

Path Constraints and Numerical Optimization

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- State and Control Equality Constraints
- Pseudoinverse
- State and Control Inequality Constraints
- Numerical Methods
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 - Steepest descent
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<http://www.princeton.edu/~stengel/MAE546.html>
<http://www.princeton.edu/~stengel/OptConEst.html>

*Minimization with
Equality Constraints*

Minimization with Equality Constraint on State and Control

$$\min_{\mathbf{u}(t)} J = \phi[\mathbf{x}(t_f)] + \int_{t_o}^{t_f} L[\mathbf{x}(t), \mathbf{u}(t)] dt$$

▪ subject to

Dynamic Constraint

$$\dot{\mathbf{x}}(t) = \mathbf{f}[\mathbf{x}(t), \mathbf{u}(t)], \quad \mathbf{x}(t_o) \text{ given}$$

$$\begin{aligned} \dim(\mathbf{x}) &= n \times 1 \\ \dim(\mathbf{f}) &= n \times 1 \\ \dim(\mathbf{u}) &= m \times 1 \end{aligned}$$

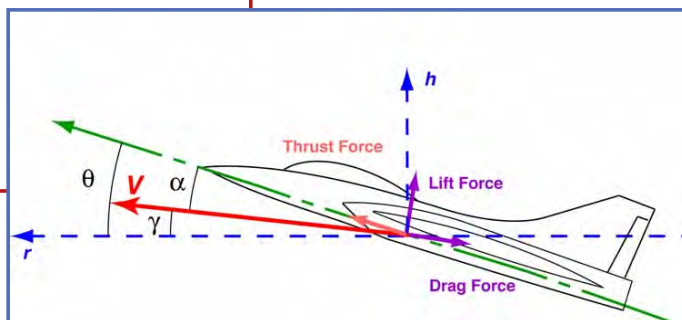
State/Control Equality Constraint

$$\mathbf{c}[\mathbf{x}(t), \mathbf{u}(t)] \equiv 0 \text{ in } (t_o, t_f); \quad \dim(\mathbf{c}) = (r \times 1) \leq (m \times 1)$$

Aeronautical Example: Longitudinal Point-Mass Dynamics

$$\begin{aligned} \dot{V} &= \left(T - C_D \frac{1}{2} \rho V^2 S \right) / m - g \sin \gamma \\ \dot{\gamma} &= \frac{1}{V} \left[\left(C_L \frac{1}{2} \rho V^2 S \right) / m - g \cos \gamma \right] \\ \dot{h} &= V \sin \gamma \\ \dot{r} &= V \cos \gamma \\ \dot{m} &= -(SFC)(T) \end{aligned}$$

$$\begin{aligned} x_1 = V &: \text{Velocity, m/s} \\ x_2 = \gamma &: \text{Flight path angle, rad} \\ x_3 = h &: \text{Height, m} \\ x_4 = r &: \text{Range, m} \\ x_5 = m &: \text{Mass, kg} \end{aligned}$$



Aeronautical Example: Longitudinal Point-Mass Dynamics

$u_1 = \delta T$: Throttle setting, %

$u_2 = \alpha$: Angle of attack, rad

$T = T_{\max_{SL}} (e^{-\beta h}) \delta T$: Thrust, N

$C_D = (C_{D_0} + \epsilon C_L^2)$ Drag coefficient

$C_L = C_{L_\alpha} \alpha$ = Lift coefficient

S = Reference area, m²

m = Vehicle mass, kg

ρ = Air density = $\rho_{SL} e^{-\beta h}$, kg/m³

g = Gravitational acceleration, m/s²

SFC = Specific Fuel Consumption, g/kN-s

Path Constraint Included in the Cost Function Hamiltonian

Constraint must be satisfied at every instant of the trajectory

Dimension of the constraint \leq dimension of the control

$$J_1 = \psi[\mathbf{x}(t_f)] + \int_{t_o}^{t_f} \{ L + \boldsymbol{\lambda}_1^T(t) [\mathbf{f} - \dot{\mathbf{x}}(t)] + \boldsymbol{\mu}^T \mathbf{c} \} dt$$

$$\mathbf{c}[\mathbf{x}(t), \mathbf{u}(t)] \equiv 0 \quad \text{in } (t_o, t_f)$$

