Industrial Control



Nov 2021

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INSTRUCTIONS

Experimental Modeling and Control Design for Industrial Systems:

- Model Estimation: Applied Standard Methods to Estimate First and Second-Order Models with Time Delay for a Process System
 - \circ First-Order: Obtaining Time Constant Parameter τ through 4 Methods:
 - Entire Step Response
 - Tangent Intersection Method
 - 2 63% Final Value
 - Information from 2 Points (with Ranges of 0.2 and 0.63 of the Final Value)
 - O Plotted Process Response Curve (with Model Curves from the Four Methods)
 - Estimated Second-Order Model with Time Delay for 3 States:
 - 1. Critical Overdamped State
 - 2. Supercritical Damped State
 - 3. Subcritical Damped State
- **Controller Design**: Selecting a First-Order Approximation and Designing a Controller with these Objectives:
 - Attaining a Phase Margin (PM) of 45 degrees by Determining the Intersection
 Point of the Nyquist Curve with a Unit Circle
 - Achieving Gain Margin (GM) of 6 decibels by Pinpointing the Frequency Cutting 180 degrees in the System's Bode Diagram
 - Pursuing both Previous PM and GM Goals
- Experimental Z-N Method
 - Implemented Controller Design in Closed and Open-Loop using the Ziegler-Nichols Method

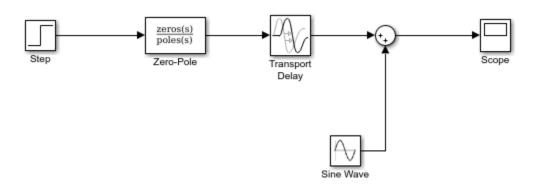
1) Model Estimation Expertise: I demonstrated proficiency in estimating first and second-order process models with time delay. This involved employing a spectrum of standard methods, including the analysis of entire step responses, tangent intersection methods with maximum slope, identifying the point where the output reaches 63% of the final value, and incorporating information from two strategically chosen points.

Model or Transformation Function in this Report:

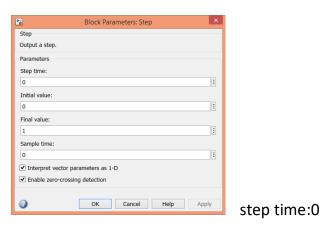
$$\frac{15}{(s+1)(s+2)(s+2)}e^{-0.6s}$$

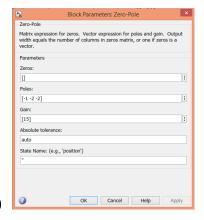
First order

Simulink:



Settings:

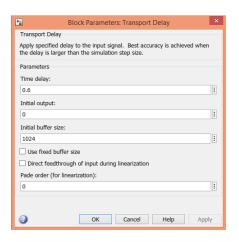




zeros : []

poles : [-1 -2 -2]

gain : [15]



Sine type determines the computational recrinque used. The parameters in the two types are related through:

Samples per period = 2*pi / (Frequency * Sample time)

Number of offset samples = Phase * Samples per period / (2*pi)

Use the sample-based sine type if numerical problems due to running for large times (e.g. overflow in absolute time) occur.

Parameters

Sine type: Time based

Time (t): Use simulation time

Amplitude:

0.015

Blas:

0

Frequency (rad/sec):

40

Phase (rad):

0

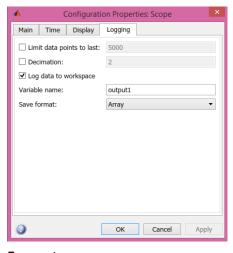
Samole time:

OK

Cancel Value Vielpidaws Apply

time delay = 0.6

amplitude: 0.015 Frequency: 40

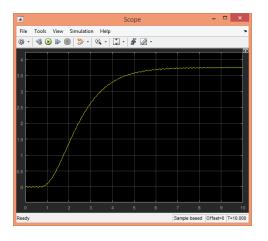


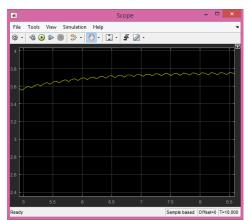
Variable name: output1

Format: array

Scope's input:

Closer look:



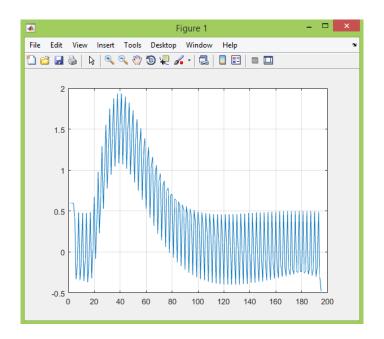


→ It can be seen that there are 196 data.

