

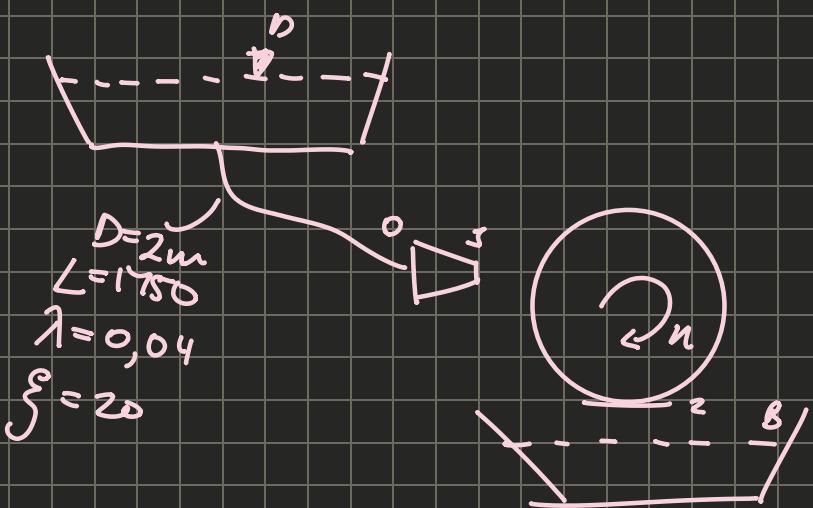
Esercitazione 4 - Hydraulic Turbines and Ducts

Exercise Pelton

$$z_0 - z_B = 1000 \text{ m}$$

$$z_0 \approx z_1 \approx z_2 \approx z_B$$

No draft tube
(only for Pelton)



$$\rho = 1000 \frac{\text{kg}}{\text{m}^3}$$

$$\beta_2 = -68^\circ$$

$$n = 600 \text{ rpm}$$

nozzles $\rightarrow i = 3$

$$D_m = 2 \text{ m}$$

↳ mean diameter
of machine

$$d = 0.22 \text{ m}$$

$\vdash \dashv$ ideal machine
 $\dashv \dashv$ optimized.
 $\rightarrow \varphi = 1, \psi = 1$

1) Velocity Triangles

2) Calculate gH_m ?

3) L^* ?

4) Is the condition optimized?

Machine = Stator + Rotor

\hookrightarrow Nozzle \hookrightarrow Turbine
 $\hookrightarrow \ell_{wN}$ $\hookrightarrow \ell_{wB} \rightarrow \psi = \frac{\omega_2}{\omega_1}$
 $\hookrightarrow \varphi$

BME ($D \rightarrow i$)

$$\ell - \ell_w - y_p = \frac{P_i - P_D}{\rho} + \frac{V_i^2 + V_D^2}{2} + g(z_i - z_D)$$

\sim since $P_i = P_D = P_{ATM}$

\downarrow
 ↳ Penstock
 $= \ell_{wN}$

$$\rightarrow \frac{v_1^2}{2} = g(z_0 - z_1) - \rho_{wn} - y_p$$

we want v_i to find the velocity triangles.

$$\rho_{wn} \rightarrow \varphi = \frac{v_i}{v'_i} \rightarrow \text{ideal velocity} \Rightarrow v_i = \varphi v'_i$$

y_p must be considered since it is not part of the nozzle. Problem since we now need to find v_0 and v_i .

$$y_p = \left(\xi + \frac{\lambda L}{D} \right) \frac{v_0^2}{2} = \textcircled{K},$$

$$Q = Q_1 \rightarrow \frac{\pi D^2}{4} v_0 = \frac{\pi d^2}{4} v_i i = \frac{\pi d^2}{4} \varphi v'_i i$$