The constraint is adjoined to the Hamiltonian

$$H \triangleq L + \boldsymbol{\lambda}_1^T \mathbf{f} + \boldsymbol{\mu}^T \mathbf{c}$$

$$\dim(\mathbf{x}) = \dim(\mathbf{f}) = \dim(\boldsymbol{\lambda}) = n \times 1$$

$$\dim(\mathbf{u}) = m \times 1$$

$$\dim(\mathbf{c}) = \dim(\boldsymbol{\mu}) = r \times 1, r \leq m$$

Euler-Lagrange Equations Including Equality Constraint

$$\boldsymbol{\lambda}(t_f) = \left\{ \frac{\partial \phi[\mathbf{x}(t_f)]}{\partial \mathbf{x}} \right\}^T$$

$$\dot{\boldsymbol{\lambda}} = - \left\{ \frac{\partial H[\mathbf{x}, \mathbf{u}, \boldsymbol{\lambda}, \mathbf{c}, \boldsymbol{\mu}, t]}{\partial \mathbf{x}} \right\}^T = - \left[\frac{\partial L}{\partial \mathbf{x}} + \boldsymbol{\lambda}^T \frac{\partial \mathbf{f}}{\partial \mathbf{x}} + \boldsymbol{\mu}^T \frac{\partial \mathbf{c}}{\partial \mathbf{x}} \right]^T = - \left[L_{\mathbf{x}}^T + \mathbf{F}^T \boldsymbol{\lambda} + \left(\frac{\partial \mathbf{c}}{\partial \mathbf{x}} \right)^T \boldsymbol{\mu} \right]$$

$$\left\{ \frac{\partial H[\mathbf{x}, \mathbf{u}, \boldsymbol{\lambda}, \mathbf{c}, \boldsymbol{\mu}, t]}{\partial \mathbf{u}} \right\}^T = - \left[\left(\frac{\partial L}{\partial \mathbf{u}} \right)^T + \mathbf{G}^T \boldsymbol{\lambda} + \left(\frac{\partial \mathbf{c}}{\partial \mathbf{u}} \right)^T \boldsymbol{\mu} \right] = \mathbf{0}$$

$$\mathbf{c}[\mathbf{x}(t), \mathbf{u}(t)] \equiv \mathbf{0} \quad \text{in } (t_o, t_f)$$

$$\begin{aligned} \dim(\mathbf{x}) &= \dim(\boldsymbol{\lambda}) = n \times 1 \\ \dim(\mathbf{F}) &= n \times n \\ \dim(\mathbf{G}) &= n \times m \\ \dim(\mathbf{u}) &= m \times 1 \\ \dim(\mathbf{c}) &= \dim(\boldsymbol{\mu}) = r \times 1, r \leq m \end{aligned}$$

No Optimization When $r = m$

- Control entirely specified by constraint
 - m unknowns, m equations

$$\mathbf{c}[\mathbf{x}(t), \mathbf{u}(t)] \equiv \mathbf{0} \Rightarrow \mathbf{u}(t) = fcn[\mathbf{x}(t)]$$

Example

$$\begin{aligned} \mathbf{c} &= \mathbf{A}\mathbf{x}(t) + \mathbf{B}\mathbf{u}(t) = \mathbf{0}; \quad \dim(\mathbf{x}) = n \times 1; \quad \dim(\mathbf{c}) = \dim(\mathbf{u}) = m \times 1 \\ \dim(\mathbf{A}) &= m \times n; \quad \dim(\mathbf{B}) = m \times m \\ \mathbf{u}(t) &= -\mathbf{B}^{-1}\mathbf{A}\mathbf{x}(t) \end{aligned}$$

- Constraint Lagrange multiplier is irrelevant but can be expressed
 - from $dH/d\mathbf{u} = \mathbf{0}$,

$$\boldsymbol{\mu} = - \left(\frac{\partial \mathbf{c}}{\partial \mathbf{u}} \right)^{-T} \left[\left(\frac{\partial L}{\partial \mathbf{u}} \right)^T + \mathbf{G}^T \boldsymbol{\lambda} \right]$$

$$\left(\frac{\partial \mathbf{c}}{\partial \mathbf{u}} \right) \text{ is square and non-singular}$$

