



Computational Modelling of Surface Tension Driven Flow and Experimental Measurements of Metallic Wicks

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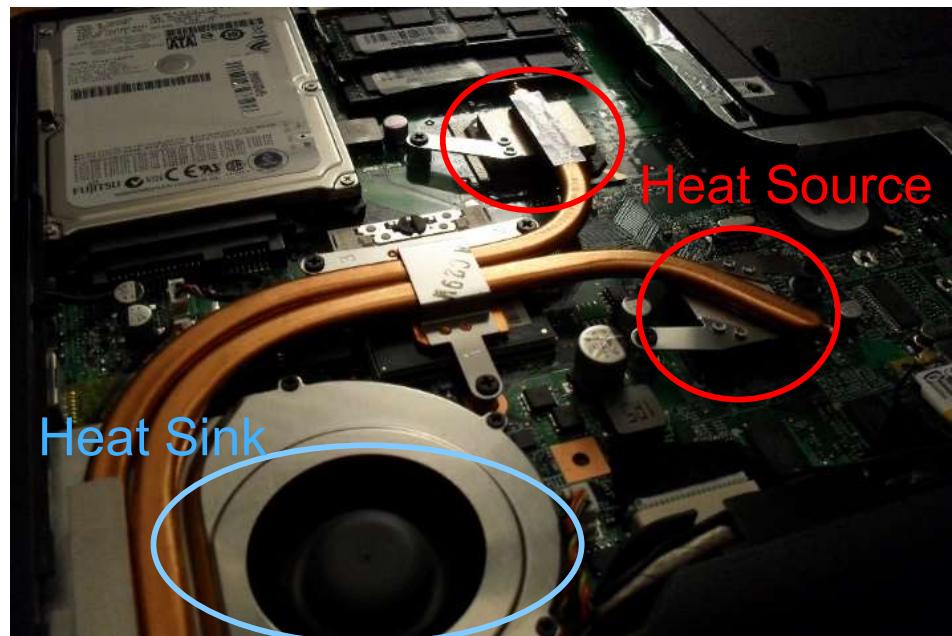
Outline

- Introduction and Motivation
- Governing Equations
- Interface Reconstruction and Advection
- Surface Tension and Curvature
- Validation Cases
- Current and Future Work
- Conclusion



Introduction

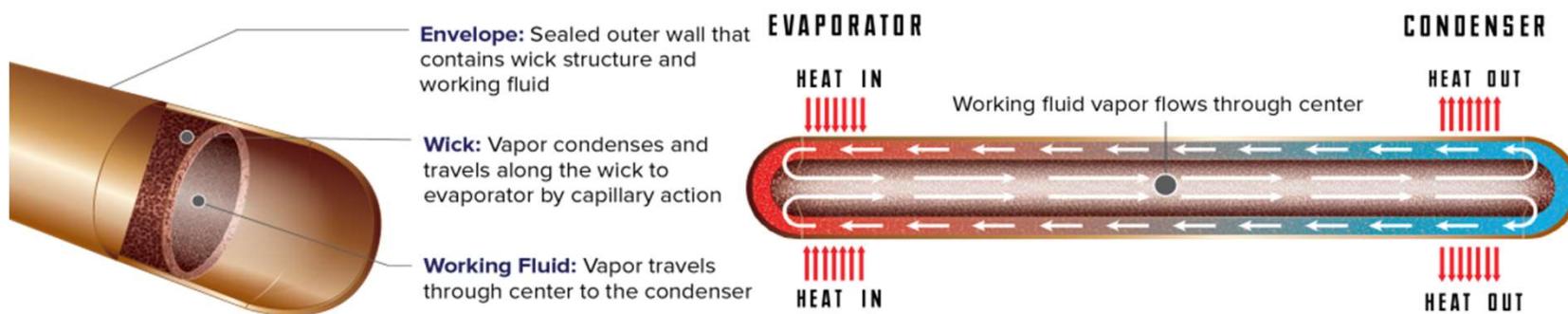
- Poor thermal management leads to total system failure
- Two-phase thermal management, is crucial to improve efficiency in miniaturized electronic, energy, and spacecraft systems
- Latent heat → Higher heat flux
- Passive cooling → Reliability, No Power!
- Long distances → Flexibility



<https://commons.wikimedia.org/w/index.php?curid=13144188>

Why wicks?

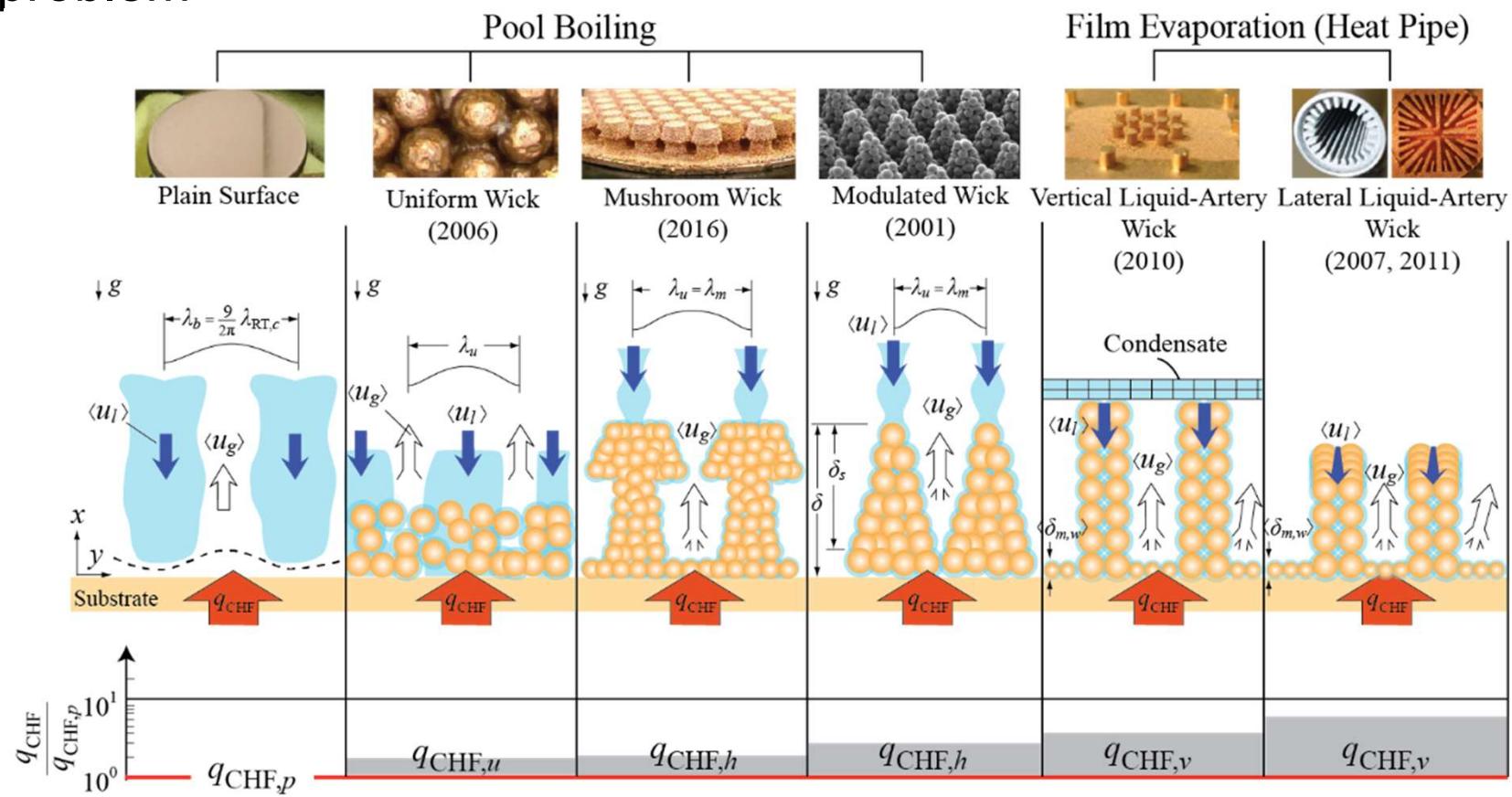
- Evaporator: Coolant is vaporized by absorbing heat
- Condenser: Phase change to liquid
- A wick structure is needed so that condensed liquid returns back to the heating region



<https://www.1-act.com/resources/heat-pipe-resources/faq/>

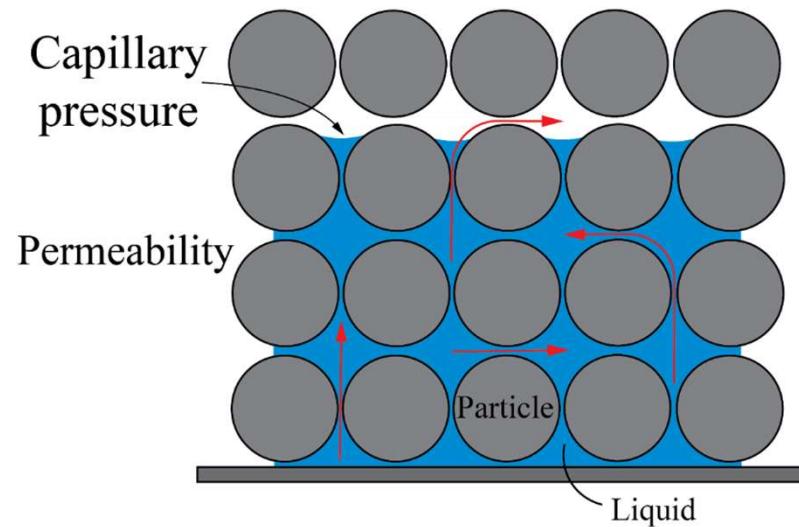
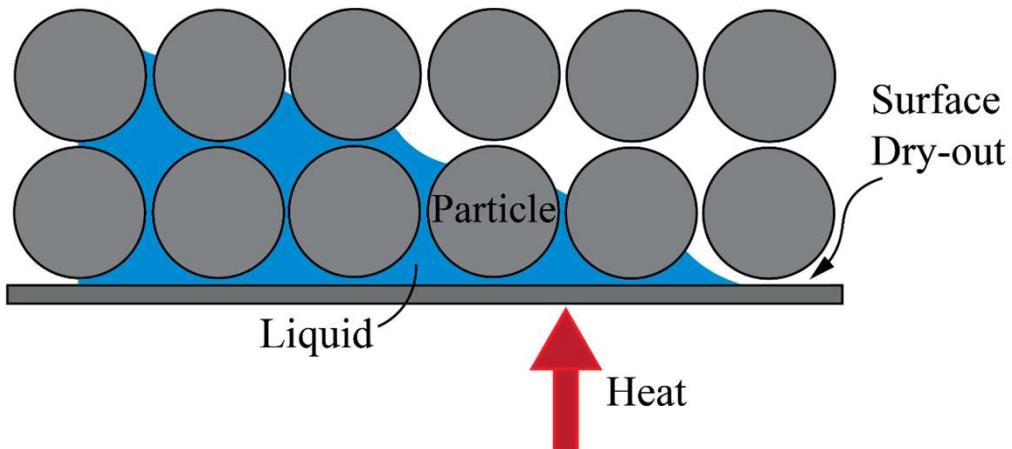
Why wicks?

- Fundamentally, wicks are used to supply cooland to the heated surface and can be applied to film evaporation and pool boiling problem



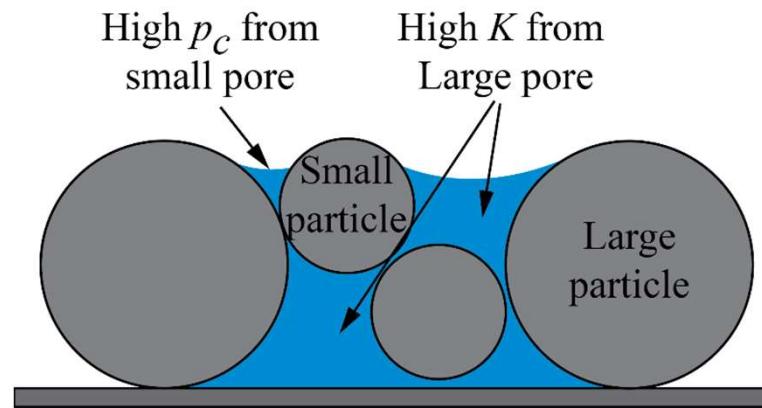
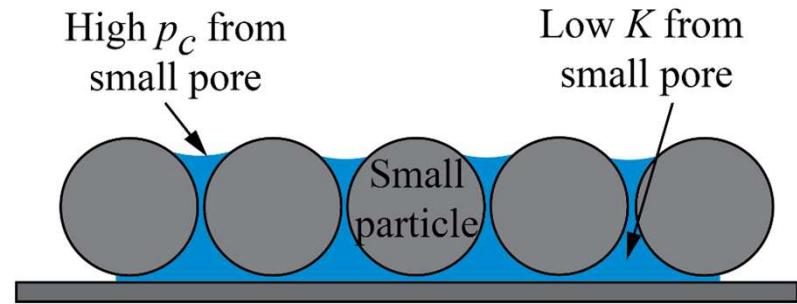
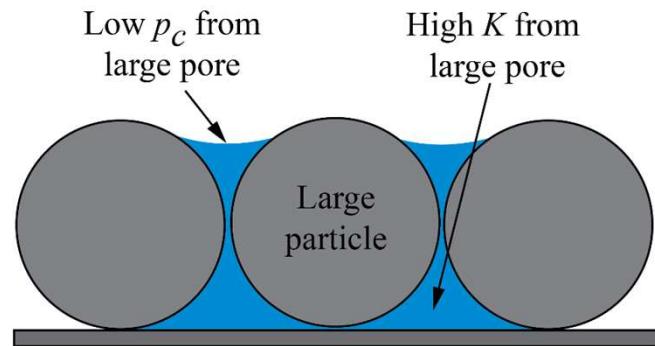
Preventing Surface Dry-Out

- It is important to keep the surface wet and prevent surface dry-out
- Limits the maximum cooling capability (critical heat flux)
- Continuous heat transfer can lead to system failure!



