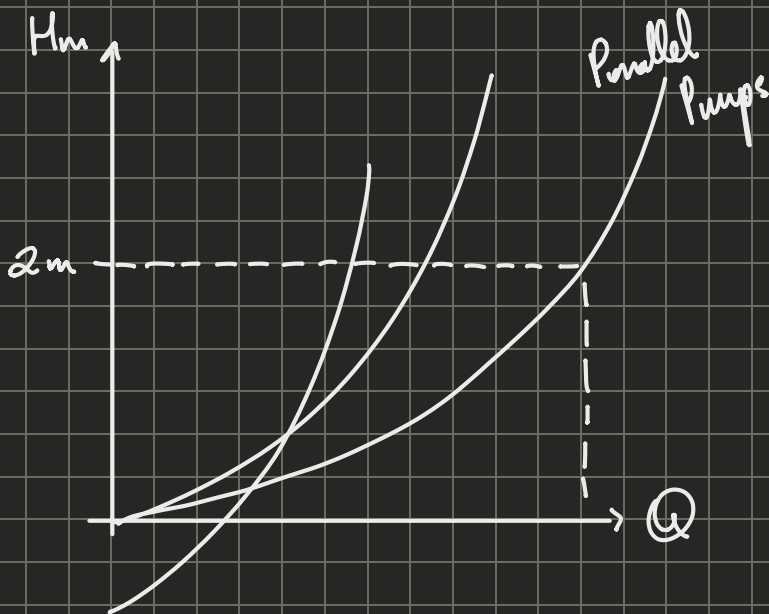


Exercitacione 10 -

Exercise 5 - Hydraulic Turbines in Parallel

$$H_m = 0.4Q^2 + 0.5Q$$



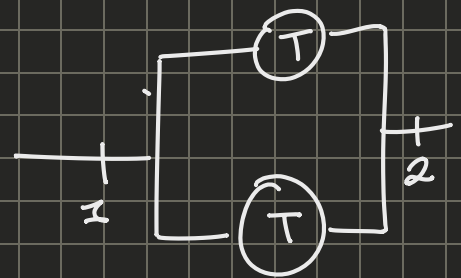
This is the first time we see the curve we see for turbines since they are generally useless, since in turbines we change geometry to regulate power.

$$H_{m//} = \frac{0.4}{4}Q^2 + \frac{0.5}{2}Q$$

$$Q_{T1} = Q_{T2} = \frac{1}{2}Q_{//} = \frac{1}{2}Q$$

$$H_{m//}(n) = 0.1Q^2 + 0.25Q\left(\frac{n}{375}\right)$$

$$\Rightarrow n = 825 \text{ rpm}$$



$$g H_m = \pi_1' - \pi_2'$$

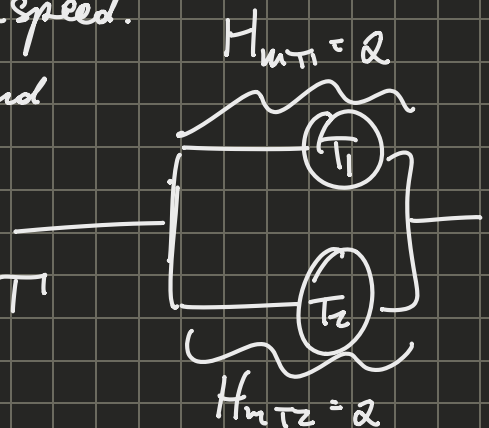
↳ For this type of exercise

Once the pumps/turbines are different we can't apply simplifying rules:

$$H_{T1} = 0.4Q_{T1}^2 + 0.5Q_{T1} = 2 \rightarrow Q_{T1} \text{ found}$$

$$H_{T2} = 0.4Q_{T2}^2 + 0.5Q_{T2}\left(\frac{n_{T2}}{375}\right) = 2$$

We know $H_{T1} = H_{T2} = 2$, because $\Delta\pi'$ does not change.



$$Q_{11} = 2.5 \text{ m}^3/\text{s} = Q_{T1} + Q_{T2} \rightarrow Q_{Te} = 0.8 \text{ m}^3/\text{s}$$

