

# Chronological summary of the work performed during my Thesis

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# Ansys

- The first couple of weeks consisted mostly of getting familiar with Ansys and starting some small parameter sweeps run from MATLAB. Some errors I ran into:
  - Extra singularity for  $\text{atan}(L33/L11) = \text{Alpha}$
  - Output of the simulations consisted of double data for the area  $u_{in} > 0$

# Clear resin

- Use flexural modulus? (2.2 GPa)
- Data on yield strength? (calculate through elongation at failure?)

	METRIC <sup>1</sup>		IMPERIAL <sup>1</sup>		METHOD
	Green <sup>2</sup>	Post-Cured <sup>3</sup>	Green <sup>2</sup>	Post-Cured <sup>3</sup>	
<b>Tensile Properties</b>					
Ultimate Tensile Strength	38 MPa	65 MPa	5510 psi	9380 psi	ASTM D 638-10
Tensile Modulus	1.6 GPa	2.8 GPa	234 ksi	402 ksi	ASTM D 638-10
Elongation at Failure	12 %	6.2 %	12 %	6.2 %	ASTM D 638-10
<b>Flexural Properties</b>					
Flexural Modulus	1.25 GPa	2.2 GPa	181 ksi	320 ksi	ASTM C 790-10
<b>Impact Properties</b>					
Notched IZOD	16 J/m	25 J/m	0.3 ft-lbf/in	0.46 ft-lbf/in	ASTM D 256-10
<b>Temperature Properties</b>					
Heat Deflection Temp. @ 264 psi	42.7 °C	58.4 °C	108.9 °F	137.1 °F	ASTM D 648-07
Heat Deflection Temp. @ 66 psi	49.7 °C	73.1 °C	121.5 °F	163.6 °F	ASTM D 648-07

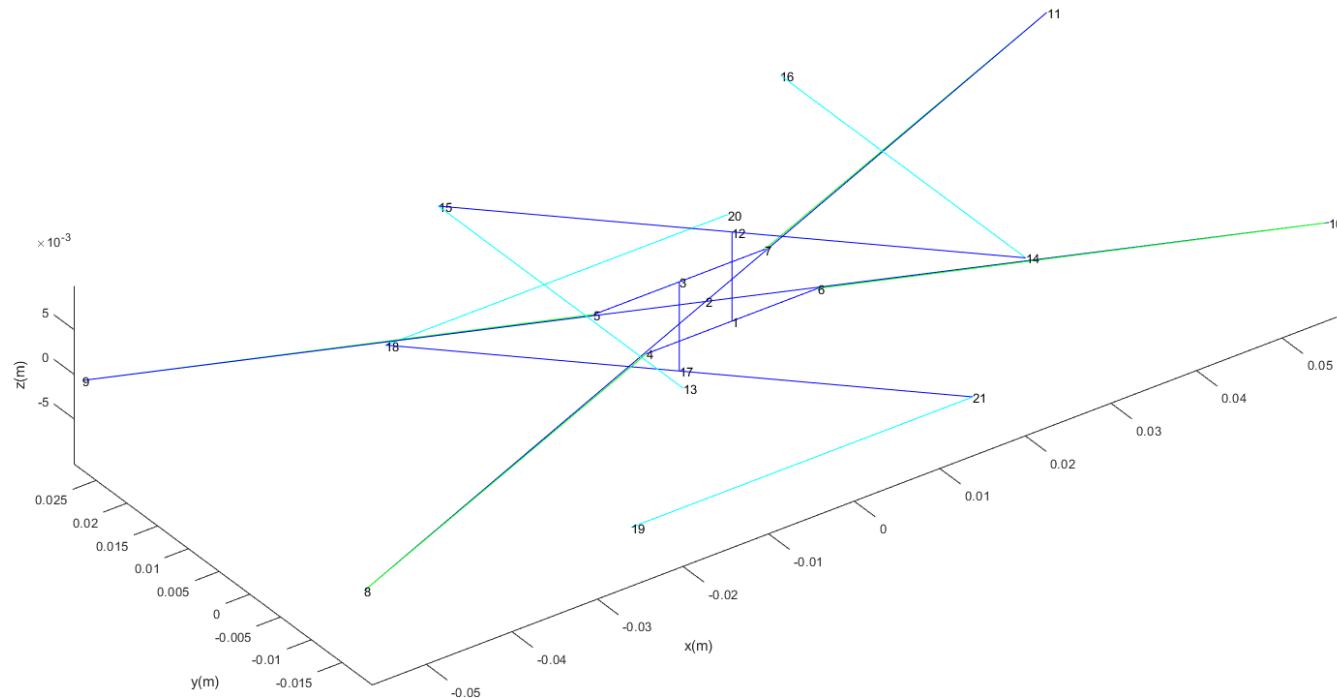
<sup>1</sup>Material properties can vary with part geometry, print orientation, print settings, and temperature.

<sup>2</sup>Data was obtained from green parts, printed using Form 2, 100 µm, Clear settings, washed and air dried without post cure.

<sup>3</sup>Data was obtained from parts printed using Form 2, 100 µm, Clear settings, and post-cured with 1.25 mW/cm<sup>2</sup> of 405 nm LED light for 60 minutes at 60 °C.

# Kinematics

- Behavior when  $\text{atan}(L_{33}/L_{11}) = \text{Alpha}$ .
  - When the flexures create a rotation point in the middle of the system.
- How to deal with this?
- 1) Run check before – dependent on parameters + limit computations
- 2) Check if Ansys produces error



# Handling of data

## Data handling

- Overview of all tested possibilities. Add new cases.
- Avoid running an existing solution again.
- Makes sure all data is saved.

Solution?:

- Object oriented
  - Save textfiles with all data per case, or save objects with data as property.
  - How to check for duplicates

## Implementing Criteria

# Optimization

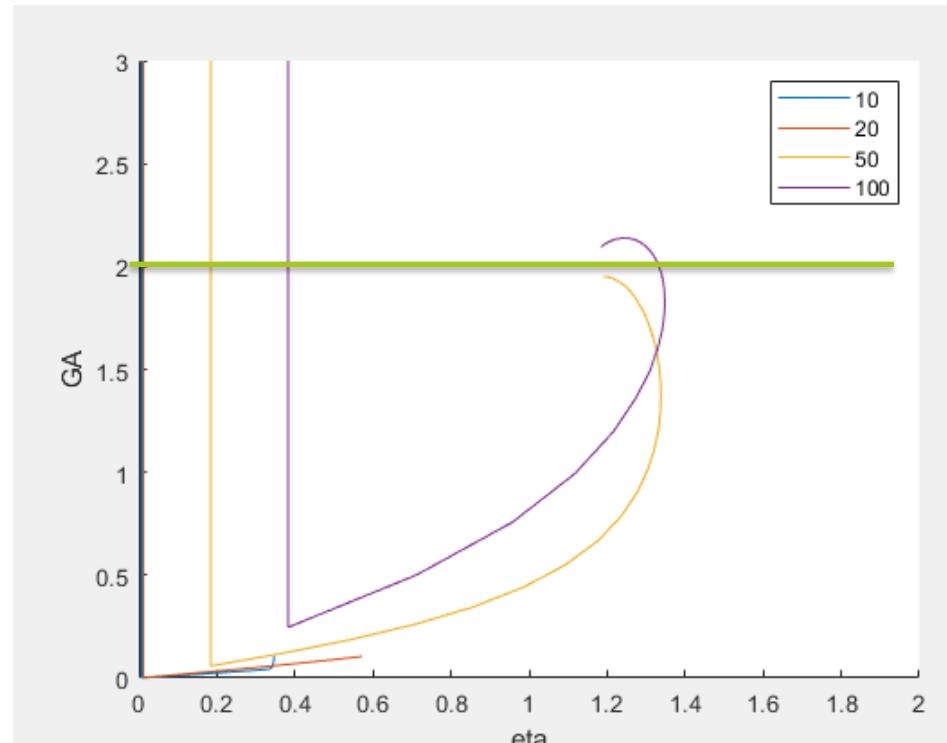
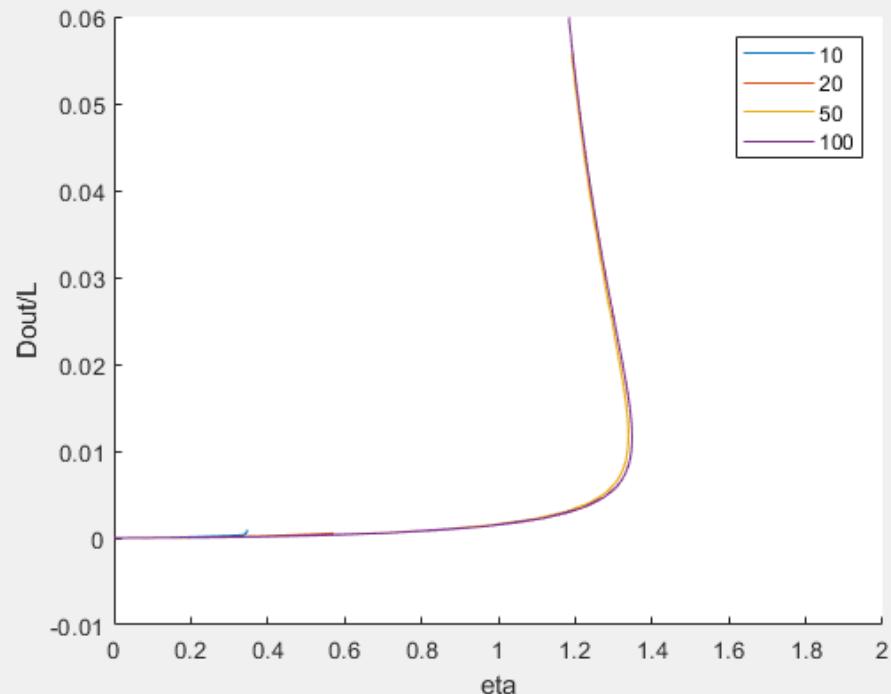
- How extensive should the search be?
- Should I investigate using optimization algorithms?
  - What should the objective be?
  - Constraints?

# Yield strength

- Ultimate strength, 65 MPa
- Highest a viable design gets to is  
 $<30 \text{ MPa}$

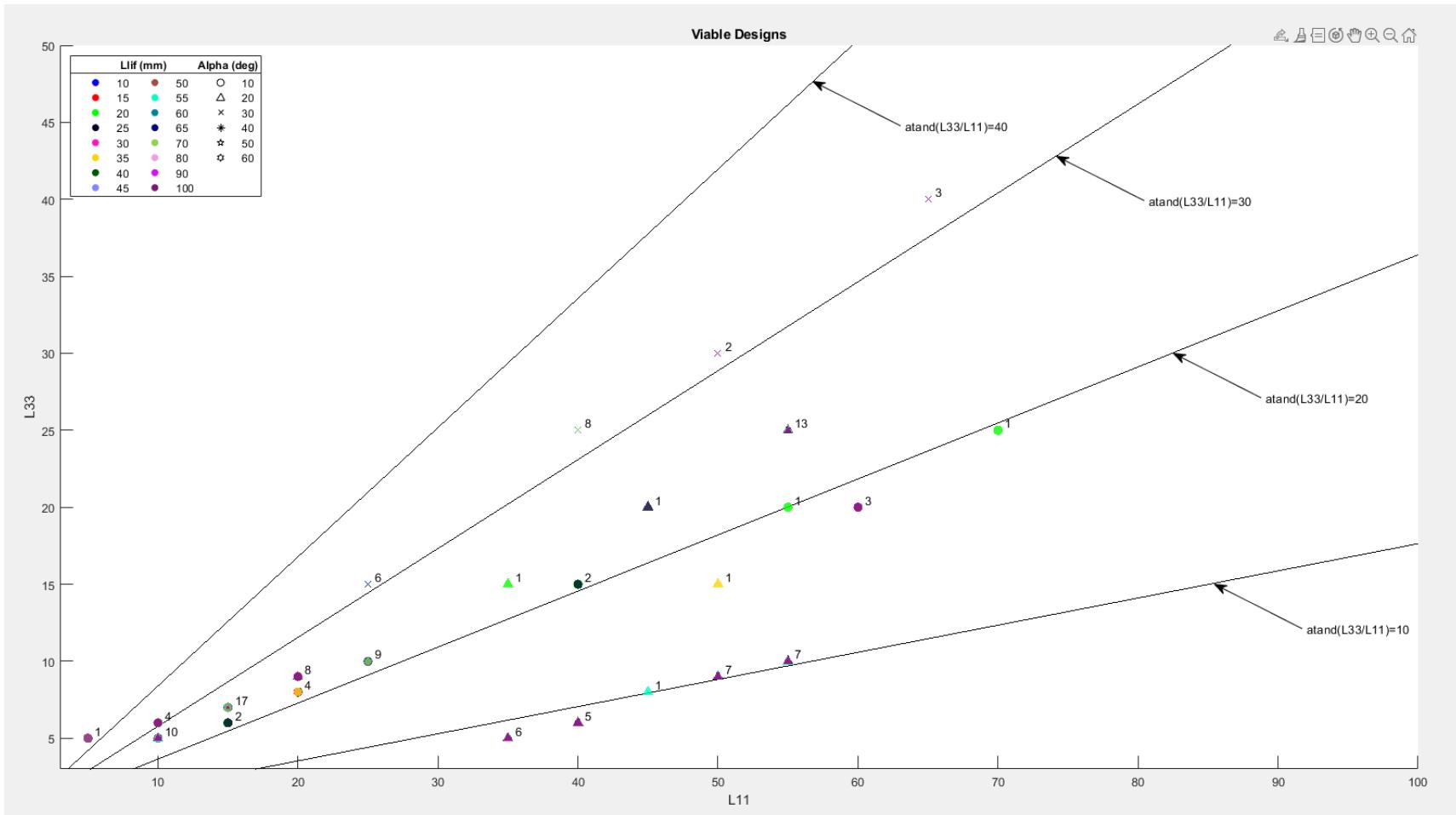
# Kinetoelastic map

- To Non-Dimensionalize a compliant mechanism
- $\eta = \frac{Fs^2}{Ebd}$ ,  $s = \frac{L}{d}$  All are average values. S  
s, was determined by changing d. By increasing the force eta changes.  
(Did have to increase b(out of plane thickness), to counter out of plane displacement)



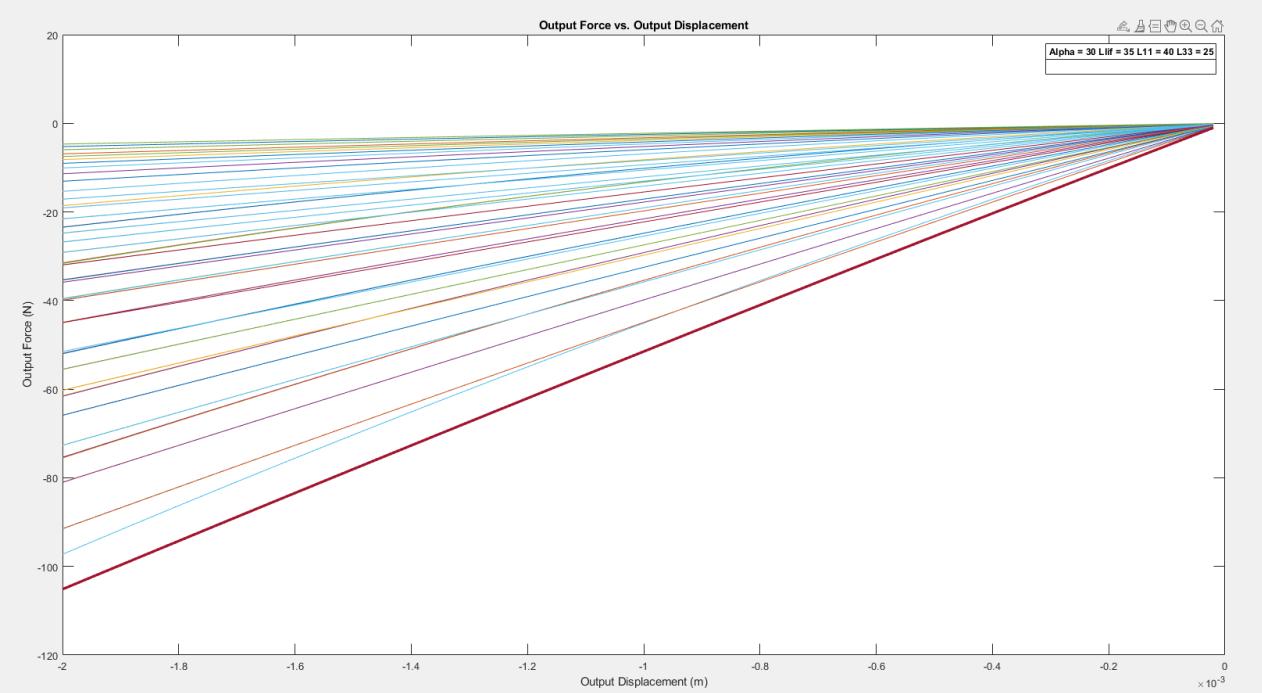
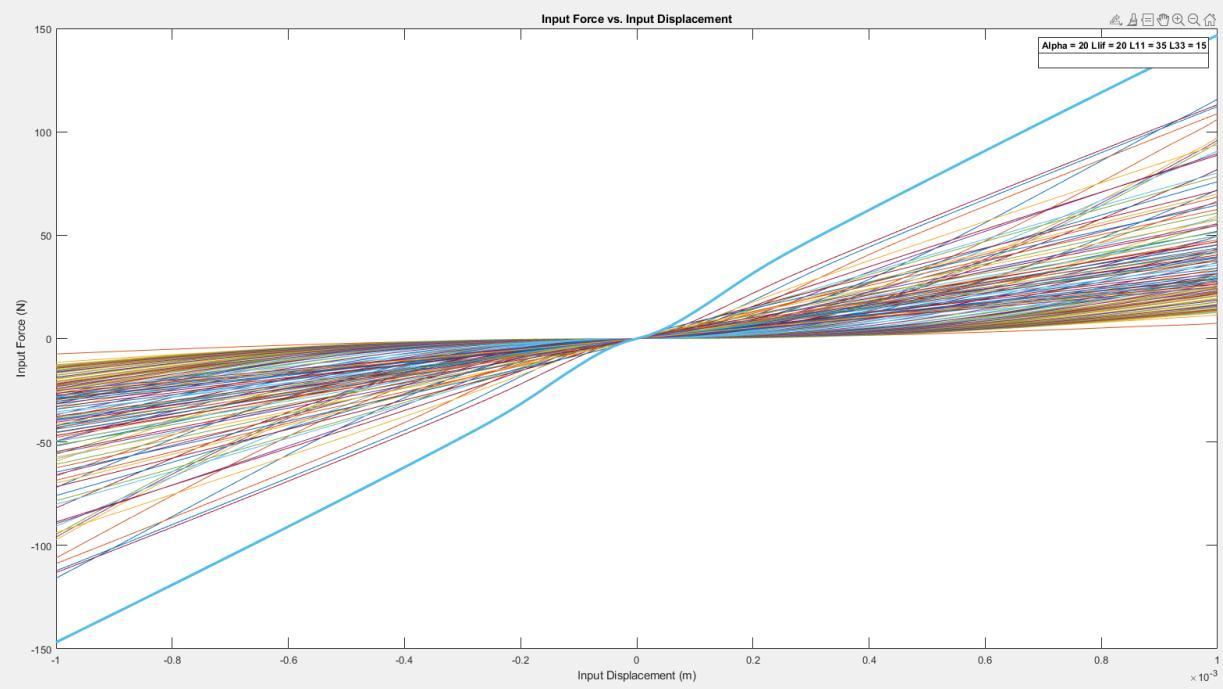
# Viable designs

```
Alpha = [10,20,30,40,50,60,70]; %deg 22, angle  
Llif = [10,15,20,25,30,35,40,45,50,55,60,65,70,80,90,100]*Dinput;  
L11 = [5,10,15,20,25,30,35,40,45,50,55,60,65,70]*Dinput; %mm 18, width  
L33 = [5,6,7,8,9,10,15,20,25,30,35,40,45,50]*Dinput; %mm 8.6, height
```

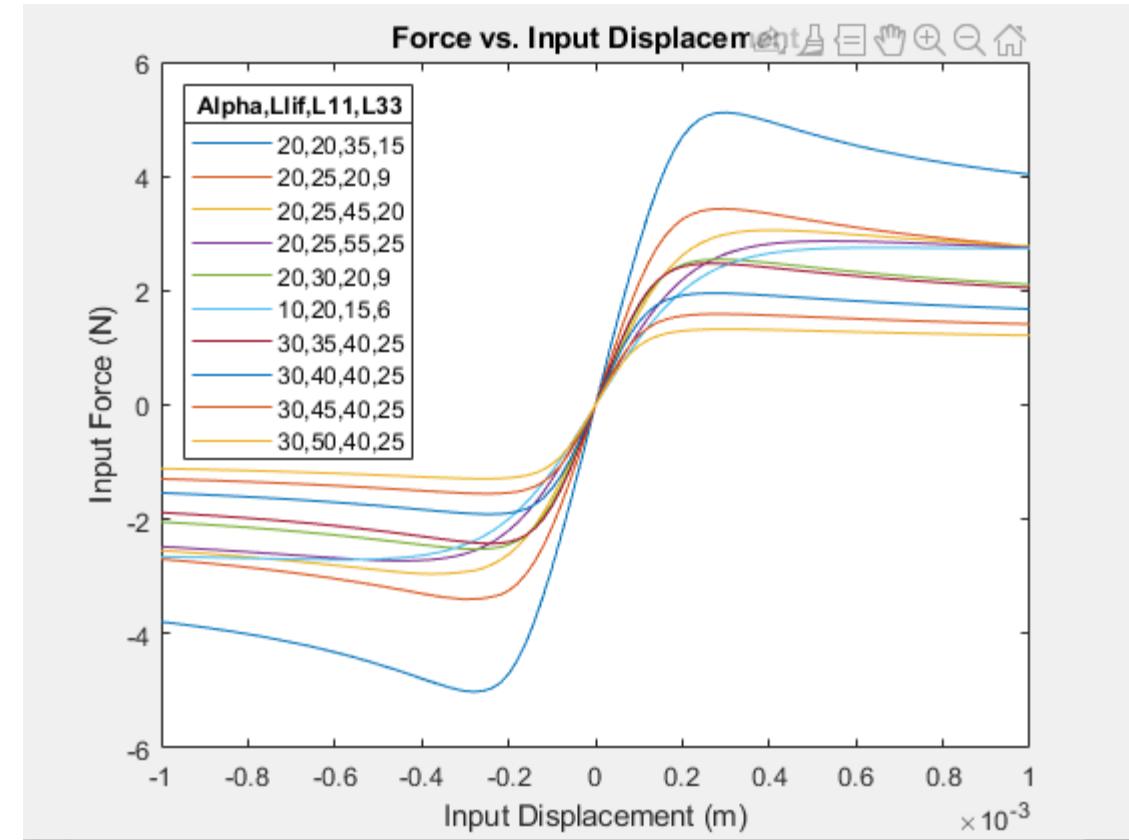
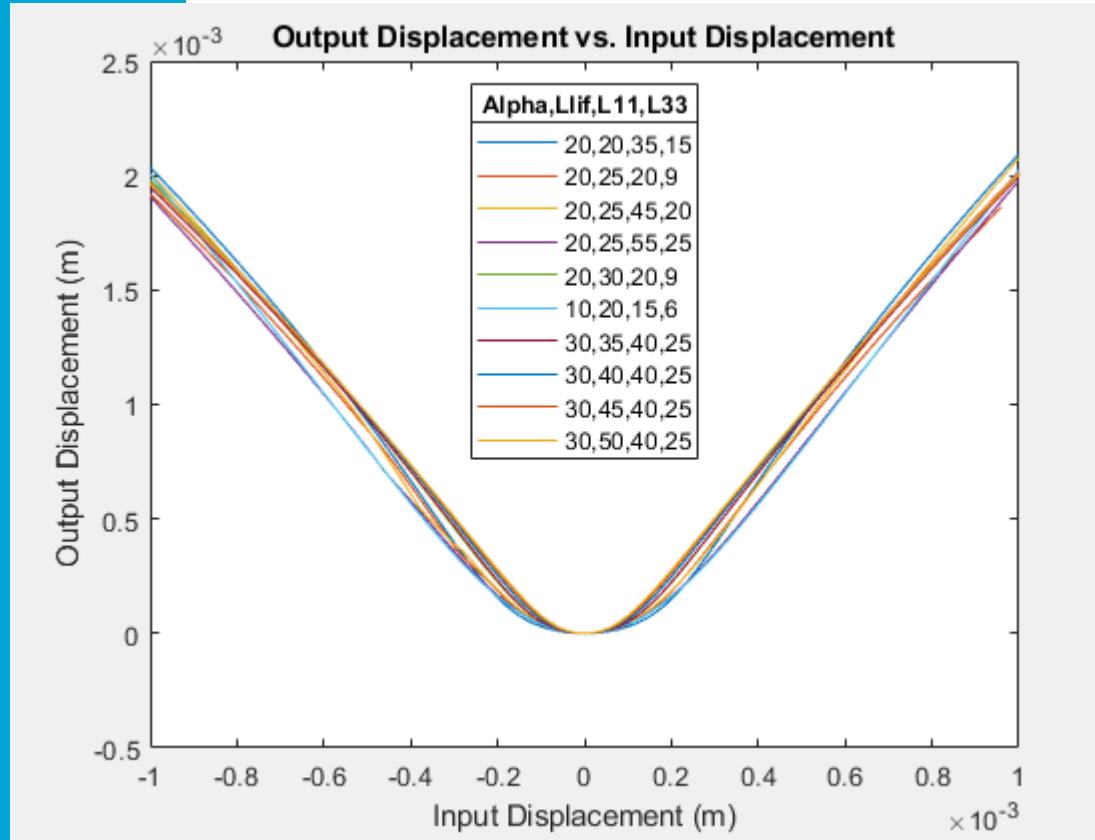


# Viable designs

- Testing input and output stiffness



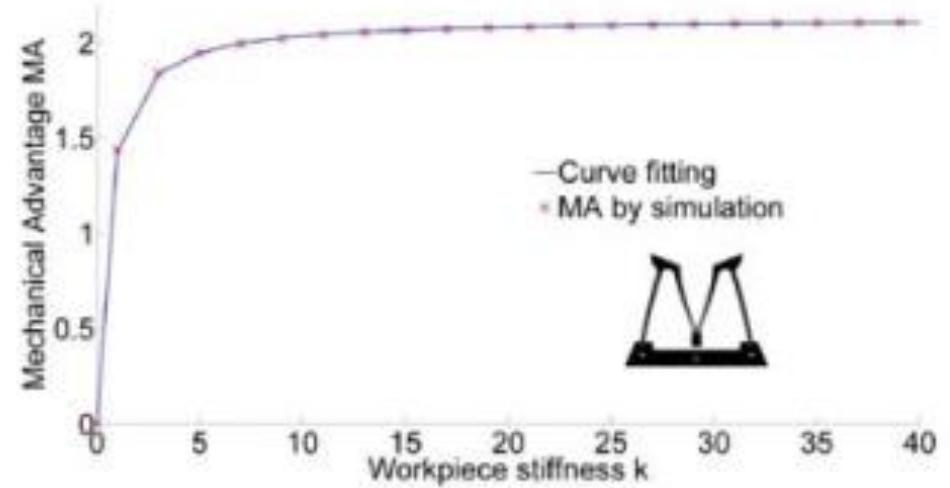
# ~10 best designs



- Next step?
- How to make sure this is the global optimum

# MA

- Did not find any formulae for MA with components of input/output stiffness and actuation stiffness.



$$MA = MA_x \left( \frac{k_w}{\sigma + k_w} \right)$$

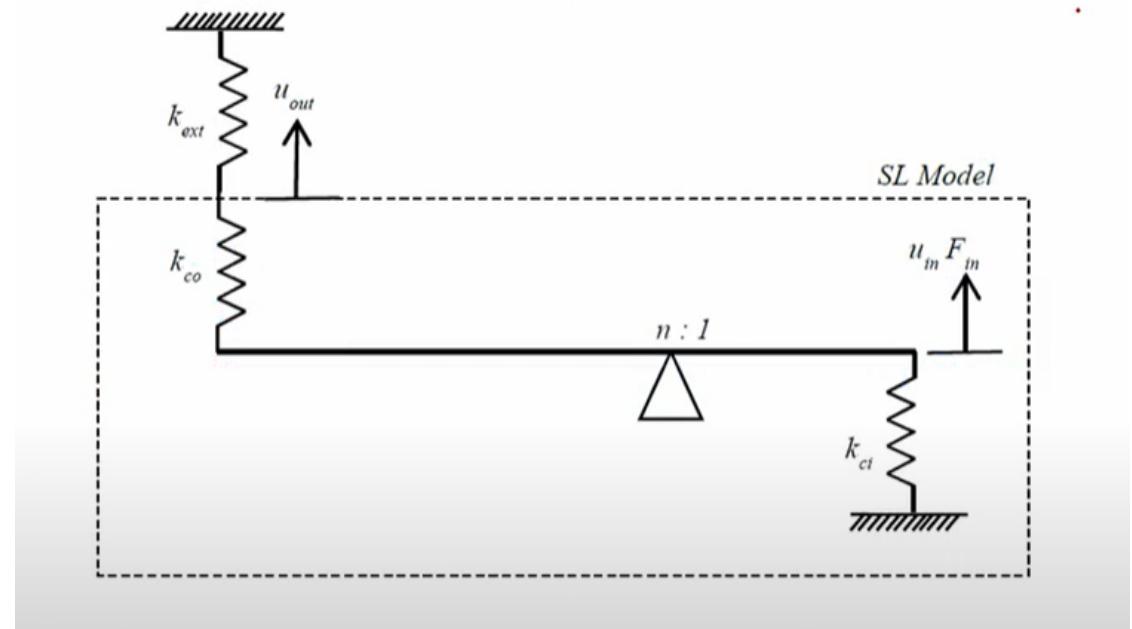
# MA

$$MA = MA_s \left( \frac{k_w}{\sigma + k_w} \right)$$

- In one of his lectures Prof. Ananthasuresh does discuss some work of a master student with the following formula's:

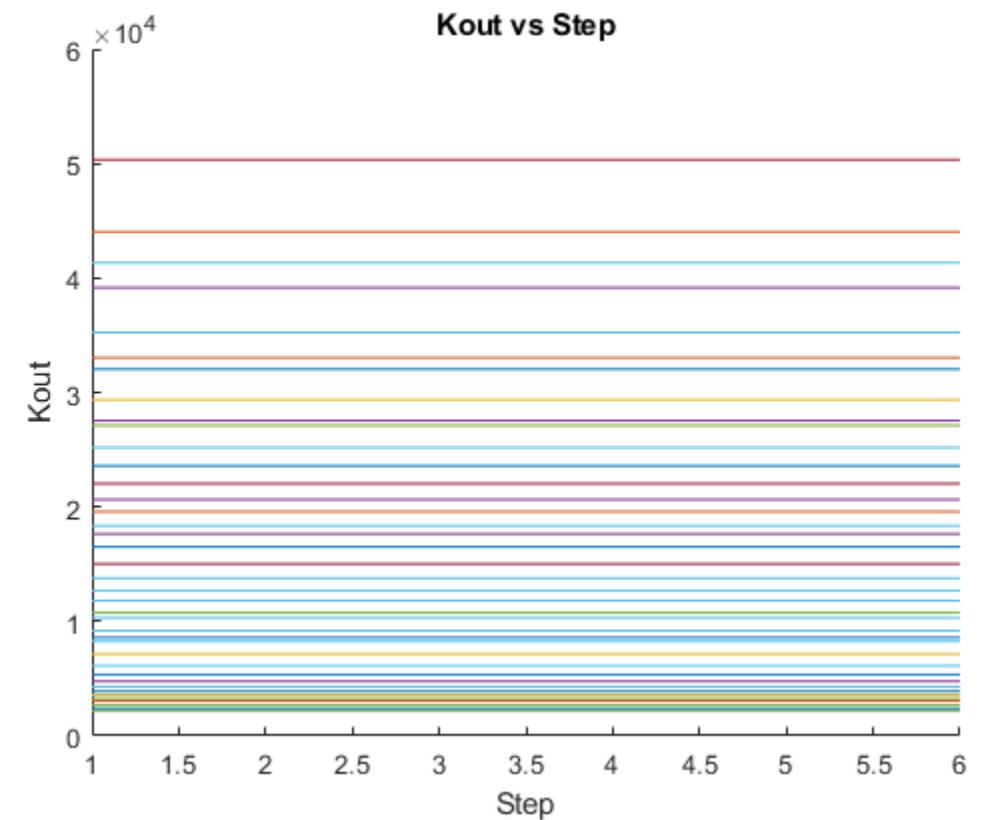
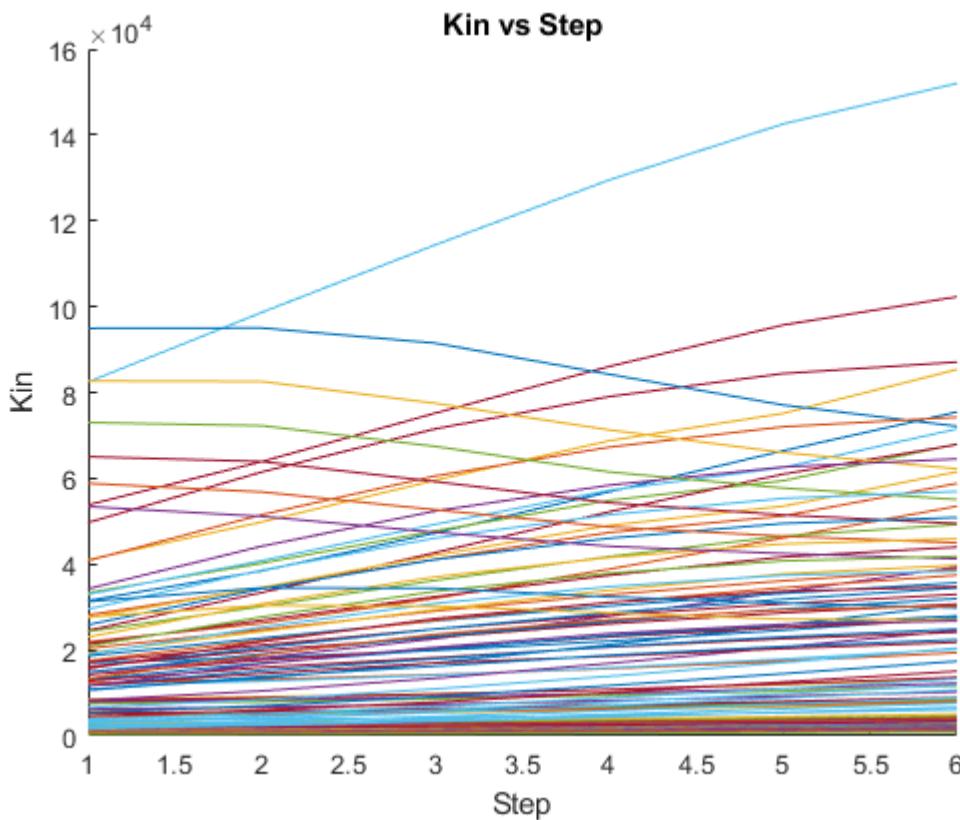
$$MA = MA_s \left( \frac{k_{ext}}{s_k + k_{ext}} \right)$$

$$\left\{ \begin{array}{l} MA_s = \frac{n k_{co}}{k_{ci} + n^2 k_{co}} \\ s_k = \frac{k_{ci} k_{co}}{k_{ci} + n^2 k_{co}} \end{array} \right\}$$

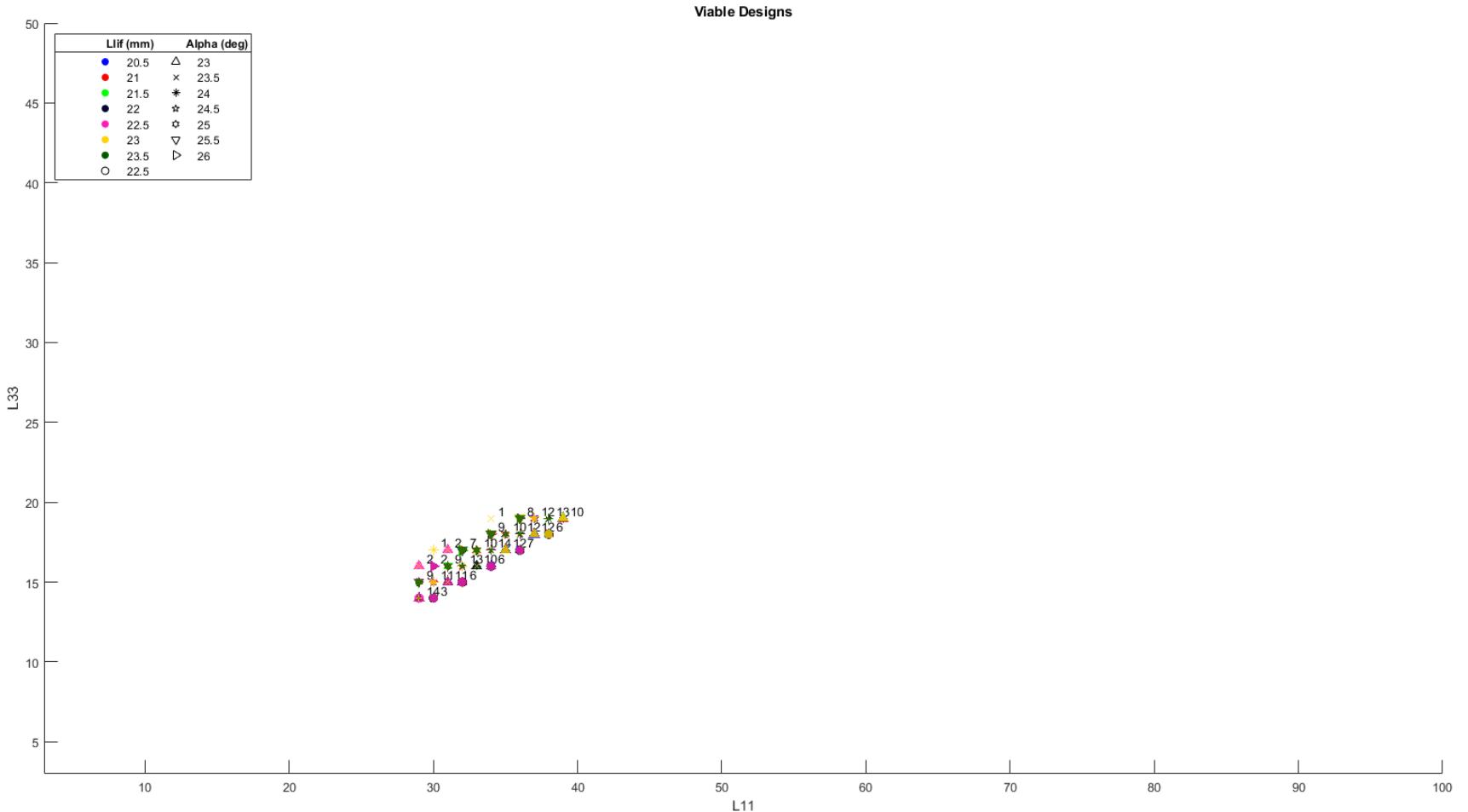


# Visual investigation

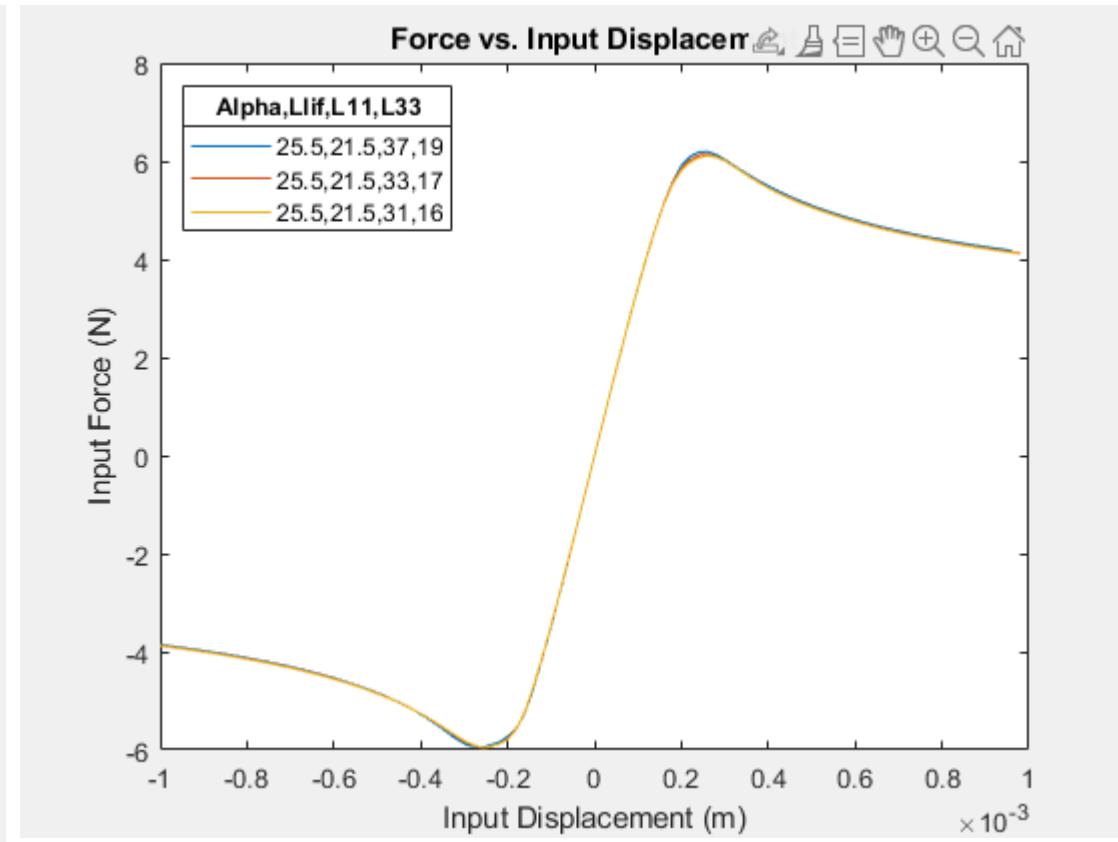
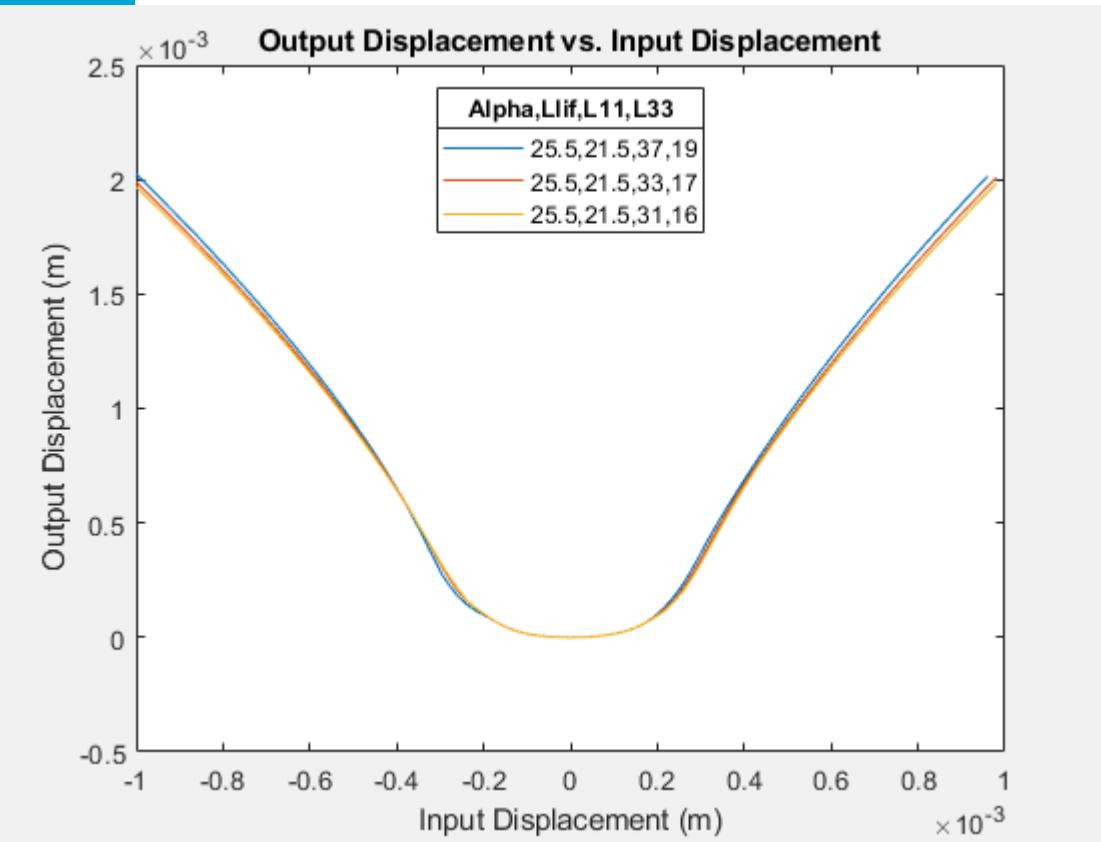
- What are some reasonable values: Now have  $0.3N$  1% discrepancy of input, as that would result in a reasonable number of solutions



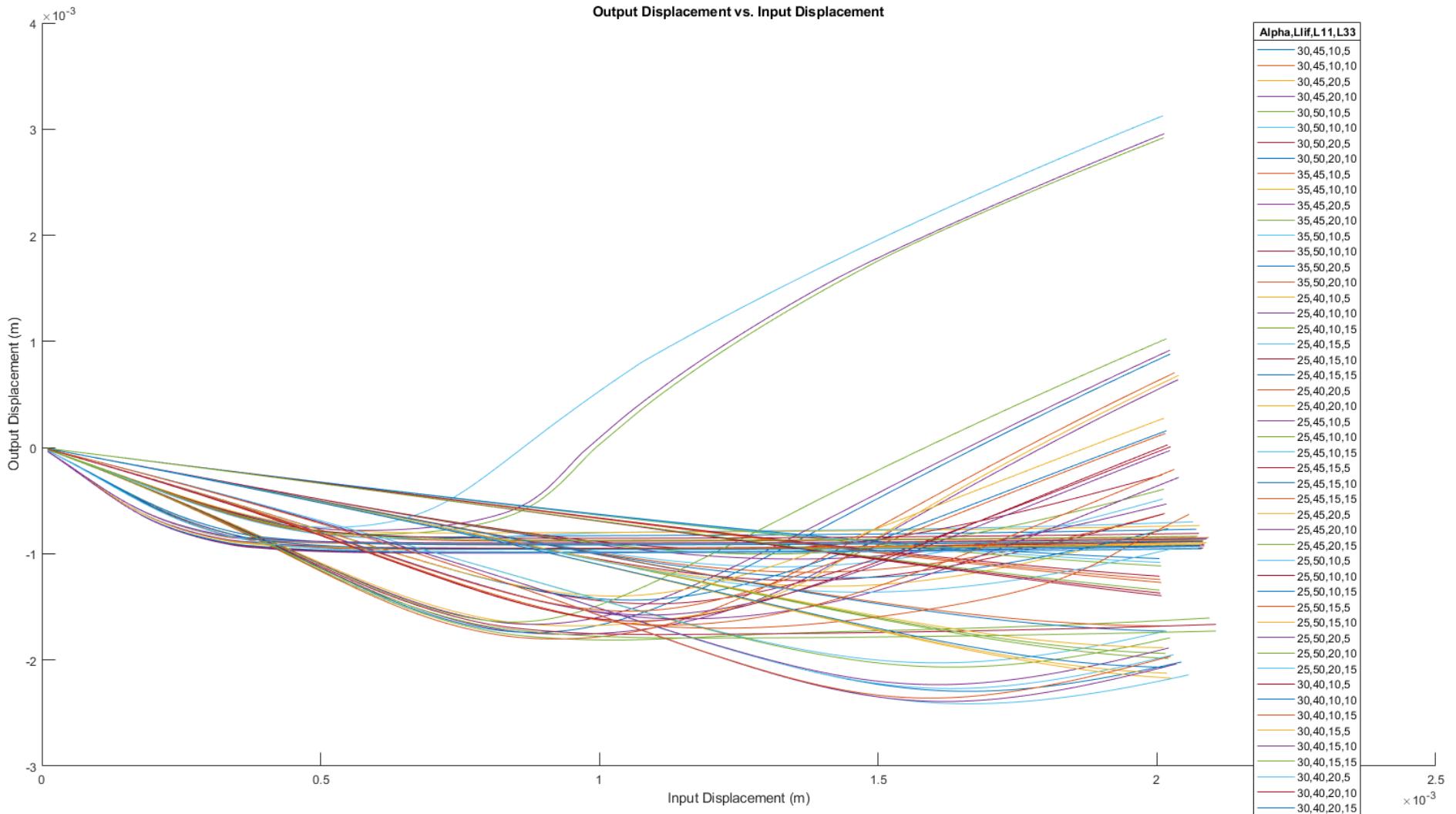
# Result after 3 iterations



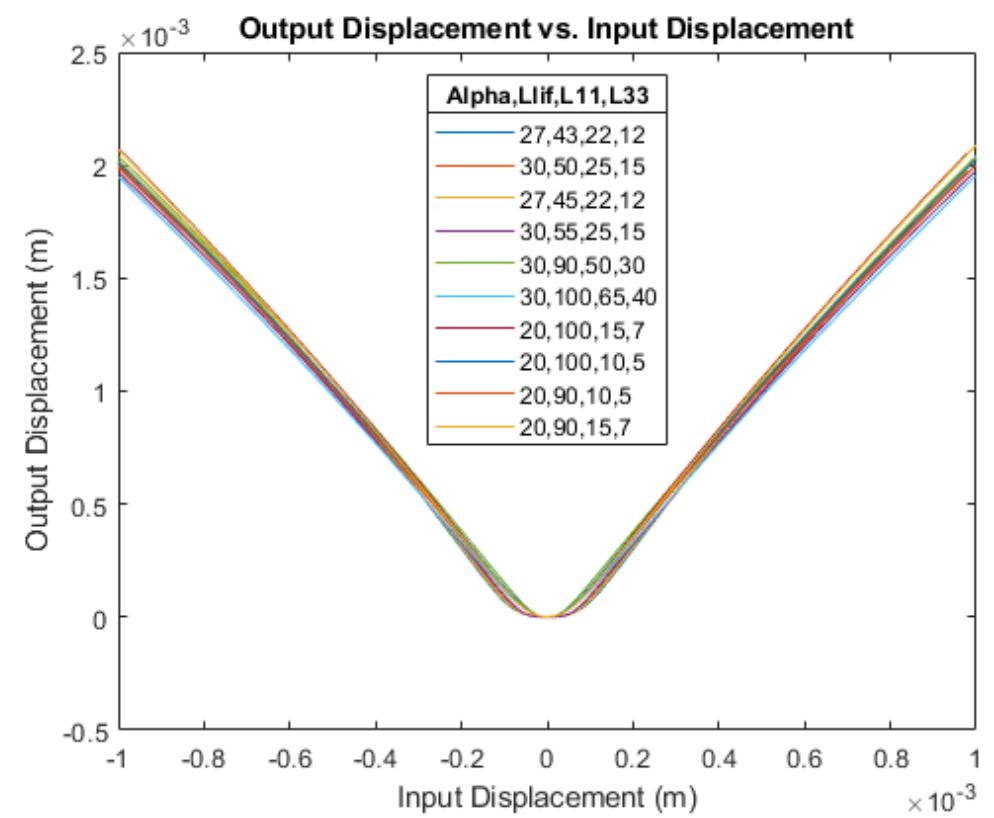
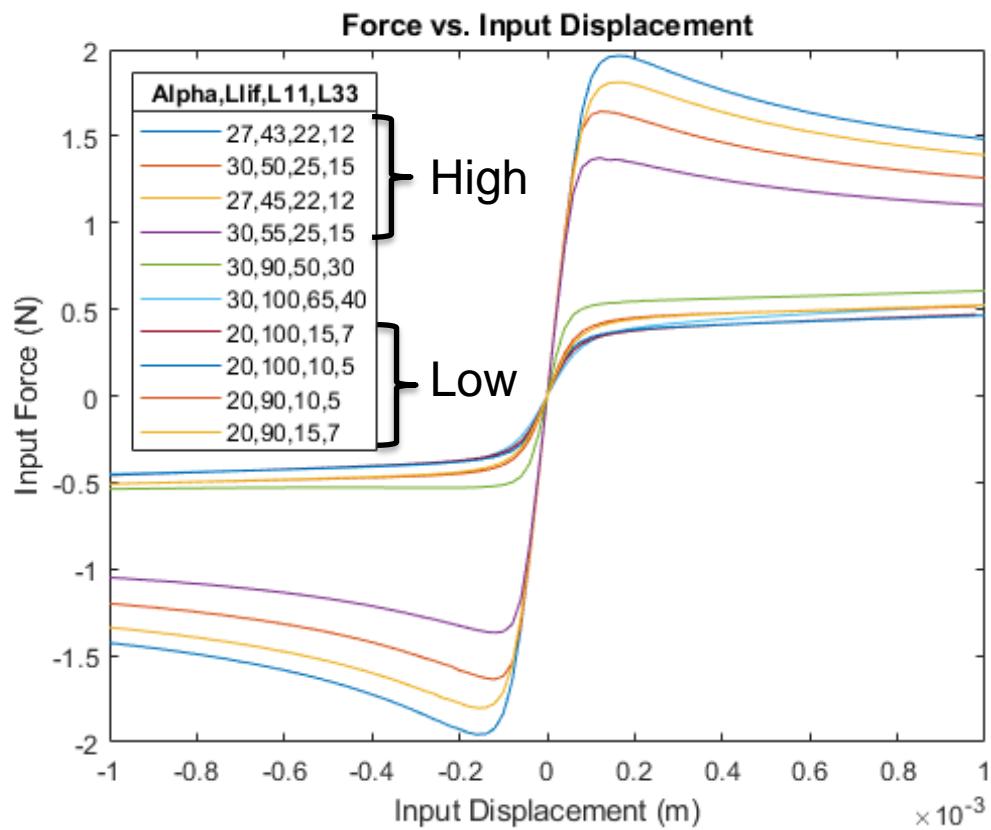
# Din/Dout and Fin/Din of best designs



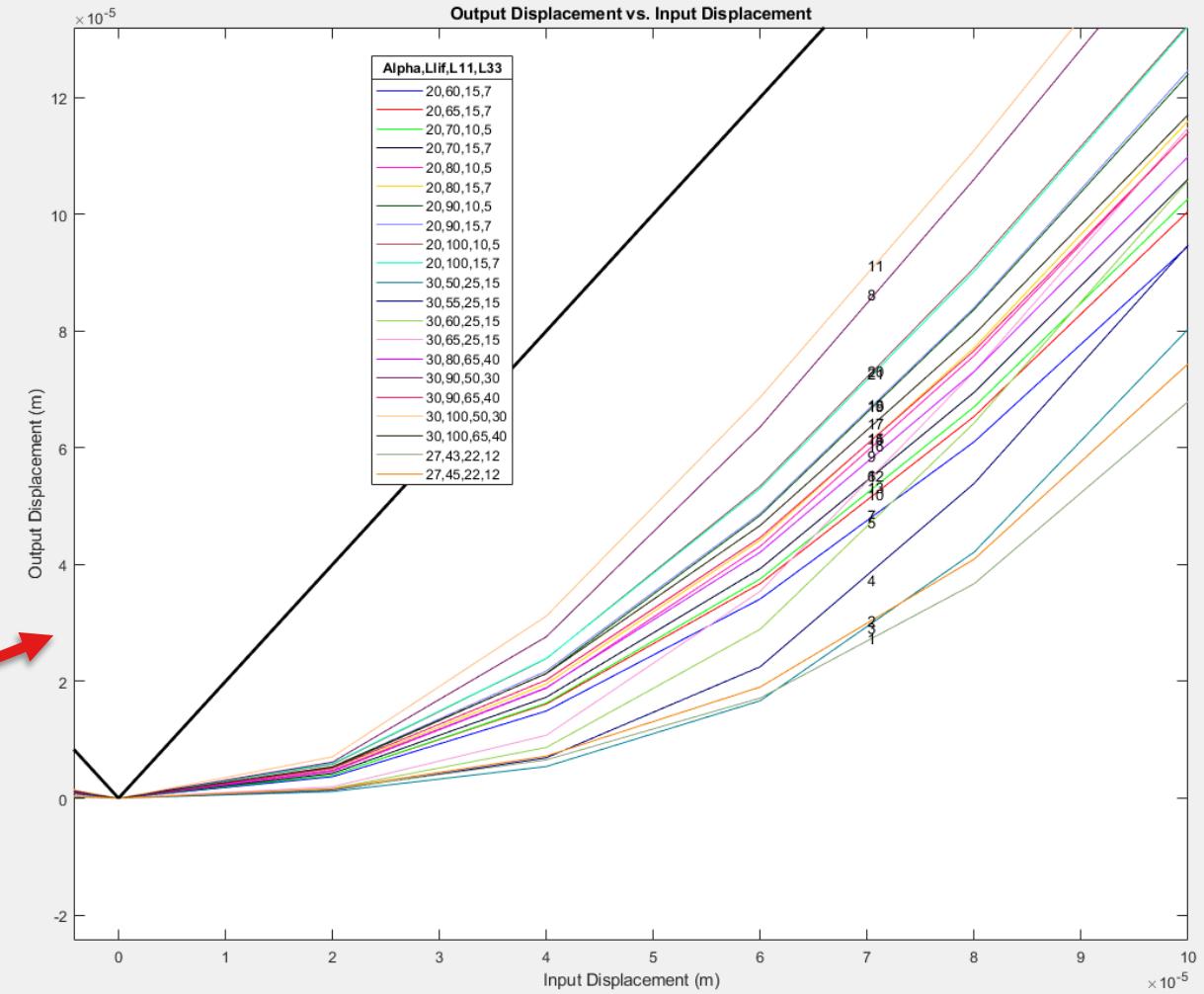
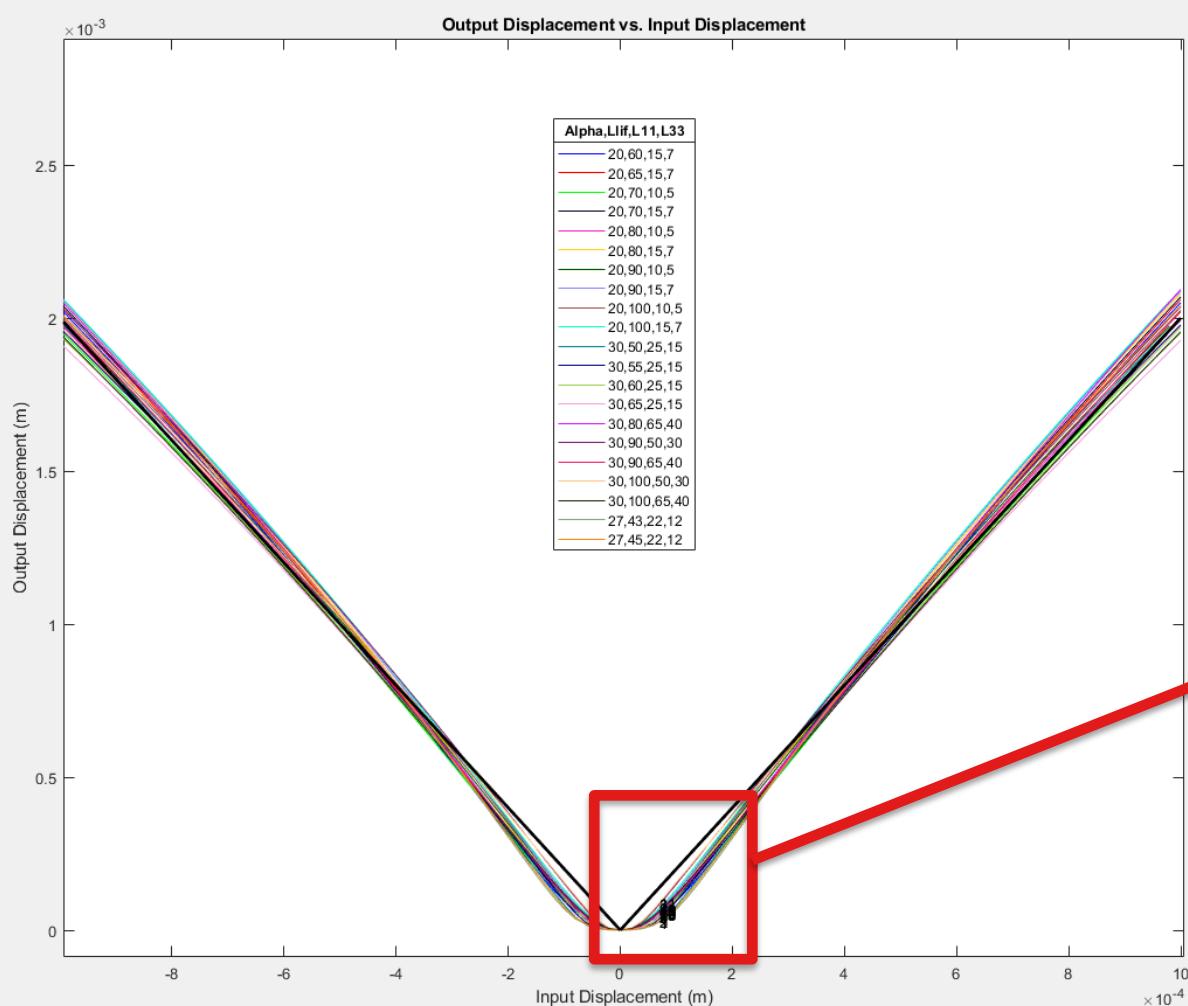
# Cosinusoidal



