Assignment 7

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Inputs

LFORM == 1 - TOTAL LAGRANGE, ELSE UPDATED LAGRANGE

IGRAD = 1 - PERFORM POST-PROCESSING

MODEL = 1 - NO ITERATION, ELSE ITERATIONS

Parameters	Value(s)
Simu	llation Parameters
ITERMAX	25
Epsilon	0.001
GAMA	0.5
LFORM	2
IGRAD	1
MODEL	2
Doma	in Inputs
XO	0
YO	0
x_length	10
y_length	1
thick	0.1
Mesh	n Inputs
NX	5
NY	1
NPE	8
NDF	2
DX	(x_length/NX)*ones(NX,1)
DY	(y_length/NX)*ones(NY,1)
NGPF	3
Materia	Properties
E	1.2E7
V	0.3
G	4.6154E6
Loading	condition
NLS	32
DPX	Zeros(NLS,1)
DPY	Zeros(NLS,1)
F	[0 0]
Essential Bour	ndary Conditions

NSPV	6
ISPV	[1,2; 1,1; 12,2; 12 ,1; 18,2; 18,1]
VSPV	[0 0 0 0 0 0]
	Natural Boundary Conditions
NSSV	20
ISSV	[19 2; 20 2; 21 2; 22 2; 23 2; 24 2; 25 2; 26 2; 27 2;
	28 2; 2 2; 3 2; 4 2; 5 2; 6 2; 7 2; 8 2; 9 2; 10 2; 11
	2]
VSSV	[0.066667 0.033333 0.066667 0.033333 0.066667
	0.033333 0.066667 0.033333 0.066667 0.016667
	0.066667 0.033333 0.066667 0.033333 0.066667
	0.033333 0.066667 0.033333 0.066667 0.016667]

Tabular Results

		3*3 Gau	uss rule		2*2 Gauss rule			
f0 = q0*h	x	у	u	v	x	у	u	v
50	9.9787	0.1145	0.0213	0.6145	9.9786	0.1163	0.0214	0.6163
30	10.0000	0.1163	0.0000	0.6163	10.0000	0.1181	0.0000	0.6181
100	9.9159	0.7181	0.0841	1.2181	9.9155	0.7225	0.0845	1.2225
100	9.9690	0.2686	0.0310	0.7686	9.9689	0.2707	0.0311	0.7707
150	9.8152	1.3010	0.1848	1.8010	9.8138	1.3091	0.1862	1.8091
130	9.8901	1.1644	0.1099	1.6644	9.8896	1.1698	0.1104	1.6698
200	9.6816	1.8554	0.3184	2.3554	9.6785	1.8688	0.3215	2.3688
200	9.8272	1.2658	0.1728	1.7658	9.8264	1.2719	0.1736	1.7719
250	9.5212	2.3758	0.4788	2.8758	9.5152	2.3960	0.4848	2.8960
230	9.6208	2.2693	0.3792	2.7693	9.6183	2.2818	0.3817	2.7818
300	9.3402	2.8593	0.6598	3.3593	9.3300	2.8874	0.6700	3.3874
300	9.5421	2.3359	0.4579	2.8359	9.5387	2.3493	0.4613	2.8493
350	9.1444	3.3047	0.8556	3.8047	9.1288	3.3417	0.8712	3.8417
550	9.2429	3.2102	0.7571	3.7102	9.2347	3.2350	0.7653	3.7350
400	8.9391	3.7125	1.0609	4.2125	8.9167	3.7587	1.0833	4.2587
400	9.1659	3.2796	0.8341	3.7796	9.1561	3.3052	0.8439	3.8052
450	8.7284	4.0846	1.2716	4.5846	8.6984	4.1398	1.3016	4.6398
430	8.8026	4.0388	1.1974	4.5388	8.7827	4.0812	1.2173	4.5812
500	8.5160	4.4229	1.4840	4.9229	8.4776	4.4871	1.5224	4.9871
300	8.7266	4.1054	1.2734	4.6054	8.7046	4.1480	1.2954	4.6480
EEO	8.3046	4.7302	1.6954	5.2302	8.2571	4.8028	1.7429	5.3028
550	8.3545	4.7083	1.6455	5.2083	8.3179	4.7692	1.6821	5.2692
600	8.0962	5.0093	1.9038	5.5093	8.0392	5.0897	1.9608	5.5897
000	8.2768	4.7839	1.7232	5.2839	8.2380	4.8448	1.7620	5.3448

