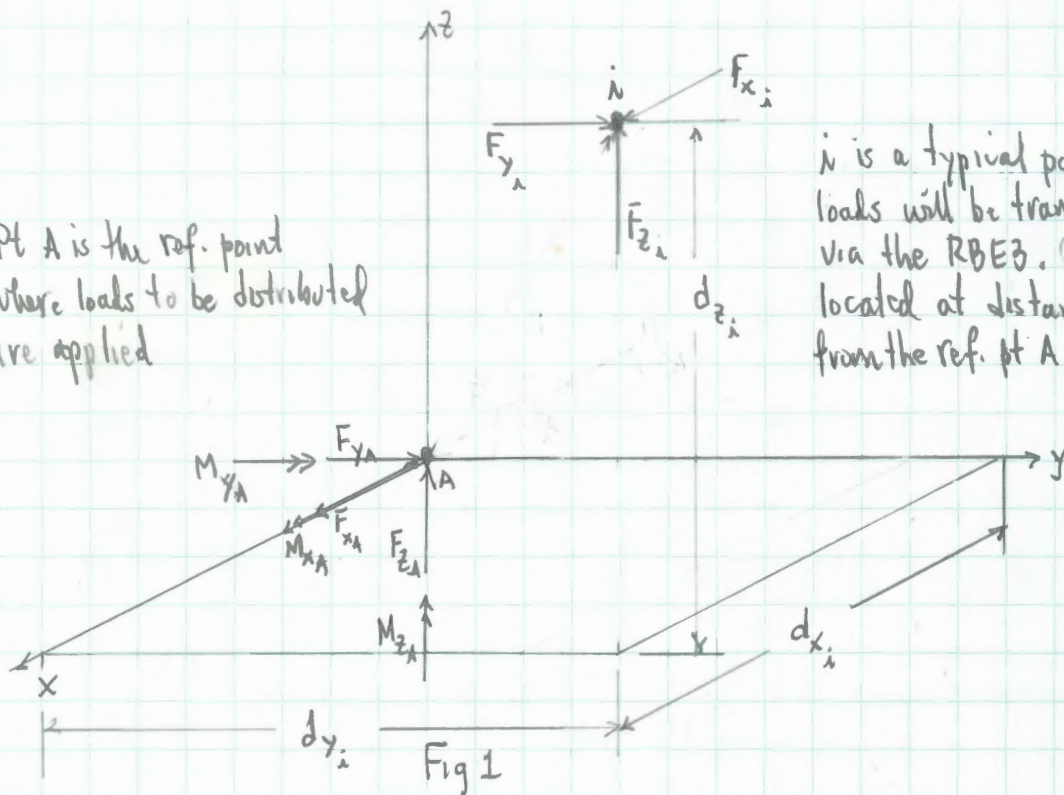


# RBE3 Equations

Pt A is the ref. point where loads to be distributed are applied



i is a typical point to which loads will be transferred to via the RBE3. (There are N points i located at distances  $d_{x_i}$ ,  $d_{y_i}$ ,  $d_{z_i}$  from the ref. pt A)

The forces on the i points must add up to the forces on A:

$$\sum_{i=1}^N F_{x_i} = F_{x_A} \quad \sum_{i=1}^N F_{y_i} = F_{y_A} \quad \sum_{i=1}^N F_{z_i} = F_{z_A} \quad (1) a, b, c$$

Moments at A due to the forces at i:

$$\sum_{i=1}^N (F_{z_i} d_{y_i} - F_{y_i} d_{z_i}) = M_{x_A}, \quad \sum_{i=1}^N (F_{x_i} d_{z_i} - F_{z_i} d_{x_i}) = M_{y_A}, \quad \sum_{i=1}^N (F_{y_i} d_{x_i} - F_{x_i} d_{y_i}) = M_{z_A} \quad (2) a, b, c$$

Assume that the  $F_{x_i}$ , etc can be written as

$$F_{x_i} = \frac{w_i}{\sum_{i=1}^N w_i} F_{x_A}, \quad F_{y_i} = \frac{w_i}{\sum_{i=1}^N w_i} F_{y_A}, \quad F_{z_i} = \frac{w_i}{\sum_{i=1}^N w_i} F_{z_A} \quad (3) a, b, c$$