# Stiffness Differences

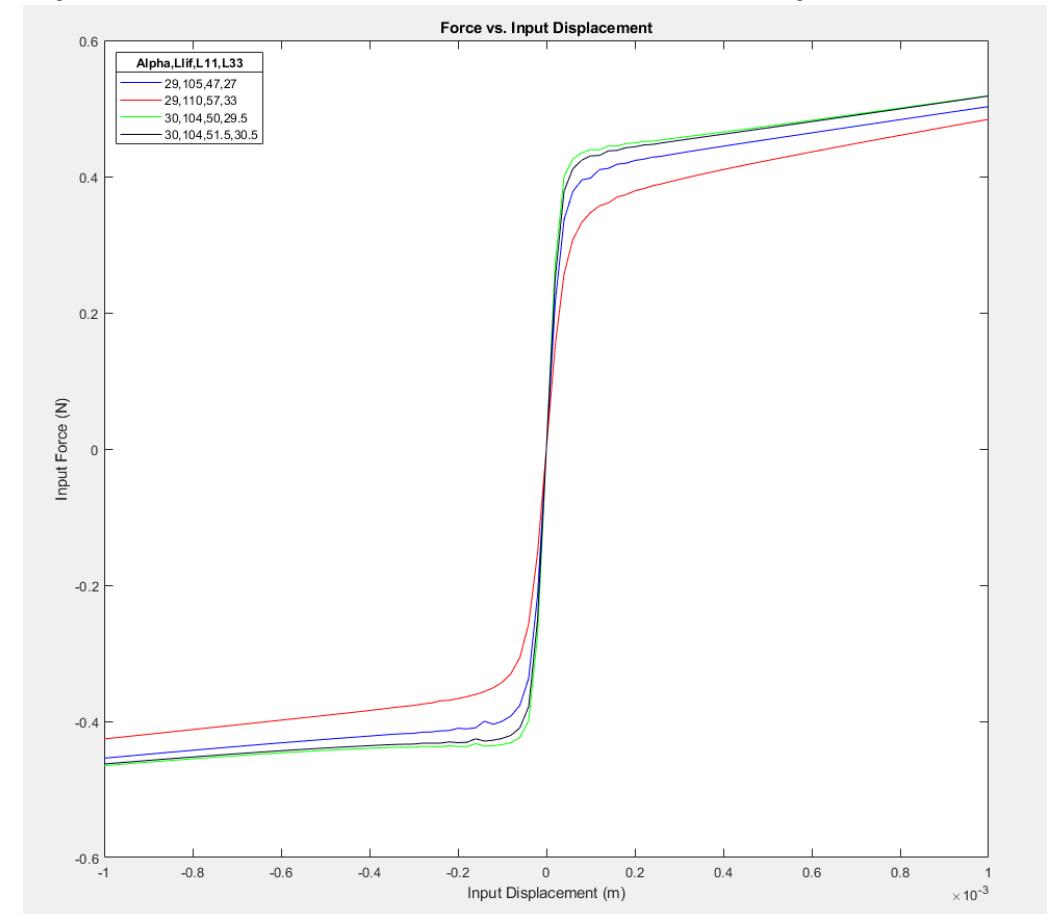
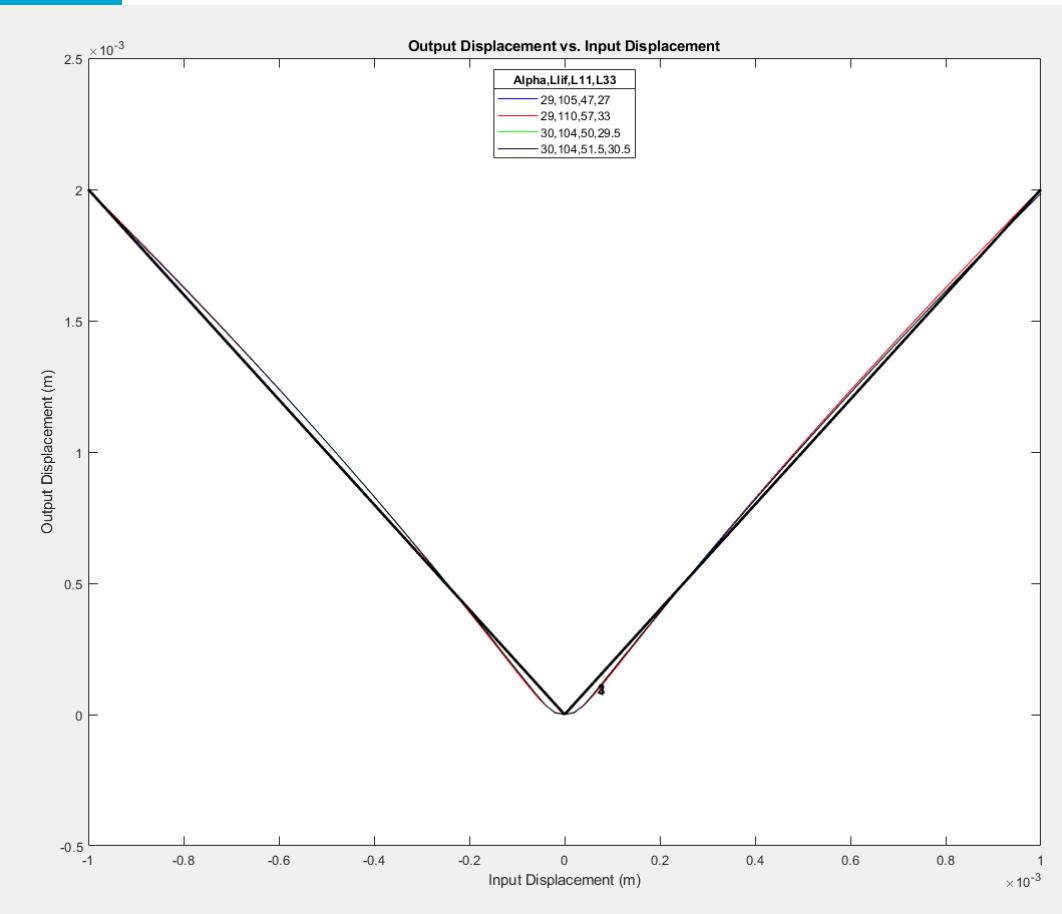


# New criteria added

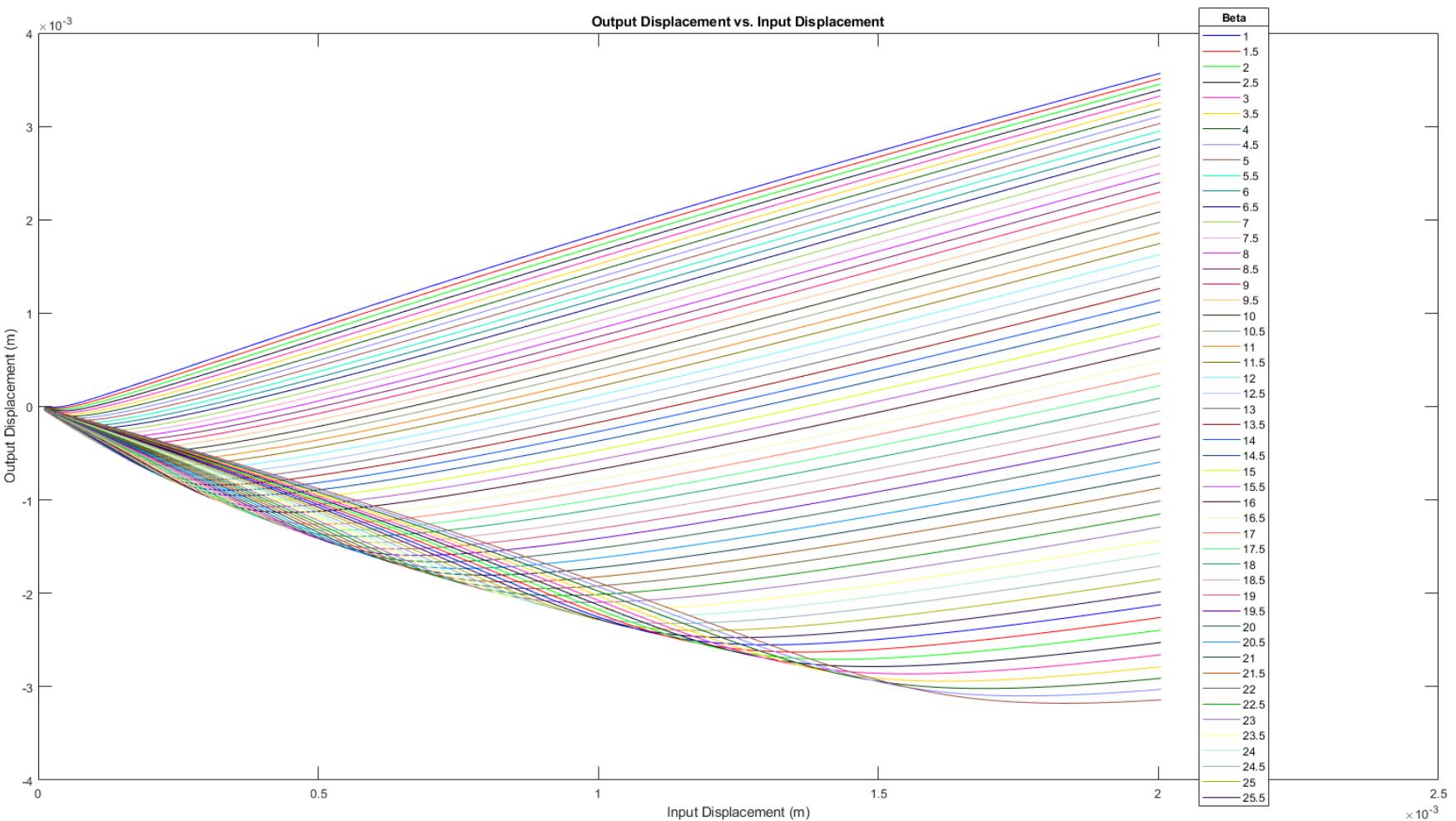


# For now

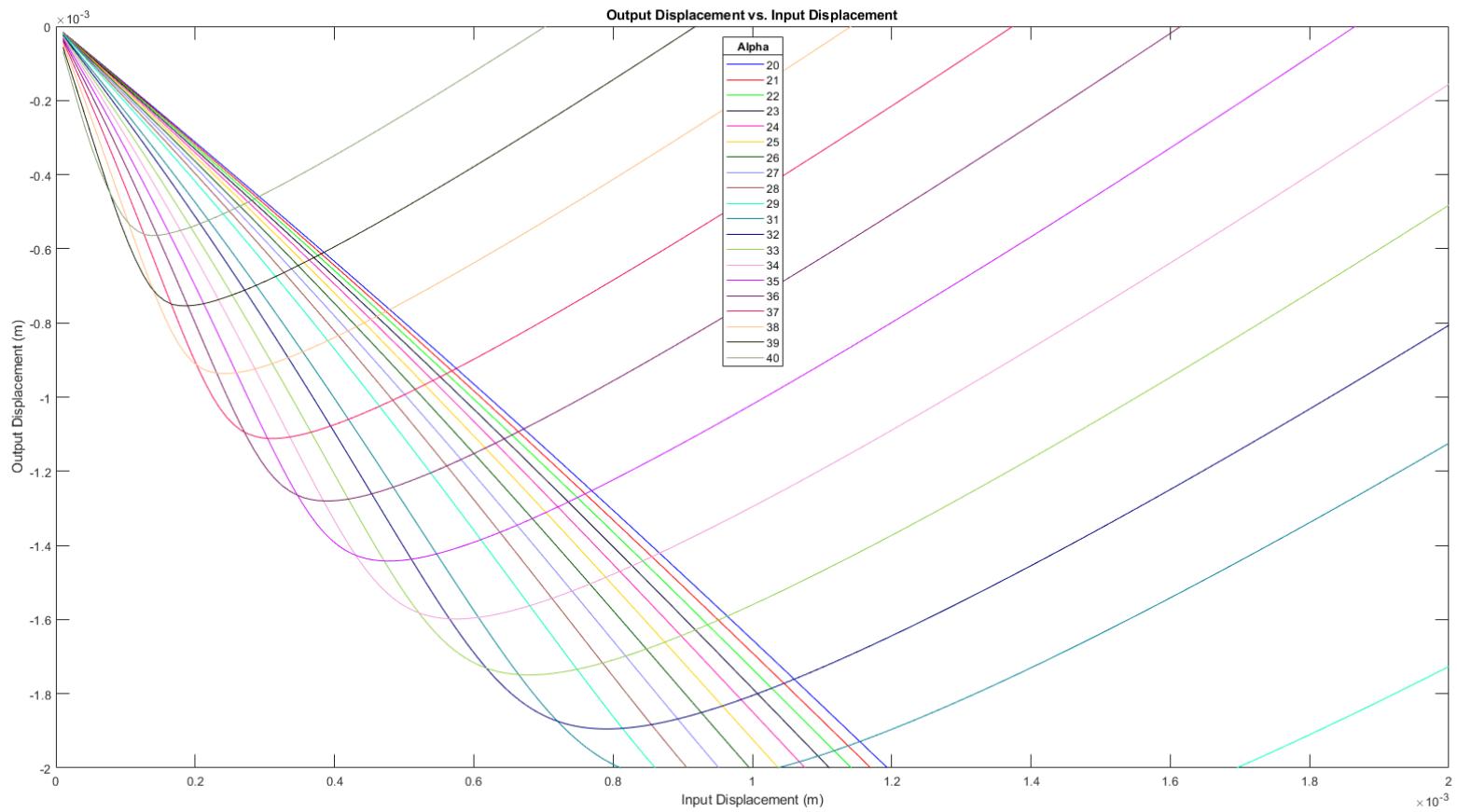
- To converge, and to make sure not too many solutions exist the criteria bounds to check linearity of the GA decrease every



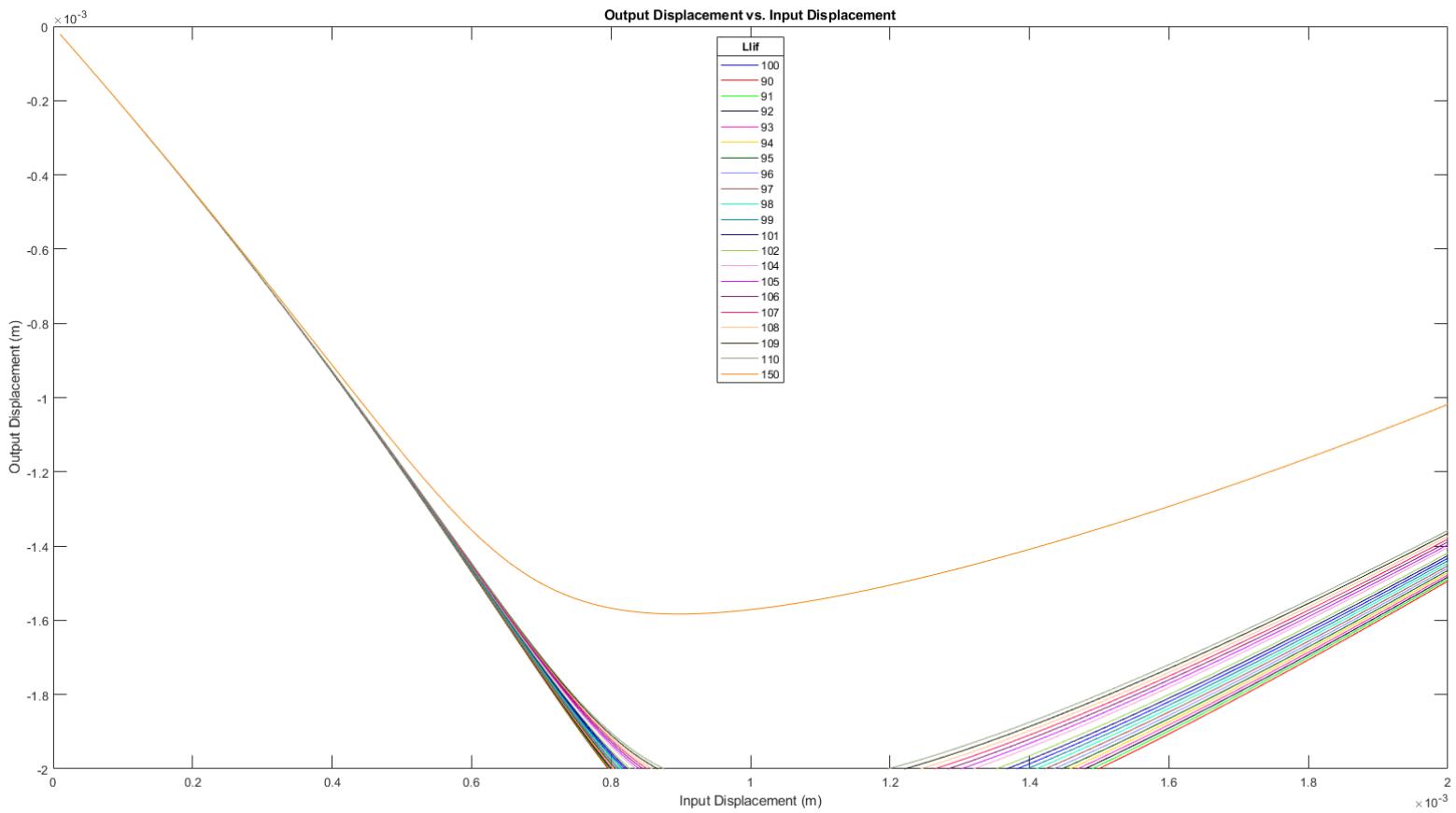
# Sinusoidal → Cosinusoidal



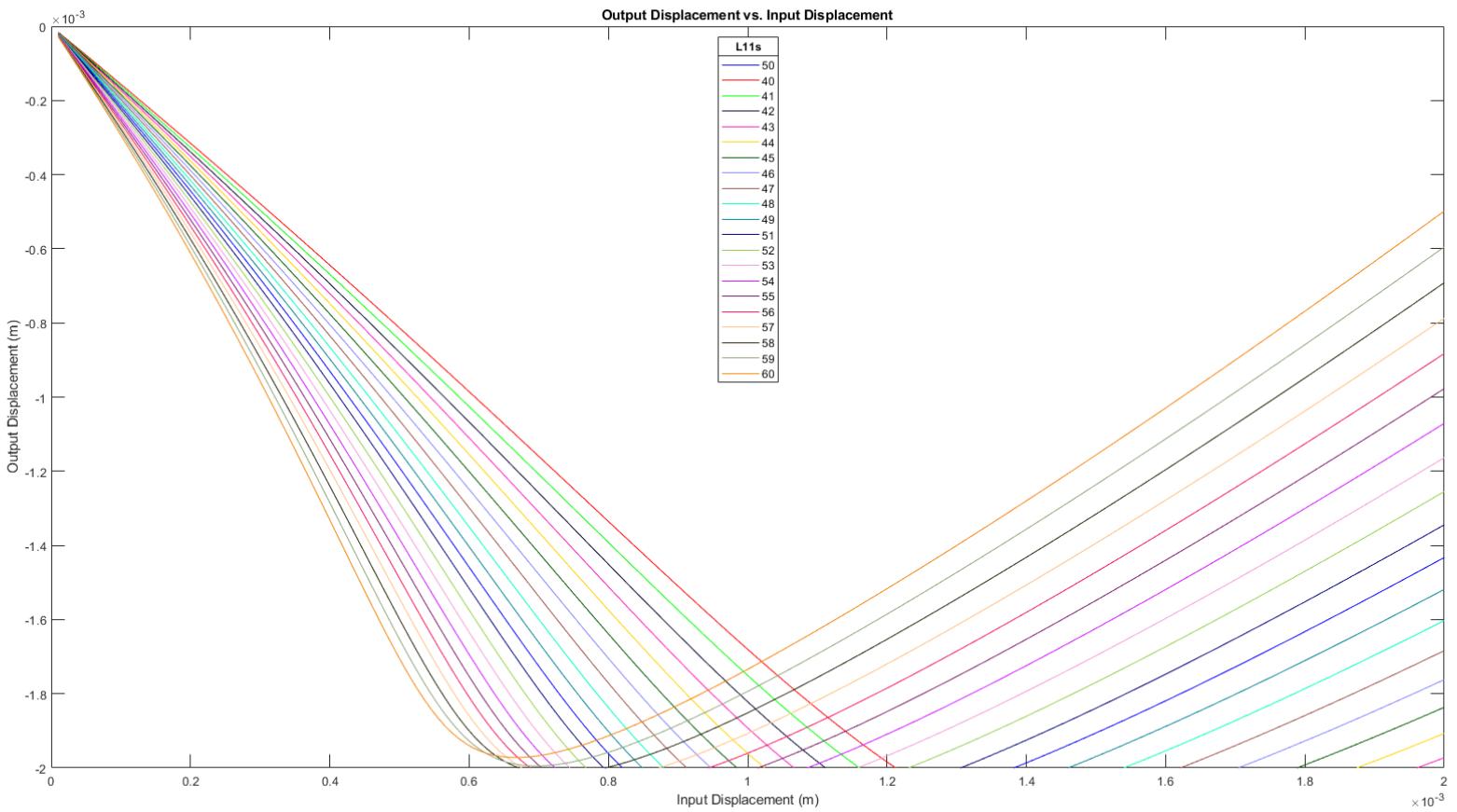
# Sinusoidal → Cosinusoidal, Alpha



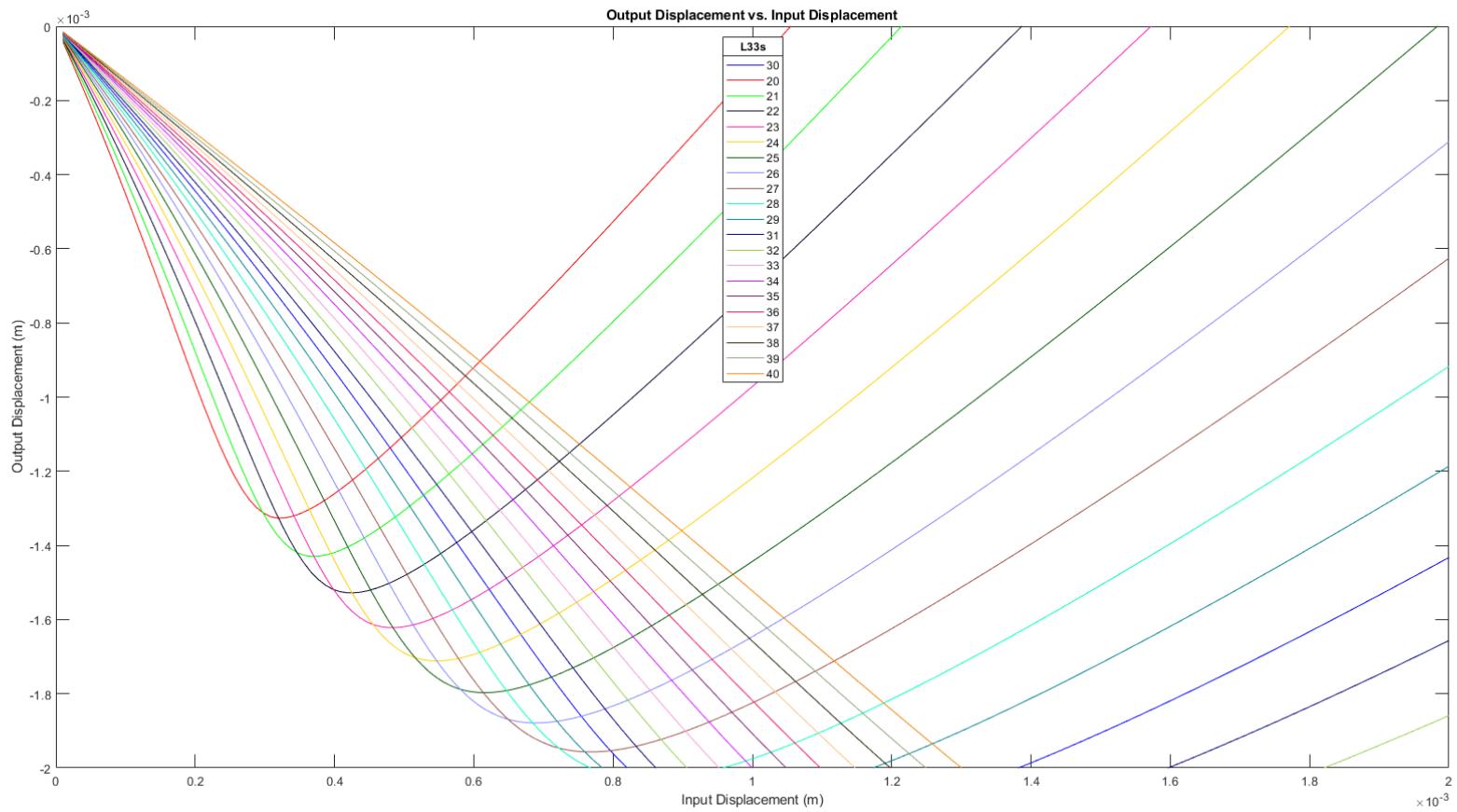
# Sinusoidal → Cosinusoidal, Llif



# Sinusoidal → Cosinusoidal, L11



# Sinusoidal → Cosinusoidal, L33



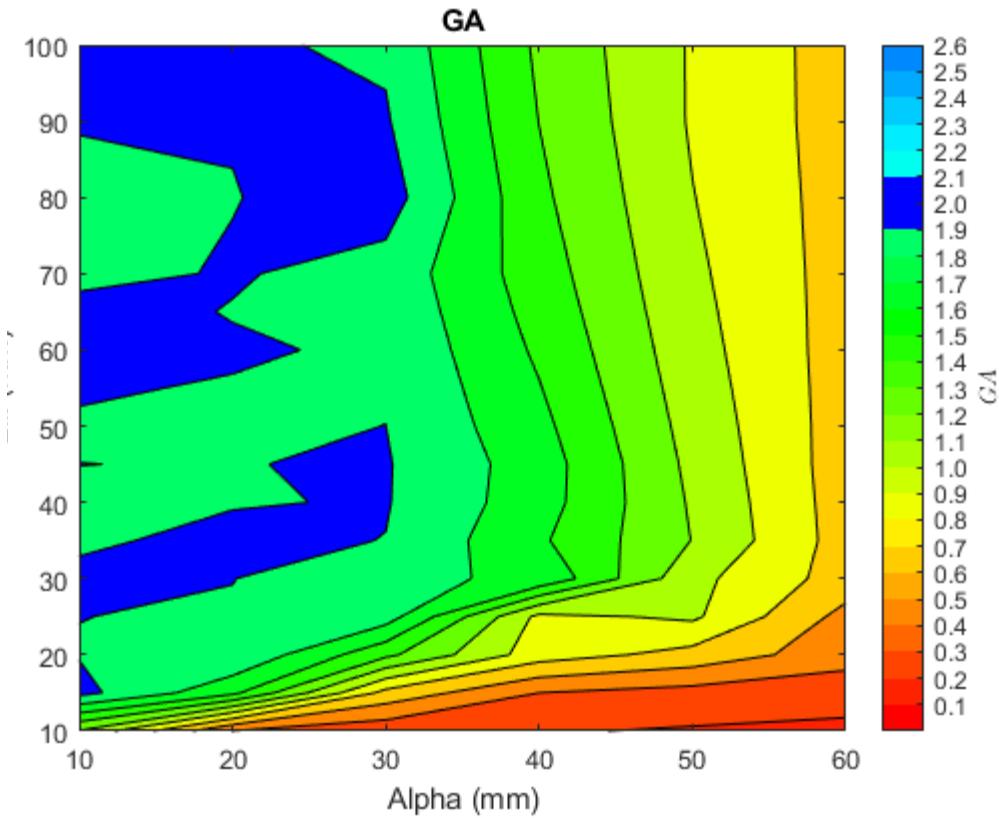
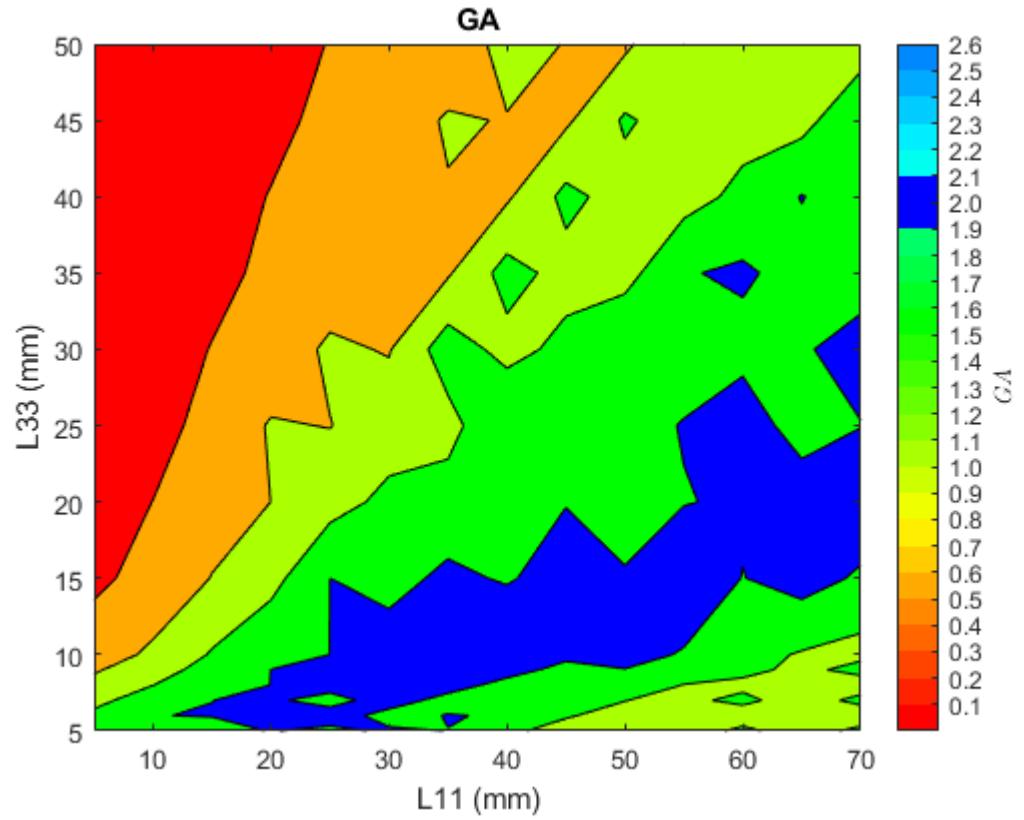
# Conclusion

- Cosinusoidal mechanism can not simply be made by picking the same parameters and altering the alpha over all flexures.

# Static Balancing

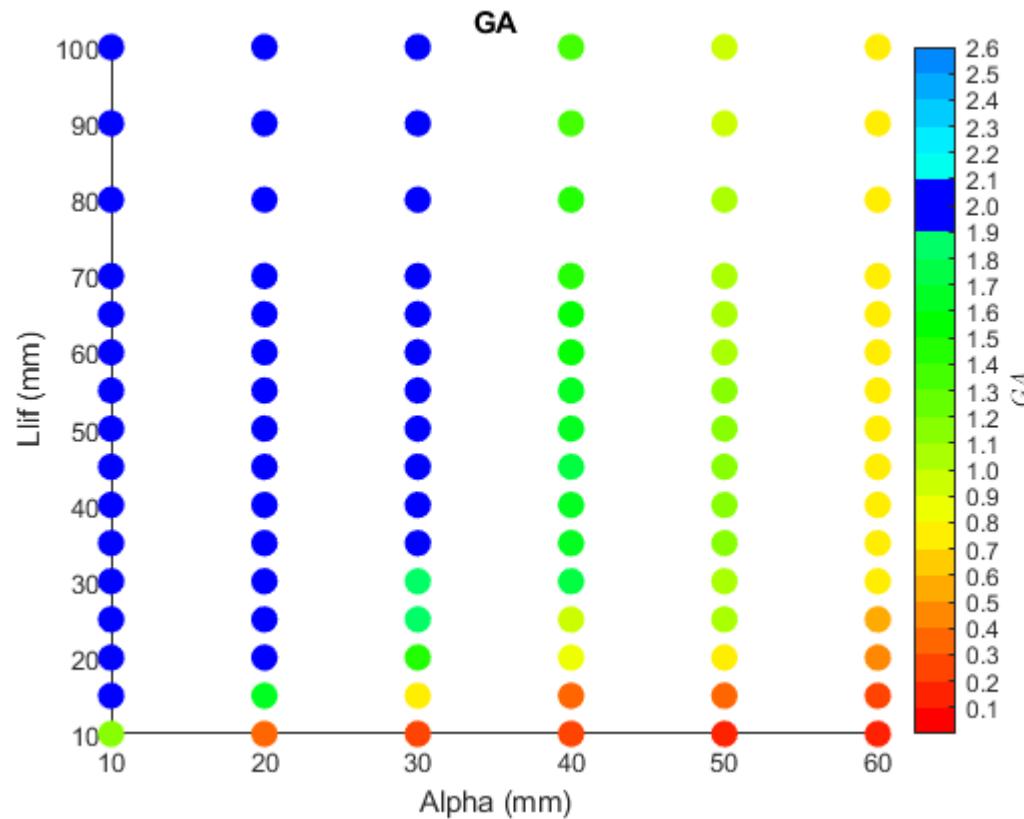
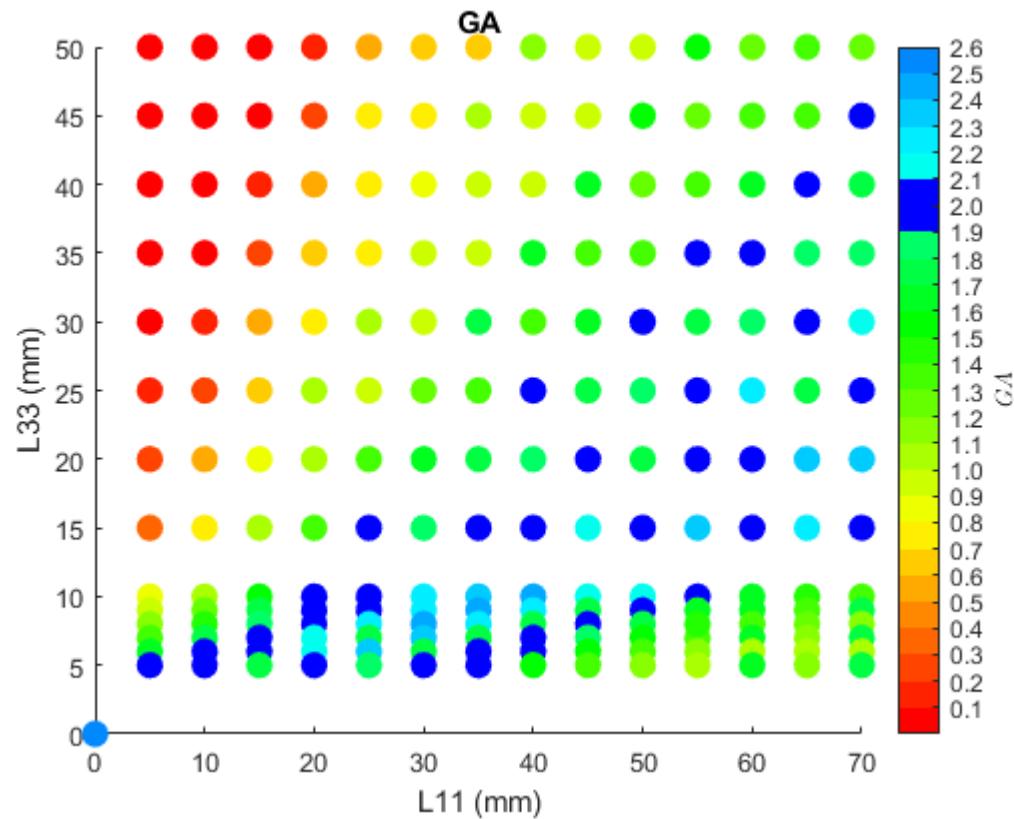
- Started searching for literature about static balancing.
- Not a lot of examples which use buckling beam
- Is there somewhere where Force-displacement diagrams can be found?

# GA Contour plot



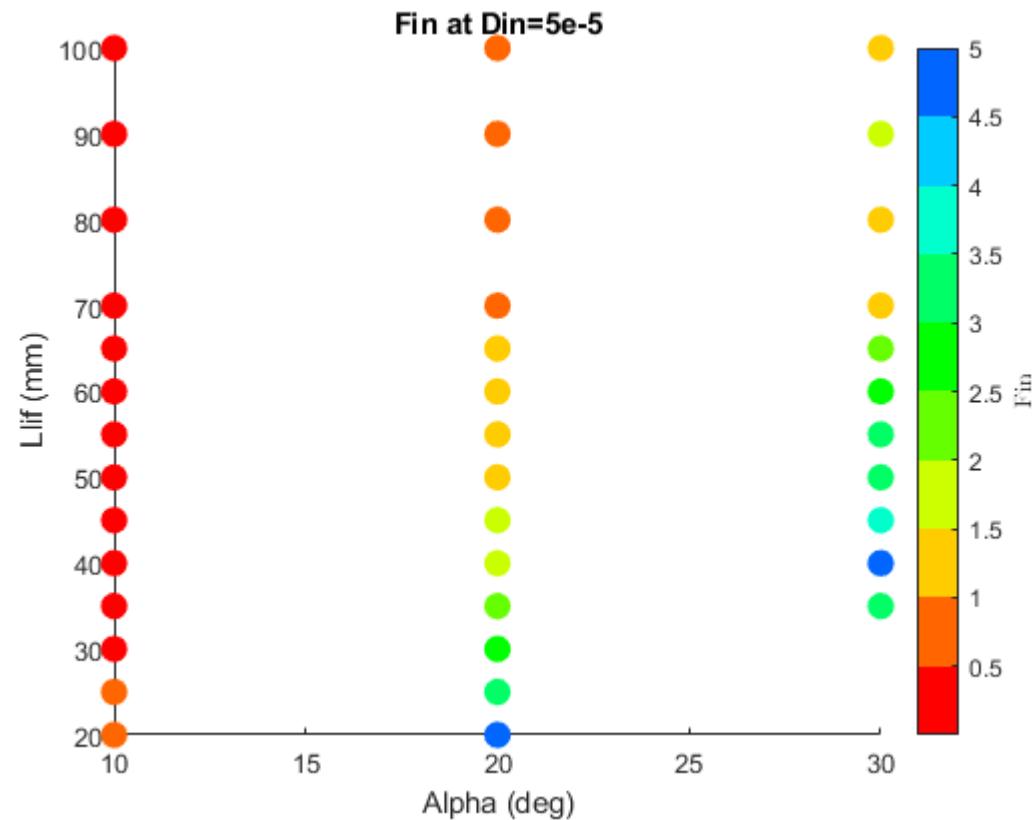
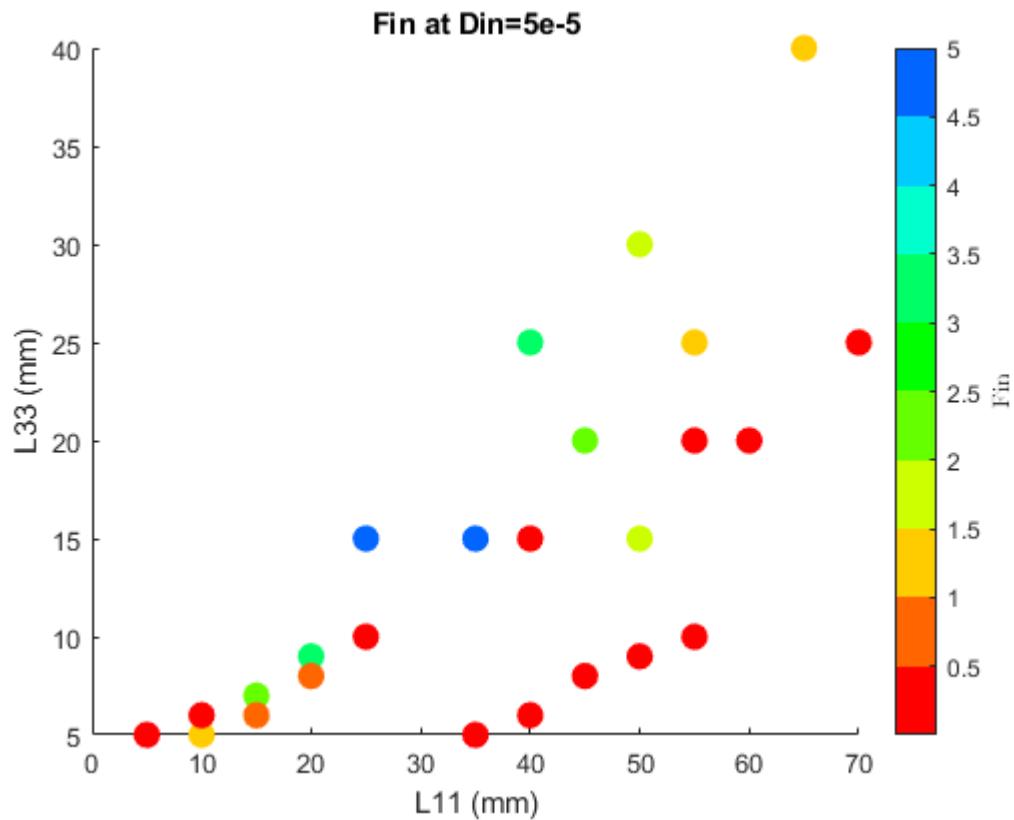
- Also made contour plot of every graph, but these do not perfectly show the behaviour due to the lack of datapoints and inter/extrapolation in between these.

# GA



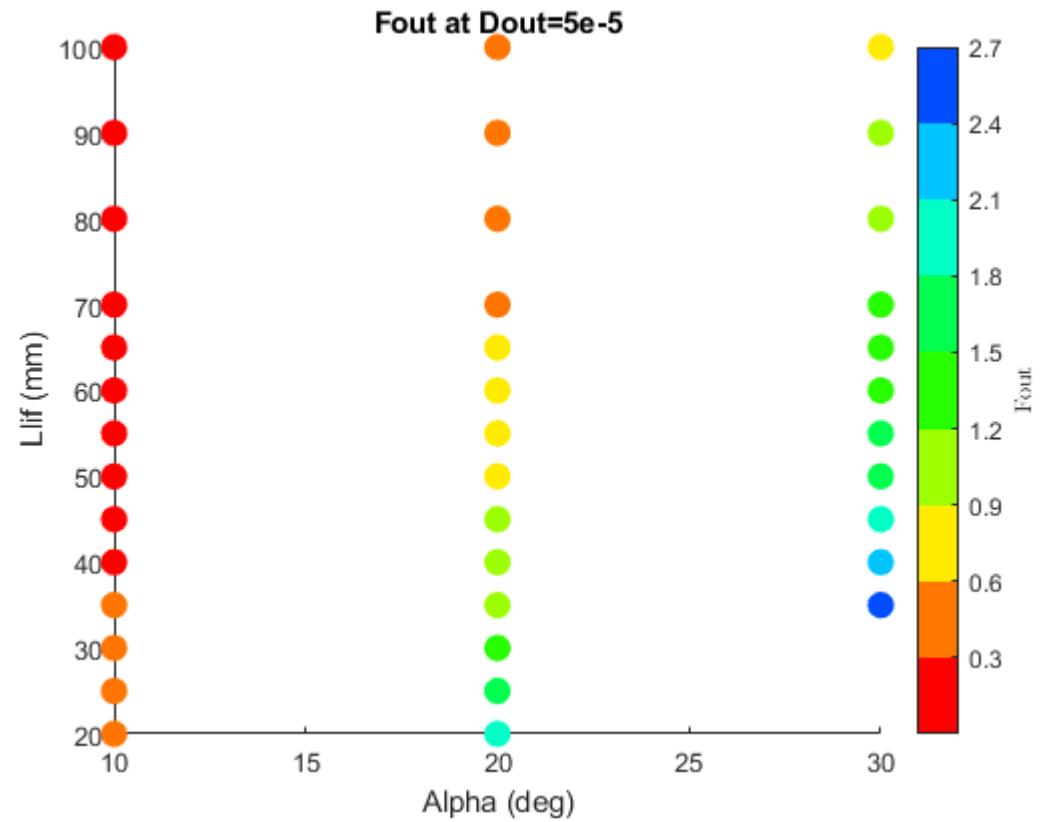
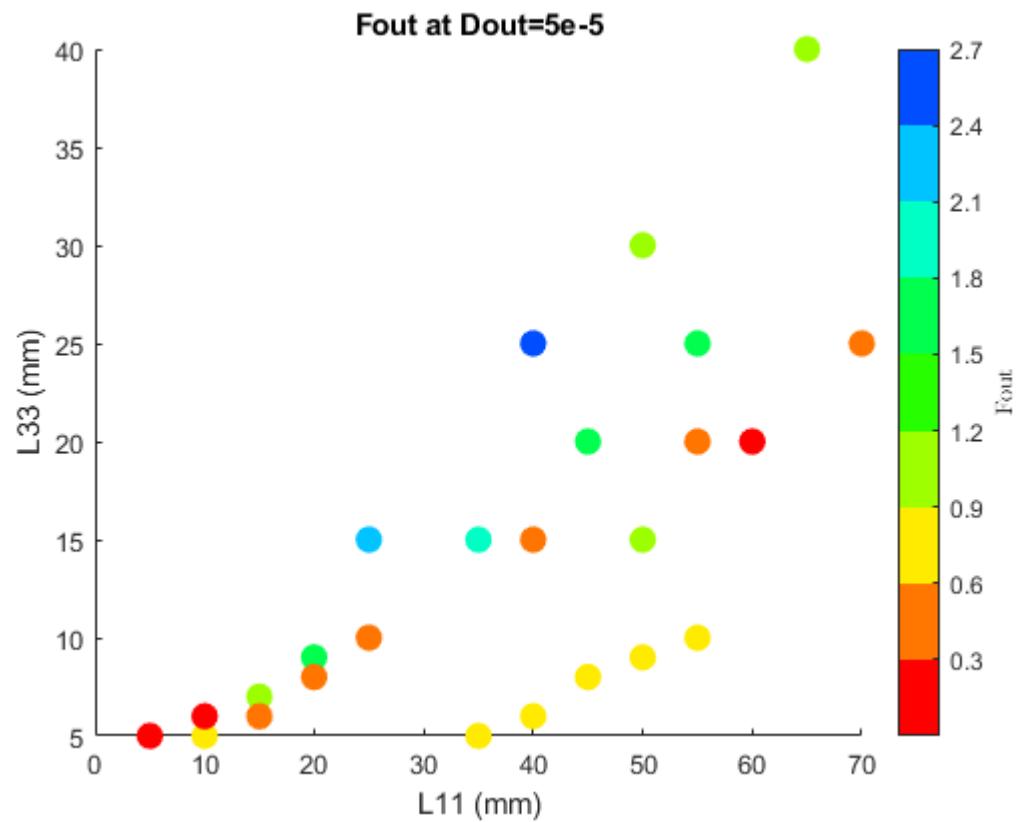
- Best points are taken to plot the graphs, (here: closer to 2 the better). Otherwise, there is overlap and only the last points plotted are visible
- This also means that you can not just take the best results from the two graphs and combine these.

# Kin

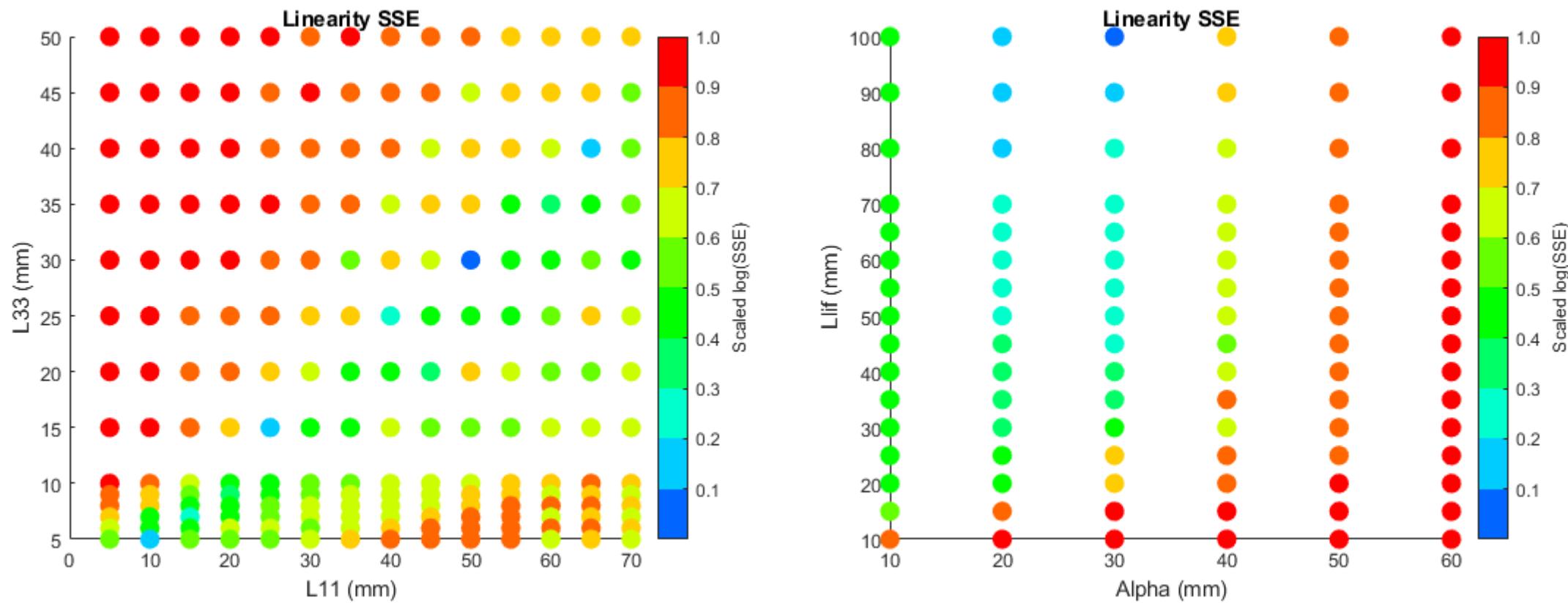


- Only solutions with  $1.9 < GA < 2.1$  are investigated and plotted
- Force is checked at 10% of total displacement

# Kout

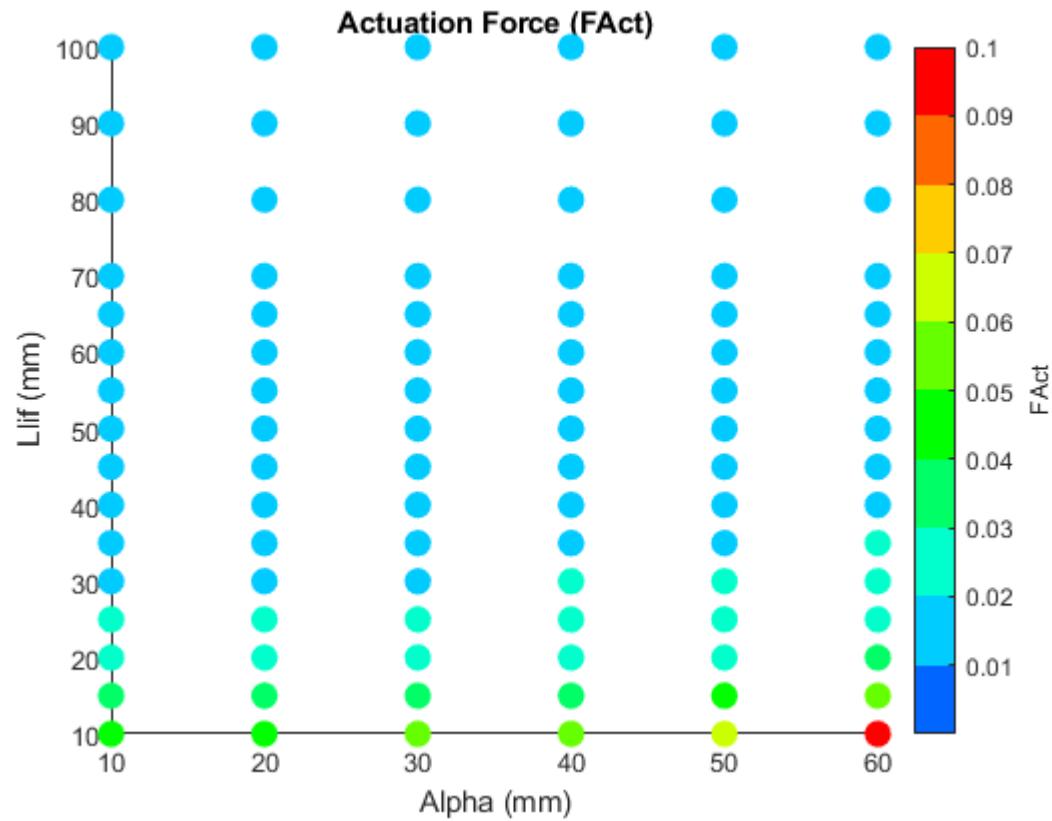
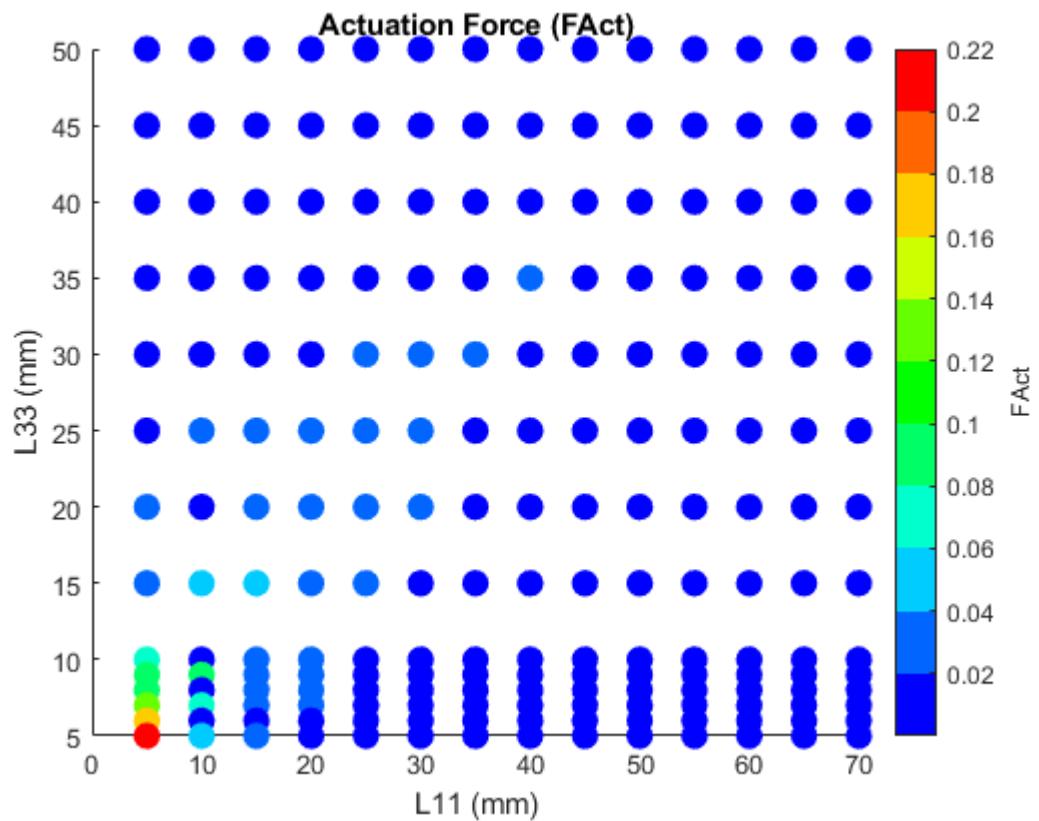