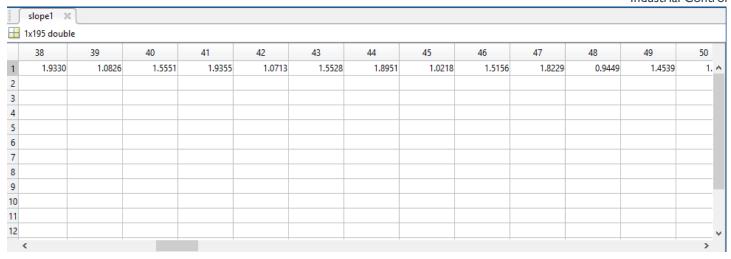
Now, we need to find the slope of the tangent line at each of these 196 points and select the maximum one and then plot it to find the τ_d Codes:



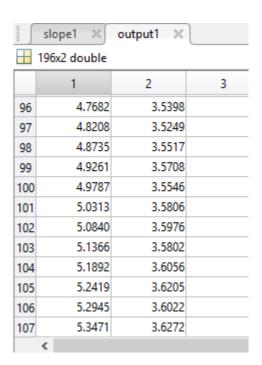
It can be seen: a1=196, b1=2



The maximum point is at point 40. Consequently, when we click on the slope, we have:

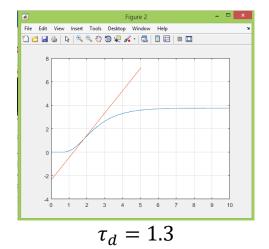


If we click on output1:



→ By the fifth second, it becomes 101

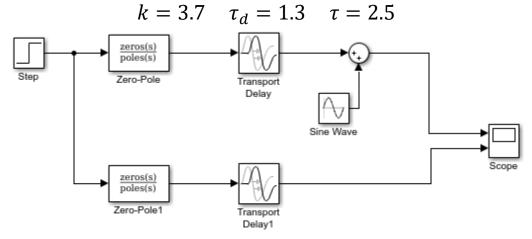
```
max_slope1=40;
mslope=1.9;
for j=1:101
    maxslopeline(j)=(mslope*output1(j,1))+(output1(max_slope1,2)-
mslope*output1(max_slope1,1));
end
figure
plot(output1(:,1),output1(:,2))
hold on
plot(output1(1:101,1),maxslopeline)
grid on
```



```
Taud=1.3;%timedelayestimation
sum1=0;
for j=1:a1-1
    area1=(output1(j+1,1)-output1(j,1))*(2*k-output1(j+1,2)-output1(j,2))*0.5;
    sum1=sum1+area1;
end
Tar=sum1/k;
```

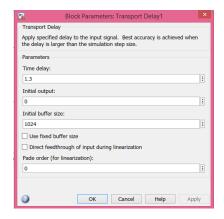
 $\tau = 2.4996 \sim 2.5$

Now we have all the necessary parameters to estimate our transformation function.

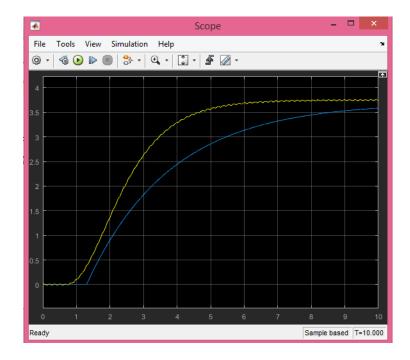


In which our new function lies:





Which we have in the final figure: (First method: Using the entire step response):



Now we need to obtain the transfer function using other methods:

Second method (Maximum slope intersect method) with K=3.7

Taud2=1.3;%ravesh2(shib-max)
Tau2=3.4-Taud2;

$$\tau_{d2} = 1.3$$

$$au_2 = 2.1$$

Third method (0.63 final value)

$$g = 0.63 * 3.7 ; %ravesh3-y(t0.63)=2.3-->t=3$$

Tau3 = 3-Taud;

$$\tau_3 = 3 - \tau_d = 1.7$$

Fourth method (2-point 0.63 and 0.28 final value)

```
g = 0.63 * 3.7; %ravesh4, t = 3

f = 0.28 * 3.7; %t(y=0.28k)=1.8

Tau4=1.5 * (3-1.8);

Taud4= (3-Tau4);
```

$$au_4 = 1.8 \\ au_{d4} = 1.2$$

Now we need to plot the process response curve along with the response curves of the models obtained using these 4 methods:

$$G_1(s) = \frac{3.7/2.5}{s + (1/2.5)}$$

$$G_2(s) = \frac{3.7/2.1}{s + (1/2.1)}$$

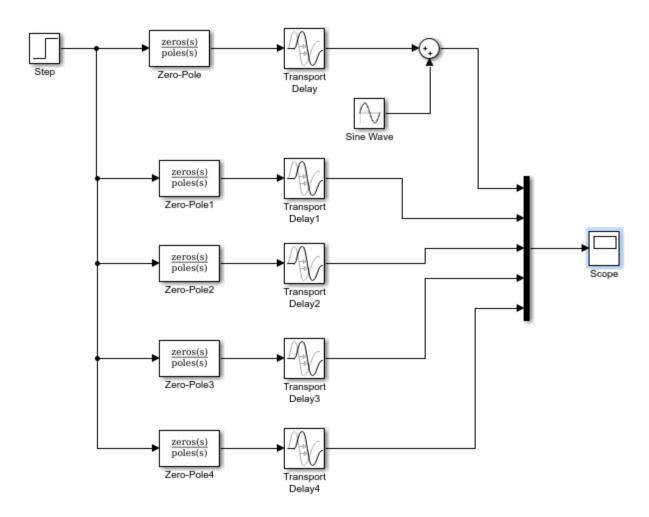
$$G_3(s) = \frac{3.7/1.7}{s + (1/1.7)}$$

$$\tau_{d1} = \tau_{d2} = \tau_{d3} = 1.3$$

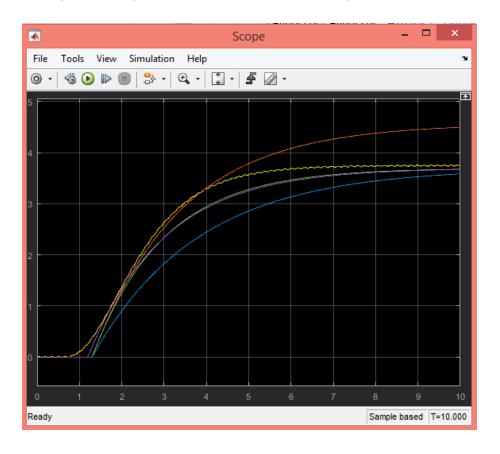
$$G_4(s) = \frac{3.7/1.8}{s + (1/1.8)}$$