Define  $W_T = \sum_{i=1}^N w_i$  ( $w_i$  are weighting factors) (4)

Substitute (3) into (1) and we see (1) is satisfied since, for example

$$\sum_{i=1}^N F_{x_i} = \frac{\sum_{i=1}^N w_i F_{x_A}}{\sum_{i=1}^N w_i} = F_{x_A} \quad (\text{since } F_{x_A} \text{ constant})$$

Substitute (3)a, (4) into (2) a:

$$+\sum_{i=1}^N \left( \frac{w_i}{W_T} F_{z_A} dy_i - \frac{w_i}{W_T} F_{y_A} dz_i \right) = M_{x_A}$$

or

$$+\left[ \frac{F_{z_A}}{W_T} \sum_{i=1}^N w_i dy_i - \frac{F_{y_A}}{W_T} \sum_{i=1}^N w_i dz_i \right] = M_{x_A} \quad (5)$$

Substitute (3)b, (4) into (2) b:

$$+\left[ \frac{F_{x_A}}{W_T} \sum_{i=1}^N w_i dz_i - \frac{F_{z_A}}{W_T} \sum_{i=1}^N w_i dx_i \right] = M_{y_A} \quad (6)$$

Substitute (3)c, (4) into (2) c:

$$+\left[ \frac{F_{y_A}}{W_T} \sum_{i=1}^N w_i dx_i - \frac{F_{x_A}}{W_T} \sum_{i=1}^N w_i dy_i \right] = M_{z_A} \quad (7)$$

Define  $\bar{d}_x = \frac{1}{W_T} \sum_{i=1}^N w_i dx_i$ ,  $\bar{d}_y = \frac{1}{W_T} \sum_{i=1}^N w_i dy_i$ ,  $\bar{d}_z = \frac{1}{W_T} \sum_{i=1}^N w_i dz_i$  (8)

Then (5)-(7) become

$$(F_{z_A} \bar{d}_y - F_{y_A} \bar{d}_z) = M_{x_A} \quad (9)$$

$$(F_{x_A} \bar{d}_z - F_{z_A} \bar{d}_x) = M_{y_A} \quad (10)$$

$$(F_{y_A} \bar{d}_x - F_{x_A} \bar{d}_y) = M_{z_A} \quad (11)$$

The total work done by the forces and moments on A must equal the work on the  $i$  points (to which  $F_{x_A}$ , etc have been distributed) due to the  $F_{x_i}, F_{y_i}, F_{z_i}$



The work due to forces, moments on A is

$$W_A = F_{x_A} u_{x_A} + F_{y_A} u_{y_A} + F_{z_A} u_{z_A} + M_{x_A} \theta_{x_A} + M_{y_A} \theta_{y_A} + M_{z_A} \theta_{z_A} \quad (12)$$

Where  $u_{x_A}, \dots, \theta_{z_A}$  are the displ's and rotations at the ref. pt. A

The work done by the  $F_{x_i}, F_{y_i}, F_{z_i}$  is

$$W_N = \sum_{i=1}^N (F_{x_i} u_{x_i} + F_{y_i} u_{y_i} + F_{z_i} u_{z_i}) \quad (13)$$

Where  $u_{x_i}, u_{y_i}, u_{z_i}$  are the 3 translation displ components at i

Substitute (3) into (12) and (13) into (14) and equate  $W_A = W_N$

$$F_{x_A} u_{x_A} + F_{y_A} u_{y_A} + F_{z_A} u_{z_A} + (F_{z_A} \bar{d}_y - F_{y_A} \bar{d}_z) \theta_{x_A} + (F_{x_A} \bar{d}_z - F_{z_A} \bar{d}_x) \theta_{y_A} + (F_{y_A} \bar{d}_x - F_{x_A} \bar{d}_y) \theta_{z_A} = \sum_{i=1}^N \left( \frac{\omega_i}{W_T} F_{x_i} u_{x_i} + \frac{\omega_i}{W_T} F_{y_i} u_{y_i} + \frac{\omega_i}{W_T} F_{z_i} u_{z_i} \right)$$