650	7.8922	5.2628	2.1078	5.7628	7.8270	5.3480	2.1730	5.8480
030	7.9257	5.2508	2.0743	5.7508	7.8691	5.3291	2.1309	5.8291
700	7.6937	5.4933	2.3063	5.9933	7.6195	5.5843	2.3805	6.0843
700	7.8407	5.3368	2.1593	5.8368	7.7820	5.4146	2.2180	5.9146
750	7.5013	5.7031	2.4987	6.2031	7.4180	5.7995	2.5820	6.2995
730	7.5280	5.6913	2.4720	6.1913	7.4510	5.7835	2.5490	6.2835
800	7.3171	5.8922	2.6829	6.3922	7.2233	5.9956	2.7767	6.4956
800	7.4303	5.7881	2.5697	6.2881	7.3514	5.8796	2.6486	6.3796
850	7.1384	6.0668	2.8616	6.5668	7.0356	6.1746	2.9644	6.6746
630	7.1618	6.0550	2.8382	6.5550	7.0657	6.1576	2.9343	6.6576
900	6.9664	6.2268	3.0336	6.7268	6.8550	6.3383	3.1450	6.8383
	7.0488	6.1600	2.9512	6.6600	6.9505	6.2621	3.0495	6.7621

Table 1: Total Displacement of node 17 (5Q8)

f00*L			C	auchy Stre	ss	Piola-Kirchhoff Stress			
f0 = q0*h	х	у	СХХ	CYY	[CXY]	SXX	SYY	SXY	
50	0.4253	0.7861	0.7776	0.1819	0.0539	0.7885	0.1840	0.0487	
30	0.4254	0.7860	0.7929	0.1840	0.0545	0.8042	0.1862	0.0491	
100	0.4280	0.7835	1.5457	0.3579	0.1198	1.5894	0.3661	0.0994	
100	0.4282	0.7833	1.5779	0.3584	0.1220	1.6236	0.3670	0.1004	
150	0.4308	0.7810	2.2962	0.5268	0.1975	2.3950	0.5448	0.1521	
130	0.4310	0.7806	2.3476	0.5225	0.2022	2.4514	0.5416	0.1540	
200	0.4335	0.7783	3.0226	0.6880	0.2861	3.1978	0.7191	0.2070	
200	0.4339	0.7779	3.0961	0.6761	0.2943	3.2812	0.7093	0.2097	
250	0.4362	0.7757	3.7200	0.8411	0.3845	3.9918	0.8880	0.2640	
230	0.4367	0.7752	3.8185	0.8193	0.3970	4.1075	0.8699	0.2671	
300	0.4389	0.7731	4.3854	0.9863	0.4915	4.7721	1.0509	0.3230	
300	0.4396	0.7724	4.5117	0.9527	0.5092	4.9255	1.0232	0.3261	
350	0.4415	0.7706	5.0172	1.1237	0.6057	5.5354	1.2076	0.3838	
330	0.4424	0.7697	5.1731	1.0766	0.6293	5.7310	1.1693	0.3864	
400	0.4441	0.7680	5.6151	1.2537	0.7258	6.2795	1.3579	0.4462	
400	0.4452	0.7670	5.8027	1.1921	0.7562	6.5221	1.3086	0.4476	
450	0.4466	0.7655	6.1793	1.3767	0.8507	7.0025	1.5018	0.5098	
430	0.4478	0.7643	6.4004	1.2997	0.8886	7.2971	1.4414	0.5096	
500	0.4491	0.7630	6.7118	1.4933	0.9793	7.7050	1.6396	0.5746	
300	0.4505	0.7617	6.9668	1.4004	1.0254	8.0548	1.5682	0.5721	
550	0.4514	0.7605	7.2141	1.6040	1.1108	8.3870	1.7716	0.6402	
330	0.4531	0.7590	7.5032	1.4948	1.1658	8.7948	1.6894	0.6349	
600	0.4536	0.7580	7.6879	1.7093	1.2444	9.0489	1.8981	0.7064	
000	0.4556	0.7564	8.0111	1.5836	1.3088	9.5173	1.8055	0.6978	