Mass Balance

$$\Rightarrow v_0 = \varphi_i \left(\frac{d}{D} \right)^2 v'_i$$

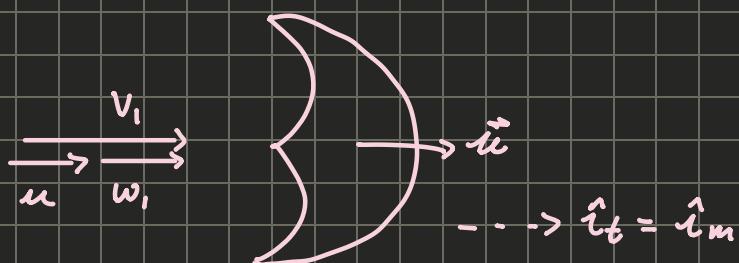
$$\textcircled{K}_i = \left(\xi + \frac{\lambda L}{D} \right) \varphi_i^2 \left(\frac{d}{D} \right)^4 \frac{v'_i^2}{2}$$

Substitute in BME

$$\frac{v_1^2}{2} = g(z_0 - z_1) - \left(\xi + \frac{\lambda L}{D} \right) \varphi_i^2 \left(\frac{d}{D} \right)^4 \frac{v'_i^2}{2}$$

$$\Rightarrow v'_i = \sqrt{\frac{2g(z_0 - z_1)}{1 + \left(\xi + \frac{\lambda L}{D} \right) \varphi_i^2 \left(\frac{d}{D} \right)^4}}$$

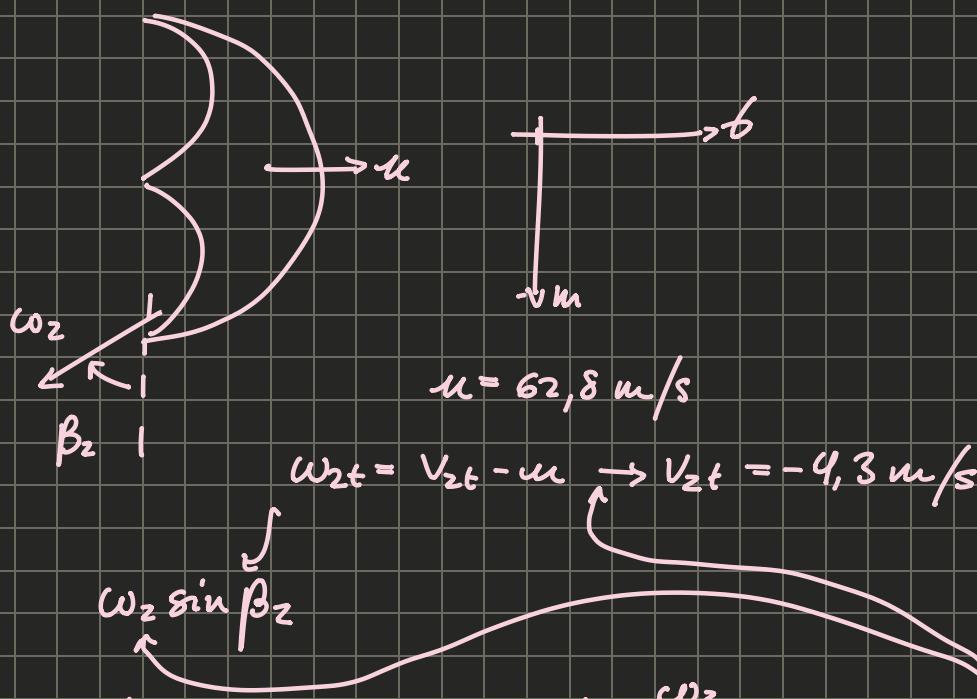
since it's an ideal machine $\rightarrow \varphi = 1 \Rightarrow v_i = v'_i = 135,2$



$$n = n \cdot \frac{2\pi}{60} \cdot \frac{D_m}{2} = 62,8 \frac{\text{m}}{\text{s}}$$

$$\omega_1 = V_1 - u = 72,4 \frac{\text{m}}{\text{s}}$$

Outlet



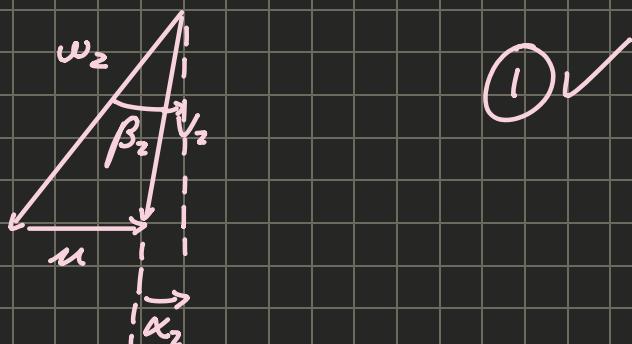
$$u = 62,8 \frac{\text{m}}{\text{s}}$$

$$w_{2t} = v_{2t} - u \rightarrow v_{2t} = -4,3 \frac{\text{m}}{\text{s}}$$

since the machine is ideal, $IV = \frac{\omega_2}{\omega_1} = 1 \Rightarrow \omega_1 = \omega_2 = 72,4 \frac{\text{m}}{\text{s}}$