# Linearity

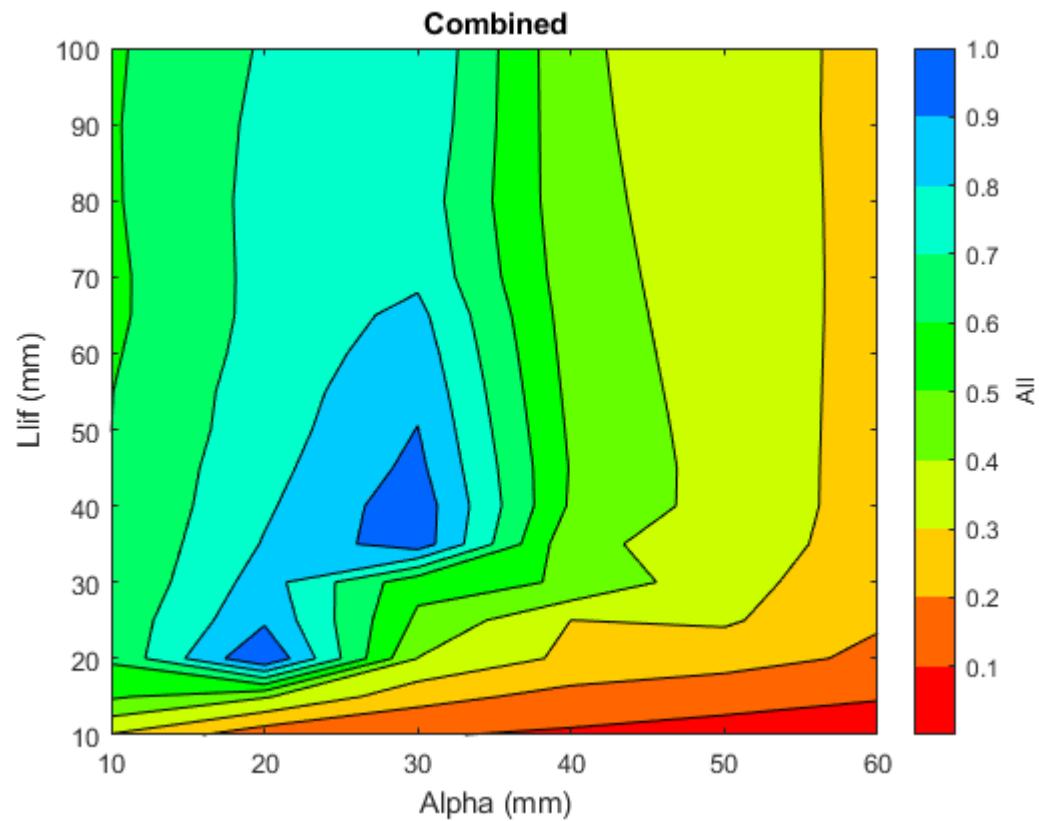
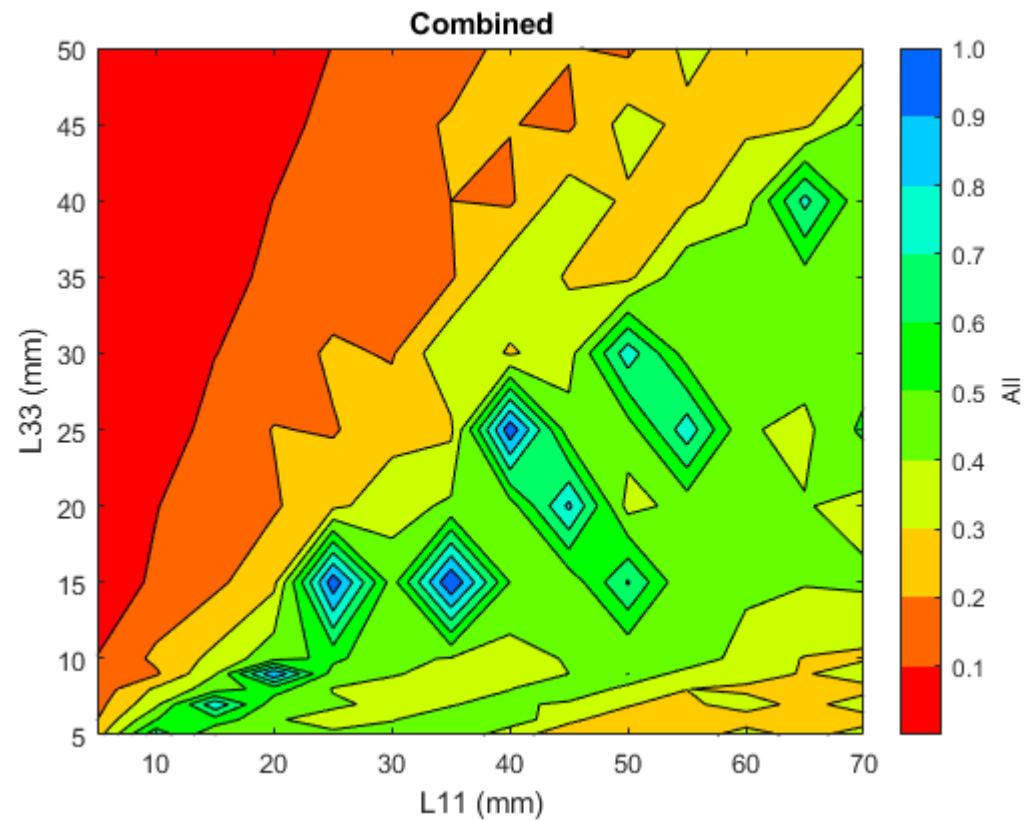


- Used sum of square error (SSE) to calculate how close a system was to linearity.
- Log10(SSE) to get order of magnitude of error
- Rescaled from 0 to 1

# Kact

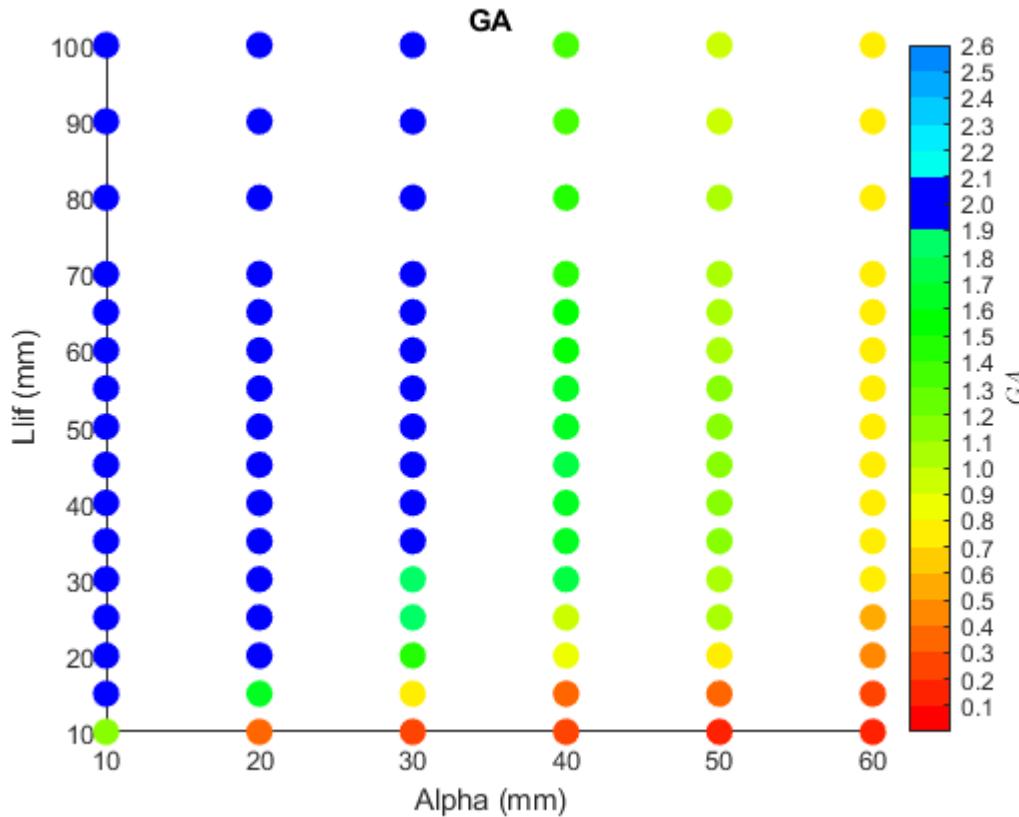
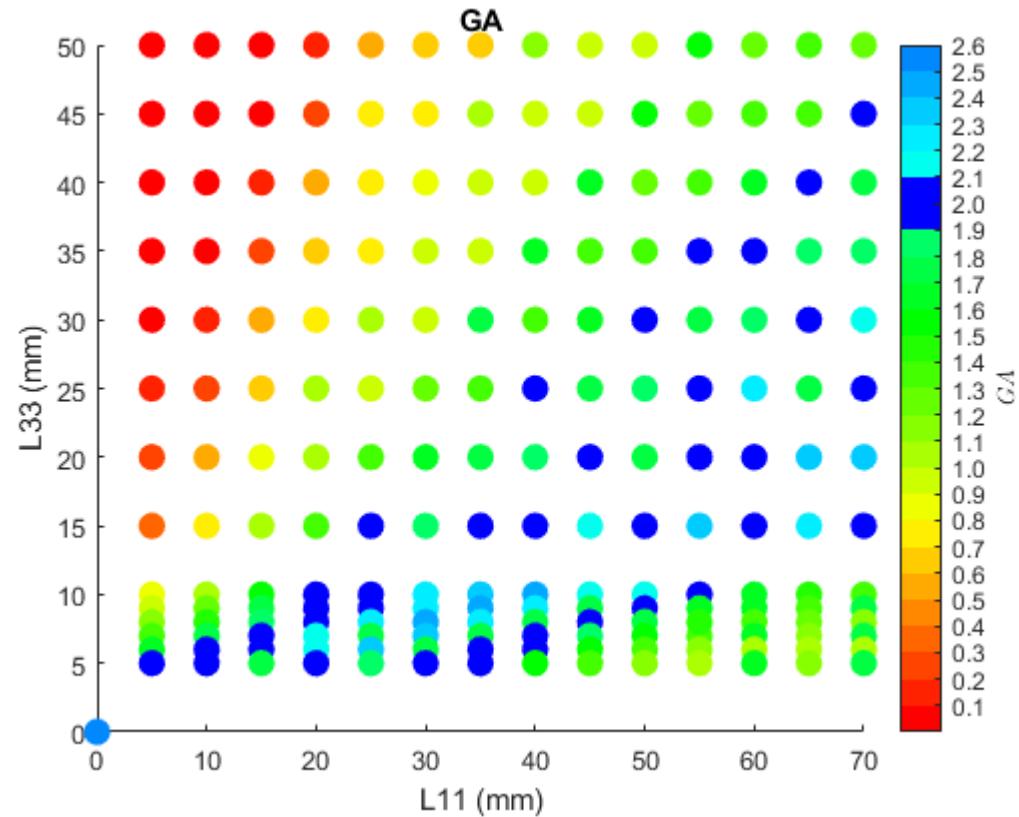


# Combined



- Just to get an idea, weights per criteria are arbitrary  
(1,0.9,0.8,0.7,0.6) : (GA, Kin, Kout, Lin, Kact)

# GA

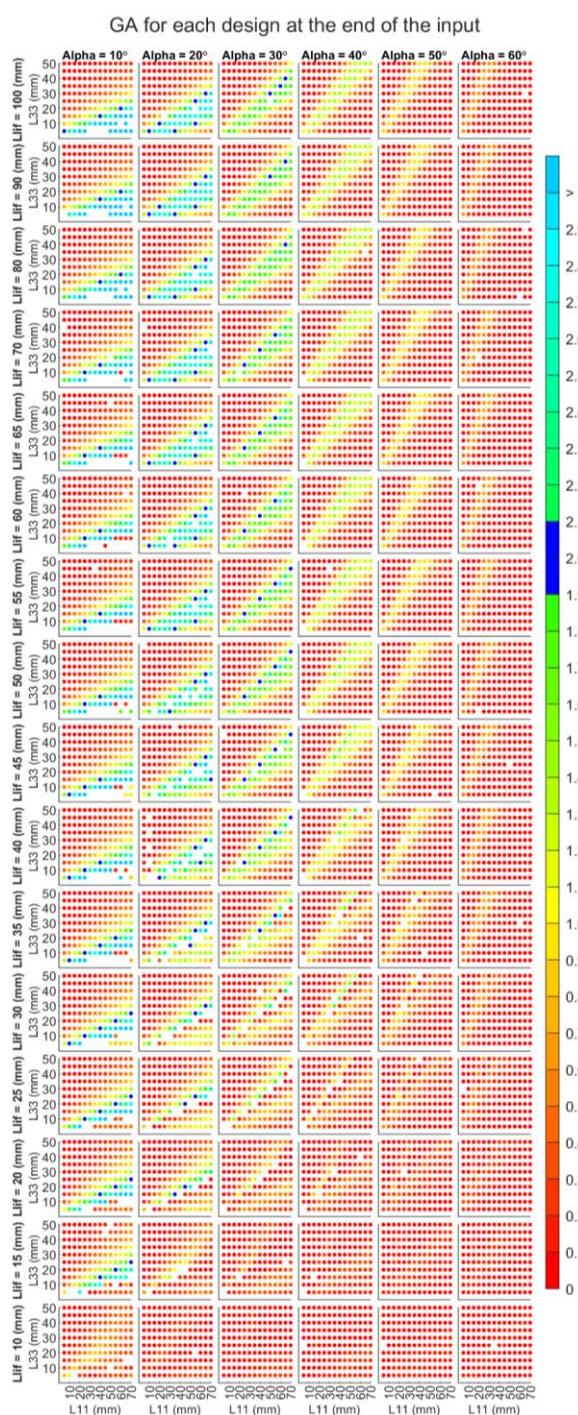


This was data first, but problem with showing 4-parameters + 1 criteria. As each point in this graph is essentially the best point chosen from all the points with the same  $L_{11}$  and  $L_{33}$  (in the first figure).

Also, more points in between the data were needed to get a better idea of where the optimum is.

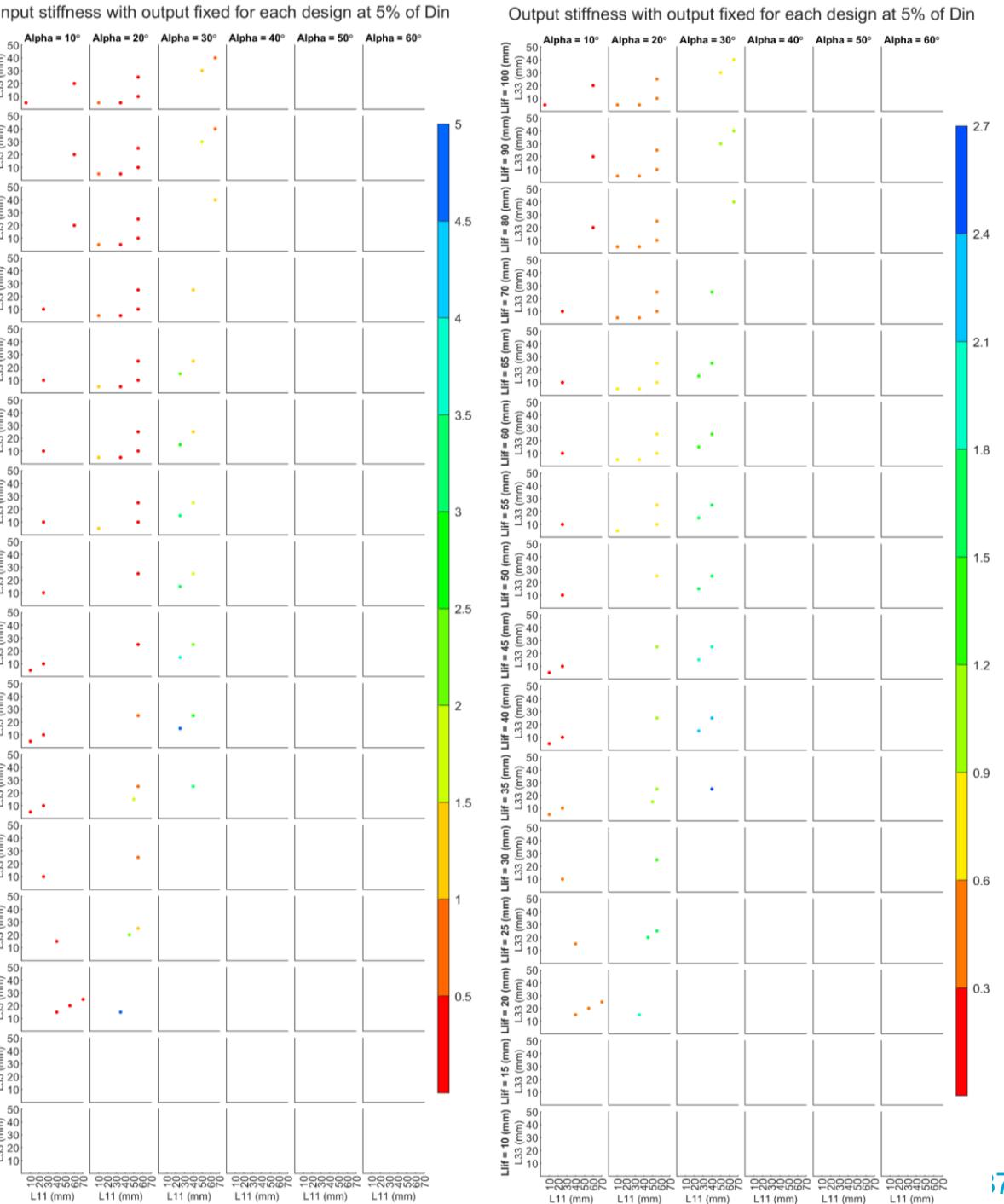
# GA

A solution I used, to show all dimensions was to make subplots and use the extra axes to show two more dimensions. The figures shown here are using the old data.



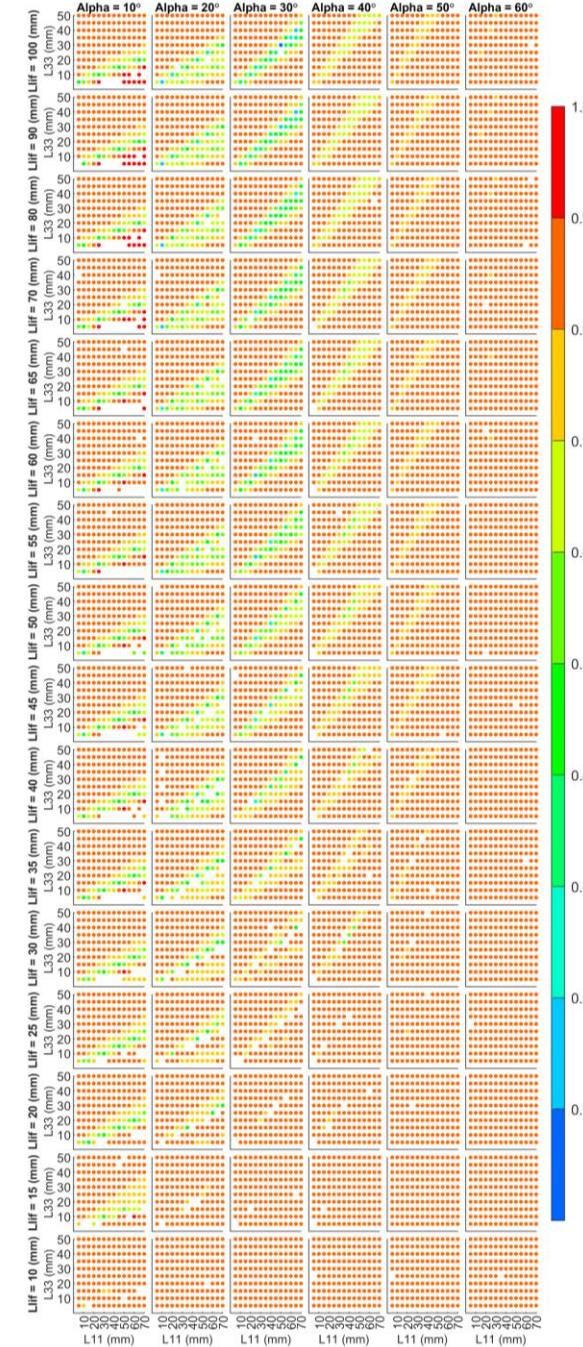
# Fin/Fout

Here only the points that have a  $1.9 < GA < 2.1$  are shown. The points are taken for the input force/ output force at 5% of the total input displacement.



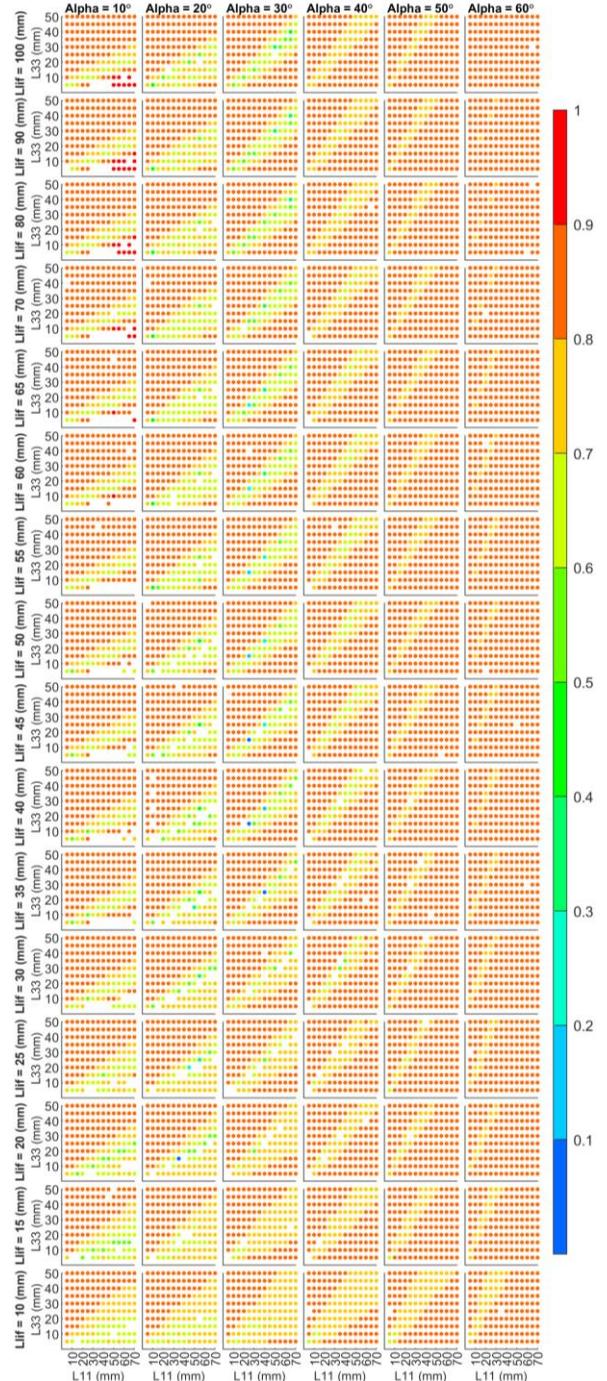
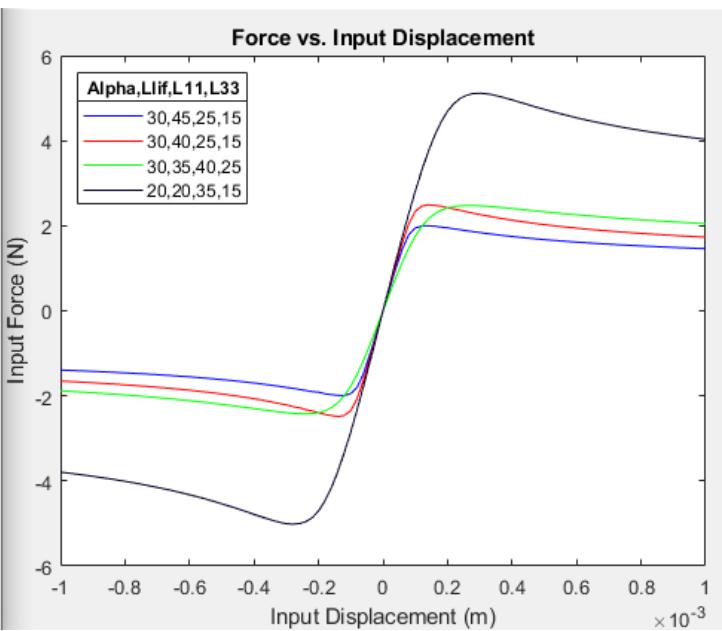
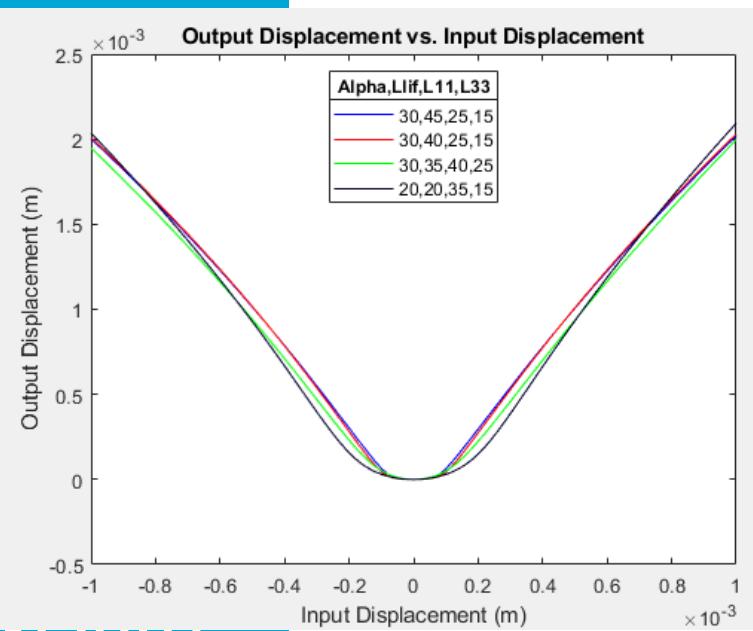
# Linearity

Scaled log(SSE) for each design at the end of the input



# Weights?

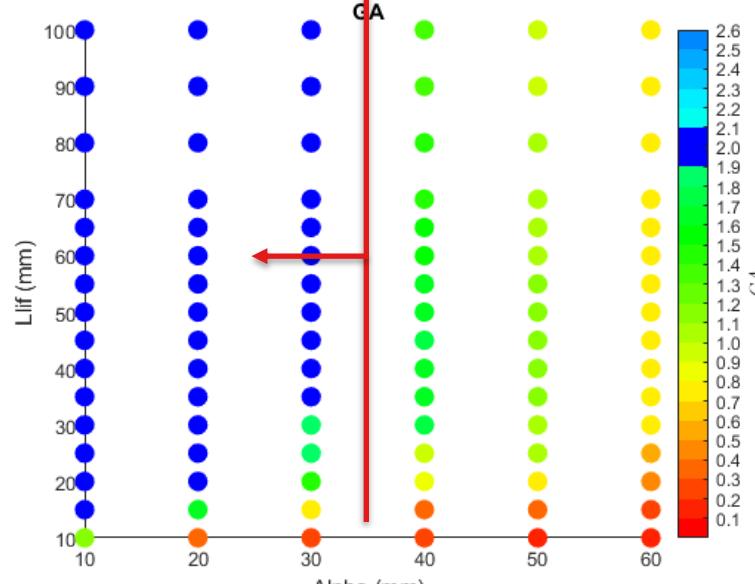
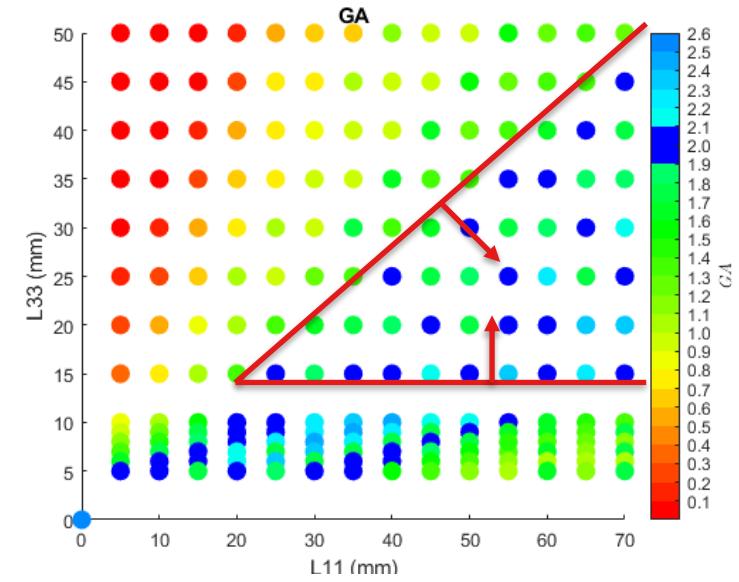
Here I combined all the criteria and minimized the objective. All criteria are scaled before and have a weight attached. In this figure the weights were all set to 1. The displacement and force-disp of the 4 best results are plotted below.



# Extra simulations

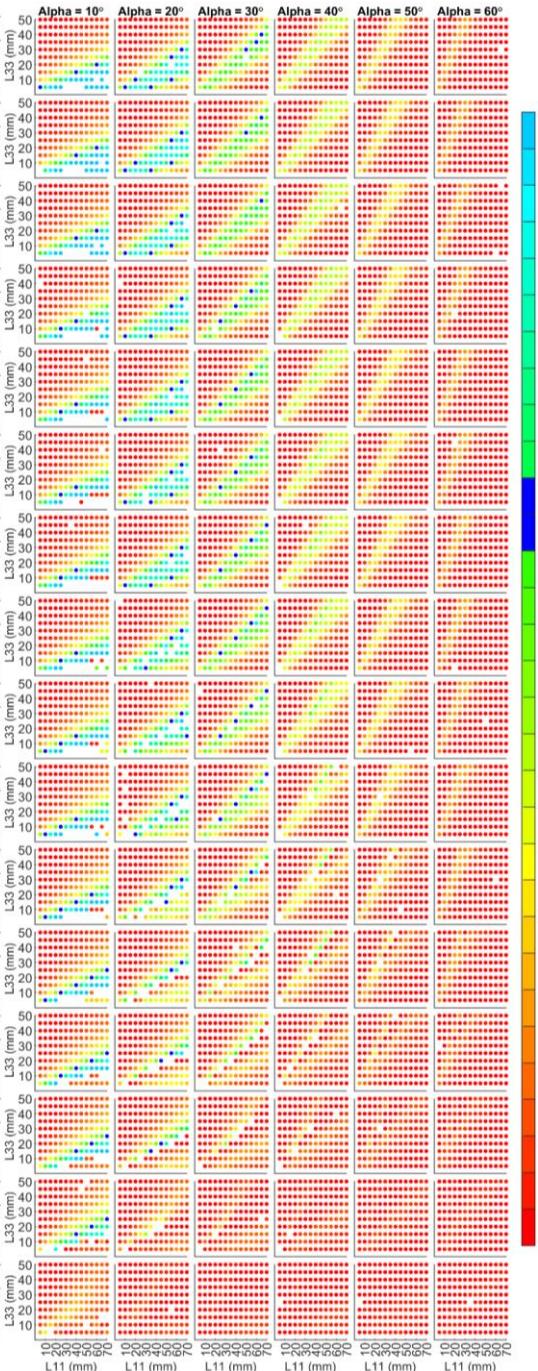
Did extra simulations in the red areas.  
Results of this will come later.

- Alpha = [15:2.5:35];
- Llif = [15:2.5:110];
- Range of L11=f(L33) res(2.5)
- L33 = [15:2.5:45];

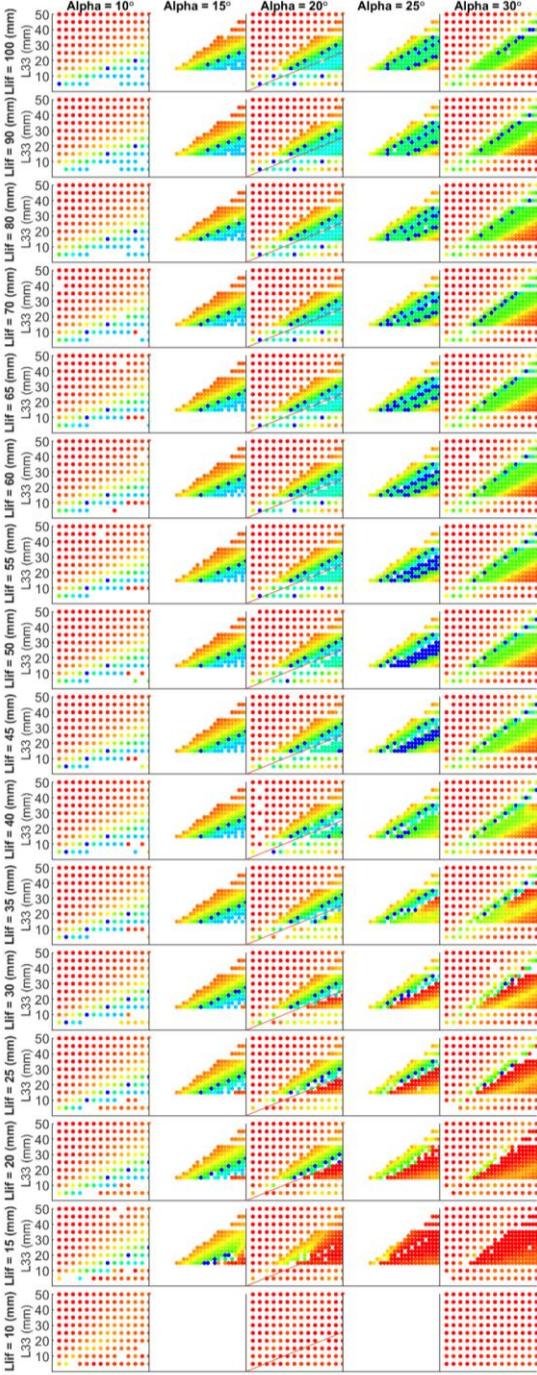


# GA

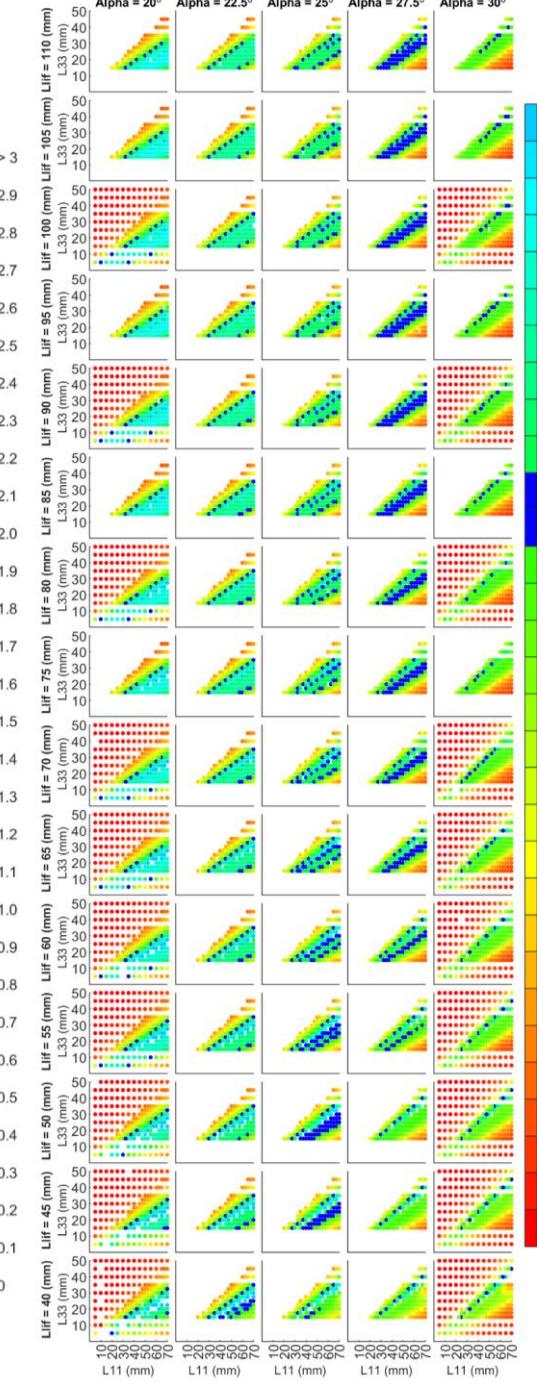
GA for each design at the end of the input



GA for each design at the end of the input



GA for each design at the end of the input



# Extra simulations

Did extra simulations in the areas indicated with the red arrows.

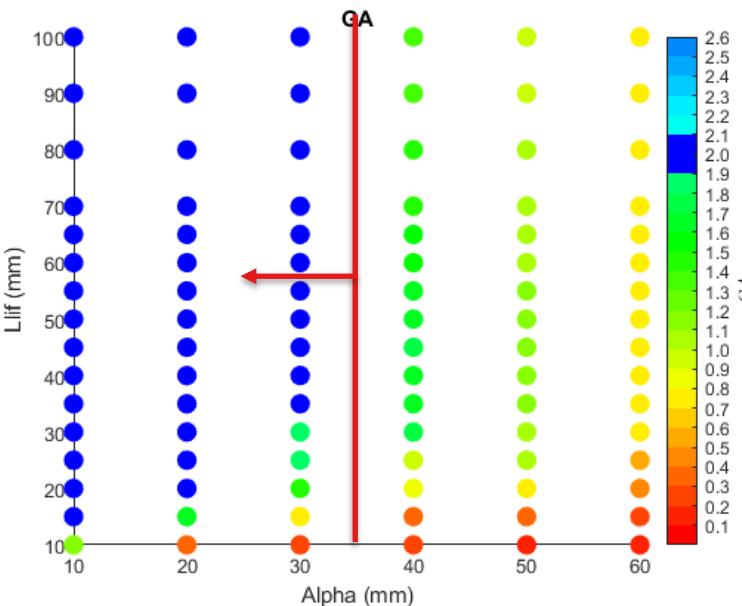
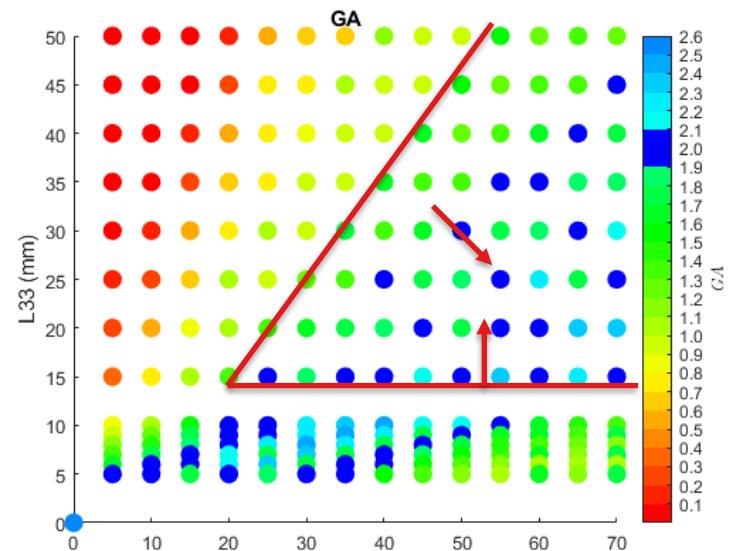
Results of this will come later.

+/- 66000 data points

- $\text{Alpha} = [15:2.5:35];$
- $\text{Llif} = [15:2.5:110];$
- Range of  $\text{L11}=f(\text{L33})$
- $\text{L33} = [15:2.5:45];$

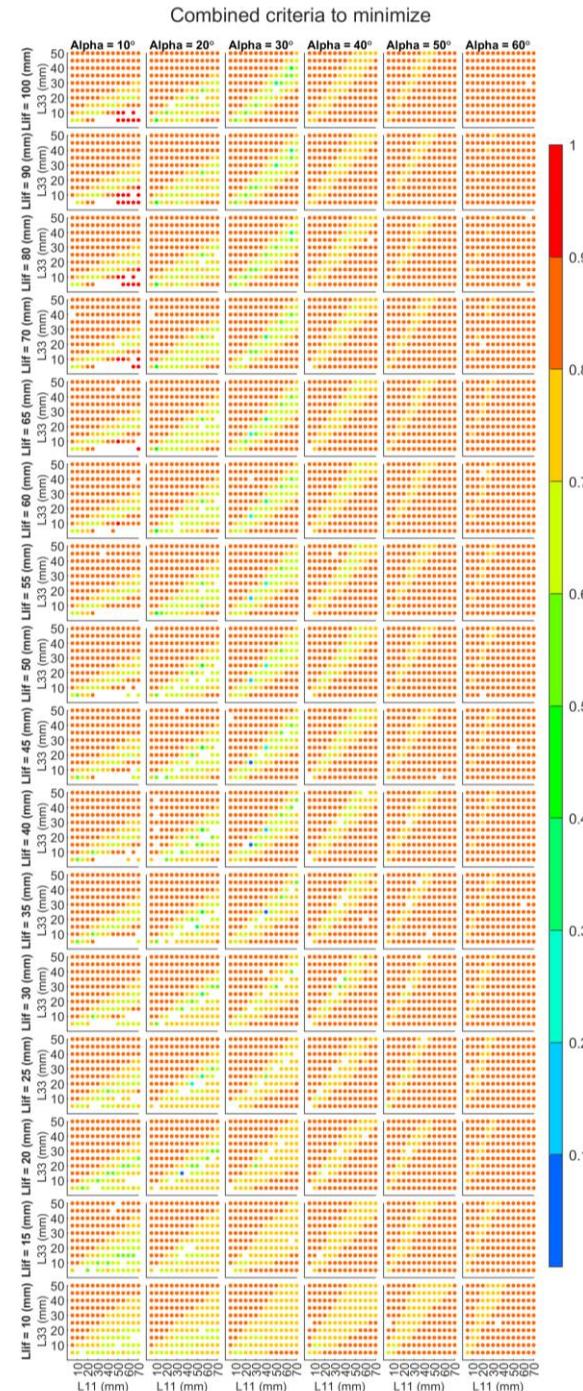
NEW

- $\text{Alpha} = [15:2.5:45];$
- $\text{Llif} = [15:2.5:110];$
- Range of  $\text{L11}=f(\text{L33})$
- $\text{L33} = [15:2.5:45];$

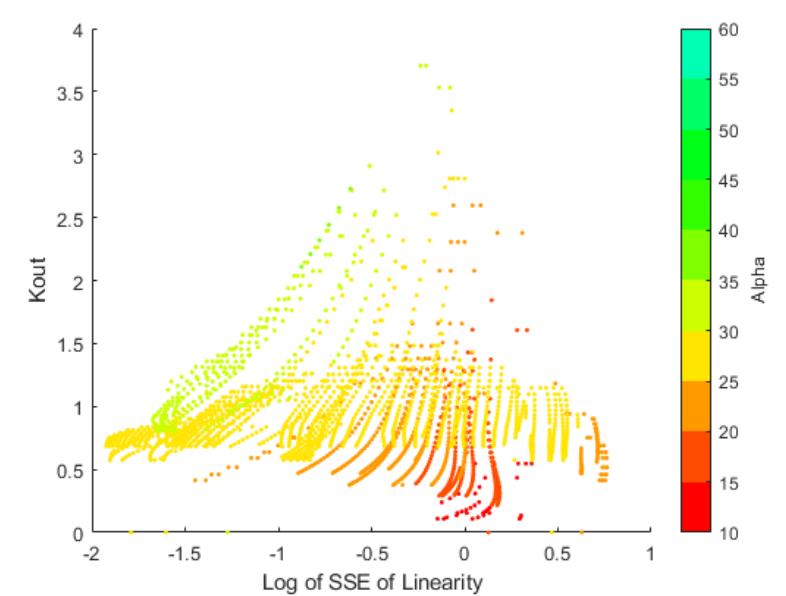
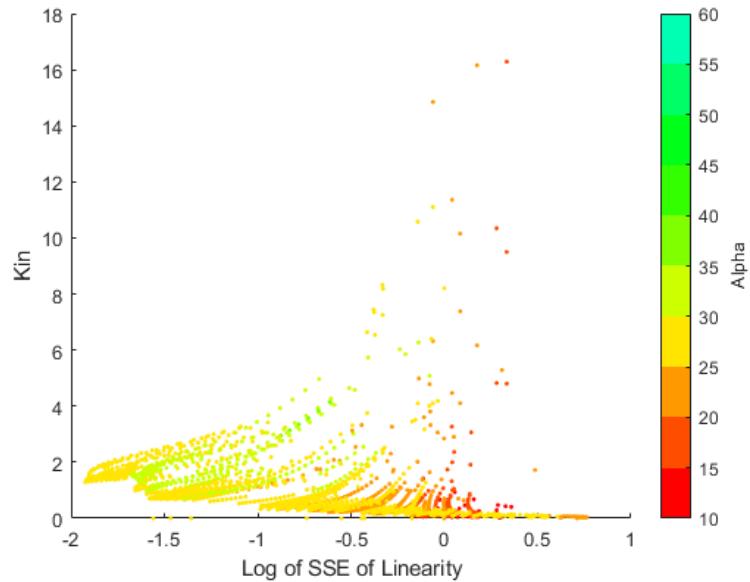
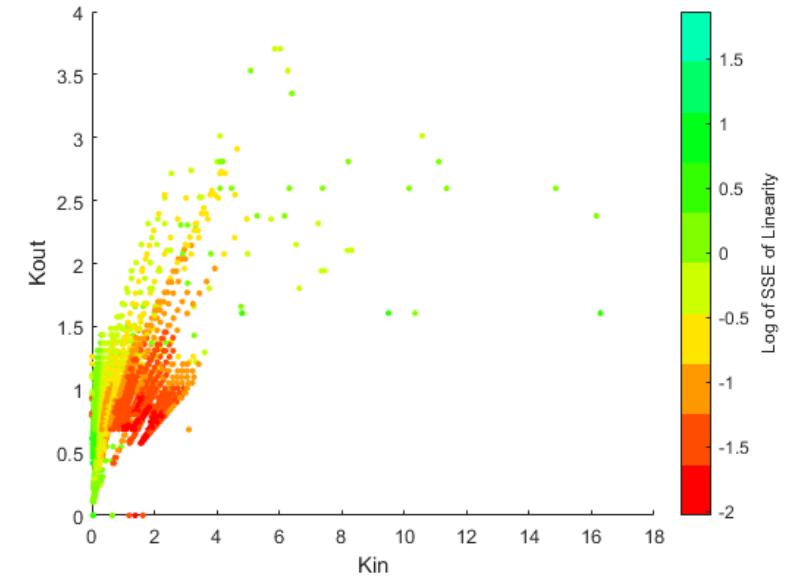
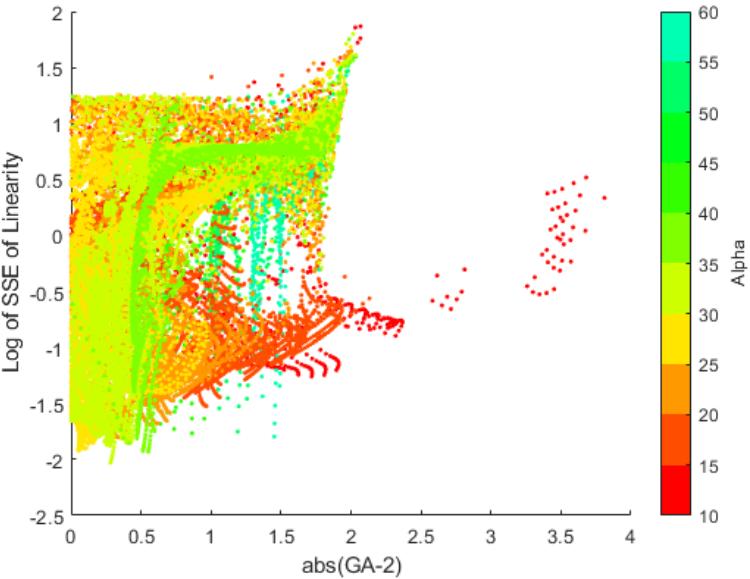


# Criteria Combined

- There is no clear best point, from just looking at individual plots, so I scaled each criteria from 0-1 and added these up.
- How to determine weights/ how to determine the objective to minimize?
- Also wrote a script to find the optimal point by minimizing an objective.



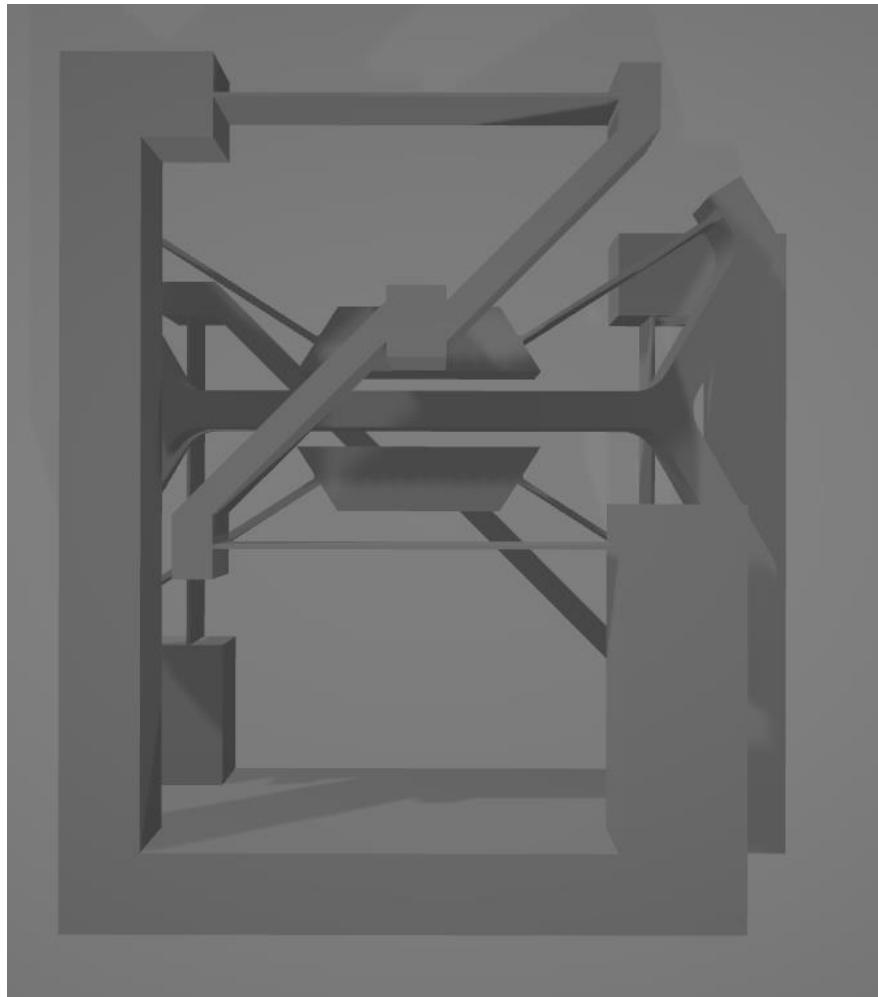
- No clear optimum
- More criteria to plot
- Dimensionless parameters?
- GA
- Linearity x
- Kin
- Kout
- Kact x



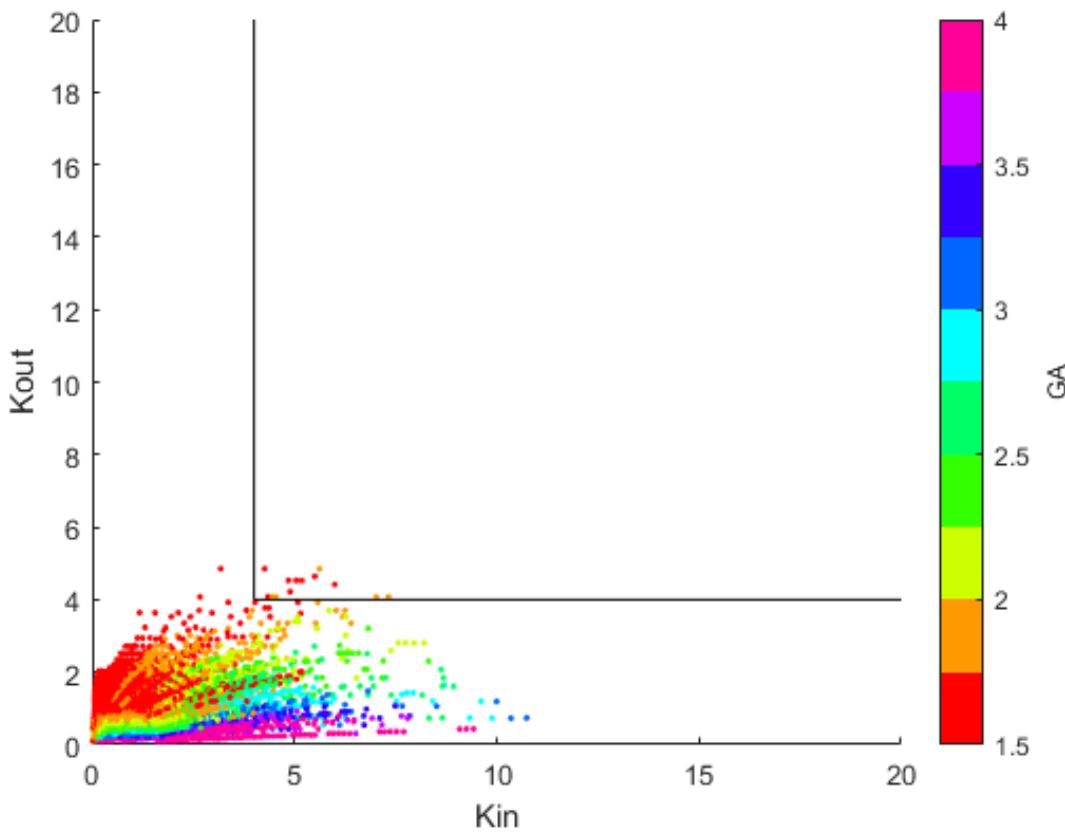
# Cosinusoidal mechanism

- Still not really have a clue how to optimize for the extra parameter. Without it taking an insane amount of time.
- Maybe some sort of regression algorithm to find a relationship for the tilted angle in relationship with the other parameters.