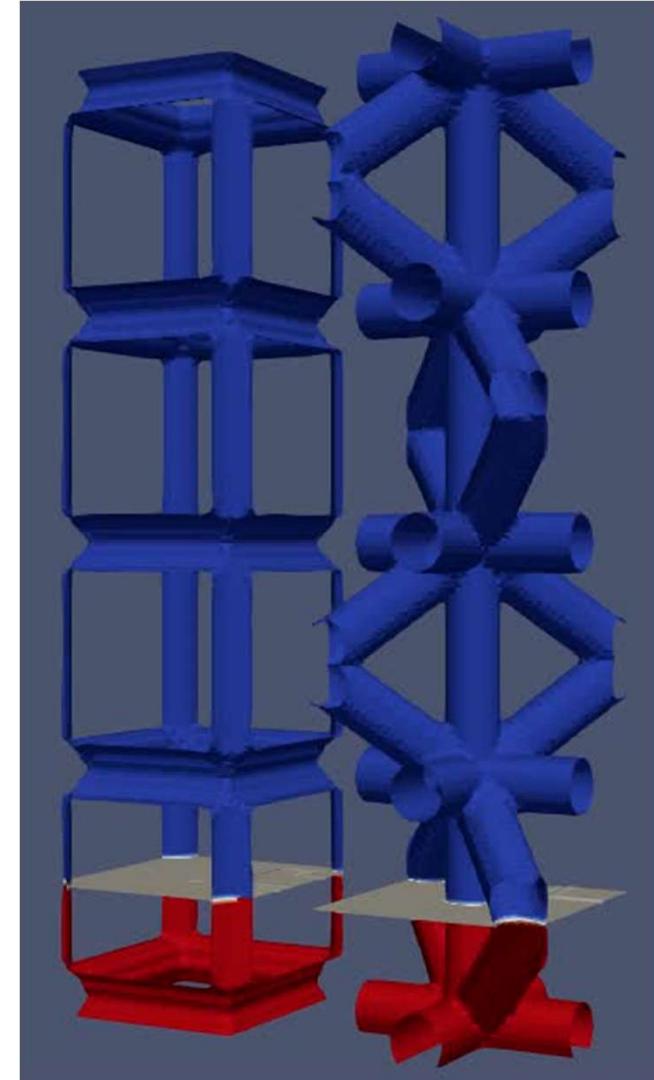
Wickability

- High permeability $K = a \frac{\varepsilon^3 r_p^2}{(1-\varepsilon)^2}$
- High capillary pressure $p_c = \frac{2\sigma}{r_c}$



Modelling using Finite Volume Method

- Permeability requires single phase, steady state, pressure driven simulation.
 - straight forward
 - cheap
- Capillary pressure requires two phase, transient, buoyancy driven simulation!
 - modelling of the vapor-liquid interface
 - accurate contact angle modelling
 - computationally expensive



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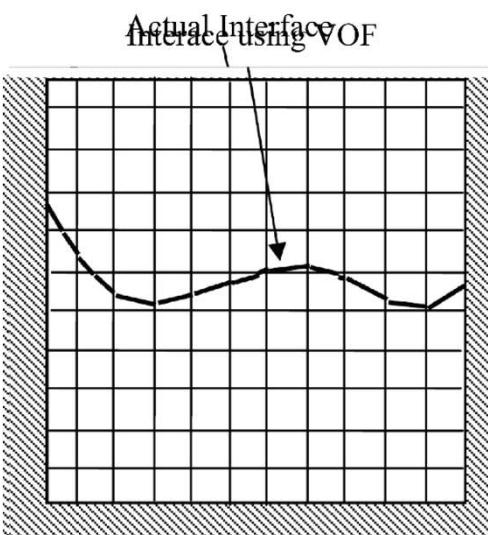
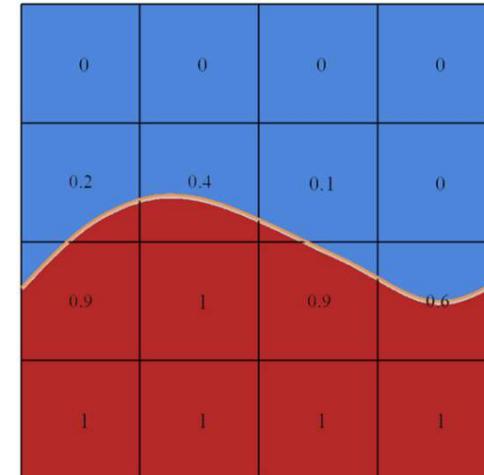
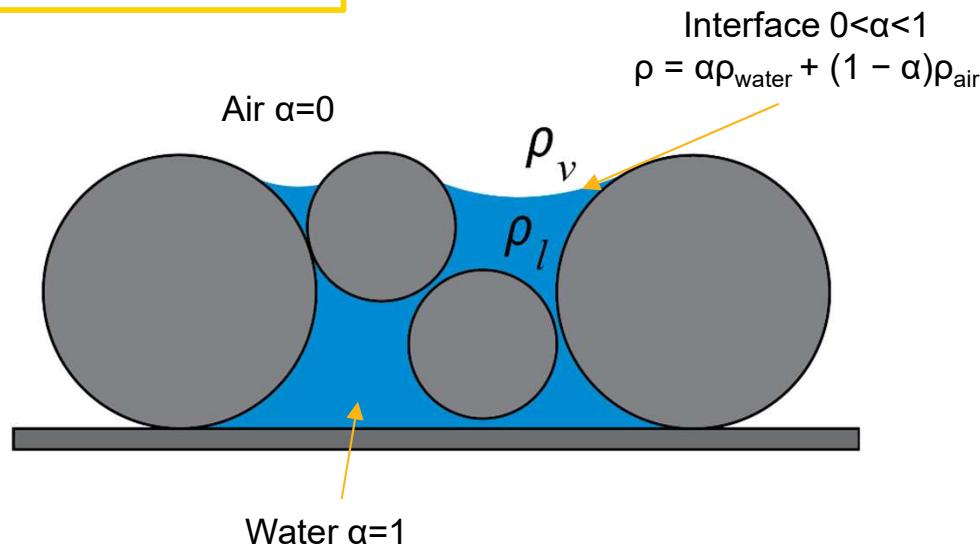
Governing Equations

$$\nabla \cdot \mathbf{u} = 0$$

$$p_{rgh} = p - \rho(\mathbf{g} \cdot \mathbf{x})$$

$$\frac{\partial(\rho\mathbf{u})}{\partial t} + \mathbf{u} \cdot \nabla(\rho\mathbf{u}) - \nabla \cdot (\mu \nabla \mathbf{u}) = -\nabla p_{rgh} - \mathbf{g} \cdot \mathbf{x} \nabla \rho + \mathbf{f}$$

$$\frac{\partial \alpha}{\partial t} + \nabla \cdot (\mathbf{u}\alpha) = 0$$

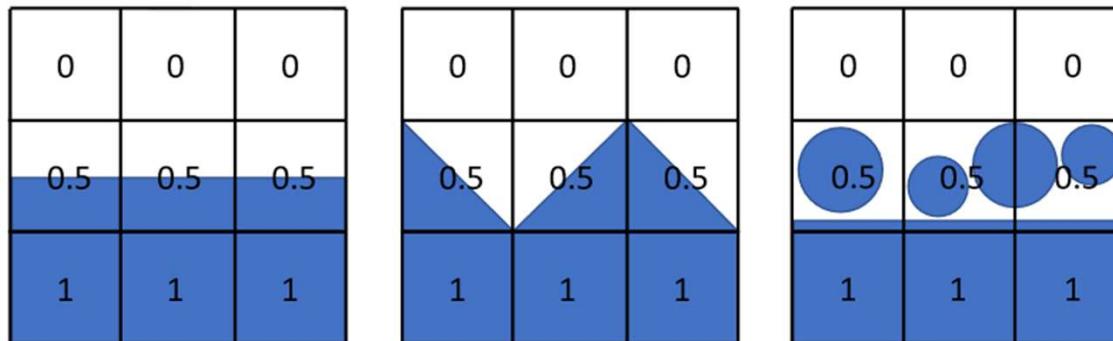


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Interface Reconstruction

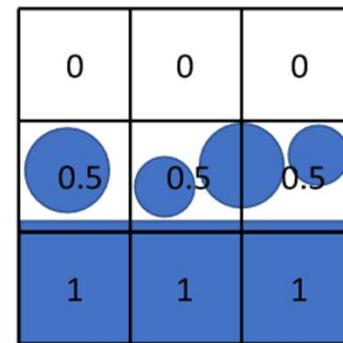
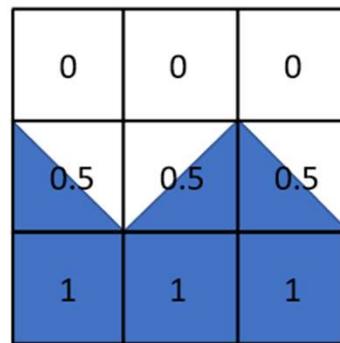
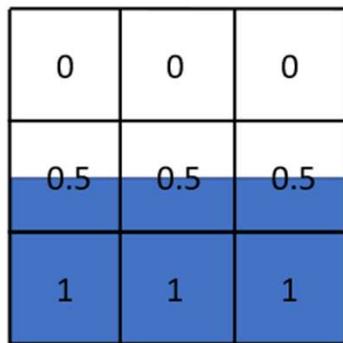
- Because we are not tracking the interface, when we solve the transport equation for the volume fraction, we only have knowledge of the scalar values.
- We have no idea on the distribution of the two fluids at the cell faces.



- A subgrid scale model is needed for cells with $0 < \alpha < 1$

Interface Reconstruction

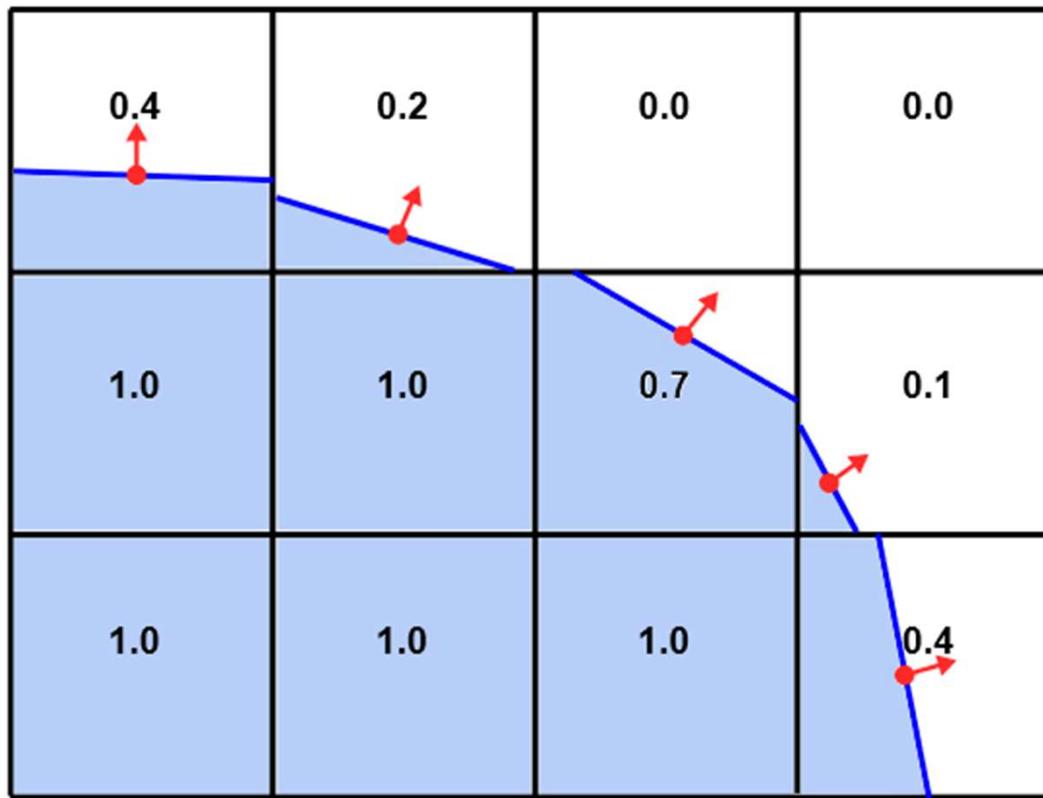
- The key assumption is that the interface is smooth (relative to grid size).
- This means that if the mesh size is much smaller than the local radius of curvature, $\Delta x \ll R$, then every cell can be approximated as a linear plane.



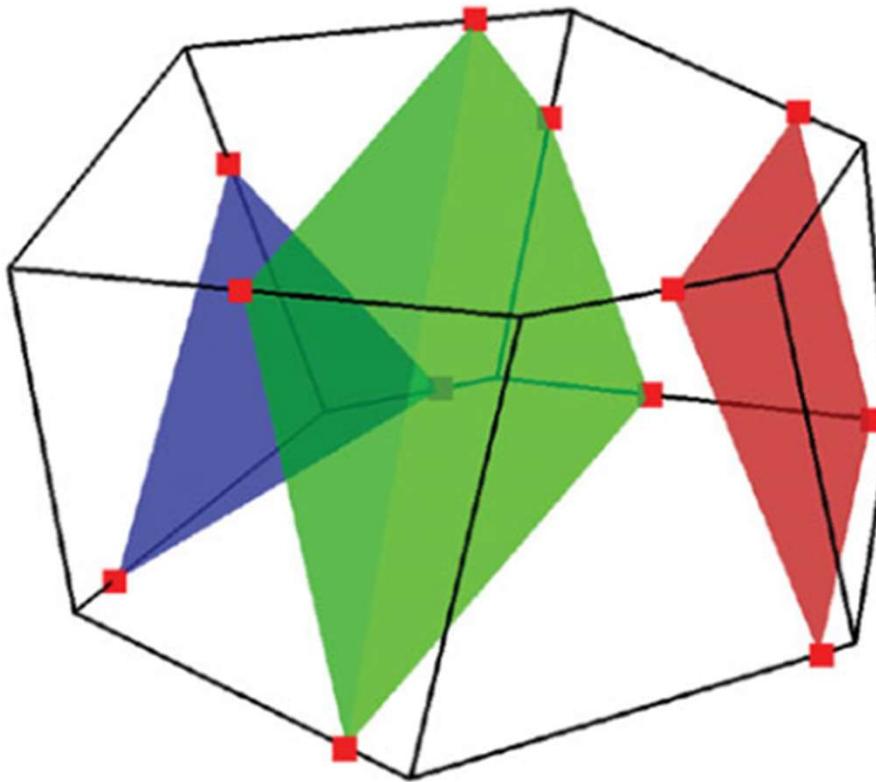
- Inherent restriction on mesh size
- Typical grid convergence not possible because refining the grids changes the subgrid modelling!

Interface Reconstruction

- We now know $\alpha(x,t)$ and the linear interface, however what is $\alpha(x,t+\Delta t)$?