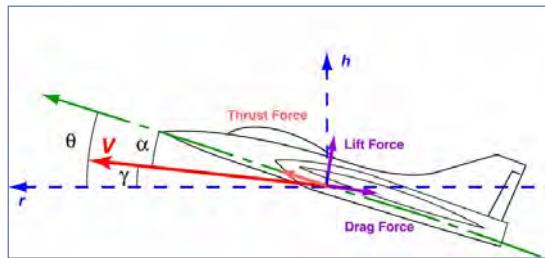
No Optimization When $r = m$

MAINTAIN CONSTANT VELOCITY AND FLIGHT PATH ANGLE

$$\mathbf{c}[\mathbf{x}(t), \mathbf{u}(t)] \equiv \mathbf{0} = \begin{bmatrix} 0 = \dot{V} = \left[T_{\max} \delta T - \left(C_{D_o} + \epsilon C_L^2 \right) \frac{1}{2} \rho V^2 S \right] / m - g \sin \gamma \\ 0 = \dot{\gamma} = \frac{1}{V} \left[\left(C_{L_\alpha} \alpha \frac{1}{2} \rho V^2 S \right) / m - g \cos \gamma \right] \end{bmatrix}$$

$\Rightarrow \mathbf{u}(t) = fcn[\mathbf{x}(t)]$

$V(0) = V_{desired}$
 $\gamma(0) = \gamma_{desired}$



Effect of Constraint Dimensionality:

$$r < m$$

MINIMIZE FUEL AND CONTROL USE WHILE MAINTAINING
CONSTANT FLIGHT PATH ANGLE

$$\min_{\mathbf{u}(t)} J = \int_{t_o}^{t_f} \left(q \dot{m}^2 + r_1 u_1^2 + r_2 u_2^2 \right) dt$$

$$\mathbf{c}[\mathbf{x}(t), \mathbf{u}(t)] \equiv 0 = \dot{\gamma} = \frac{1}{V(t)} \left[\frac{\left(C_{L_\alpha} \alpha \frac{1}{2} \rho V^2(t) S \right)}{m} - g \cos \gamma(t) \right]$$

$\gamma(0) = \gamma_{desired}$

Effect of Constraint Dimensionality: $r < m$

$$\begin{aligned}\dim(\mathbf{x}) &= n \times 1 \\ \dim(\mathbf{u}) &= m \times 1 \\ \dim(\mathbf{c}) &= r \times 1\end{aligned}$$

$$\begin{aligned}\left(\frac{\partial \mathbf{c}}{\partial \mathbf{u}}\right) &\text{ is not square when } r < m \\ \therefore \left(\frac{\partial \mathbf{c}}{\partial \mathbf{u}}\right) &\text{ is not strictly invertible}\end{aligned}$$

- **Three approaches to constrained optimization**
 - Algebraic solution for r control variables using an invertible subset of the constraint
 - Pseudoinverse of control effect
 - “Soft” constraint

$$\begin{aligned}\dim(\mathbf{u}) &= m \times 1 \\ \dim(\mathbf{c}) &= r \times 1\end{aligned}$$

Effect of Constraint Dimensionality: $r < m$

Algebraic solution for r control variables using an invertible subset of the constraint

Example 1

$$\begin{aligned}\dim(\mathbf{x}) &= n \times 1; \quad \dim(\mathbf{A}_r) = r \times n \\ \dim(\mathbf{u}) &= m \times 1; \quad \dim(\mathbf{u}_r) = r \times 1; \quad \dim(\mathbf{B}_r) = r \times r \\ \mathbf{c} &= \mathbf{A}_r \mathbf{x}(t) + \mathbf{B}_r \mathbf{u}_r(t) = \mathbf{0}; \quad \det(\mathbf{B}_r) \neq 0 \\ \mathbf{u}_r(t) &= -\mathbf{B}_r^{-1} \mathbf{A}_r \mathbf{x}(t)\end{aligned}$$

Example 2

$$\begin{aligned}\dim(\mathbf{u}) &= m \times 1; \quad \dim(\mathbf{u}_r) = r \times 1; \quad \dim(\mathbf{B}_1) = r \times r \\ \mathbf{c} &= \begin{bmatrix} \mathbf{B}_1 & \mathbf{B}_2 \end{bmatrix} \begin{bmatrix} \mathbf{u}_r \\ \mathbf{u}_{m-r} \end{bmatrix} = \mathbf{B}_1 \mathbf{u}_r + \mathbf{B}_2 \mathbf{u}_{m-r} = \mathbf{0}; \quad \det(\mathbf{B}_1) \neq 0 \\ \mathbf{u}_r(t) &= -\mathbf{B}_1^{-1} \mathbf{B}_2 \mathbf{u}_{m-r}(t)\end{aligned}$$