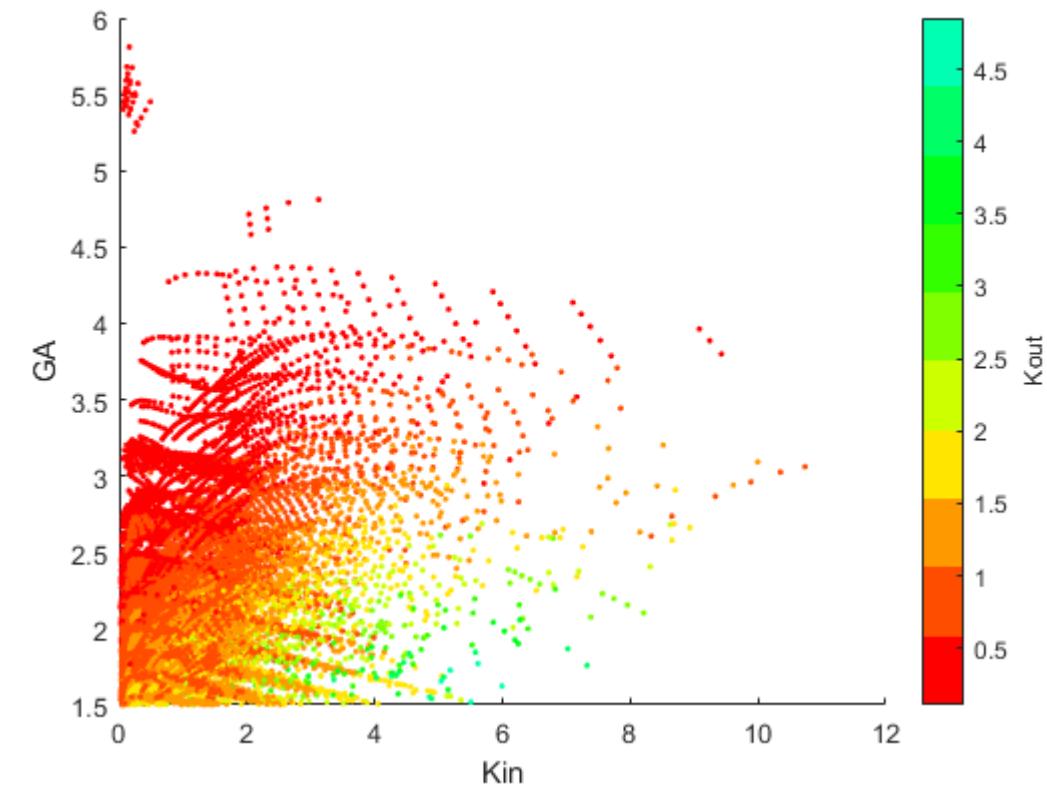
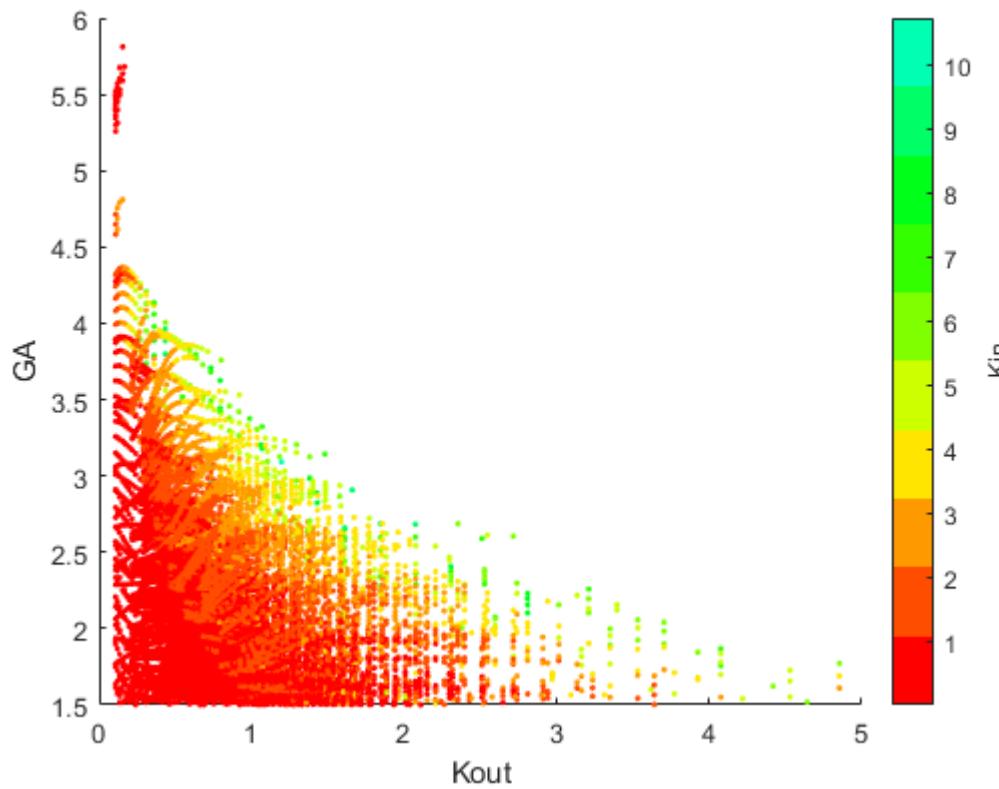
# Iteration 1 using SLA printer



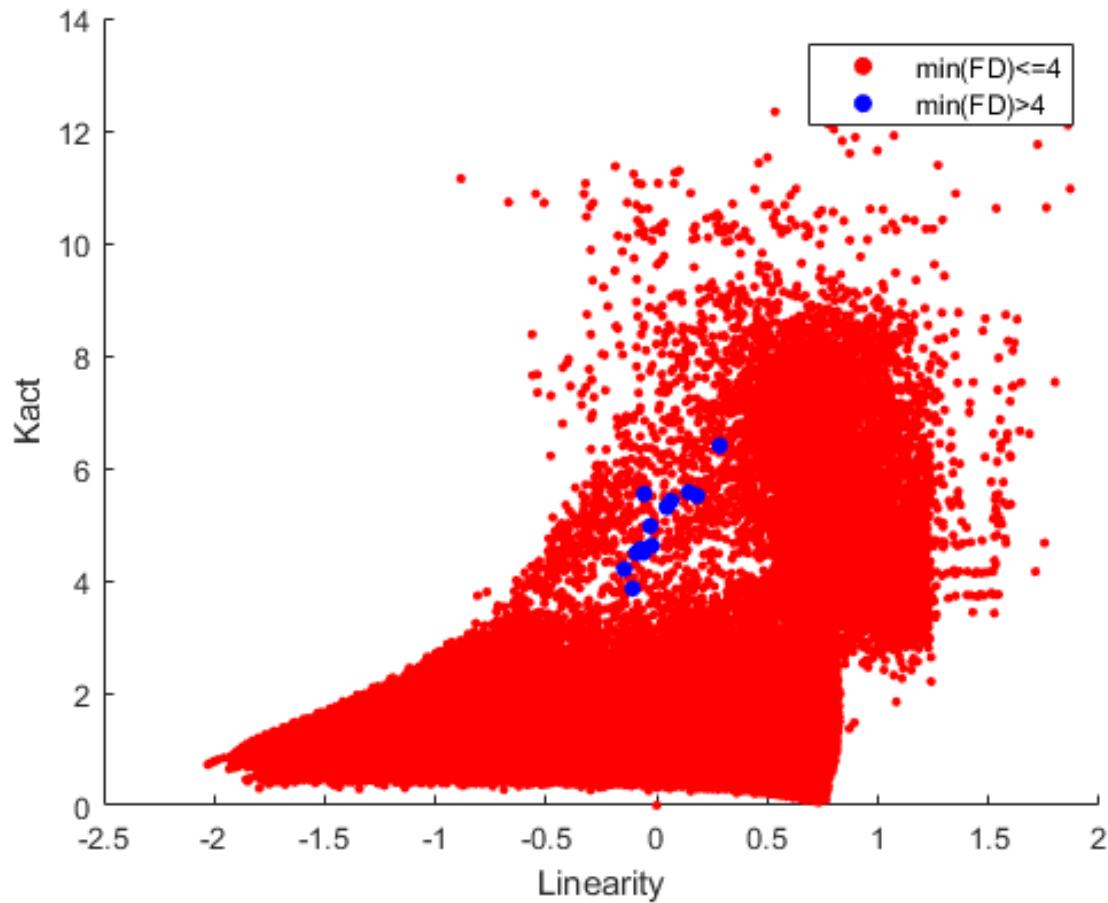
13 points that have  $\min(K_{in}, K_{out}) > 4$



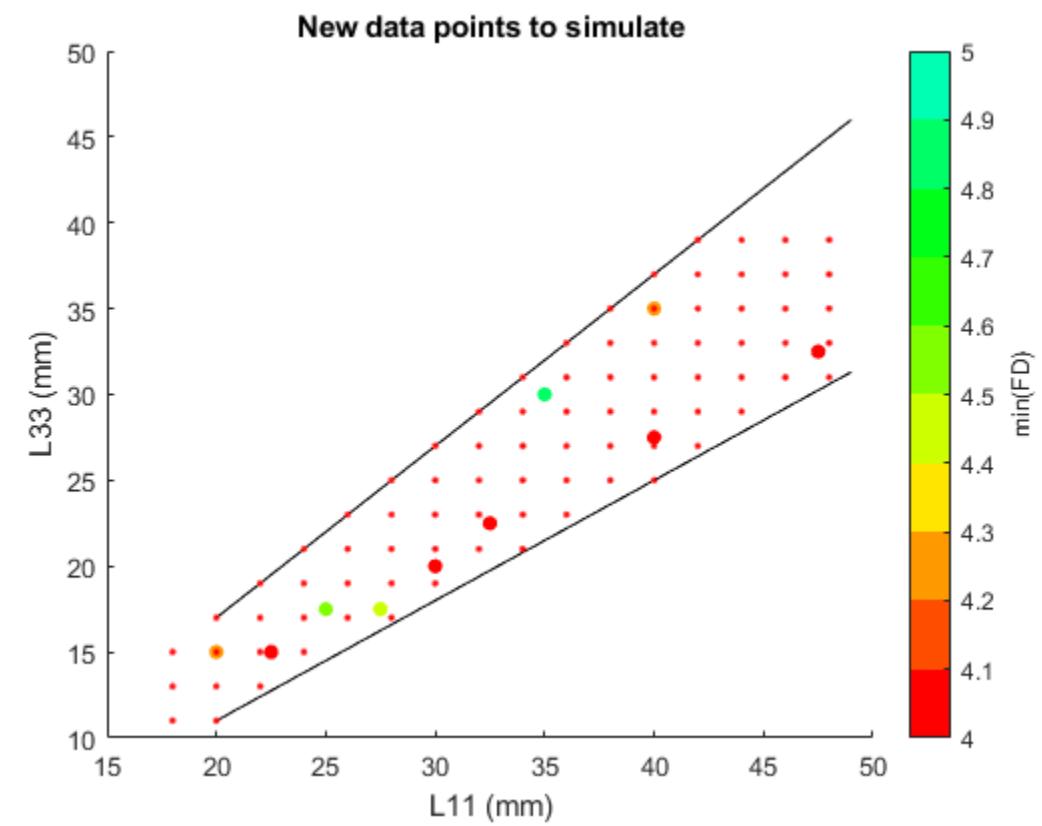
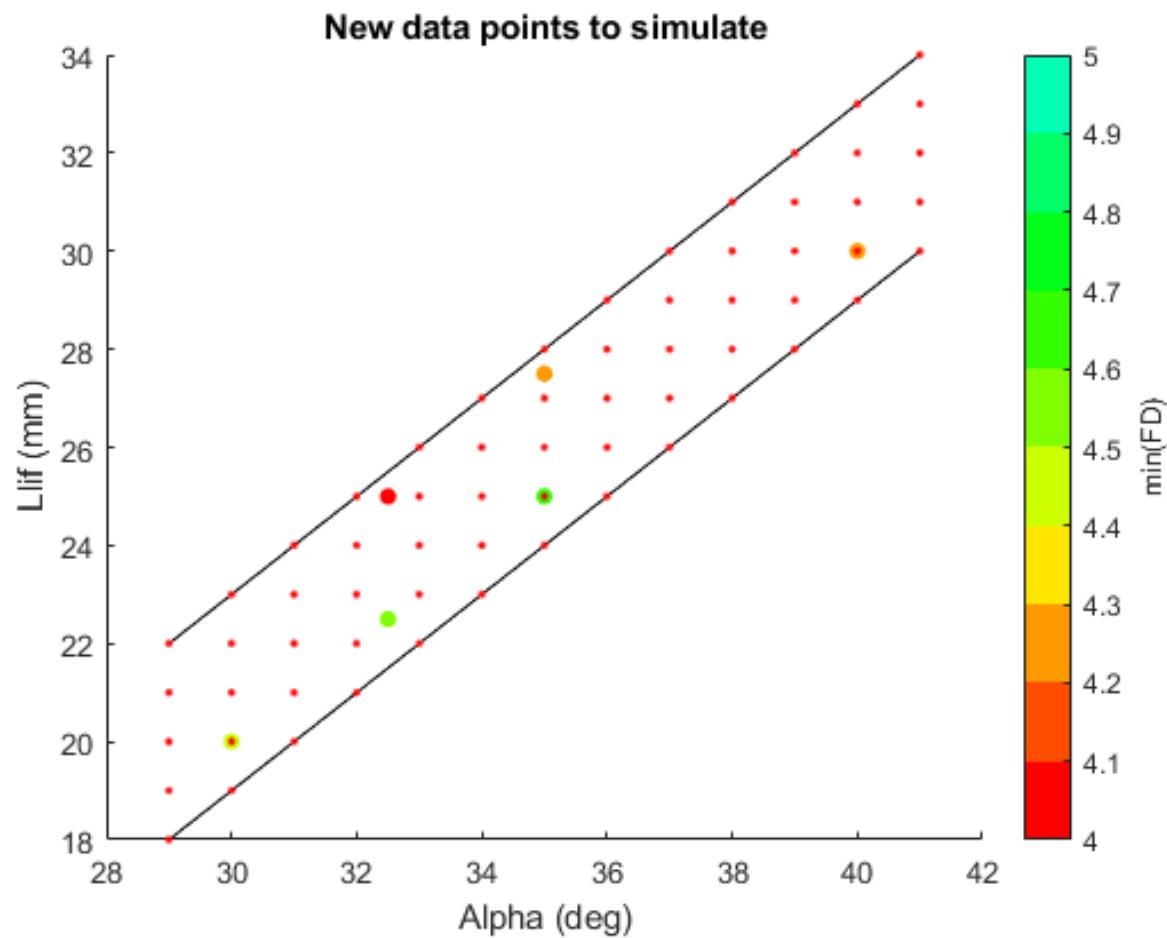
Clear that for a large GA a small Kout is necessary



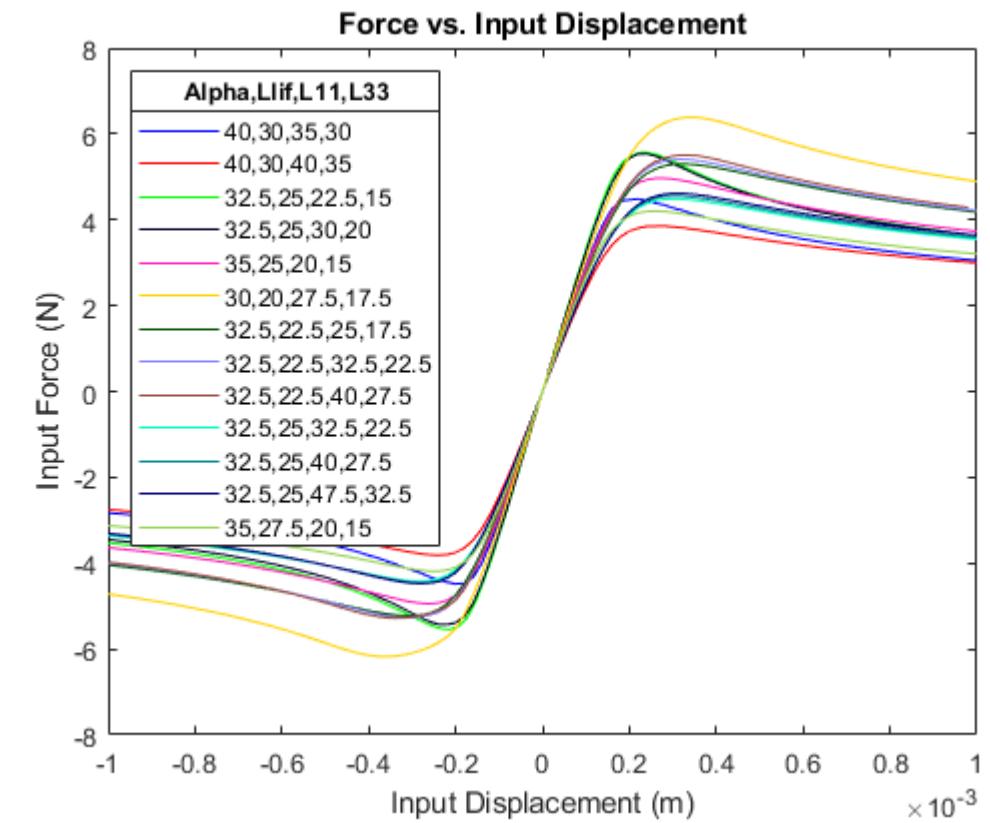
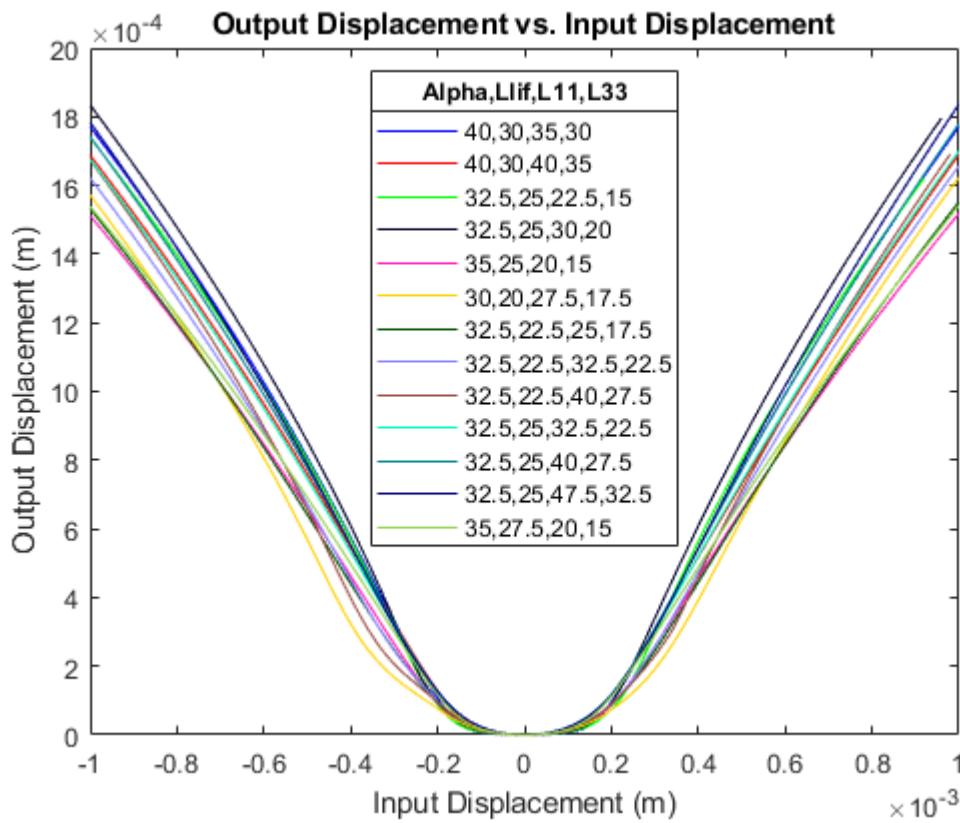
Linearity and KAct for the the points with  $\min(Kin, Kout) > 4$ .  
Mostly middle of the pack.



For the points  $\min(\text{Kin}, \text{Kout}) > 4$ , there is a clear relationship between Alpha-Llif and L11-L33. This was used for determining where more data points will be evaluated.

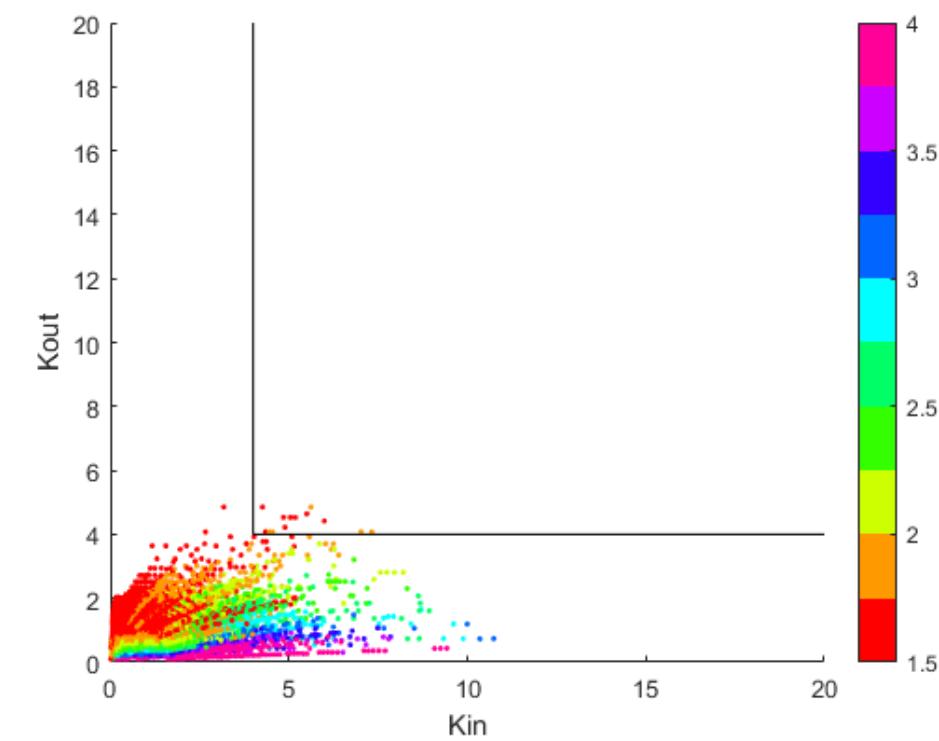
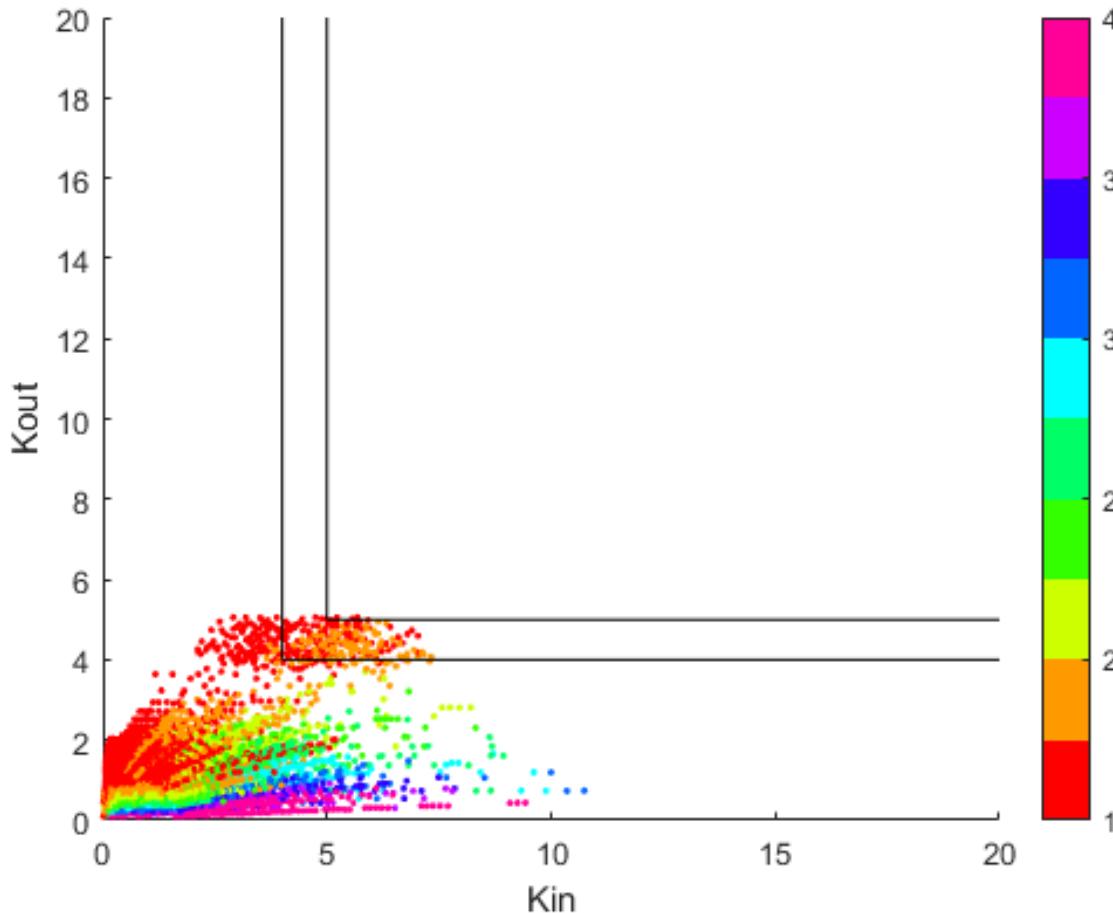


## DD and FD diagram for the mechanisms that have $\min(\text{Kin}, \text{Kout}) > 4$

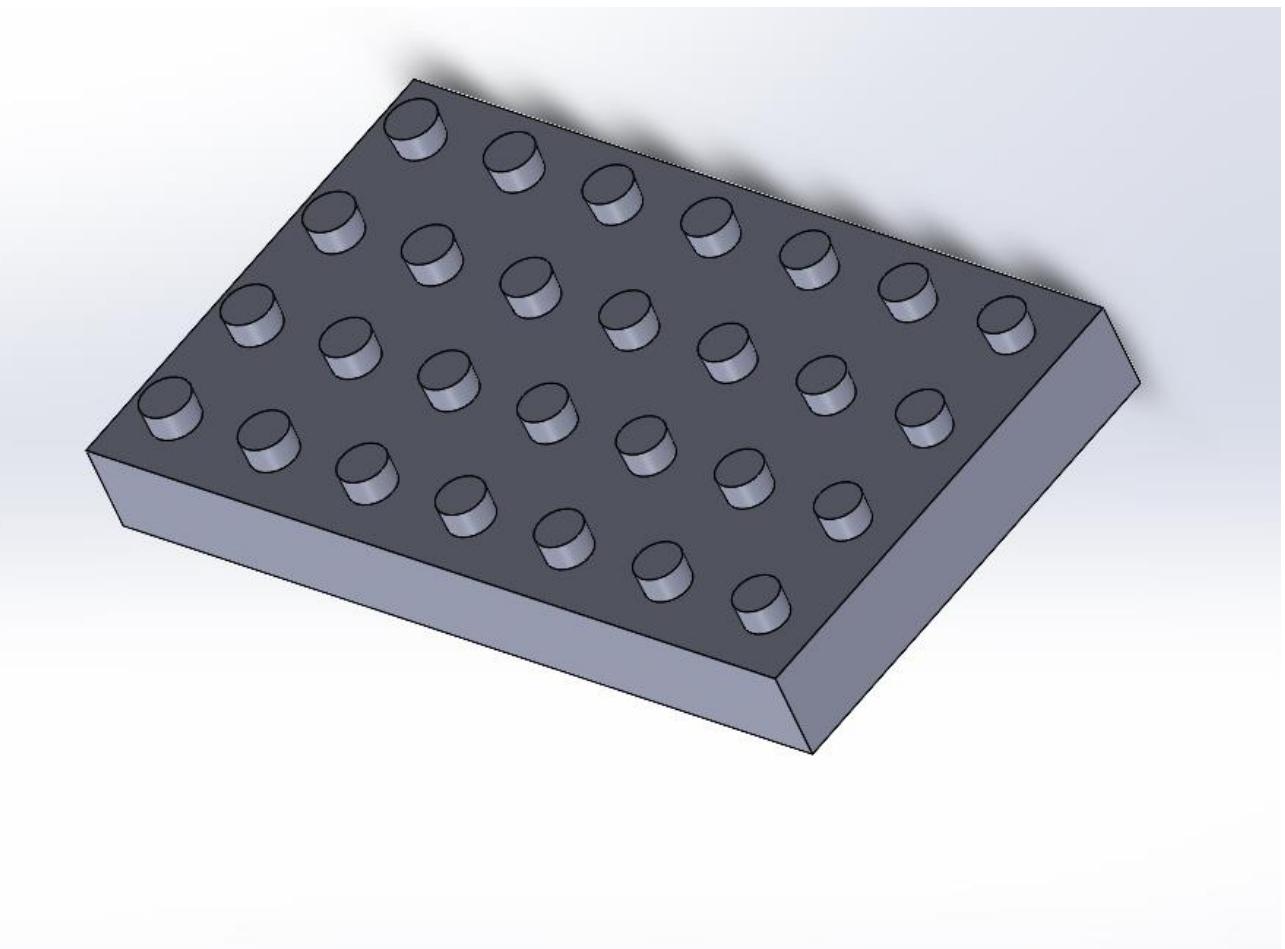
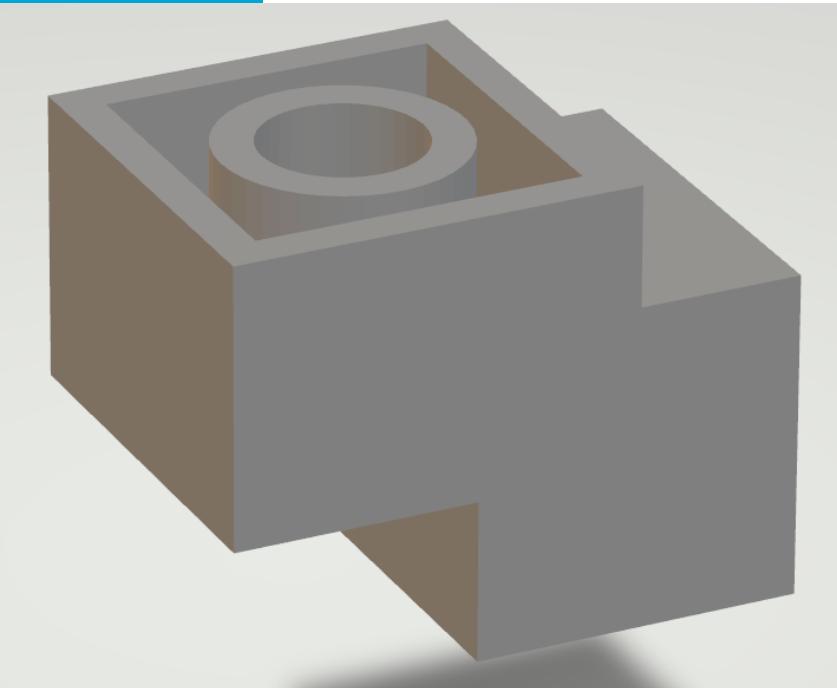


New Data, now 226 points have  $\min(\text{Kin}, \text{Kout}) > 4$ , and 4 even  $> 5$

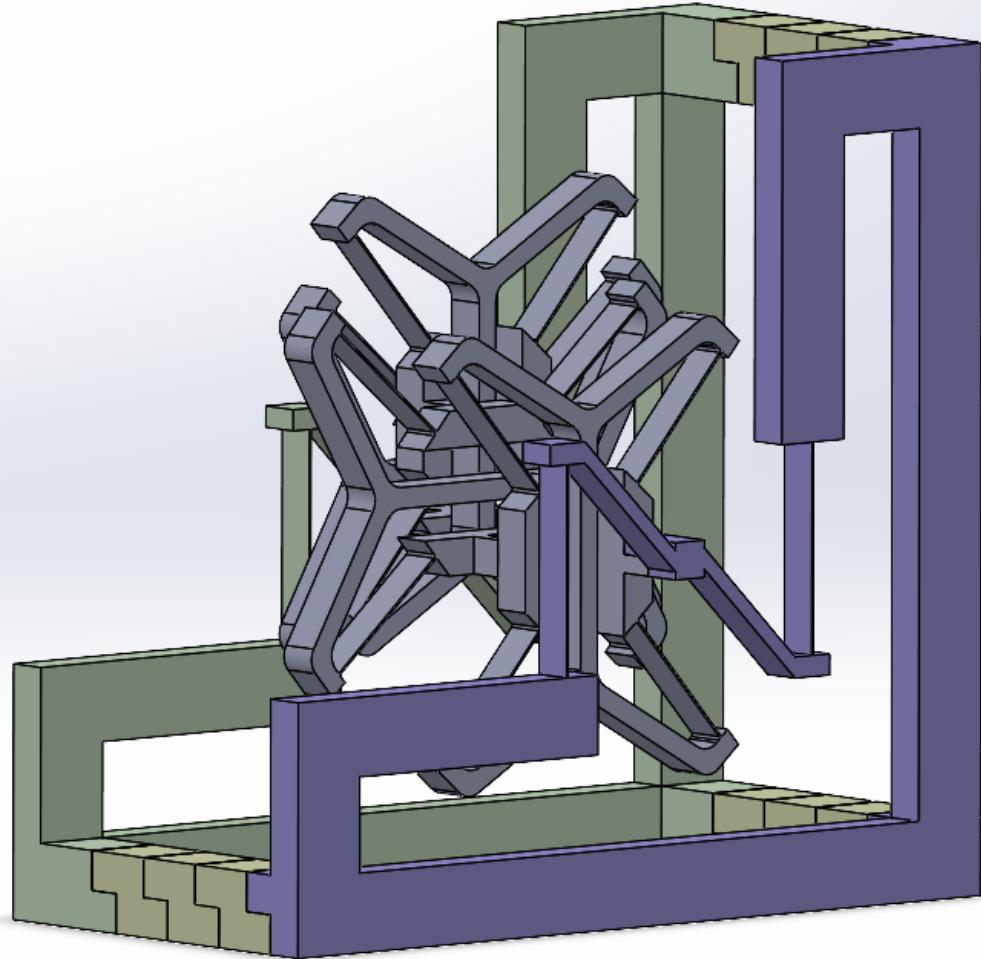
Best point: [Alpha, Llif, L11, L33] = [41, 30, 26, 23]



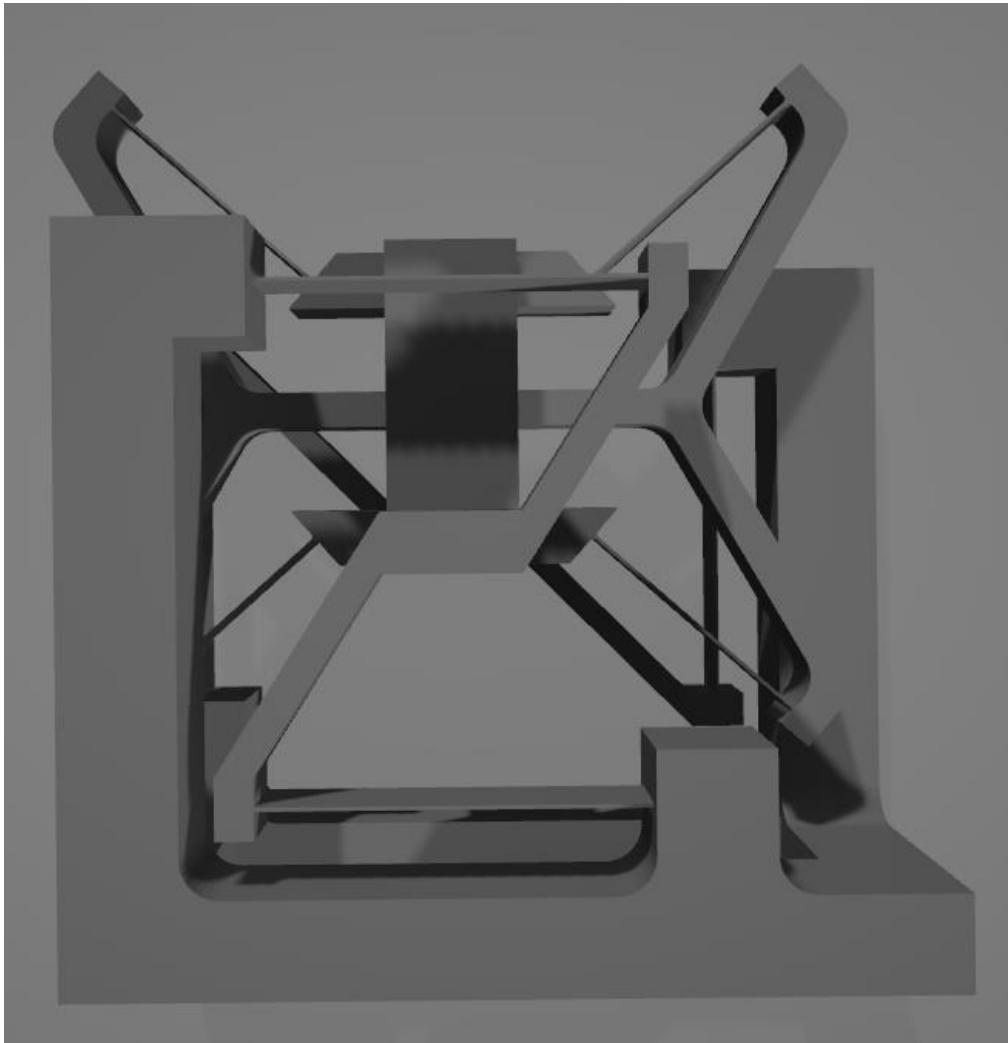
# To test pin-hole sizes to connect mechanisms



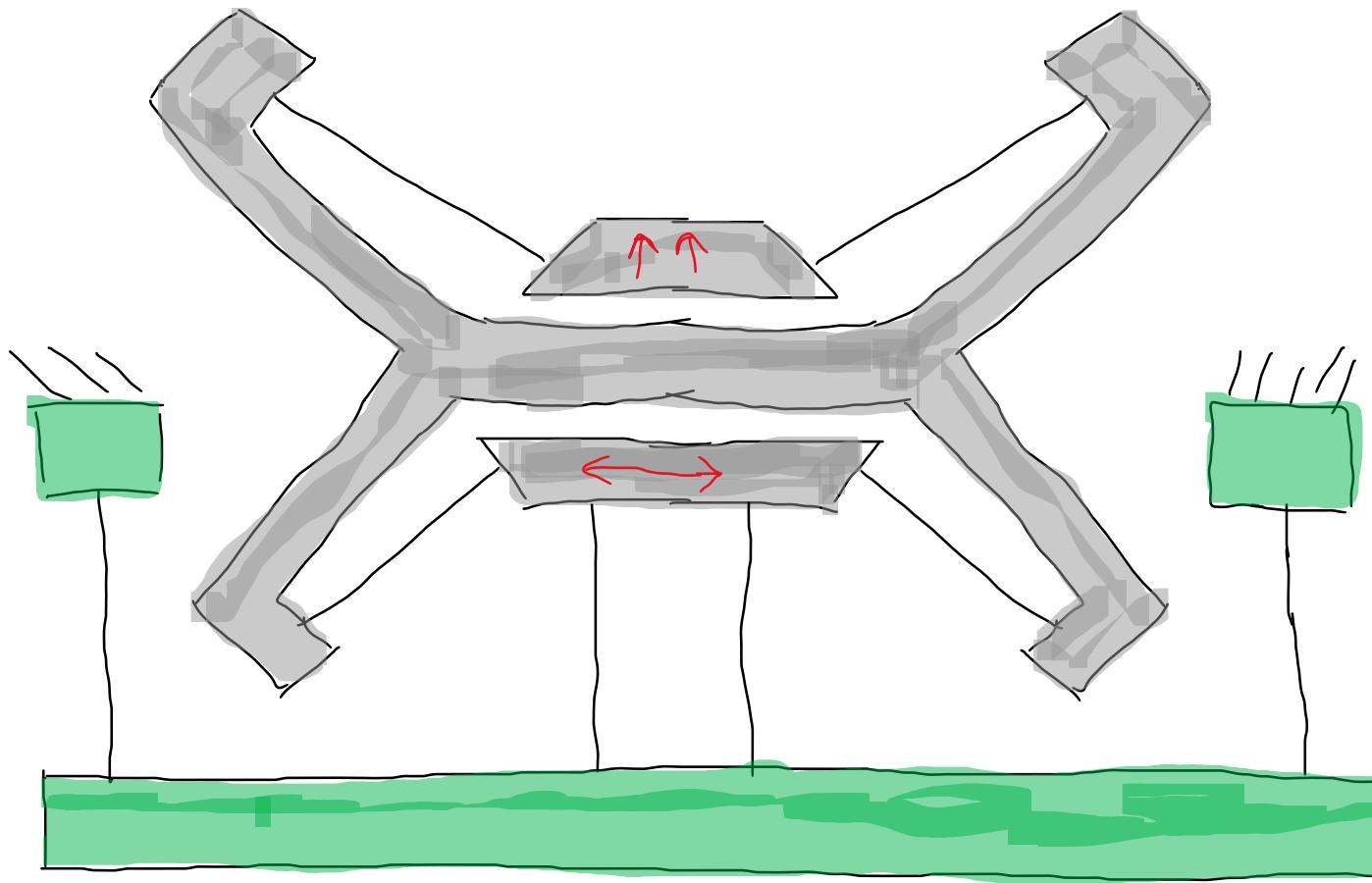
Solidworks: - Made it tileable  
- Easy manipulation with functions



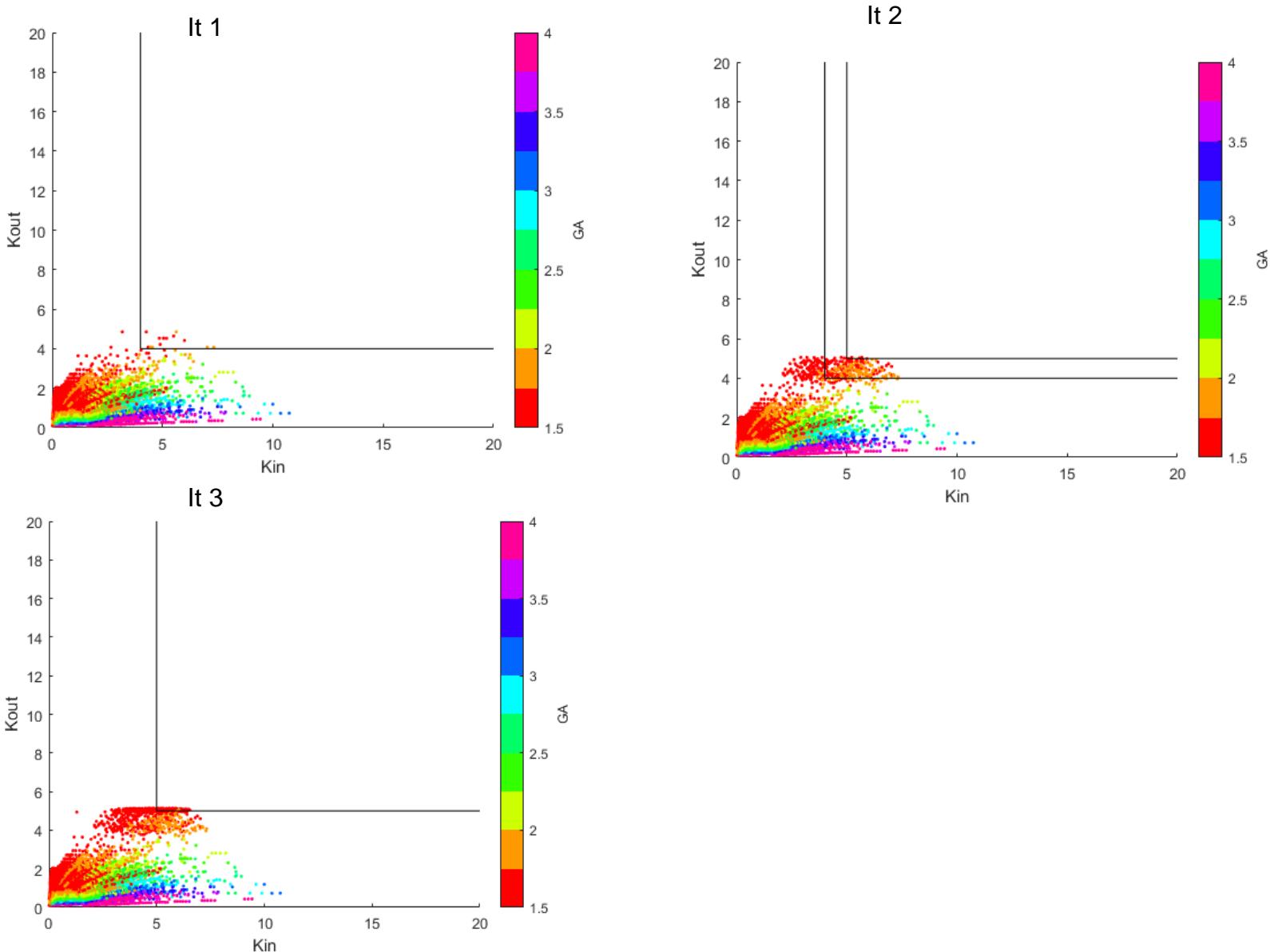
# Iteration 2 SLA printer



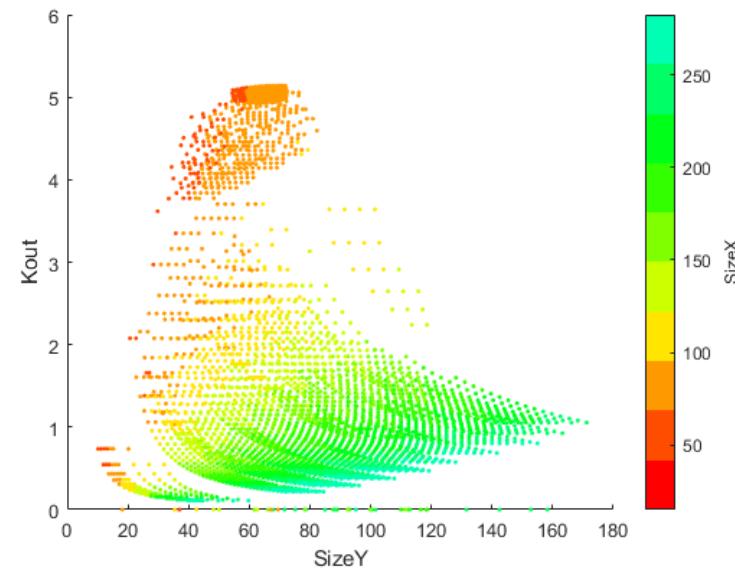
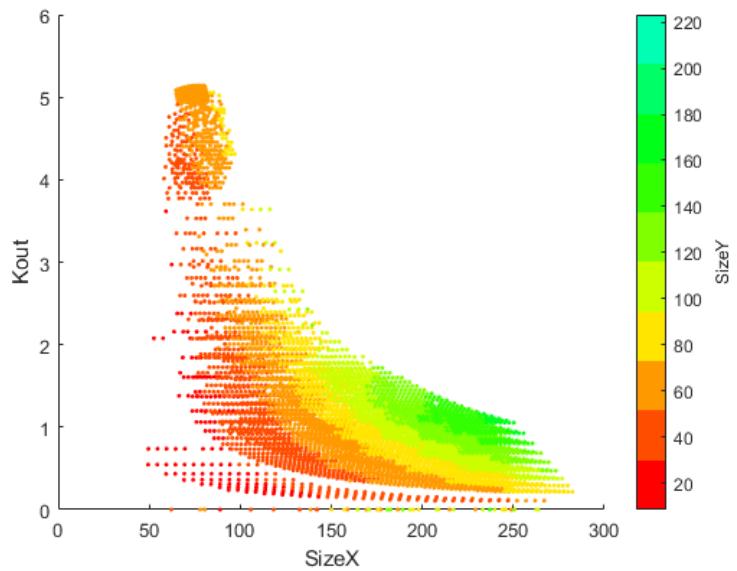
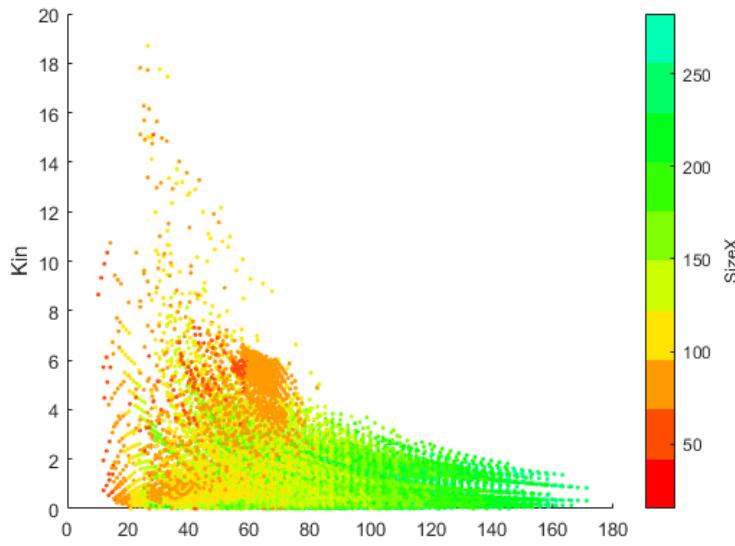
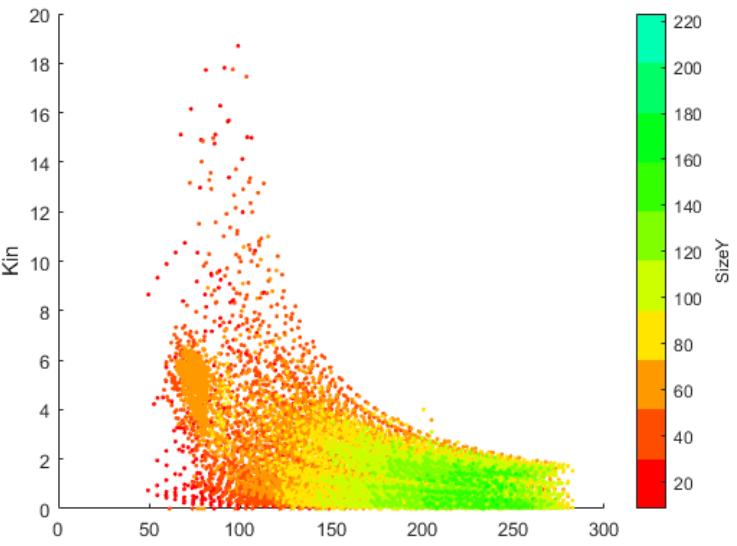
## Prototype for new mechanism to not have large moment from input



Kout does not seem to improve, but more points with high force capabilities

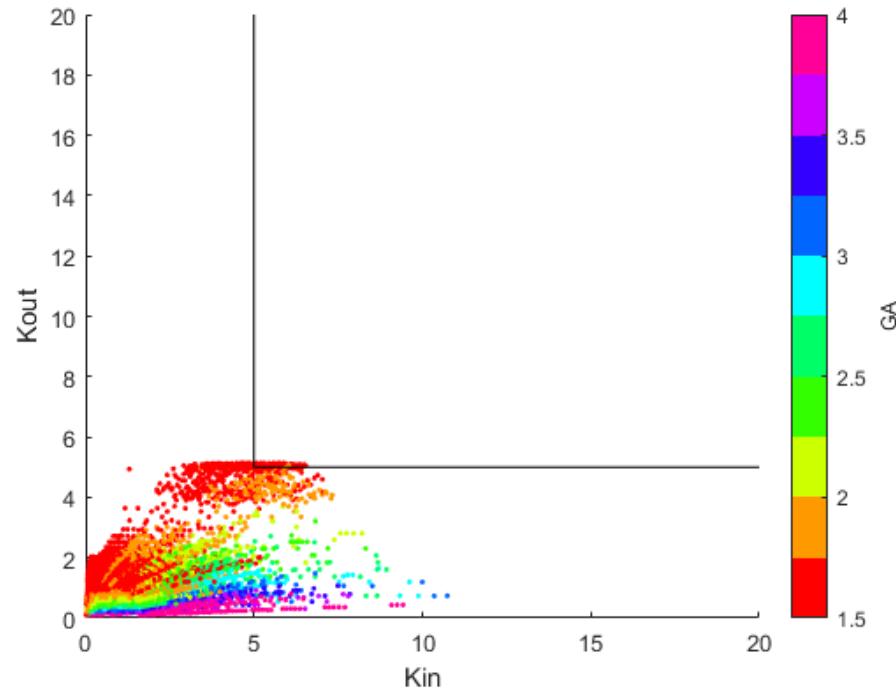


## Size dependency, smaller seems better and limited by manufacturing



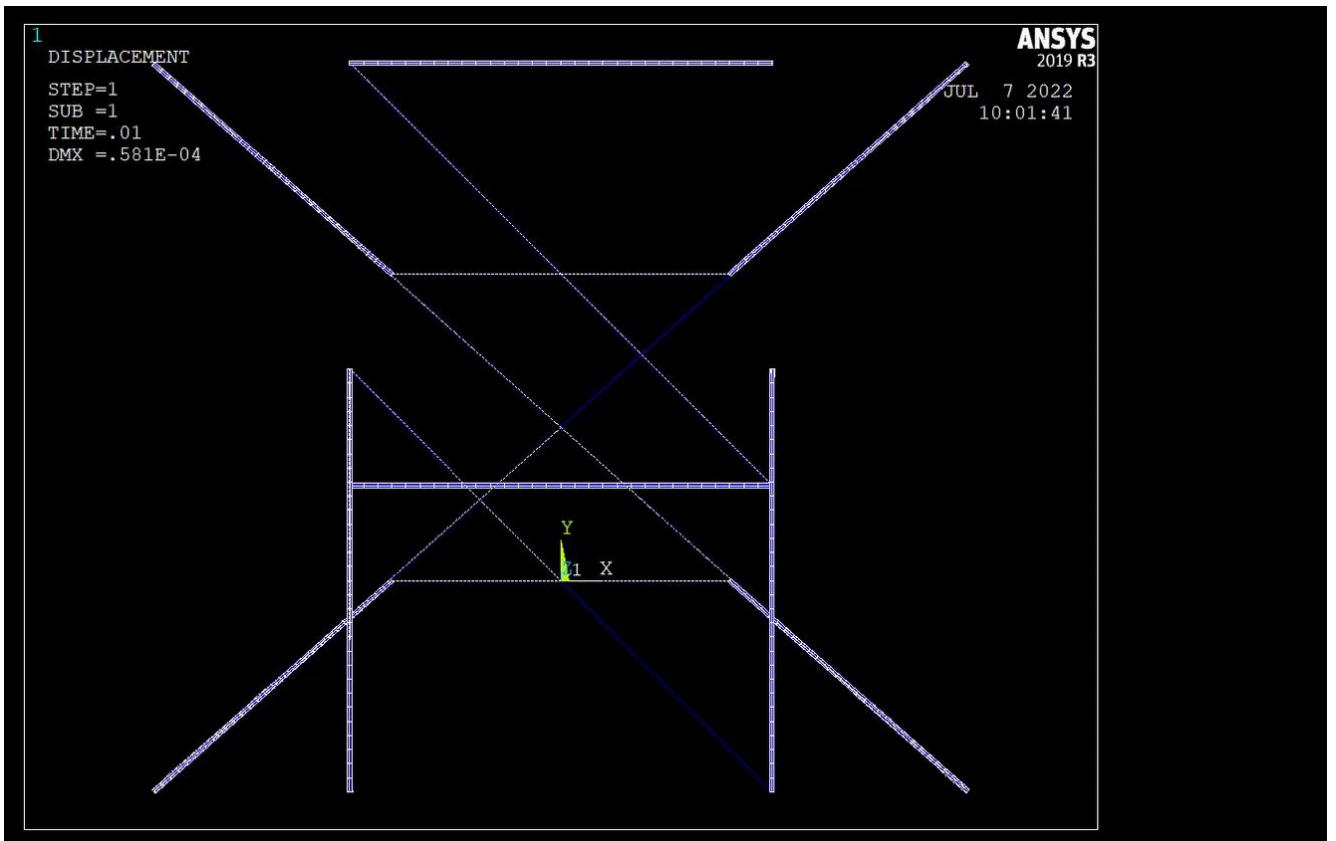
## Some notes

- Sharing files
- Behavioral analysis approach
- Testing?
- Still have to run stress analysis
- Linear guidances between mechanism?

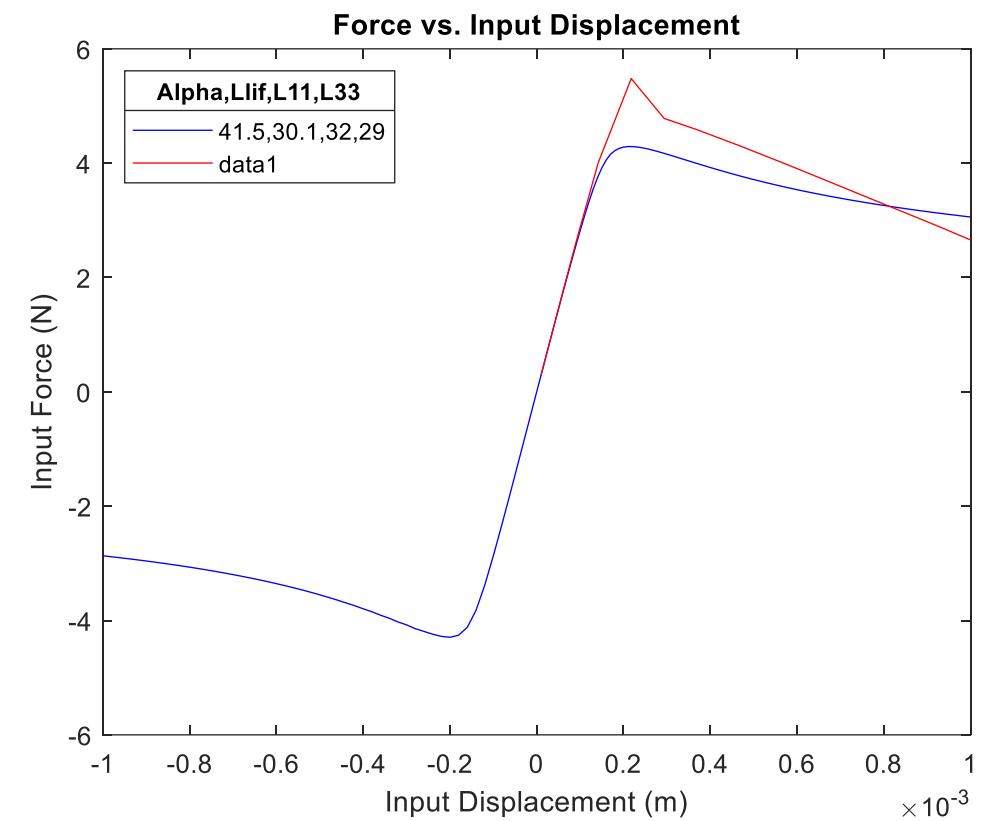
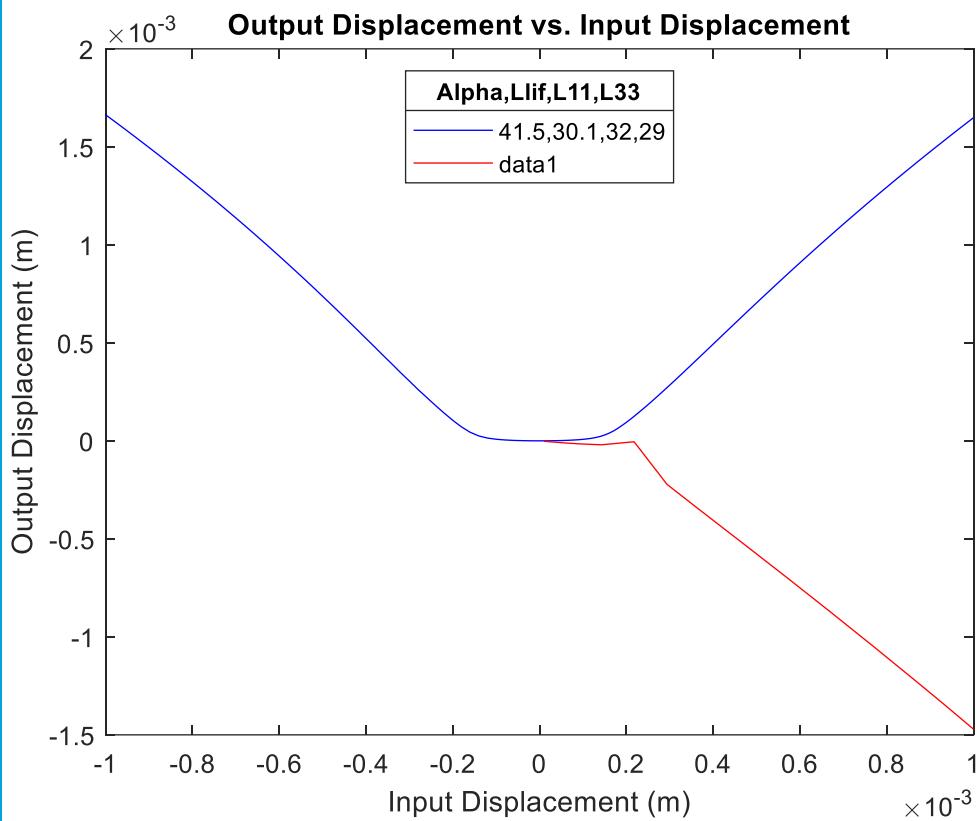


## Comparing Stress

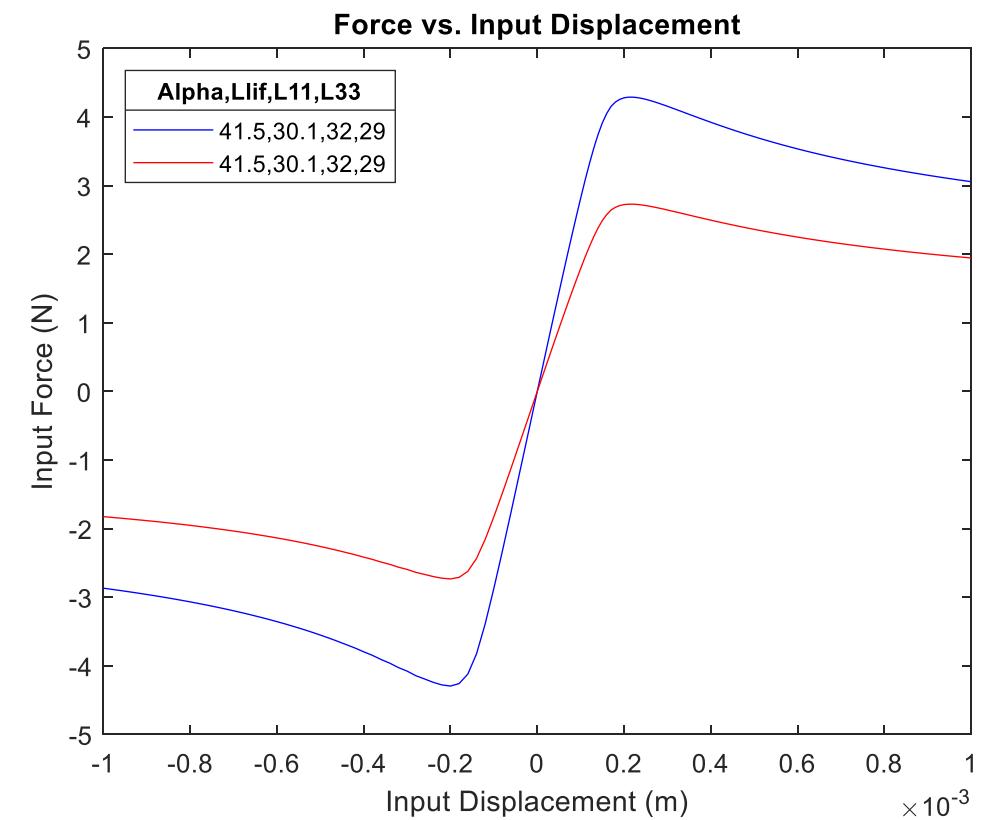
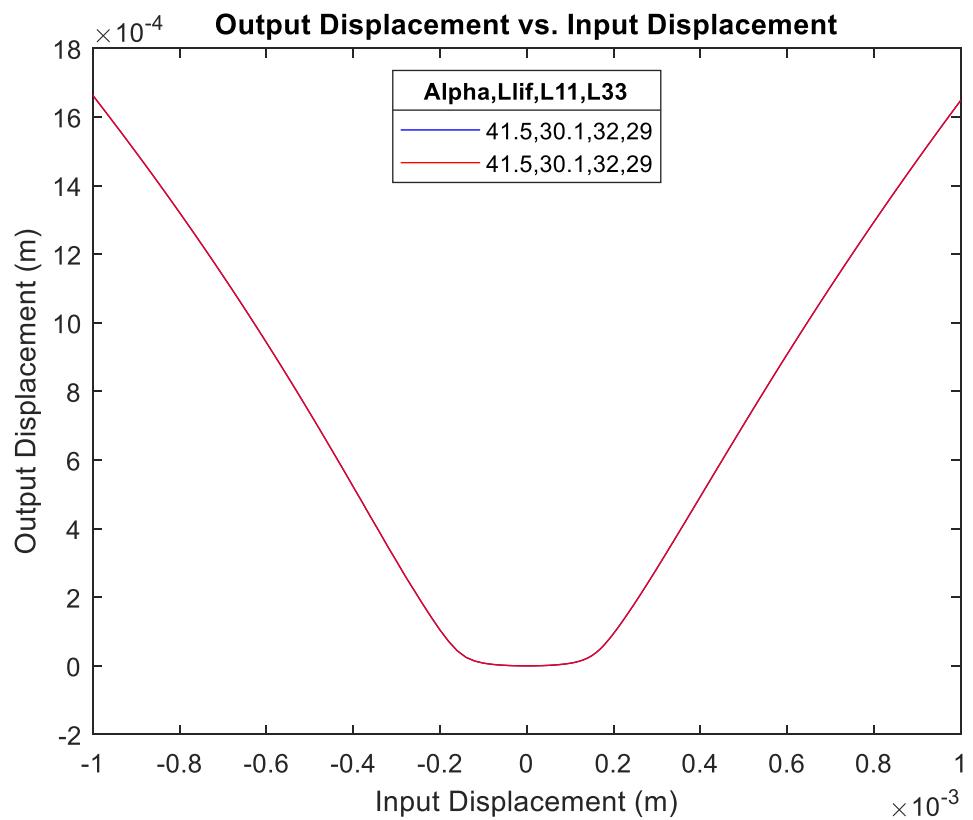
- Output fixed looking at 5% of Din
  - Smax = 3.32 MPa
  - F = 4.11893 N
- Load at output
  - Smax = 38 MPa
  - Opposing force?
  - Too high