Interface Advection



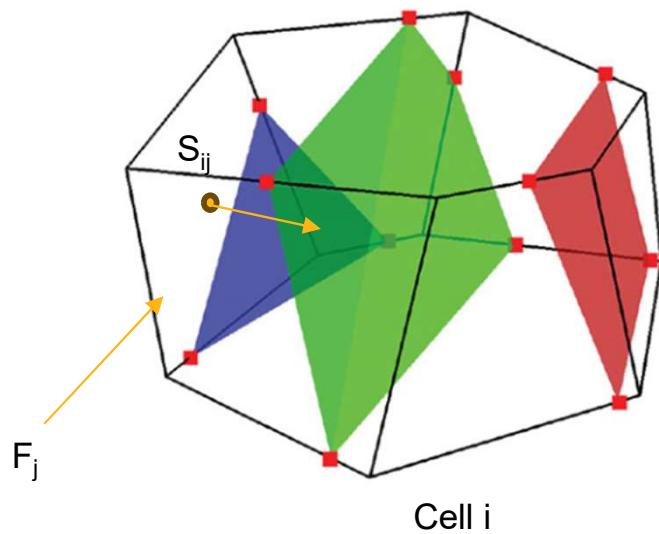
Interface Advection

This sums the net volume transported for all face of cell i

This is an exact solution and is valid for all two-phase solver

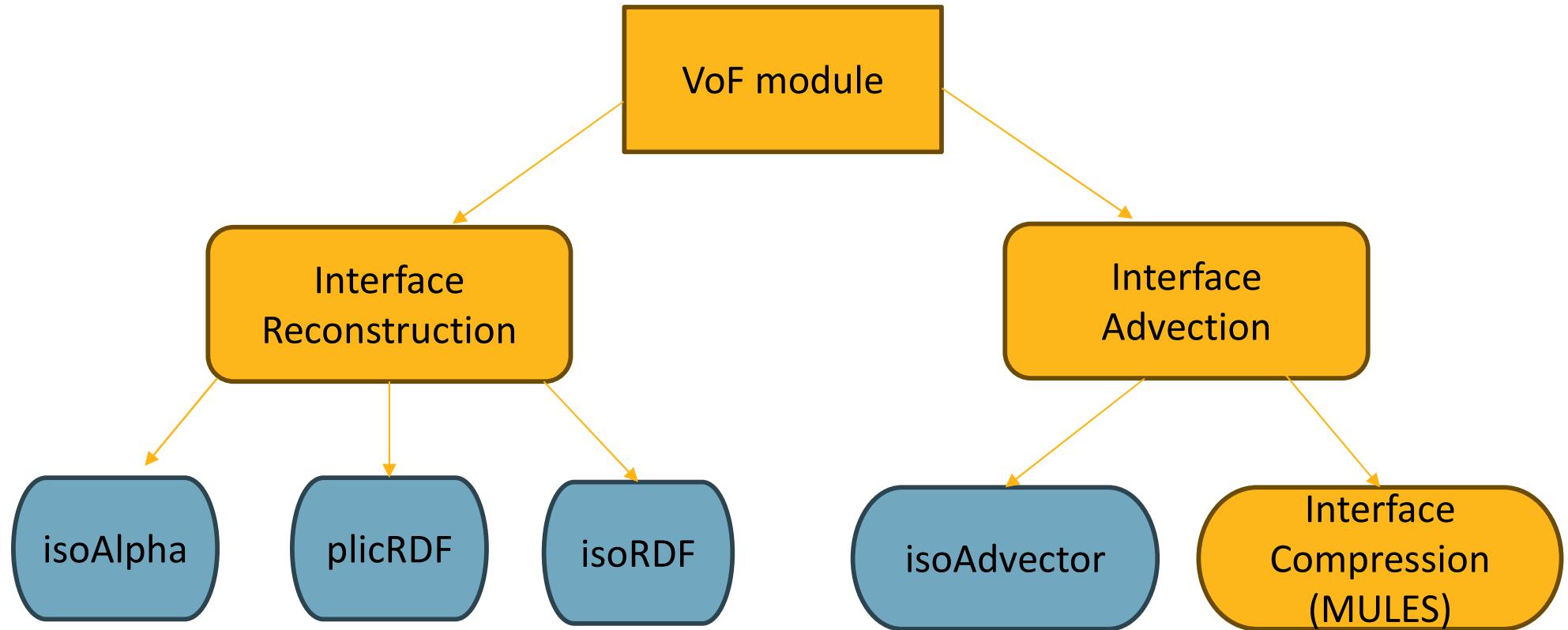
$$\alpha_i(t + \Delta t) = \alpha_i(t) - \frac{1}{V_i} \sum_{j \in B_i} s_{ij} \int_t^{t+\Delta t} \int_{F_j} H(\mathbf{x}, t) \mathbf{u}(\mathbf{x}, t) d\mathbf{S} dt$$

We have no idea what's going on at the future time step!

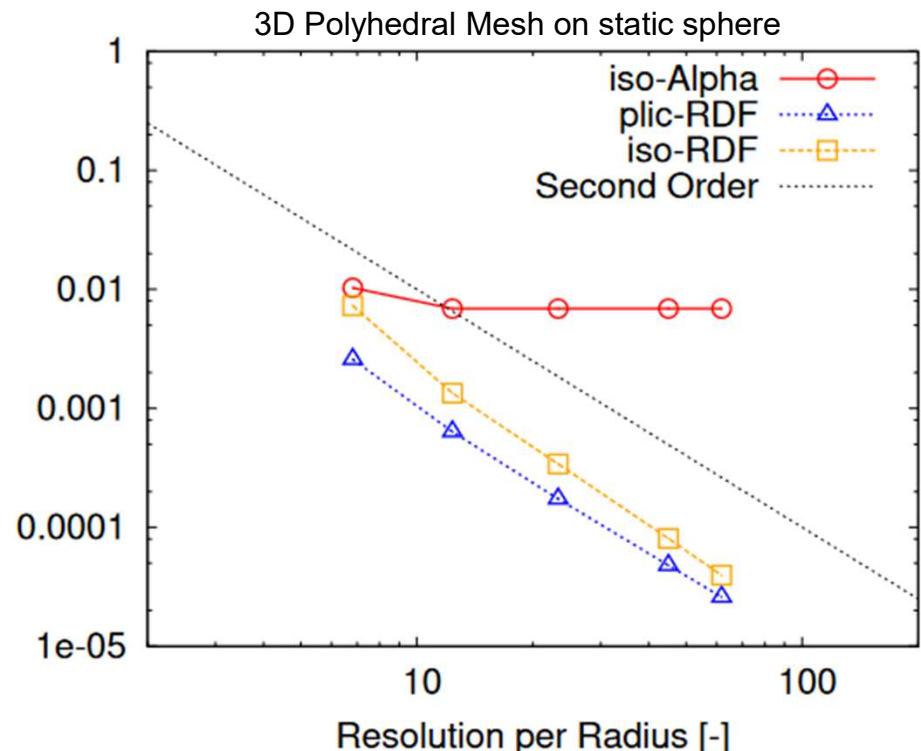
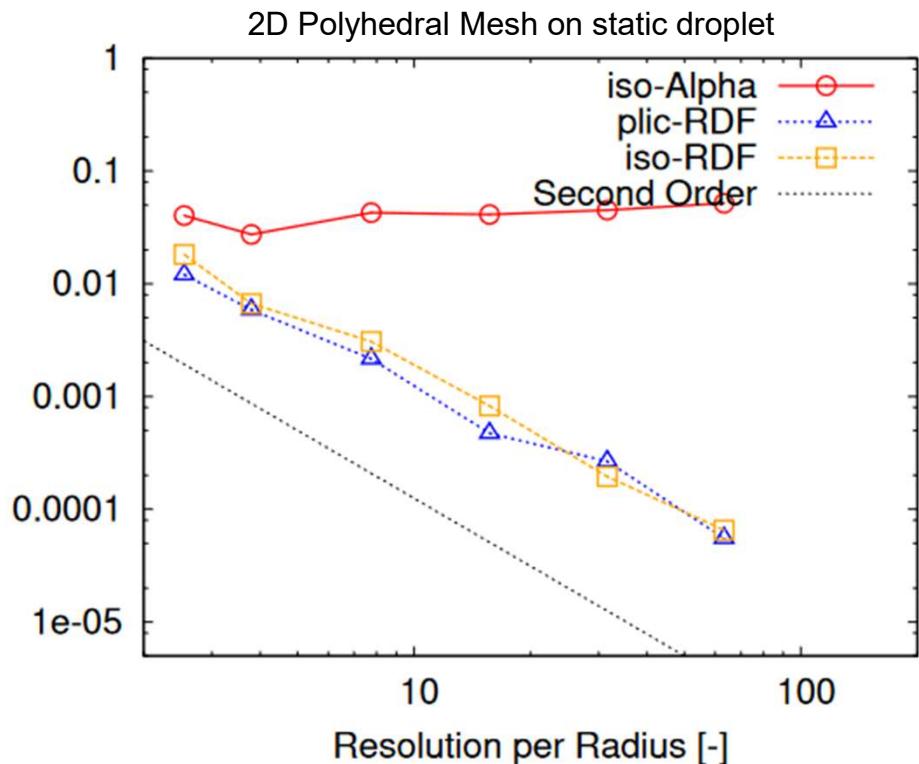


This is just the total volume of water that is transported across a single face j between t and $t + \Delta t$

Overview of twoPhaseFlow library

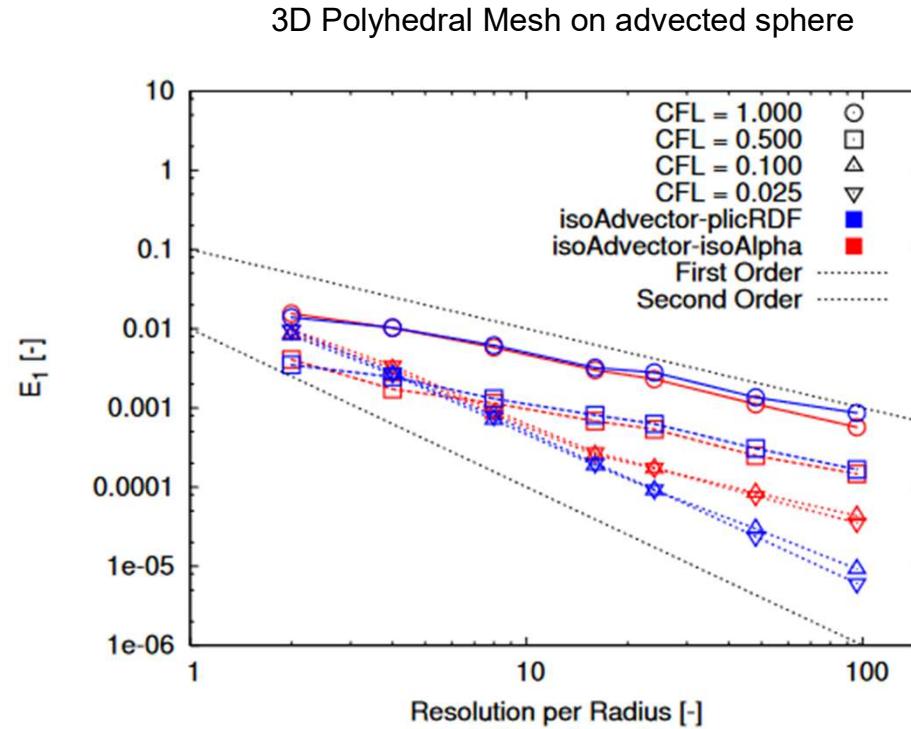


Notes on Accuracy



- plicRDF and isoRDF yields second-order convergence for 2D and 3D unstructured meshes in static equilibrium conditions
- Much better accuracy than iso-Alpha (standard OpenFOAM)

Notes on Accuracy



- plicRDF generally yields second-order convergence for 3D unstructured meshes
- plicRDF with isoAdvector scheme should be used for most applications

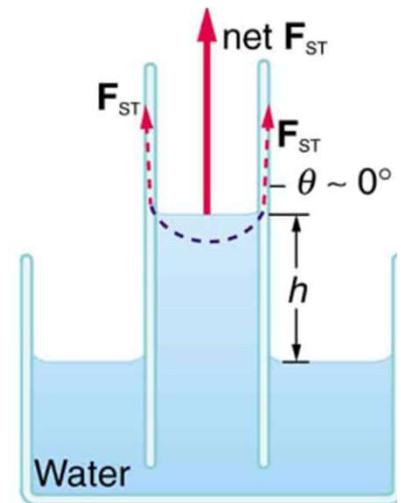
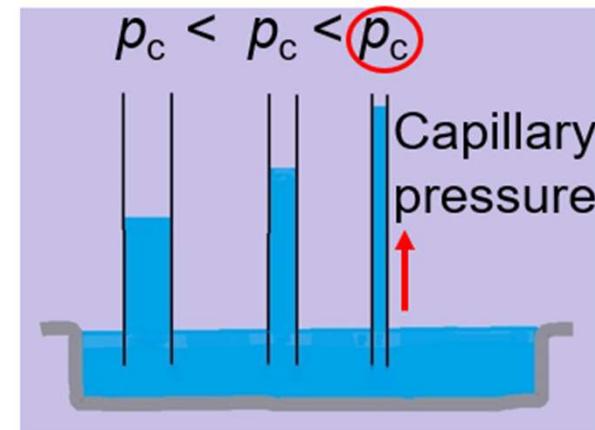
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Surface Tension and Curvature

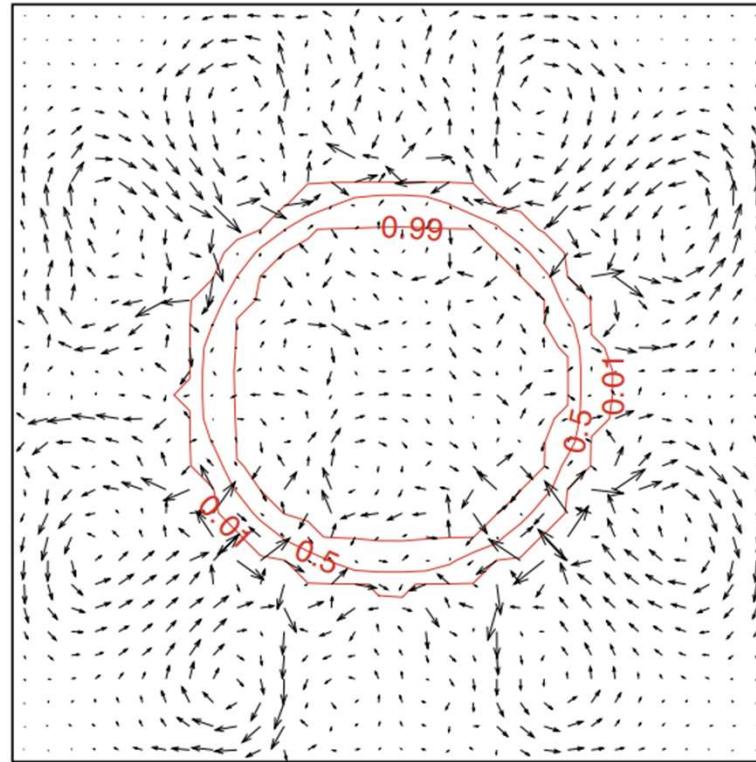
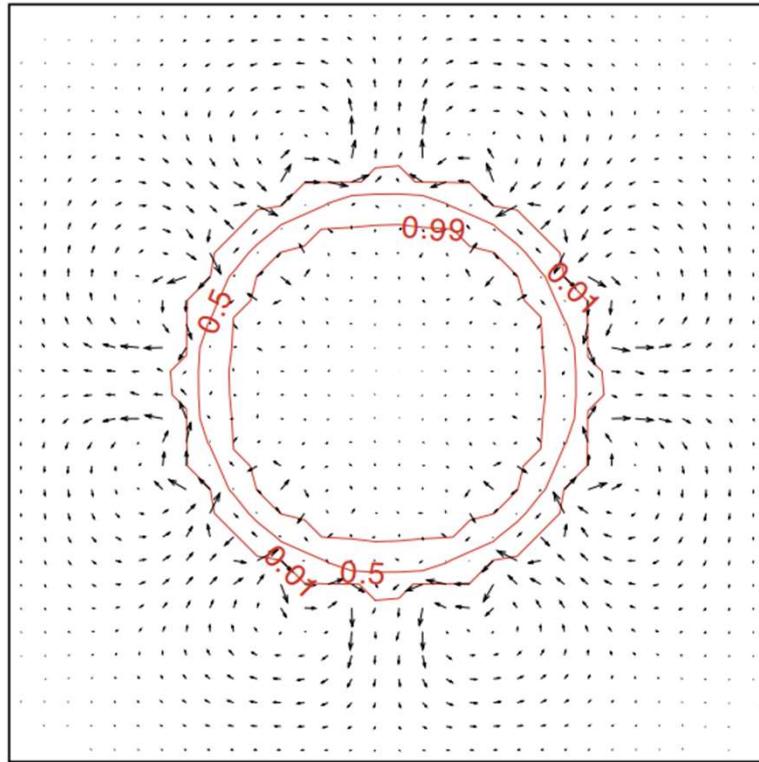
$$\frac{\partial(\rho\mathbf{u})}{\partial t} + \mathbf{u} \cdot \nabla(\rho\mathbf{u}) - \nabla \cdot (\mu \nabla \mathbf{u}) = -\nabla p_{rgh} - \mathbf{g} \cdot \mathbf{x} \nabla \rho + \mathbf{f}$$