Second Approach: Satisfy Constraint Using Left Pseudoinverse: $r < m$

$\left[\left(\frac{\partial \mathbf{c}}{\partial \mathbf{u}} \right)^T \right]^L$ is the **left pseudoinverse** of control sensitivity

$$\dim \left[\left(\frac{\partial \mathbf{c}}{\partial \mathbf{u}} \right)^T \right]^L = r \times m$$

Lagrange multiplier

$$\mu_L = - \left[\left(\frac{\partial \mathbf{c}}{\partial \mathbf{u}} \right)^T \right]^L \left[\left(\frac{\partial L}{\partial \mathbf{u}} \right)^T + \mathbf{G}^T \boldsymbol{\lambda} \right]$$

Pseudoinverse of Matrix

$$\mathbf{y} = \mathbf{A} \mathbf{x}$$

$$\begin{aligned} \dim(\mathbf{x}) &= r \times 1 \\ \dim(\mathbf{y}) &= m \times 1 \end{aligned}$$

$$(m \times 1) = (m \times r)(r \times 1)$$

$r = m$, \mathbf{A} is square and non-singular

$$\mathbf{x} = \mathbf{A}^{-1} \mathbf{y}$$

$$(r \times 1) = (r \times m)(m \times 1) = (r \times r)(r \times 1)$$

$r \neq m$, \mathbf{A} is not square

Use **pseudoinverse** of \mathbf{A}

$$\mathbf{x} = \mathbf{A}^\# \mathbf{y} = \mathbf{A}^\dagger \mathbf{y}$$

$$(r \times 1) = (r \times m)(m \times 1)$$

Maximum rank of \mathbf{A} is r or m ,
whichever is smaller

See http://en.wikipedia.org/wiki/Moore-Penrose_pseudoinverse

Left Pseudoinverse

$$\dim(\mathbf{x}) = r \times 1$$

$$\dim(\mathbf{y}) = m \times 1$$

Maximum rank of \mathbf{A} is r or m ,
whichever is smaller

$$\dim(\mathbf{A}^T \mathbf{A}) = r \times r$$

$$\dim(\mathbf{A} \mathbf{A}^T) = m \times m$$

$r < m$, Left pseudoinverse is appropriate

$$\mathbf{A} \mathbf{x} = \mathbf{y}$$

$$\mathbf{A}^T \mathbf{A} \mathbf{x} = \mathbf{A}^T \mathbf{y}$$

Averaging
solution

$$\mathbf{x} = (\mathbf{A}^T \mathbf{A})^{-1} \mathbf{A}^T \mathbf{y}$$

$$\mathbf{A}^L \triangleq (\mathbf{A}^T \mathbf{A})^{-1} \mathbf{A}^T$$

$$\mathbf{x} = \mathbf{A}^L \mathbf{y}$$

$$(r \times 1) = (r \times m)(m \times 1)$$

$$\dim(\mathbf{x}) = r \times 1$$

$$\dim(\mathbf{y}) = m \times 1$$

Right Pseudoinverse

$r > m$, Right pseudoinverse is appropriate

$$\dim(\mathbf{A}^T \mathbf{A}) = r \times r$$

$$\dim(\mathbf{A} \mathbf{A}^T) = m \times m$$

$$\mathbf{A} \mathbf{x} = \mathbf{y} = \mathbf{I} \mathbf{y}$$

$$\mathbf{A} \mathbf{x} = (\mathbf{A} \mathbf{A}^T)(\mathbf{A} \mathbf{A}^T)^{-1} \mathbf{y}$$

$$= \mathbf{I} \mathbf{y}$$

Minimum Euclidean
error norm solution

$$\mathbf{A} \mathbf{x} = \mathbf{A} \left[\mathbf{A}^T (\mathbf{A} \mathbf{A}^T)^{-1} \mathbf{y} \right]$$