## DD and FD graph of mechanism with added load

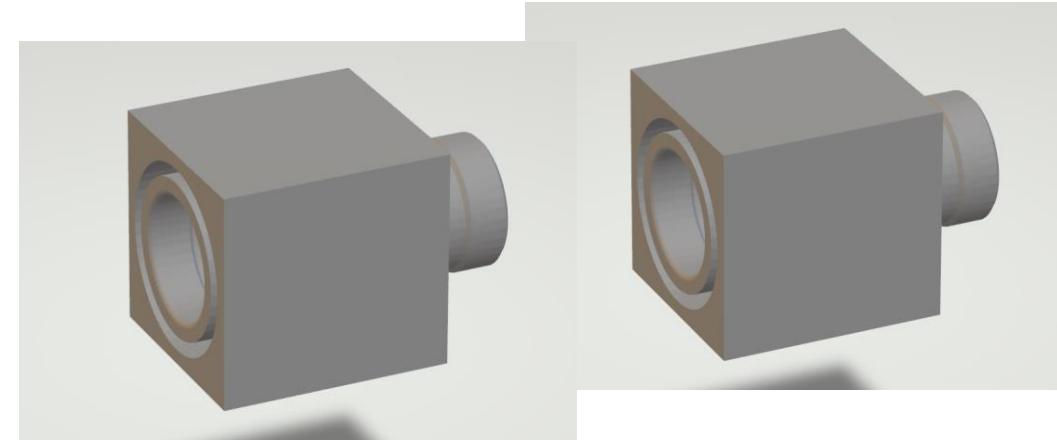
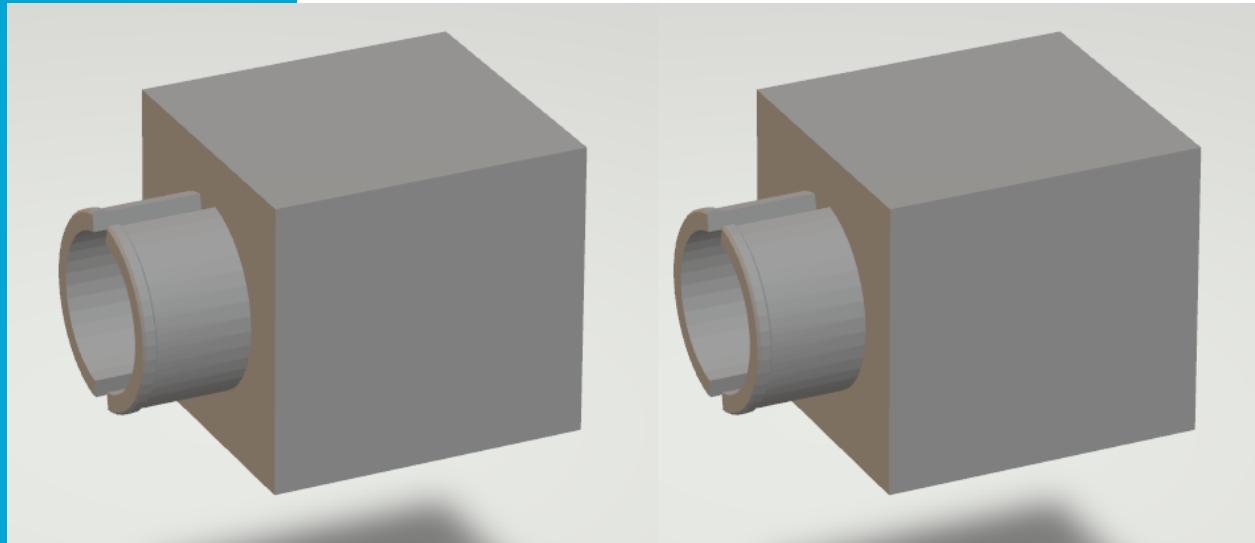


Blue: Clear resin, Red: Tough1500 resin



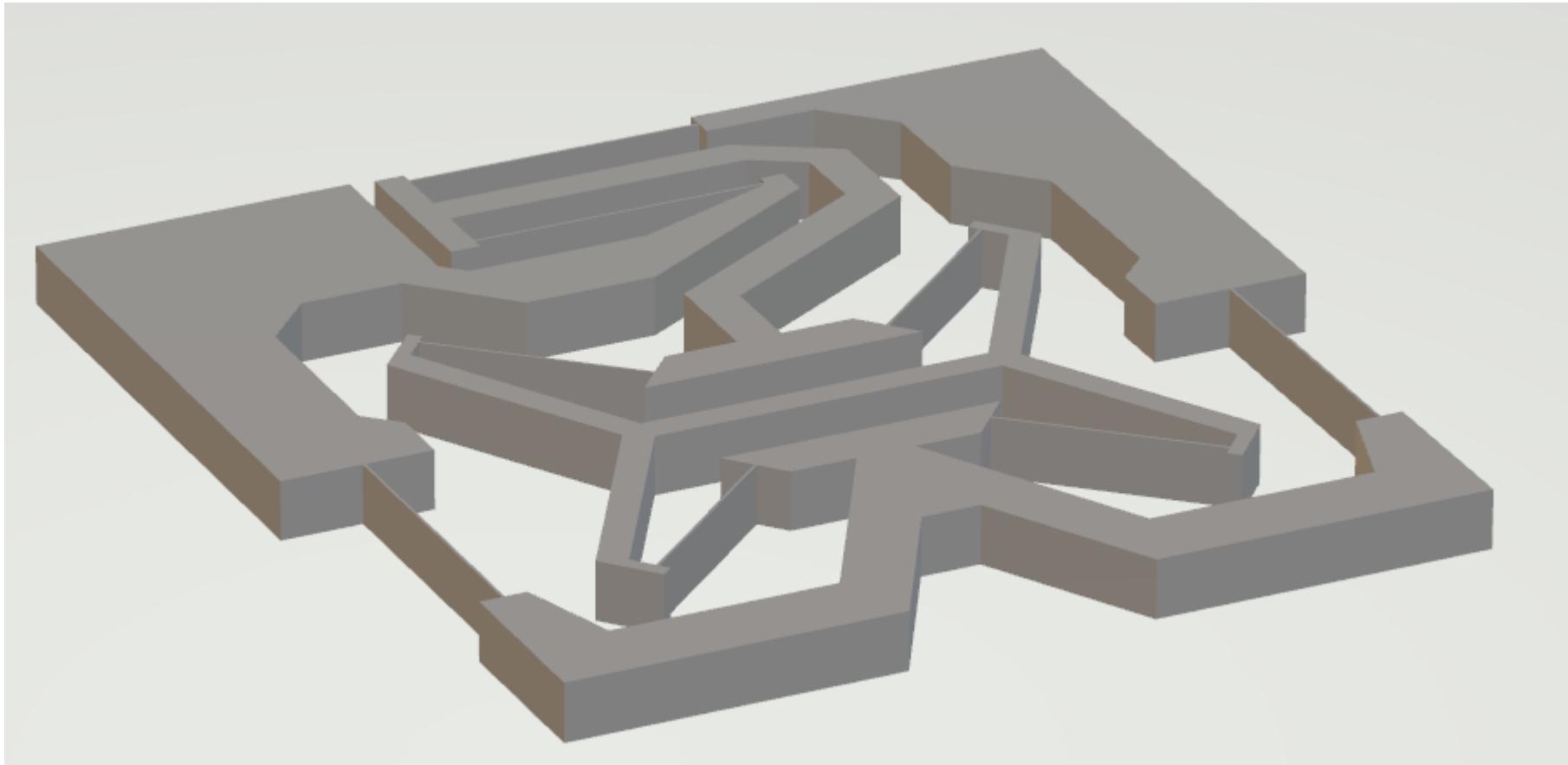
# Combining the planes

- Some ideas to attach the input shuttle to butterfly flexures and output shuttle

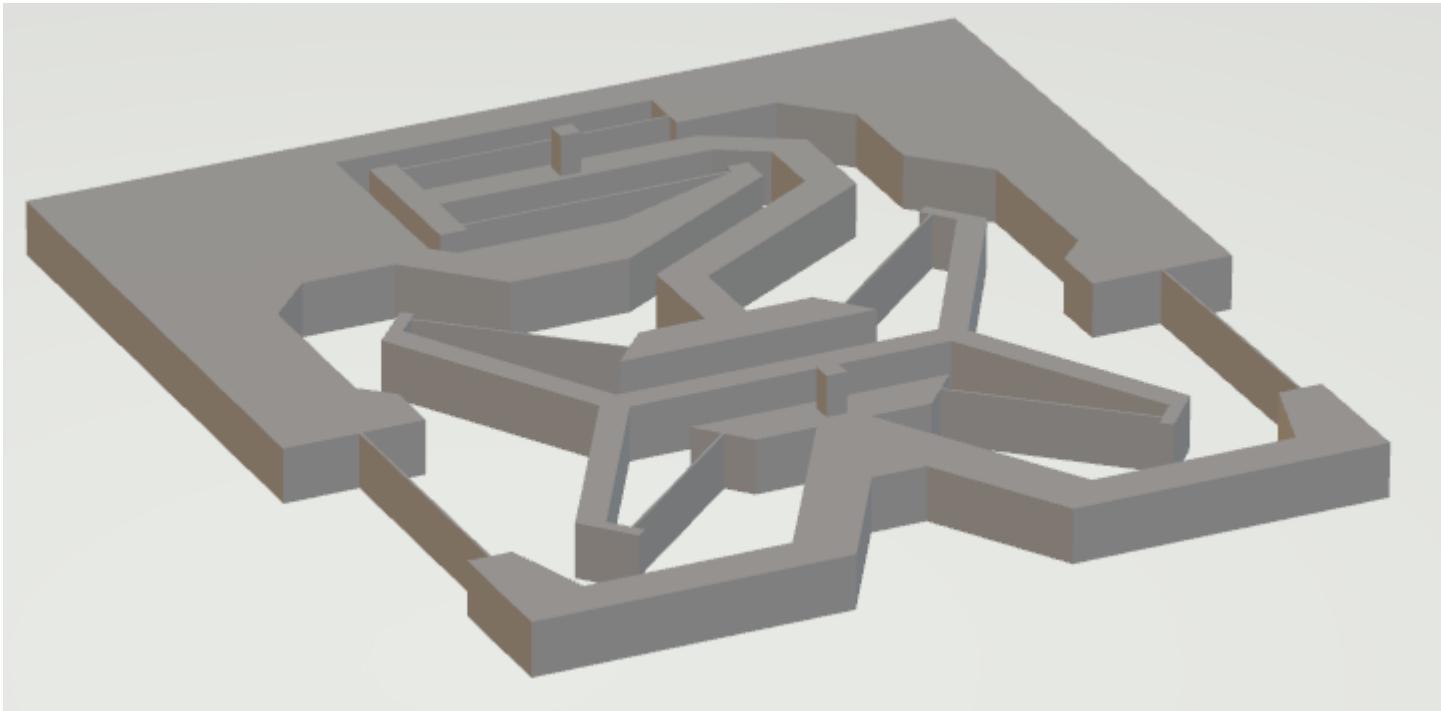


# Trying to make manufacturing work

- Iteration 1 planar mechanism



- Iteration 2 of planar mechanism

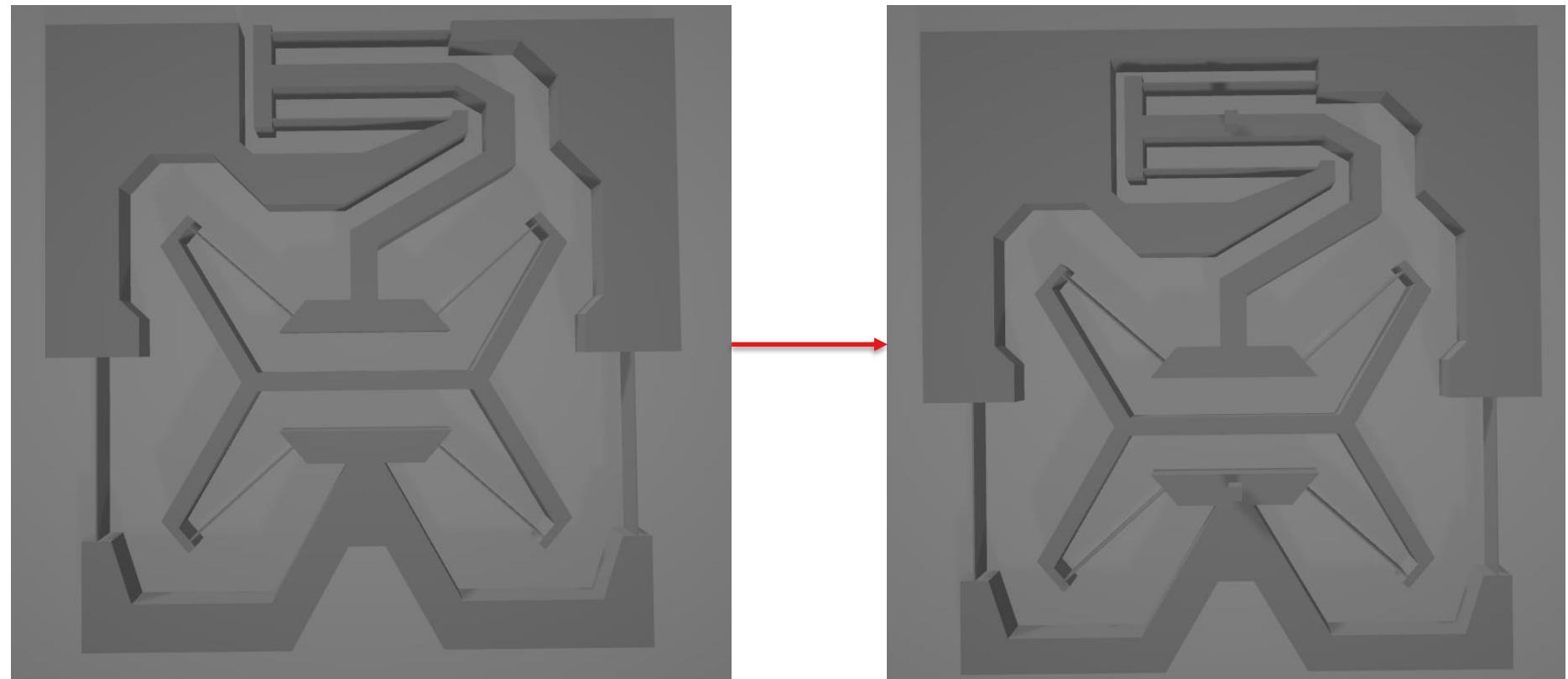


- Matlab figures for behaviour now working
  - Adding stiffness to linear design?
  - Is linearity preferred?
  - Kinetoelastic (non-dimensional analysis)
    - 1. Take only flexures into account
    - 2. The whole mechanism
- Problem is that for same s and eta the mechanism still behaves different
- Planar design just for manufacturing?

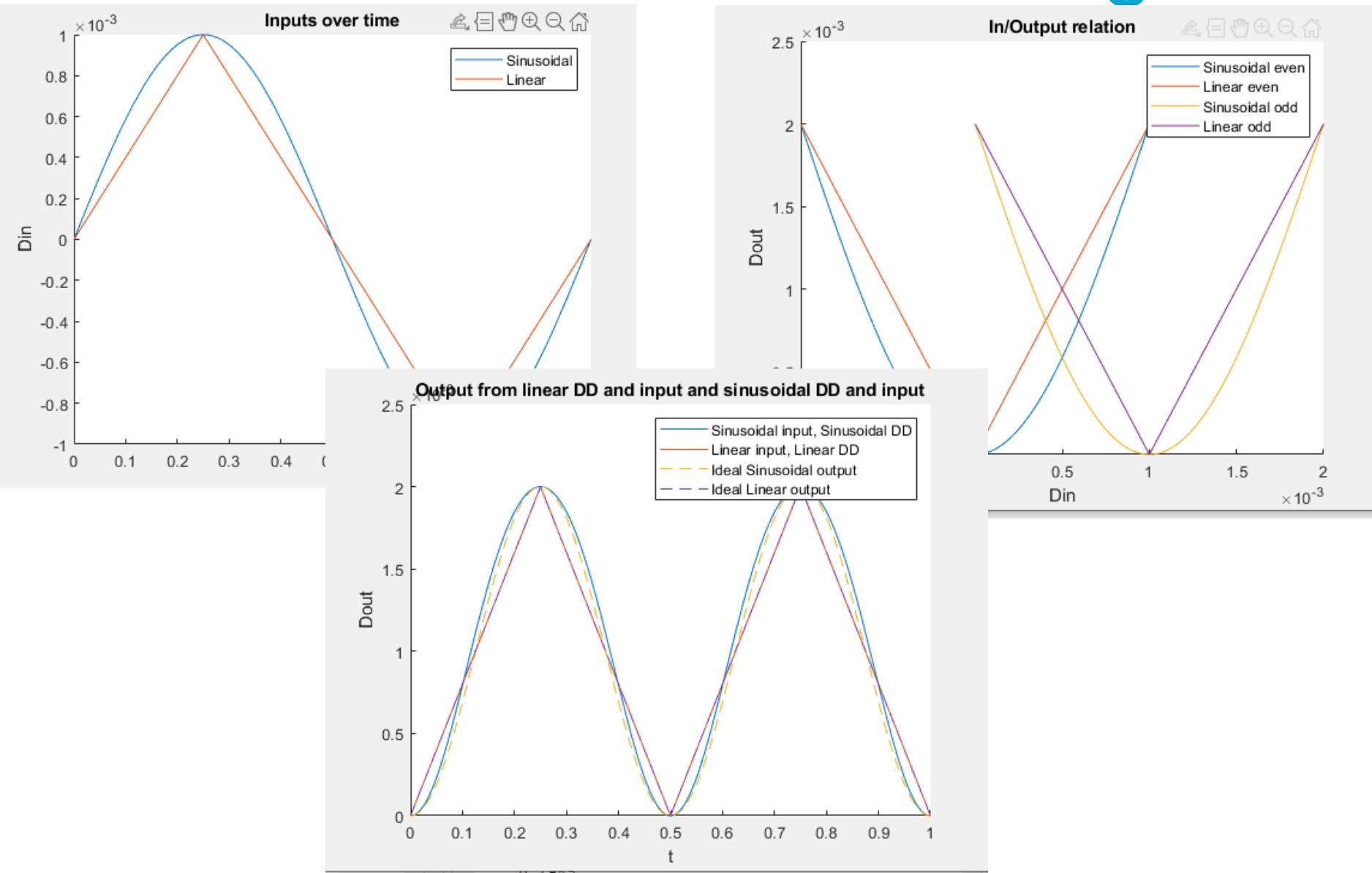
$$s = \frac{\bar{L}}{\bar{d}}$$

$$\eta = \frac{Fs^2}{E\bar{b}\bar{d}}$$

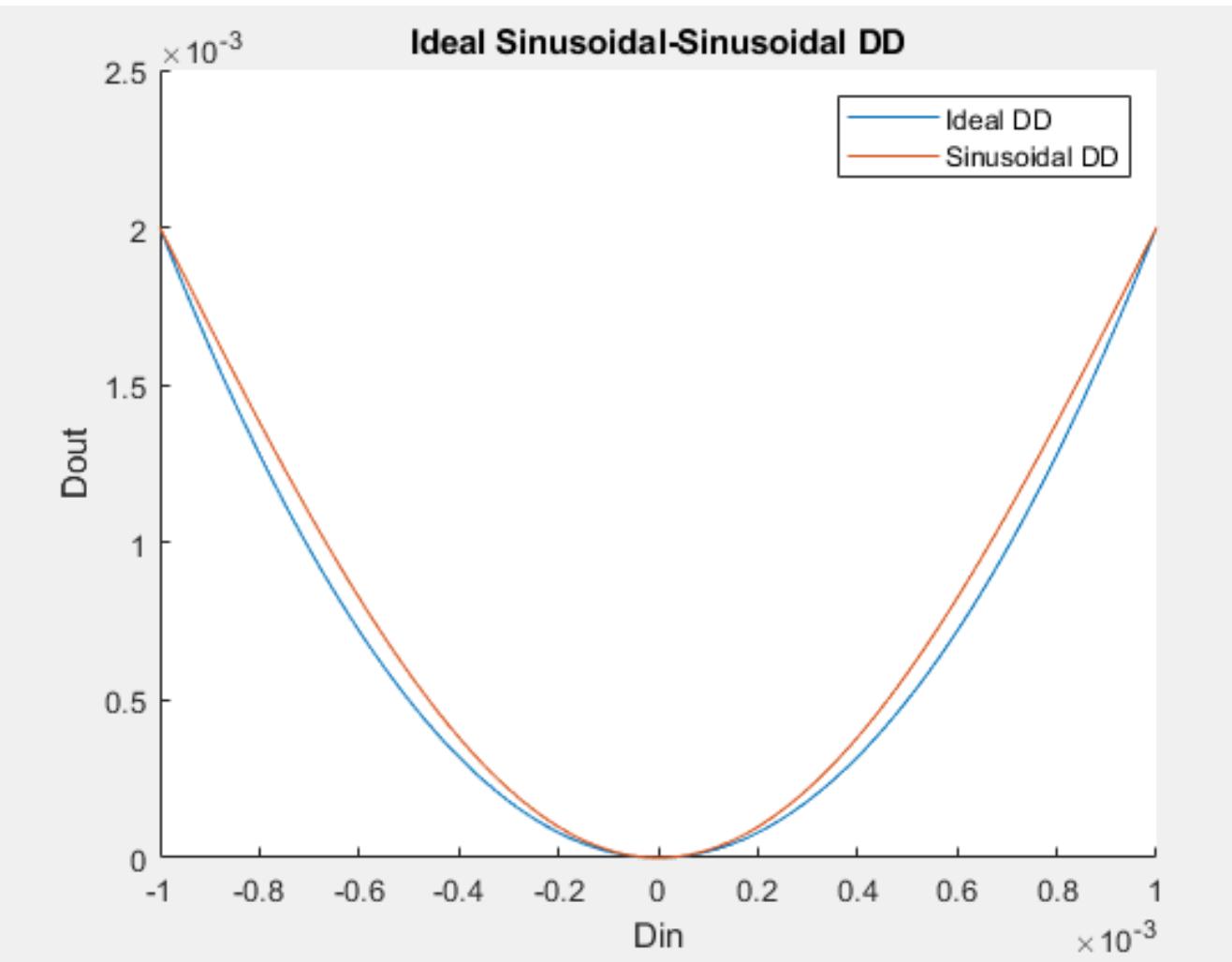
- Sent planar design for printing



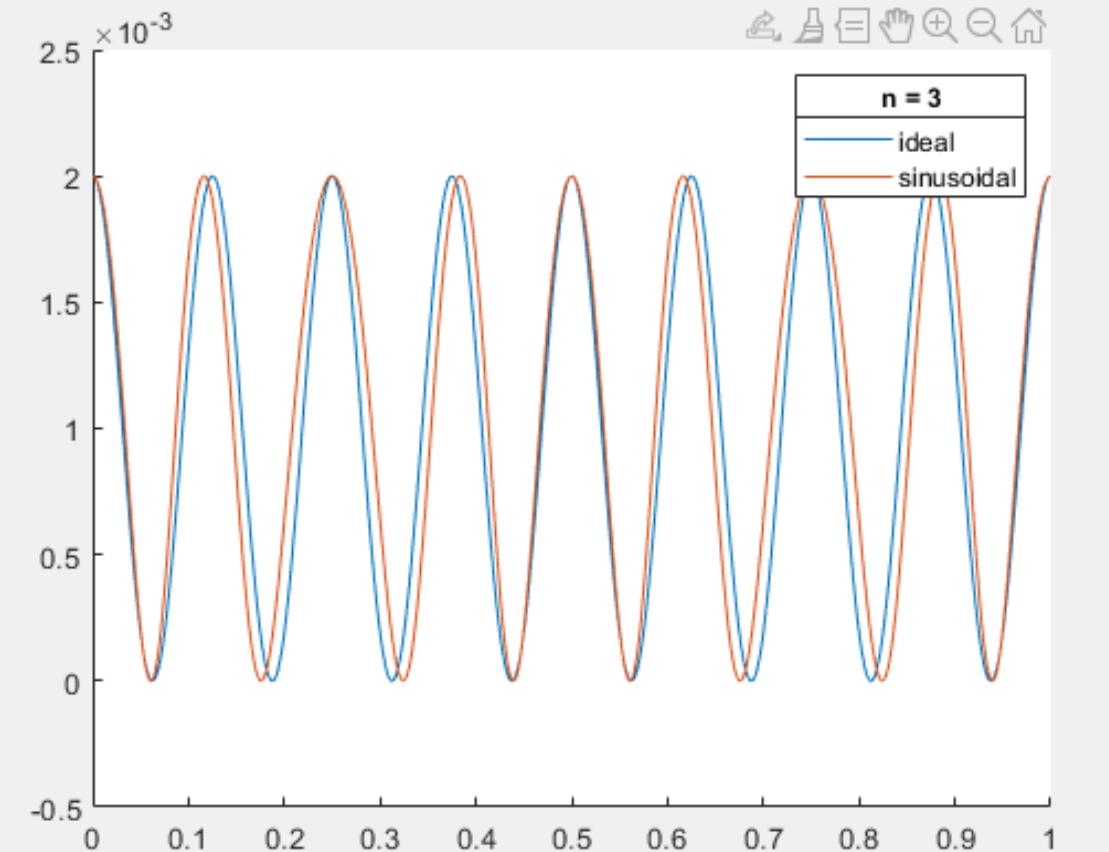
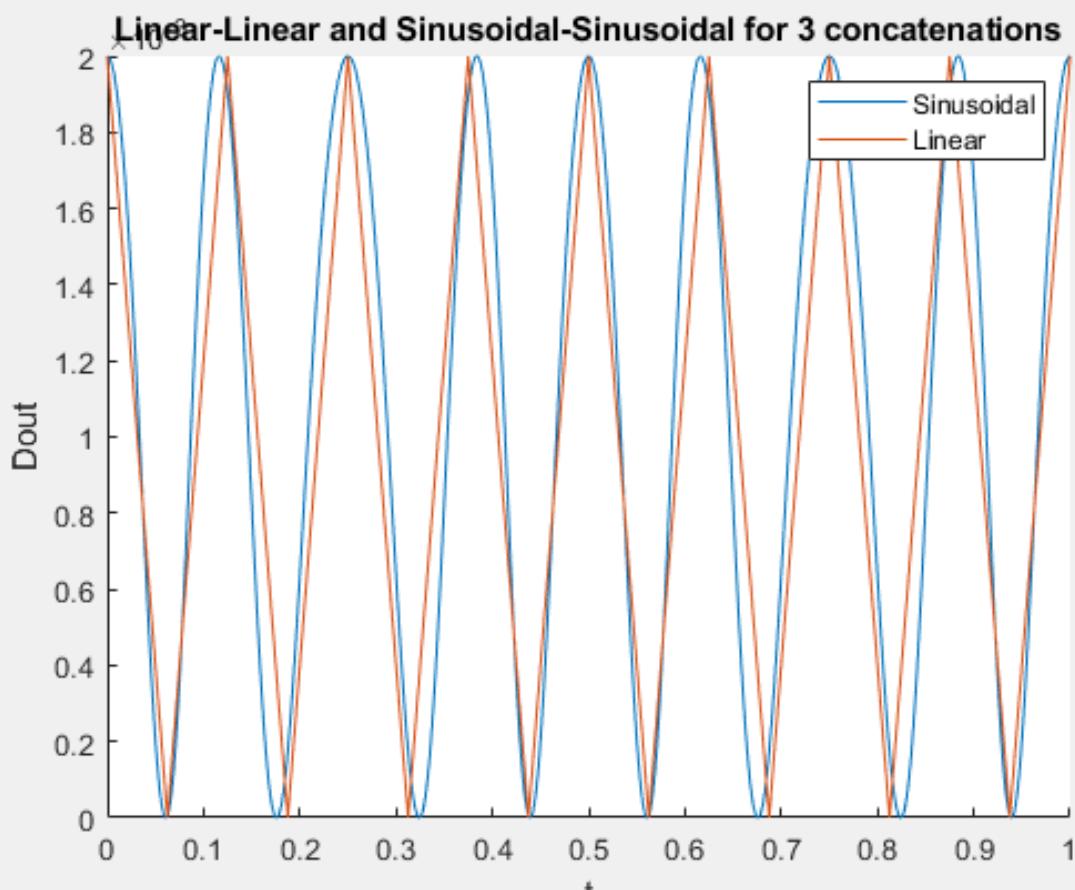
# Linear vs Sinusoidal Design?



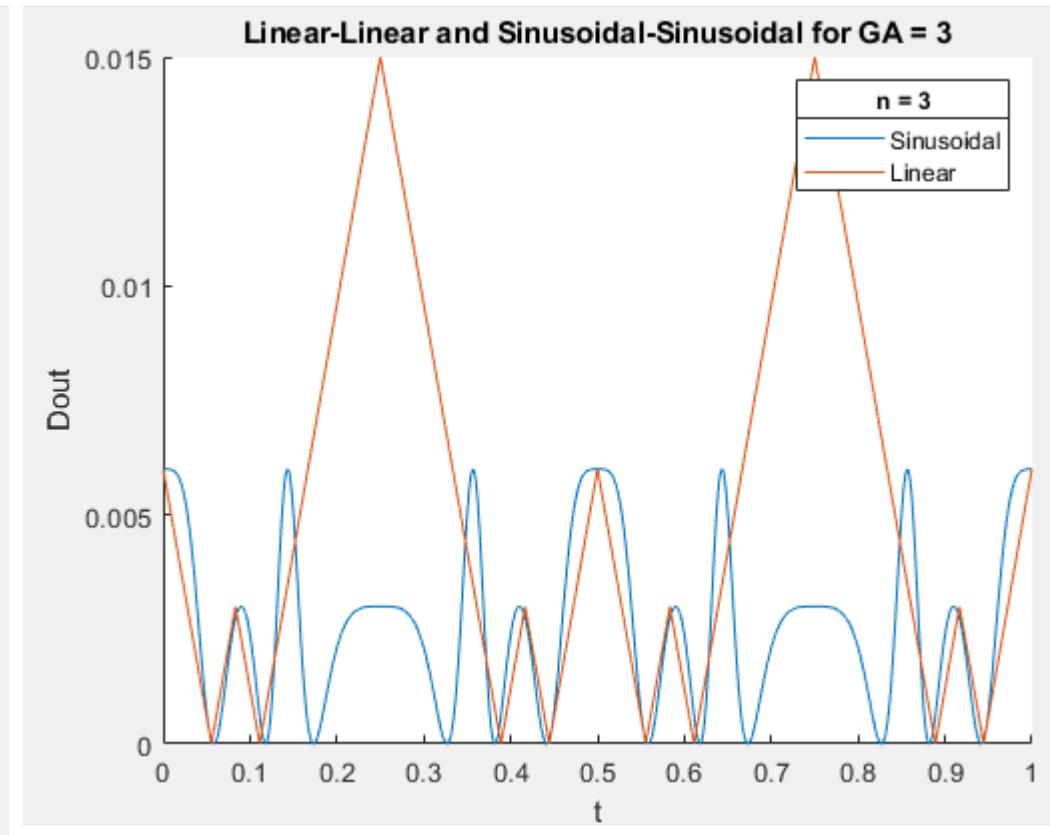
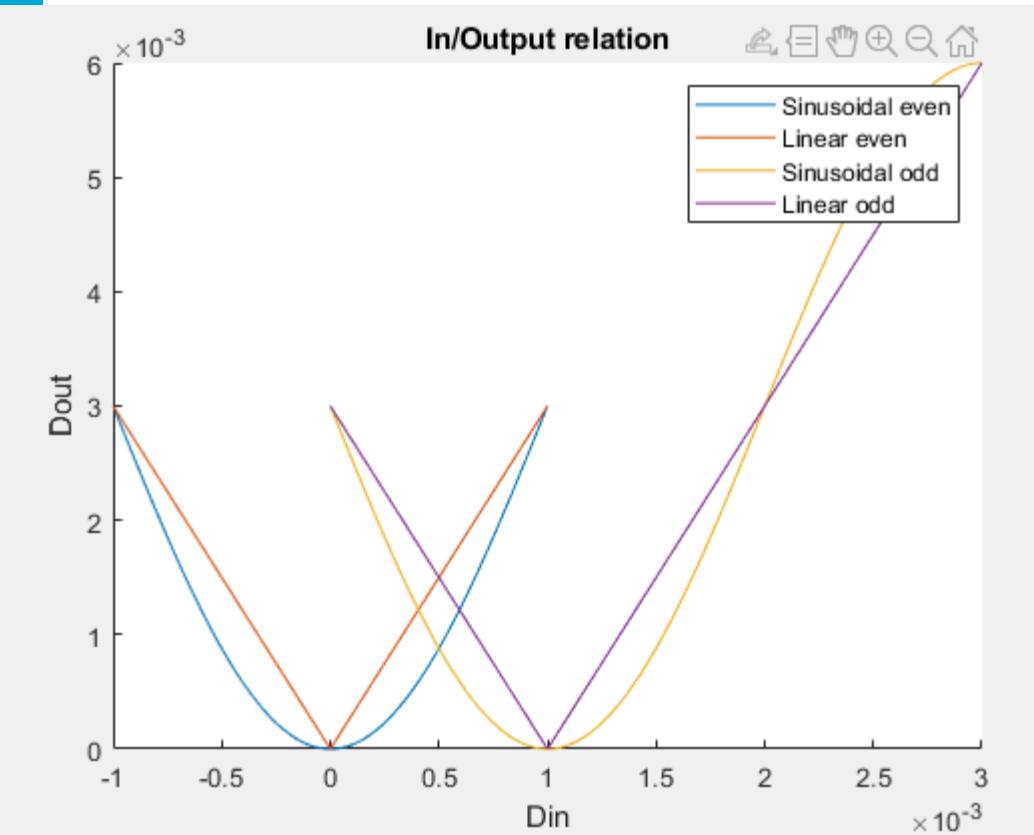
# Input/Output relation GA=2



# 3 concatenations

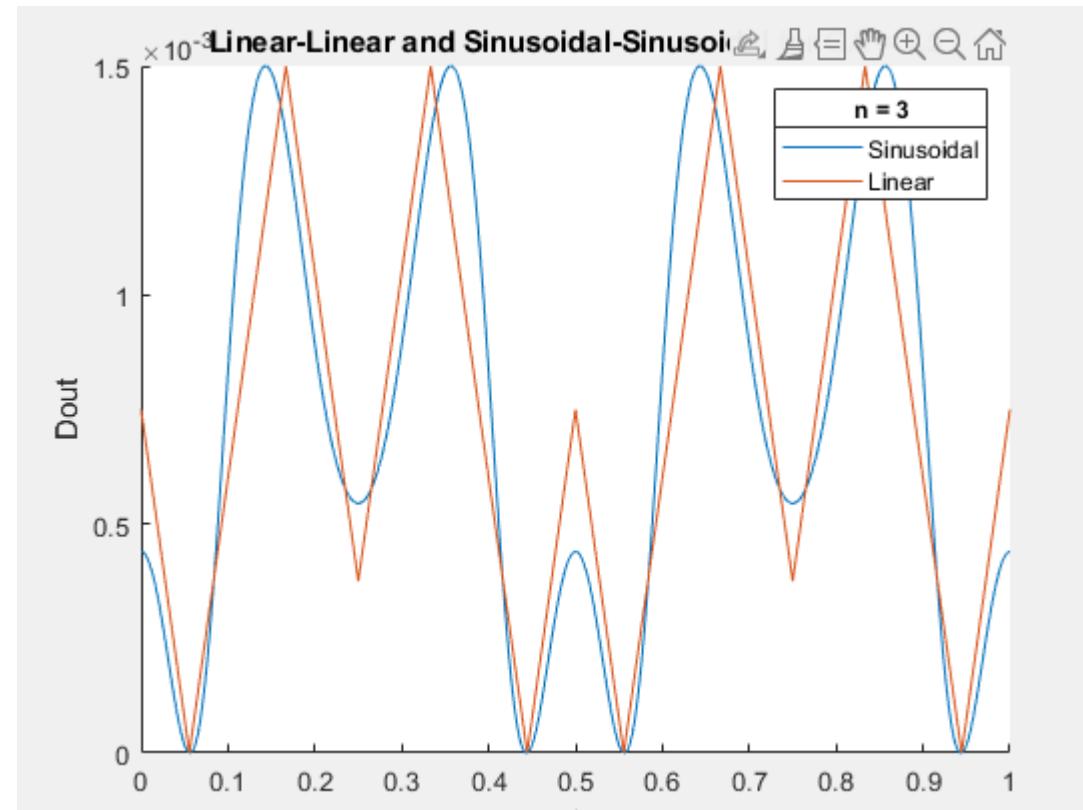
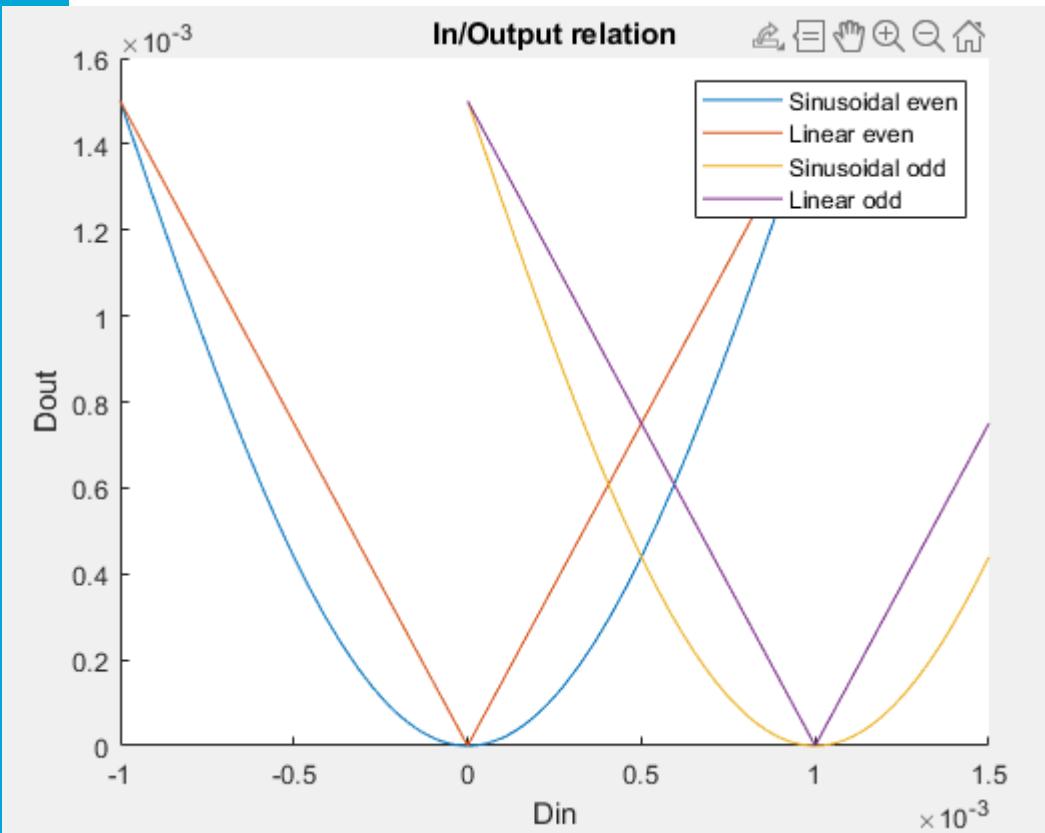


# GA = 3



- How to deal with increase of output displacement?

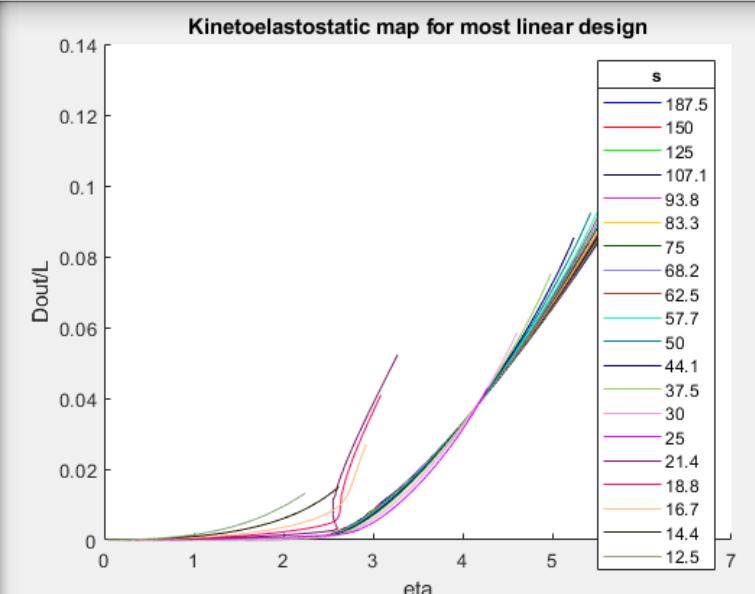
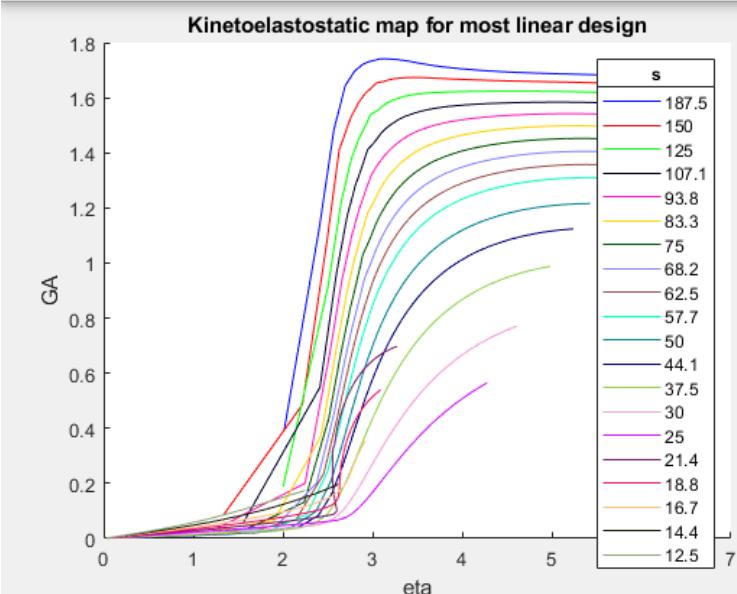
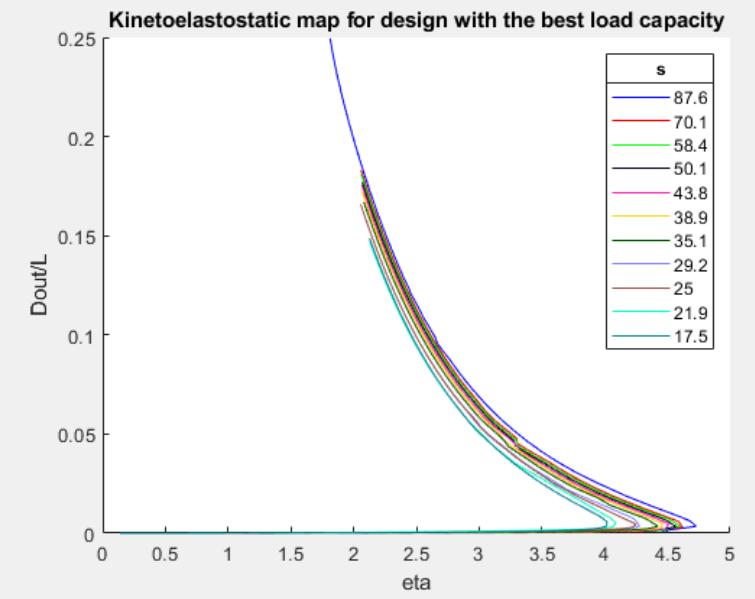
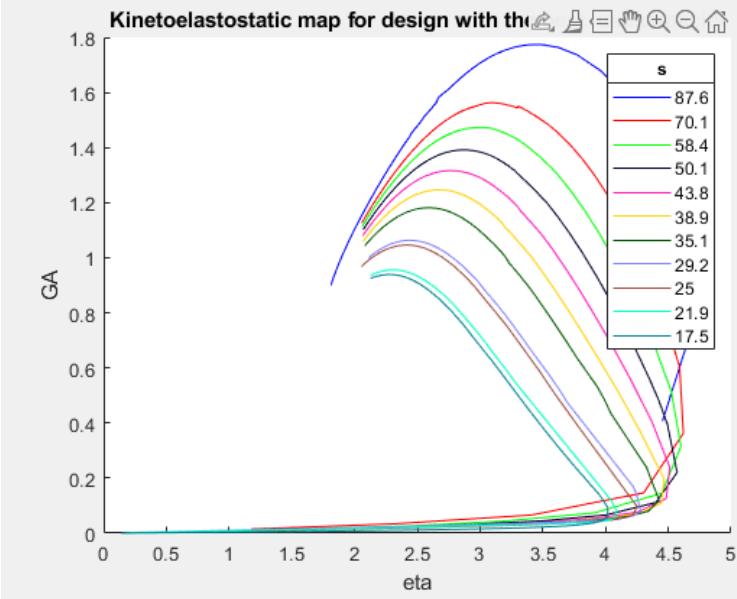
# GA = 1.5



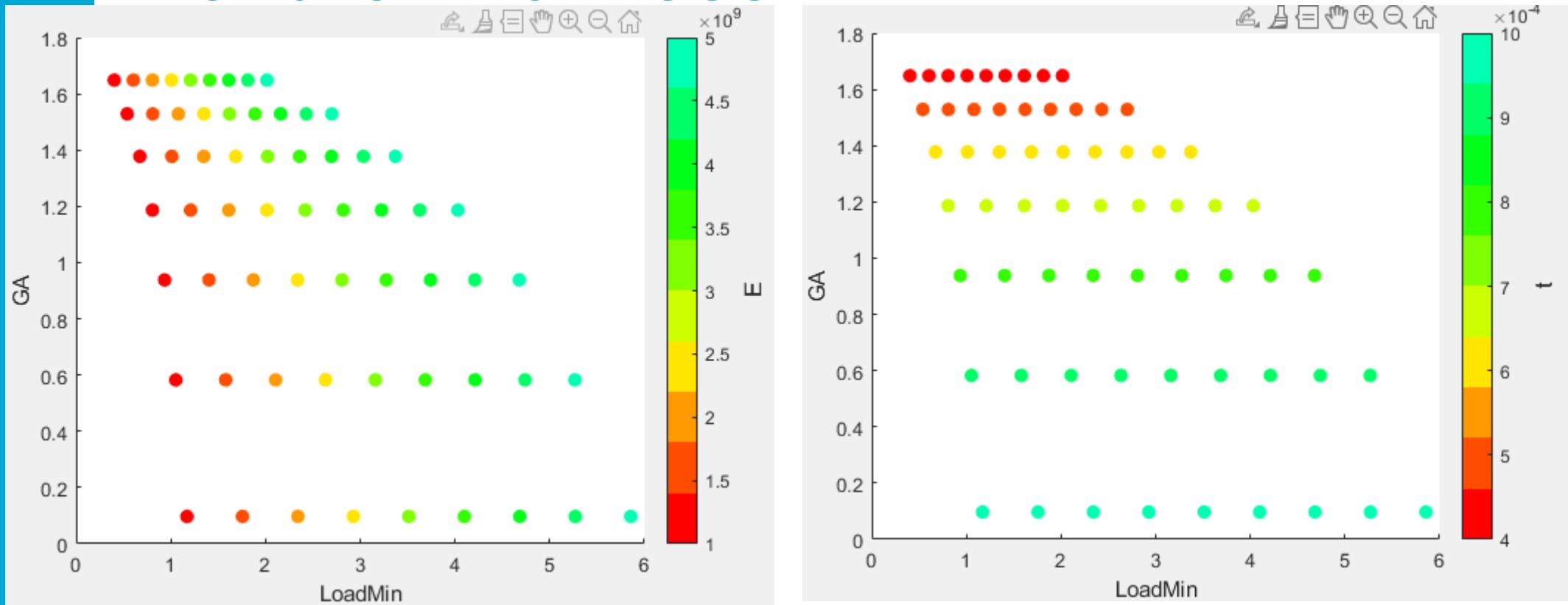
- Looks more manageable than  $GA=3$

# Kinetoelastic

$$\eta = \frac{Fs^2}{Ebd}, S = \frac{L}{d}$$

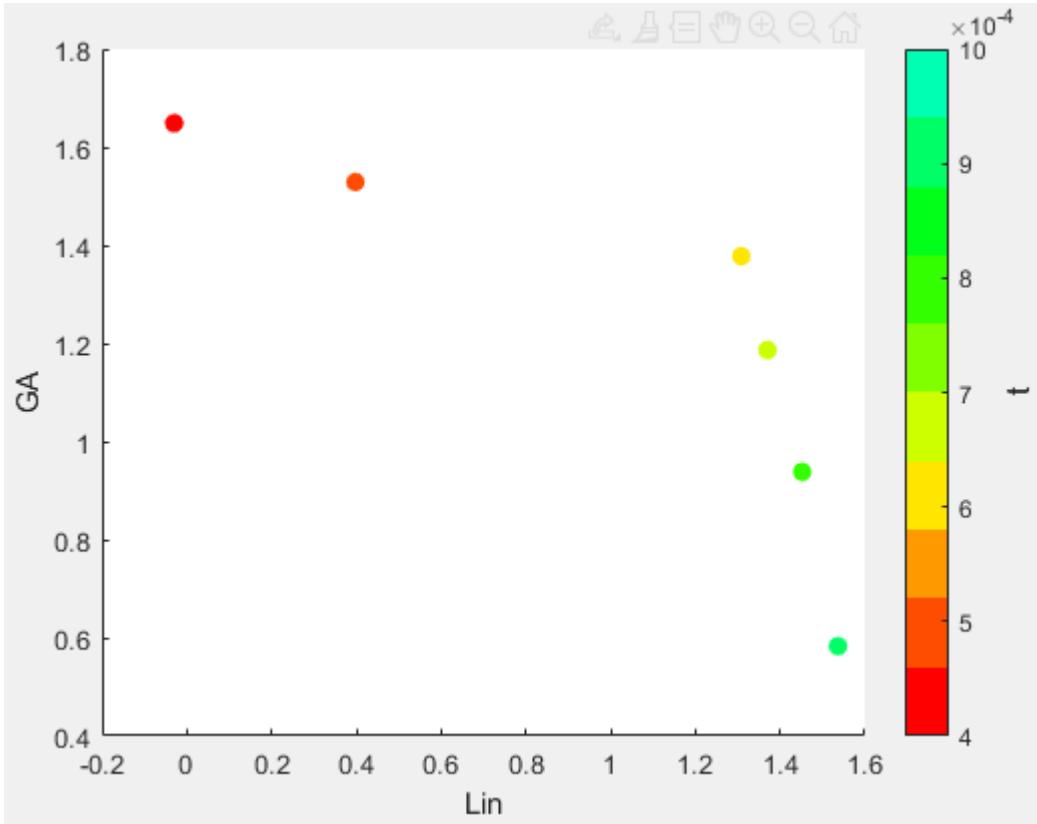


# Varying material property E and flexure thickness t



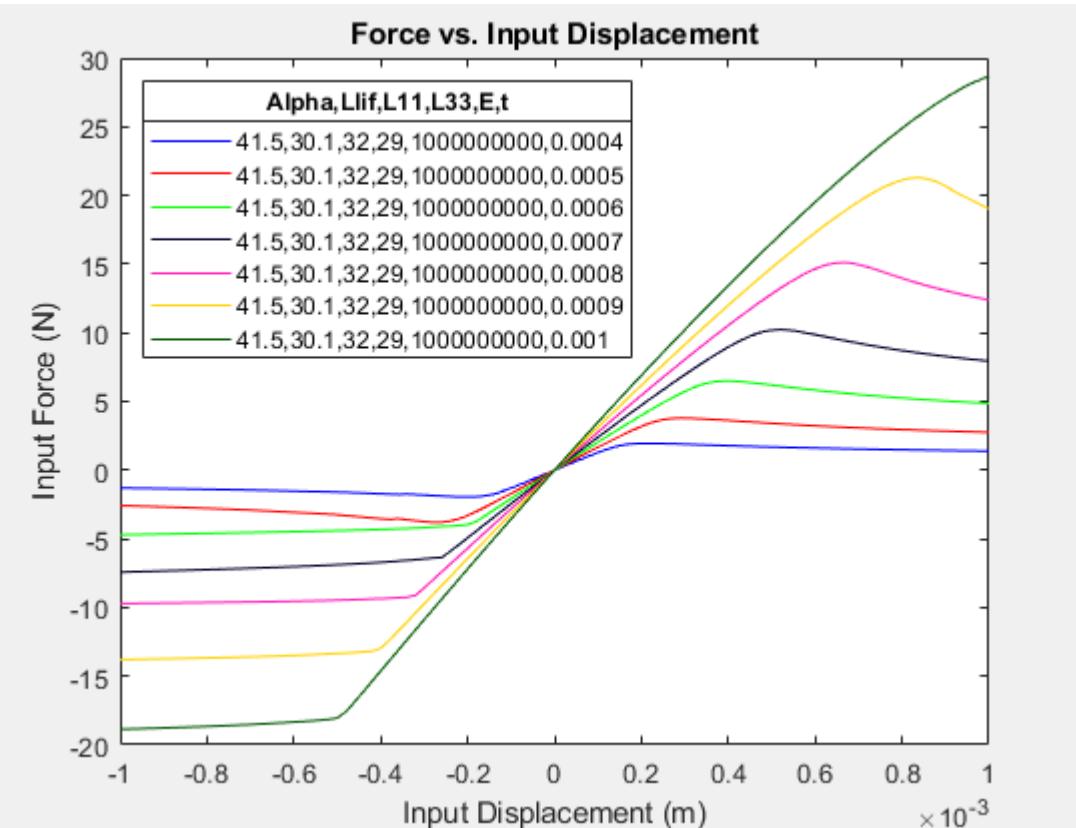
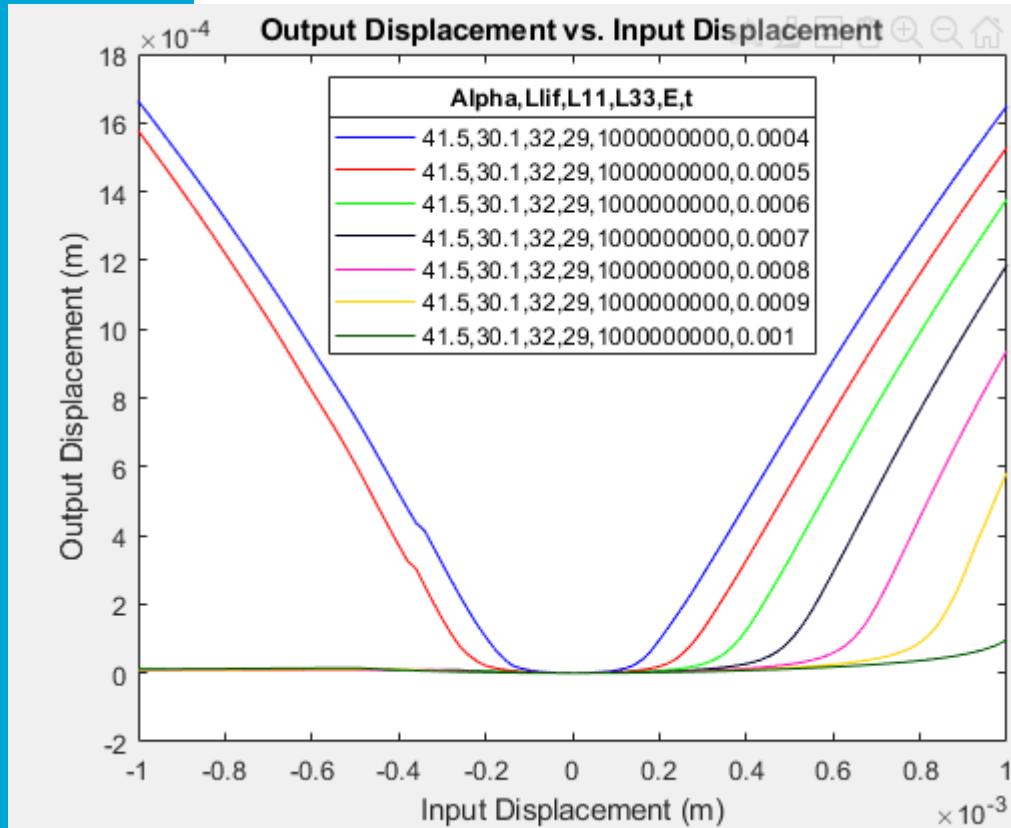
- ↑E means ↑ LoadMin
- ↑t means ↓ GA, E has no effect

# Varying E and t

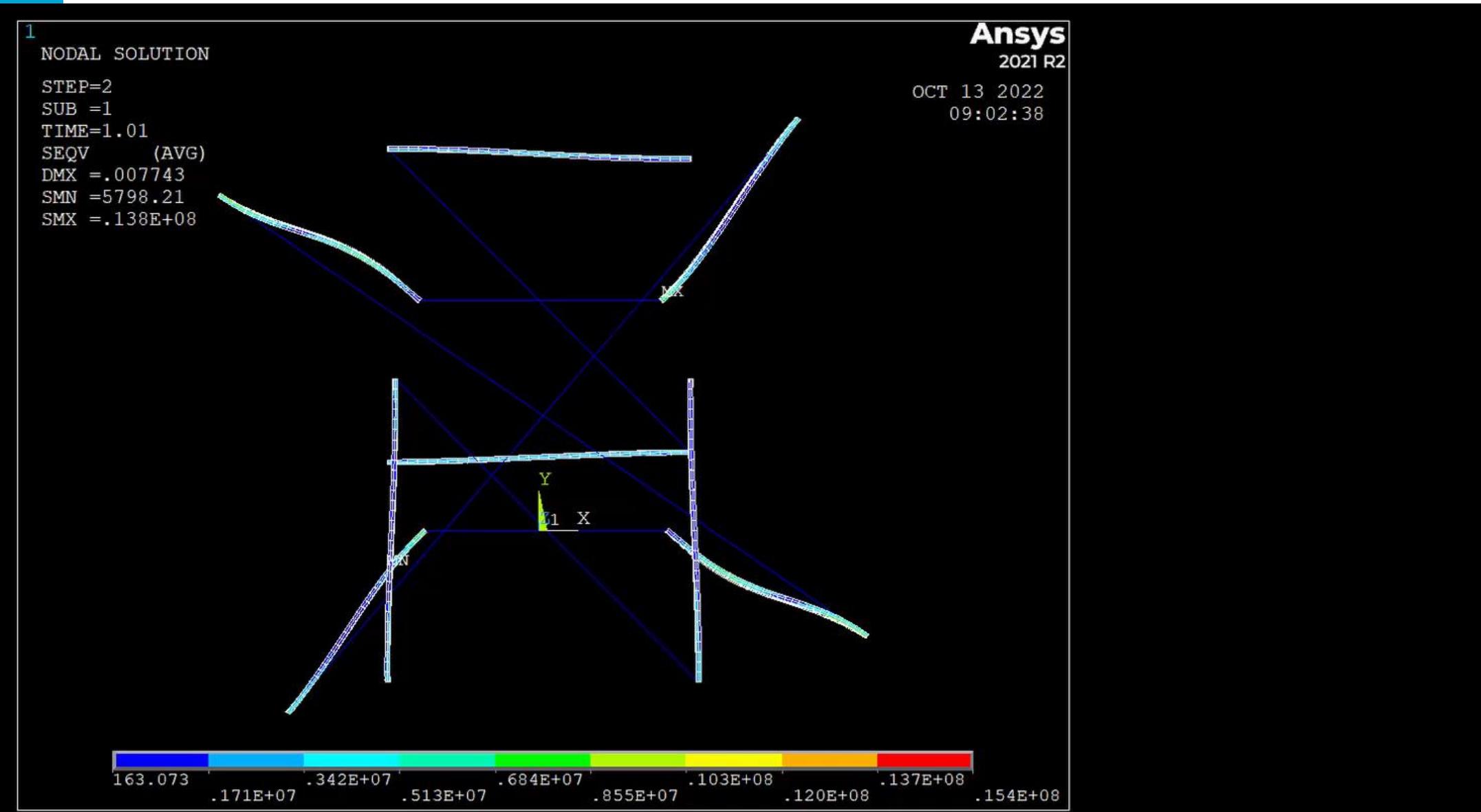


- $\uparrow t$  means  $\downarrow$  GA and worse linearity

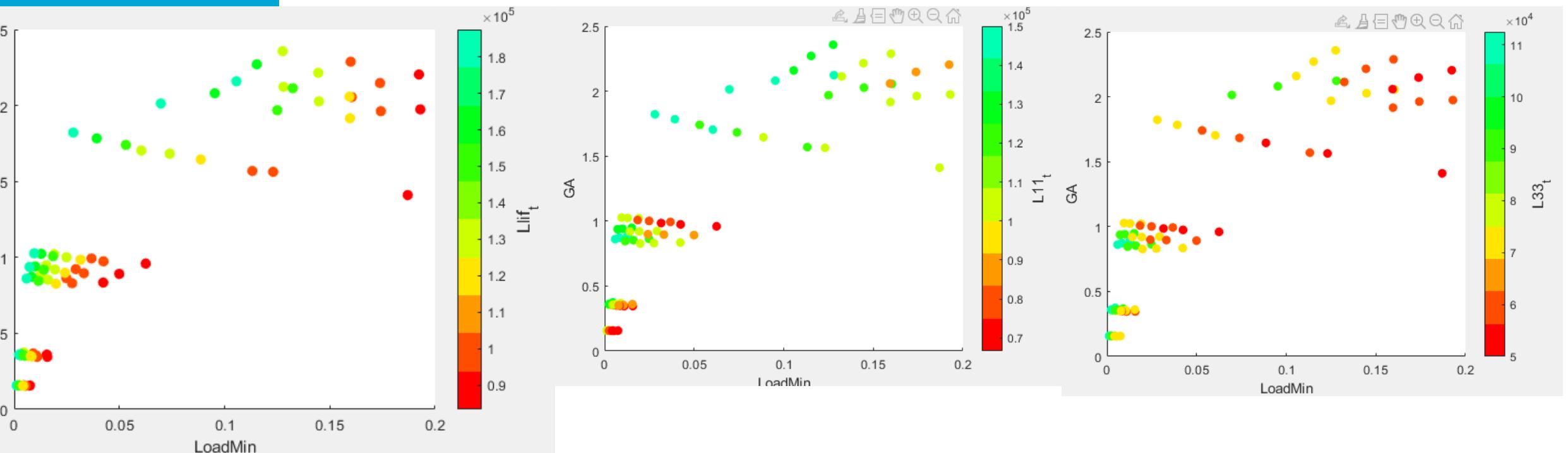
# Varying E and t



# Varying E and t (buckling of output)



# $L_{lift}/t$ , $L_{11}/t$ , $L_{33}/t$ constant



# Resin Printer disadvantages

- Curvature due to curing.
- Not enough space for sufficient stiffness in case of planar design.
- And for non-planar design it is hard to remove flexures.
- Imperfections in the flexure, seemingly large.
- Long waiting time
- Pooling of resin
- Haven't been able to make the design modular

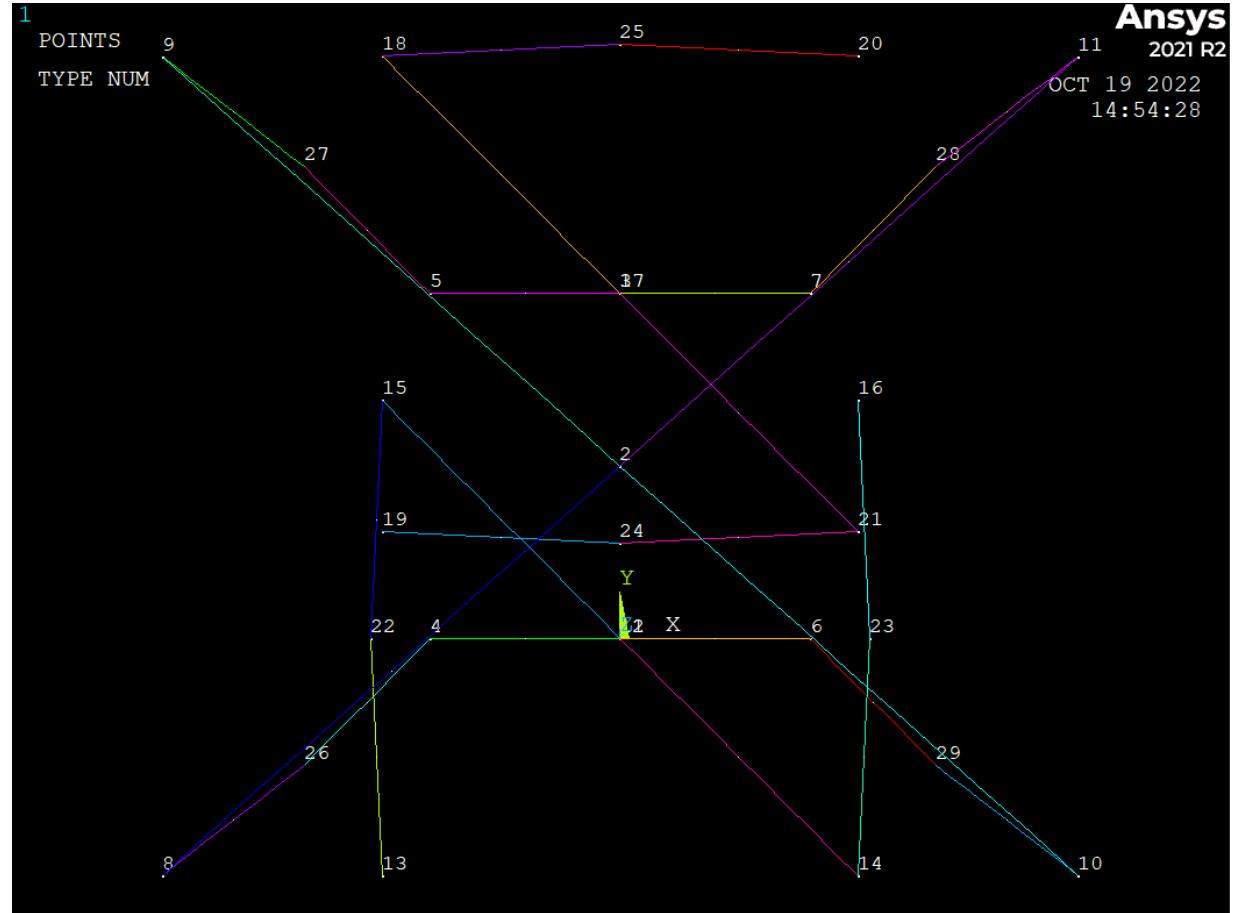
# Options for printing?