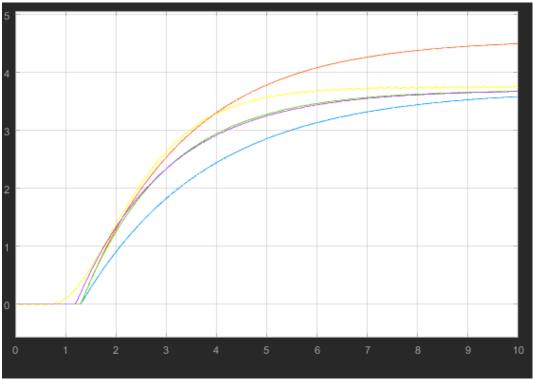
$$\tau_{d4} = 1.2$$

Symbolic representation of the obtained models:



The symbolic representation is as follows: Output1

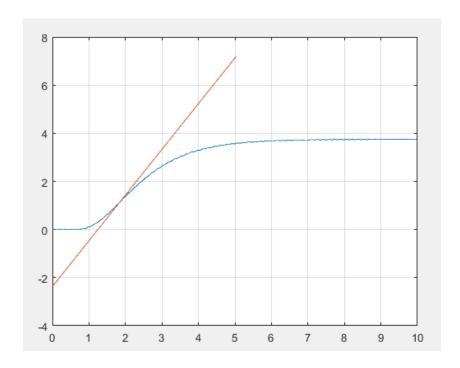




Second order:

First case: Critical Damped Second Order Model

Given the output shape and maximum slope from the previous section, we have:



We use point $y(\tau + \tau_d) = 0.264k$ to determine the parameters of this model

$$m = 0.264 * 3.7 ; %y(tau+taud) = y(0.9) = 1.8$$

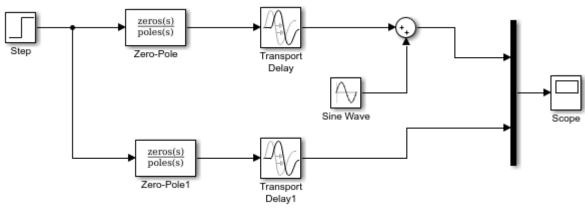
Tau = 1.8 - 1.3;

$$\tau = 0.5$$

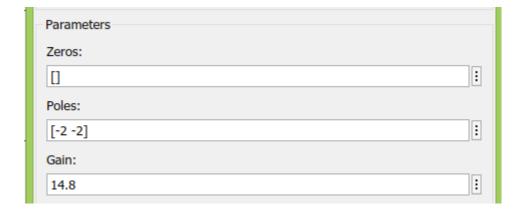
Consequently, the second-order model becomes as follows:

$$G(s) = \frac{k}{(\tau s + 1)^2} e^{-\tau_d s} = \frac{3.7}{0.25s^2 + s + 1} e^{-1.3s}$$

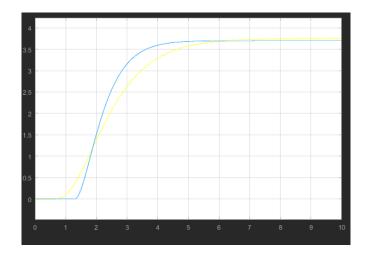
Scope's outputs:



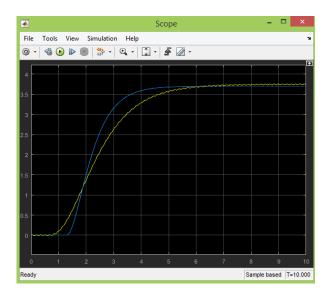
Settings for the first second-order model:



Output:



More specifically:



Second case: Overdamped Second Order Model

Second way
$$(T_{ar}= au_1+ au_2+ au_d)$$

$$G(s) = \frac{k}{(\tau_1 s + 1)(\tau_2 s + 1)} e^{-\tau_d s}$$

To obtain the parameters in this method, we need to derive the following formulas: Given the graph with maximum slope:

$$t_{m} = 5$$

$$t_{\chi} = 0.3$$

$$T_{ar} = sum1 = 2.5$$

$$\frac{k_{s}}{k} = \frac{1}{t_{m} - t_{\chi}} \to \frac{k_{s}}{3.7} = \frac{1}{4.7} \to k_{s} = 0.7$$

$$\lambda = \frac{(t_{m} - T_{ar})k_{s}}{k} = 0.4$$

$$\lambda = 0.4 = \chi e^{-\chi} \to \chi = \frac{\ln \eta}{\eta - 1} \to \eta \approx 0.6$$