Calculate the first equation

$$Q_{T2} = 0.8 \text{ m}^3/\text{s} \rightarrow n_{T2}$$

Exercise 2

Ideal $\rightarrow \Delta S = 0$

Impulse $\chi = 0 \Rightarrow h_1 = h_2 = h_3$

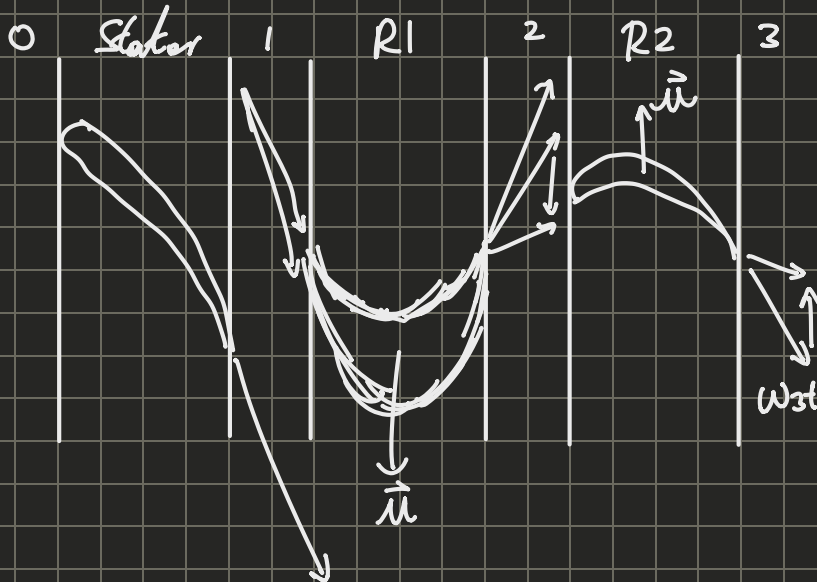
$$P_1 = P_2 = P_3 = P_{\text{out}}$$

Axial $\Rightarrow \Delta u = 0$

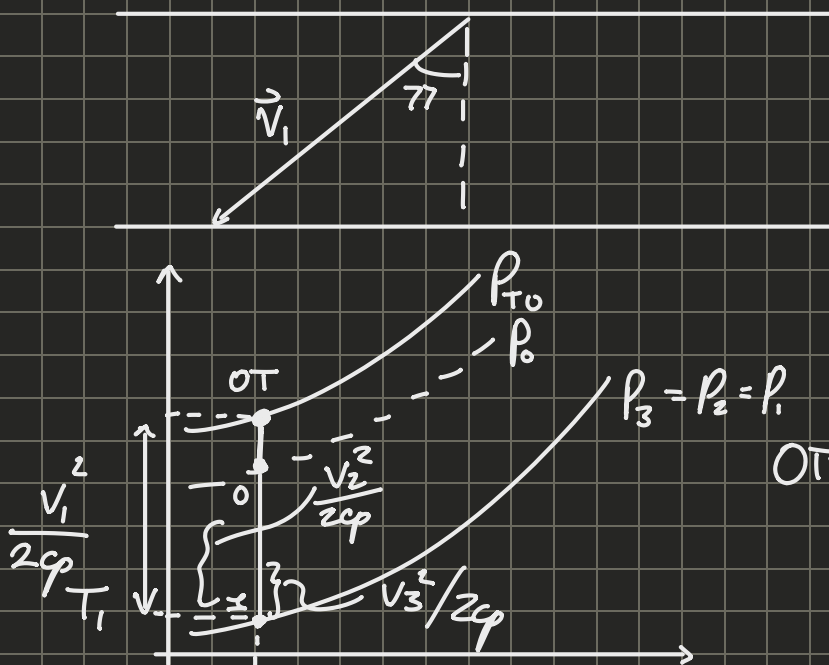
Not optimize:

$$V_{3t} \neq 0$$

Axial Velocity = constant
+ impulse \Rightarrow rotors are symmetric.

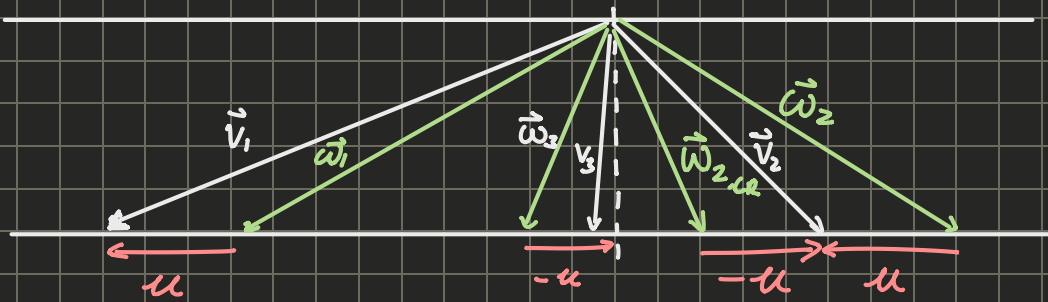


Isentropic flow
 \Rightarrow same blade height along flow.