Rearrange

$$\left( u_{x_A} + \bar{d}_z \theta_{y_A} - \bar{d}_y \theta_{z_A} - \sum_{i=1}^N \frac{\omega_i}{W_T} u_{x_i} \right) F_{x_A} + \left( u_{y_A} + \bar{d}_z \theta_{x_A} + \bar{d}_x \theta_{z_A} - \sum_{i=1}^N \frac{\omega_i}{W_T} u_{y_i} \right) F_{y_A} + \left( u_{z_A} + \bar{d}_y \theta_{x_A} - \bar{d}_x \theta_{y_A} - \sum_{i=1}^N \frac{\omega_i}{W_T} u_{z_i} \right) F_{z_A} = 0 \quad (14)$$

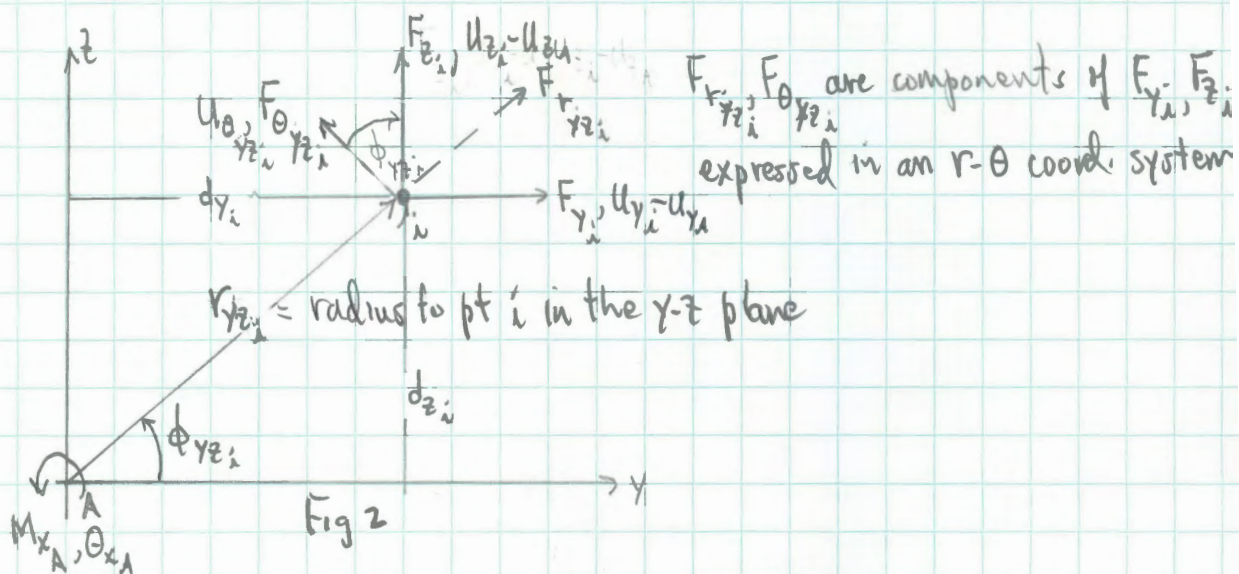
Since the  $F_{x_A}, F_{y_A}, F_{z_A}$  are independent and, in general, not zero, (14) requires

$$\begin{cases} u_{x_A} + \bar{d}_z \theta_{y_A} - \bar{d}_y \theta_{z_A} - \sum \frac{\omega_i}{W_T} u_{x_i} = 0 \\ u_{y_A} + \bar{d}_z \theta_{x_A} + \bar{d}_x \theta_{z_A} - \sum \frac{\omega_i}{W_T} u_{y_i} = 0 \\ u_{z_A} + \bar{d}_y \theta_{x_A} - \bar{d}_x \theta_{y_A} - \sum \frac{\omega_i}{W_T} u_{z_i} = 0 \end{cases} \quad (15)$$

There are 3 constraint eqns for the RBE3 (but we need others for  $\theta'_A$ , since (14) represents 3 eqns but with 6 unknowns

There are also eqns relating the  $F_{x_i}, F_{y_i}, F_{z_i}$  to the actual moments at A.

