

# UAV Outback Challenge 2016

## Deliverable #1

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## 1 Aircraft Design

### 1.1 Flight

In order to achieve the required flight range (up to 60km) and endurance (60 minutes), the UAV will be a hybrid aircraft, fusing both a traditional fixed-wing aircraft and a Vertical TakeOff and Landing (VTOL) multi-rotor aircraft. A Skywalker X8 frame is being used, with 3 motors in Y3 configuration, as shown by the prototype in Figure 1.

These motors will act as a tri-copter when in VTOL mode, with a servo mounted in the back stabilising the aircraft's yaw by controlling the back motor, offsetting the rotation caused by the 3 motors. A servo system in the front of the aircraft allows the front motors to rotate forwards, transitioning the aircraft to fixed wing flight mode. The back motor is then disengaged, and the servos controlling the ailerons are engaged.



Figure 1: Outback Challenge aircraft - Prototype #1

### 1.2 Sensing and Control

Figure 2 outlines the on-board sensing capabilities that will be available to the aircraft. Each sensor is detailed below, as well as the control devices that will facilitate the autonomous behaviour of the aircraft.

#### Raspberry Pi

The Raspberry Pi will act as the aircraft's on-board computing platform, providing autonomy by giving flight commands to the flight controller, as well as the processing and intelligence for path planning, and object detection. It will also pull flight data from the PixHawk and other sensors, and generate detailed flight logs for later review.

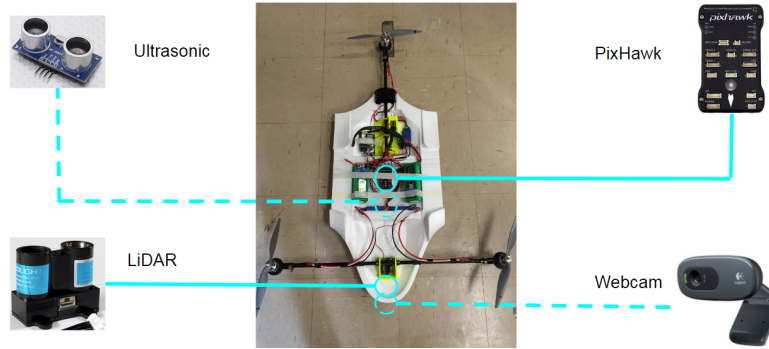


Figure 2: Onboard sensing capabilities for Prototype #1

### PixHawk Flight Controller

The PixHawk will control the aircraft's flight functionality, such as controlling motors and ailerons, and executing flight paths and commands from the Raspberry Pi. It also has several in-built or plug-and-play sensors, including a 3-axis accelerometer, altimeter, compass, and GPS. The PixHawk will provide the aircraft's telemetry to the Raspberry Pi and the base station, which will be augmented by the additional sensors below.

### Ultrasonic Module

The ultrasonic module will be mounted underneath the aircraft. The GPS and altimeter will provide altitude measurements during fixed-wing flight; the ultrasonic will augment these by providing a more reliable and controllable height measurement during rotor-based flight, assisting with search and landing.

### Webcam

The webcam will be mounted beneath the nose of the aircraft. It will provide vision for the aircraft's obstacle avoidance manoeuvres, and will form the basis for identifying Joe using his hat and blue jeans.

### LiDAR

The LiDAR will be mounted in the nose of the aircraft. The LiDAR can only measure the range of objects directly in front of it, so it will be mounted on a dual servo system that allows it to sweep a hemisphere in front of the aircraft (see Figure 3). It will provide a 3D map of the environment in front of the aircraft, and will assist in path planning and obstacle avoidance.

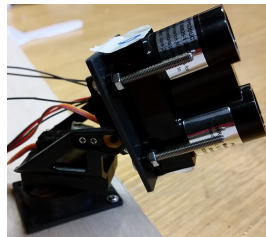


Figure 3: LiDAR mounting

### 1.3 Geofence System

A monitoring system will be implemented on the aircraft to detect proximity to, and crossing of, Geofence boundaries, as per 3.1.3. Although the PixHawk has inbuilt Geofence capabilities, the UAV will instead use the Raspberry Pi, as it will be at the highest level of control.

