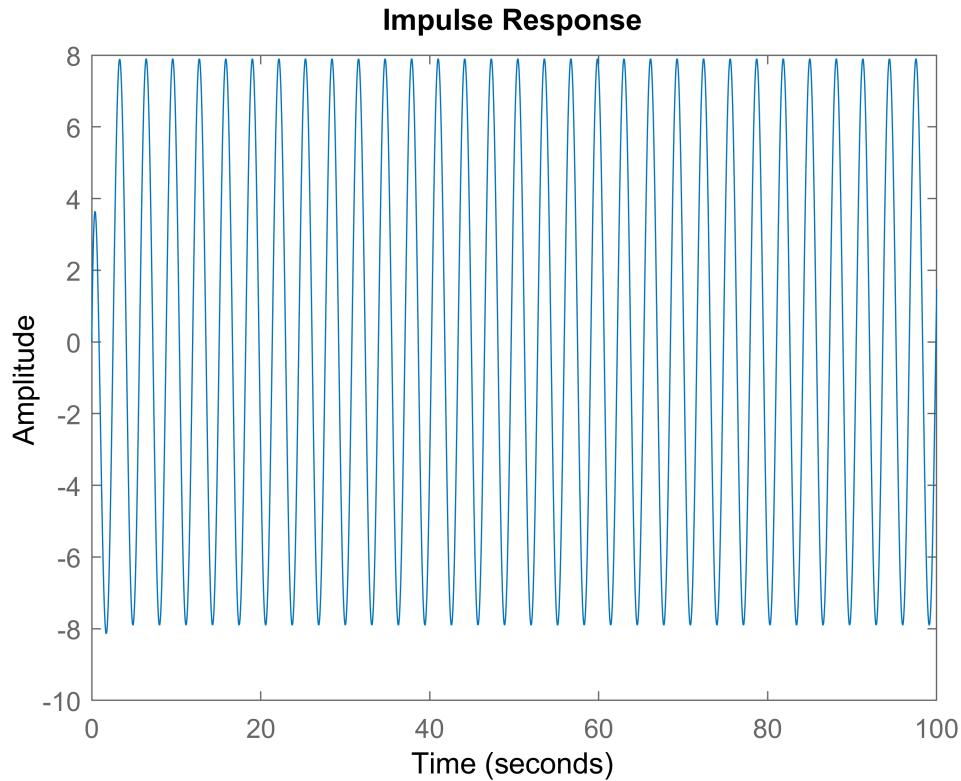


Qs1

$$M(s) = \frac{20(s-1)}{(s+2)(s^2+4)}$$

Poles at $-2, \pm 2j$: - 2 poles on imaginary axis and one pole in LHP, so it is marginally stable, as seen from the impulse response of the transfer function.

```
tf1=zpk(1,[-2,2i,-2i],20);  
impulse(tf1,100)
```



QS. 2

Characteristic equation,

$$2s^4 + s^3 + 3s^2 + 5s + 10 = 0$$

```
y1=rhc([2,1,3,5,10],1);
```

Routh-Hurwitz Table:

rhTable = 5x3

2.0000	3.0000	10.0000
1.0000	5.0000	0
-7.0000	10.0000	0
6.4286	0	0

```
10.0000      0      0
```

```
if y1==0
    fprintf("System is unstable \n")
else
    fprintf("System is stable \n")
end
```

```
System is unstable
```

QS. 3

Characteristic equation,

$$s^4 + s^3 + 2s^2 + 2s + 3 = 0$$

```
y2=rhc([1,1,2,2,3],1);
```

```
Routh-Hurwitz Table:
rhTable = 5x3
10^4 x
0.0001    0.0002    0.0003
0.0001    0.0002    0
0.0000    0.0003    0
-2.9998    0        0
0.0003    0        0
```

```
if y2==0
    fprintf("System is unstable \n")
else
    fprintf("System is stable \n")
end
```

```
System is unstable
```

QS. 4

Characteristic equation,

$$s^5 + 4s^4 + 8s^3 + 8s^2 + 7s + 4 = 0$$

```
y3=rhc([1,4,8,8,7,4],1);
```

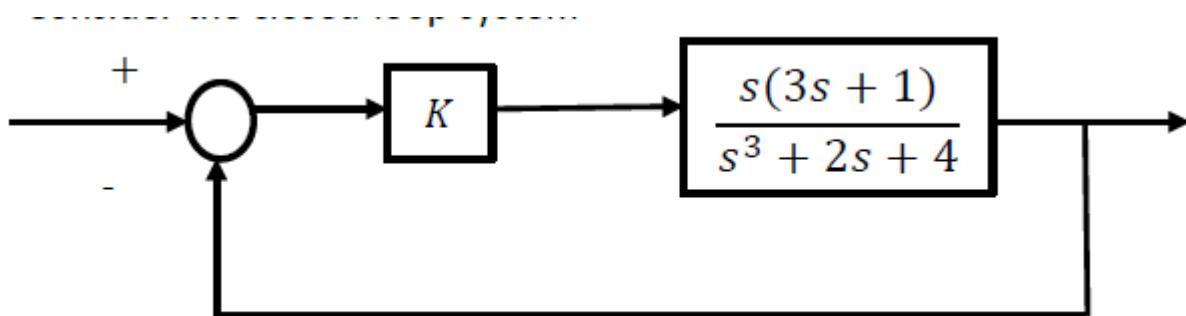
```
Routh-Hurwitz Table:
rhTable = 6x3
1.0000    8.0000    7.0000
4.0000    8.0000    4.0000
6.0000    6.0000    0
4.0000    4.0000    0
0.0001    0        0
4.0000    0        0
```

```
if y3==0
    fprintf("System is unstable \n")
else
    fprintf("System is stable \n")
```

end

System is stable

QS. 5

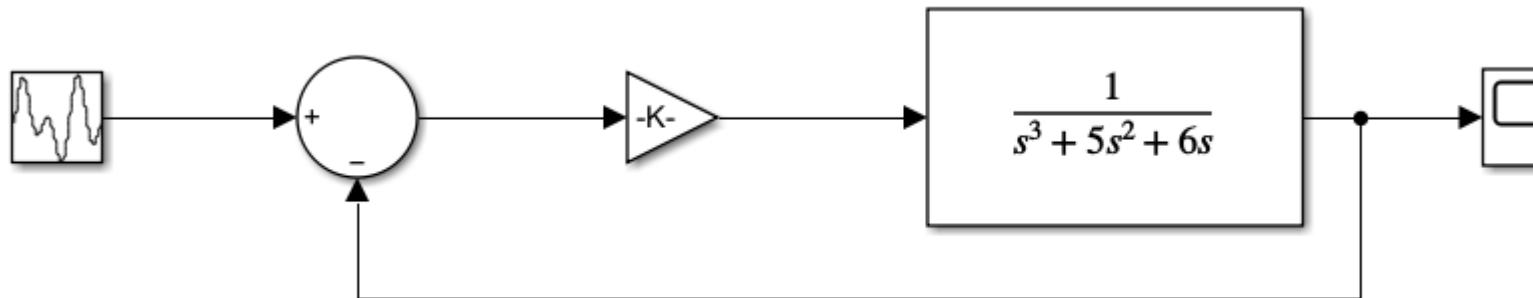
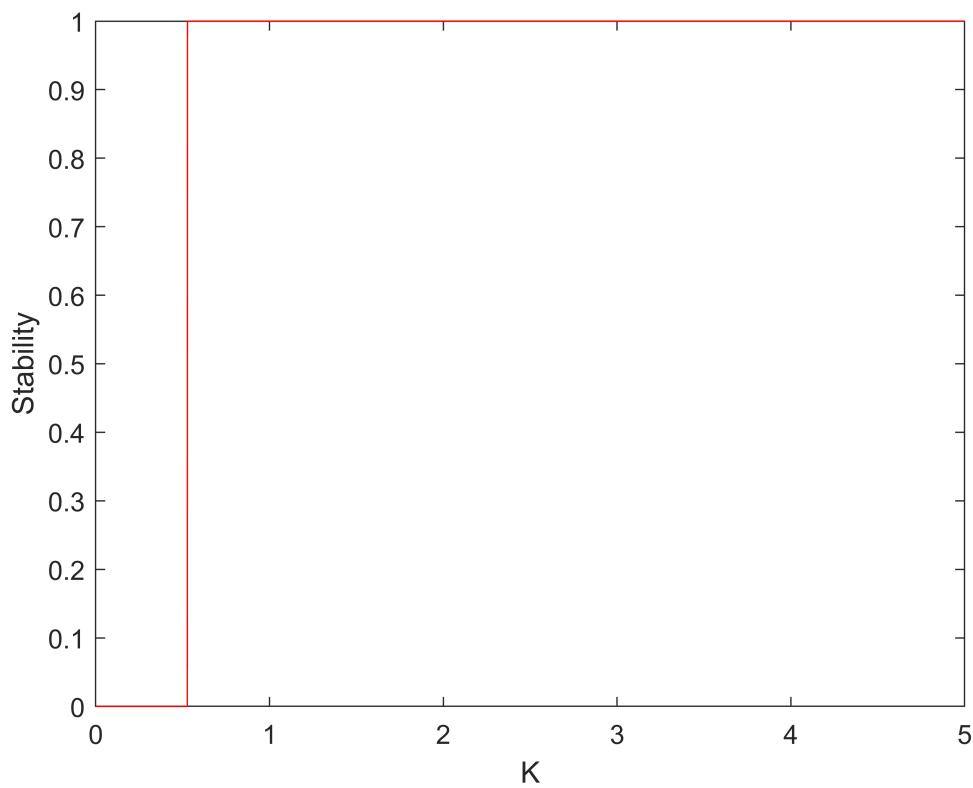


For $K \geq -1 + \sqrt{\frac{7}{3}}$ the system is stable

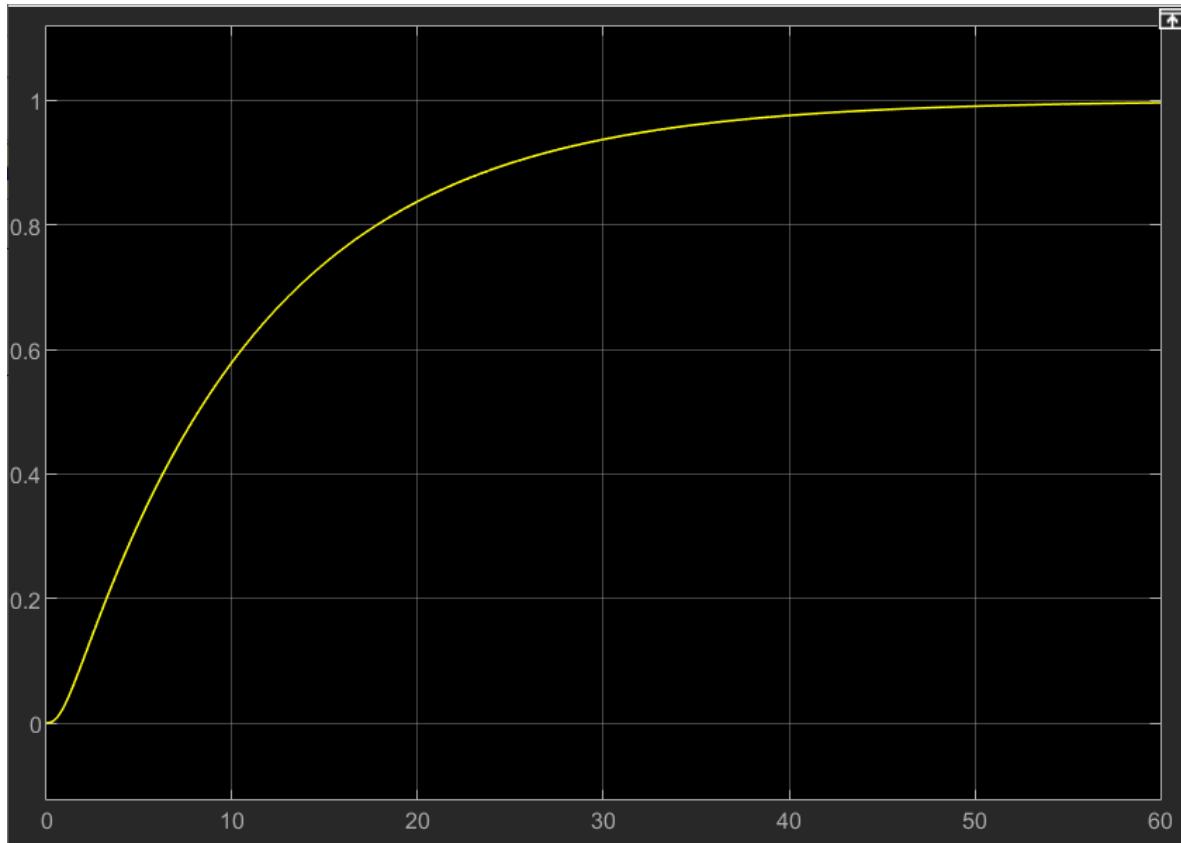
For $K = -1 + \sqrt{\frac{7}{3}}$ it is marginally stable.

```
k=1e-3:1e-3:5;
y=zeros(size(k));

for i =1:length(k)
    y(i)=rhc([1,3*k(i),k(i)+2,4],0);
end
figure;
plot(k,y, "r-")
xlabel("K");
ylabel("Stability");
```