$$\mathbf{x} = \mathbf{A}^T (\mathbf{A} \mathbf{A}^T)^{-1} \mathbf{y}$$

$$\mathbf{A}^R \triangleq \mathbf{A}^T (\mathbf{A} \mathbf{A}^T)^{-1}$$

$$\mathbf{x} = \mathbf{A}^R \mathbf{y}$$

$$(m \times 1) = (m \times r)(r \times 1)$$

Left Pseudoinverse Example

$$\mathbf{Ax} = \mathbf{y}, \quad r < m$$

$$\mathbf{x} = (\mathbf{A}^T \mathbf{A})^{-1} \mathbf{A}^T \mathbf{y}$$

$$\begin{aligned} \begin{bmatrix} 1 \\ 3 \end{bmatrix} x &= \begin{bmatrix} 2 \\ 6 \end{bmatrix} \\ x &= \left\{ \begin{bmatrix} 1 & 3 \end{bmatrix} \begin{bmatrix} 1 \\ 3 \end{bmatrix} \right\}^{-1} \begin{bmatrix} 1 & 3 \end{bmatrix} \begin{bmatrix} 2 \\ 6 \end{bmatrix} \\ &= \frac{1}{10}(20) = 2 \end{aligned}$$

Unique solution

Right Pseudoinverse Example

$$\mathbf{Ax} = \mathbf{y}, \quad r > m$$

$$\mathbf{x} = \mathbf{A}^T (\mathbf{A} \mathbf{A}^T)^{-1} \mathbf{y}$$

$$\begin{aligned} \mathbf{Ax} = \mathbf{y}; \quad \begin{bmatrix} 1 & 3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} &= 14 \\ \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} &= \begin{bmatrix} 1 \\ 3 \end{bmatrix} \left\{ \begin{bmatrix} 1 & 3 \end{bmatrix} \begin{bmatrix} 1 \\ 3 \end{bmatrix} \right\}^{-1} 14 \\ &= \frac{\begin{bmatrix} 1 \\ 3 \end{bmatrix}}{10} (14) = \begin{bmatrix} 1.4 \\ 4.2 \end{bmatrix} \end{aligned}$$

At least two solutions

$$\begin{aligned} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} &= \begin{bmatrix} 2 \\ 4 \end{bmatrix} \text{ satisfies the equation, but } \|\mathbf{x}\|_2 = \sqrt{20} \\ \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} &= \begin{bmatrix} 1.4 \\ 4.2 \end{bmatrix} \text{ and } \|\mathbf{x}\|_2 = \sqrt{19.6} \end{aligned}$$

Minimum - norm solution

Necessary Conditions Use Left Pseudoinverse for $r < m$

Optimality conditions

$$\lambda(t_f) = \left\{ \frac{\partial \phi[\mathbf{x}(t_f)]}{\partial \mathbf{x}} \right\}^T$$

$$\dot{\lambda} = - \left[L_x^T + \mathbf{F}^T \lambda + \left(\frac{\partial \mathbf{c}}{\partial \mathbf{x}} \right)^T \mu \right]$$

$$\left[\left(\frac{\partial L}{\partial \mathbf{u}} \right)^T + \mathbf{G}^T \lambda + \left(\frac{\partial \mathbf{c}}{\partial \mathbf{u}} \right)^T \mu \right] = 0$$

with

$$\mathbf{c}[\mathbf{x}(t), \mathbf{u}(t)] \equiv 0$$

Third Approach: Penalty Function Provides “Soft” State-Control Equality Constraint: $r < m$

$$L \triangleq L_{original} + \varepsilon \mathbf{c}^T \mathbf{c}$$

ε : Scalar penalty weight

Euler-Lagrange equations are adjusted accordingly

$$\lambda(t_f) = \left\{ \frac{\partial \phi[\mathbf{x}(t_f)]}{\partial \mathbf{x}} \right\}^T$$

$$\begin{aligned} \dot{\lambda} &= - \left[L_x^T + \mathbf{F}^T \lambda \right] \\ &= - \left[\left(\frac{\partial L_{orig}}{\partial \mathbf{x}} + 2\varepsilon \mathbf{c}^T \frac{\partial \mathbf{c}}{\partial \mathbf{x}} \right)^T + \mathbf{F}^T \lambda \right] \end{aligned}$$

$$\left[\left(\frac{\partial L_{orig}}{\partial \mathbf{u}} + 2\varepsilon \mathbf{c}^T \frac{\partial \mathbf{c}}{\partial \mathbf{u}} \right)^T + \mathbf{G}^T \lambda \right] = 0$$

$$\mathbf{c}[\mathbf{x}(t), \mathbf{u}(t)] \equiv 0$$

Equality Constraint on State Alone

$$\mathbf{c}[\mathbf{x}(t)] \equiv 0 \quad \text{in } (t_o, t_f)$$

$$J_1 = \psi[\mathbf{x}(t_f)] + \int_{t_o}^{t_f} \left\{ L + \boldsymbol{\lambda}_1^T(t) [\mathbf{f} - \dot{\mathbf{x}}(t)] + \boldsymbol{\mu}^T \mathbf{c} \right\} dt$$