- Capillary Pressure \rightarrow Surface Tension
- Errors:
 - Interface reconstruction and advection
 - Temporal and spatial discretization
 - External force discretization
- In flows that is driven by surface tension, these errors have a noticeable impact in predicted equilibrium height



Surface Tension and Curvature

- Imperfect discretization of the pressure discontinuity leads to artificially induced velocities around the interface causing parasitic currents



Surface Tension of Curvature

Surface Tension Force

$$\mathbf{F}_{st} = \sigma \kappa \hat{\mathbf{n}}_s \delta_s$$

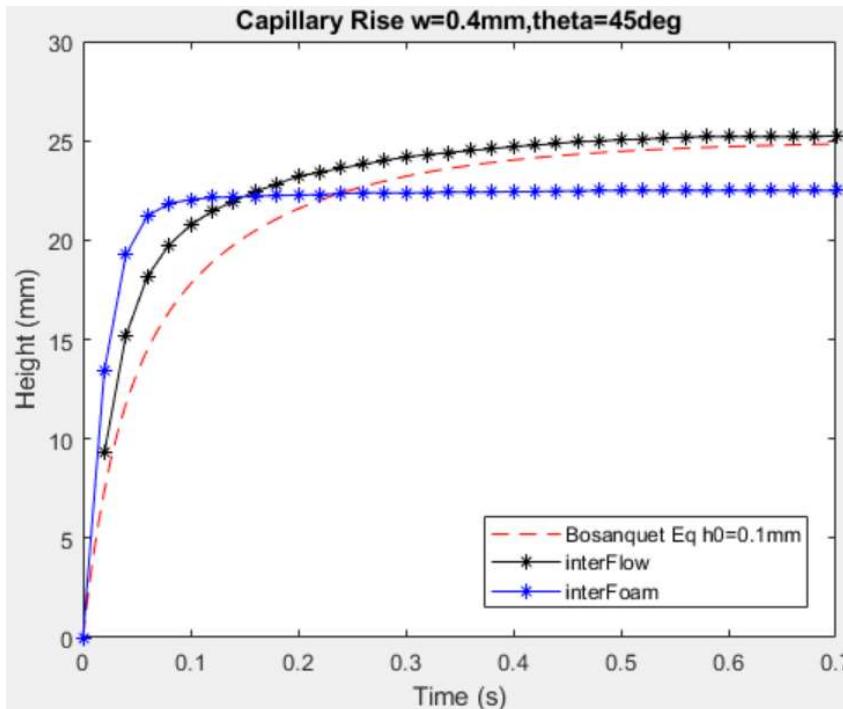
where

σ is surface tension

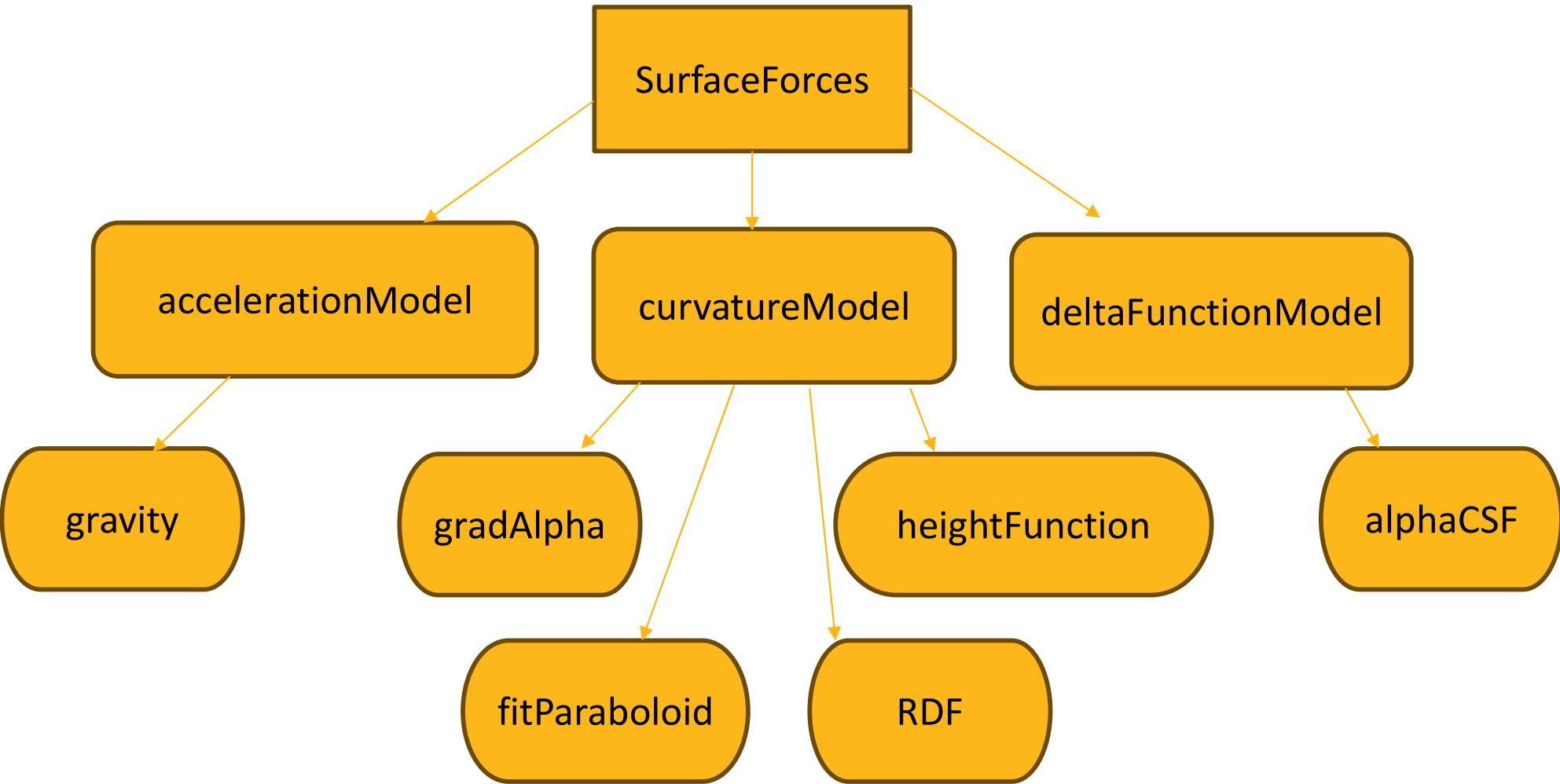
κ is the curvature

$\hat{\mathbf{n}}_s$ is the unit interface normal (pointing into the heavy liquid)

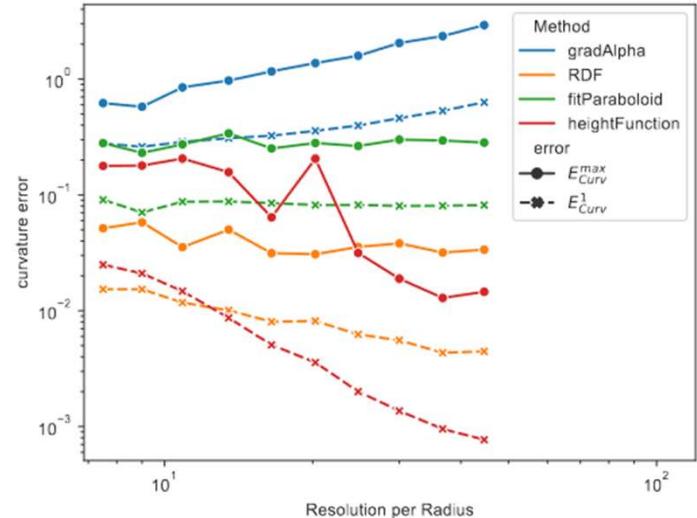
δ_s is the Dirac delta function and is only nonzero at the fluid interface



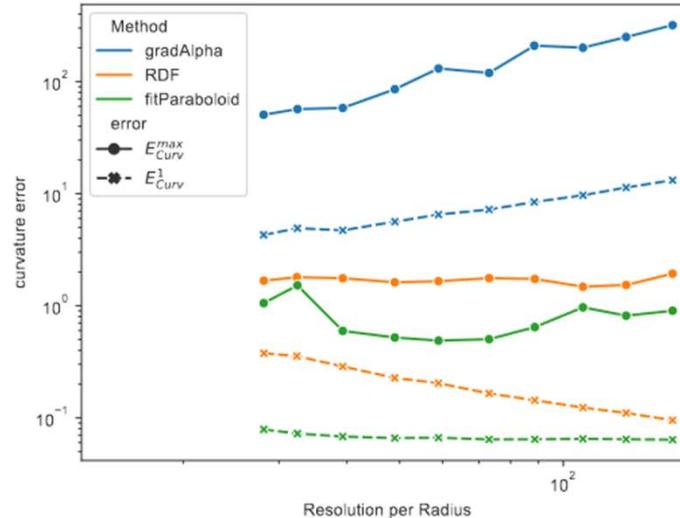
Overview of Surface Tension Module



Notes on Accuracy



(a) Hexahedral grid



(b) Tetrahedral grid

- For unstructured 3D grids, RDF is first order accurate for all grids and accuracy can be improved by using least square method for the gradient discretization
- Much better than gradAlpha (standard OpenFOAM)

Outline

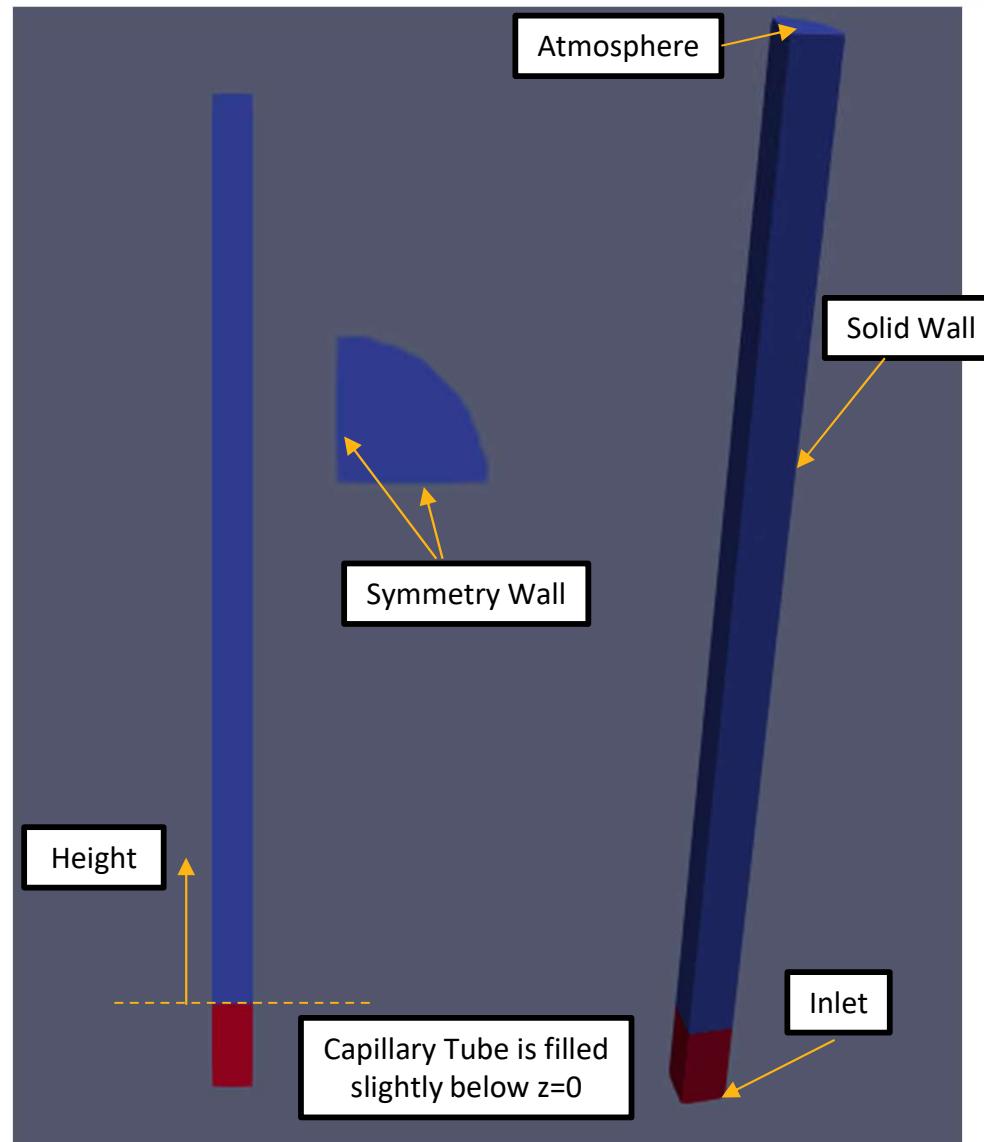
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Validation 1