$$0T = 1T > 2T > 3$$

$$S_0 = S_1 = S_2 = S_3$$



Symmetric $\beta_1 = \beta_2$
 $\beta_{2cr} = \beta_3$

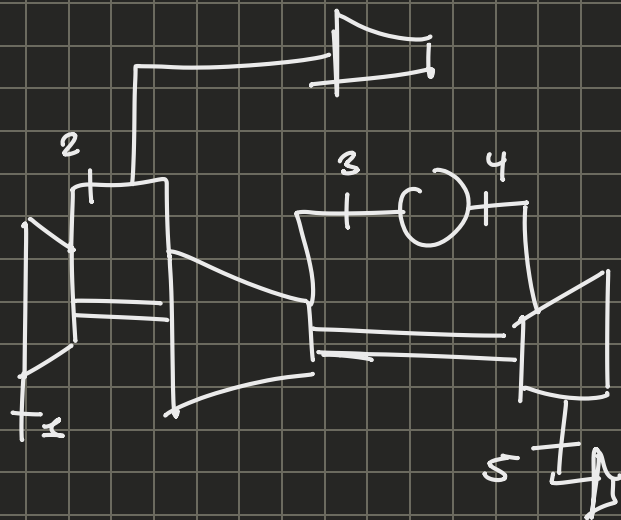
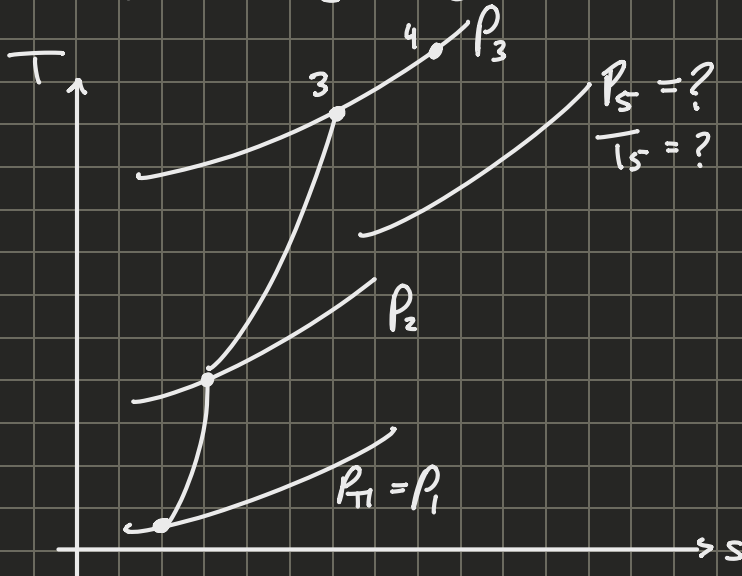
This is like a Curtis which exchanges a lot of energy at the expense of some high tangential flow.

$$w_{2cr,3} = v_{20} - (u_{cr}) = v_{2t} - (-u) = v_{2t} + u$$

$$v_{30} = w_{3t} + u_{cr} = w_{3t} - u$$

Exercise 3

Total = Static since considered black box



Balance Shaft:

$$\dot{L}_F + \dot{L}_C + \dot{L}_E = 0$$

$$\dot{m}_{in} c_p (T_2 - T_1) + \dot{m}_{com} c_p (T_3 - T_2) + \dot{m}_{FG} c_p (T_5 - T_4) = 0$$

$\dot{m}_{in} = \dot{m}_{com} = \dot{m}_{FG}$

We need $P_5 > P_0$ since this is propulsion system so we need residual pressure to propel us.

$$\rightarrow T_s$$

$$\hookrightarrow \dot{m}_{\text{cool}} + \dot{m}_F$$

$$\alpha = \frac{\dot{m}_{\text{cool}}}{\dot{m}_F}$$

$$\frac{T_s}{T_4} = \left(\frac{P_s}{P_4} \right)^{\frac{n_T - 1}{n_T}}$$

Nozzles: (Always isentropic for exam)

We find β , if $\beta > \beta_{\text{crit}}$ and the nozzle is purely convergent we can impose $M=1$