The PixHawk will provide the Raspberry Pi with the GPS, altimeter and accelerometer measurements, which will be used to estimate and monitor the position of the aircraft. Because the Raspberry Pi is the highest level of control and autonomy, a breach of the Geofence boundary will result in the immediate disengagement of control to all subsystems, and activation of the flight termination system, detailed below.

### 1.4 Flight Termination System

As per 3.1.4. there will be two options for flight termination on the completed craft. The higher level Raspberry Pi will be able to send termination signals to the Pixhawk directly if termination is required. On a lower level, the Pixhawk will be able to automatically comply with termination requirements through a failsafe if necessary (such as crossing a geofence or losing signal). The Raspberry Pi will act as a last resort if the Pixhawk fails to terminate, as it relies on a separate power supply.

In the event that the aircraft must terminate its flight, either by the Pixhawk or Pi, controls will be completely overridden to ensure servo positions are in termination position for fixed-wing flight, or the throttle is closed during multirotor flight (VTOL) as per 3.1.5.

As per 3.1.6. Flight termination will be automatically activated if the aircraft crosses a Geofence boundary, if the Geofence detection system fails, or if the autopilot has failed. Manual termination can also be activated by sending a signal to the Pi at the request of judges/range personnel, for example, if the aircraft appears out of control.

### 1.5 Miscellaneous

#### Communications

Will make use of INSERT HERE to maintain telemetry radio communications during flight, as per item 6 of the *General Requirements*.

Need to conform with Class Licences: Radiocommunications (Low Interference Potential Devices) Class Licence 2000 and Radiocommunications (Radio-Controlled Models) Class Licence 2002.

#### Safety Systems

The aircraft will be equipped with an external emergency stop button, red in colour with yellow surrounding disk, to disengage power, as per item 7 of the *General Requirements*. It will also be equipped with an external arming switch, and a visual state indicator to indicate armed (red) and disarmed (green) states, as per item 8 of the *General Requirements*.

#### Storage Compartment

The aircraft will be fitted with a storage compartment of appropriate dimensions in its center, allowing for Joe to deposit the sample describes in 1.4.1.

#### Mission Display

The ground station will make use of Ardupilot Mission Planner to provide a graphical display and data feed of the aircraft's mission, per 3.2.2. In addition, bespoke software will be developed to visualise the data sent back by the aircraft.

## 2 Use of UAVs

Execution of the mission will begin once the arming switch is pressed, and the aircraft becomes armed, at which point the aircraft will immediately begin autonomous operations; there will be no commands given by the UAV controller to initiate the mission.

Upon arming, the aircraft will perform vertical takeoff from the Base using its VTOL mode, and will ensure it remains within the base Geofence. Once it has reached an appropriate altitude (yet to be tested, no more than 1500ft AGL), the aircraft will transition to its fixed-wing flight mode by rotating its front motors 90° and disengaging the back motor.

Once fixed-wing mode is engaged, the aircraft will begin its transit to the remote landing site, through the provided waypoints, whilst continuing to monitor its position relative to the Geofence boundaries.

Upon reaching the remote landing site, the aircraft will begin performing sweeps within the landing site Geofence, attempting to get an approximate position for Joe's location using the webcam, while remaining in the air.

Once Joe has been located, the aircraft will transition back to VTOL mode. It will then use the webcam and LiDAR to maintain vision of Joe, while avoiding obstacles in the environment, in order to land in the safe zone (30m-80m) around Joe.

After landing, the Raspberry Pi will disengage all power to the motors, and will display the "disarmed" signal to Joe. He may then approach the aircraft to place the Sample in the Storage Compartment, and engage the arming switch.

Having been rearmed, the aircraft will wait one minute before initiating vertical takeoff, and reversing the manoeuvres to return to base; transitioning to fixed-wing flight, traveling through the transit corridor, and returning to VTOL mode to land at the Base. The aircraft will then disarm itself once more, and the mission will be completed.

This behaviour is consistent with the rules outlined in section 3. If at any time during the mission the Raspberry Pi determines a Geofence was breached, the flight termination system will be immediately activated, per sections 3.1.4-3.1.6.

## 3 Risk Assessment

See attached "Risk Assessment" document.

## 4 Risk Management

See attached "Risk Management" document.

Arming switch E-stop button