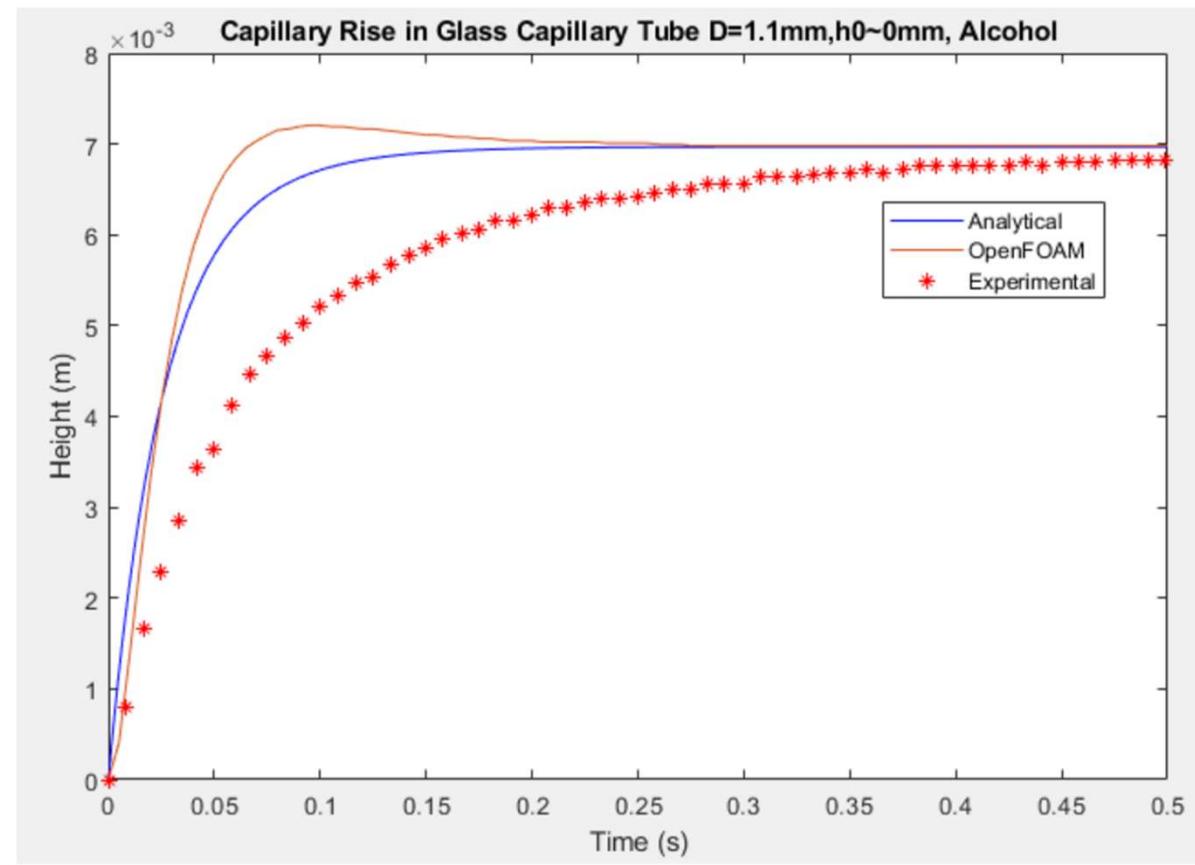
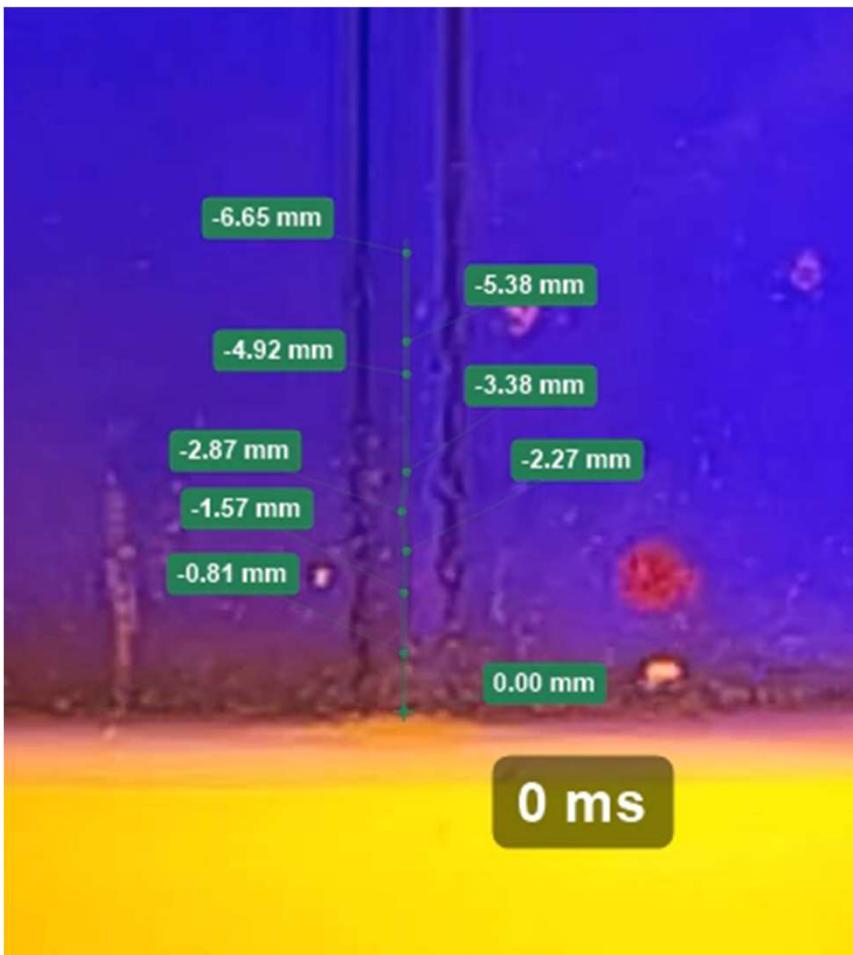
- Capillary rise of Alcohol in 1.1mm capillary tube
- Analytical solution is obtained by solving using ODE45

$$\frac{4\sigma \cos \theta}{D} = \frac{32\mu h}{D^2} \dot{h} + \rho \frac{d(h\dot{h})}{dt} + \rho gh$$

- A constant contact angle is assumed which will lead to error in the transient behaviour.



Results



Validation 2

- Capillary rise in Body Centered Cubic lattice column
- Lattice properties

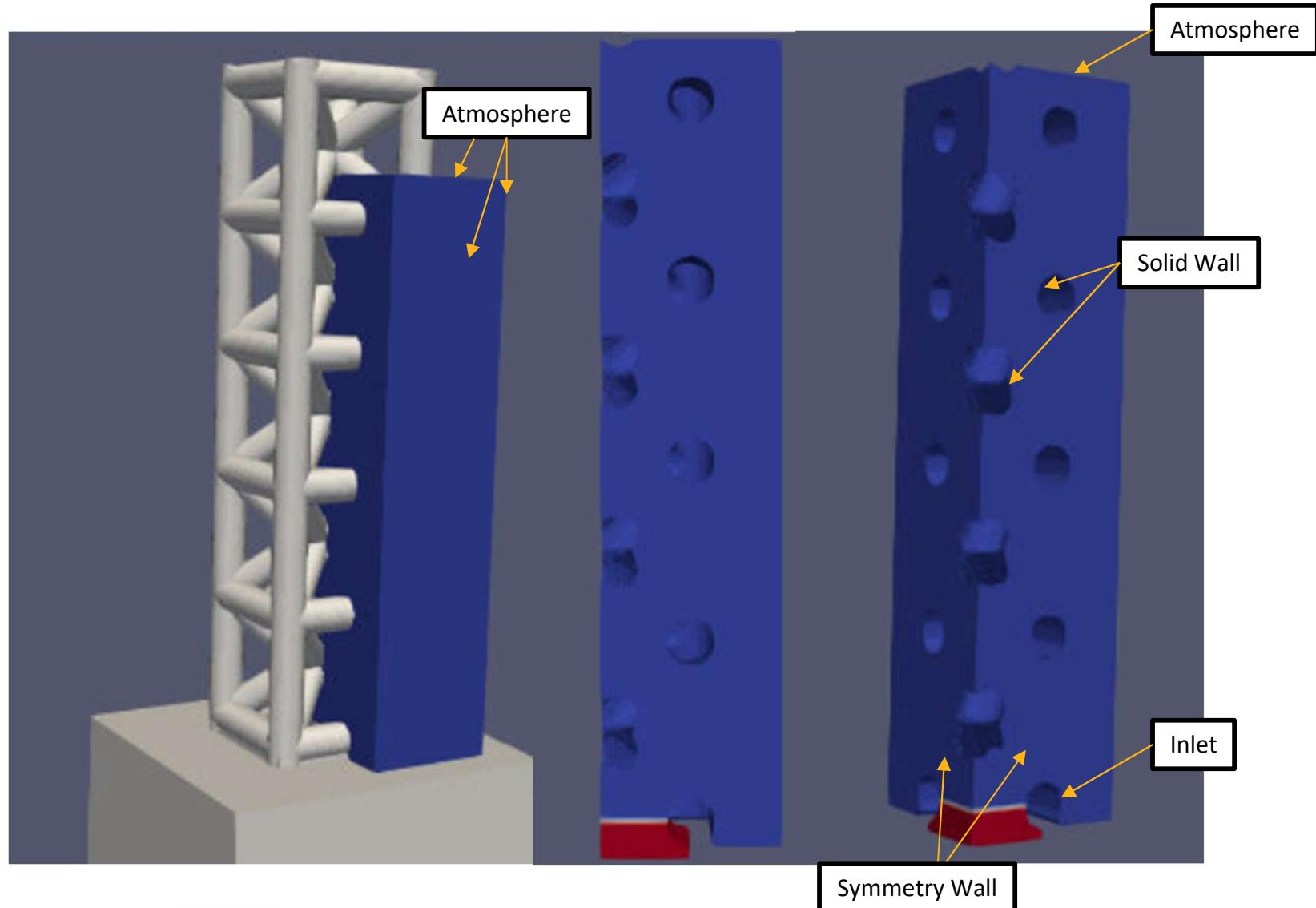
Unit length, $L=1.5\text{mm}$

Strut diameter, $D=0.4\text{mm}$



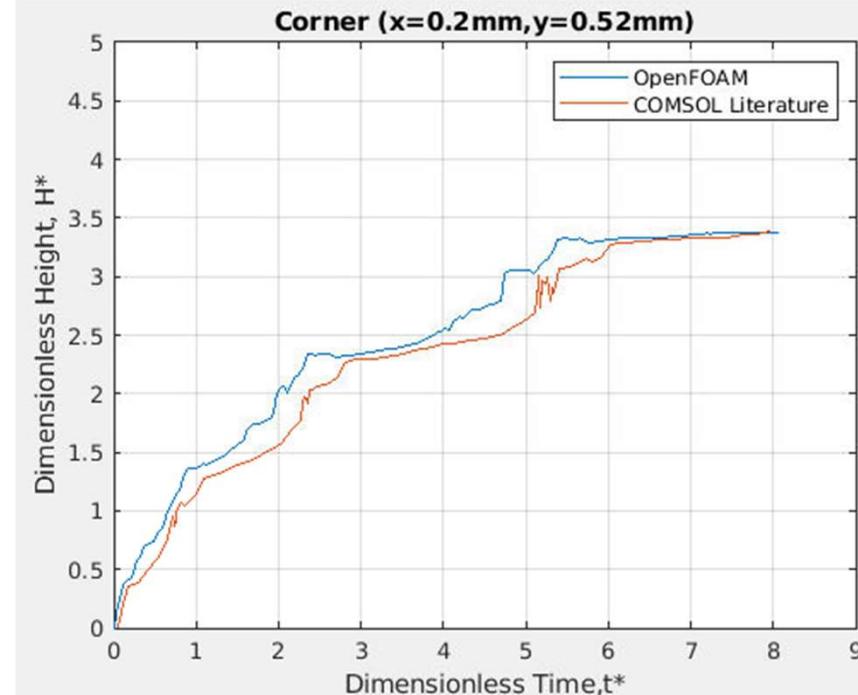
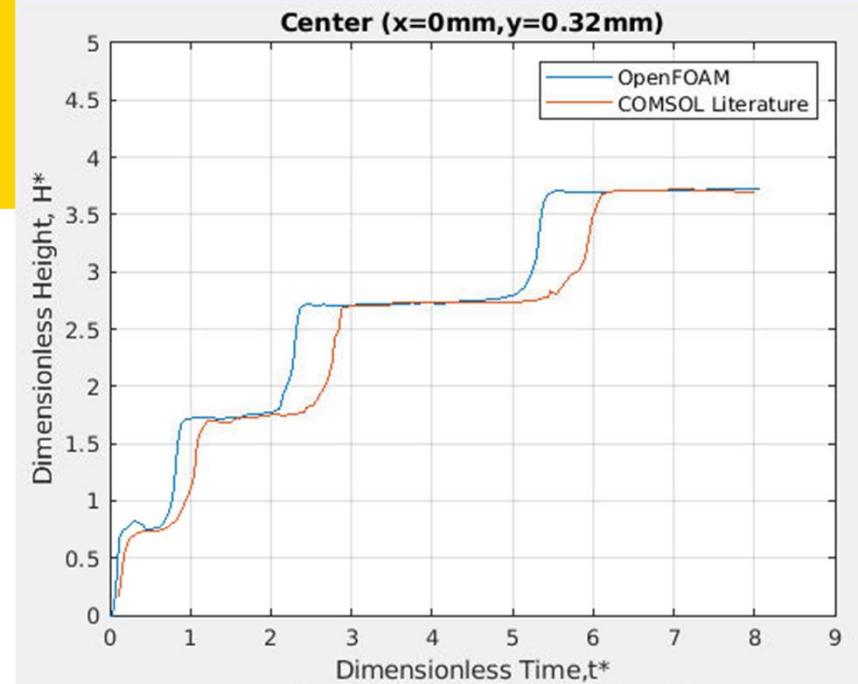
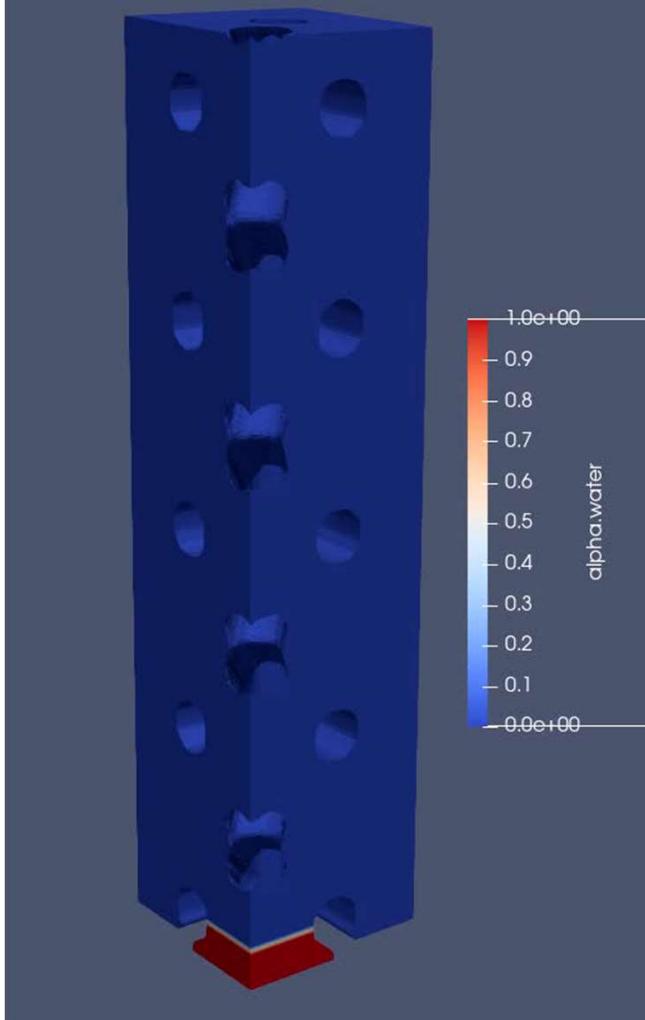
- Dimensionless height, $h^*=h/L$
- Dimensionless time, $t^*=t/\sqrt{L/g}$

