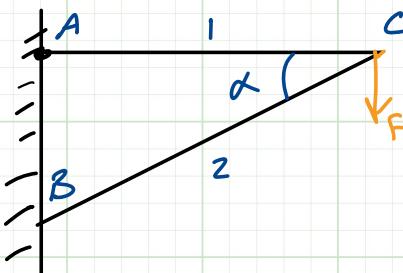
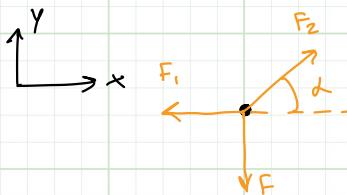


1) Stress

SchematicAssuming
Stahl

a) Draw FBD of point C



b) Find the axial forces on bar 1,2

$$\sum F_y = 0 = F_2 \sin \alpha - F$$

$$\sum F_x = 0 = F_2 \cos \alpha - F_1$$

$$\rightarrow F_1 = \frac{F}{\sin \alpha}$$

$$\rightarrow F_2 = F \cot \alpha$$

c) Find the average stresses in the bars

$$\sigma_1 = \frac{F_1}{A_1} = \frac{F}{A_1 \sin \alpha}$$

$$\sigma_2 = \frac{F_2}{A_2} = \frac{F \cot \alpha}{A_2}$$

d) Determine the cross sectional areas A_1, A_2

$$A_1 = F_1 / \sigma_1 = \frac{F}{\sigma_1 \sin \alpha}$$

$$A_2 = F_2 / \sigma_2 = \frac{F \cot \alpha}{\sigma_2}$$

e) Determine what bars are in tension and compression.

For static stability, Bar 2 must apply an upward force to balance force F. So it is in Compression.

Because F_2 has an x component for stability F_1 must oppose that force. Because F_1 points toward Bar 1 this bar is in Tension

$$f) \text{ given } A_1 = A_2 = 3 \text{ in}^2$$

$$F = 5 \text{ lbs} \quad \alpha = 30^\circ$$

$$\sigma_1 = \frac{F}{A_1 \sin \alpha} = \frac{(5 \text{ lbs})}{(3 \text{ in}^2) \sin(30^\circ)}$$

$$\rightarrow \boxed{\frac{10}{3} \text{ psi}}$$

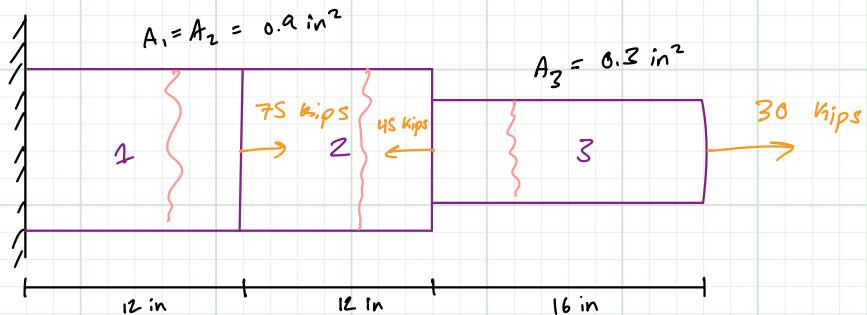
$$\sigma_2 = \frac{F \cot \alpha}{A_2} = \frac{(5 \text{ lbs}) \cot 30^\circ}{3 \text{ in}^2}$$

$$\rightarrow \boxed{2.89 \text{ psi}}$$

2) Deflection

Note: $E_{steel} = 29 \cdot 10^6 \text{ psi}$ Deformation (δ)

$$\delta = \sum_i \frac{P_i L_i}{A_i E_i}$$

Schematic

Determine The horizontal deformation

$$F_3 \xleftarrow{\square} \xrightarrow{\square} 30 \text{ kips} \quad F_2 = 30 \text{ kips} \quad \sigma_3 = \frac{F_3}{A_3} = \frac{30 \text{ kips}}{0.3 \text{ in}^2} \rightarrow 100 \text{ ksi}$$

$$F_2 \xrightarrow{\square} \xleftarrow{\square} 30 \text{ kips} \quad -F_2 = (30 - 45) \text{ kips} \quad \sigma_2 = \frac{F_2}{A_2} = \frac{15 \text{ kips}}{0.9 \text{ in}^2} \rightarrow 16.67 \text{ ksi}$$

$$\rightarrow F_2 = 15 \text{ kips}$$

$$F_1 \xleftarrow{\square} \xrightarrow{\square} 75 \text{ kips} \quad \xrightarrow{\square} 30 \text{ kips} \quad \xleftarrow{\square} 45 \text{ kips} \quad F_3 = (75 + 30 - 45) \text{ kips} \quad \sigma_1 = \frac{F_1}{A_1} = \frac{60 \text{ kips}}{0.9 \text{ in}^2} \rightarrow 66.67 \text{ ksi}$$

$$\rightarrow F_3 = 60 \text{ kips}$$

Stages 1 and 3 are in Tension, Since the restoring force points toward the Beam. Stage 2 points away from the beam indicating that it is in compression

$$\delta_T = \sum \delta = \frac{1}{E} (L_1 \sigma_1 - L_2 \sigma_2 + L_3 \sigma_3) = \frac{1}{29 \cdot 10^6 \text{ psi}} ((12 \text{ in})(66.67 \text{ ksi} - 16.67 \text{ ksi}) + (16 \text{ in})(100 \text{ ksi}))$$

$$\begin{aligned} \delta_2 &= \sigma \\ S &= L \sigma \\ &= \frac{L \sigma}{E} = \frac{PL}{AE} \end{aligned}$$

$$\delta_T = 0.076 \text{ in}$$

or just use the formula

3) Material properties

All values @ room temperature

	metric	imperial	metric	imperial	metric	imperial
	Aluminum 2024-T6	Aluminum 6061-T6	Aluminum 7075-T6			
Modulus of Elasticity (E)	72.4 GPa 10800 ksi	68.9 GPa 10000 ksi	71.7 GPa 10400 ksi			
Poisson's Ratio (n)	0.33	0.33	0.33			
Ultimate Stress (σ)	476 MPa 69 ksi	310 MPa 45 ksi	572 MPa 83 ksi			
Yield Strength (σ)	393 MPa 57 ksi	276 MPa 40 ksi	503 MPa 73 ksi			
Density (ρ)	2.78 g/cm³ 0.100 lb/in³	2.70 g/cm³ 0.0975 lb/in³	2.81 g cm³ .102 lb in³			

of these materials 7075 is the strongest, however 2024 has the highest strain to stress ratio (E) which I find interesting

ii) This took me about 2 hours to complete.