stable response on $k > 0.527$



QS. 6

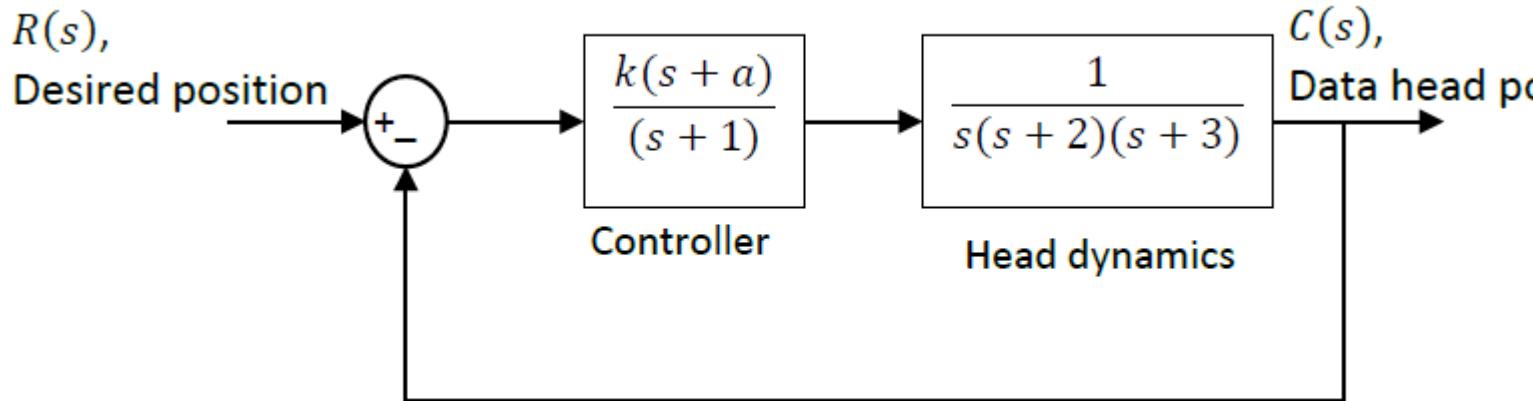


Fig. 1: Disk-storage data-head positioning feedback system

$0 < K \leq 60$; $0 \leq a \leq \frac{10}{k} + \frac{3}{2} - \frac{k}{36}$; The system is stable

$K = 60$; $a = 0$; The system is marginally stable with pole at 0

```

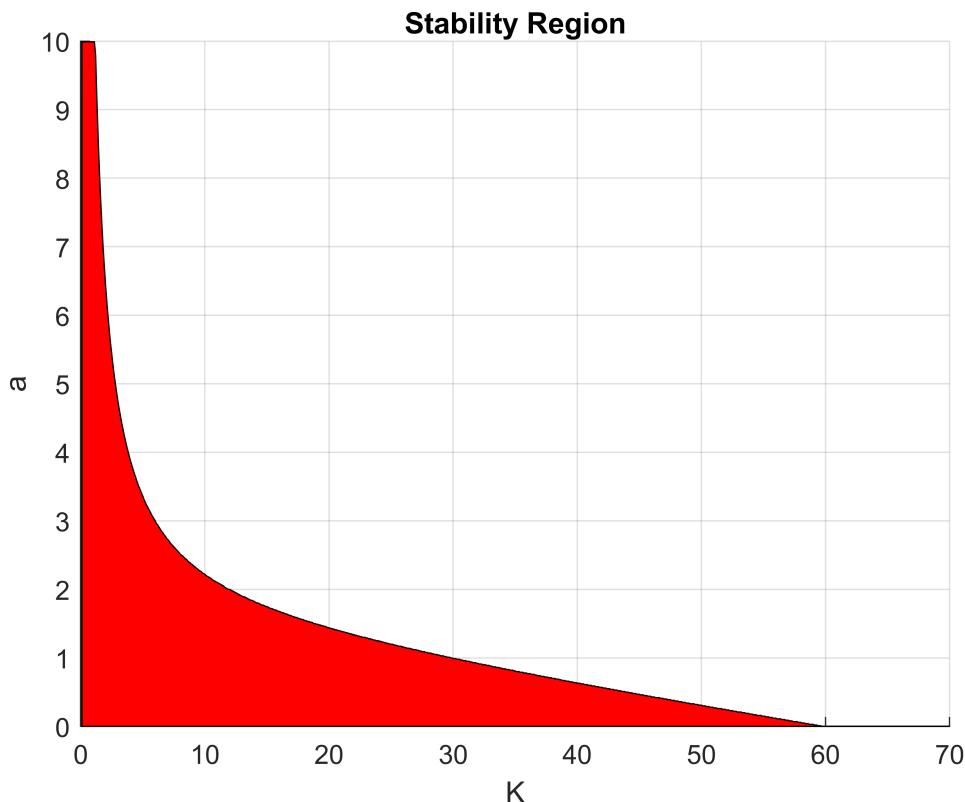
k=0.1:0.1:70;
a=0:0.01:10;
y=zeros(size(k));
  
```

```

for i=1:length(k)
    for j=1:length(a)-1
        c=rhc([1,6,11,6+k(i),k(i)*a(j)],0);
        b=rhc([1,6,11,6+k(i),k(i)*a(j+1)],0);
        if(c~=b)|| (b==1 && j==length(a)-1)
            y(i)=a(j);
        end
    end
end

figure;
patch([k,fliplr(k)],[y,zeros(size(y))], 'r')
xlabel("K");
ylabel("a");
title("Stability Region");
grid on

```



```

A=[0.4,0.8,1.0,1.2,1.6,2.0];
CM= hsv(6);
figure;

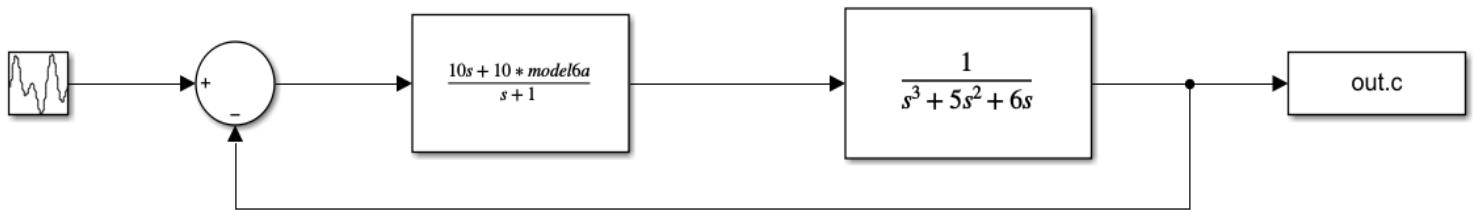
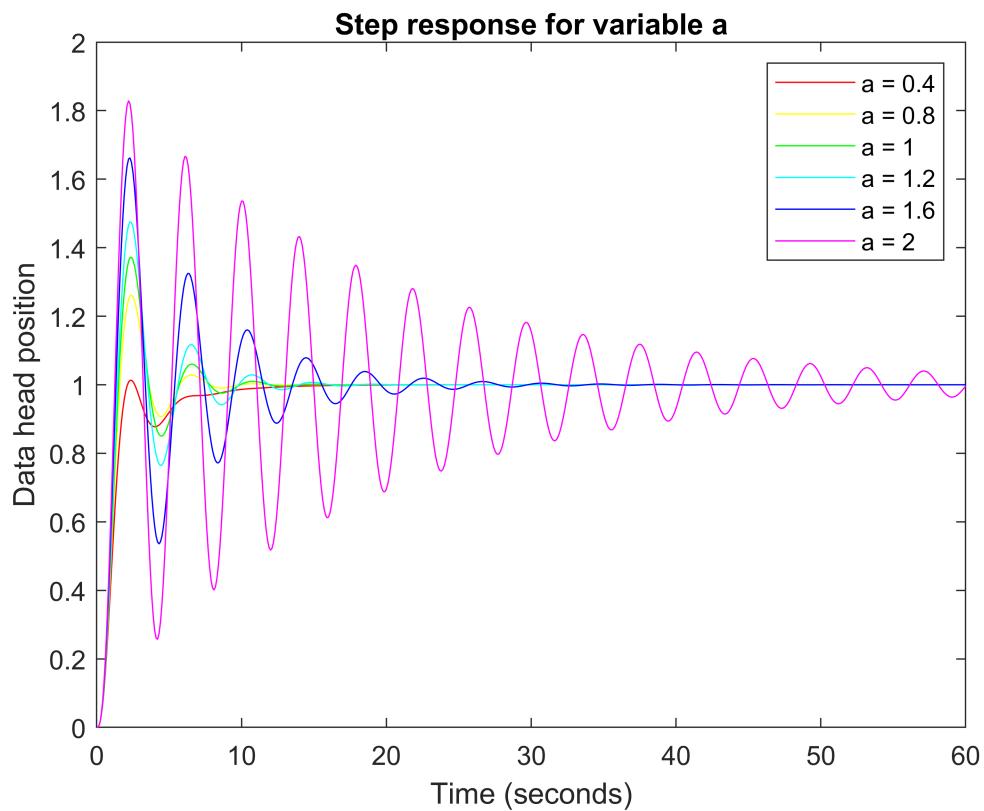
for i=1:6
    model6a=A(i);
    out=sim("model_6.slx");
    plot(out.c,"Color",CM(i,:),"DisplayName",strcat("a = ",num2str(model6a)));
    hold on
end

```

```

hold off
legend
ylabel("Data head position");
title("Step response for variable a");