Boundary Conditions



Results

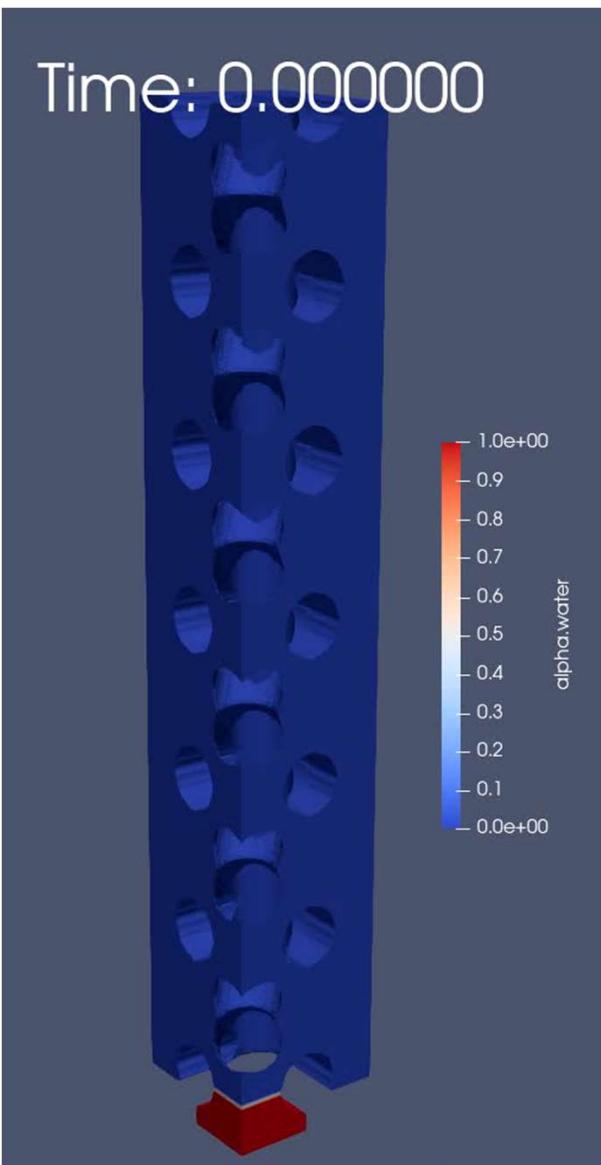
Time: 0.000000



Outline

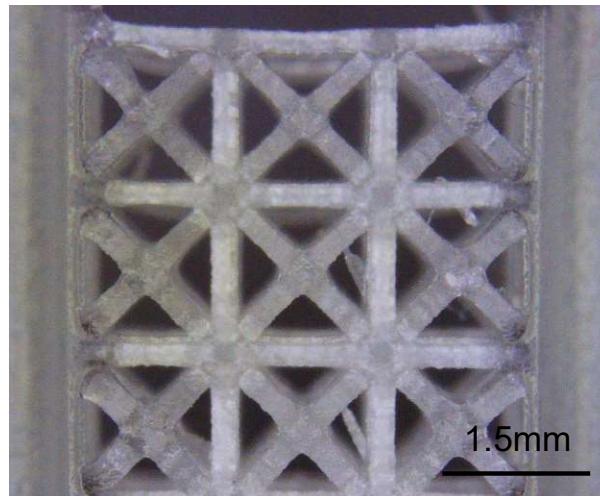
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Experimental Validation of Uniform and Non-Uniform Lattice

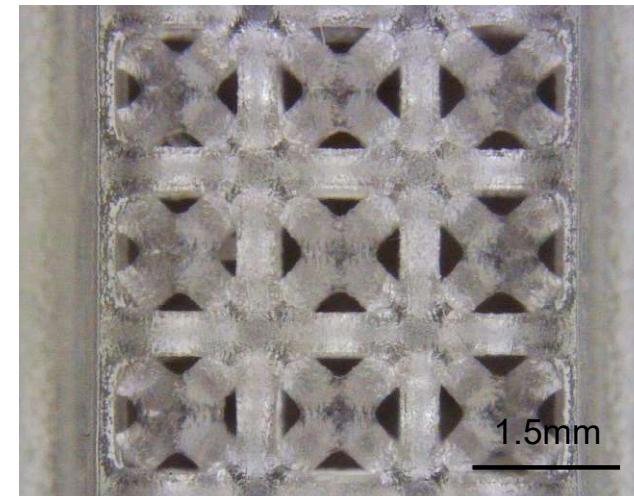


- Similar to BCC structure but vary strut diameter
- Reduction of pore size leads to air bubble entrapment. Mathematically, this is possible however, it is unlikely to appear in physical experiments
- More research on the limitation ongoing with in-house experiments
 - Effects of pore size
 - Effects of periodic versus atmospheric B.C.

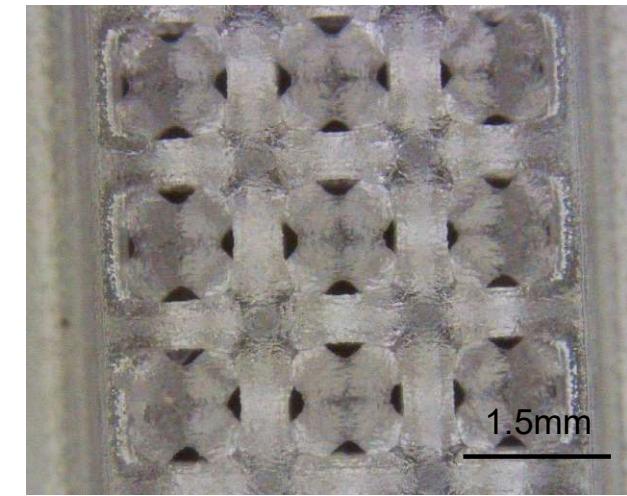
Experimental Validation of Uniform and Non-Uniform Lattice



150 μm



300 μm



400 μm

Rate-of-Rise Test: Working Principle

Hydrostatic Pressure

$$\Delta p_{hs} = \rho g h$$

Pressure due to friction
(Darcy's Law)

$$\Delta p_f = \frac{\varepsilon \mu}{K} h \frac{dh}{dt}$$

Capillary pressure
(Young-Laplace Equation)

$$\Delta p_{cap} = \frac{2\sigma \cos \theta}{r_p} = \frac{2\sigma}{r_{eff}}$$

$$h \frac{dh}{dt} = \frac{K}{\varepsilon \mu} \left(\frac{2\sigma}{r_{eff}} - \rho g h \right)$$

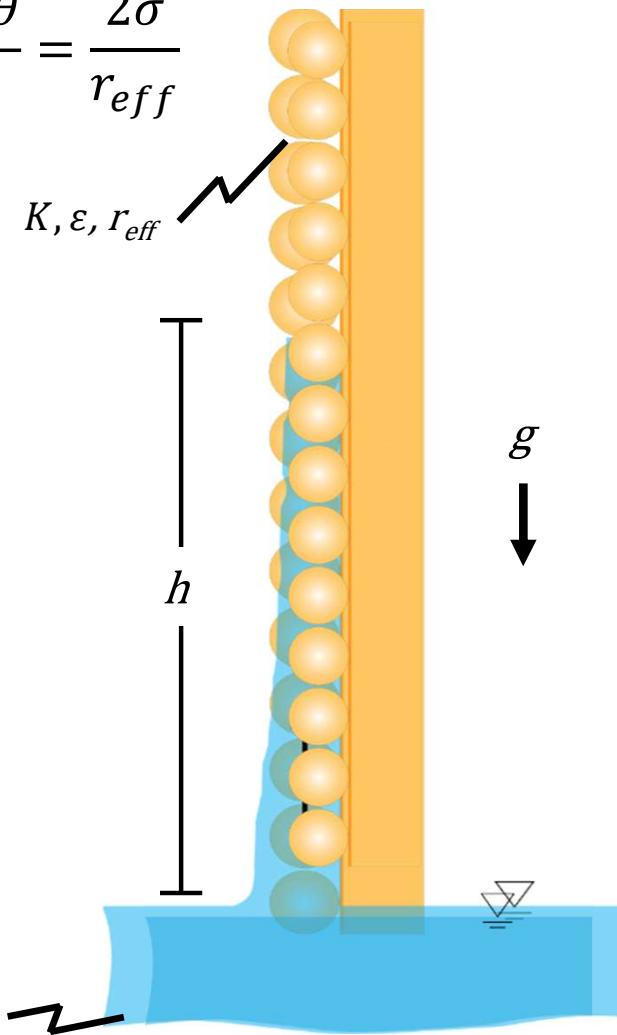
$\mu, \sigma, \rho \rightarrow$ Known fluid properties

$\varepsilon \rightarrow$ Porosity

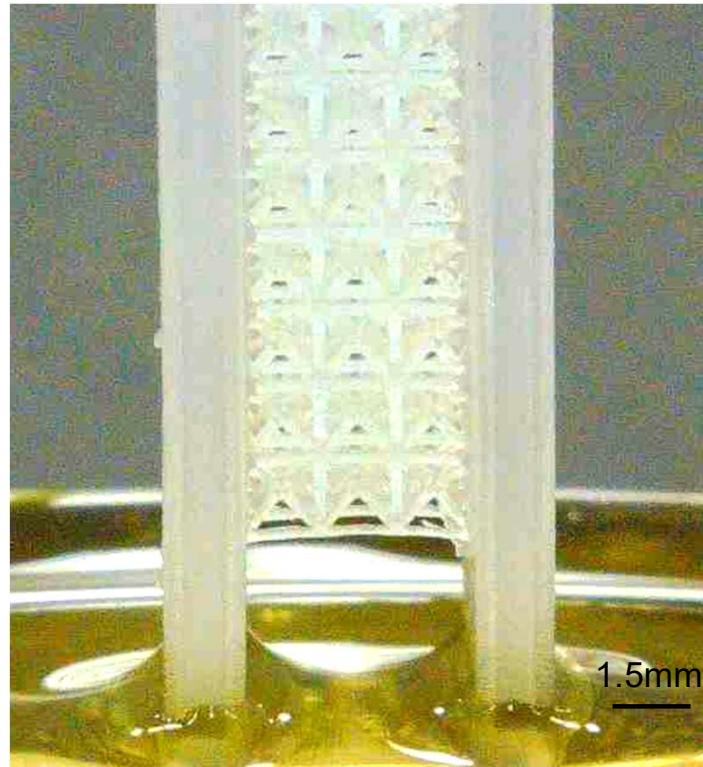
$r_{eff} \rightarrow$ Effective Pore Radius

$h \rightarrow$ Fluid height

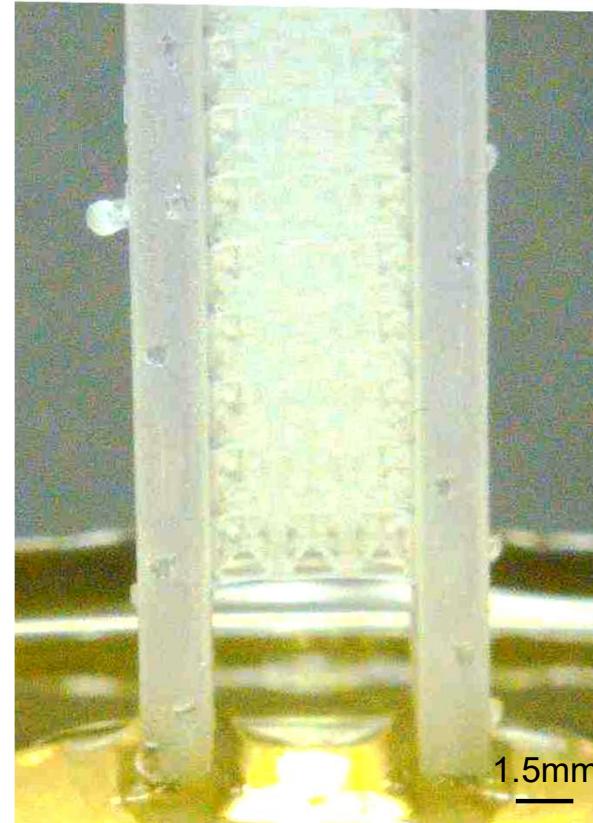
$K \rightarrow$ Permeability



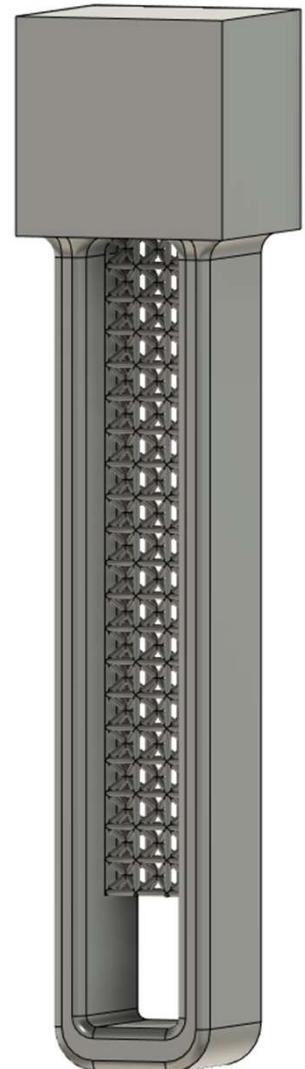
Experimental Validation of Uniform and Non-Uniform Lattice