- PTU,PTG?  $t = 0.6\text{-}0.7\text{mm}$ 
  - Imperfections in flexures? Of what magnitude?
- Leaf spring steel?  $t = 0.1\text{mm}$ 
  - No imperfections due to printing
  - A lot of evidence it works in literature

# Imperfections

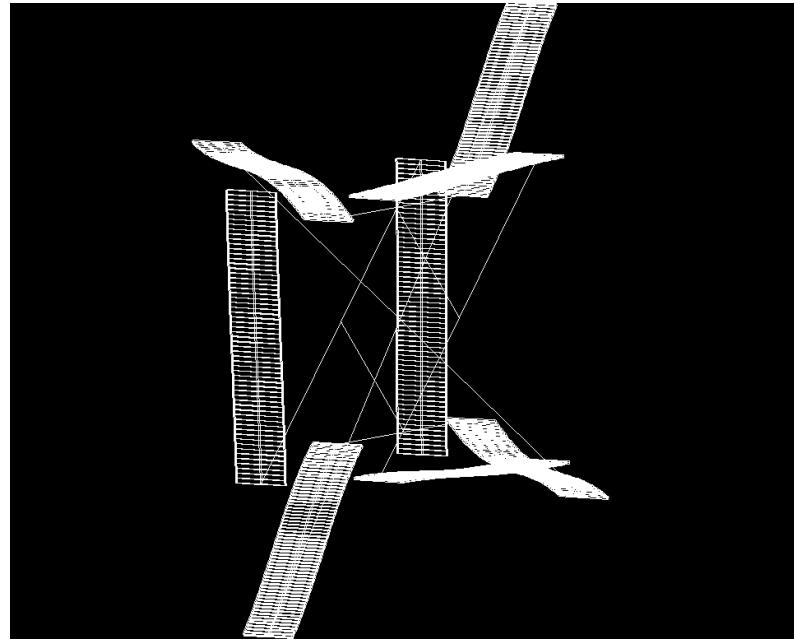
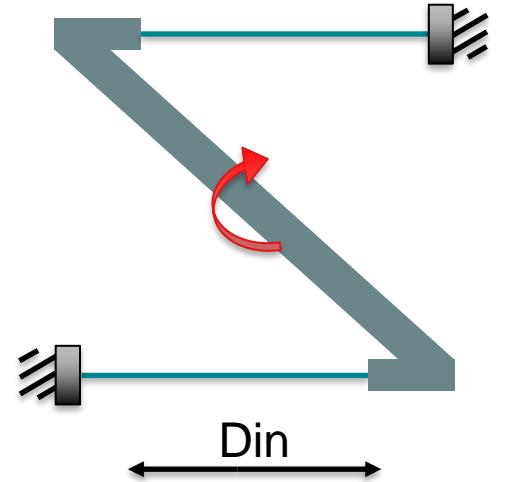
Exaggerated in the image

- Buckling of output happens earlier



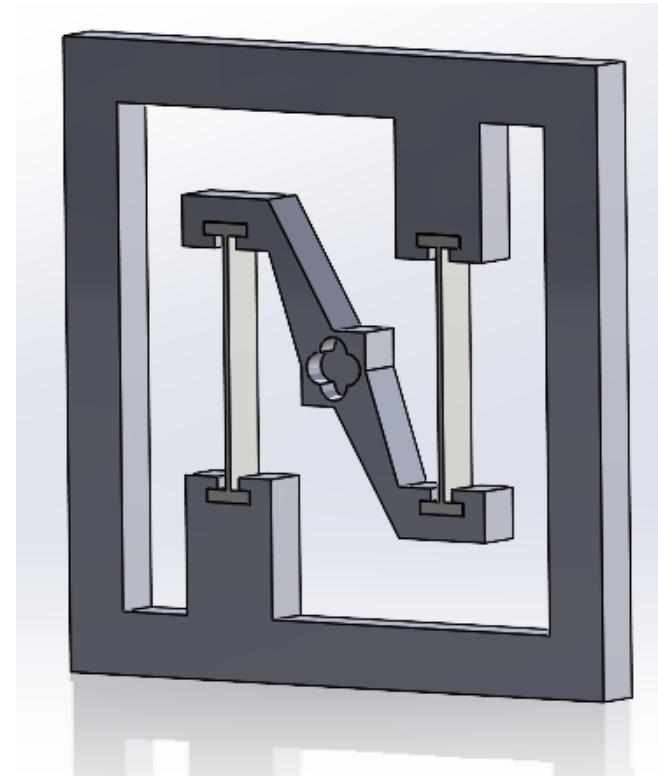
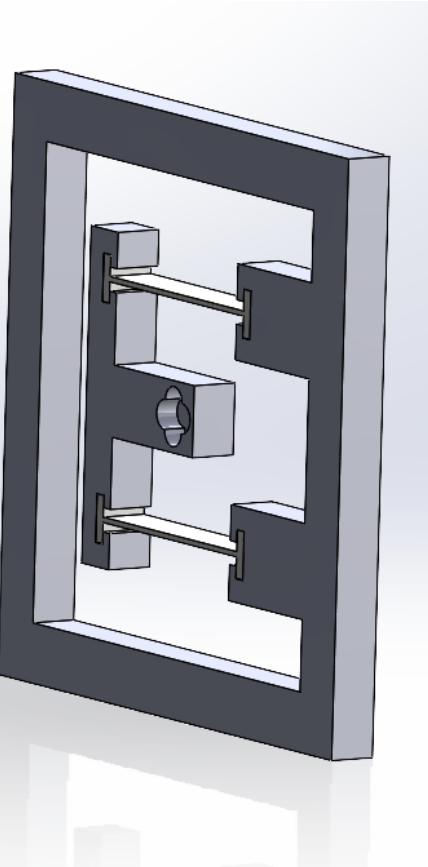
# Improving stiffness of output

- Improve stiffness?
- Tried to displace output shuttle to not cause a resulting moment.
  - Output at same height as input. X
  - Output and Input both at same height as middle of mechanism. Works!

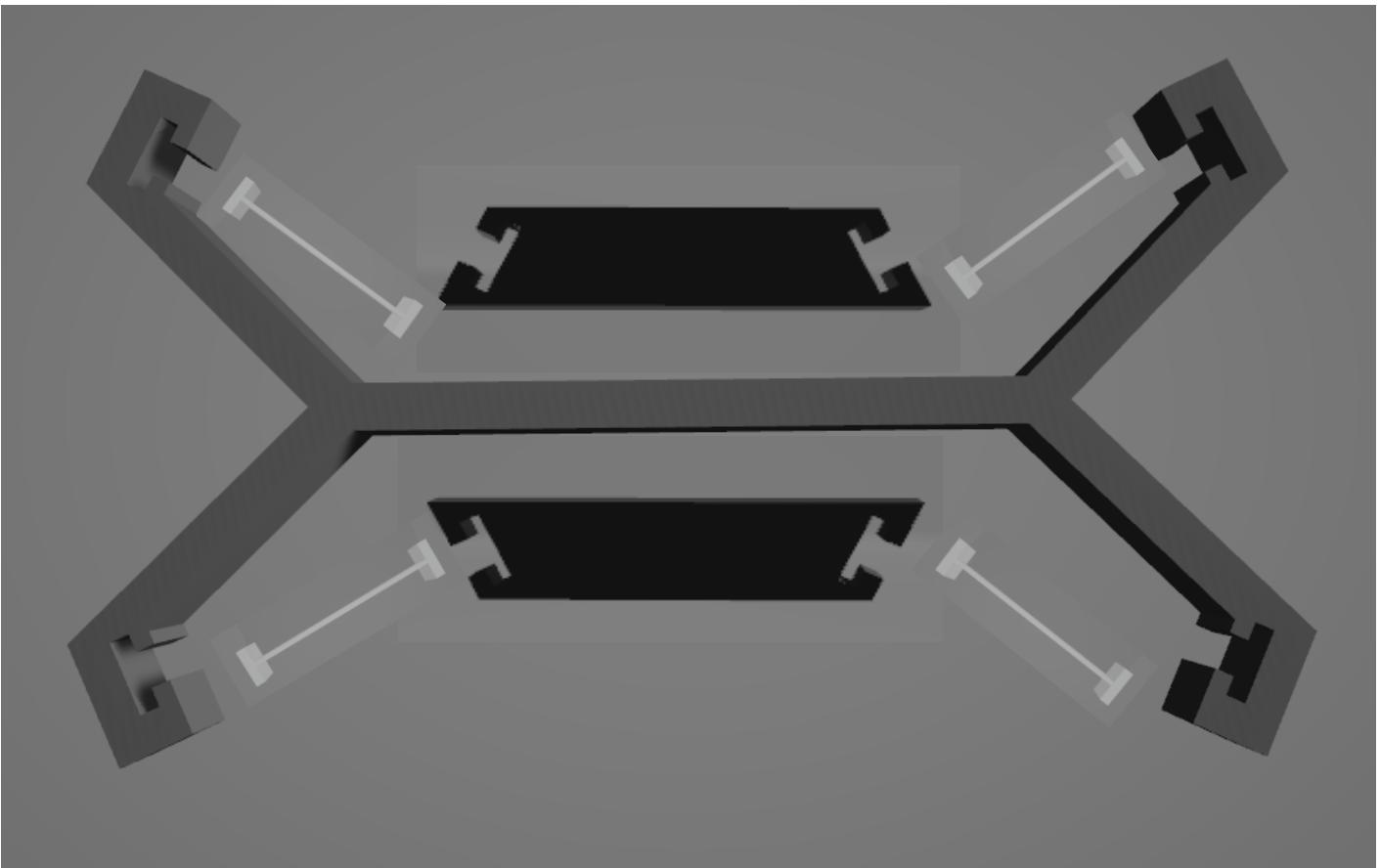


# Modular design, to increase manufacturing robustness

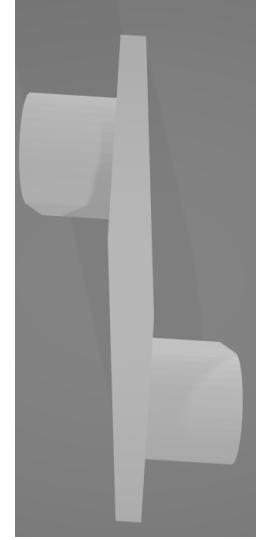
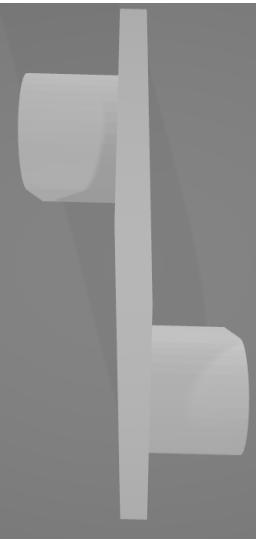
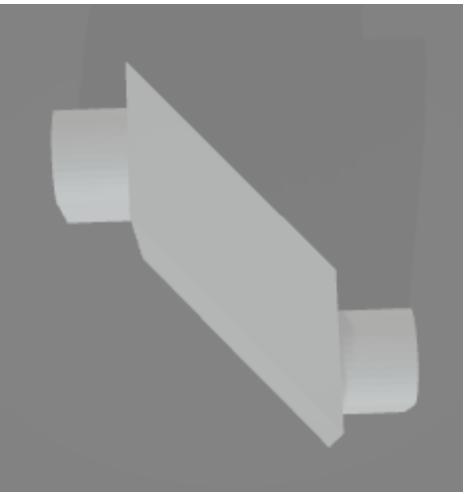
- T-shaped input/output structure or diagonal?



# Butterfly flexure



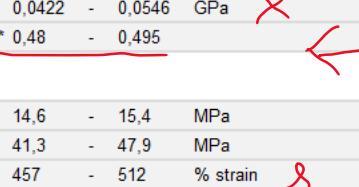
# Iteration process of connection piece



# Material properties

- Poisson's ratio = 0.49
- Yield strength = 8 MPa

## TPU D45, edupack

TPU (Ether, aromatic, Shore D45)			
Datasheet view:		Materials data for simulation	Show/Hide
<b>Composition overview</b>			
Compositional summary <small>i</small>			
(NH-R-NH-CO-O-R'-O-CO)n where R is from an aromatic diisocyanate, usually MDI or TDI (hard segment), and R' is from a short or long chain polyether diol (soft segment)			
Material family	<small>i</small>	Elastomer (thermoplastic, TPE)	
Base material	<small>i</small>	TPU (Thermoplastic polyurethane elastomer)	
Polymer code	<small>i</small>	TPU	
<b>Physical</b>			
Density	<small>i</small>	1,13e3 - 1,15e3 kg/m <sup>3</sup>	
<b>Elastic</b>			
Young's modulus	<small>i</small>	0,0422 - 0,0546 GPa	X
Poisson's ratio	<small>i</small>	* 0,48 - 0,495	
<b>Plastic</b>			
Yield strength (elastic limit)	<small>i</small>	14,6 - 15,4 MPa	
Tensile strength	<small>i</small>	41,3 - 47,9 MPa	
Elongation	<small>i</small>	457 - 512 % strain	

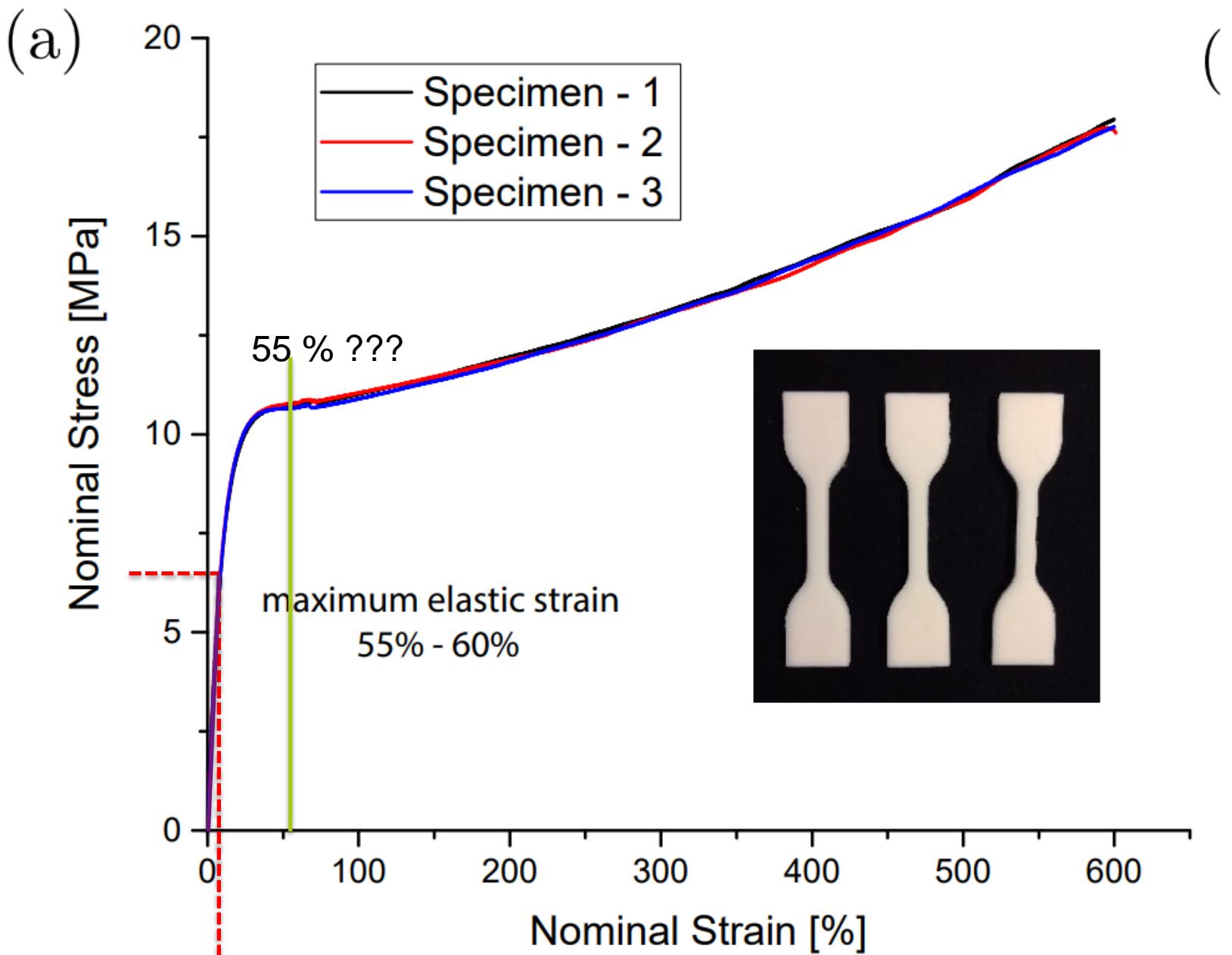
## TPC Flex 45, Datasheet

### Physical properties

Description	Testmethod	Typical value
Specific gravity	ISO 1183	1,14 g/cc
Melt volume flow rate	ISO 1133	39 cm <sup>3</sup> /10 min
Stress at break	ISO 527	24
Strain at break	ISO 527	530%
Tensile modulus	ISO 527	95 Mpa
Impact strength Charpy method 23°C	ISO 179	Notched No break
Shore D Hardness	ISO 868	45

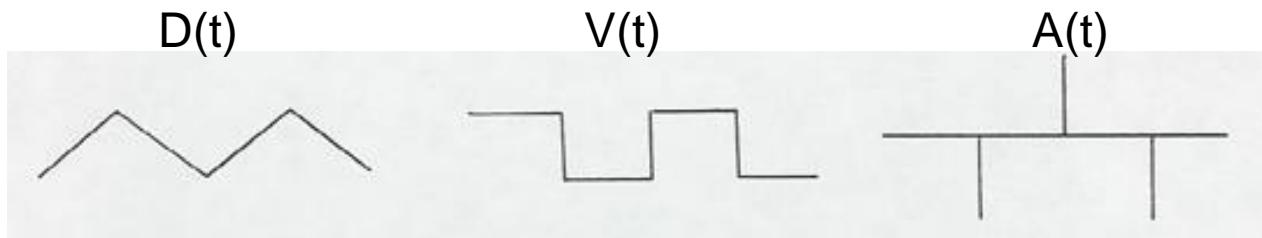
## TPC D40, edupack

TPC (Shore D40)			
Datasheet view:		Materials data for simulation	Show/Hide
<b>Compositional summary</b> <small>i</small>			
Block copolymer of a polyester (semi-crystalline hard phase - usually PBT, polybutylene terephthalate) and polyether (amorphous soft phase).			
Material family	<small>i</small>	Elastomer (thermoplastic, TPE)	
Base material	<small>i</small>	TPC (Thermoplastic copolyester-ether elastomer)	
Polymer code	<small>i</small>	TPC	
<b>Physical</b>			
Density	<small>i</small>	1,14e3 - 1,18e3 kg/m <sup>3</sup>	
<b>Elastic</b>			
Young's modulus	<small>i</small>	0,0708 - 0,115 GPa	
Poisson's ratio	<small>i</small>	* 0,476 - 0,495	
<b>Plastic</b>			
Yield strength (elastic limit)	<small>i</small>	6,64 - 9,48 MPa	
Tensile strength	<small>i</small>	17,5 - 22,7 MPa	
Elongation	<small>i</small>	226 - 324 % strain	



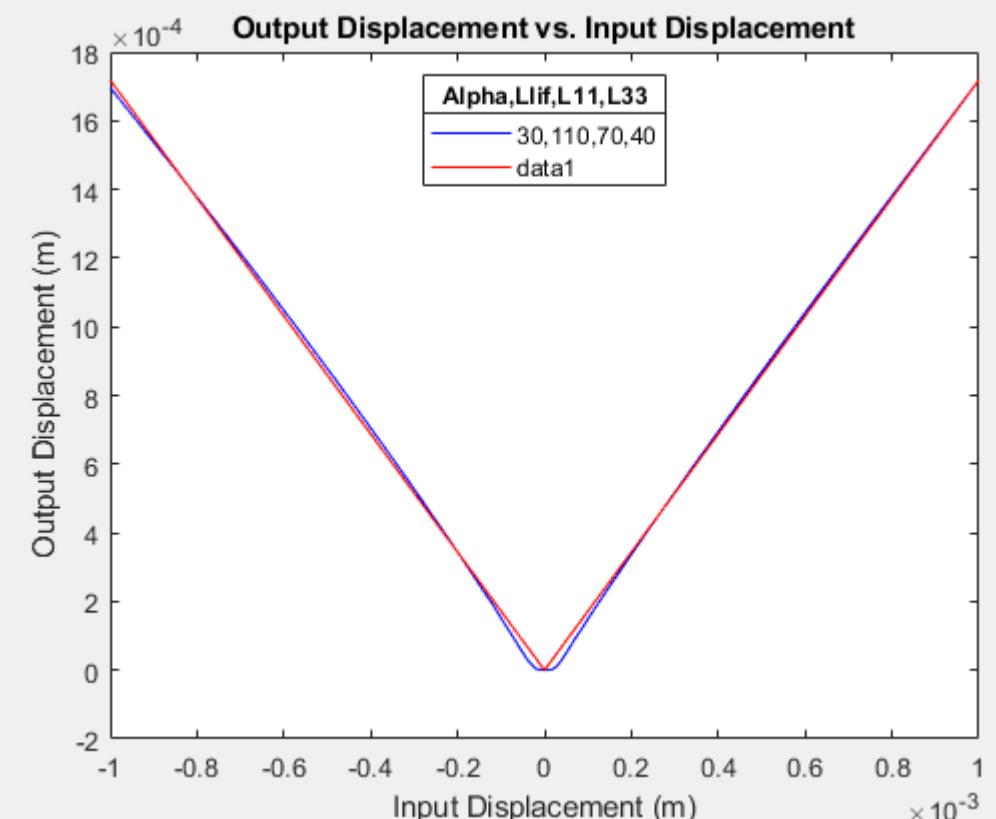
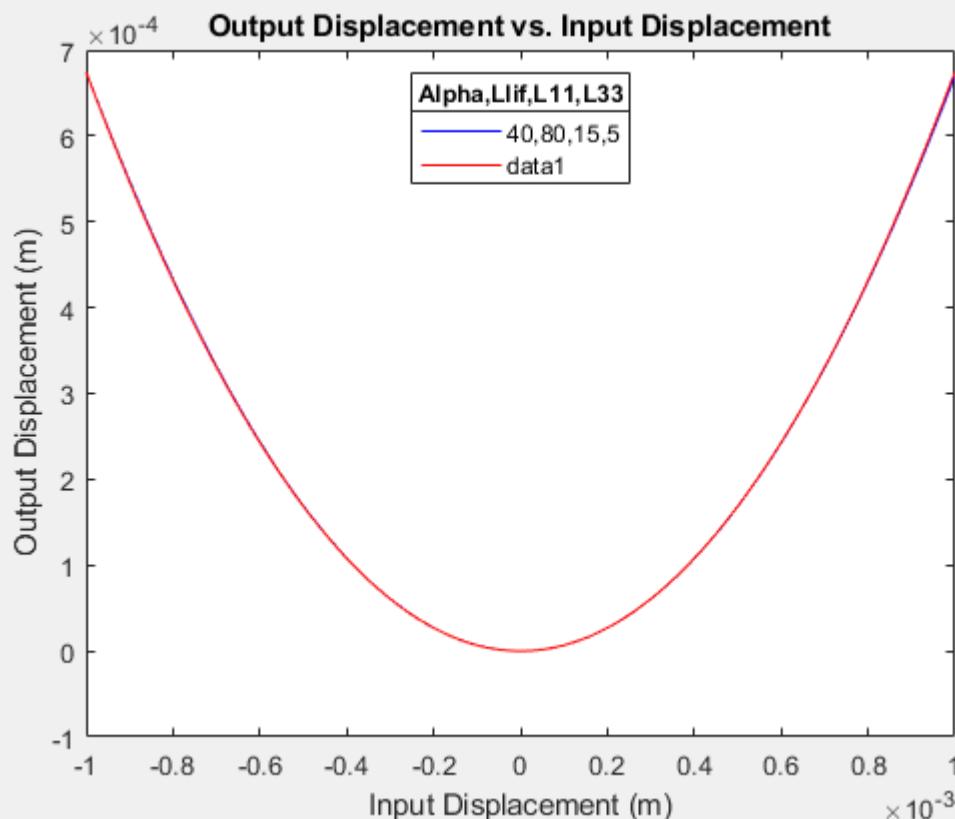
# Sinusoidality Vs. Linearity

- More achievable (see next slide)
  - From available literature can already see the struggle to create perfect linear.
  - Harder to make due to infinite acceleration when input/output switch directions
  - Same for input, needs high harmonics to be created and will always be an approximation.
- More cyclical behavior
  - Would be preferable for a lot of applications
  - No high frequency harmonics and large scale of frequencies, which could drive an unwanted eigenmode of the system
    - [https://en.wikipedia.org/wiki/Triangle\\_wave](https://en.wikipedia.org/wiki/Triangle_wave)
    - For actuation would be big disadvantage



# Achievable?

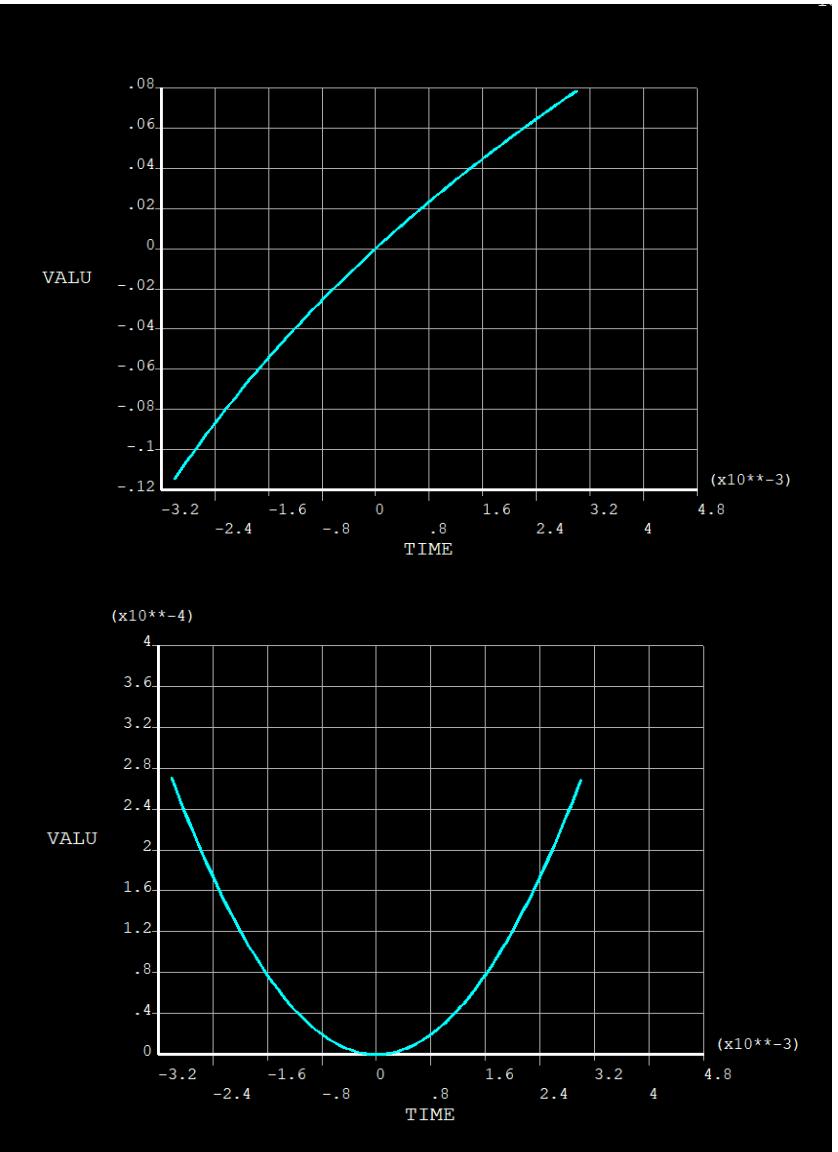
- Red = ideal, Blue = best sinusoidality or linearity



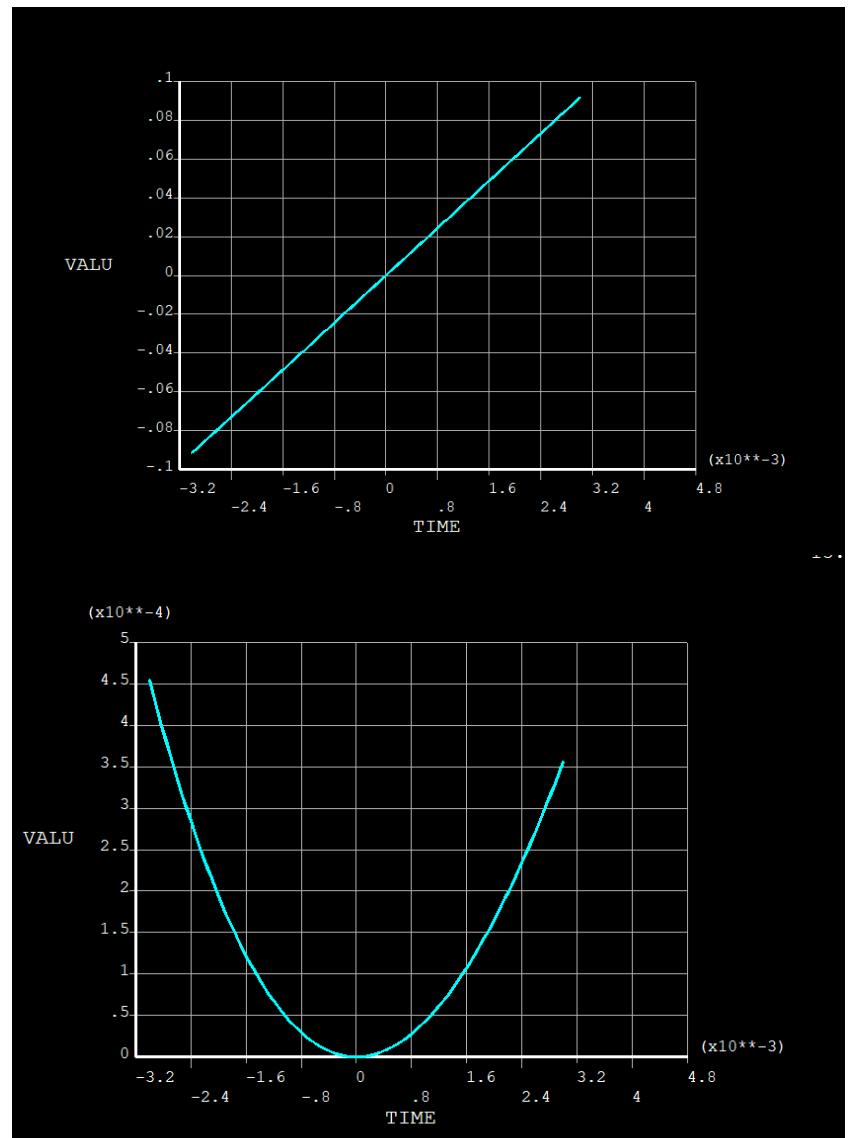
# Show planning

- Only looking at Sinusoidal DD
- Manufacturing to concatenate once, might take too long

# Parallel vs. Crossed [10,45,25,55]

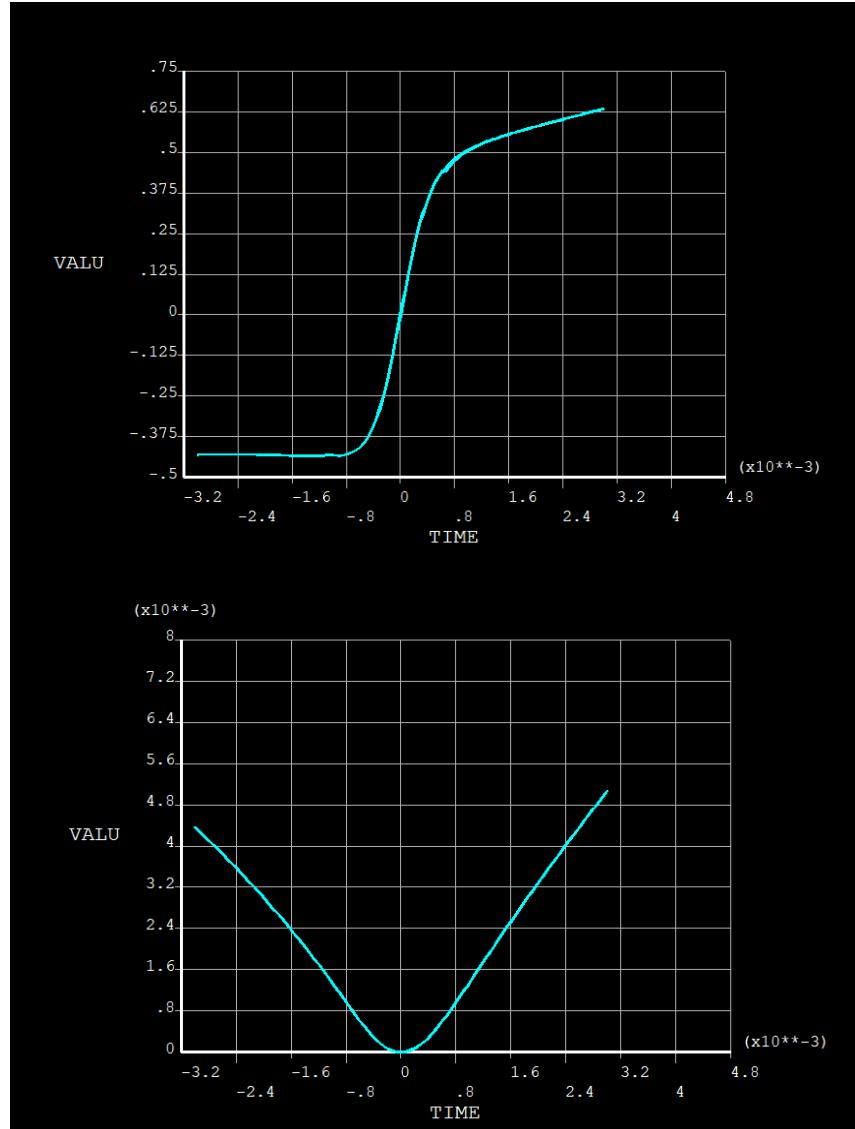


Parallel

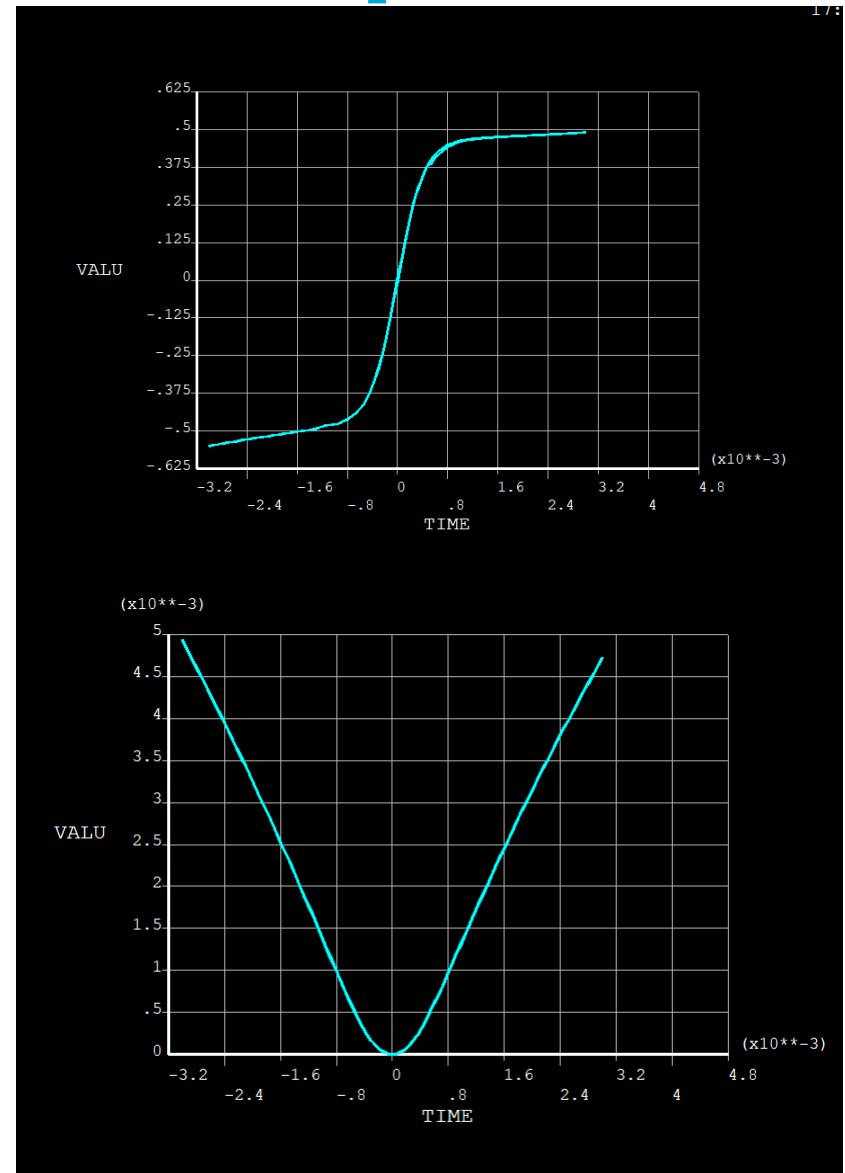


Crossed (original)

# Parallel vs. Crossed [20,45,40,20]



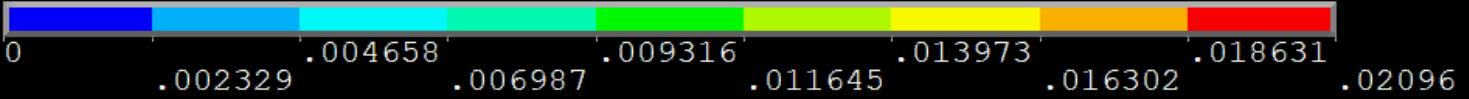
Parallel



Crossed (original)

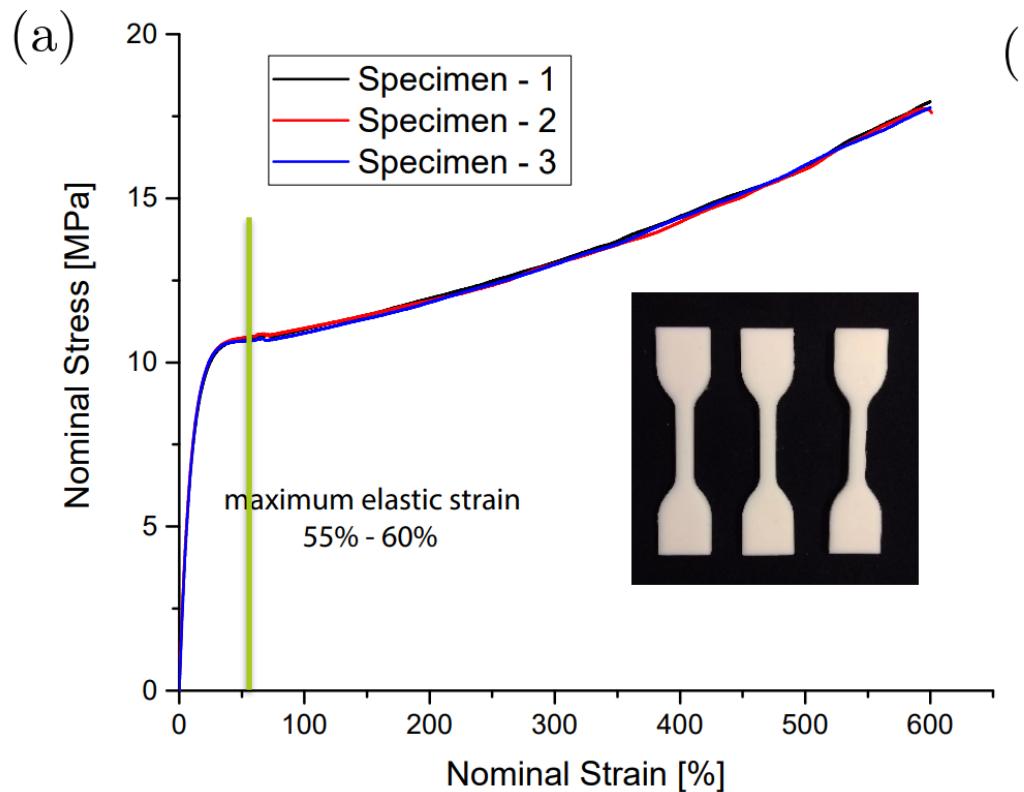
NOV 22 2022  
18:24:22

1  
NODAL SOLUTION  
STEP=2  
SUB =1  
TIME=1.01  
USUM (AVG)  
RSYS=0  
DMX =.02096  
SMN =0  
SMX =.02096



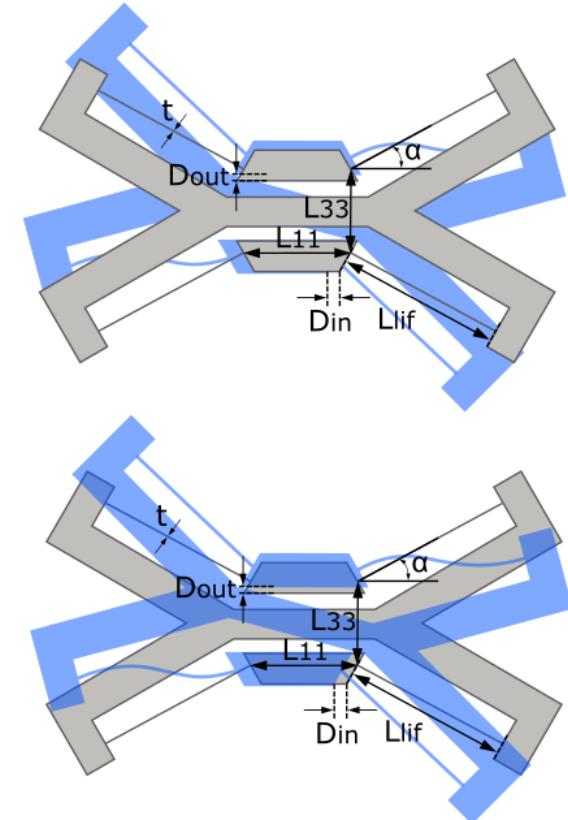
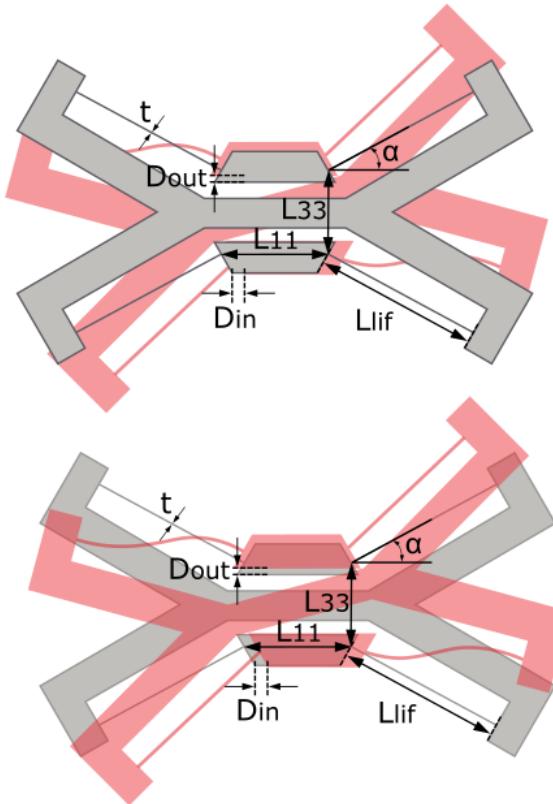
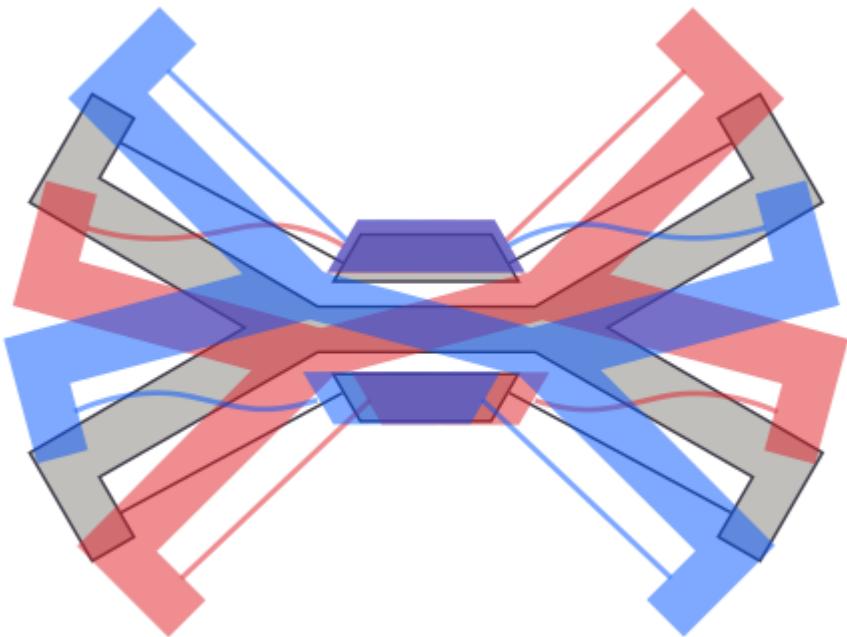
# Material properties

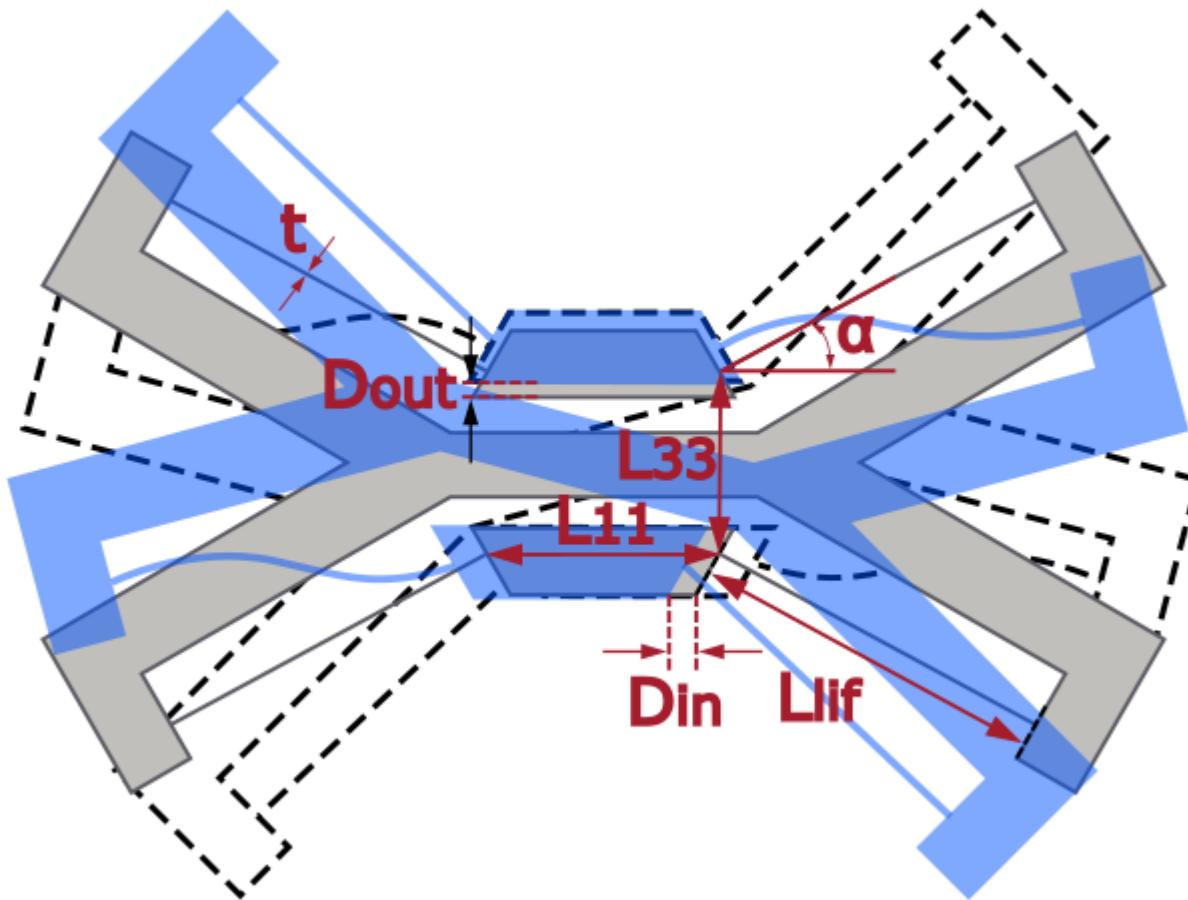
- Poisson's ratio = 0.4
- Yield strength = 7 MPa
- $E = 95 \text{ MPa}$ , only until  
~7-8MPa



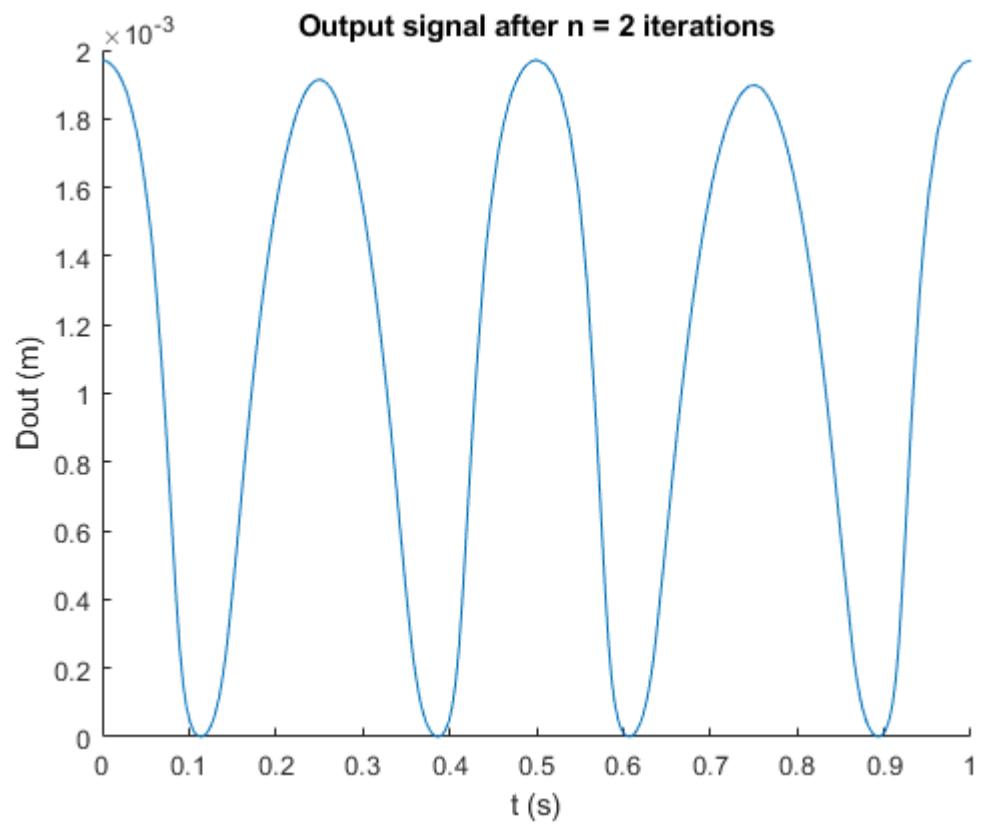
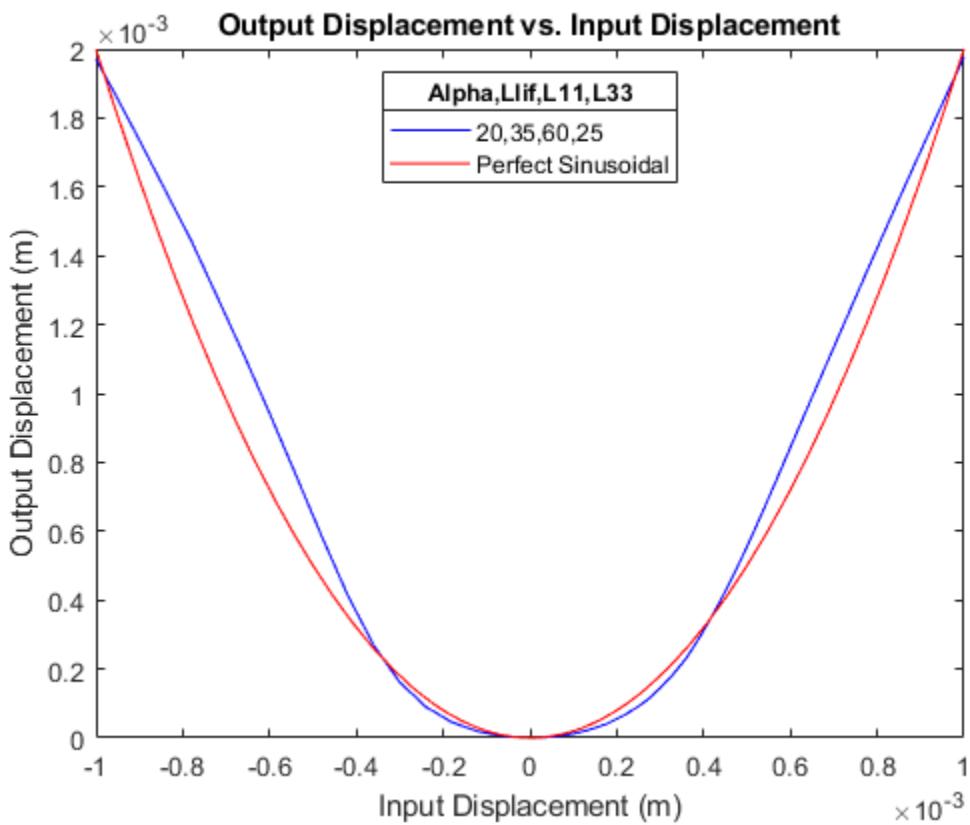
Zhang, Y., Tichem, M., & van Keulen, F. (2022). Concept and design of a metastructure-based multi-stable surface. *Extreme Mechanics Letters*, 51, 101553.  
<https://doi.org/10.1016/j.eml.2021.101553>