To see this look at the forces in 1 plane (say y-z):



Using  $F_{yzi}, F_{zzi}$  components we can take moments about A to get

$$\sum_{i=1}^N F_{\theta_{yzi}} r_{yzi} = M_{xA}$$

(16)

If we take

$$F_{\theta_{yzi}} = \frac{w_i r_{yzi}}{\sum w_i r_{yzi}^2} M_{xA}$$

(17)

and subst (17) into (16) we see that

$$\sum \left( \frac{w_i r_{yzi}}{\sum w_i r_{yzi}^2} \right) M_{xA} r_{yzi} = M_{xA}$$

or

$$\frac{\sum w_i r_{yzi}^2}{\sum w_i r_{yzi}^2} M_{xA} = M_{xA}$$

$$\therefore M_{xA} = M_{xA}$$

so that (17) is a valid expression for the  $F_{\theta_{yzi}}$  components of force



The work done by  $M_{x_A}$  must equal that due to all of the  $F_{\theta_i}$ :

$$\sum F_{\theta_{yz_i}} u_{\theta_{yz_i}} = M_{x_A} \theta_{x_A} \quad (18)$$

where  $u_{\theta_{yz_i}}$  is the tangential displ. at  $i$  in the  $y$ - $z$  plane

Subst (17) into (18)

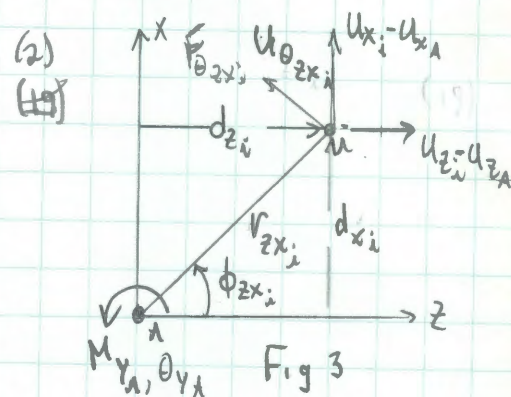
$$\sum \frac{\omega_i r_{zi}}{\sum \omega_i r_{zi}^2} M_{x_A} u_{\theta_{yz_i}} = M_{x_A} \theta_{x_A}$$

or

$$\theta_{x_A} = \frac{\sum \omega_i r_{zi} u_{\theta_{yz_i}}}{\sum \omega_i r_{zi}^2} \quad (19)$$

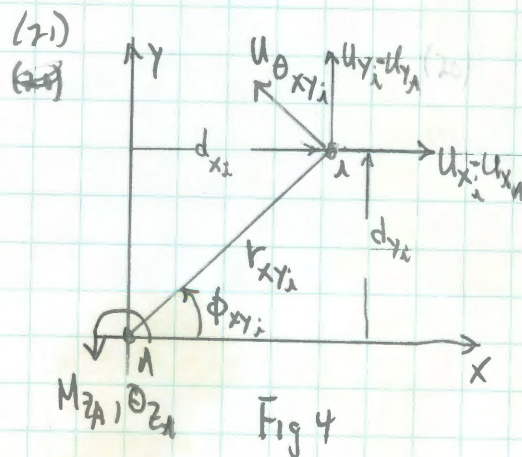
similarly for the  $x$ - $z$  and  $x$ - $y$  planes we can get

$$\theta_{y_A} = \frac{\sum \omega_i r_{zx_i} u_{\theta_{zx_i}}}{\sum \omega_i r_{zx_i}^2}$$



and

$$\theta_{z_A} = \frac{\sum \omega_i r_{xy_i} u_{\theta_{xy_i}}}{\sum \omega_i r_{xy_i}^2}$$



We can rewrite <sup>(19)-(21)</sup>~~(18)-(20)~~ in terms of  $u_{x_i}, u_{y_i}, u_{z_i}$  (rel to  $u_{x_A}, u_{y_A}, u_{z_A}$ ) instead of the  $u_\theta$ . From Fig 2:

$$\begin{aligned} u_{\theta_{yz_i}} &= (u_{z_i} - u_{z_A}) \cos \phi_{yz_i} - (u_{y_i} - u_{y_A}) \sin \phi_{yz_i} \\ &= (u_{z_i} - u_{z_A}) \frac{dy_i}{r_{yz_i}} - (u_{y_i} - u_{y_A}) \frac{dz_i}{r_{yz_i}} \end{aligned}$$

$$\therefore r_{yz_i} u_{\theta_{yz_i}} = (u_{z_i} - u_{z_A}) dy_i - (u_{y_i} - u_{y_A}) dz_i \quad (22)$$