$$v_{2x} = \omega_{2x} = \omega_2 \cos \beta_2 \Rightarrow v_{2x} = 27,2 \frac{\text{m}}{\text{s}}$$

$$\alpha_2 = \tan \frac{V_{2t}}{V_{2x}} = -9^\circ$$



$$3) \quad \dot{L} = ? \quad -4,3 \frac{\text{m}}{\text{s}} = m \cdot l = 134,9 \text{ MW}$$

$$l = u \left(v_{2t} - v_{1t} \right) = -8760,6 \frac{\text{J}}{\text{kg}}$$

$\hookrightarrow u_1 = u_2$

$$\dot{m} = \rho Q = \rho \cdot \frac{\pi d^2}{4} V_1 = 15400 \frac{\text{kg}}{\text{s}}$$

2) gH_m ?

$$\eta = \frac{|e|}{gH_m}$$

Assuming we have a completely ideal turbine ($\eta = 1$), they would be the same, but we don't.

$$gH_m = T_0 - T_2 = T_D - y_p - T_2 = \frac{P_D - P_2}{\rho} + \frac{V_D^2 - V_2^2}{2} + g(z_D - z_2) - y_p$$

↳ The energy available to the turbine is the difference between D and B .

$$\text{With } T_B = \underbrace{g(z_D - z_B)}_{\text{--- --- --- --- ---}} - y_p \rightarrow \text{The one we use.}$$

BME ($D \rightarrow I$):

$$-y_p = \frac{P_I - P_D}{\rho} + \frac{V_I^2 - V_D^2}{2} + g(z_B - z_D) \rightarrow \underbrace{\frac{V_I^2}{2}}_{gH_m} = g(z_D - z_P) - y_p$$

$$\Rightarrow gH_m = \frac{V_I^2}{2} = 9139,52 \text{ J/kg}$$

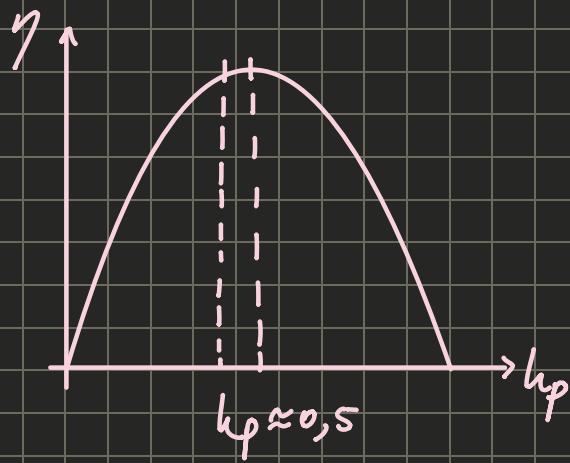
↳ This is always true, independent on whether the machine is ideal or not.

$$\eta = \frac{|e|}{gH_m} = 95,9\%$$

Does the machine work in the optimised condition?

↳ What is the definition of optimised condition?

|



when $\alpha = \frac{v_i}{\omega}$

$$h_{p,\text{opt}} = \frac{\varphi}{2}$$

$$h_p = \frac{\alpha}{v_i'} \quad \text{or}$$

$$h_p = \frac{\alpha}{v_i'} = 0,464 \neq 0,5$$

We need to increase α , to reach $h_{p,\text{opt}}$

$$h_{p,\text{opt}}(\varphi=1)$$

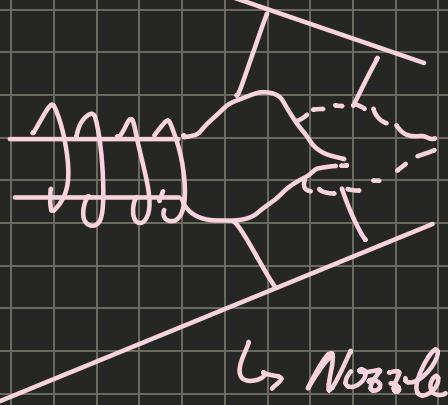
Load regulation in Pelton turbine (second part of exercise)