```



Reducing the value of a , reduces the overshoot and oscillation

For higher value of a , the system reaches the steady state value faster

but it oscillates for long time

QS.7

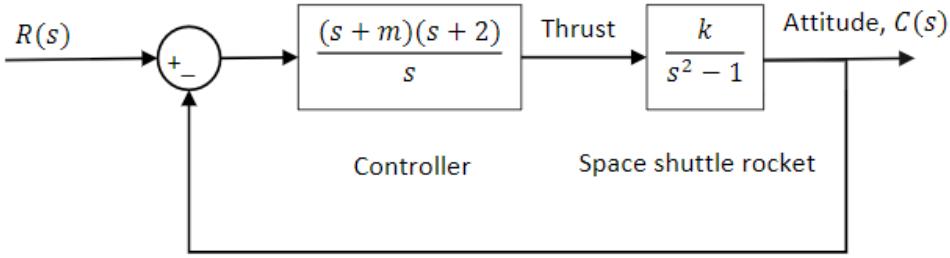


Fig. 2: Attitude control system of a space shuttle

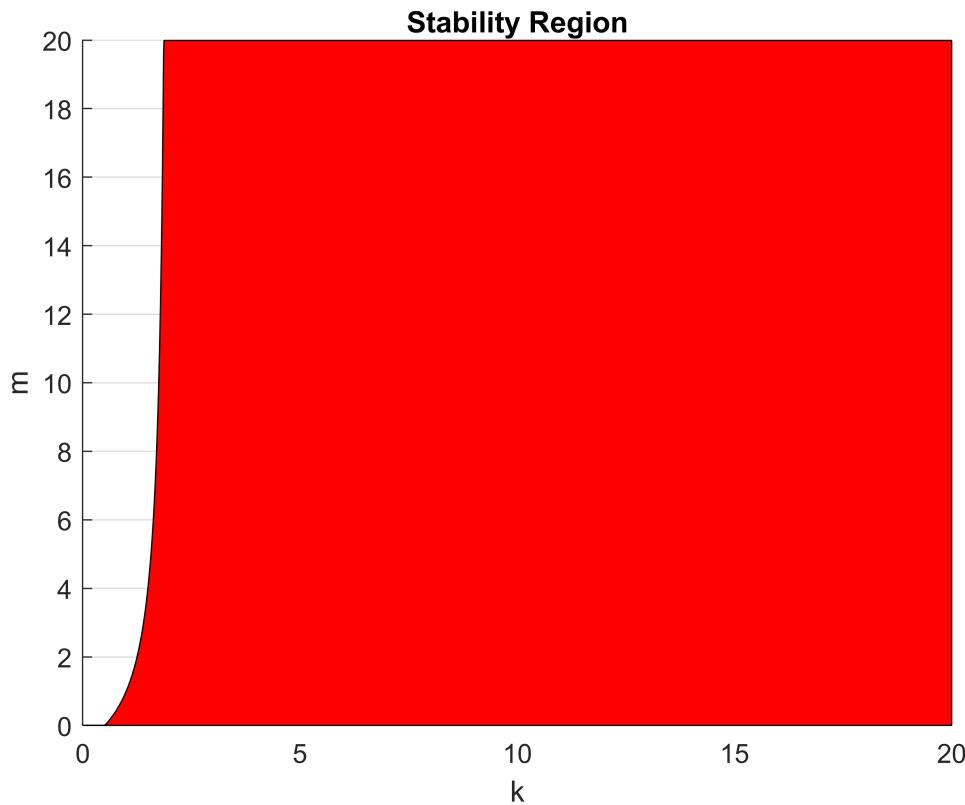
```

k=0.01:0.01:20;
m=0:0.01:20;
y=zeros(size(k));

for i=1:length(k)
    for j=1:length(m)-1
        a=rhc([1,k(i),k(i)*(m(j)+2)-1,2*k(i)*m(j)],0);
        b=rhc([1,k(i),k(i)*(m(j+1)+2)-1,2*k(i)*m(j+1)],0);
        if(a~=b)|| (b==1 && j==length(m)-1)
            y(i)=m(j);
        end
    end
end

figure;
patch([k,fliplr(k)],[y,zeros(size(y))], 'r')
xlabel("k");
ylabel("m");
title("Stability Region");
grid on

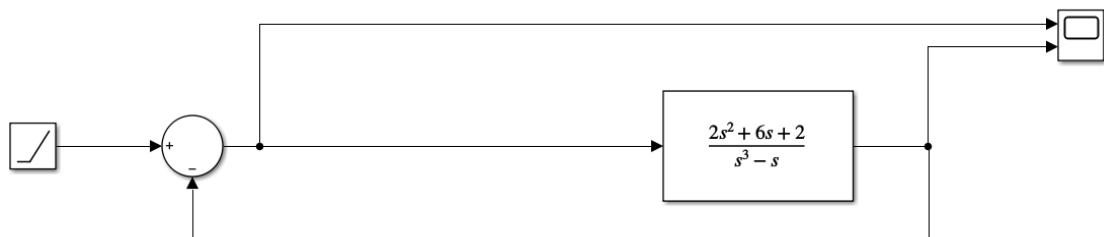
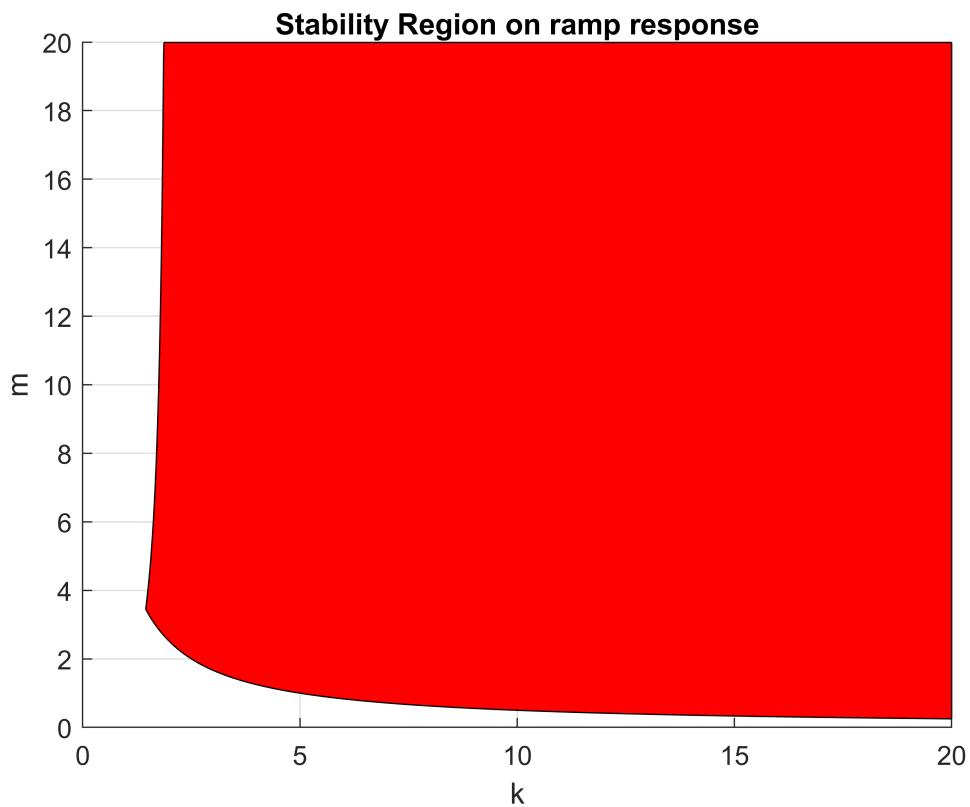
```

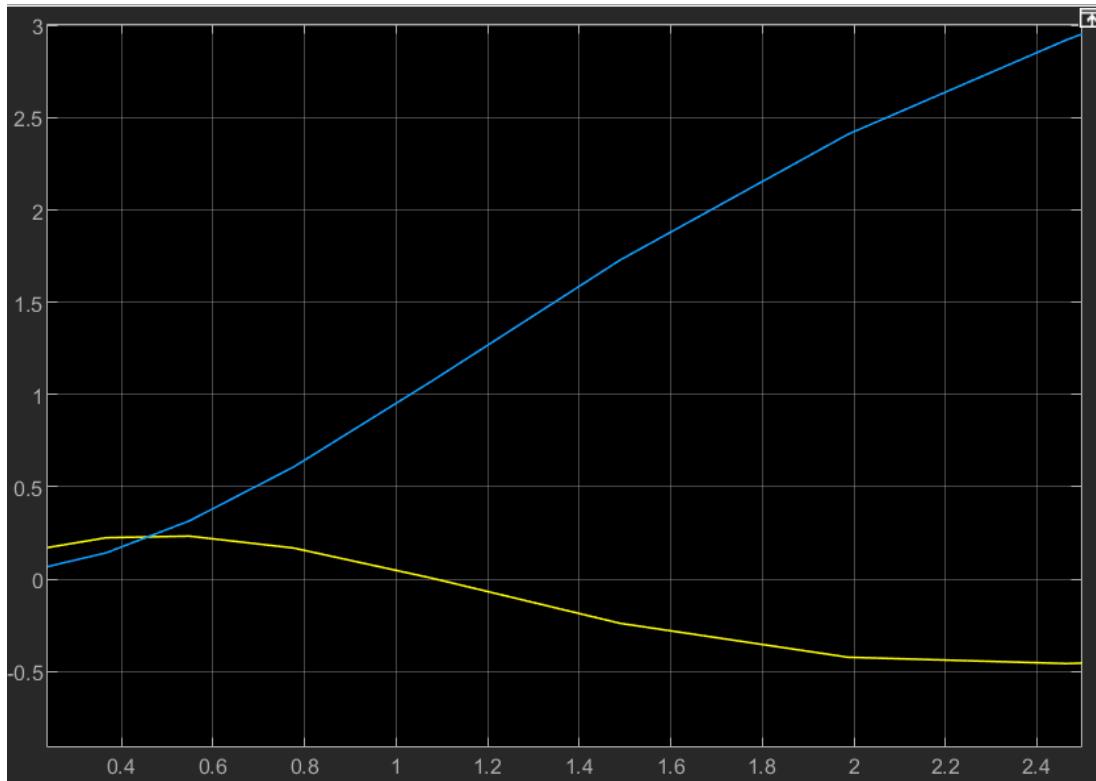


Stability region:

```
y1=5./k;
Li=(y>=y1);
figure;

patch([k(Li),fliplr(k(Li))],[y(Li),fliplr(y1(Li))],'r');
xlabel("k");
ylabel("m");
grid on
title("Stability Region on ramp response");
```





QS. 8

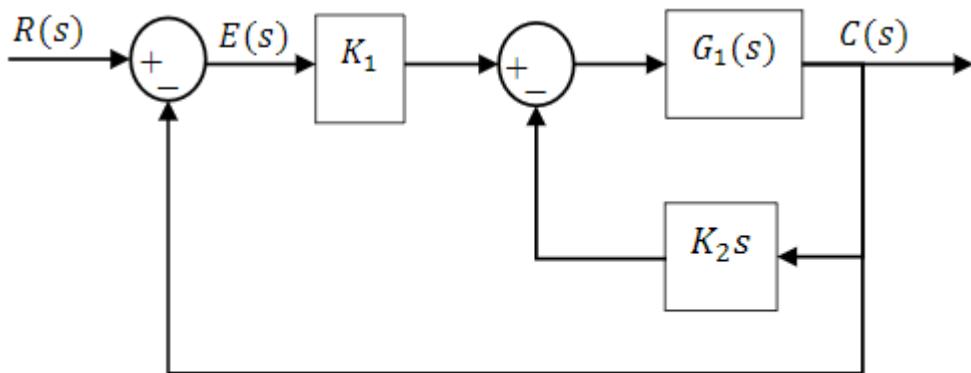


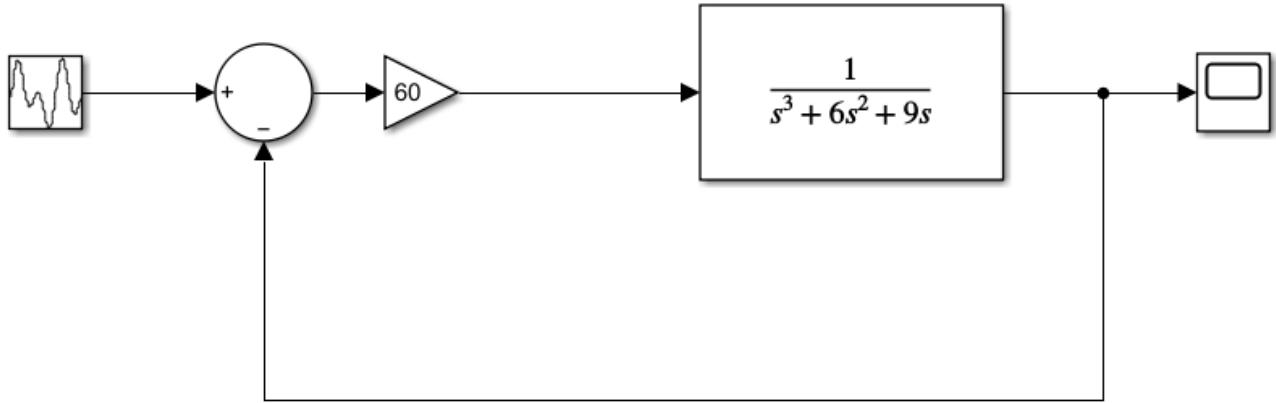
Fig. 3: A two-loop feedback system

At $K_2=1$, value of K_1 varies in the range of $0 < K_1 < 54$

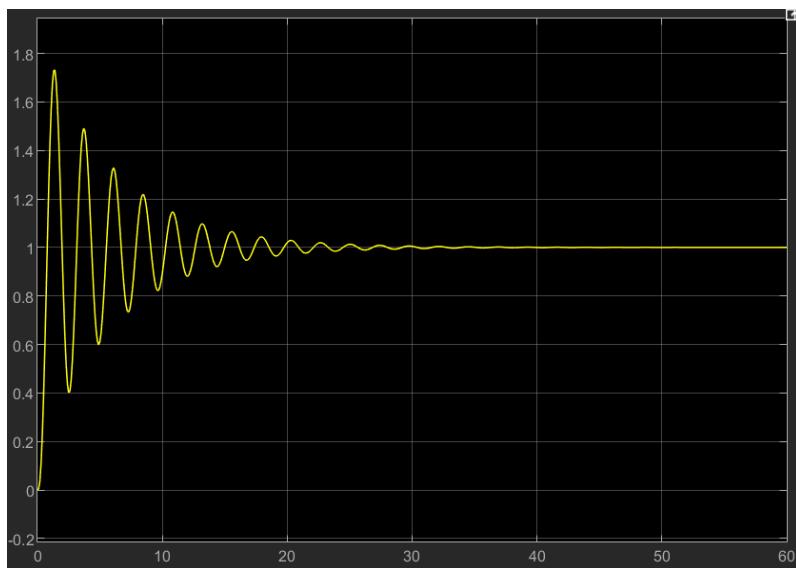
At $K_1=54$, the system is marginally stable i.e. oscillates

Frequency of Oscillation will be 3rad/sec.