From Fig 3

$$\begin{aligned} u_{\theta_{zx_i}} &= (u_{x_i} - u_{x_A}) \cos \phi_{zx_i} - (u_{z_i} - u_{z_A}) \sin \phi_{zx_i} \\ &= (u_{x_i} - u_{x_A}) \frac{dz_i}{r_{zx_i}} - (u_{z_i} - u_{z_A}) \frac{dx_i}{r_{zx_i}} \end{aligned}$$

$$\therefore r_{zx_i} u_{\theta_{zx_i}} = (u_{x_i} - u_{x_A}) dz_i - (u_{z_i} - u_{z_A}) dx_i \quad (23)$$

From Fig 4

$$\begin{aligned} u_{\theta_{xy_i}} &= (u_{y_i} - u_{y_A}) \cos \phi_{xy_i} - (u_{x_i} - u_{x_A}) \sin \phi_{xy_i} \\ &= (u_{y_i} - u_{y_A}) \frac{dx_i}{r_{xy_i}} - (u_{x_i} - u_{x_A}) \frac{dy_i}{r_{xy_i}} \end{aligned}$$

$$\therefore r_{xy_i} u_{\theta_{xy_i}} = (u_{y_i} - u_{y_A}) dx_i - (u_{x_i} - u_{x_A}) dy_i \quad (24)$$

where

$$r_{xy_i} = \sqrt{d_{x_i}^2 + d_{y_i}^2}$$

$$r_{yz_i} = \sqrt{d_{y_i}^2 + d_{z_i}^2}$$

$$r_{zx_i} = \sqrt{d_{z_i}^2 + d_{x_i}^2}$$

(25)



Define

$$\bar{e}_{xy} = \frac{1}{W_T} \sum_{i=1}^N \omega_i r_{xyi}^2 = \frac{1}{W_T} \sum_{i=1}^N \omega_i (d_{xi}^2 + d_{yi}^2)$$

$$\bar{e}_{yz} = \frac{1}{W_T} \sum_{i=1}^N \omega_i r_{yzi}^2 = \frac{1}{W_T} \sum_{i=1}^N \omega_i (d_{yi}^2 + d_{zi}^2)$$

$$\bar{e}_{zx} = \frac{1}{W_T} \sum_{i=1}^N \omega_i r_{zxi}^2 = \frac{1}{W_T} \sum_{i=1}^N \omega_i (d_{zi}^2 + d_{xi}^2)$$

(23) (24)

Using (9), (11), (22), (24), (25) in (19)

$$\Theta_{x_A} = \frac{1}{W_T \bar{e}_{yz}} \left[ \sum_{i=1}^N \omega_i (\check{u}_{zi} - \check{u}_{z_A}) \check{d}_{yi} - \sum_{i=1}^N \omega_i (\check{u}_{yi} - \check{u}_{y_A}) \check{d}_{zi} \right]$$

$$= \frac{1}{W_T \bar{e}_{yz}} \left[ - \left( \sum_{i=1}^N \omega_i \check{d}_{yi} \right) \check{u}_{z_A} + \left( \sum_{i=1}^N \omega_i \check{d}_{zi} \right) \check{u}_{y_A} + \sum_{i=1}^N \omega_i \check{d}_{yi} \check{u}_{zi} - \sum_{i=1}^N \omega_i \check{d}_{zi} \check{u}_{yi} \right]$$

$$\therefore \Theta_{x_A} = \frac{1}{W_T \bar{e}_{yz}} \left[ -W_T \bar{d}_y \check{u}_{z_A} + W_T \bar{d}_z \check{u}_{y_A} + \sum_{i=1}^N \omega_i \check{d}_{yi} \check{u}_{zi} - \sum_{i=1}^N \omega_i \check{d}_{zi} \check{u}_{yi} \right] \quad (25) (26)$$

Using (9), (11), (22), (24) in (26):

$$\Theta_{y_A} = \frac{1}{W_T \bar{e}_{zx}} \left[ \sum_{i=1}^N \omega_i (\check{u}_{xi} - \check{u}_{x_A}) \check{d}_{zi} - \sum_{i=1}^N \omega_i (\check{u}_{zi} - \check{u}_{z_A}) \check{d}_{xi} \right]$$

$$\therefore \Theta_{y_A} = \frac{1}{W_T \bar{e}_{zx}} \left[ -W_T \bar{d}_z \check{u}_{x_A} + W_T \bar{d}_x \check{u}_{z_A} + \sum_{i=1}^N \omega_i \check{d}_{zi} \check{u}_{xi} - \sum_{i=1}^N \omega_i \check{d}_{xi} \check{u}_{zi} \right] \quad (27) (28)$$