Hamiltonian

$$H \triangleq L + \boldsymbol{\lambda}_1^T \mathbf{f} + \boldsymbol{\mu}^T \mathbf{c}$$

Constraint is insensitive to control perturbations to first order

$$\Delta \mathbf{c} = \left(\frac{\partial \mathbf{c}}{\partial \mathbf{x}} \right) \Delta \mathbf{x} + \cancel{\left(\frac{\partial \mathbf{c}}{\partial \mathbf{u}} \right) \Delta \mathbf{u}} = \left(\frac{\partial \mathbf{c}}{\partial \mathbf{x}} \right) \Delta \mathbf{x}$$

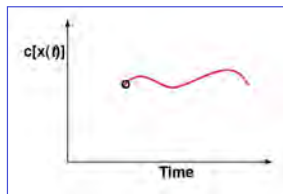
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Example of Equality Constraint on State Alone

**MINIMIZE FUEL AND CONTROL USE WHILE MAINTAINING
CONSTANT ALTITUDE**

$$\min_{\mathbf{u}(t)} J = \int_{t_o}^{t_f} \left(q \dot{m}^2 + r_1 u_1^2 + r_2 u_2^2 \right) dt$$

$$c[\mathbf{x}(t), \mathbf{u}(t)] = c[\mathbf{x}(t)] = 0 = h(t) - h_{desired}$$



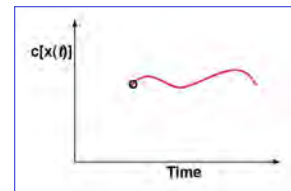
Introduce Time-Derivative of Equality Constraint

Equality constraint has no effect on optimality condition

$$\begin{aligned} \left\{ \frac{\partial H}{\partial \mathbf{u}} \right\}^T &= - \left[\left(\frac{\partial L}{\partial \mathbf{u}} \right)^T + \mathbf{G}^T \boldsymbol{\lambda} + \left(\frac{\partial \mathbf{c}}{\partial \mathbf{u}} \right)^T \boldsymbol{\mu} \right] \\ &= - \left[\left(\frac{\partial L}{\partial \mathbf{u}} \right)^T + \mathbf{G}^T \boldsymbol{\lambda} \right] \end{aligned}$$

Solution: Incorporate time-derivative of $\mathbf{c}[\mathbf{x}(t)]$ in optimization

Introduce Time-Derivative of Equality Constraint



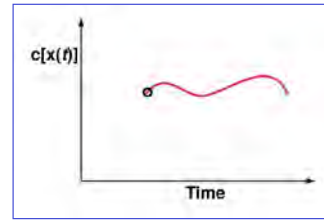
Define $\mathbf{c}[\mathbf{x}(t)]$ as the zeroth-order equality constraint

$$\mathbf{c}[\mathbf{x}(t)] \triangleq \mathbf{c}^{(0)}[\mathbf{x}(t)] \equiv 0$$

Compute first-order equality constraint

$$\begin{aligned} \frac{d\mathbf{c}^{(0)}[\mathbf{x}(t)]}{dt} &= \frac{\partial \mathbf{c}^{(0)}[\mathbf{x}(t)]}{\partial t} + \frac{\partial \mathbf{c}^{(0)}[\mathbf{x}(t)]}{\partial \mathbf{x}} \mathbf{f}[\mathbf{x}(t), \mathbf{u}(t)] \\ &\triangleq \mathbf{c}^{(1)}[\mathbf{x}(t), \mathbf{u}(t)] = 0 \end{aligned}$$

Time-Derivative of Equality Constraint



Optimality condition now includes derivative of equality constraint

$$\left\{ \frac{\partial H}{\partial \mathbf{u}} \right\}^T = - \left[\left(\frac{\partial L}{\partial \mathbf{u}} \right)^T + \mathbf{G}^T \boldsymbol{\lambda} + \left(\frac{\partial \mathbf{c}^{(1)}}{\partial \mathbf{u}} \right)^T \boldsymbol{\mu} \right]$$

Subject to

$$\mathbf{c}^{(0)}[\mathbf{x}(t_o)] \equiv 0 \quad \underline{\text{or}} \quad \mathbf{c}^{(0)}[\mathbf{x}(t_f)] \equiv 0$$

- With equality constraint satisfied at beginning or end of trajectory, $\mathbf{c}^{(1)} = \mathbf{0}$ assures that constraint is satisfied throughout
- If $\partial \mathbf{c}^{(1)} / \partial \mathbf{u} = \mathbf{0}$, differentiate again, and again, ...