$$\dot{L} = m \cdot l$$

$$\rho Q$$

$$\rightarrow i \frac{\pi d^2}{4} \varepsilon \varphi v_i'$$

$$0 < \varepsilon < 1$$



coefficient of decrease of d.

$$\varepsilon = 0,6 \quad n = 600$$

$$\xi = 20$$

$$\dot{L}^o = ?$$

$$g H_m = ?$$

$$\eta = ?$$

optimized?

BME
(Δ → 1)

$$-y_p = \frac{v_i'^2}{2} + g(z_s - z_o)$$

$$\left(\xi + \frac{\lambda L}{D} \right) \frac{v_o^2}{2} = \left(\xi + \frac{\lambda L}{D} \right) \varphi_i^2 \varepsilon^2 \left(\frac{d}{D} \right)^4 \frac{v_i'^2}{2}$$

$$Q_o = Q_i \sim \frac{\pi D^2}{4} v_o = \frac{\pi D^2}{4} i \varepsilon \varphi v_i' \rightarrow v_o = \varphi_i \varepsilon \frac{d^2}{D^2} v_i'$$

$$v_1' = \sqrt{\frac{2g(z_D - z_B)}{1 + \left(\frac{f + \gamma L}{D}\right) \varphi_i^2 \epsilon^2 \left(\frac{d}{D}\right)^4}} = 139,2 \frac{m}{s}$$

Before we have 135 m/s,

since $\left(\frac{d}{D}\right)^4 \approx 10^{-4} \rightarrow$ this is negligible, so the velocity will be nearly the same.

We are able to modify the power while maintaining relatively good efficiency.

Same process as before

$$u = 62,8 \frac{m}{s}$$

$$\omega_1 = 76,4 \frac{rad}{s}$$

$$\mathcal{N} = 1 \rightarrow \omega_2 = \omega_1 = 76,4 \frac{rad}{s}$$

$$\begin{cases} \omega_{2x} \\ \omega_{2t} = -70,8 \frac{rad}{s} \end{cases}$$

$$v_{2t} = -8 \frac{m}{s}$$

$$\ell = u(v_{2t} - v_{1t}) = -9244,2 \text{ J/kg}$$

$$L = \rho Q \ell = -58,7 \text{ MW}$$

$\hookrightarrow \frac{i\pi d^2}{4} \epsilon v_1 = 6,3 \frac{m^3}{s}$

\rightarrow Before it was $\approx 134 \text{ MW}$, so we have been able to reduce L by reducing d .

$$g H_m = \frac{v_1'^2}{2} = 95,4\% \rightarrow$$
 we have only lost 0,5%, while

reducing L by a lot.

$$h_{p, \text{opt}} = \frac{\varphi}{2} = .5 \quad h_p = \frac{u}{V_i} = 0,451$$

↳ We are not optimised.

How can we change n so that the efficiency is the same as before.

$$h_{pa} = 0,464$$

↳ $h_{p, \text{opt}}$ case a.

since now $h_p = 0,451 = \frac{u}{V_i}$, we need

to increase u to have the same
 h_p as before

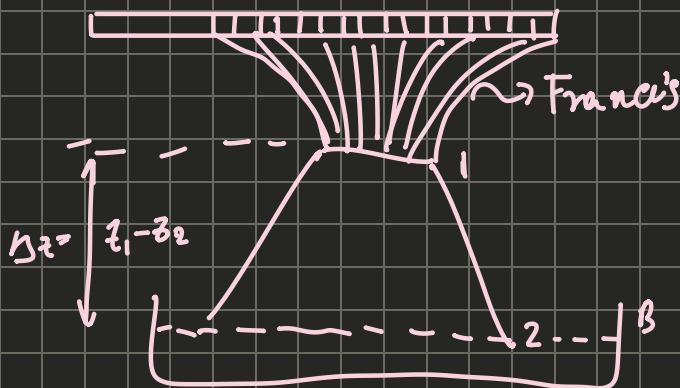
$$h_{pa} = h_{ps} = 0,464$$

$$\frac{u}{V_i} = 0,464 \rightarrow u = 0,464 V_i \Rightarrow n = 617 \text{ rpm}$$

$$\hookrightarrow u \cdot \frac{2\pi}{60} \cdot \frac{D_m}{\lambda}$$

↳ speed at which
we need to run the
turbine to have the
same h_p and η .

