\alpha} d\eta = \frac{A}{h} \frac{1}{\sin\alpha} \ln\left(1 + \frac{h}{a} \frac{\sin\alpha}{\cos\alpha}\right)$$
$$\simeq \frac{A}{h} \frac{1}{\sin\alpha} \frac{h}{a} \frac{\sin\alpha}{\cos\alpha}$$
$$= \frac{A}{a\cos\alpha}$$

For term 2, there is:

$$\frac{2B}{h} \int_{0}^{1} \frac{\eta}{\frac{a}{h} \cos \alpha + \eta \sin \alpha} d\eta = \frac{2B}{h} \frac{1}{\sin^{2} \alpha} \left( \sin \alpha - \frac{a \cos \alpha}{h} \ln \left( 1 + \frac{h}{a} \frac{\sin \alpha}{\cos \alpha} \right) \right)$$
$$\simeq \frac{2B}{h} \frac{1}{\sin \alpha} \left( 1 - \frac{a}{h} \frac{\cos \alpha}{\sin \alpha} \frac{h}{a} \frac{\sin \alpha}{\cos \alpha} \right)$$
$$= 0$$

Due to

$$ar{u} = \left(rac{1}{2}A + rac{1}{3}B
ight)u_s \ = rac{2}{3}u_s$$

Thus this green integration becomes:

$$\frac{\sin \alpha}{h} \int_0^h \frac{1}{h_2} \partial_b u db = \frac{A}{ah} u_s \tan \alpha$$

$$= \frac{1}{ah} A \frac{3}{2} \tan \alpha \, \bar{u}$$

$$= \frac{3}{ah} \tan \alpha \, \bar{u}$$

$$= \frac{c_2}{ah} \tan \alpha \, \bar{u}$$