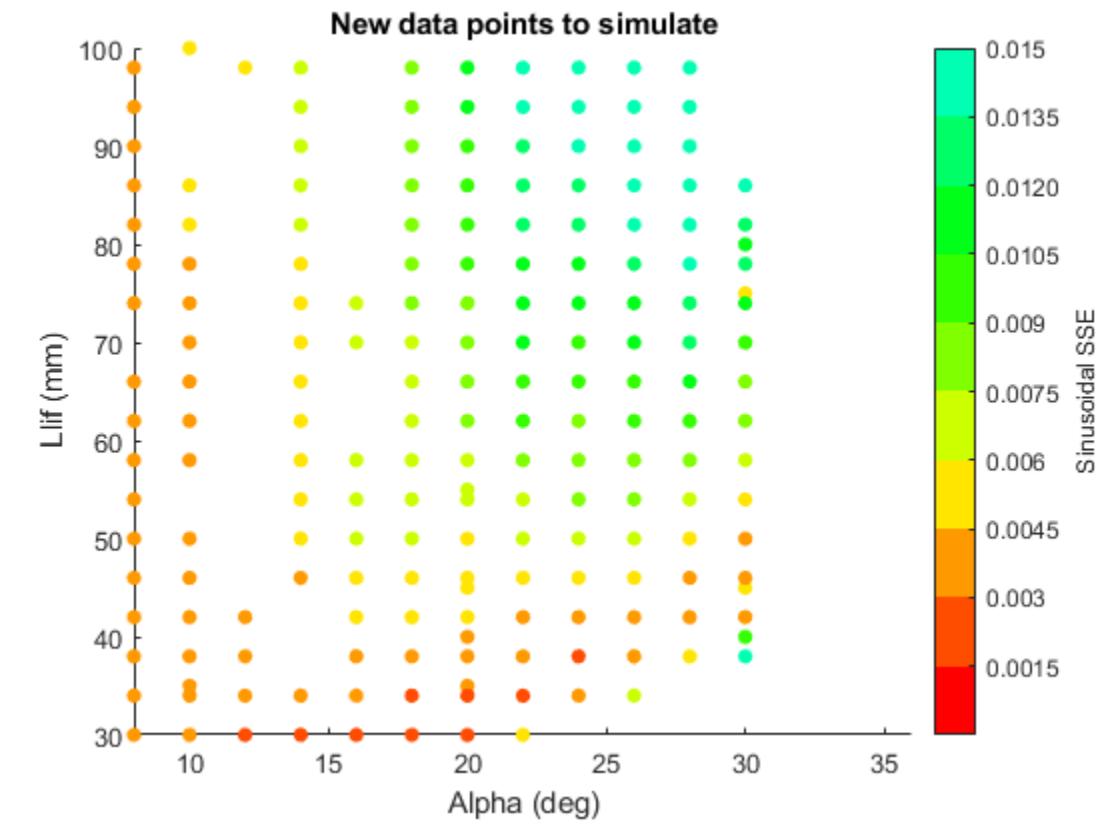
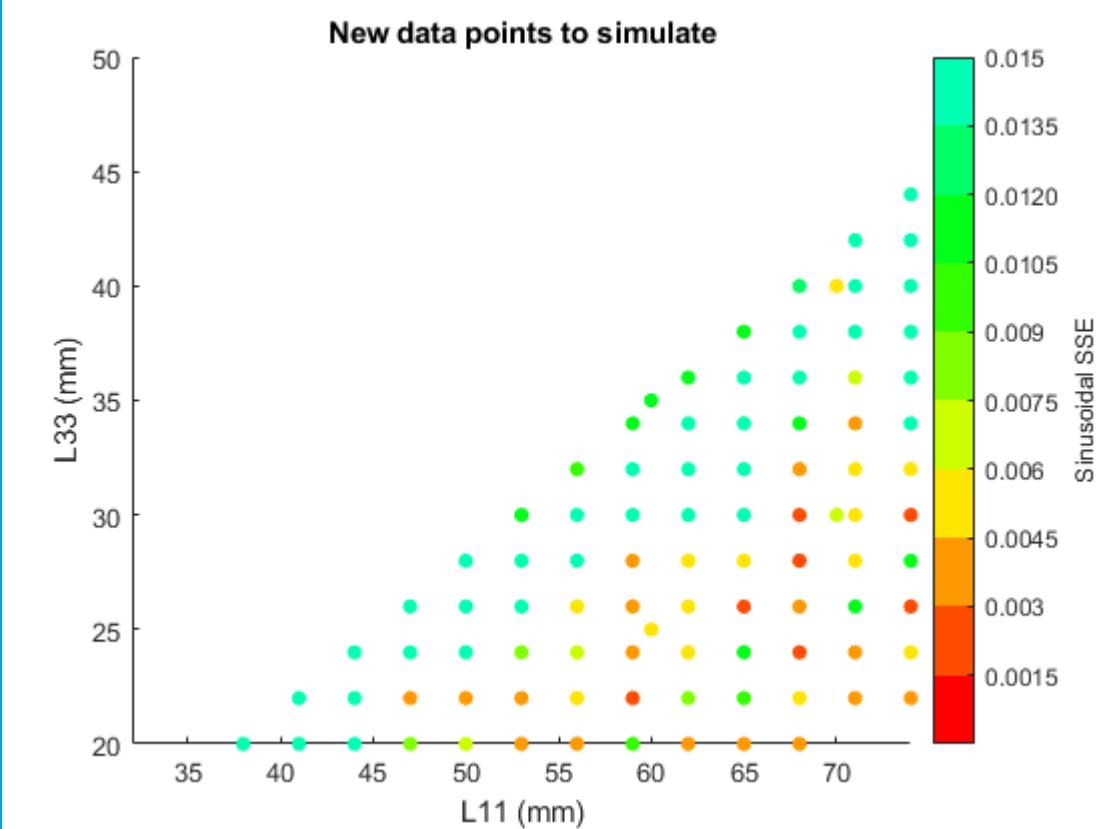
# Figures



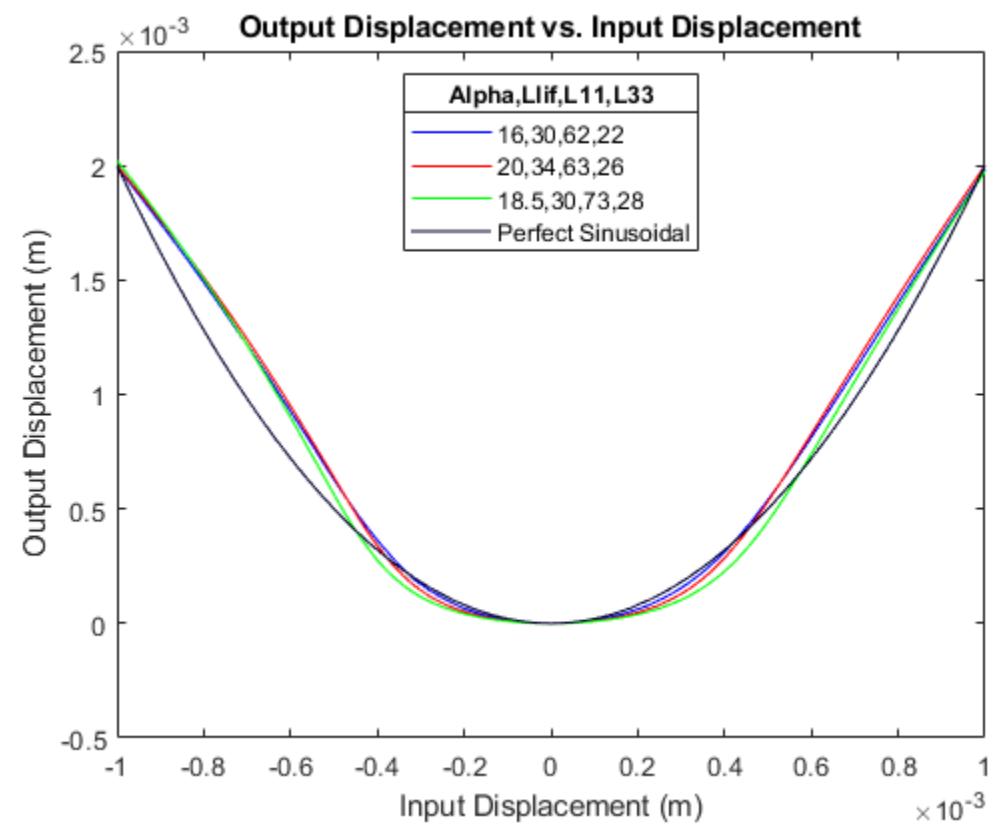
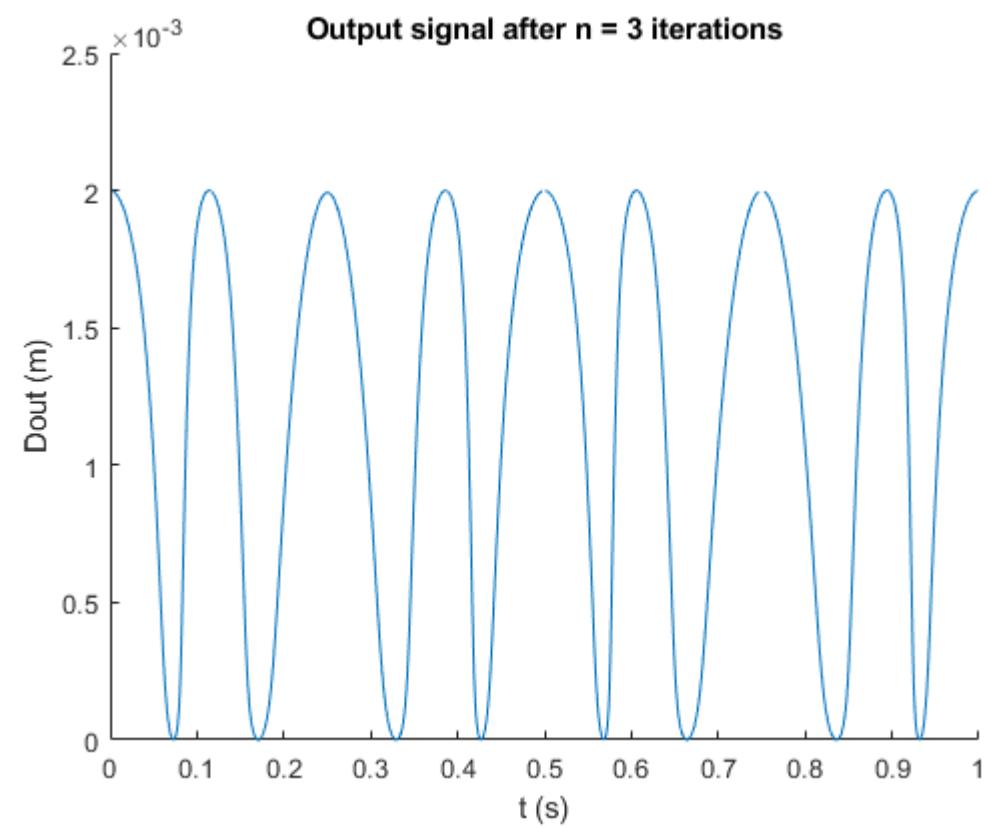


# After full investigation

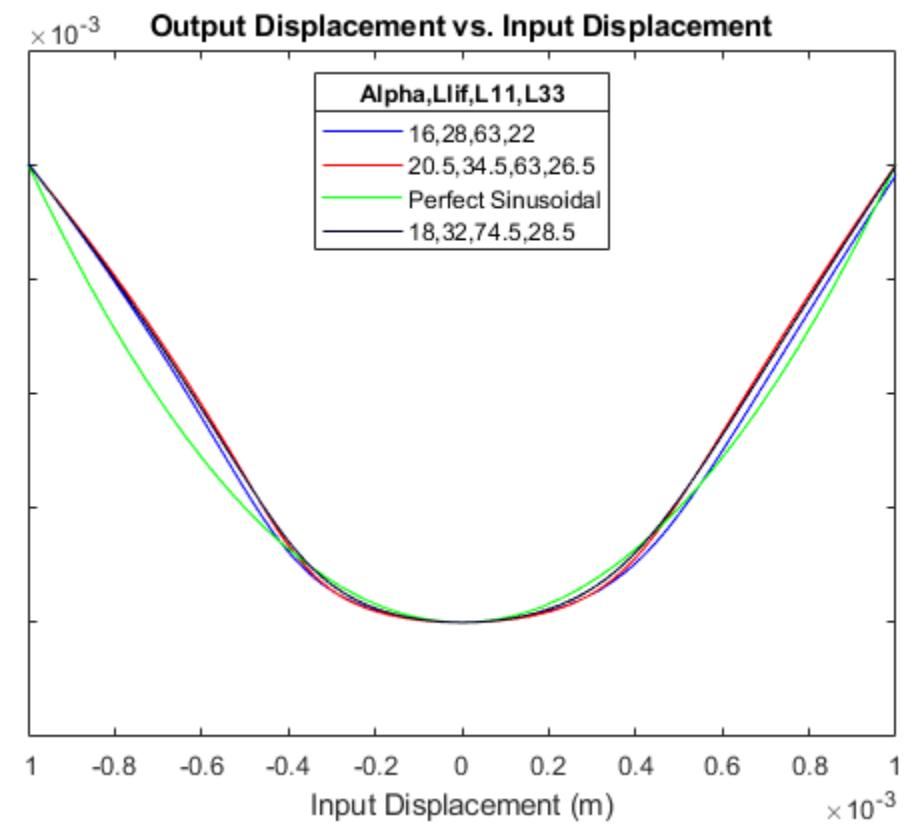
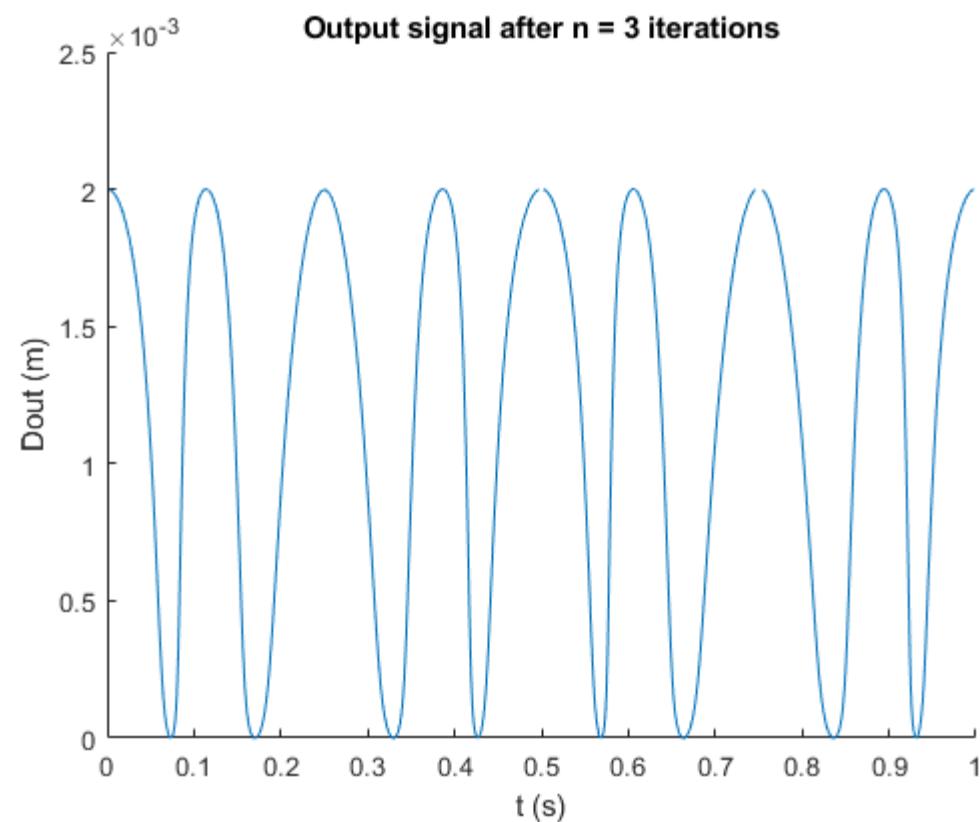




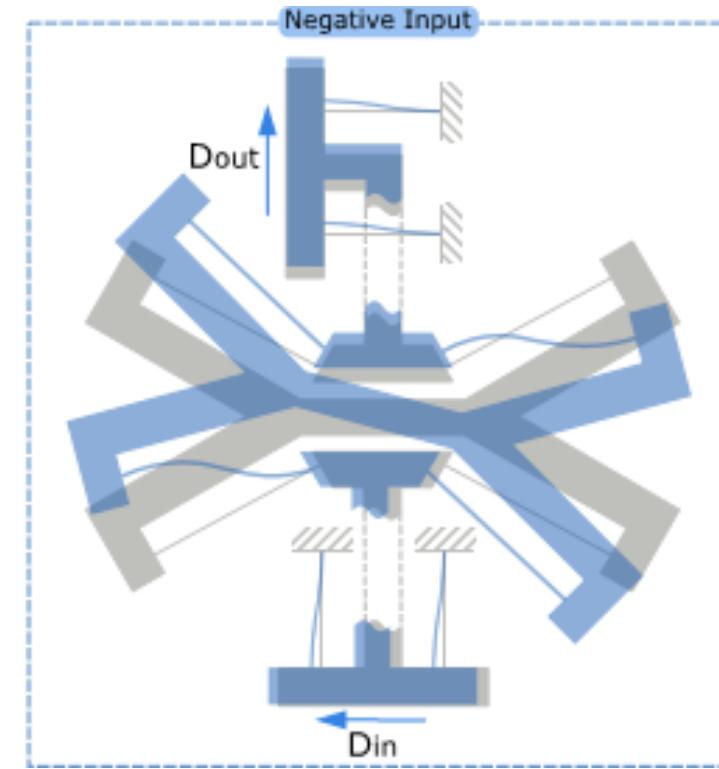
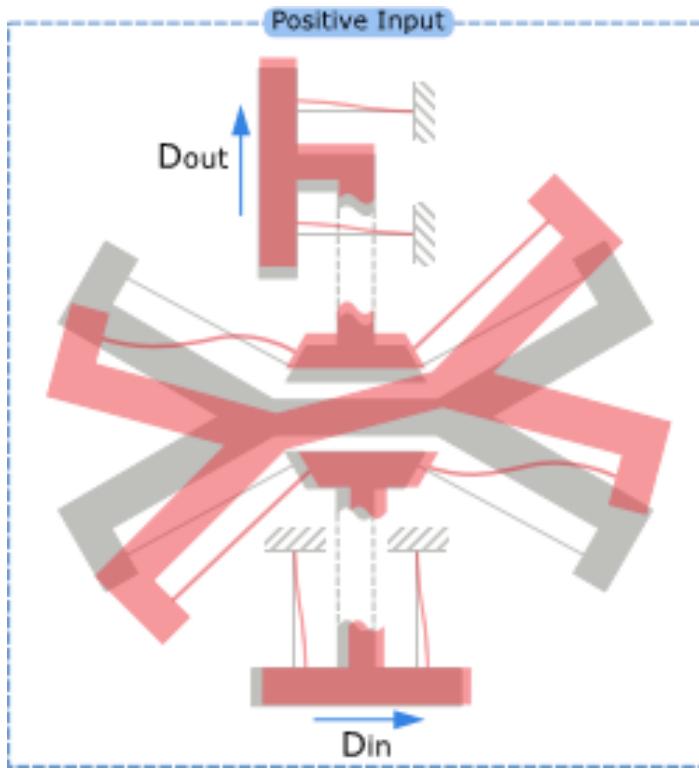
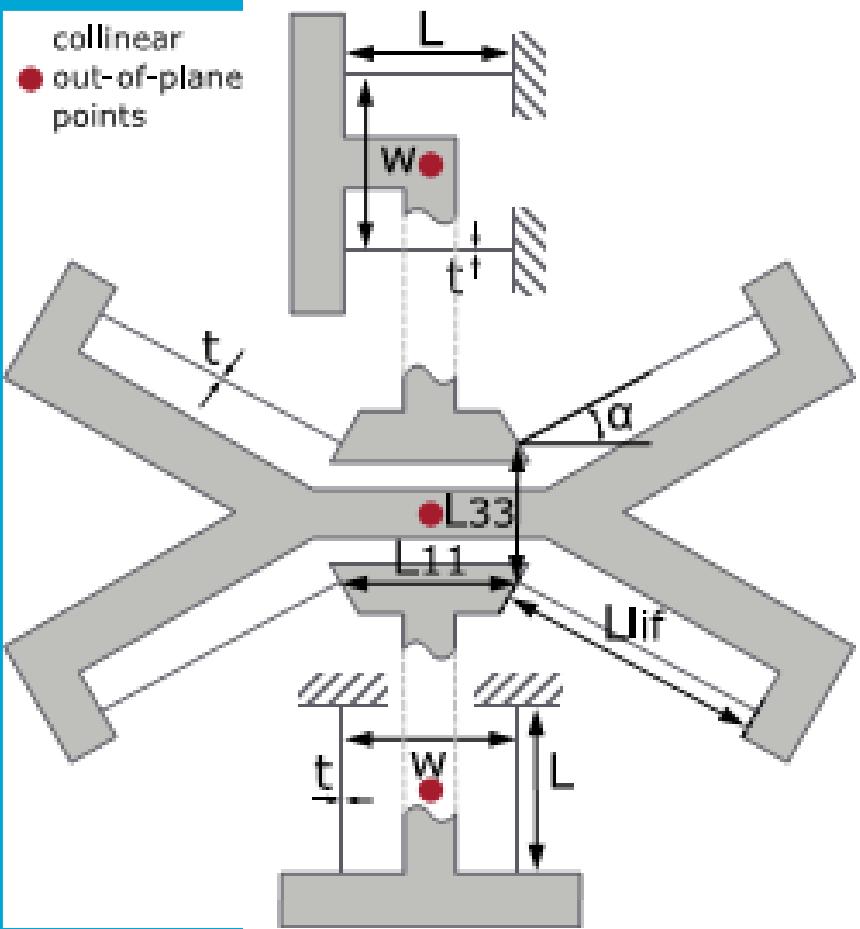
# Iteration 1



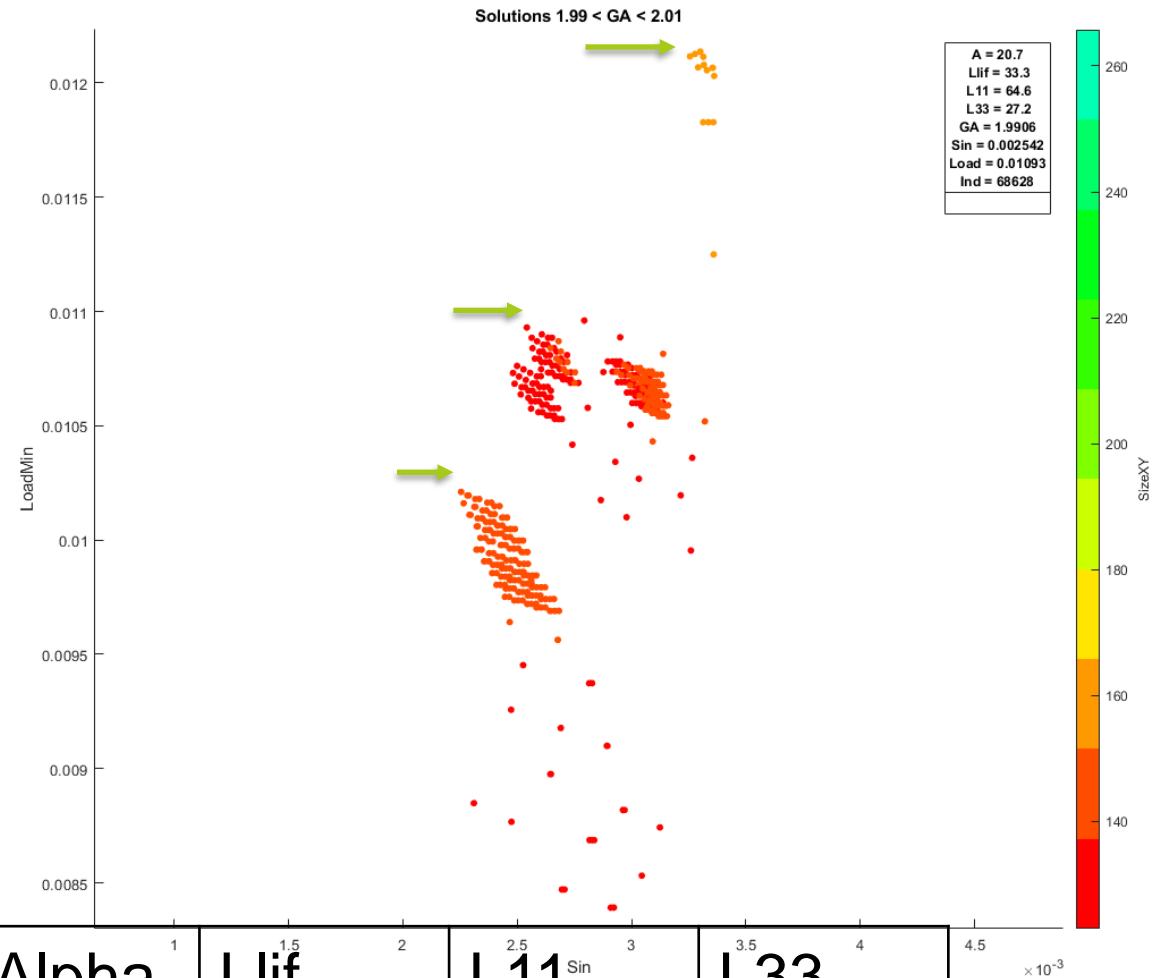
# Iteration 2



# Figures



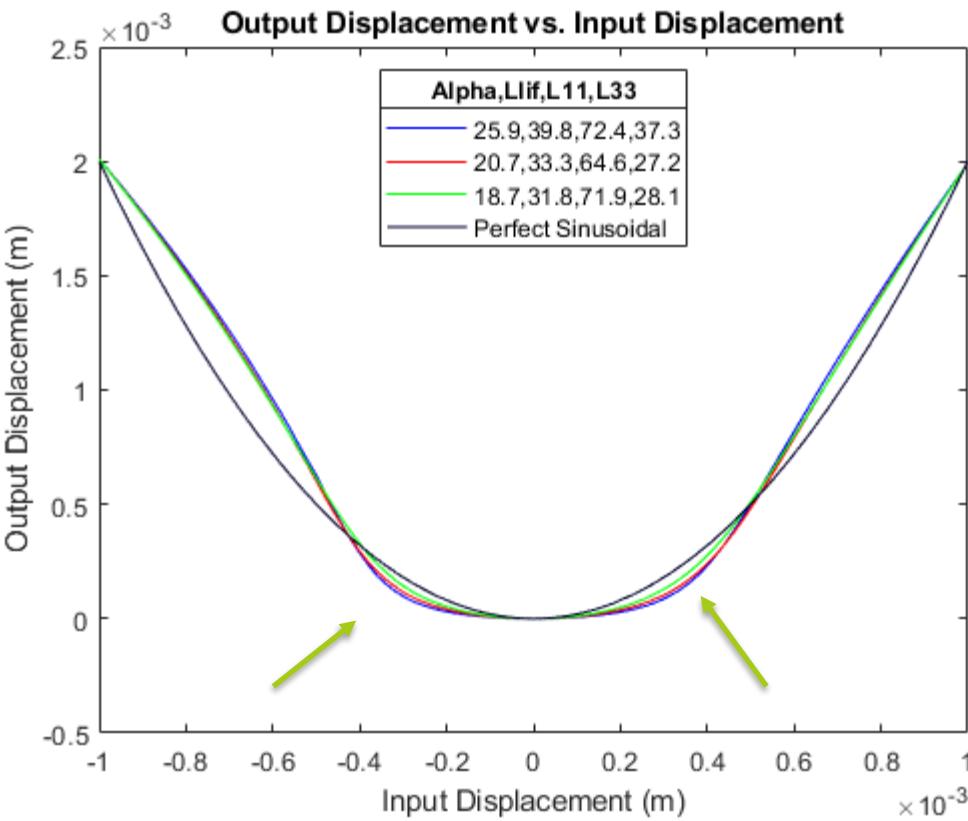
# After 4 iterations



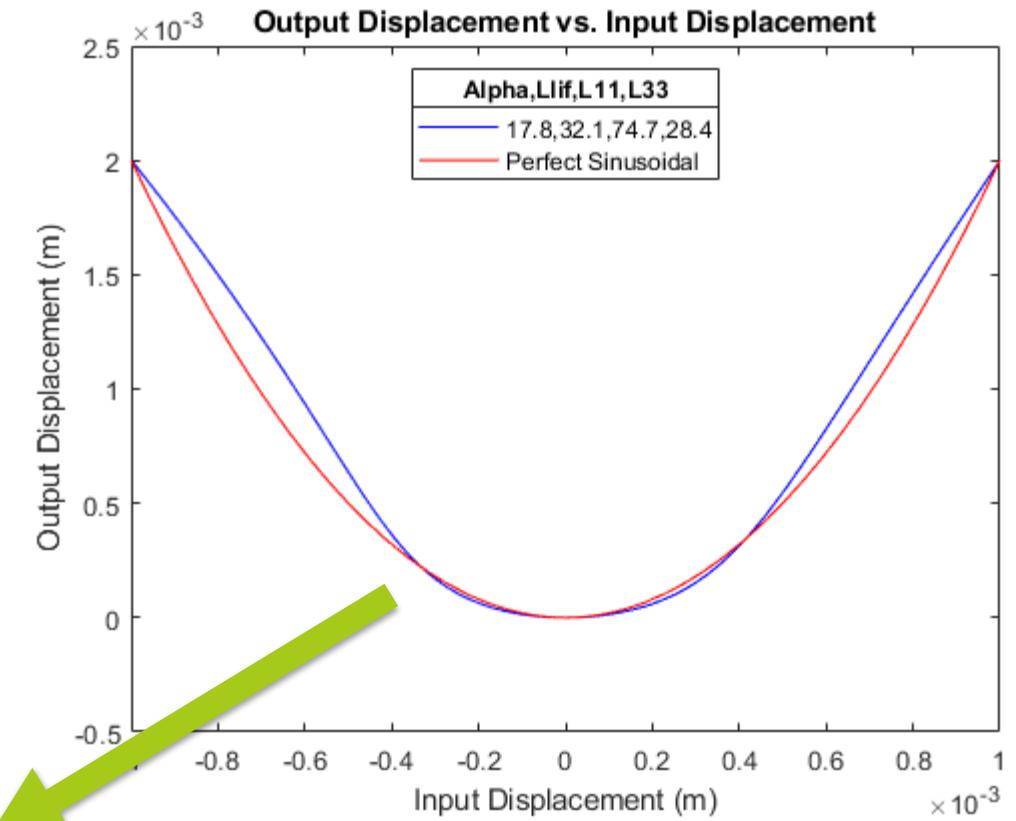
GA	Sin	Load	Alpha	$L_{lif}$	$L_{11}$	$Sin$	$L_{33}$
2.0095	0.00325	0.0121	25.9	39.8	72.4	37.3	37.3
1.9906	0.00254	0.0109	20.7	33.3	64.6	27.2	27.2
2.0098	0.00225	0.0102	18.7	31.8	71.9	28.1	28.1

# DD

New with load as criteria

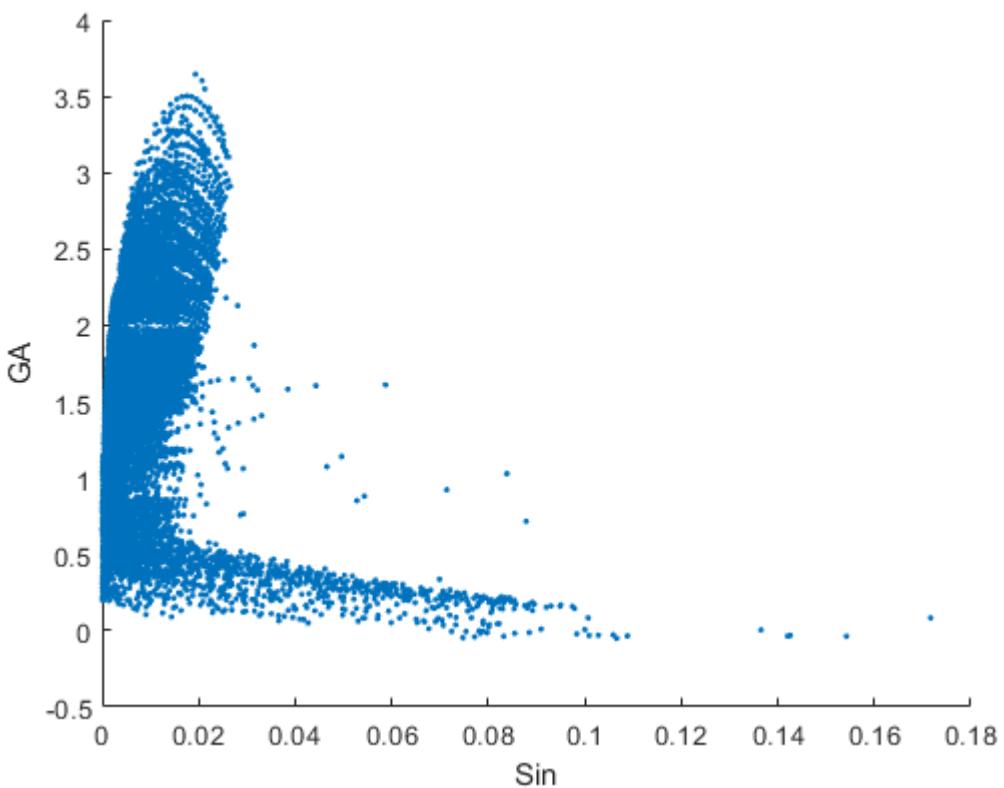


Old just looking at GA and Sin



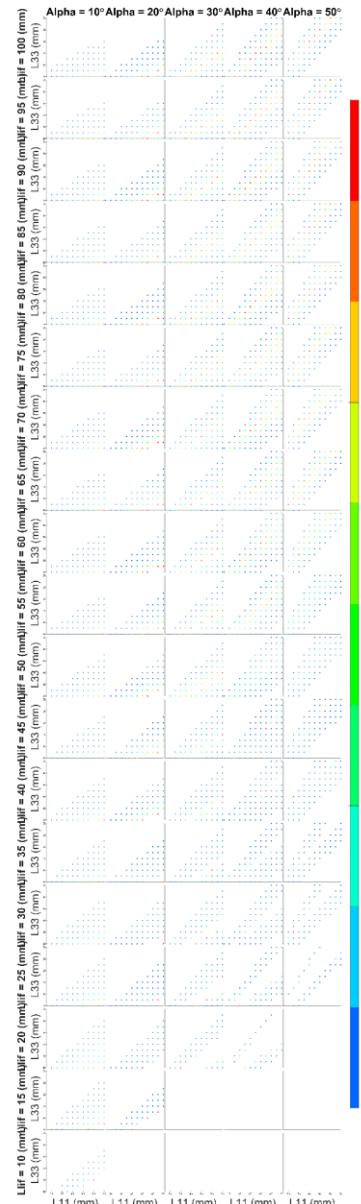
GA	Sin	Load	Alpha	Llif	L11	L33
1.9989	0.0026	0.0097	17.8	32.1	74.7	28.4

# 68733 points analysed

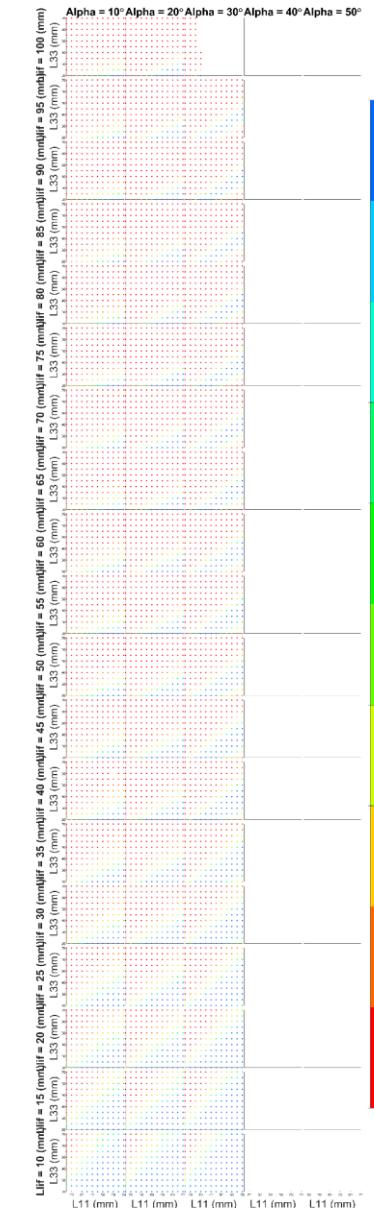


# Tile plots

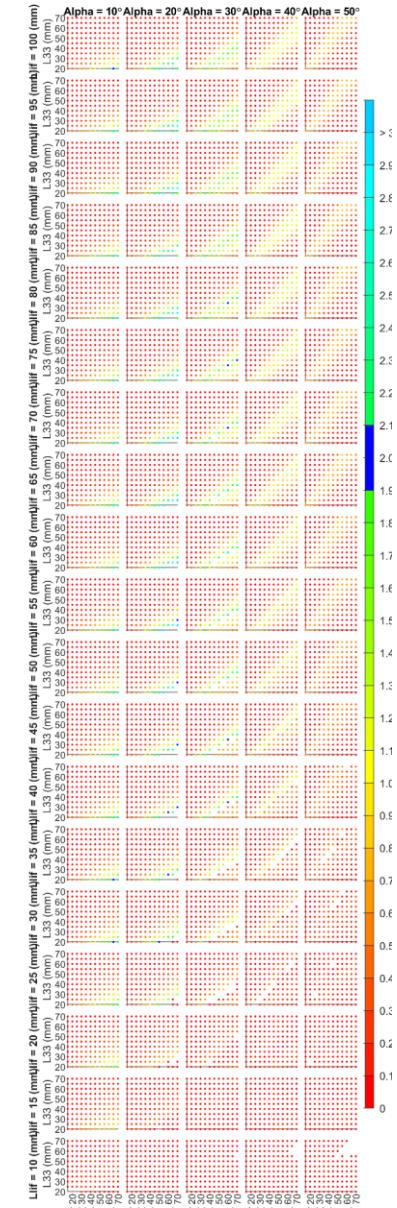
Sinusoidal sum of square error for each design



Force capacity for each design



GA for each design at the end of the input



# Results and discussion

1. Introduce mechanism and capabilities [Schematic of design with parameters + design with deformation](#)
2. Showing how different parameters alter the results qualitatively and quantitatively [Displacement relation](#)
3. What behaviour is desired? Setup of criteria..
4. First investigation to decrease design space [Parameter plots of GA, Sin, Force capacity](#)
5. Full investigation into decreased design space, and iterate while: [Resulting criteria plot](#)
  1. Narrowing search space
  2. Tightening constraints
6. Validate the results with tested prototype [DD and FD graph with FEM and Test](#)
7. Mention best result

# Plots

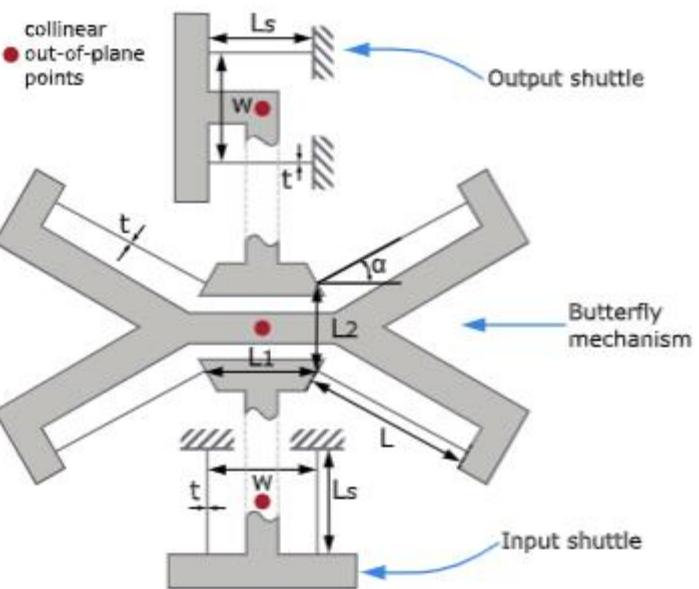


Figure 1: Schematic of the eight-bar mechanism based frequency doubler building block

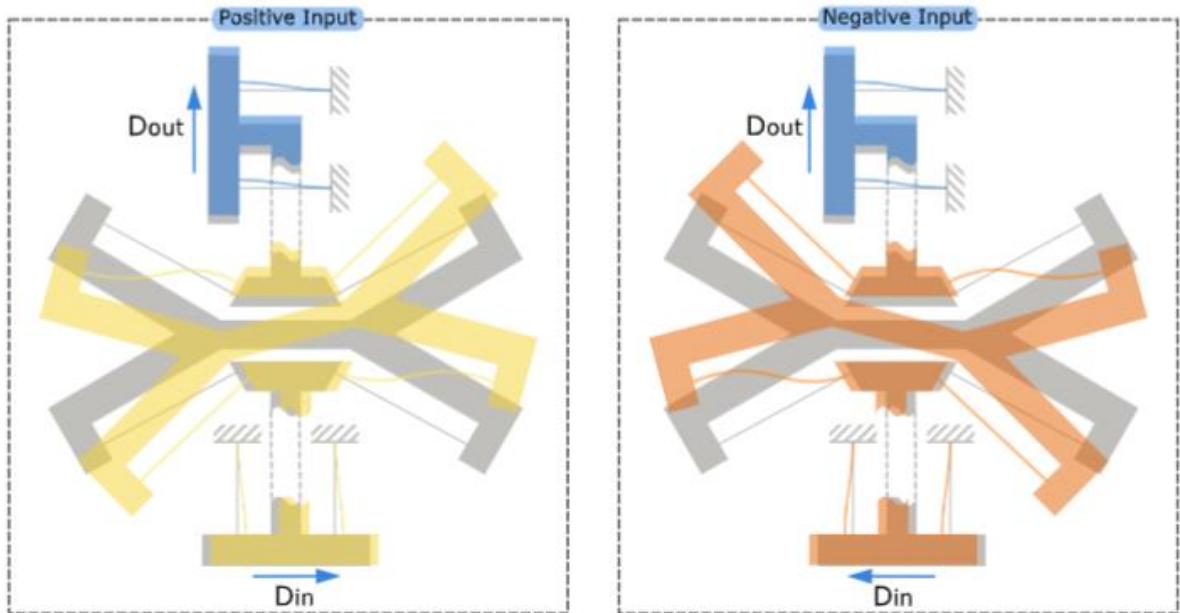
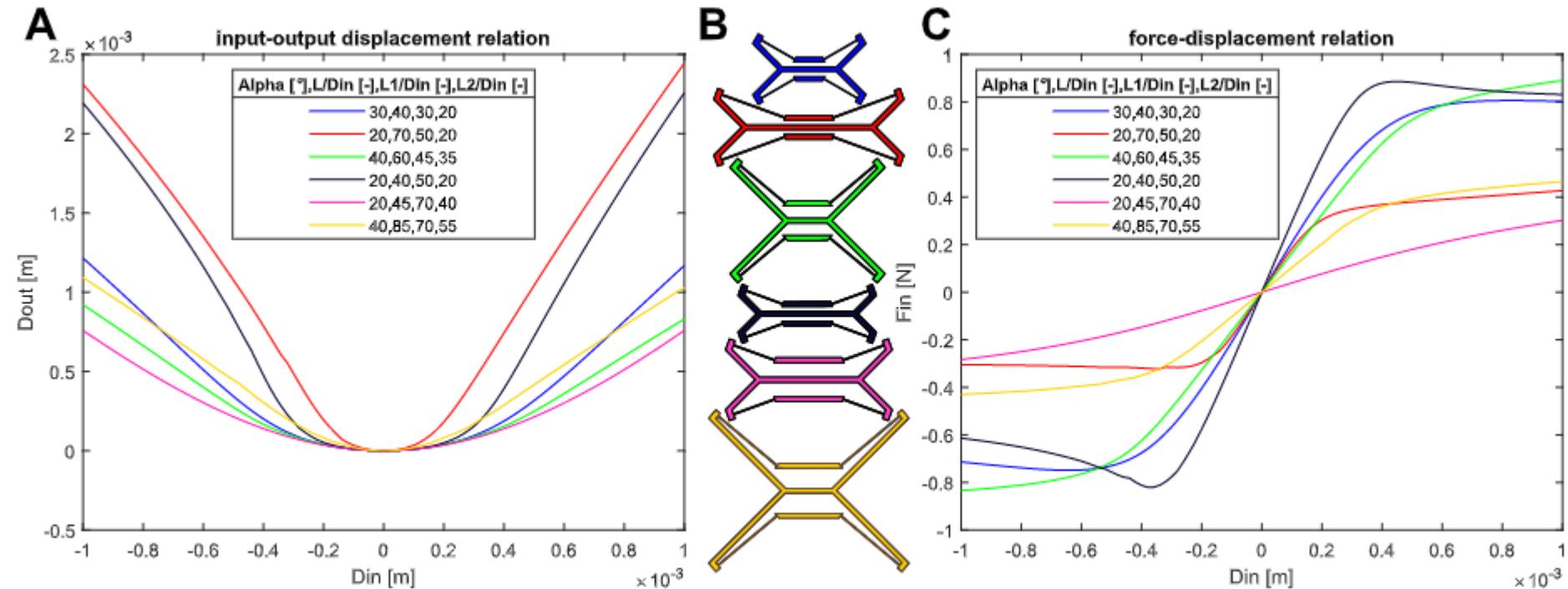
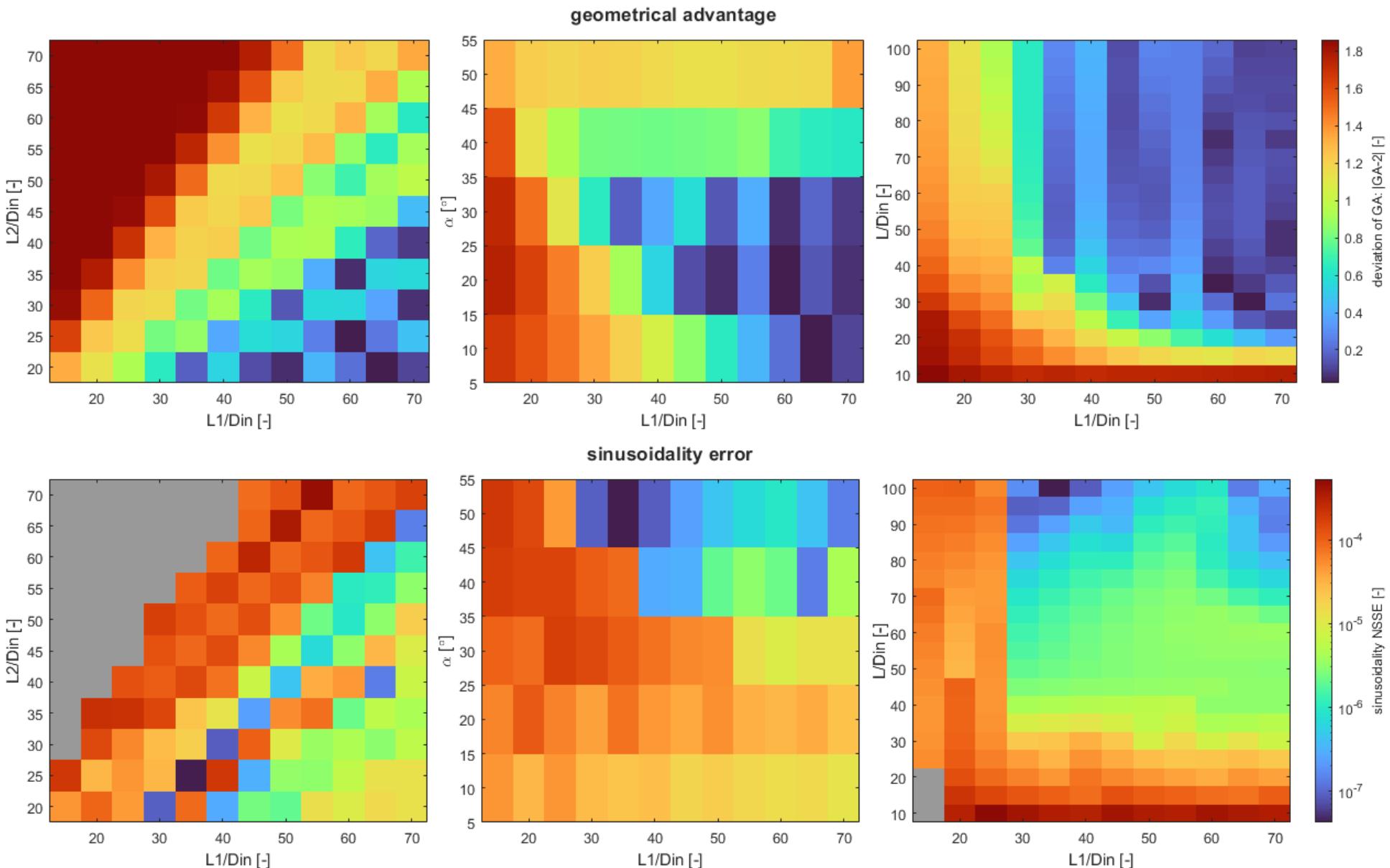


Figure 3: Input-output displacement characteristics for a positive input displacement (left) and a negative input displacement (right).