Using (9), (11), (24), (26) in (28)

$$\Theta_{z_A} = \frac{1}{W_T \bar{e}_{xy}} \left[ \sum_{i=1}^N \omega_i (\check{u}_{xi} - \check{u}_{x_A}) \check{d}_{yi} - \sum_{i=1}^N \omega_i (\check{u}_{yi} - \check{u}_{y_A}) \check{d}_{xi} \right]$$

$$\therefore \Theta_{z_A} = \frac{1}{W_T \bar{e}_{xy}} \left[ -W_T \bar{d}_x \check{u}_{y_A} + W_T \bar{d}_y \check{u}_{x_A} + \sum_{i=1}^N \omega_i \check{d}_{xi} \check{u}_{yi} - \sum_{i=1}^N \omega_i \check{d}_{yi} \check{u}_{xi} \right] \quad (29) (30)$$

The constraint eqns on  $u_{xA}, u_{yA}, u_{zA}, \theta_{xA}, \theta_{yA}, \theta_{zA}$  are summarized below:

$$u_{xA} + \bar{d}_z \theta_{yA} - \bar{d}_y \theta_{zA} - \frac{1}{W_T} \sum_{i=1}^N w_i u_{xi} = 0$$

$$u_{yA} - \bar{d}_z \theta_{xA} + \bar{d}_x \theta_{zA} - \frac{1}{W_T} \sum_{i=1}^N w_i u_{yi} = 0$$

$$u_{zA} + \bar{d}_y \theta_{xA} - \bar{d}_x \theta_{yA} - \frac{1}{W_T} \sum_{i=1}^N w_i u_{zi} = 0$$

$$\bar{e}_{yz} \theta_{xA} - \bar{d}_z u_{yA} + \bar{d}_y u_{zA} + \frac{1}{W_T} \sum_{i=1}^N w_i d_{zi} u_{yi} - \frac{1}{W_T} \sum_{i=1}^N w_i d_{yi} u_{zi} = 0$$

$$\bar{e}_{zx} \theta_{yA} + \bar{d}_z u_{xA} - \bar{d}_x u_{zA} - \frac{1}{W_T} \sum_{i=1}^N w_i d_{zi} u_{xi} + \frac{1}{W_T} \sum_{i=1}^N w_i d_{xi} u_{zi} = 0$$

$$\bar{e}_{xy} \theta_{zA} - \bar{d}_y u_{xA} + \bar{d}_x u_{yA} - \frac{1}{W_T} \sum_{i=1}^N w_i d_{yi} u_{xi} - \frac{1}{W_T} \sum_{i=1}^N w_i d_{xi} u_{yi} = 0$$

where

$$a) W_T = \sum_{i=1}^N w_i, \quad \bar{e}_{xy} = \frac{1}{W_T} \sum_{i=1}^N w_i (d_{xi}^2 + d_{yi}^2), \quad \bar{e}_{yz} = \frac{1}{W_T} \sum_{i=1}^N w_i (d_{yi}^2 + d_{zi}^2), \quad \bar{e}_{zx} = \frac{1}{W_T} \sum_{i=1}^N w_i (d_{zi}^2 + d_{xi}^2)$$

$$b) \bar{d}_x = \frac{1}{W_T} \sum_{i=1}^N w_i d_{xi}, \quad \bar{d}_y = \frac{1}{W_T} \sum_{i=1}^N w_i d_{yi}, \quad \bar{d}_z = \frac{1}{W_T} \sum_{i=1}^N w_i d_{zi}$$

c)  $d_{xi}, d_{yi}, d_{zi}$  are the coords of pt  $i$  relative to pt A in the global coord. sys. at A

d)  $u_{xi}, u_{yi}, u_{zi}$  are displ's at  $i$  transformed from global coord sys. at  $i$  to global coord sys at A

e)  $u_{xA}, u_{yA}, u_{zA}, \theta_{xA}, \theta_{yA}, \theta_{zA}$  are displ's at A in the global coord. sys. at A (obviously)

f) The weights  $w_i$  are assigned by the user (and are often 1.0)