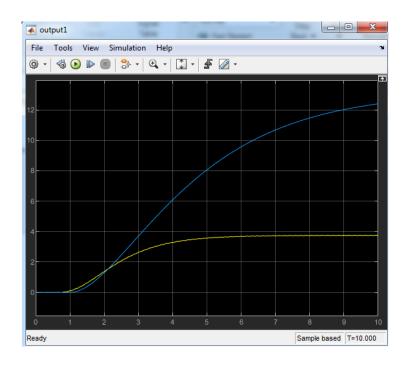
$$\frac{k_{s}}{k} = \frac{\eta^{\frac{1}{1 - \eta}}}{\tau_{1}} \to \tau_{1} = 1.4$$

$$\eta = \frac{\tau_{1}}{\tau_{2}} \to \tau_{2} = 2.3$$

$$\tau_{d} = T_{ar} - \tau_{1} - \tau_{2} = 1.02$$

$$3.7$$

$$G(s) = \frac{3.7}{(1.4s + 1)(2.3s + 1)} e^{-s}$$



Third method: Similar to the second method but without time delay Third case (Fourth method): Subcritical Second Order Model In this case, we have:

$$G(s) = \frac{k\omega^2}{s^2 + 2\xi\omega + \omega^2}$$

We must obtain $\omega_{\,\underline{\varrho}}\xi\,$:

$$\omega = \frac{2\pi}{T_P \sqrt{1 - \xi^2}}$$

$$\xi = \frac{1}{\sqrt{1 + (\frac{2\pi}{lnd})^2}}$$

Now we should get d:

$$d = \frac{e_2}{e_1}$$

Codes:

```
maxoutput1=0;
for i=1:a1-1
    if output1(i,2)> maxoutput1
        maxoutput1=output1(i,2);
        maxloc=i;
    end
end
minoutput1=maxoutput1;
for j=maxloc : a1-1
    if output1(j,2)<minoutput1</pre>
    minoutput1=output1(j,2);
    minloc=j;
    end
maxout2=minoutput1;
for c = minloc:a1-1
    if output1(c,2)>maxout2
        maxout2=output1(c,2);
        maxloc2=c;
    end
end
d=(\max 2-3.7)/(\max 1-3.7);
zita=1/sqrt(1+(2*pi/log(d))^2);
w = (2*pi)/((output1(maxloc2,1)-output1(maxloc,1))*sqrt(1-zita^2));
```

results:

Workspace		ூ
Name 📤	Value	
<u>₩</u> a1	196	
	-8.7521e-04	
<mark>⊞</mark> b1	2	
<mark>⊞</mark> c	195	
∐ d	0.7381	
<mark>⊞</mark> i	195	
⊞ j	195	
<mark>⊞</mark> k	3.7000	
max_slope1	40	
maxloc	194	
maxout2	3.7462	
maxoutput1	3.7627	
maxslopeline	1x101 double	
minloc	195	
minoutput1	3.7462	
mslope	1.9000	
output1	196x2 double	
slope1	1x195 double	
sum1	9.2484	
 Tar	2.4996	
Taud	1.3000	
tout	196x1 double	
w	119.5199	
zita	0.0483	

$$d = 0.7$$

 $\xi = 0.048 \cong 0.05$
 $\omega = 119$

As a result, our approximate model looks like this:

$$G(s) = \frac{52395}{s^2 + 11.9s + 14161}$$

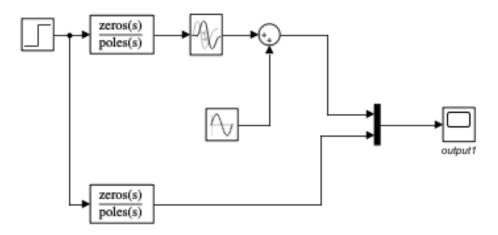
Maxloc2 error: Despite placing the program in MATLAB, it seems to output 'C' instead. Instead, the final answer appears to be incorrect and significantly larger than reasonable, so I calculated it manually and with an error of 'w'.

$$Tp \cong 140 \rightarrow \omega \cong 0.04$$

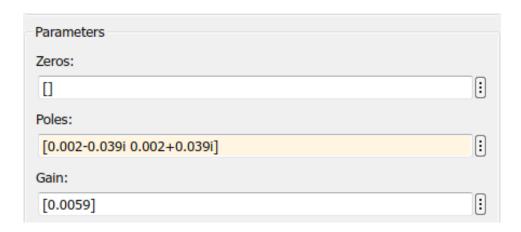
The final transformation function becomes as follows:

$$G(s) = \frac{0.0059}{s^2 + 0.004s + 0.0016}$$

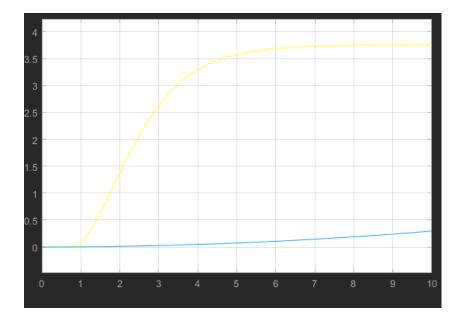
Simulink:



→ As observed, my Simulink program displays the transformation function section as above, and I need to enter the zeros and poles in the settings. However, the poles of my transformation function come out as complex, and after running the program, it gives an error:



And the output of the scope is:



- → Modeling a second-order system for a third-order function is better. Some of the issues with first-order modeling include the fact that at the start, the derivative change in the output response is not zero, and also that a first-order model cannot capture oscillations effectively.
- 2) Controller Design Mastery: With a keen focus on precision, I excelled in selecting a first-order approximation and adeptly designing controllers to achieve specific objectives. This encompassed attaining a Phase Margin (PM) of 45 degrees by determining the intersection point of the Nyquist curve with a unit circle, achieving Gain Margin (GM) of 6 decibels by pinpointing the frequency cutting 180 degrees in the system's Bode diagram, and simultaneously pursuing both PM and GM goals.