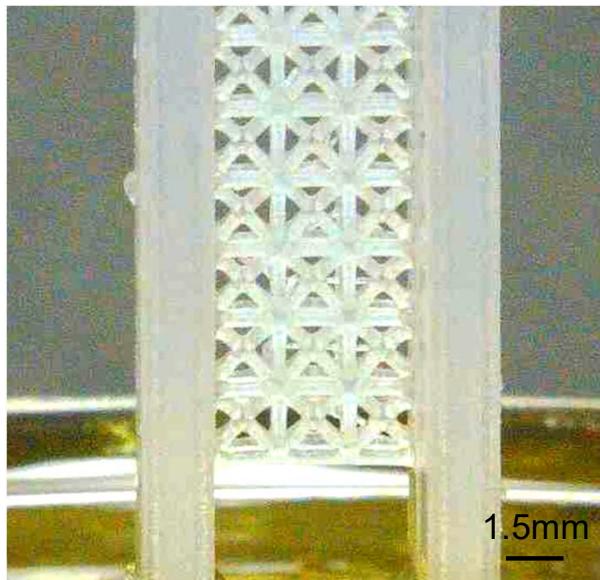
3x3 Lattice. Strut diameter=150 μm



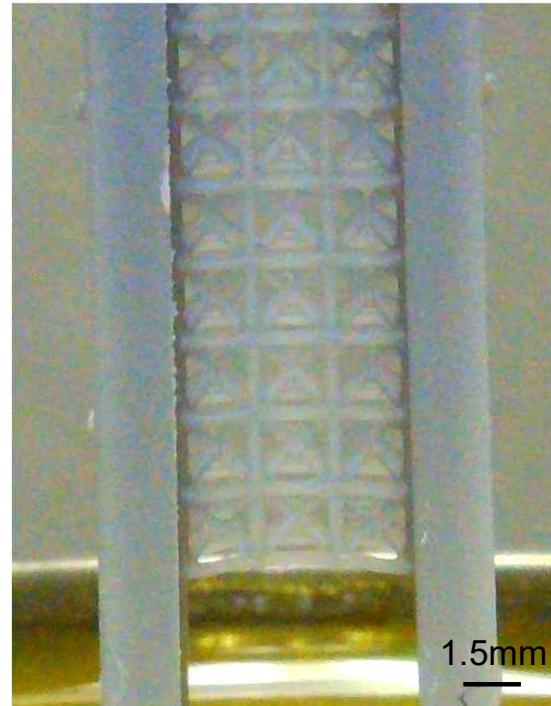
3x3 Lattice. Strut diameter=300 μm



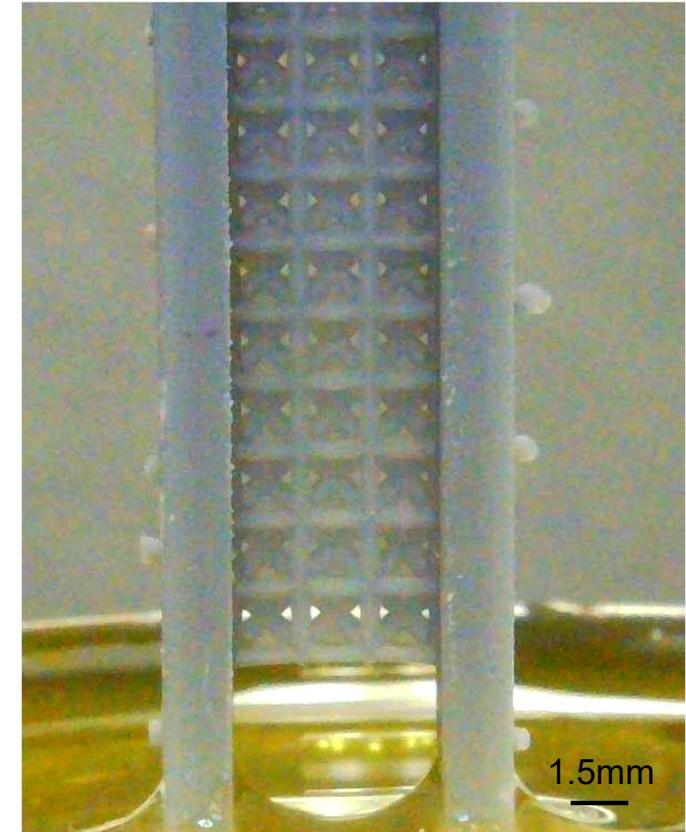
Experimental Validation of Uniform and Non-Uniform Lattice



1x3 Lattice. Strut diameter=150um



1x3 Lattice. Strut diameter=300um



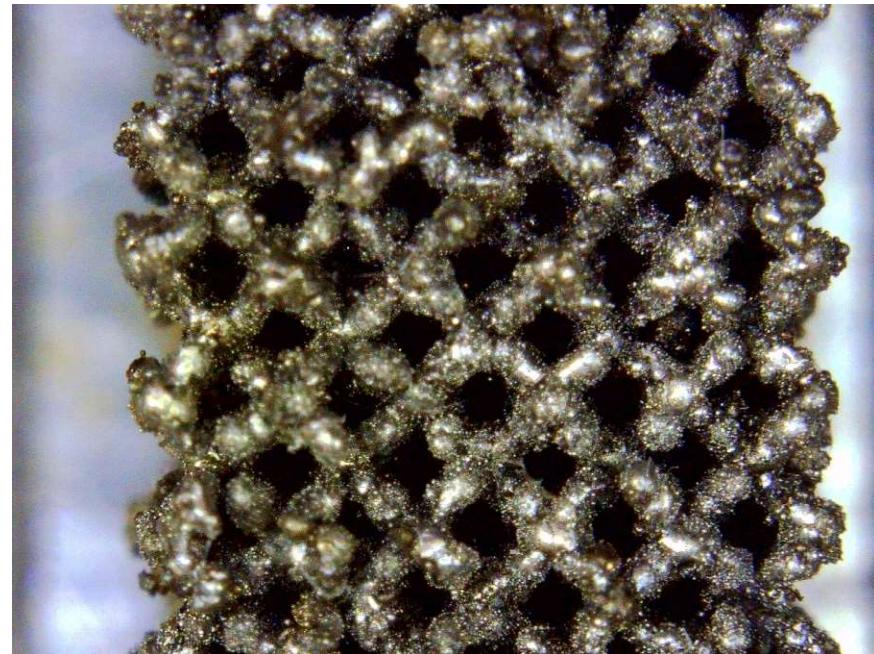
1x3 Lattice. Strut diameter=400um

Experimental Validation of Additively Manufactured Wicks



L5 Sample

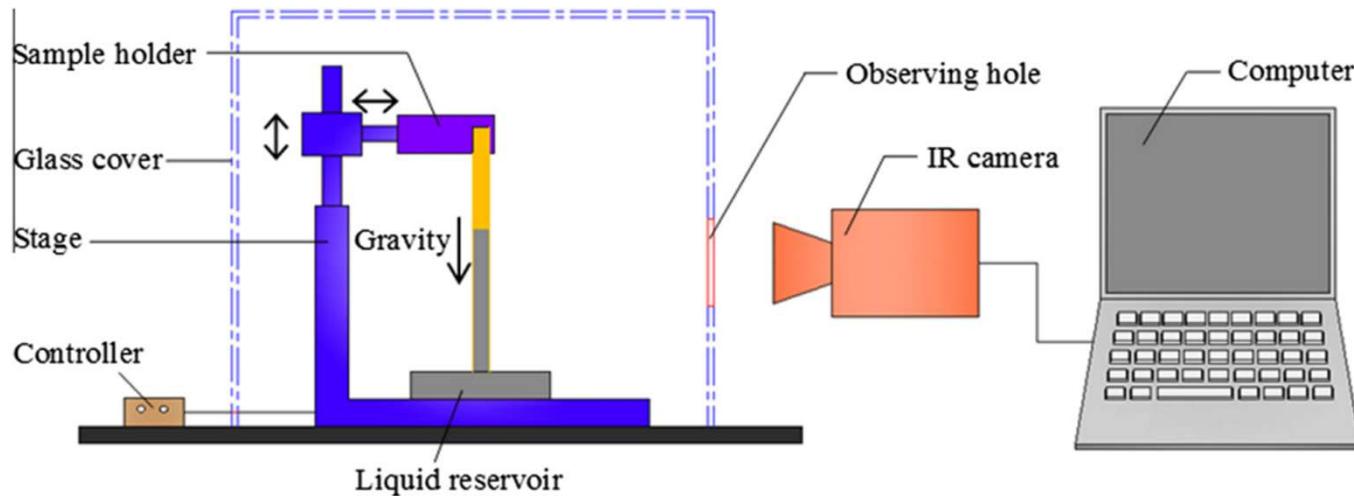
1mm



L3 Sample

1mm

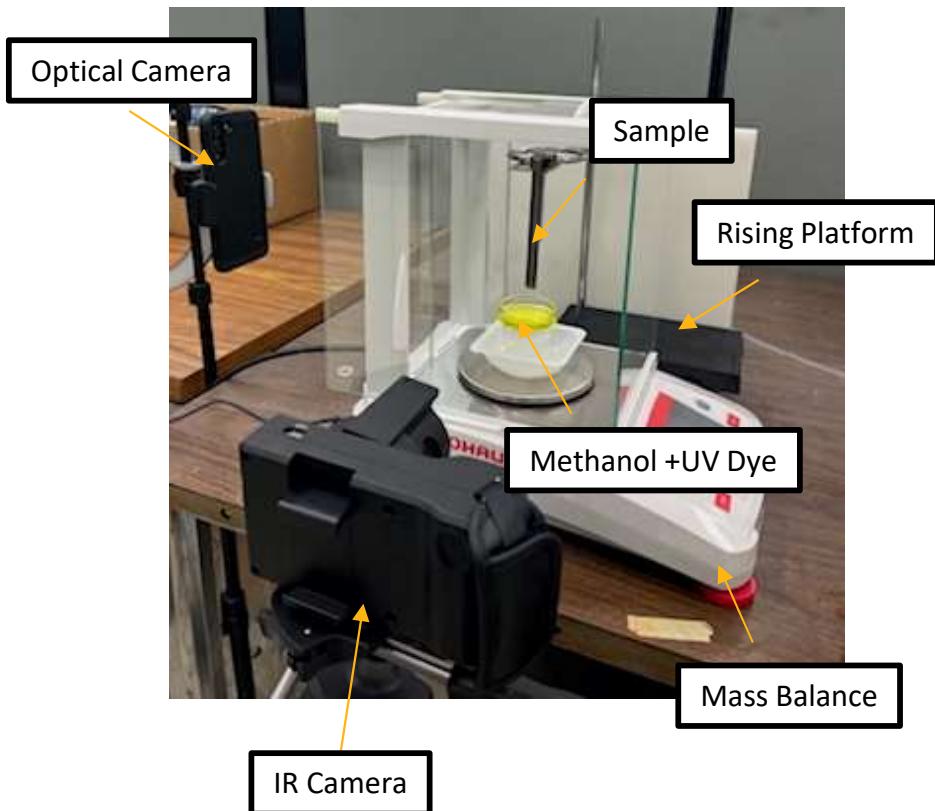
Experimental Validation of Additively Manufactured Wicks



- We cannot measure optically for most metal wicks
- Infrared camera can measure the difference in emissivity (and temperature) difference between dry area and wetted area
- However, camera must be extremely sensitive otherwise the interface will appear smeared

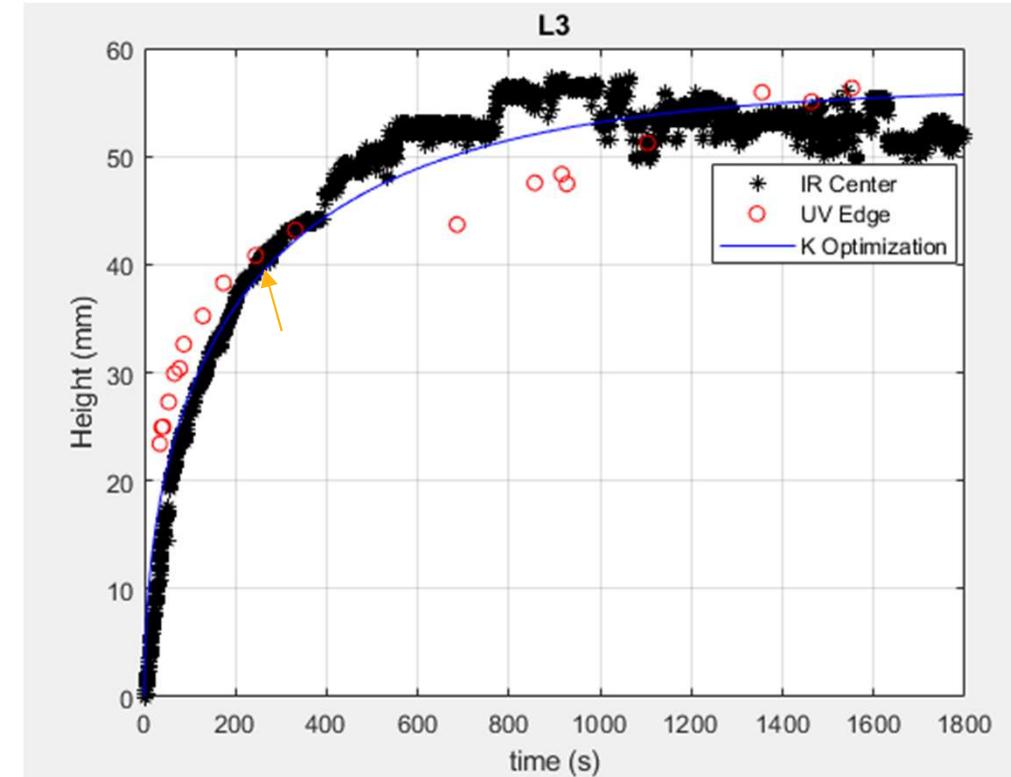
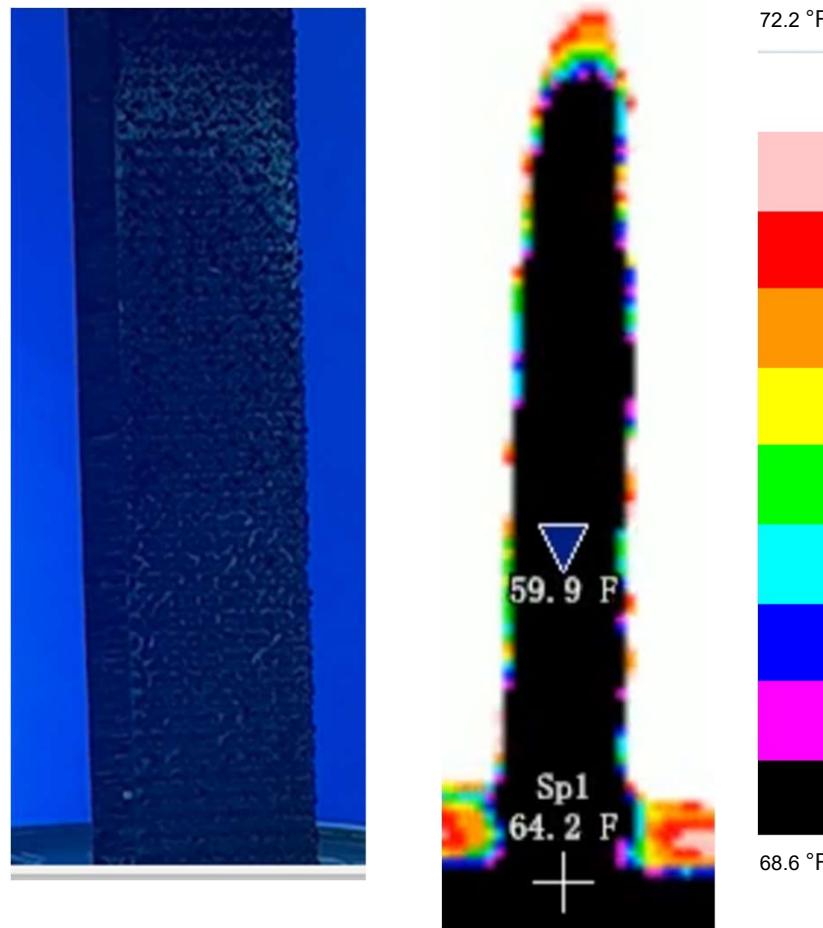
Experimental Validation of Additively Manufactured Wicks

- Methanol-UV solution shows the interface clearly at the edges, but cannot be seen in the center which is most important.
- IR camera shows the center of the wick but the interface is not clear.
- A hybrid approach is used where the optical camera corrects the temperature range of IR data, accurate within 0.3°F



Experimental Validation of Additively Manufactured Wicks

At t=245s



Conclusion

- Non-uniform wick structures facilitates high wickability for efficient high heat flux thermal management system.
- Modelling surface tension driven two phase flow is computationally challenging and modelling errors can have a large impact to the predicted equilibrium height.
- The standard OpenFOAM solver should not be applied for this application, instead twoPhaseFlow library should be used.
- CFD results are comparable to analytical solutions and literature numerical studies, however it is difficult the initial transient dynamics because of the assumption of static contact angle.
- Validation with AM wicks are needed, and the improved rate-of-rise setup have promising accuracy.