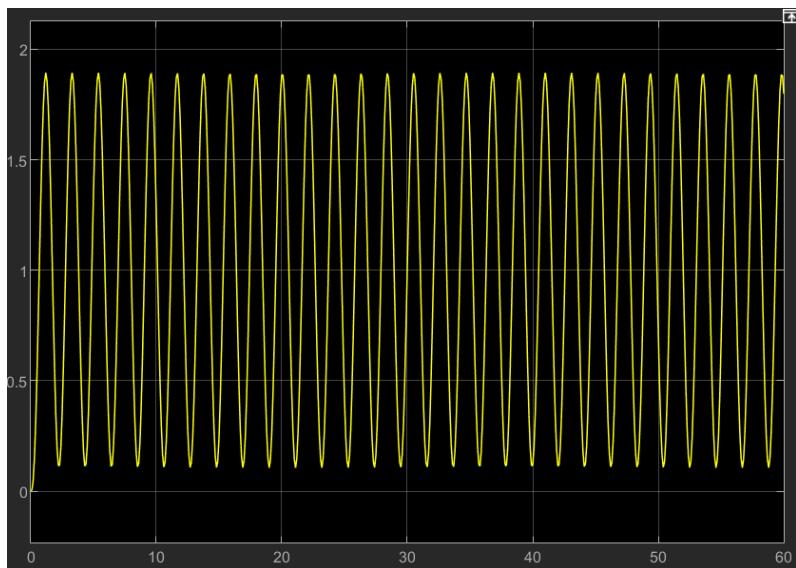
System becomes unstable when $K_1 > 54$



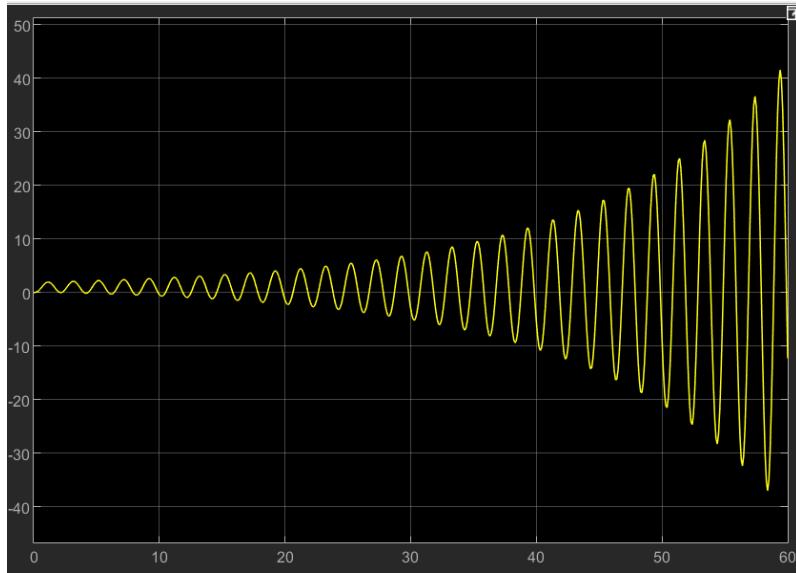
K1=40, stable



K1=54, marginally stable



K1=60, unstable

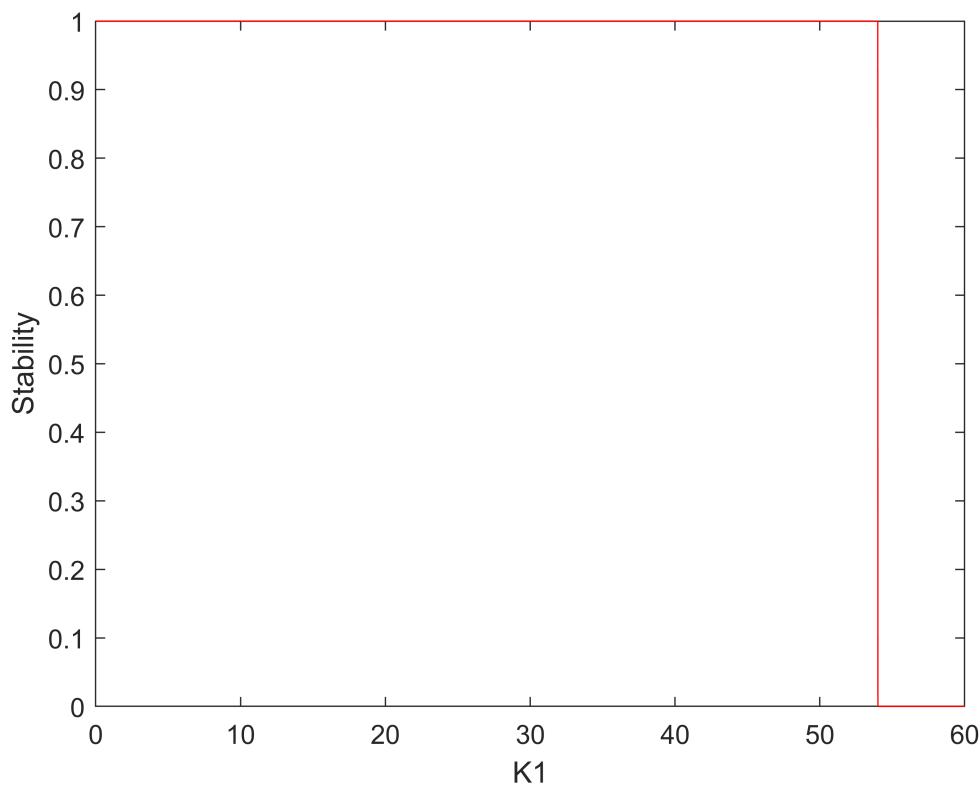


```
%a,b
```

```
k1=1e-2:1e-2:60;
y=zeros(size(k));

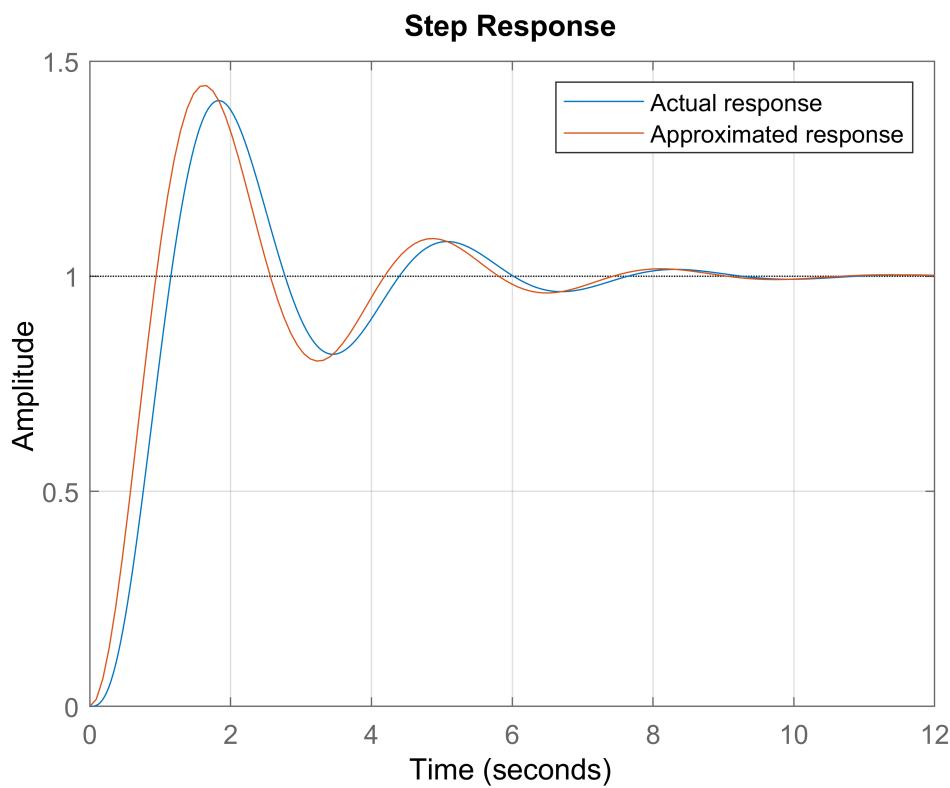
for i =1:length(k1)
    y(i)=rhc([1,6,9,k1(i)],0);
end

figure;
plot(k1,y, "r-")
xlabel("K1");
ylabel("Stability");
```



```
%c
```

```
tf1=tf(20,[1,6,9,20]);
tf2=tf(4,[1,1,4]);
step(tf1,tf2);
legend("Actual response","Approximated response")
grid on
```



QS. 9

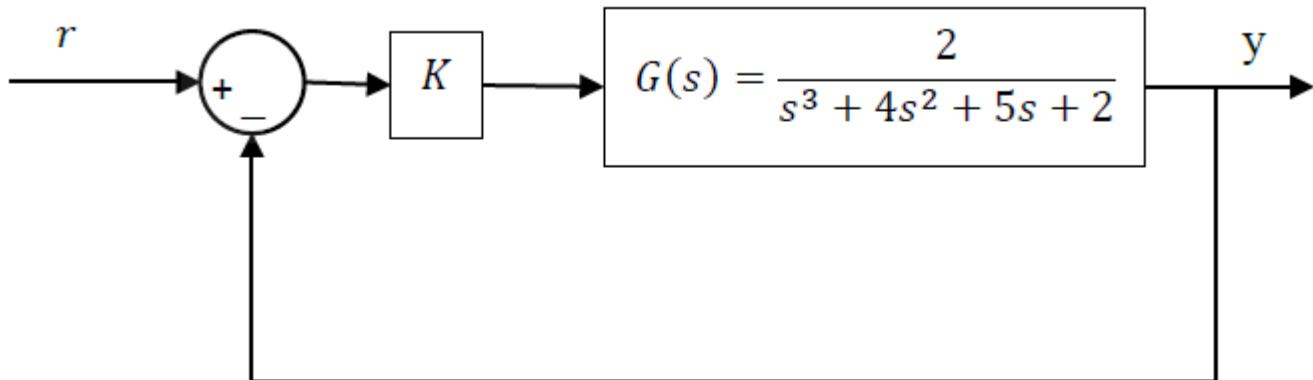
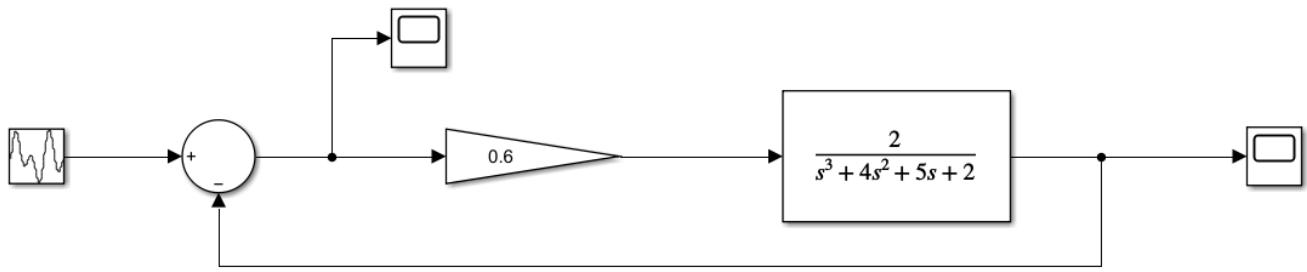
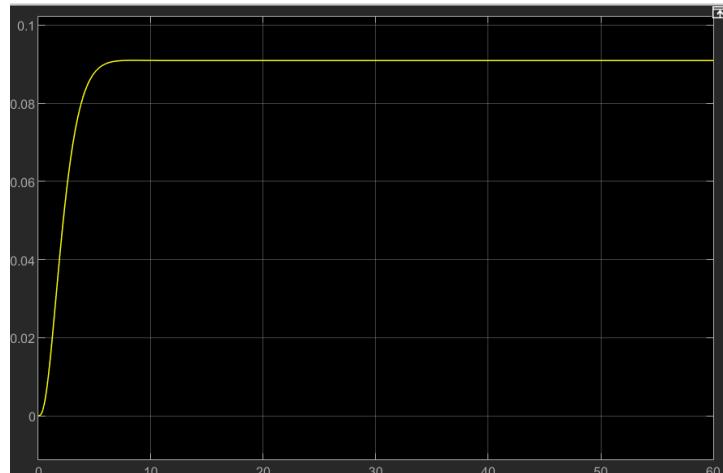