650	0.4558	0.7556	8.1352	1.8096	1.3795	9.6916	2.0193	0.7731
030	0.4580	0.7540	8.4794	1.6631	1.4502	10.2038	1.9118	0.7597
700	0.4579	0.7532	8.5579	1.9055	1.5157	10.3161	2.1356	0.8402
700	0.4603	0.7514	8.9338	1.7422	1.5961	10.8897	2.0184	0.8225
750	0.4600	0.7508	8.9577	1.9973	1.6525	10.9233	2.2474	0.9074
750	0.4626	0.7489	9.3650	1.8176	1.7431	11.5602	2.1212	0.8851
800	0.4619	0.7486	9.3272	2.0816	1.7862	11.4997	2.3511	0.9736
800	0.4649	0.7464	9.7742	1.8897	1.8907	12.2153	2.2205	0.9474
850	0.4638	0.7463	9.6857	2.1660	1.9230	12.0742	2.4544	1.0408
630	0.4670	0.7440	10.1628	1.9588	2.0385	12.8557	2.3166	1.0094
000	0.4657	0.7440	10.0265	2.2473	2.0598	12.6351	2.5542	1.1080
900	0.4692	0.7416	10.5322	2.0253	2.1864	13.4821	2.4097	1.0710

Table 2: Stresses (*10^-5) evaluated at left-most gauss point nearest the top of element 1 (5Q8)

f0 = q0*h	х	у	u	v	х	у	СХХ	[CXY]
F0	9.978224	0.121824	0.021776	0.621824	0.4254	0.7860	0.775954	0.05491
50					0.4226	0.7887	0.787678	0.048532
100	9.914179	0.73232	0.085821	1.23232	0.4282	0.7833	1.542921	0.123887
100					0.4226	0.7887	1.59038	0.098568
150	9.811476	1.321405	0.188524	1.821405	0.4311	0.7806	2.292431	0.2065
150					0.4226	0.7887	2.399751	0.150361
200	9.675441	1.881042	0.324559	2.381042	0.434	0.778	3.017521	0.301793
200					0.4226	0.7887	3.208063	0.204049
250	9.512325	2.40571	0.487675	2.90571	0.4368	0.7753	3.713073	0.408444
230					0.4226	0.7887	4.008743	0.259663
300	9.328547	2.892354	0.671453	3.392354	0.4396	0.7726	4.375838	0.524941
300					0.4226	0.7887	4.79665	0.317144
350	9.13013	3.340026	0.86987	3.840026	0.4424	0.7699	5.00423	0.649737
530					0.4226	0.7887	5.568099	0.376365
400	8.922353	3.749389	1.077647	4.249389	0.445	0.7673	5.597979	0.78136
400					0.4226	0.7887	6.320713	0.437161
450	8.709509	4.122399	1.290491	4.622399	0.4476	0.7646	6.157186	0.918516
430					0.4226	0.7887	7.052462	0.499396
500	8.495278	4.461095	1.504722	4.961095	0.4501	0.7621	6.684028	1.059958
300					0.4226	0.7887	7.763859	0.562797
550	8.282279	4.768418	1.717721	5.268418	0.4525	0.7595	7.180008	1.20475
330					0.4226	0.7887	8.454772	0.627231
600	8.072524	5.047198	1.927476	5.547198	0.4549	0.757	7.646999	1.352064
000					0.4226	0.7887	9.125678	0.692534
650	7.867456	5.300213	2.132544	5.800213	0.4572	0.7545	8.086976	1.501223
650					0.4226	0.7887	9.777373	0.758562

700	7.668068	5.530089	2.331932	6.030089	0.4594	0.752	8.50188	1.651674
700					0.4226	0.7887	10.4108	0.825187
750	7.475004	5.739248	2.524996	6.239248	0.4615	0.7496	8.89357	1.802969
750					0.4226	0.7887	11.02697	0.892295
800	7.288644	5.929892	2.711356	6.429892	0.4635	0.7472	9.263788	1.954747
800					0.4226	0.7887	11.62692	0.959791
850	7.111413	6.101288	2.888587	6.601288	0.4655	0.7449	9.605038	2.102051
630					0.4226	0.7887	12.19591	1.026032
900	6.939197	6.260413	3.060803	6.760413	0.4674	0.7426	9.936603	2.253453
900					0.4226	0.7887	12.76516	1.093911
950	6.773775	6.406435	3.226225	6.906435	0.4693	0.7403	10.2515	2.404826
930					0.4226	0.7887	13.32174	1.162021
1000	6.615008	6.540705	3.384992	7.040705	0.4711	0.738	10.55069	2.555848
1000					0.4226	0.7887	13.86603	1.230252
1050	6.462707	6.664441	3.537293	7.164441	0.4728	0.7357	10.83527	2.706392
1050					0.4226	0.7887	14.39883	1.298558
1100	6.316653	6.778721	3.683347	7.278721	0.4745	0.7334	11.10626	2.85636
1100					0.4226	0.7887	14.92089	1.366898