First, we use one of the first and second-order approximations obtained in the first project. Original system:

$$G(s) = \frac{15}{(s+1)(s+2)(s+2)}e^{-0.6s}$$

First-order approximation:

$$G_1(s) = \frac{3.7/2.5}{s + (1/2.5)}e^{-1.3s}$$

- 1) Obtain a Phase Margin (PM) of 45 degrees.
- 2) Obtain a Gain Margin (GM) of 6 dB.
- 3) Achieve both a PM of 45 degrees and a GM of 6 dB.

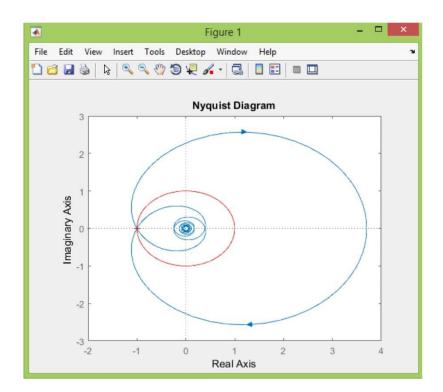
1-In the first scenario, where we aim to achieve a phase margin (PM) of 45 degrees, we first need to plot the Nyquist diagram of the system. We want to design a controller that can bring the PM of this system to 45 degrees and improve the system's performance. Therefore, we plot the unit circle on the

Nyquist diagram of the system to find the current PM. Then, we design the controller using the following mathematical relationships:

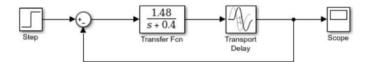
$$A: \qquad G(j \, \omega_0) = r_A e^{j \, \varphi_A} \\ B: \qquad GC(j \, \omega_0) = r_B e^{j \, \varphi_B} \\ Controller: \quad C(j \, \omega_0) = r_C e^{j \, \varphi_C}$$
 \Rightarrow
$$\begin{cases} r_C = \frac{r_B}{r_A} \\ \varphi_C = \varphi_B - \varphi_A \end{cases}$$

```
s=tf('s');
G=3.7/(2.5*s+1);
G1=G*exp(-1.3*s);
nyquist(G1);
r=1;
for j=0:360
    x(j+1)=r*cos((pi/180)*j);
    y(j+1)=r*sin((pi/180)*j);
end
hold on
plot(x,y,'r')
```

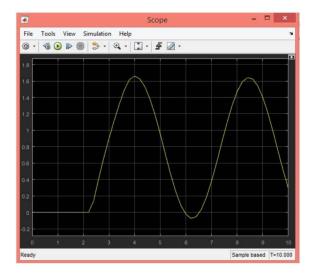
now we have:



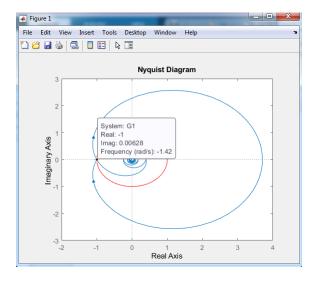
We need to find the intersection point of the curve with the unit circle, which is observed to be exactly at (x=-1, y=0). The reason for this is the instability of my system and its complete oscillation.



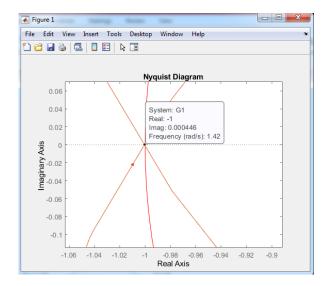
Output:



The intersection point of the Nyquist curve with the unit circle:

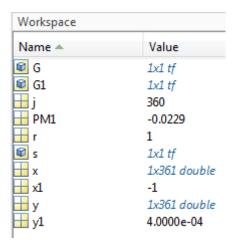


Closer look:



```
x1=-1;
y1=0.0004;
PM1=atan(y1/x1)*(180/pi);
```

So we have:



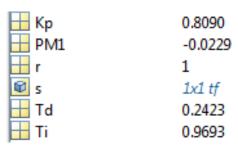
Now we will determine the PID coefficients based on the following formulas:

$$Kp = \frac{rB}{rA}\cos(\varphi B - \varphi A)$$

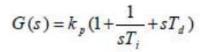
$$Ti = -\frac{1}{\tan(\varphi B - \varphi A)\,\omega 0}$$

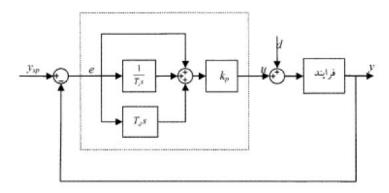
$$Td = 0.25Ti$$

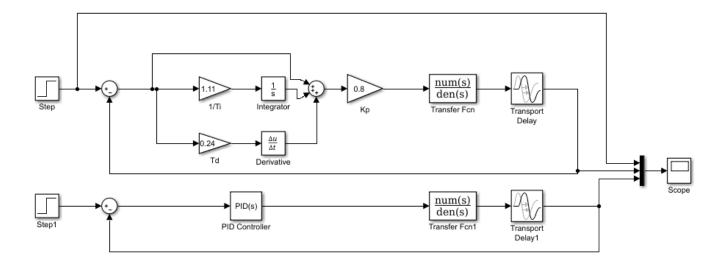
```
w0=1.42;
Kp=cos(36*(pi/180));
Ti=1/(w0*tan(36*(pi/180)));
Td=0.25*Ti;
```