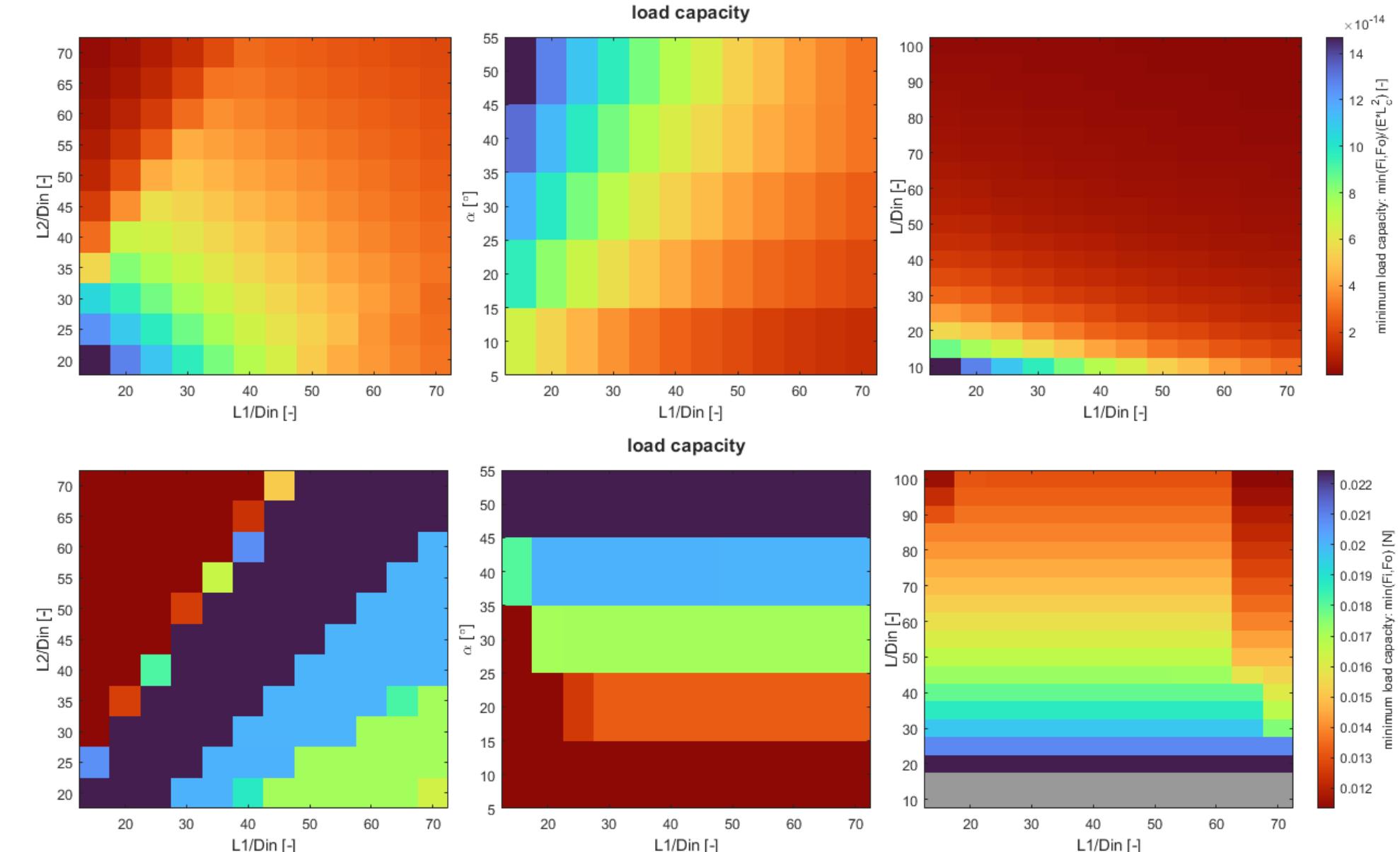
# Frequency doubling capabilities and changing parameters



# Plots



# Dimensionless Force? $F/(E^*[m^2])$



# Criteria plot

Alpha: 26.1000

Llif: 40.2000

L11: 72.6000

L33: 37.7000

Alpha: 18.7000

Llif: 31.8000

L11: 71.9000

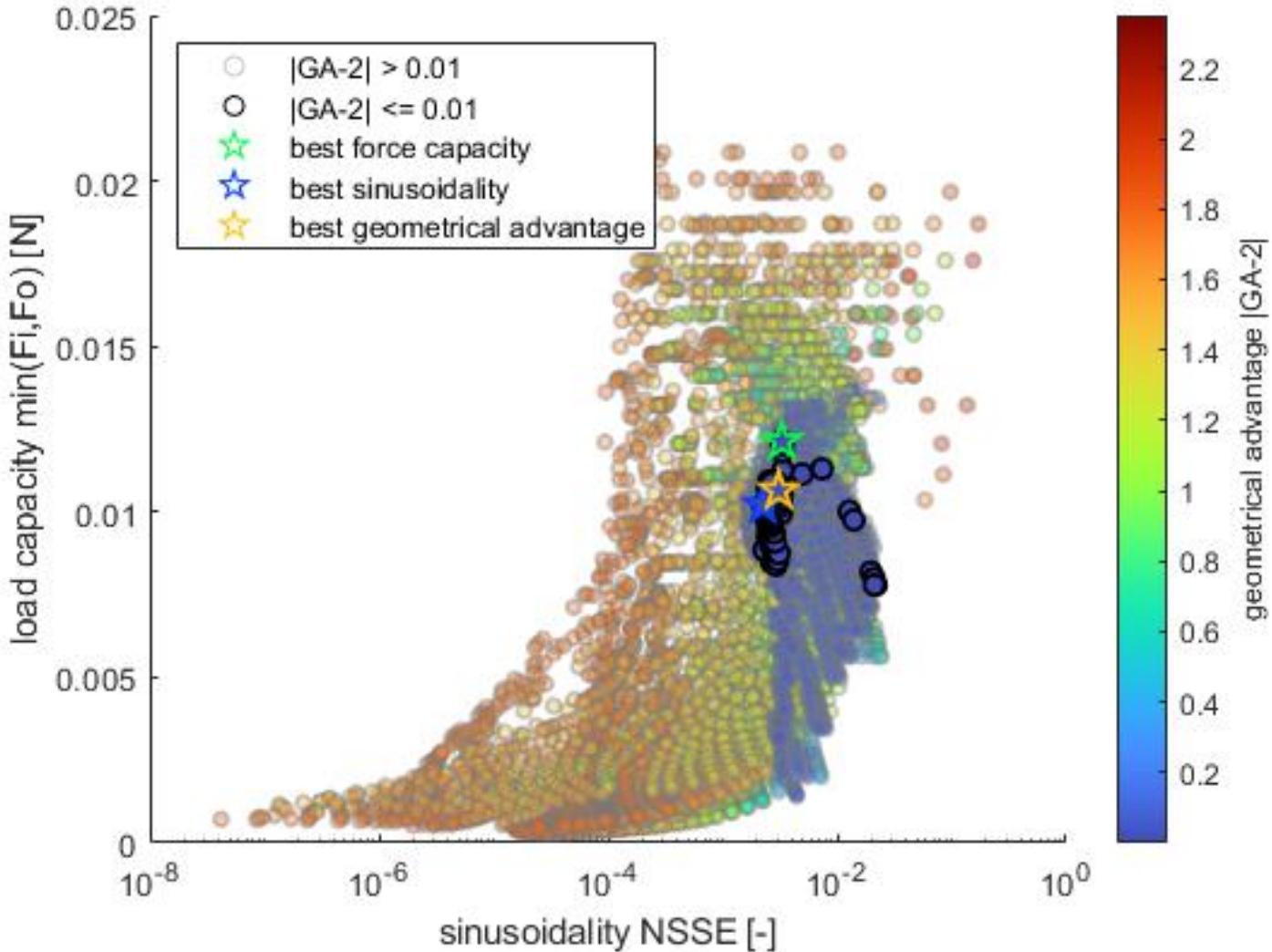
L33: 28.1000

Alpha: 20.5000

Llif: 34.6000

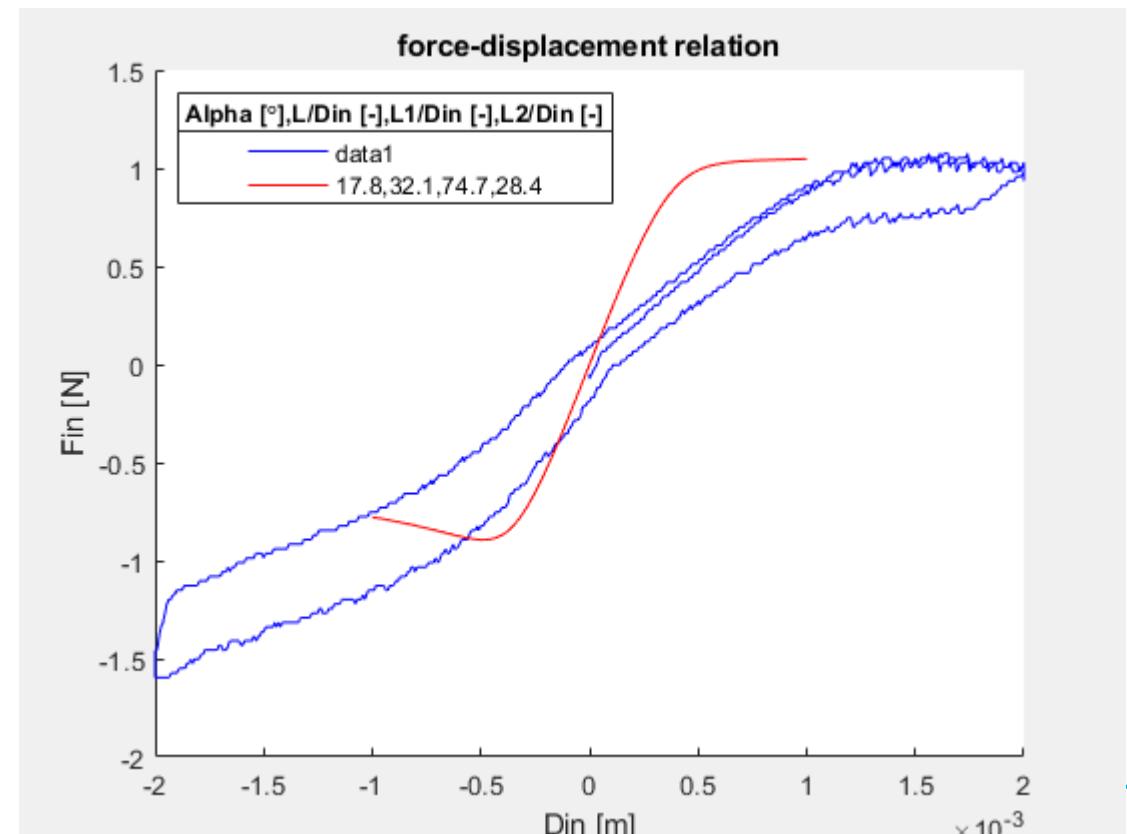
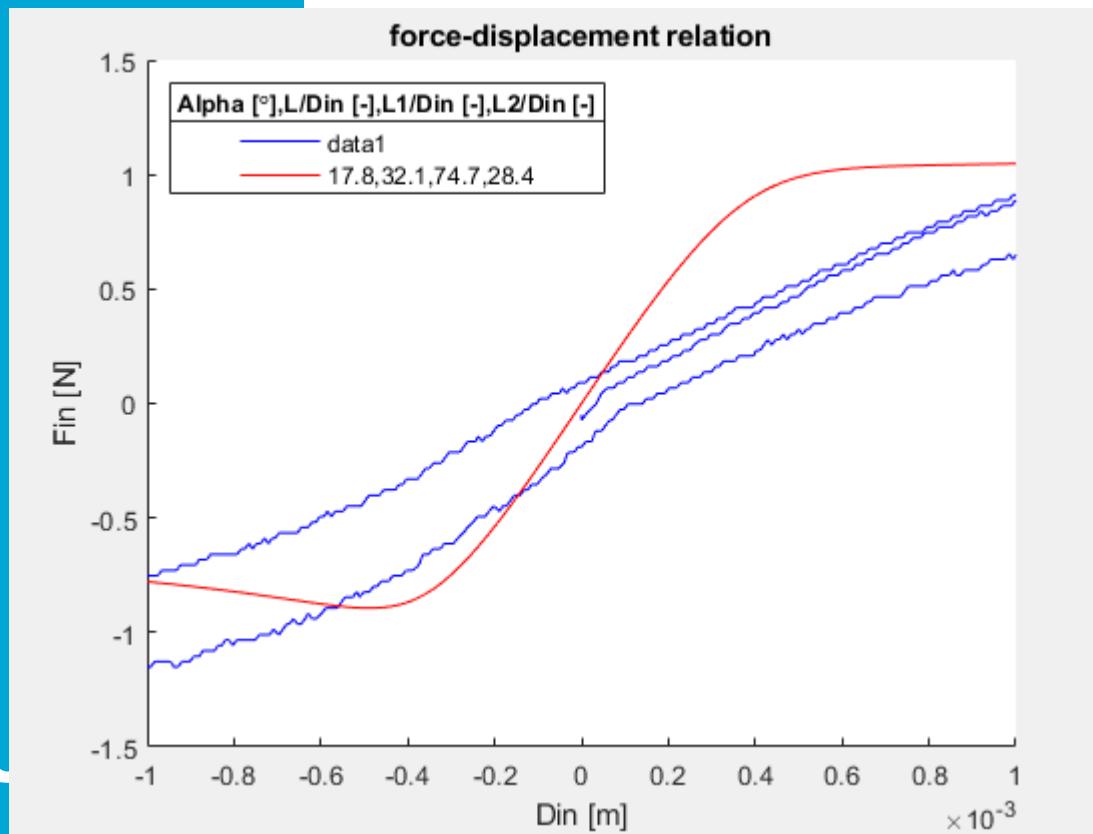
L11: 63.2000

L33: 26.6000



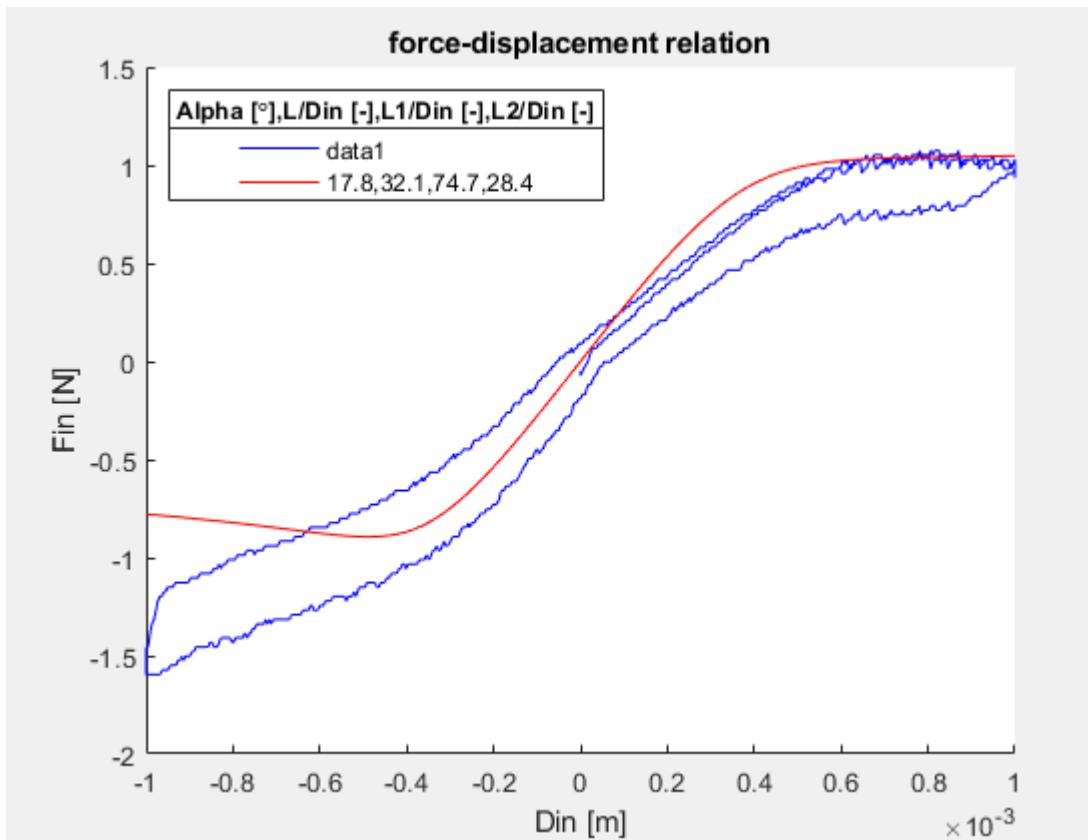
# Testing

- Hysteresis
- Backlash
- Inaccuracy of 3d printer
- Inaccuracy in assembly

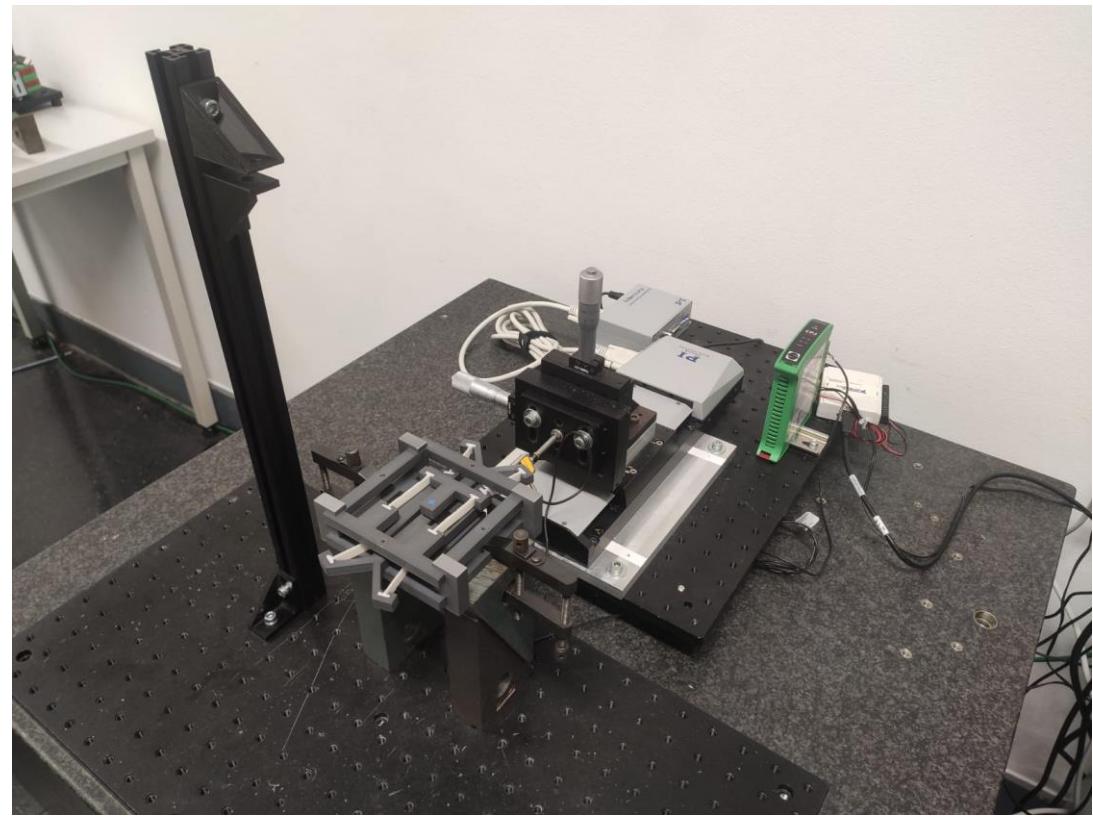
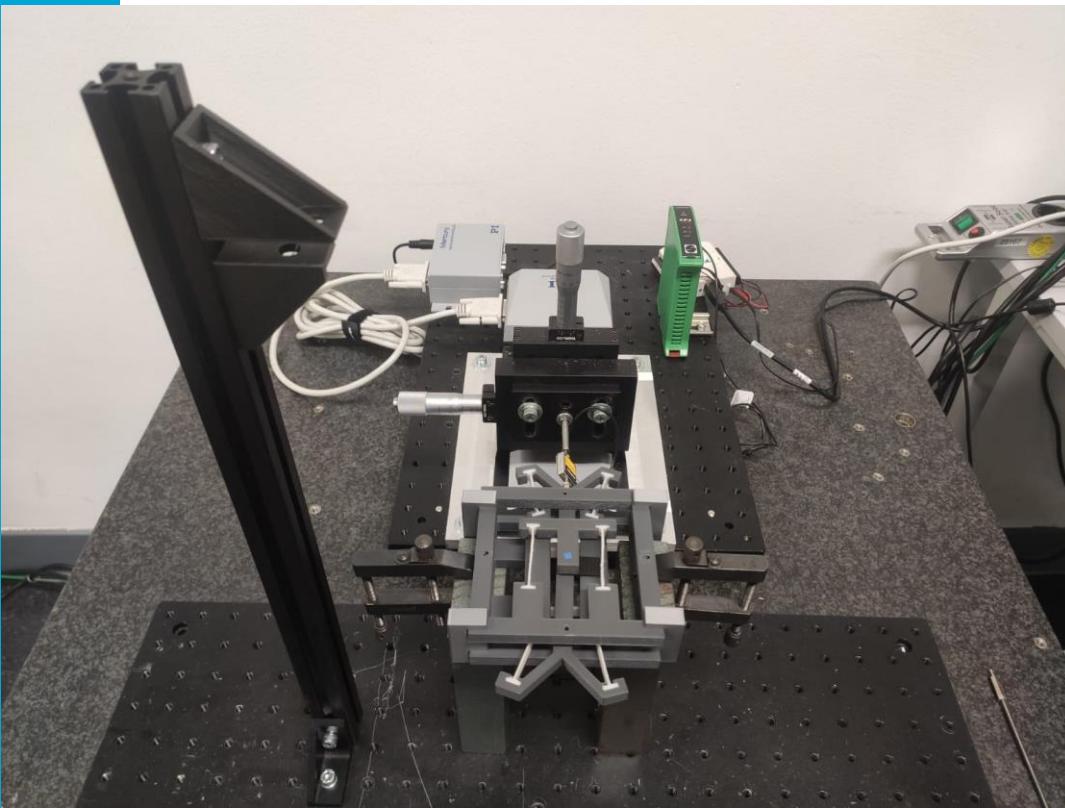


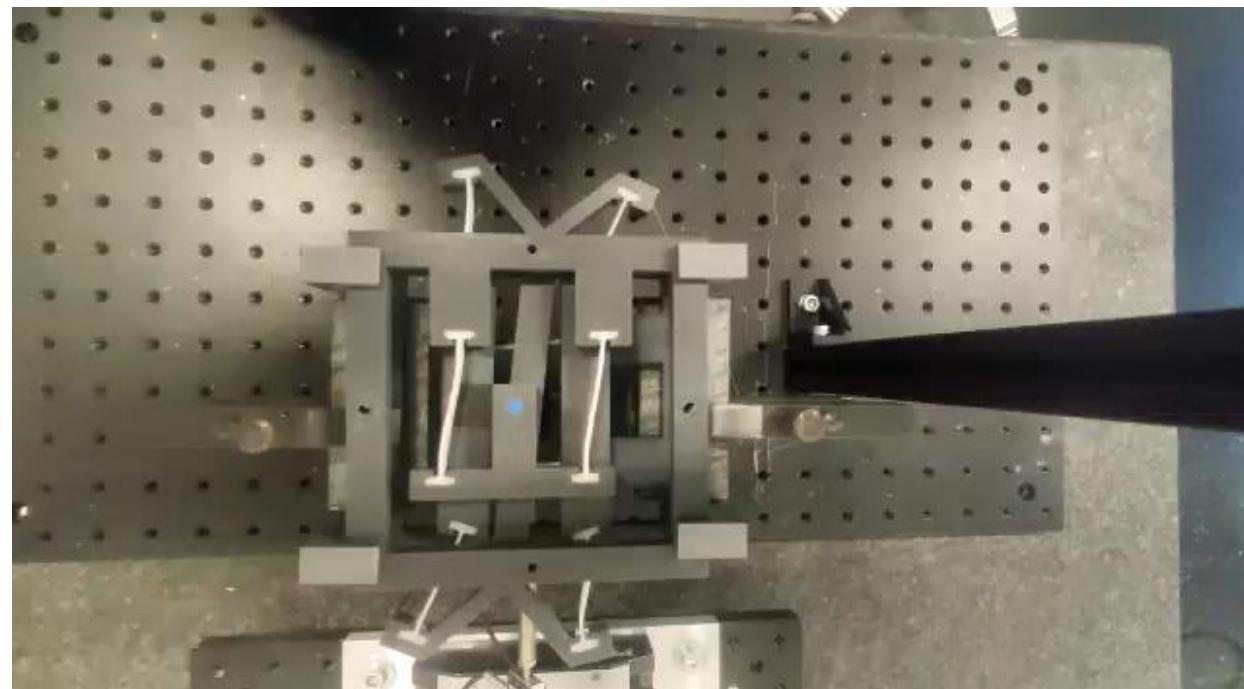
# Testing

- If input displacement/2

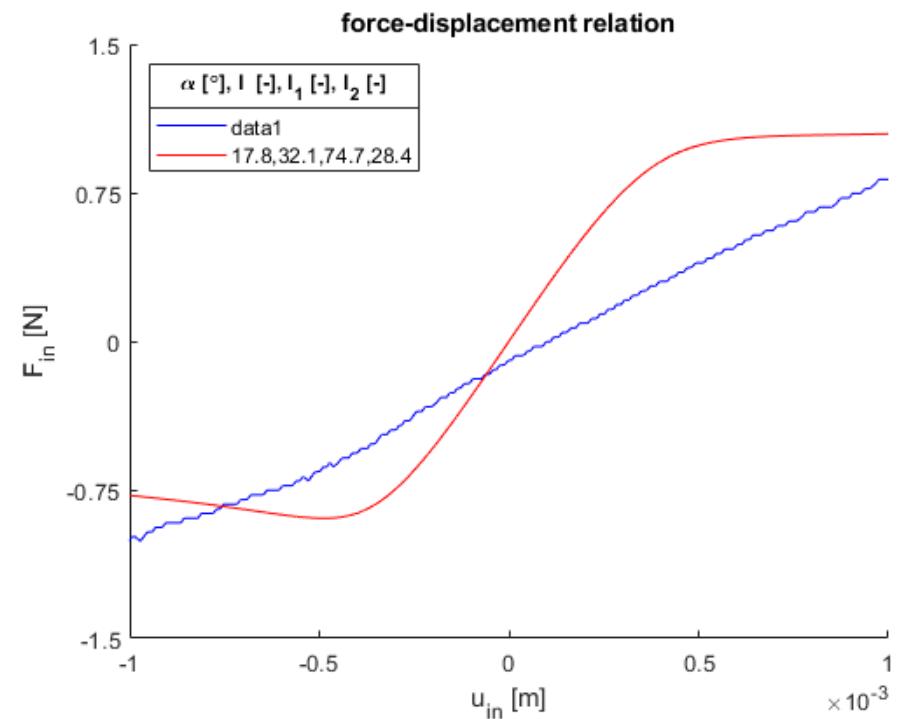
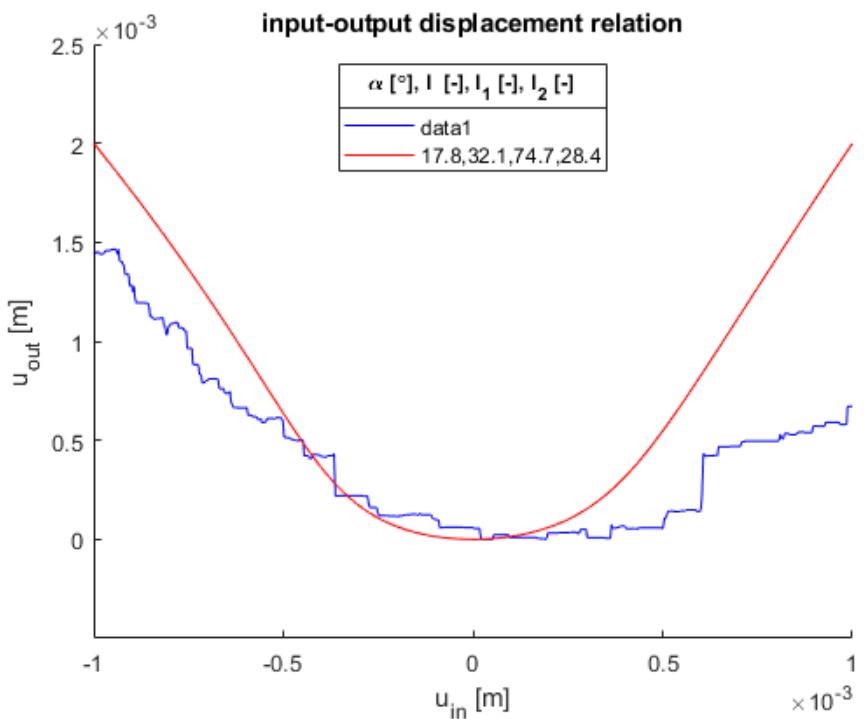


Mechanism with TPU flexures was tested, but did not work good enough

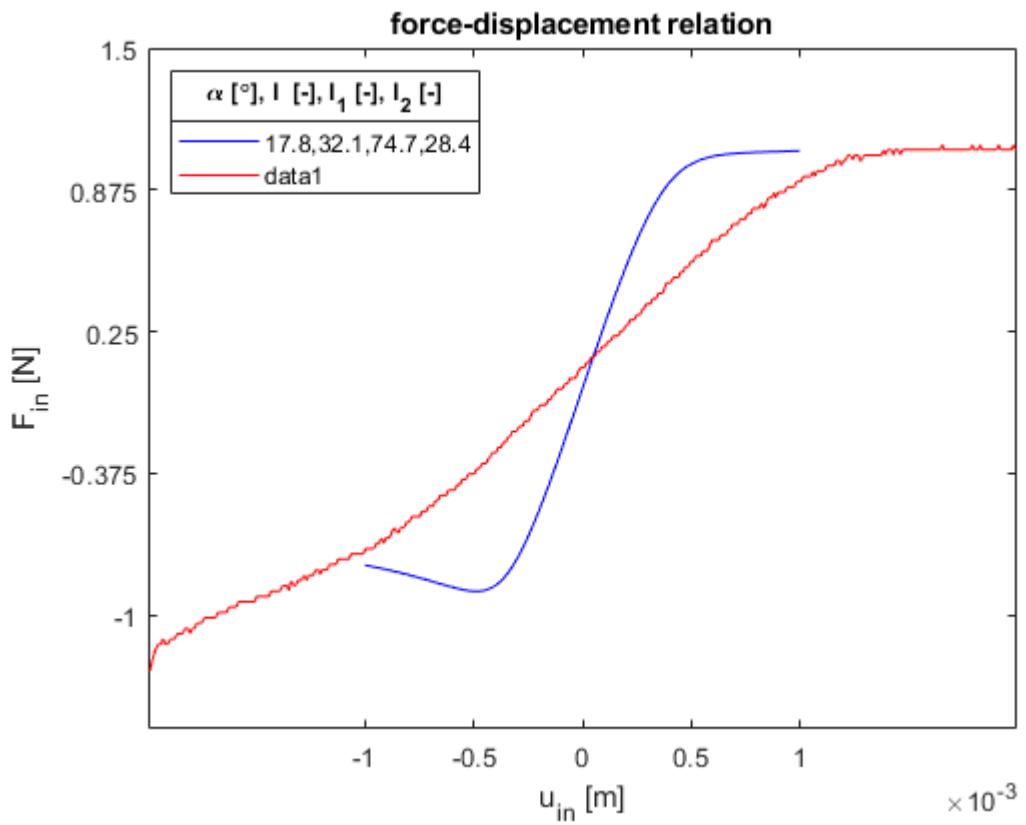




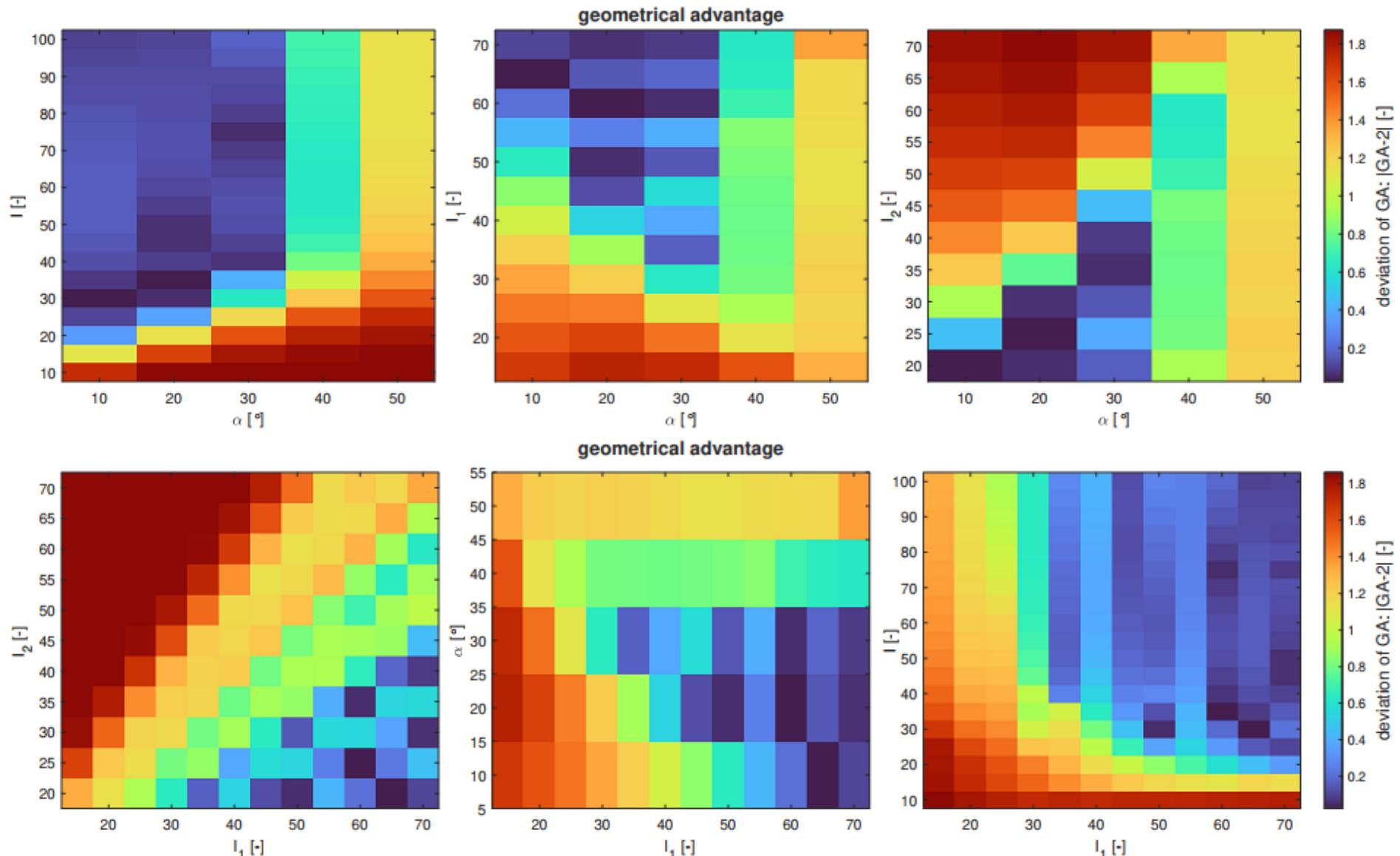
# Testresults with TPU

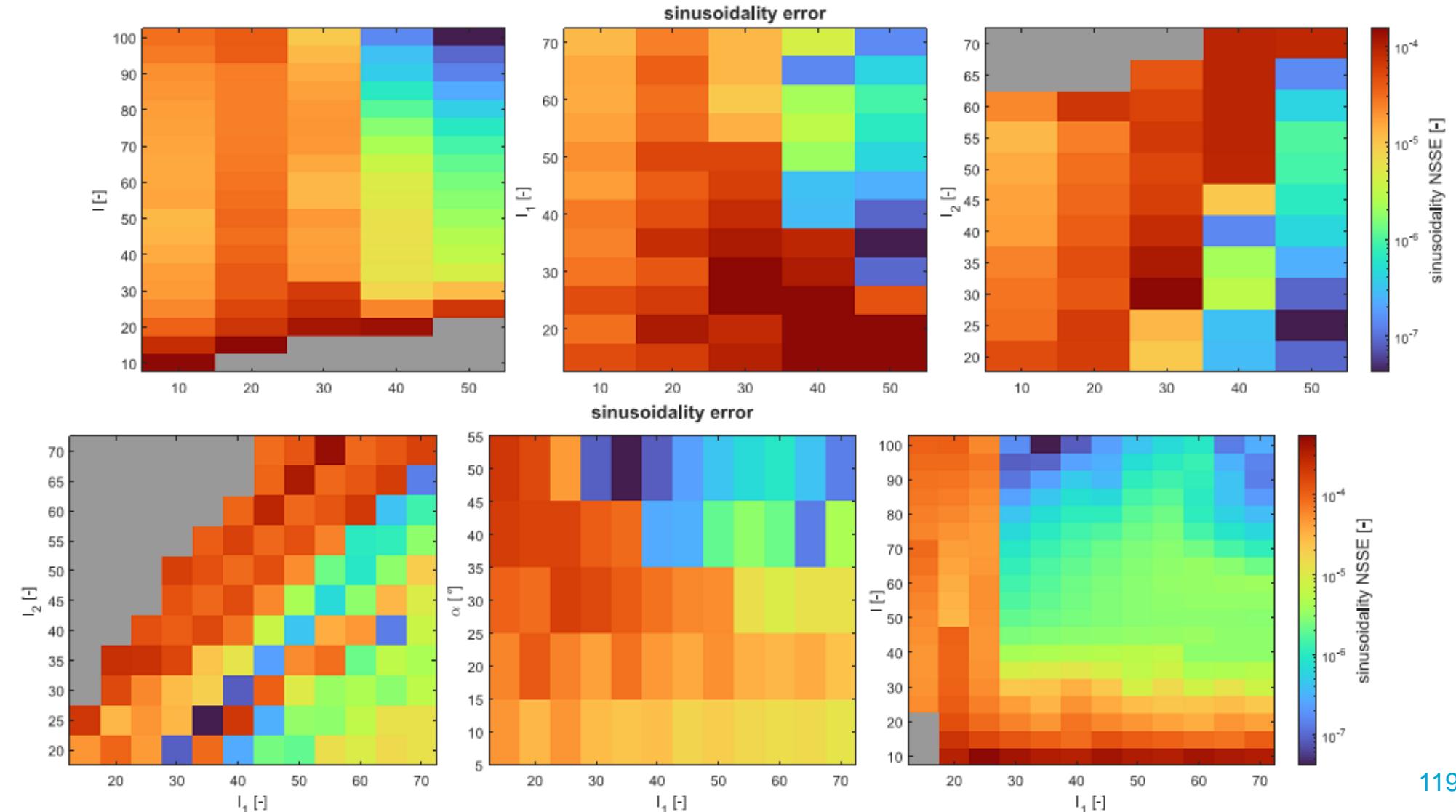


# Behavior for larger domain



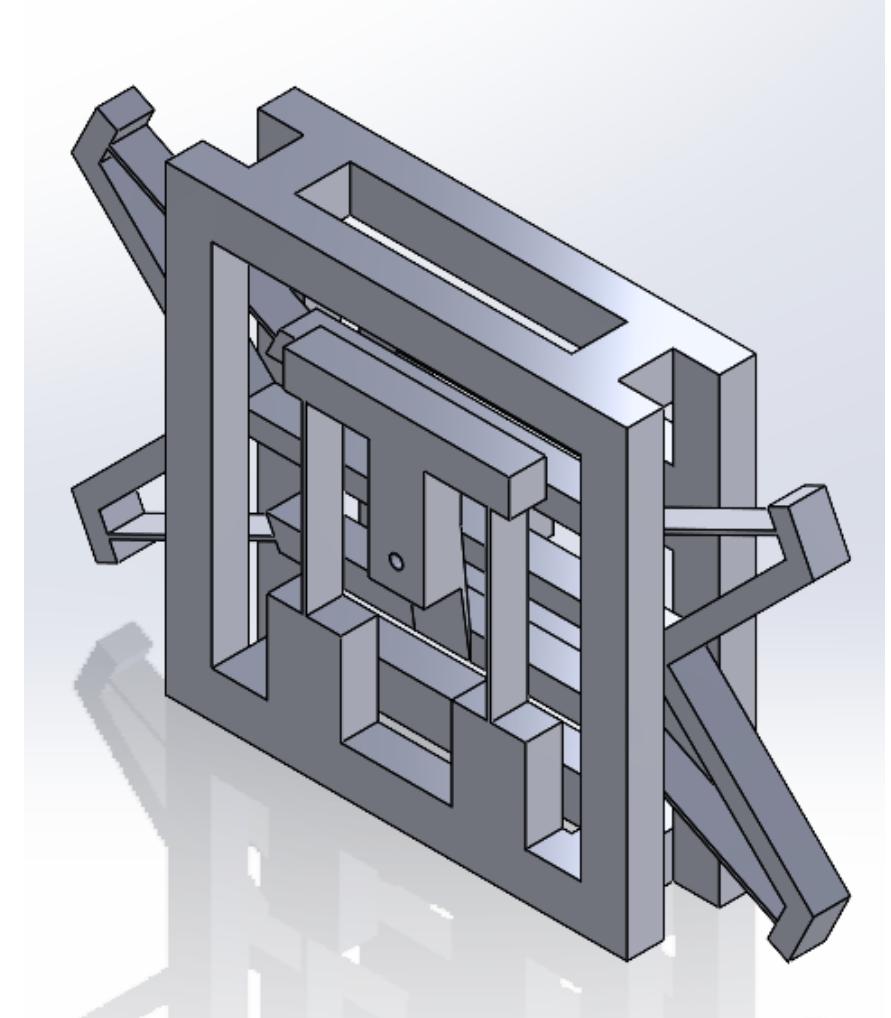
# Updated contour plots (Alpha vs $I_1$ )





# Problem with materialize

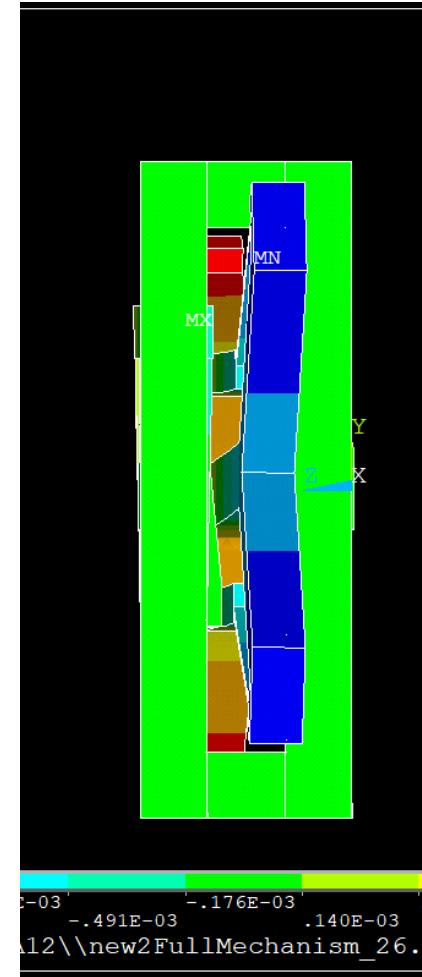
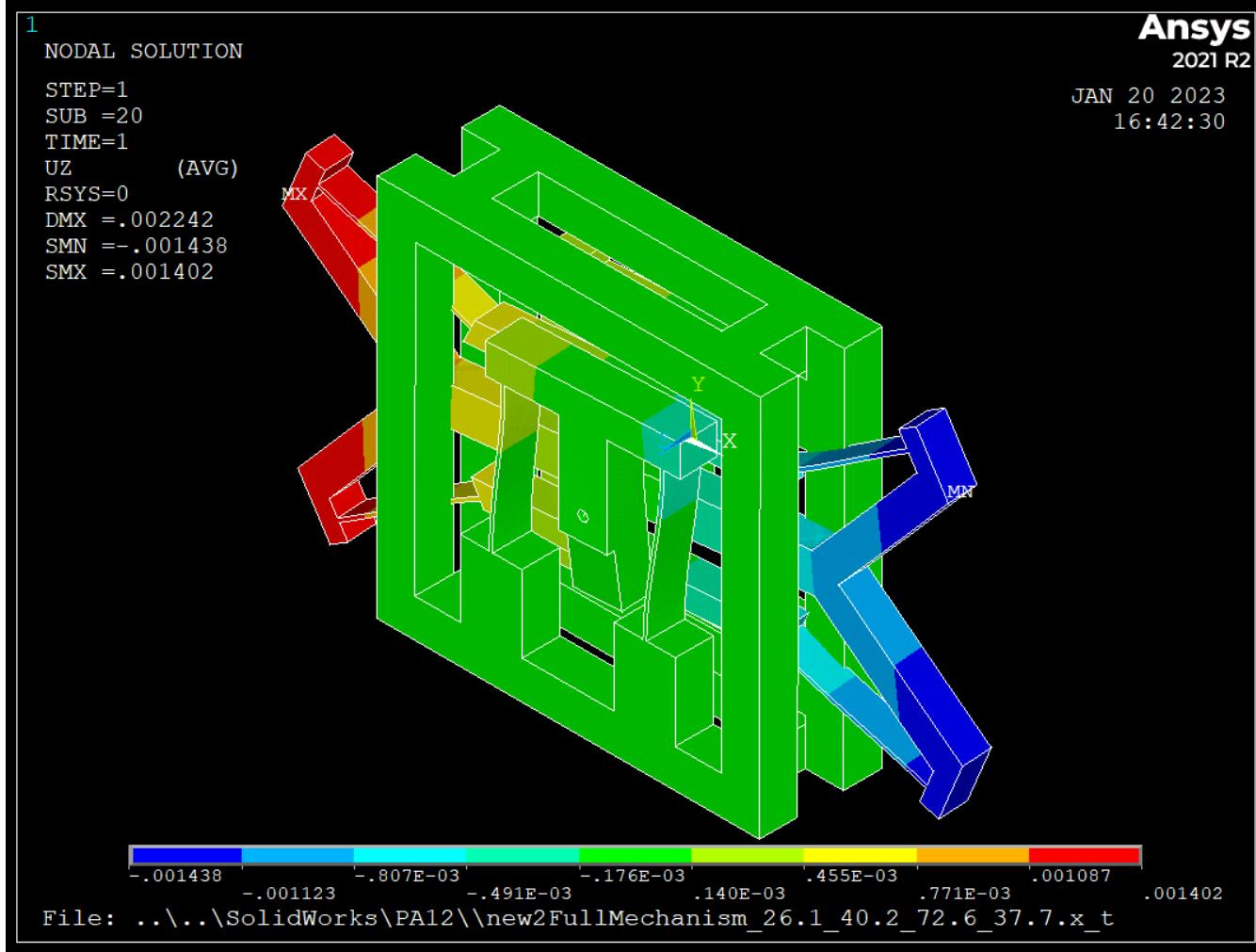
- Connection piece way too flexible
- Overall structure way too flexible
- Found out that using this part in FEM analysis, did not lead to frequency doubling



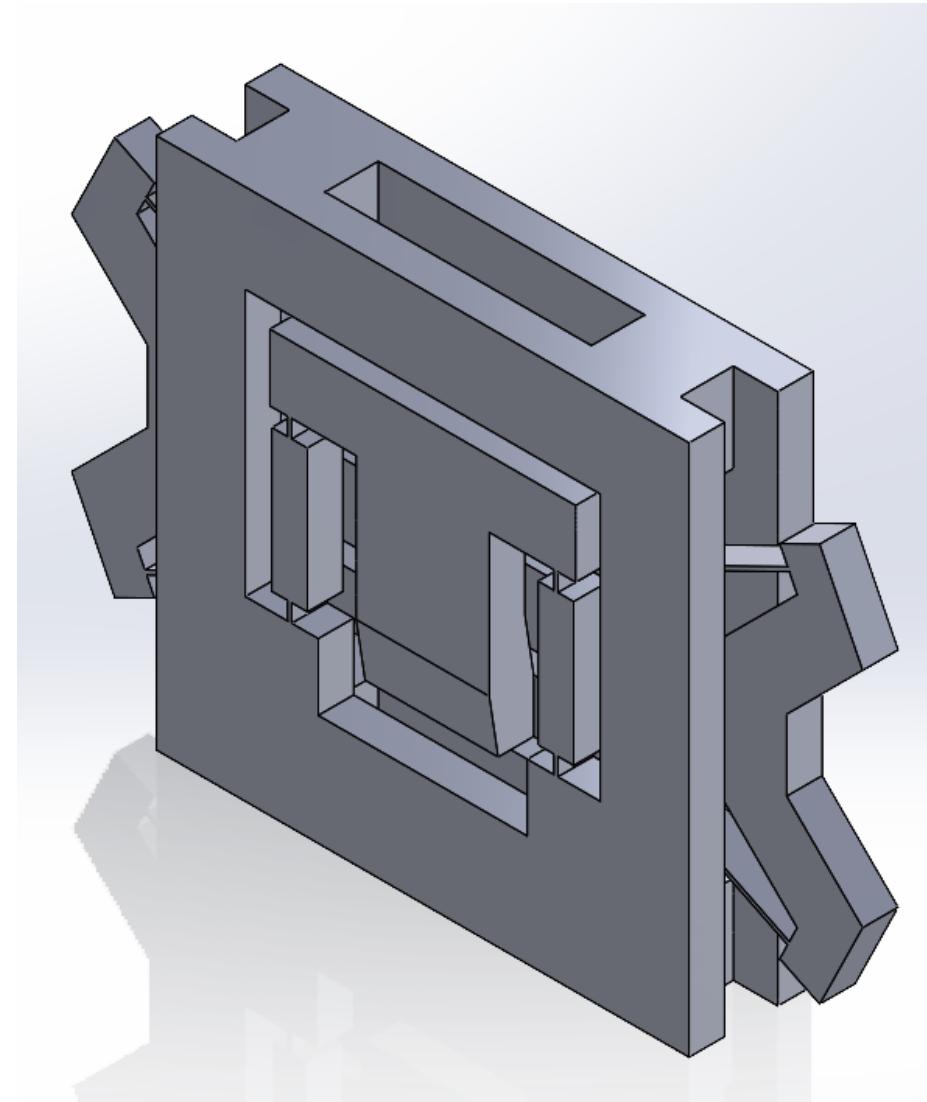
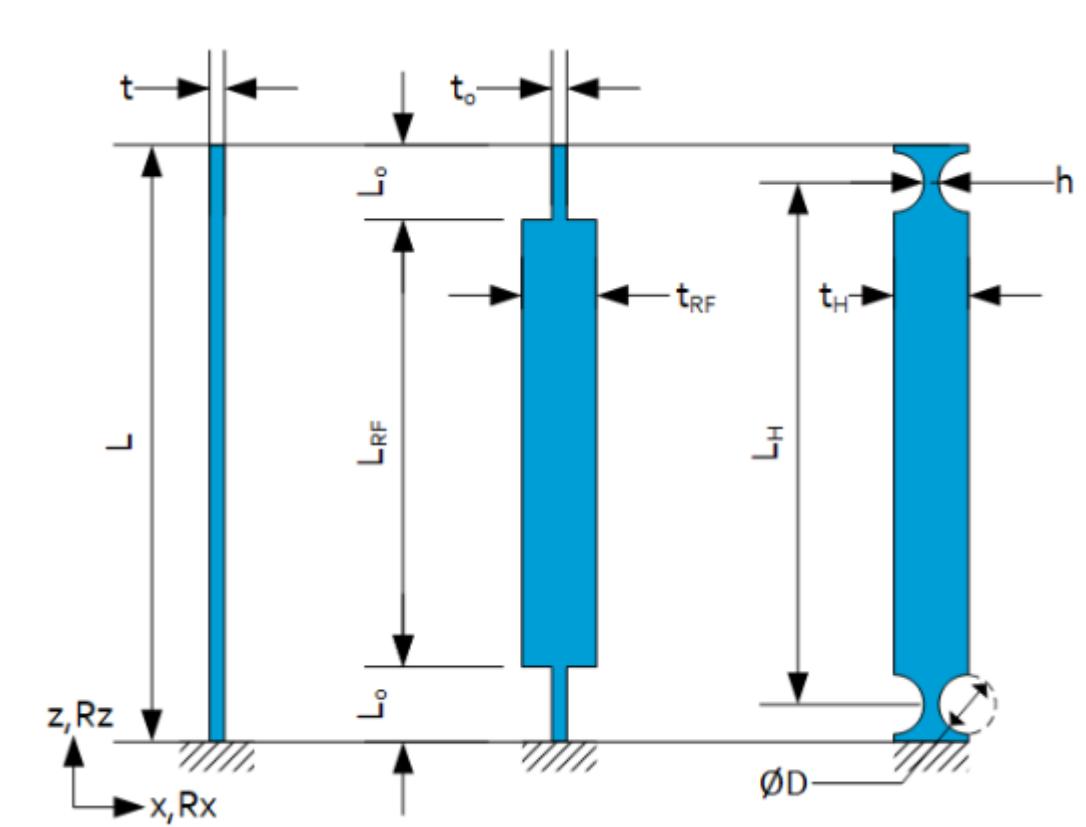
# Summary of FEM attempts

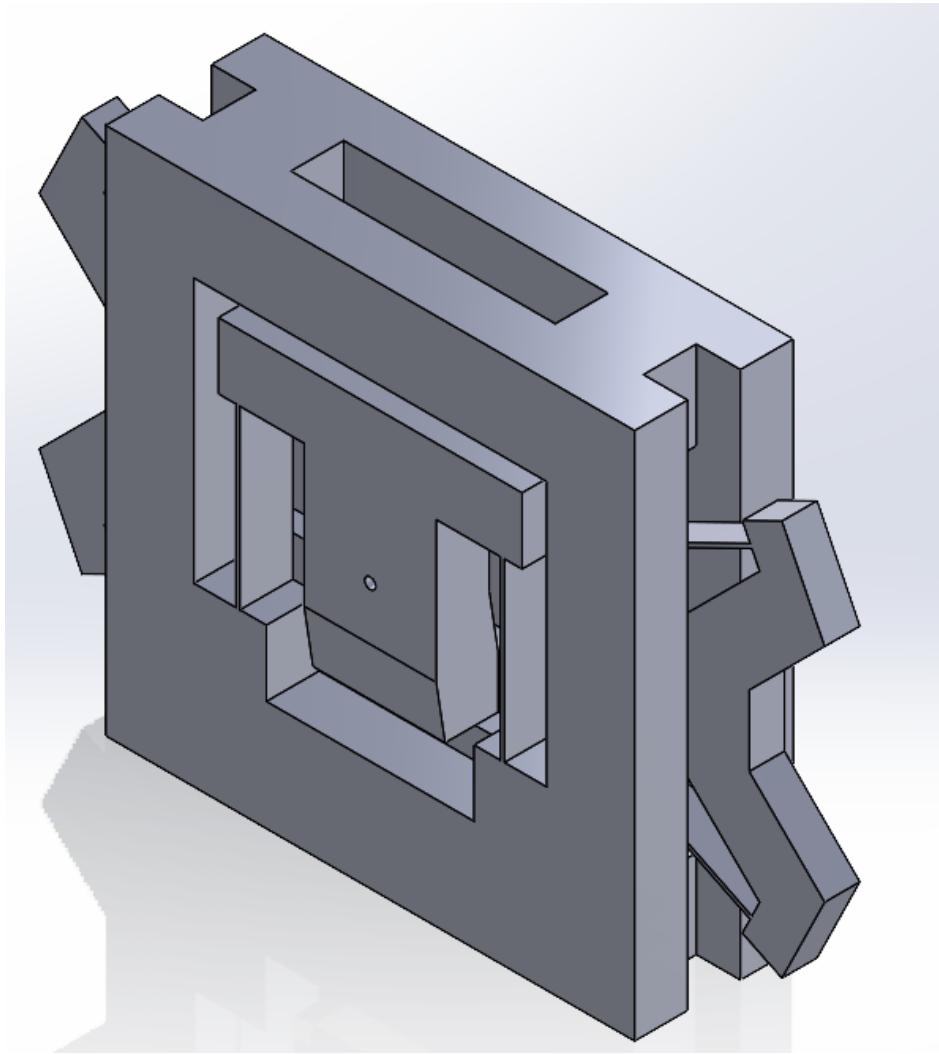
- Make connection piece thicker
- Widen the distance between input flexures from 40 – 60 mm such that connection piece can be made super thick
- Made in/output shuttle thicker

# Parasitic rotation??

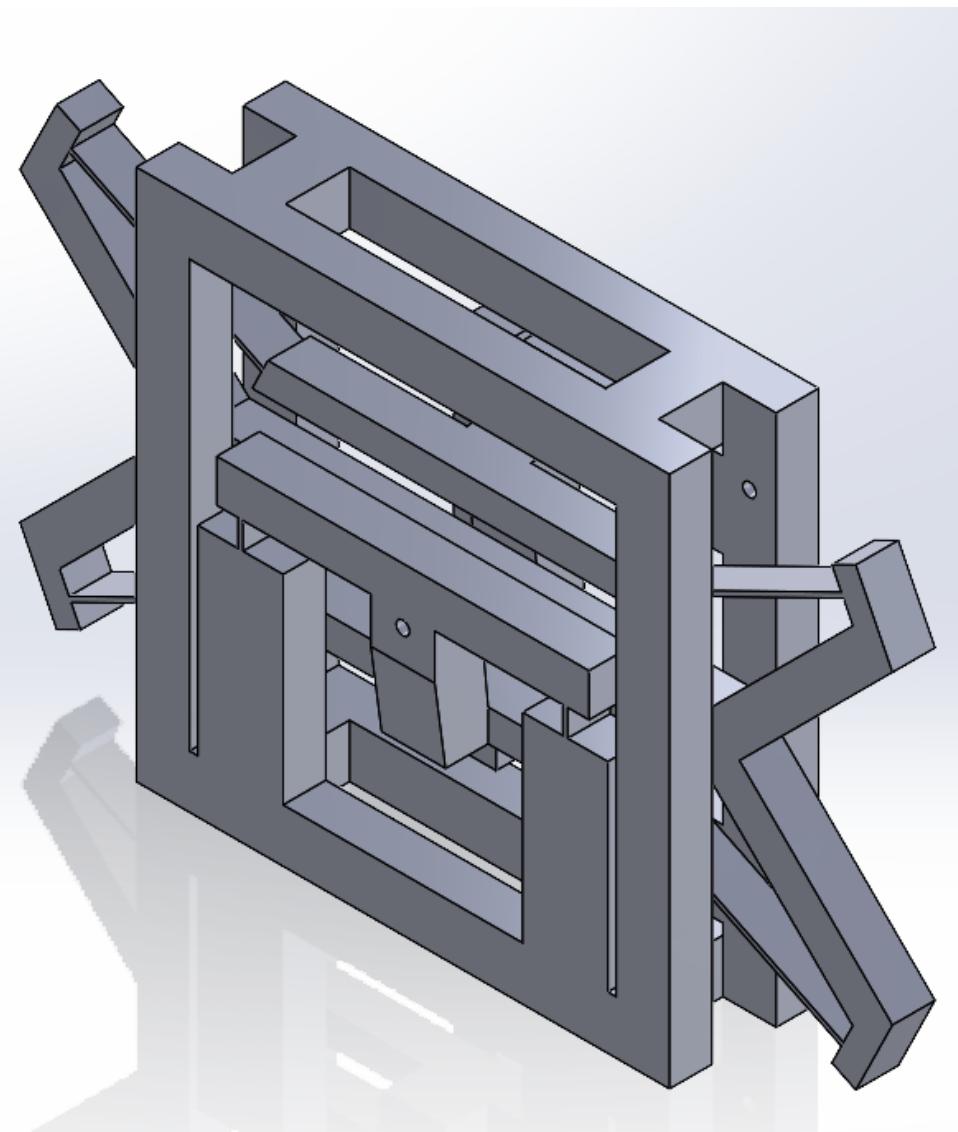
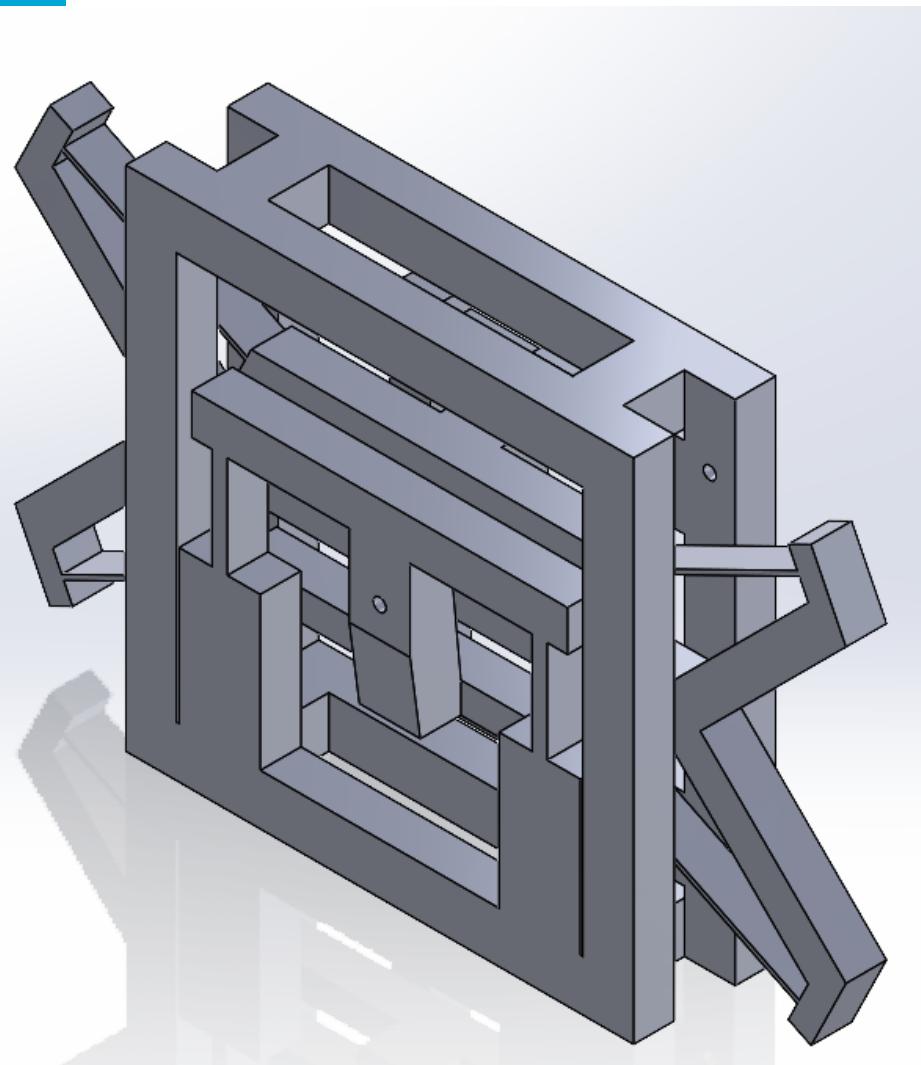


# Tried Lumped flexures





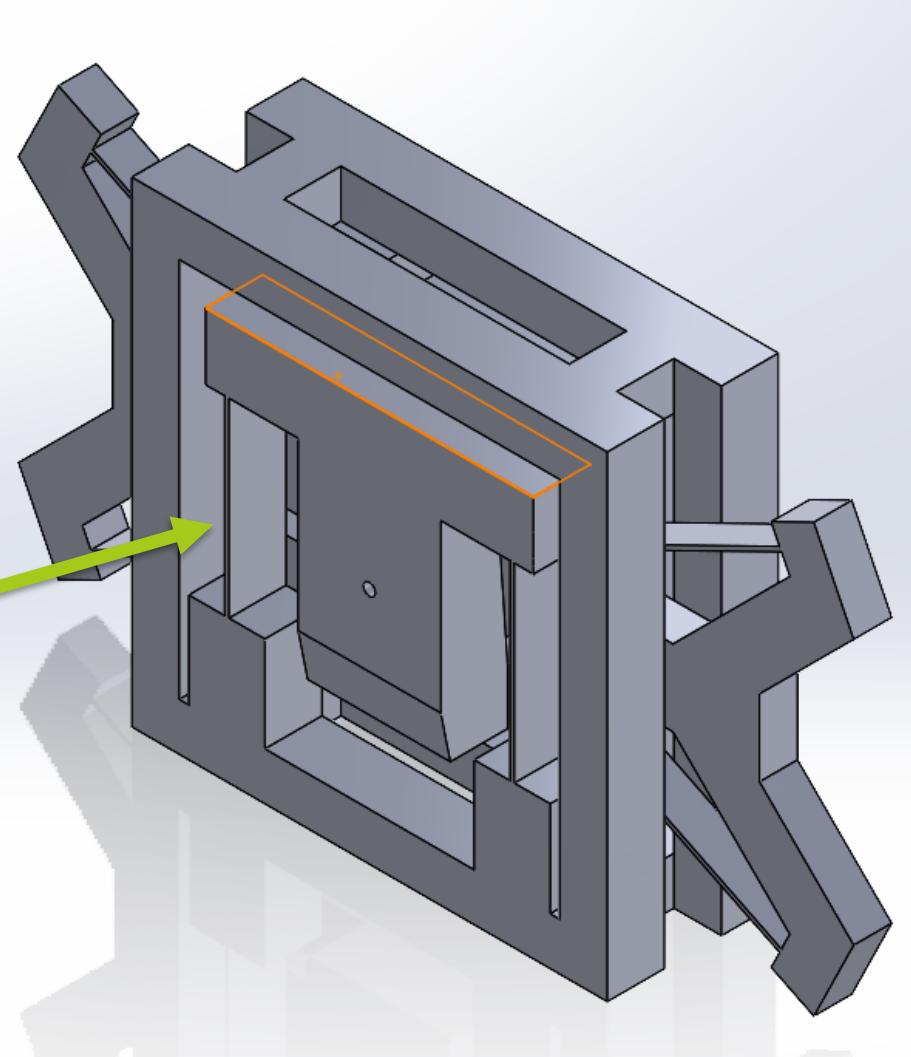
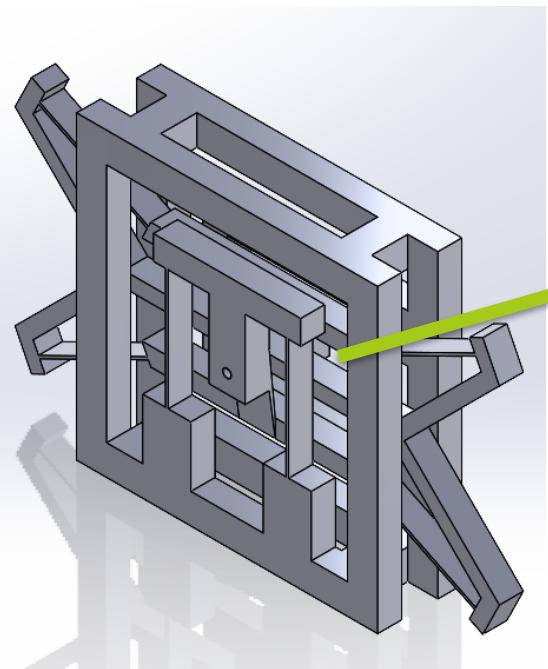
# Some weird attempts



# Final resulting prototype

## Changes

- Thicker in/output structure
- Made overall structure thicker wherever possible
- Old:



# Testing