Table 3: Total displacement of node 22 and stresses (*10^-5) (5Q9)

Plots

Node 17 Displacement

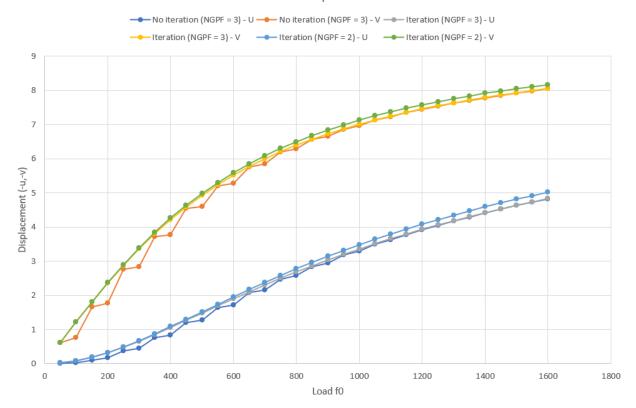


Figure 1: Node 17 displacements (-u,-v) vs load f0 = q0*h

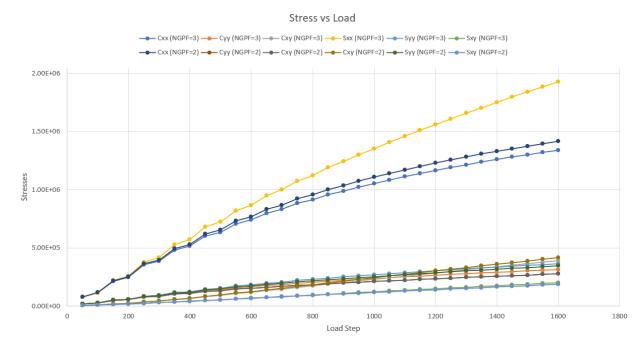


Figure 2: Stresses vs load f0 = q0*h

```
%% Pseudo Code
% Name: Manay Kothari
%UIN: 133008008
%% Assignment 7
Inputs
Call MeshR function to generate mesh
Call PlotMesh function to plot the mesh
Define DOF NOD
Calculate the components of C
Initialize GCU, GPU, and GLS
for NL = 1:NLS % Load Step for loop
    Calculate the load for that particular step
    while iter<=ITERMAX && convergence == 0</pre>
        Initialize GLK and GLF
        for N = 1:NEM
            Calculate ELS and ELXY for every element
            Call ELEMATRICS2D function to calculate ELK and ELF
            Assemble the ELK and ELF matrix into GLK and GLF
        end
        Call CONTBCS function to impose boundary conditions
        Calculate iterative solution DELU
        % Note, the formulation of this method inherents NI✓
```

```
algorithm.
        % That means, this method already imposes NI
       for I = 1:NNM
           Update the solution vector and nodal coordinates
       end
       if MODEL ~= 1
           Perform iterative method to solve for the solution
       else
           Do not perform iteration and calculate solution ✓
based on 1st iteration itself
       end
   end % End of iterative loop
    %% Post processing of results
    if IGRAD ~= 0 %IGRAD = 0 means don't calculate stresses
       for I = 1:NPE
           Calculate ELXY and ELS from updated GLXY and GLS /
for the required element
           Call Stress2D function to get all the required✓
stresses and strains.
       end
   end
end % End of load step loop
용 🗸
function [ELK,ELF] = ELEMATRICS2D(NDF,NPE,ELXY,ELS,NGPF,C,√
thick, F, LFORM)
```

```
Initialize ELK and ELF
for NI = 1:NGPF
    for NJ = 1:NGPF % Perform full integration
        Call INTERPLN2D function for SFL, GDSFL, and JAC
        Calculate Green Strain and 2nd Piola-Kirchhoff stress✓
for Total lagrange
        OR
        Calculate Euler Strain and Cauchy stress for Updated√
Lagrange
        for I = 1:NPE
            Calculate components of ELF matrix
            for J = 1:NPE
                calculate components of ELK matrix
            end
       end
   end
end
end
응 🗸
----- %
function SS = STRESS2D(NDF, ELXY, ELS, LGP, NPE, C)
for NI = 1:LGP
    for NJ = 1:LGP
        Calculate XC, YC, U1X, U1Y, V1X, V1Y for the particular ✓
gauss point
        Calculate Euler strain and Cauchy Stress Tensor
```

	Update ELXY
tensor	Calculate Green Strain and 2nd Piola Kirchhoff Stress✓
end end	
Export end	the required stress tensor and strains
	₋