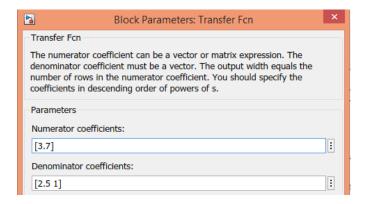


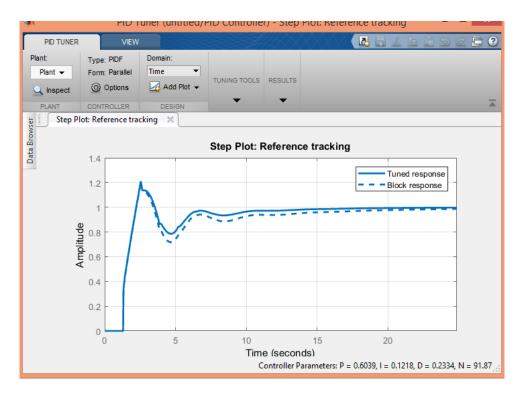
Now, in the Simulink section of MATLAB, the controller is added, and coefficient tuning is performed using the PID controller defined in MATLAB. We try to adjust the coefficients in such a way that the system response behaves similarly to the untuned system, ultimately resulting in responses as follows:











 Source:
 internal

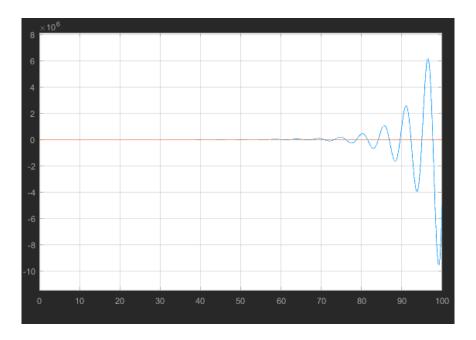
 Proportional (P):
 0.622972825824049

 Integral (I):
 0.090825884216773

 Derivative (D):
 0.222984633718735

 Filter coefficient (N):
 96.19538044861

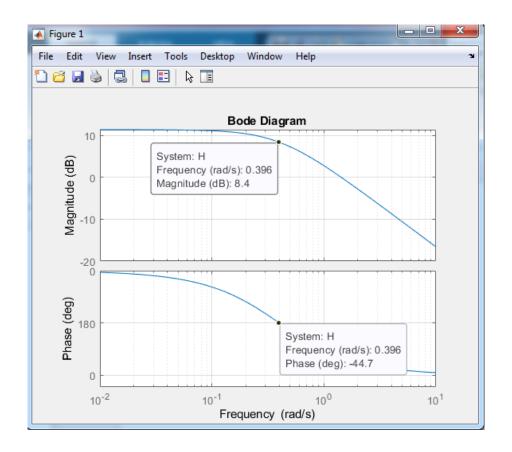
Output:



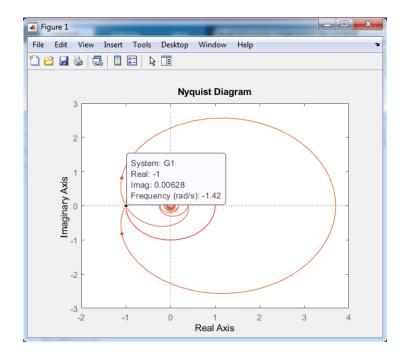
2: to increase the gain margin to 6 dB

In this scenario, we aim to adjust the system's gain margin to 6 dB. To do this, we first need to obtain the Bode plot of the system to find the frequency at which the phase margin crosses 180 degrees in the phase plot. Then, we evaluate the magnitude at that frequency in the magnitude plot and multiply it by a negative factor to find the current desired gain margin of the system:

```
clear all
close all
clc
s=tf('s');
H = (3.7/(2.5*s+1));
bode(H)
grid on
```



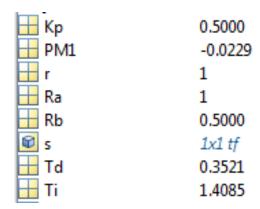
And also, we need to determine the intersection point with the real axis in the Nyquist diagram of the desired system. Additionally, we must find the intersection point with the real axis for the 6 dB magnitude in the controller we intend to design, which can be obtained using the following relationships:

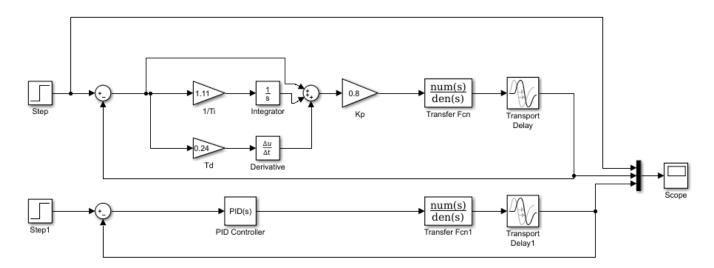


x1=-1; GM1=8.4; Ra=1;