Questions?

Progress on Computational Modelling of Surface Tension Driven Flow and Experimental Measurements of Metallic Wicks

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12/1/2023



Supplemental Materials

We will assume 2 incompressible fluid in laminar flow with no phase change. α is the volume fraction of the tracked phase in a control volume, for simplicity, liquid fluid will be water ($\alpha=1$) and gaseous fluid will be air ($\alpha=0$).

Continuity Equation

$$\frac{\partial \rho}{\partial t} + \nabla \cdot \mathbf{u} = 0 \quad (1)$$

where \mathbf{u} is the fluid velocity

Momentum Equation

$$\frac{\partial(\rho\mathbf{u})}{\partial t} + \mathbf{u} \cdot \nabla(\rho\mathbf{u}) - \nabla \cdot (\mu \nabla \mathbf{u}) = -\nabla p + \rho(\mathbf{g} \cdot \mathbf{x}) + \mathbf{f} \quad (2)$$

where

ρ is the weighted density given as $\rho = \alpha\rho_{water} + (1 - \alpha)\rho_{air}$

μ is the weighted dynamic viscosity given as $\mu = \alpha\mu_{water} + (1 - \alpha)\mu_{air}$

p is the pressure

\mathbf{g} is the gravity vector

\mathbf{x} is the position vector

\mathbf{f} accounts for additional source terms such as surface tension

Numerically, it is easier to define boundary condition of pressure by considering an auxillary quantity p_{rgh} which excludes the hydrostatic contribution. It is defined as,

$$p_{rgh} = p - \rho(\mathbf{g} \cdot \mathbf{x}) \quad (3)$$

It can be shown that this leads to the momentum equation in the form,

$$\frac{\partial(\rho\mathbf{u})}{\partial t} + \mathbf{u} \cdot \nabla(\rho\mathbf{u}) - \nabla \cdot (\mu \nabla \mathbf{u}) = -\nabla p_{rgh} - \mathbf{g} \cdot \mathbf{x} \nabla \rho + \mathbf{f} \quad (4)$$

To transport the volume fraction, an additional advection equation is required,

$$\frac{\partial \alpha}{\partial t} + \nabla \cdot (\mathbf{u}\alpha) = 0 \quad (5)$$

Interface Advection

Consider a pure advection problem, momentum equation simplifies to

$$\frac{\partial \mathbf{u}}{\partial t} + \mathbf{u} \cdot \nabla \mathbf{u} = 0 \quad (6)$$

The solution does not depend of the density of the individual fluid. We will define a continuous indicator field based on the density of the fluids such that $H=1$ when cell is in water and $H=0$ when cell is in air

$$H(\mathbf{x}, t) = \frac{\rho(\mathbf{x}, t) - \rho_{air}}{\rho_{water} - \rho_{air}} \quad (7)$$

Based on this, we can define the volume fraction as,

$$\alpha_i(t) = \frac{1}{V_i} \int_{C_i} H(\mathbf{x}, t) dV \quad (8)$$

where i is the cell number, V is the volume and C_i is a single cell.

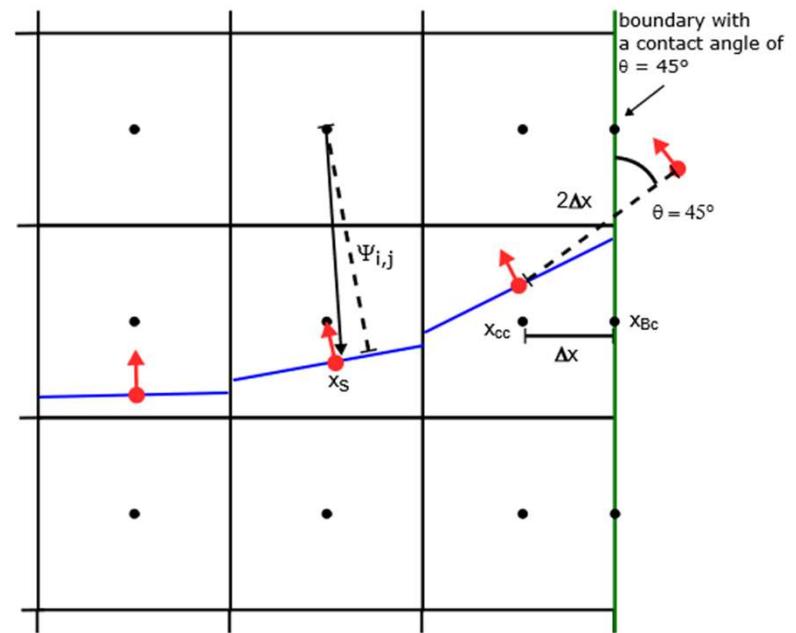
Substituting Eq. 8 into the discretized advection equation, we obtain

$$\alpha_i(t + \Delta t) = \alpha_i(t) - \frac{1}{V_i} \sum_{j \in B_i} s_{ij} \int_t^{t+\Delta t} \int_{F_j} H(\mathbf{x}, t) \mathbf{u}(\mathbf{x}, t) d\mathbf{S} dt \quad (9)$$

where B_i is the list of all faces F_j belonging to cell i , s_{ij} is used to orient the flux going out relative from cell i , dS is the unit normal surface vector

Reconstructed Distance Function

- At this point, we know have marched 1 iteration and we know what the interface looks like.
- To begin the next iteration, we need predict the surface tension force.
- Signed distance function is calculated for each cell close to the interface, Ψ_{cc}
- It is piecewise continuous function that calculates the distance nearest to a narrow band around the interface and changes sign when across interface
- Similar to idea with diagram except we are only going consider the closest interface centroid, \mathbf{x}_s



Reconstructed Distance Function

The value of ψ is calculated for each cell center (cc) near the interface

$$\psi_{cc} = \hat{\mathbf{n}}_s \cdot (\mathbf{x}_{cc} - \mathbf{x}_s) \quad (12)$$

where \mathbf{x}_s is the position of the centroid of the nearest interface segment

Then we calculate the gradient to find the unit vector perpendicular to the RDF

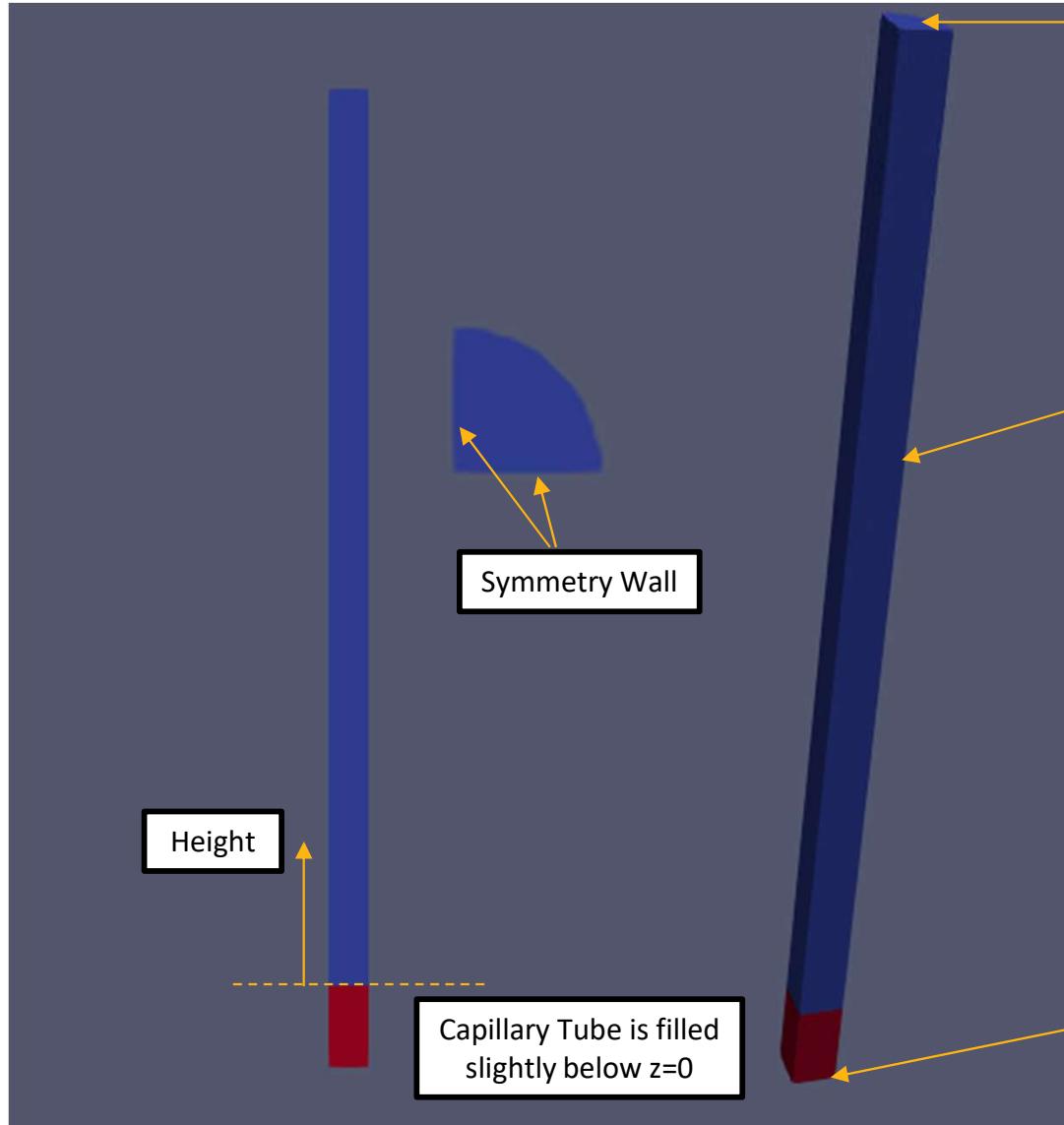
$$\hat{\mathbf{n}}_\psi = \frac{\nabla \psi}{|\nabla \psi|} \quad (13)$$

where $\hat{\mathbf{n}}_\psi$ is the normal vector of the RDF.

The curvature is computed as

$$\kappa = \text{trace}(\nabla \hat{\mathbf{n}}_\psi) \quad (14)$$

Boundary Conditions

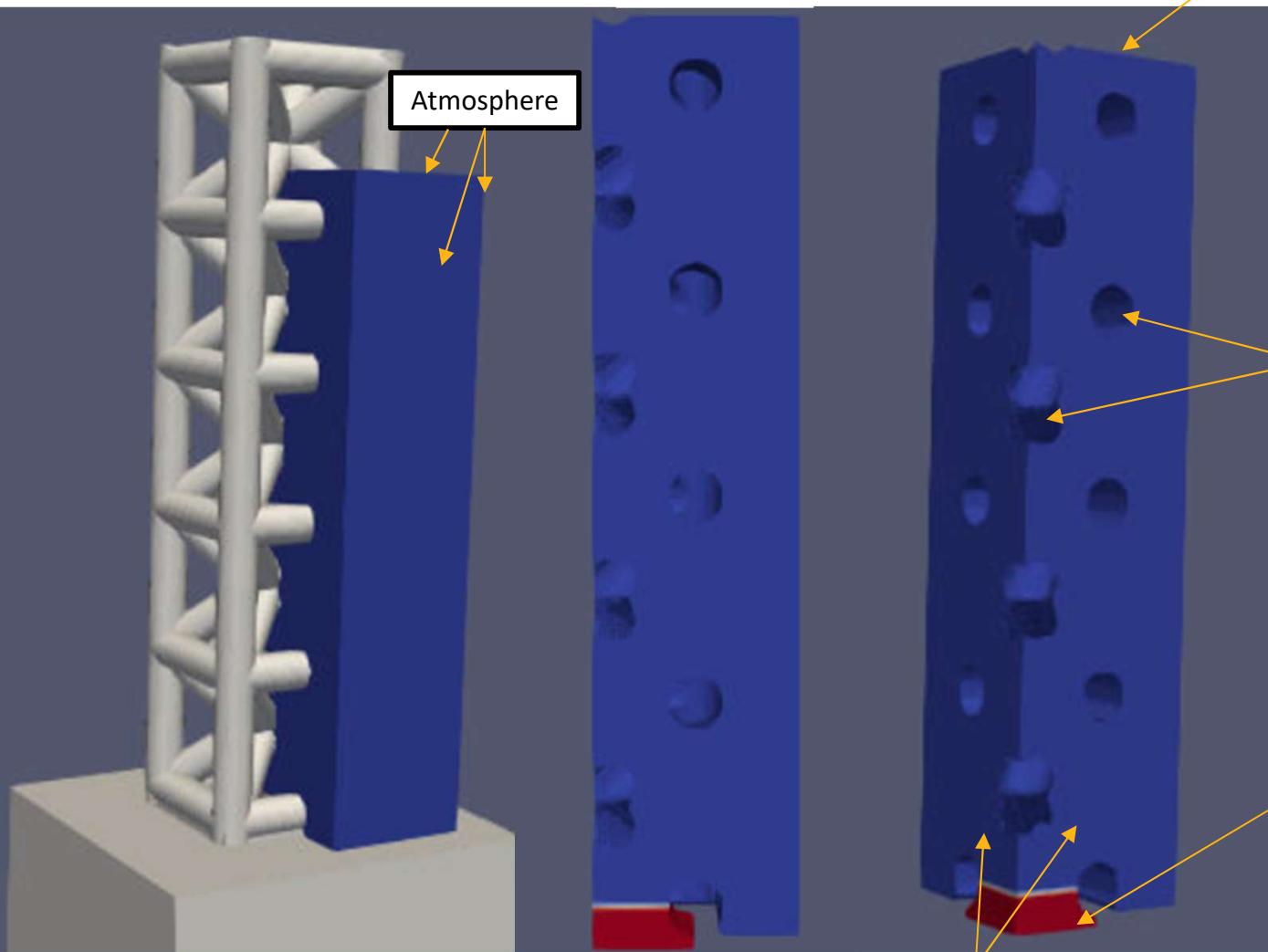


```
Atmosphere  
alpha  
zeroGradient;  
P_rgh  
fixedValue;  
uniform 0;  
U  
pressureInletOutletVelocity;  
uniform (0 0 0);
```

```
Capillary Tube  
alpha  
constantAlphaContactAngle;  
64;  
gradient;  
uniform 0;  
P_rgh  
fixedFluxPressure;  
U  
noSlip;
```

```
Inlet  
alpha  
inletOutlet;  
uniform 1;  
inletValue uniform 1;  
P_rgh  
fixedValue;  
uniform 0;  
U  
pressureInletOutletVelocity;  
uniform (0 0 0);
```

Boundary Conditions



Atmosphere

alpha

zeroGradient;

P_rgh

fixedValue;

uniform 0;

U

pressureInletOutletVelocity;

uniform (0 0 0);

Lattice

alpha

constantAlphaContactAngle;

22.5;

gradient;

uniform 0;

P_rgh

fixedFluxPressure;

U

noSlip;

Inlet

alpha

inletOutlet;

uniform 1;

inletValue uniform 1;

P_rgh

fixedValue;

uniform 0;

U

pressureInletOutletVelocity;

uniform (0 0 0);