Exercise 5 - Draft Tube



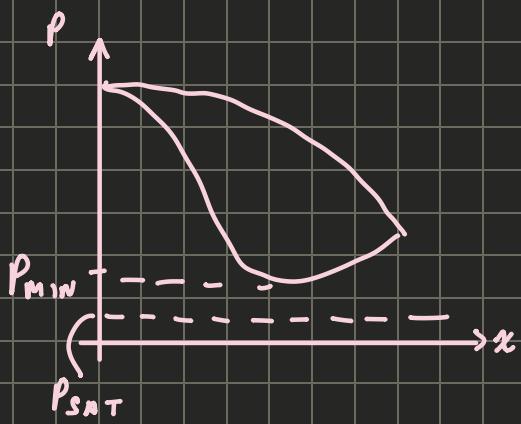
$$S_2 = 4 \text{ m}^2$$

$$S_1 = 0,75 \text{ m}^2$$

$\xi_{\text{diff}} = 2$ (evaluated at
exit)
(excluding outlet loss)

$$\frac{P_{1, \text{min}}}{\rho g} = 0,75 \text{ m}$$

$$Q = 10 \frac{\text{m}^3}{\text{s}}$$



To not have cavitation
 $P_{\min} > P_{\text{SAT}}$

BME \rightarrow B

$$-y_{DT} = \frac{P_B - P_1}{\rho} + \frac{y_B^2 - v_1^2}{2g} + g(z_B - z_1)$$

?

$$z_1 - z_B = \frac{P_{DTm} - P_1}{\rho g} + \frac{v_1^2}{2g} - \frac{y_d}{g}$$

$$Q_1 = Q = S_1 V_1 \rightarrow V_1 = 13,3 \text{ m/s}$$

$$Q_2 = Q = S_2 V_2 \rightarrow V_2 = 2,5 \text{ m/s}$$

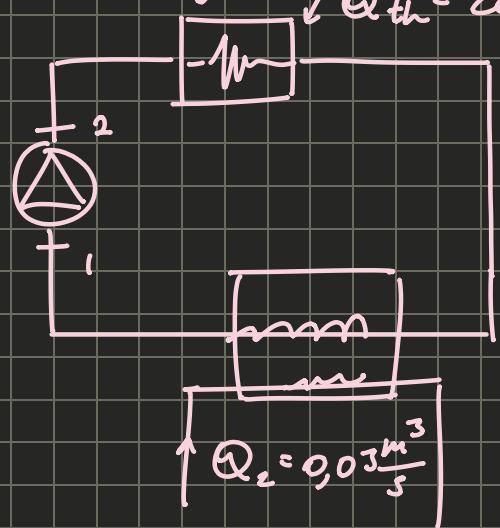
$$y_d = f \frac{V_2^2}{2} + \frac{V_2^2}{2} = (f+1) \frac{V_2^2}{2}$$

$\hookrightarrow \sum f_{\text{conv}} + \frac{\pi L}{Dg} \rightarrow$ distributed losses
 exit losses
 / discharge

$$\rightarrow z_1 - z_B = \frac{P_{DTm}}{\rho g} - \frac{P_{1,\min}}{\rho g} - \frac{V_1^2}{2g} + (f+1) \frac{V_2^2}{2g} = 1,38 \text{ m}$$

If the undivine is placed at a higher height, there will be cavitation.