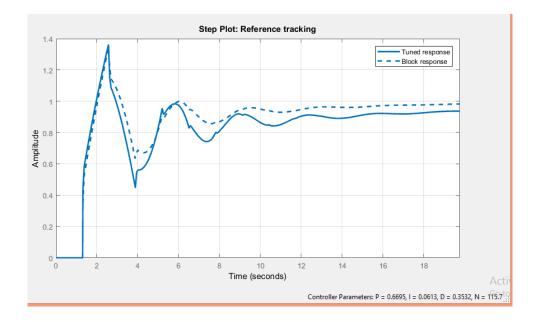
```
GM2=6;
Rb=0.5;
w0=1.42;
Kp=(Rb/Ra);
Td=1/(2*w0);
Ti=4*Td;
```

Result:





In the controller settings, we have:



 Source:
 internal

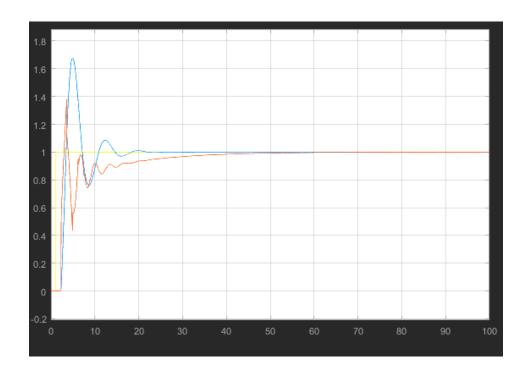
 Proportional (P):
 0.669543854076909

 Integral (I):
 0.0613034641044271

 Derivative (D):
 0.353196183364406

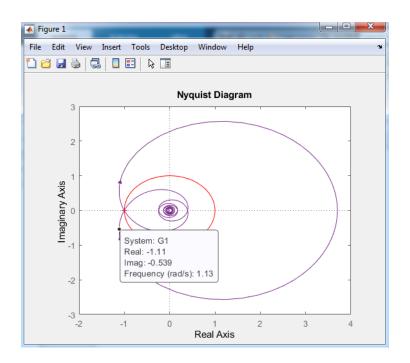
 Filter coefficient (N):
 115.652284687855

Output:



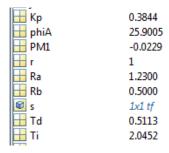
Third case: A scenario where both 45 degrees phase margin and 6 dB gain margin are achieved.

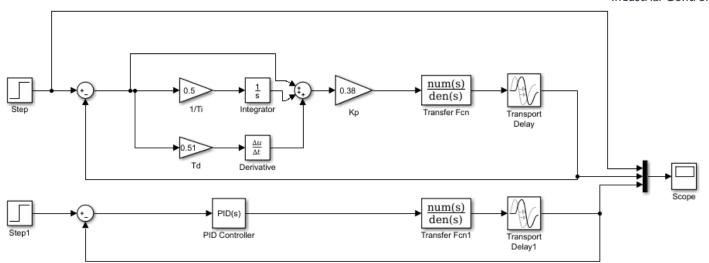
In the third part of this project, we need to adjust both GM = 6 dB and PM = 45 degrees. For this purpose, we selected a point between the intersection with the real axis and the unit circle. We obtained Phi A based on that point, and with the specifications of this point and the ratio Rb/Ra we have, we derive the controller coefficients. This point is defined as follows, and the corresponding code for this section is as follows:



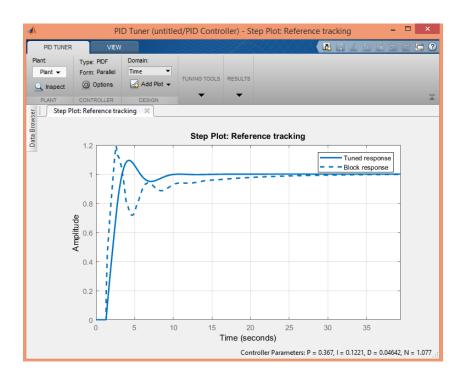
```
X2=-1.11;
Y2=-0.539;
W0=1.13;
phiA=atan(Y2/X2)*(180/pi);
Rb=0.5;
Ra=1.23;
Kp=(Rb/Ra)*cos(19*(pi/180));
Ti=1/(w0*tan(19*(pi/180)));
Td=0.25*Ti;
```

We have:





Settings:



 Source:
 internal

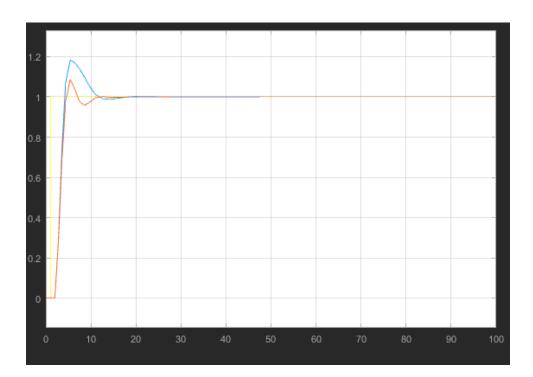
 Proportional (P):
 0.367046517528655

 Integral (I):
 0.122067040471561

 Derivative (D):
 0.0464171649393475

 Filter coefficient (N):
 1.07672704849522

Output:



3) **Implementation Prowess:** In the implementation phase, I successfully executed controller designs using the Ziegler-Nichols (ZN) method in both closed and open-loop configurations. This hands-on experience allowed for practical tuning and a comprehensive assessment of controller performance.