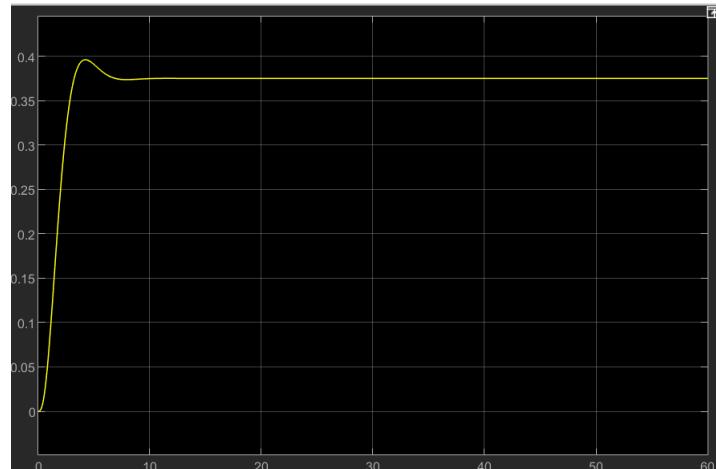
Fig. 4: A feedback system with proportional controller



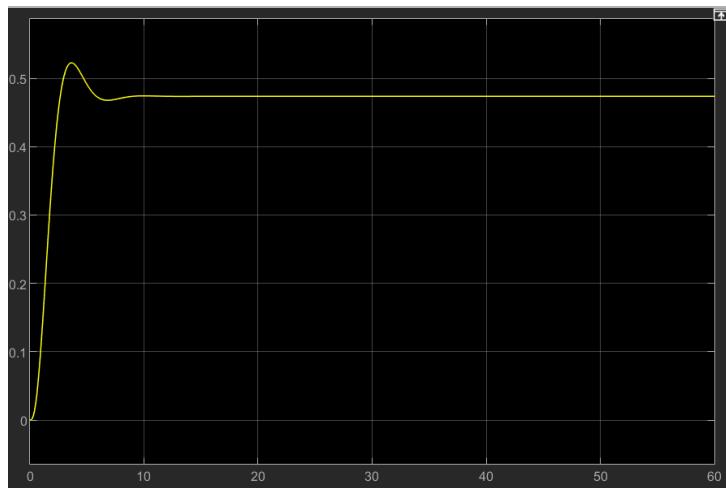
Y curve(output)



Gain=0.1,



Gain=0.6,

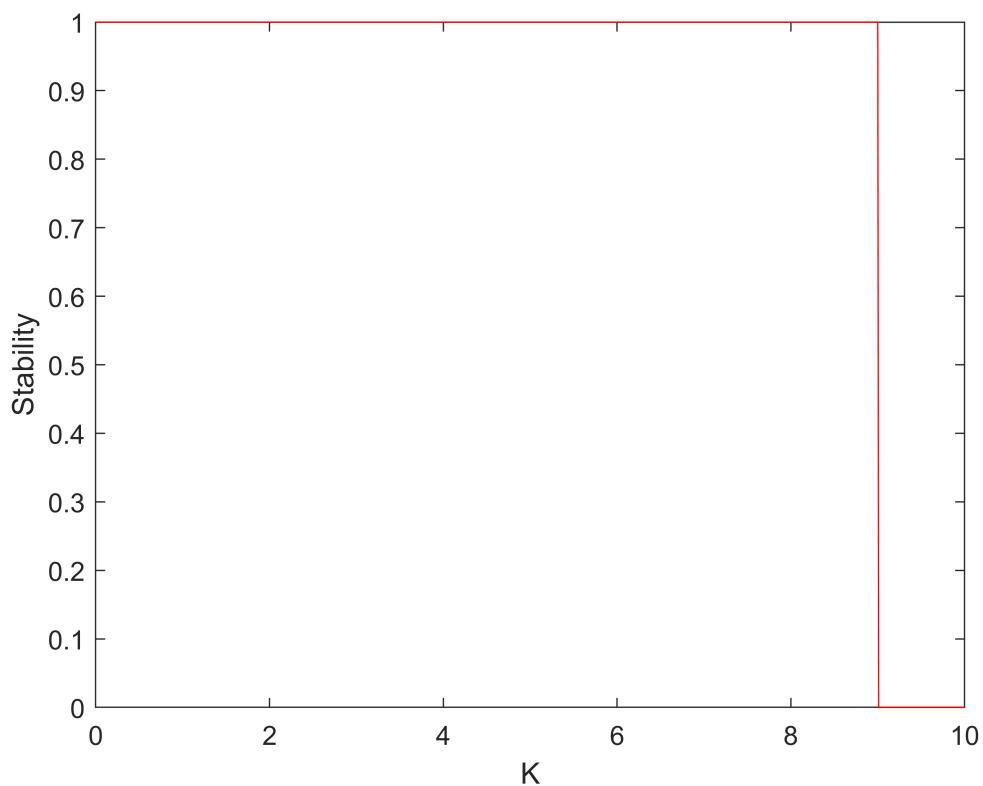


Gain=0.9,

```
k=0:0.01:10;
y=zeros(size(k));

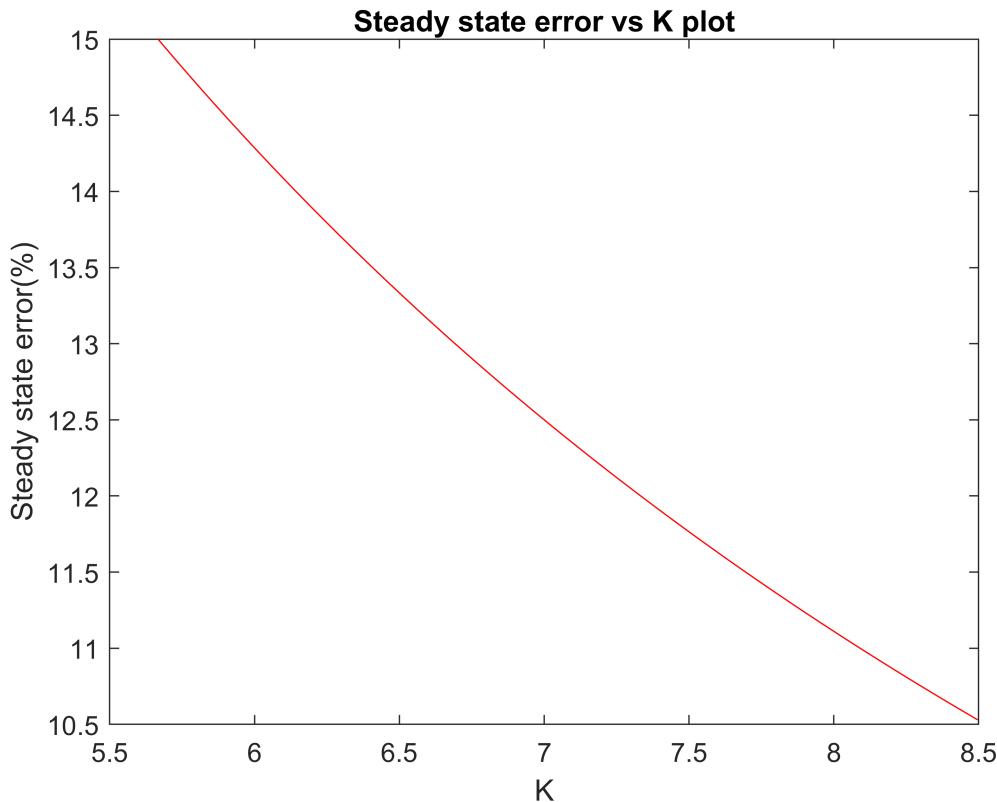
for i =1:length(k)
y(i)=rhc([1,4,5,2+2*k(i)],0);
end

figure;
plot(k,y,"r-")
xlabel("K");
ylabel("Stability");
```

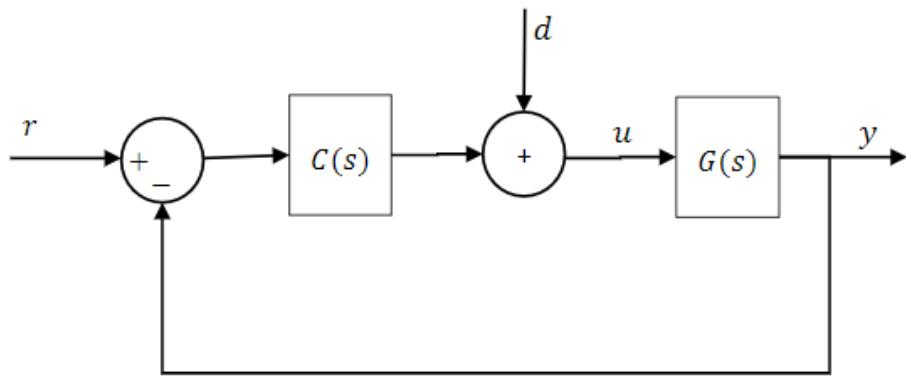


plot e_{ss} vs K graph.

```
% The error gradually decreases till.  
% Then the system becomes unstable  
  
K=17/3:1e-2:8.5;  
error=zeros(size(K));  
  
for i=1:length(K)  
    [y,t]=step(tf(2*K(i),[1,4,5,2+2*K(i)]),500);  
    error(i)=100*abs(1-y(end));  
end  
  
figure;  
plot(K,error, 'r');  
xlabel("K");  
ylabel("Steady state error(%));  
title("Steady state error vs K plot");
```



QS. 10



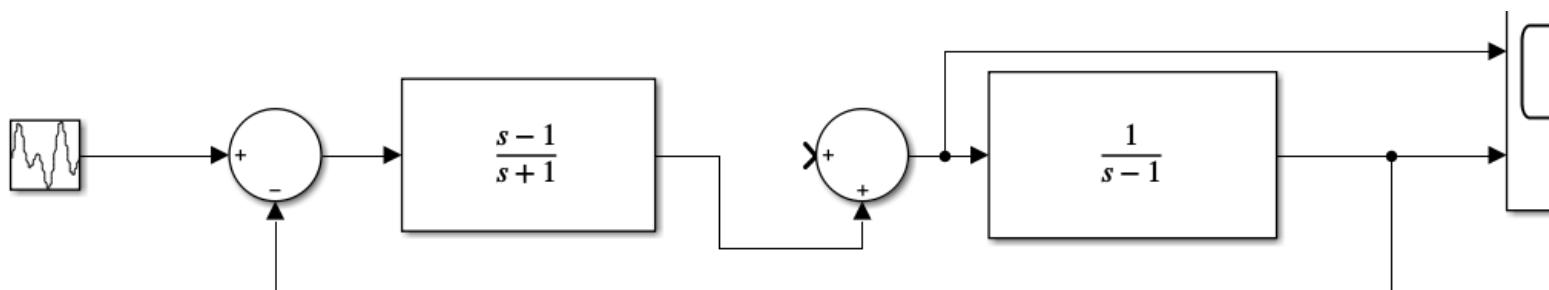
Case-I:

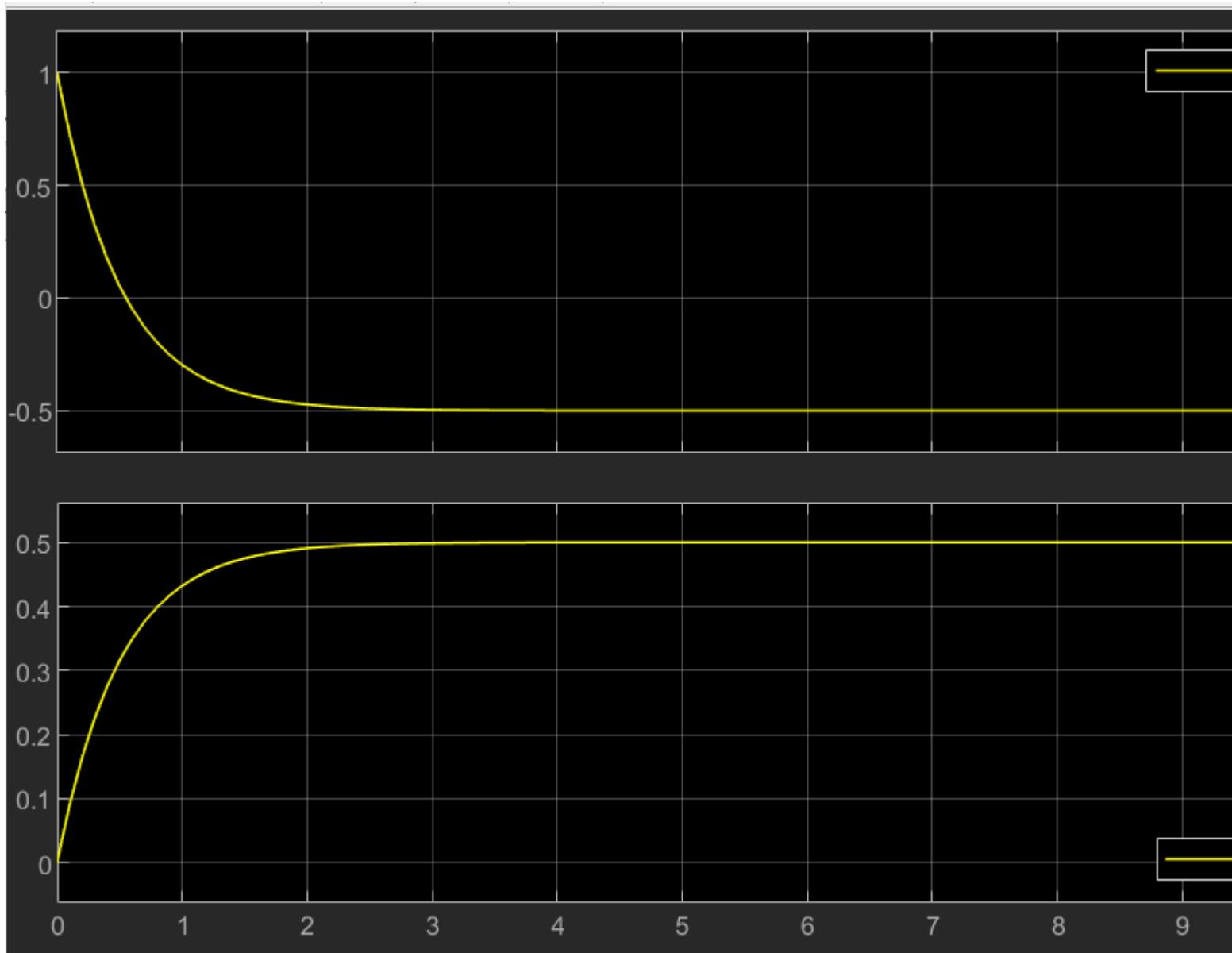
$$G(s) = \frac{1}{s - 1}$$

$$C(s) = \frac{s - 1}{s + 1}$$

r-to-y and r-to-u:

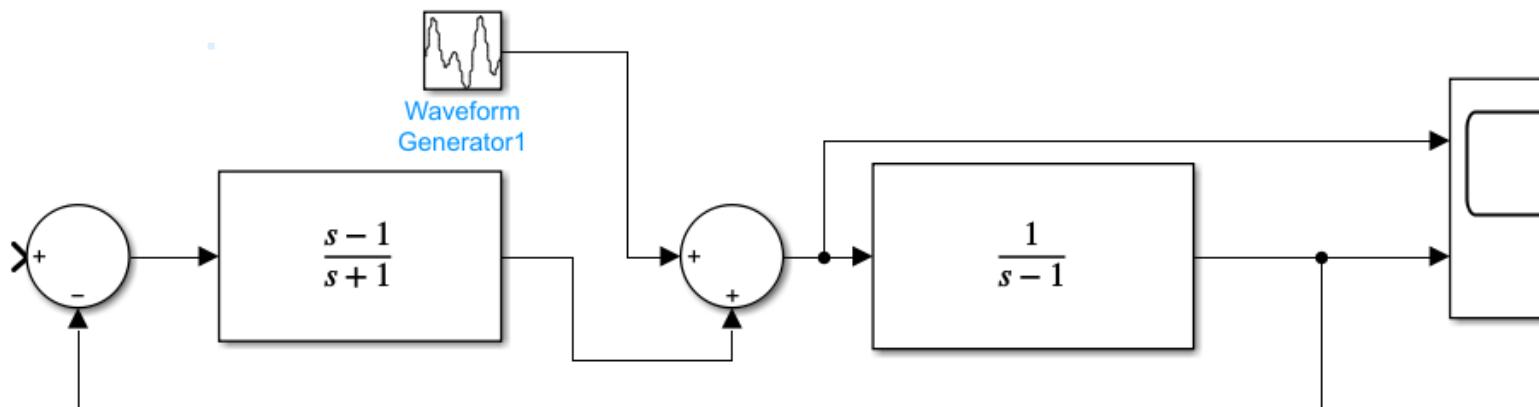
Model:

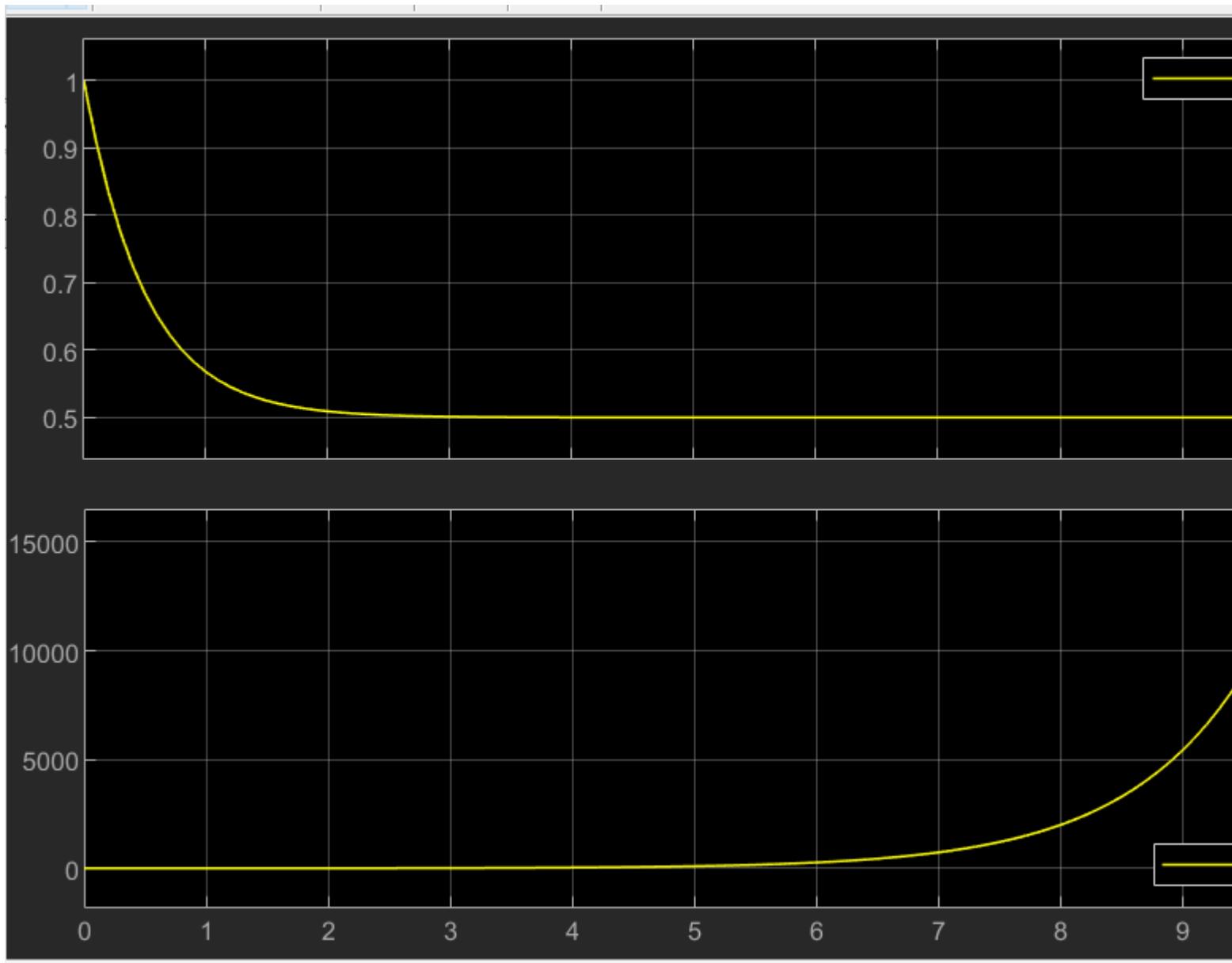




d-to-y and d-to-u:

Model:





With d as disturbance in system, the system behaves uncontrollably,

as $u(t)$ is unstable for d -to- u .

So for even a little disturbance in system, output at u will reach saturation,

changing the system output y .

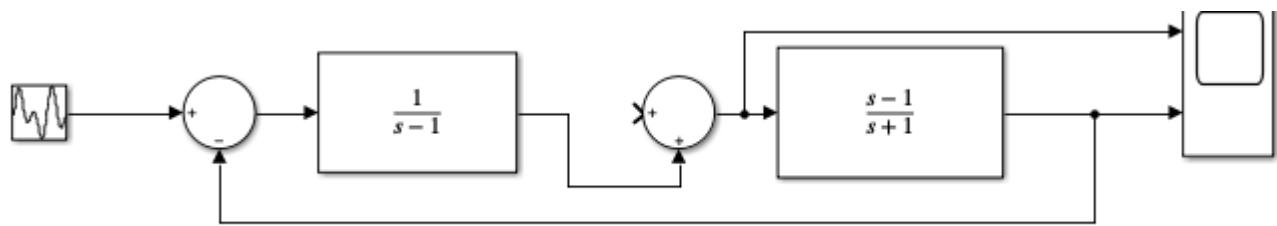
Case II:

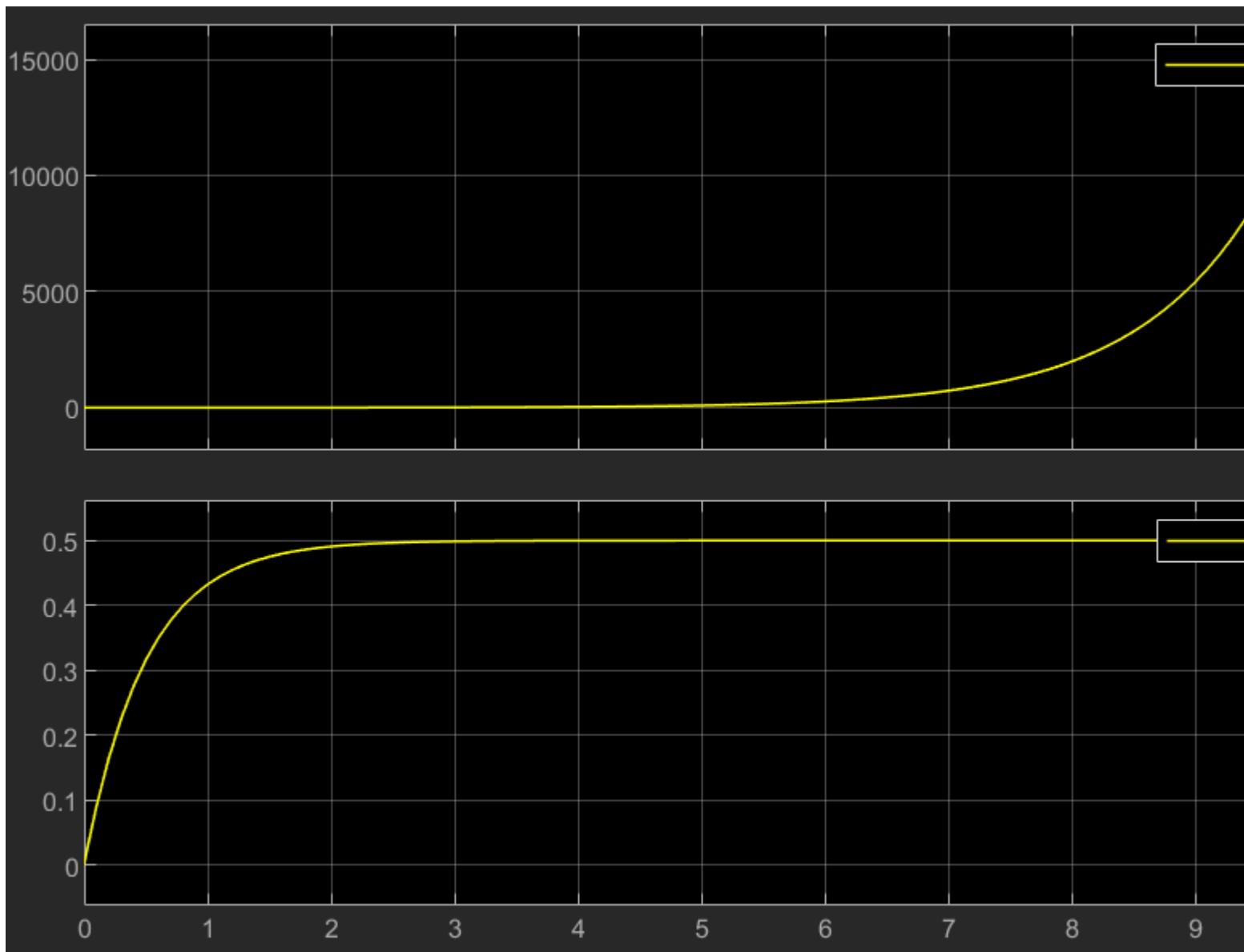
$$G(s) = \frac{s - 1}{s + 1}$$

$$C(s) = \frac{1}{s - 1}$$

r-to-y and r-to-u:

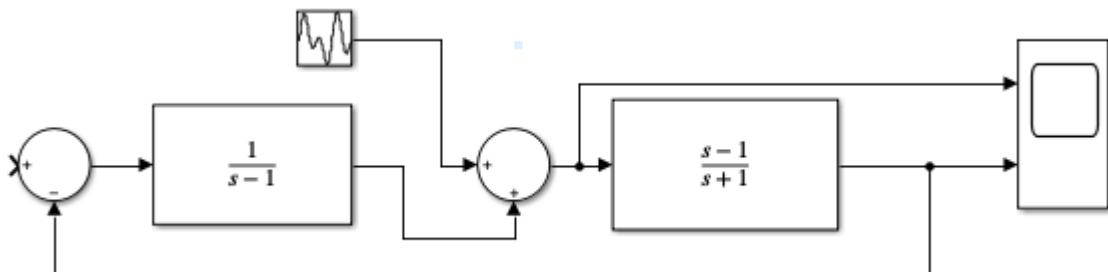
Model:

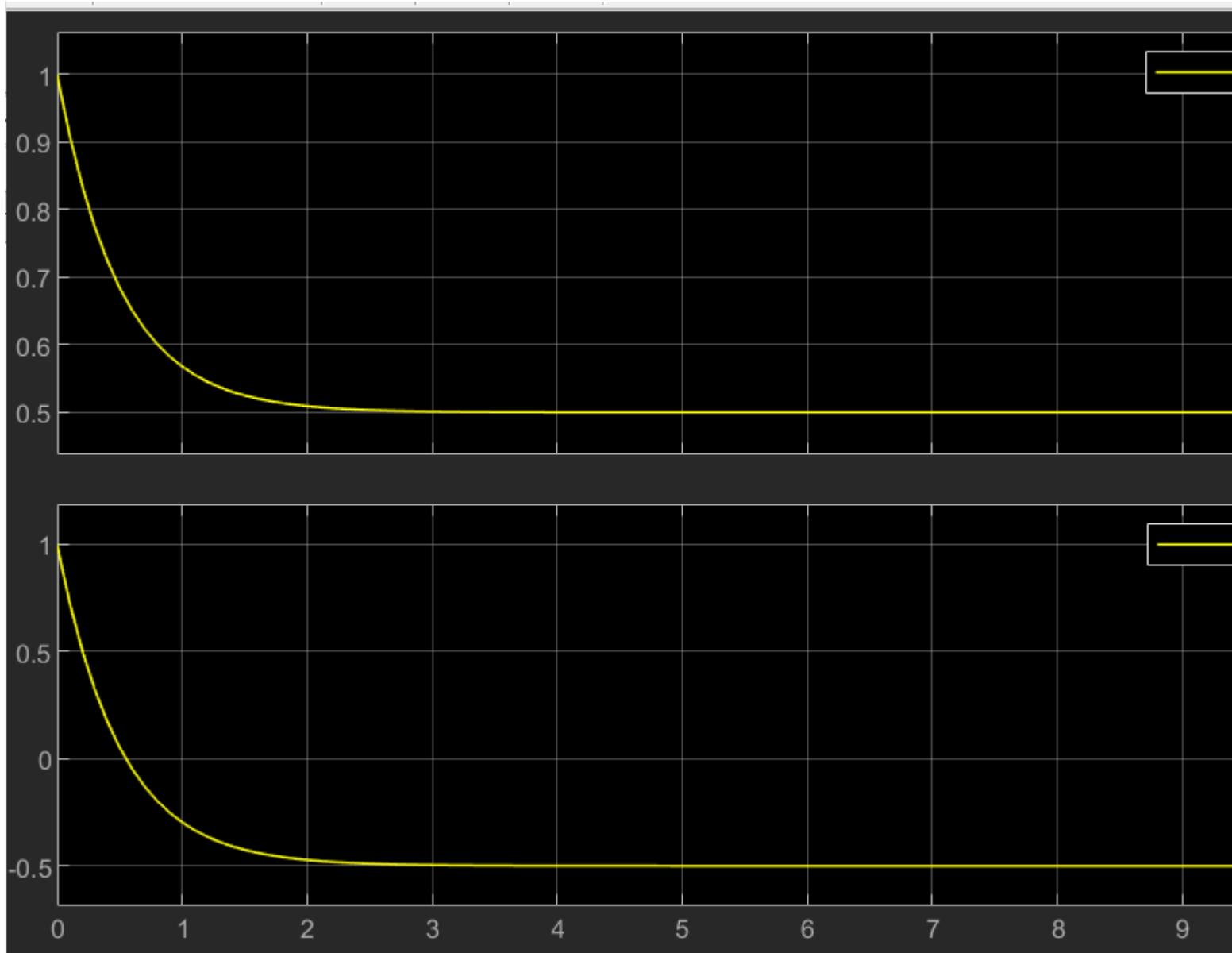




d-to-y and d-to-u:

Model:





Disturbance is controlled, as both output d-to-u and d-to-y are stable.

But the output response at u for reference input is uncontrollable,

which will reach saturation faster and change the output y.

```

function y=rhc(coeffVector,x)
ceoffLength = length(coeffVector);
rhTableColumn = round(ceoffLength/2);
rhTable = zeros(ceoffLength,rhTableColumn);
rhTable(1,:) = coeffVector(1,1:2:ceoffLength);
if (rem(ceoffLength,2) ~= 0)
    rhTable(2,1:rhTableColumn - 1) = coeffVector(1,2:2:ceoffLength);
else
    rhTable(2,:) = coeffVector(1,2:2:ceoffLength);
end
epss=0.0001;
for i = 3:ceoffLength

```

```

if rhTable(i-1,:) == 0
    order = (ceoffLength - i);
    cnt1 = 0;
    cnt2 = 1;
    for j = 1:rhTableColumn - 1
        rhTable(i-1,j) = (order - cnt1) * rhTable(i-2,cnt2);
        cnt2 = cnt2 + 1;
        cnt1 = cnt1 + 2;
    end
end
for j = 1:rhTableColumn - 1
    firstElemUpperRow = rhTable(i-1,1);
    rhTable(i,j) = ((rhTable(i-1,1) * rhTable(i-2,j+1))-(rhTable(i-2,1) * rhTable(i-1,j+1)));
end
if rhTable(i,1) == 0
    rhTable(i,1) = epss;
end
end
unstablePoles = 0;
for i = 1:ceoffLength - 1
    if sign(rhTable(i,1)) * sign(rhTable(i+1,1)) == -1
        unstablePoles = unstablePoles + 1;
    end
end
if x==1
    fprintf('\n Routh-Hurwitz Table:\n')
    rhTable %#ok<NOPRT>
end
if unstablePoles == 0
    y=1;
else
    y=0;
end
end

```

Lab 3.

Camlin Page

Date / /

$$Q1. M = \frac{20(s-1)}{(s+2)(s^2+4)}$$

↳ 1 LHP. ↳ 2 imaginary poles
=(-2) pole

∴ M(s) is marginally stable.

$$Q2. 2s^4 + s^3 + 3s^2 + 5s + 10 = 0.$$

using Routh criterion:

$$\begin{array}{ccccc} s^4 & 2 & 3 & 10 \\ s^3 & 1 & 5 & 0 \\ s^2 & -2 & 10 & 0 \\ s^1 & 4 & 0 & 0 \\ s^0 & 10 & 0 & 0 \end{array}$$

$$-\left| \begin{array}{cc} 2 & 3 \\ 1 & 5 \end{array} \right| = -7 \quad -\left| \begin{array}{cc} 2 & 10 \\ 1 & 0 \end{array} \right| = 10$$

$$-\frac{1}{7} \left| \begin{array}{cc} 1 & 5 \\ -7 & 10 \end{array} \right| = \frac{45}{7}$$

as there is a sign change

∴ system UNSTABLE

$$Q3. s^4 + s^3 + 2s^2 + 2s + 3 = 0.$$

$$\begin{array}{ccccc} s^4 & 1 & 2 & 3 \\ s^3 & 1 & 2 & 0 \\ s^2 & 0 & 3 & 0 \\ s^1 & -3 & 0 & 0 \\ s^0 & 0 & 0 & 0 \end{array}$$

UNSTABLE.

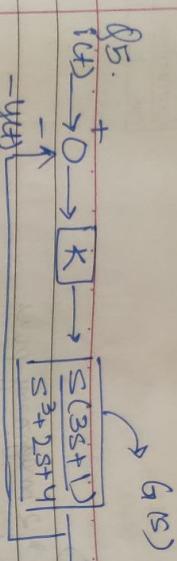
$$Q4. s^5 + 4s^4 + 8s^3 + 7s^2 + 4s + 4 = 0.$$

$$\begin{array}{ccccc} s^5 & 1 & 8 & 7 & 0 \\ s^4 & 4 & 8 & 4 & 0 \\ s^3 & 6 & 6 & 0 & 0 \\ s^2 & 4 & 4 & 0 & 0 \\ s^1 & 0 & 0 & 0 & 0 \\ s^0 & 4 & 0 & 0 & 0 \end{array}$$

No sign change

∴ system is

STABLE



$K \rightarrow$ gain of proportional controller

$$1 + KG(s) = 0$$

$$K(G(s)) (i(t) - y(t)) = y(t).$$

$$1 + \frac{s^3 + 2s + 4}{K.s(3s+1)} = 0$$

$$\begin{aligned} s^3 + 2s + 4 + Ks(3s+1) &= 0 \\ s^3 + (2+3K)s + 4 + 3Ks^2 &= 0 \end{aligned}$$

$K > 0$. ∂ h table.

$$\begin{array}{ccccc} s^3 & 1 & & & 2+K \\ s^2 & 3K & & & -4 \\ s^1 & -4K & \xrightarrow{\substack{K+2 \\ 0}} & & 0 \\ s^0 & & & \xrightarrow{-\frac{1}{3K}} & (4-3K^2-6K) \end{array}$$

as $K > 0$

$$3K^2 + 6K - 4 > 0$$

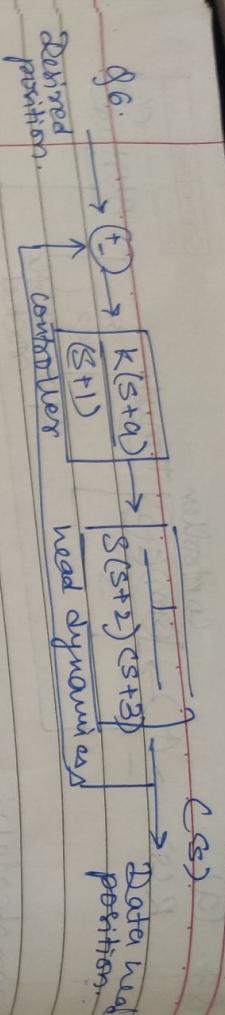
$$K > -1 + \sqrt{\frac{7}{3}}$$

when $K = -1 + \sqrt{\frac{7}{3}}$ both elements of 3rd row is 0.

both elements of 3rd row is 0.
stable (marginally).

$$\begin{array}{ccccc} s^3 & 1 & & & K+2 \\ s^2 & 3K & 4 & & \\ s^1 & 0 & 0 & \therefore & K > -1 + \sqrt{\frac{7}{3}} \\ s^0 & 0 & 0 & & \end{array}$$

System stable for



Defined position.

Characteristic equation:

$$1 + \frac{K(s+a)}{s(s+1)(s+2)(s+3)} = 0.$$

$$s^4 + 6s^3 + 11s^2 + (6+k)s + ka = 0.$$

as $k > 0 \therefore \alpha > 0$.

$$\begin{array}{cccc} s^4 & 1 & 11 & ka \\ s^3 & 6 & 6+k & 0 \\ s^2 & -\frac{1}{6} & \frac{1}{6} & \frac{1}{6} \\ s^1 & \frac{6ka - (6+k)10 - k}{10 - k/6} & 0 & 0 \end{array}$$

$$s^0 \quad ka \quad 0 \quad 0$$

20

$$\therefore 10 - \frac{k}{6} > 0 \quad \therefore 0 < k < 60.$$

$$0 < k \leq 60 \quad \text{at } k=60 \quad a=0$$

$0 \leq a \leq \frac{10+3-k}{2} \Rightarrow$
System \rightarrow marginally stable

25

\therefore Stable region: $K \in [0, 60]$
 $a \in [0, 10]$.

30

Q2.

(a) $R(s) \rightarrow \text{Controller} \xrightarrow{\frac{(sm)(k+2)}{s}} \text{Space shuttle} \xrightarrow{\frac{R}{s^2-1}} \text{Altitude}$

else

Characteristic equation:

$$1 + \frac{k(s+m)}{s(s^2-1)}(s+2km) = 0.$$

$$s^3 - s + ks^2 + k(m+2)s + 2km = 0.$$

for stability:
 $k > 0$
 $m > 0$.
 $2km > 0$.
 $\therefore m > 0$.

