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- 1) An aircraft in level and unaccelerated flight with a velocity of $v_\infty = 300$ m/s requires a power of $9 \times 10^6 W$. If the aircraft weighs $1.5 \times 10^5 N$, the lift-to-drag ratio $\frac{L}{D}$ is _____.
- 2) The percentage change in the lift-off distance for a 20 % increase in aircraft weight is _____.
- 3) Consider a monoplane wing and a biplane wing with identical airfoil sections, wingspans and incidence angles in identical conditions in a wind tunnel. As compared to the monoplane, the biplane experiences
 - a) a higher lift and a higher drag
 - b) a higher lift and a lower drag
 - c) a lower lift and a lower drag
 - d) a lower lift and a higher drag
- 4) A statically stable trimmed aircraft experiences a gust and the angle of attack reduces momentarily. As a result, the center of pressure of the aircraft
 - a) shifts forward
 - b) shifts rearward
 - c) does not shift
 - d) coincides with the neutral point
- 5) Consider a wing of elliptic planform, with its aspect ratio $AR \rightarrow \infty$. Its lift-curve slope, $\frac{dC_L}{d\alpha} =$ _____.
- 6) An ideal gas in a reservoir has a specific stagnation enthalpy of h_0 . The gas is isentropically expanded to a new specific stagnation enthalpy of $\frac{h_0}{2}$ and velocity u . The flow is one-dimensional and steady. Then $\frac{u^2}{h_0} =$ _____.
- 7) The Reynolds number, Re is defined as $\frac{U_\infty L}{\nu}$ where L is the length scale for a flow, U_∞ is its reference velocity and ν is the coefficient of kinematic viscosity. In the laminar boundary layer approximation, comparison of the dimensions of the convection term $u \frac{\partial u}{\partial x}$ and the viscous term $\nu \frac{\partial^2 u}{\partial x^2}$ leads to the following relation between the boundary layer thickness δ and Re
 - a) $\delta \propto \sqrt{Re}$
 - b) $\delta \propto 1/\sqrt{Re}$
 - c) $\delta \propto Re$
 - d) $\delta \propto 1/Re$
- 8) Isentropic efficiencies of an aircraft engine operating at typical subsonic cruise conditions with the following components - intake, compressor, turbine and nozzle - are denoted by η_i, η_c, η_t and η_n , respectively. Which one of the following is correct?
 - a) $\eta_i < \eta_c < \eta_t < \eta_n$
 - b) $\eta_t < \eta_i < \eta_c < \eta_n$
 - c) $\eta_c < \eta_t < \eta_i < \eta_n$
 - d) $\eta_c < \eta_i < \eta_t < \eta_n$
- 9) A rocket nozzle is designed to produce maximum thrust at an altitude, $H = 8km$ from the sea level. The nozzle operates in
 - a) under-expanded condition for $H > 8km$
 - b) under-expanded condition for $H < 8km$
 - c) sonic exit condition for $H < 8km$

d) unchoked condition for $H < 8km$

- 10) In the solution of $\frac{d^2y}{dx^2} - 2\frac{dy}{dx} + y = 0$, if the values of the integration constants are identical and one of the initial conditions is specified as $y(0) = 1$, the other initial condition $y'(0) = \underline{\hspace{2cm}}$.
- 11) For $x > 0$, the general solution of the differential equation $\frac{dy}{dx} = 1 - 2y$ asymptotically approaches $\underline{\hspace{2cm}}$.
- 12) For a parabola defined by $y = ax^2 + bx + c$, $a \neq 0$, the coordinates (x, y) of the extremum are
- $\left(\frac{-b}{2a} + \frac{\sqrt{b^2 - 4ac}}{2a}, 0 \right)$
 - $\left(\frac{-b}{2a}, \frac{-b^2 + 4ac}{2a} \right)$
 - $\left(\frac{-b}{2a}, \frac{-b^2 + 4ac}{4a} \right)$
 - $(0, c)$
- 13) The 2-D stress state at a point P in the x - y coordinate system is $\begin{bmatrix} 60 & 50 \\ 50 & -40 \end{bmatrix} MPa$. The magnitude of the tangential stress (in MPa) on a surface normal to the x -axis at P is $\underline{\hspace{2cm}}$.