Conducting the experiment empirically:

In the next section, we select one of the controller design methods from the table and obtain the coefficients according to the table. We have chosen the Ziegler-Nichols open-loop method.

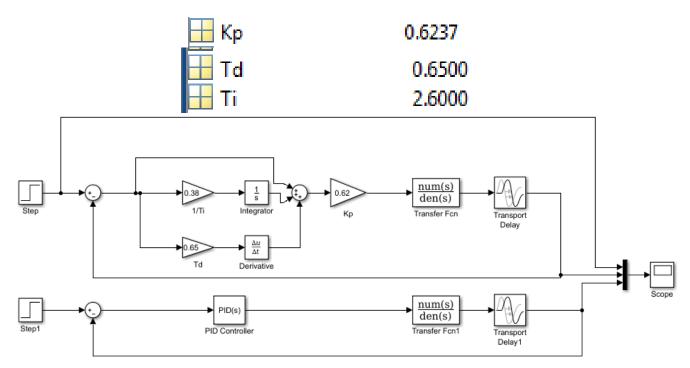
1) Open-loop method

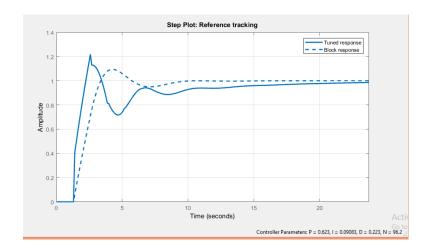
جدول ۵-۳: ضرایب کنترل کننده PID غیر تداخلی استاندارد بر اساس روش Z-N حلقه باز و مدل مرتبه یک با تاخیر

ZNsr	k_p	T ₁	T_d
P	$\frac{1}{k\alpha}$		
PI	$\frac{\cdot \cdot \cdot \cdot}{k\alpha}$	r.rr _d	
PID	$\frac{1.7}{k\alpha}$	YTd	\cdot . $\delta \tau_d$

Open-loop ZN PID

```
k=3.7;
tau=2.5;
taud=1.3;
alpha=taud/tau;
Kp=1.2/(k*alpha);
Ti=2*taud;
Td=0.5*taud;
```





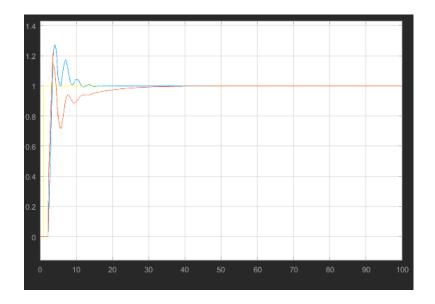
 Source:
 internal

 Proportional (P):
 0.622972825824049

 Integral (I):
 0.090825884216773

 Derivative (D):
 0.222984633718735

 Filter coefficient (N):
 96.19538044861



2) ZN Closed-loop method

جدول ۵-۲: ضرایب کنترل کننده PID غیرتداخلی استاندارد بر اساس روش Z-N حلقه بسته

ZNcl	k_p	T_i	T_{cl}
P	•.5k _u	-	-
PI	·. +0k,	$T_{\nu}/1/\Upsilon$	
PID	*.9k _u	T, / Y	T _n /A

Closed-loop zn

Ku=3.7;

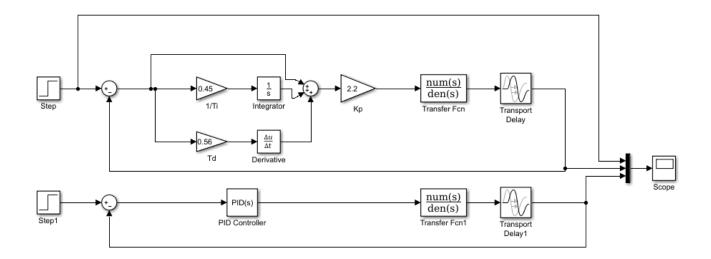
Tu=4.5;

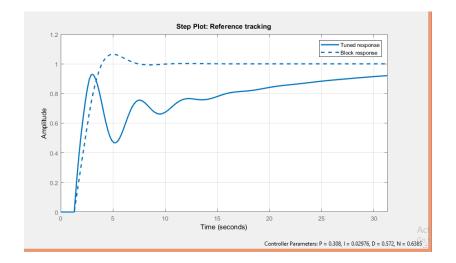
Kp=0.6*Ku;

Ti=Tu/2;

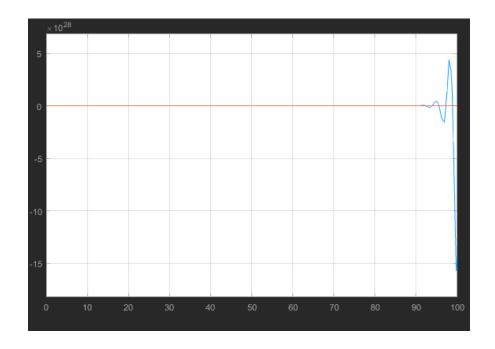
Td=Tu/8;







Source:	internal
Proportional (P):	0.307976842043044
Integral (I):	0.0297629463145374
Derivative (D):	0.572000990986316
Filter coefficient (N):	0.638518306862483



→ Based on the results obtained from all controllers designed in all sections with varying GM, PM, and both, it is estimated that the system response for the controller obtained by the Ziegler-Nichols open-loop method provides the best response among all. Therefore, the ZN method, as obtained empirically, is a good and acceptable method for designing system controllers.