10

$$\begin{matrix} s^3 & 1 & k(m+2) - 1 \\ -s^2 & k & 2km \\ s^1 & \frac{-1}{k} & \frac{k(m+2)-1}{2km} \end{matrix} \quad 0.$$

20

$$\begin{matrix} G \\ (k-2)(m+2k-1) \end{matrix} > 0. \quad \left\{ \begin{array}{l} m \geq 0 \quad \text{if } k \geq 2 \\ m \leq 0 \quad \text{if } k < \frac{1}{2} \end{array} \right. \\ \left. \begin{array}{l} \geq 0 \quad \text{if } m < \frac{2k-1}{2-k} \\ \geq 0 \quad \text{if } m \geq \frac{2k-1}{2-k}. \end{array} \right. \quad 25$$

ii) System is $\begin{cases} m = \frac{2k-1}{2-k} & k \in \left[\frac{1}{2}, 2 \right] \end{cases}$
 marginally stable

∴ for
 $k \geq 2, m \geq 0$.

$$\text{or } k \in \left[\frac{1}{2}, 2 \right], m \leq \frac{2k-1}{2-k}$$

$$Q_9. (b) G_1(s) = \frac{(s+4)(s+2)}{s}$$

$$G_2(s) = \frac{k}{s^2 - 1}$$

Steady state error $\rightarrow e = \lim_{s \rightarrow 0} s G_1(s) G_2(s)$. New response $s \rightarrow 0$

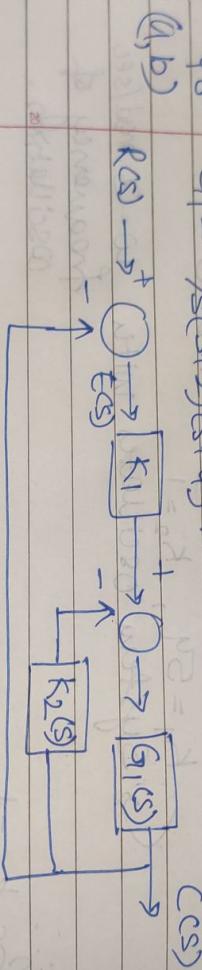
$$= (-1/2 \text{ km})$$

Given $b = 5 \pm 0.1$

$$\frac{-1}{2 \text{ km}} > -0.1$$

$$|m > 5/k|$$

$$Q_8. G_1 = \frac{1}{s(s+2)(s+4)}$$



$$1 + K_2 s G_1(s) = 0.$$

\hookrightarrow will have 2 equal -ve roots.

$$(s+2)(s+4) + K_2 = 0.$$

$$K_2 = 1$$

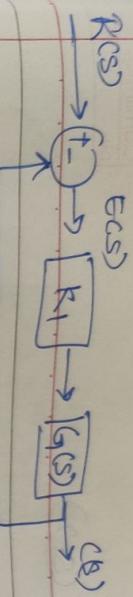
$$s^2 + 6s + 9 = 0 \quad \begin{cases} -3 \\ 2 \text{ equal} \end{cases}$$

$$s^2 + 6s + 9 = 0 \quad \begin{cases} -3 \\ 2 \text{ equal} \end{cases}$$

∴

Q 7.

Given $b = 5 \pm 0.1$
for K
System stable
chan



$$1 + K_1 G(s) = 0$$

$$1 + \frac{K_1}{S(S+2)(S+4)+3} = 0.$$

$$S^3 + 6S^2 + 9S + K_1 = 0 \quad K_1 > 0.$$

$$\begin{aligned} S^3 &= 1 + 9 \\ S^2 &= 6 \quad K_1 \\ S &= \frac{S+K_1}{6} \end{aligned}$$

$0 < K_1 < 54$

for stable system.

$$\begin{aligned} S^3 + 6S^2 + 9S + 54 &= 0 \\ (S+6)(S^2 + 9) &= 0 \end{aligned}$$

(20) $K_1 = 54, \quad K_2 = 1$
 \hookrightarrow System oscillates with $\omega = 3 \text{ rad/sec}$
 frequency of oscillation.

for $K_2 = 1$
 $\hookrightarrow K_1 \approx 0.54$

At $K_1 = 54$. \rightarrow system is marginally stable.
 when $K_1 = 6(8 + K_2)$.

if K_2 not specified, choose $S^2 = -\omega^2$
 $\omega = \sqrt{8 + K_2} \text{ rad/sec}$
 when $K_1 = 6(8 + K_2)$.

Q8 (c) as closed loop pole at will make system unstable, we won't able to approximate Step response as underdamped.

$$s^3 + 6s^2 + (8+k_2)s + k_1 = 0$$

$$s= -5 \rightarrow \text{pole}$$

$$k_1 = -s(6s+k_2)$$

~~10~~

$$k_1 = s(k_2+3)$$

$$(8+5)(s^2 + s + k_2 + 3) = 0.$$

1 pole at $s = -5$

both complex poles have real = $-4\frac{1}{2}$.

real pole is 10 times away from img axis.

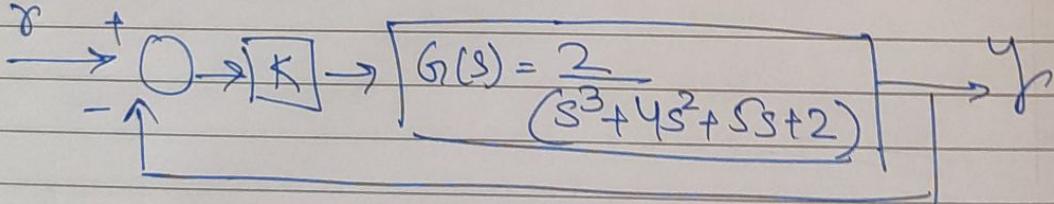
Dominant pole approximation

$$k_2 = 1 \quad k_1 = 20.$$

$$G(s) = \frac{20}{s^3 + 6s^2 + 9s + 20}.$$

$$G_{opp}(s) = \frac{4}{s^2 + s + 4}.$$

Q9.



$$y(s) = \frac{K G(s)}{1 + K G(s)} r(s).$$

$$E(s) = \frac{R(s)}{1 + K G(s)}.$$

$$R(s) = 1/s.$$

$$E = \frac{1}{1 + K \lim_{s \rightarrow 0} G(s)}.$$

$$e = \frac{1}{1+k}$$

$G(s) = (-2, -1, -1)$ in LHP.

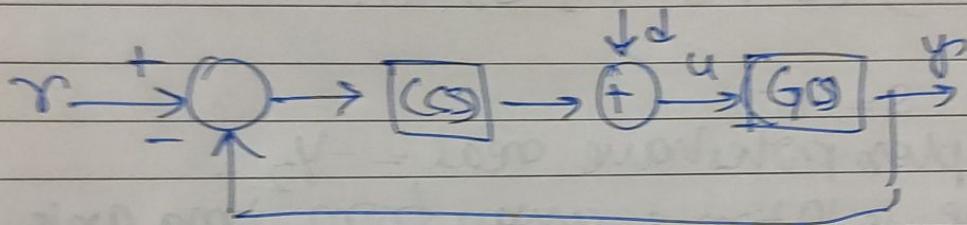
$$\frac{1}{1+k} \leq 0.15$$

$$k \geq 17/3$$

Characteristic eqn^u $1+k G(s)=0$

$$s^3 + 4s^2 + 5s + 2 + 2k = 0$$

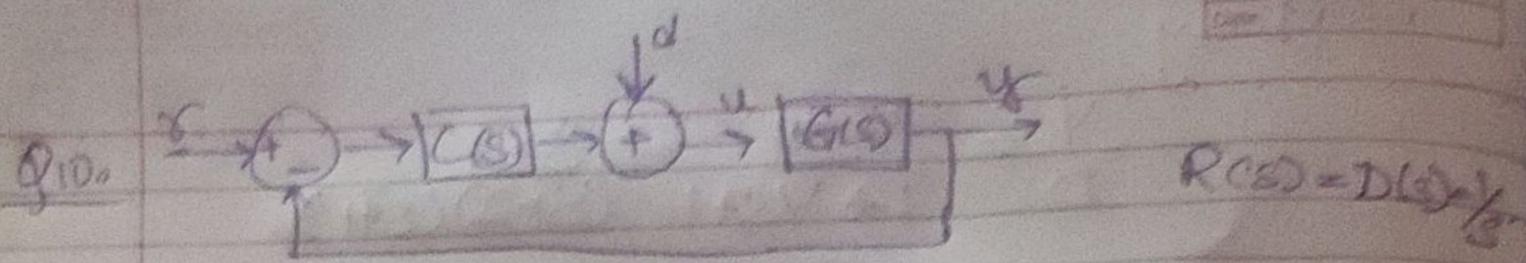
Q10



$$G(s) = \sqrt{s+1}$$

$$C(s) = \frac{s+1}{s+2}$$

case 1.



$$G(s)(C(s)(R(s) - Y(s)) + D(s)) = Y(s)$$

$$Y(s) = U(s)G(s)$$

$\tau \rightarrow y$

$$G(s)(\zeta s)R(s) = Y(s)(1 + G(s)\zeta s)$$

$$\frac{Y(s)}{R(s)} = \frac{G\zeta}{1 + G\zeta}$$

$d \rightarrow y$

$$G.D = Y(1 + G)$$

$$\frac{Y}{R} = \frac{G}{1 + G}$$

$\tau \rightarrow u$

$$\frac{UG}{R} = \frac{G\zeta}{1 + G}$$

$$\frac{U}{R} = \frac{C}{1 + GC}$$

$d \rightarrow u$

$$\frac{GU}{D} = \frac{G}{1 + GC}$$

$$\frac{U}{D} = \frac{1}{1 + GC}$$

Case ①

$$C = \frac{s-1}{s+1} \quad G = \frac{1}{s-1}$$

$\tau \rightarrow y$

$$\frac{Y}{R} = \frac{\frac{1}{s+1}}{1 + \frac{1}{s+1}} = \frac{1}{s+2}$$

$$Y = \frac{1}{s+2} \cdot \frac{1}{s}$$

$$Y(t) = \frac{1}{2} (u(t) - e^{-2t} u(t))$$

$$Y = \frac{1}{2} \left(\frac{1}{s} - \frac{1}{s+2} \right)$$

$d \rightarrow y$

$$\frac{Y}{D} = \frac{V_{S+1}}{1 + Y_{S+1}} \Rightarrow \frac{s+1}{(s+2)(s+5)}$$

$$Y = \frac{S+1}{(s+2)(s+1)(s)} \rightarrow Y = \underbrace{\frac{1}{2s} + \frac{2}{3(s-1)} - \frac{1}{6(s+5)}}$$

$$y(t) = \left(\frac{1}{2} + \frac{2}{3}e^t - \frac{1}{6}e^{-2t} \right) u(t)$$

$d \rightarrow u$.

$$\frac{U}{D} = \frac{1}{1 + Y_{S+1}} = \frac{s+1}{s+2}$$

$$U = \frac{s+1}{s(s+2)}$$

$$U = \frac{1}{2} \left(\frac{1}{s+2} + \frac{1}{s} \right)$$

$$u(t) = \underbrace{\left(\frac{1}{2} + \frac{1}{2}e^{-2t} \right) u(t)}$$

$v \rightarrow u$.

$$\frac{U}{R} = \frac{s+1}{1 + Y_{S+1}} = \frac{s+1}{s+2}$$

$$U = \frac{s+1}{(s+2)s} = \frac{3}{2} \frac{1}{s+2} - \frac{1}{2s}$$

$$u(t) = \underbrace{\left(\frac{3}{2}e^{-\frac{t}{2}} - \frac{1}{2} \right) u(t)}$$

case (2) $G(s) = \frac{s+1}{s+1}$, $C(s) = Y_{S+1}$

$d \rightarrow u$ $\frac{U}{R} = \frac{1}{1+GC} \rightarrow y(t) = \left(\frac{1+1}{2}e^{-2t}\right)u(t)$.
remains the same.

$r \rightarrow y$ $\frac{Y}{R} = \frac{GC}{1+GC} \rightarrow y(t) = \left(\frac{1-1}{2}e^{-2t}\right)u(t)$.
remains the same

$r \rightarrow u$ $\frac{U}{R} = \frac{Y_{S+1}}{1+Y_{S+1}} \rightarrow$ Same
as case (1)
 $d \rightarrow y$.

$$u(t) = \left(-\frac{1}{2} + \frac{2}{3}e^{-2t} - \frac{5}{6}e^{-2t}\right)u(t)$$

$d \rightarrow y$. $\frac{Y}{R} = \frac{S+1/S+1}{1+S+1/S+1} \rightarrow$ Same
as case (1)

$r \rightarrow u$.

$$y(t) = \left(\frac{3}{2}e^{-2t} - \frac{1}{2}\right)u(t)$$