Exercise 2 - Hydraulic Ducts



$$Q_p = 50 \frac{\ell}{s} = \frac{.05 \text{ m}^3}{s}$$

$$L_p = ?$$

$$\Delta T_2 = ?$$

$$C_L = 4186 \frac{J}{kg \cdot K}$$

$$\eta_{pump} = 0.6$$

$$\xi = 150$$

$$D = 100 \text{ mm}$$

The role of this pump is to overcome the losses and impose Q .

$$(l - l_w = y)$$

Proof:

BME 2 → 1

$$\cancel{l - l_w - y} = \frac{P_1 - P_2}{\rho} + \cancel{\frac{V_1^2 - V_2^2}{2}} + g(z_1 - z_2)$$

$$y = \frac{P_2 - P_1}{\rho} \quad \textcircled{X}$$

BME 1 → 2 "we don't have a duct"

$$l - l_w - y \neq \frac{P_2 - P_1}{\rho} + \cancel{\frac{V_2^2 - V_1^2}{2}} + g(z_2 - z_1)$$

$$l - l_w = \frac{P_2 - P_1}{\rho} \quad \textcircled{X} \textcircled{X}$$

$$l - l_w = y$$

$$y = \int \frac{V^2}{2} \rightarrow Q_p = \frac{\pi D^2}{4} V \rightarrow V = \frac{4 Q_p}{\pi D^2}$$

$$y = \int \frac{8 Q_p^2}{\pi^2 D^4} = 3072 \text{ J/kg}$$

$$\eta_{hydr} = \frac{l - l_w}{l} \rightarrow l - l_w = \eta_{hydr} l$$

$$\rightarrow \eta_{hydr} l = y \rightarrow l = \frac{y / \eta_{hydr}}{1 - \eta_{hydr}} = \frac{3072}{.6} = 5120 \text{ J/kg}$$

$$\dot{L} = \rho Q_p l = 256 \text{ kW}$$

$$2) \Delta T = ?$$

We need the heat exchanger to remove the heat from the other heat exchanger, and also the heat from l_w , since

$$l_w = T dS_{IRK}$$

$$\delta q = T \Delta S - \underbrace{T \delta s_{IRK}}_{l_w}$$

$$\delta q + \delta l_w = T \Delta S$$

$$\rightarrow q + l_w = \int T dS = C_L \Delta T_p \quad \rightarrow \frac{\dot{Q}_{th}}{m} + (1-\eta)l = C_L \Delta T_p$$

$$dh = T dS + V dP$$

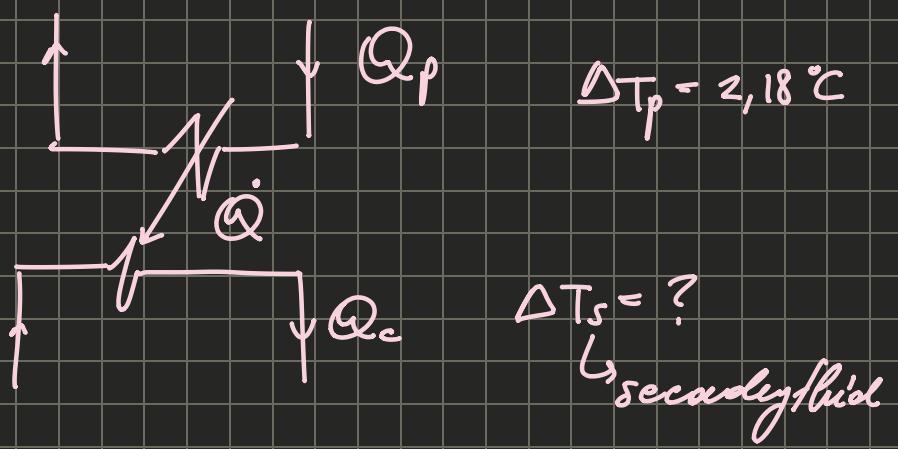
$$dH = T dS - P dV = C_L \Delta T$$

$$\eta = \frac{l - l_w}{l} \rightarrow l_w = (1 - \eta)l$$

primary fluid

$$\Rightarrow \Delta T_p = 2,18^\circ C$$

To compensate ΔT_p we have the secondary heat exchanger



$$c_L \rho Q_p \Delta T_p = c_L \rho Q_c \Delta T_s$$

$$\Delta T_s = \frac{Q_p}{Q_c} \Delta T_p = 3,6 \text{ } ^\circ\text{C}$$