State Equality Constraint Example

$$c[\mathbf{x}(t)] \triangleq c^{(0)}[\mathbf{x}(t)] = 0 = h(t) - h_{desired}$$

No control in the constraint; differentiate

$$\begin{aligned} \frac{d\mathbf{c}^{(0)}[\mathbf{x}(t)]}{dt} &= \frac{\partial \mathbf{c}^{(0)}[\mathbf{x}(t)]}{\partial t} + \frac{\partial \mathbf{c}^{(0)}[\mathbf{x}(t)]}{\partial \mathbf{x}} \mathbf{f}[\mathbf{x}(t), \mathbf{u}(t)] \\ &\triangleq \mathbf{c}^{(1)}[\mathbf{x}(t)] = 0 = \dot{h}(t) = V(t) \sin \gamma(t) \end{aligned}$$

Still no control in the constraint; differentiate again...

State Equality Constraint Example

Still no control in the constraint; differentiate again

$$\begin{aligned}
 \frac{d\mathbf{c}^{(1)}[\mathbf{x}(t)]}{dt} &= \frac{\partial \mathbf{c}^{(1)}[\mathbf{x}(t)]}{\partial t} + \frac{\partial \mathbf{c}^{(1)}[\mathbf{x}(t)]}{\partial \mathbf{x}} \mathbf{f}[\mathbf{x}(t), \mathbf{u}(t)] \\
 &\triangleq \mathbf{c}^{(2)}[\mathbf{x}(t)] = 0 = \ddot{h}(t) = \frac{dV}{dt}(t) \sin \gamma(t) + V(t) \frac{d[\sin \gamma(t)]}{dt} \\
 &= \left[\left(T_{\max_{SL}} (e^{-\beta h}) \delta T - C_D \frac{1}{2} \rho V^2 S \right) / m - g \sin \gamma \right] \sin \gamma(t) \\
 &\quad + \cos \gamma(t) \left[\left(C_{L_\alpha} \alpha \frac{1}{2} \rho V^2 S \right) / m - g \cos \gamma \right]
 \end{aligned}$$

State Equality Constraint Example

- Control appears in the 2nd-order equality constraint

$$\begin{aligned}
 \mathbf{c}^{(2)}[\mathbf{x}(t), \mathbf{u}(t)] &= 0 \\
 &= \left[\left(T_{\max_{SL}} (e^{-\beta h}) \delta T - C_D (\alpha) \frac{1}{2} \rho(h) V^2(t) S \right) / m(t) - g \sin \gamma(t) \right] \sin \gamma(t) \\
 &\quad + \cos \gamma(t) \left[\left(C_{L_\alpha} \alpha \frac{1}{2} \rho(h) V^2(t) S \right) / m(t) - g \cos \gamma(t) \right]
 \end{aligned}$$

$$H \triangleq L + \lambda_1^T \mathbf{f} + \mu^T \mathbf{c}^{(2)}$$

- 0th- and 1st-order constraints satisfied at some point on the trajectory (e.g., t_0)

$$\begin{aligned}
 c^{(0)}[\mathbf{x}(t_0)] &= 0 \Rightarrow h(t_0) = h_{desired} \\
 c^{(1)}[\mathbf{x}(t_0)] &= 0 \Rightarrow \gamma(t_0) = 0
 \end{aligned}$$

Minimization with Inequality Constraints

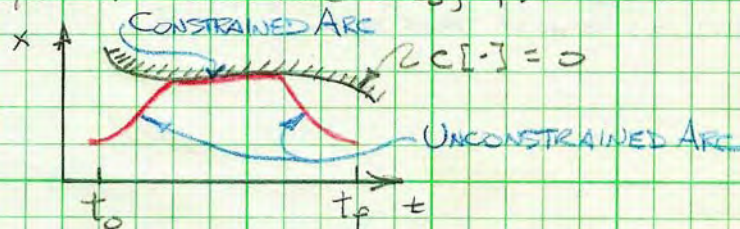
“Hard” Inequality Constraints

INEQUALITY CONSTRAINTS ON STATE AND CONTROL in (t_0, t_f)

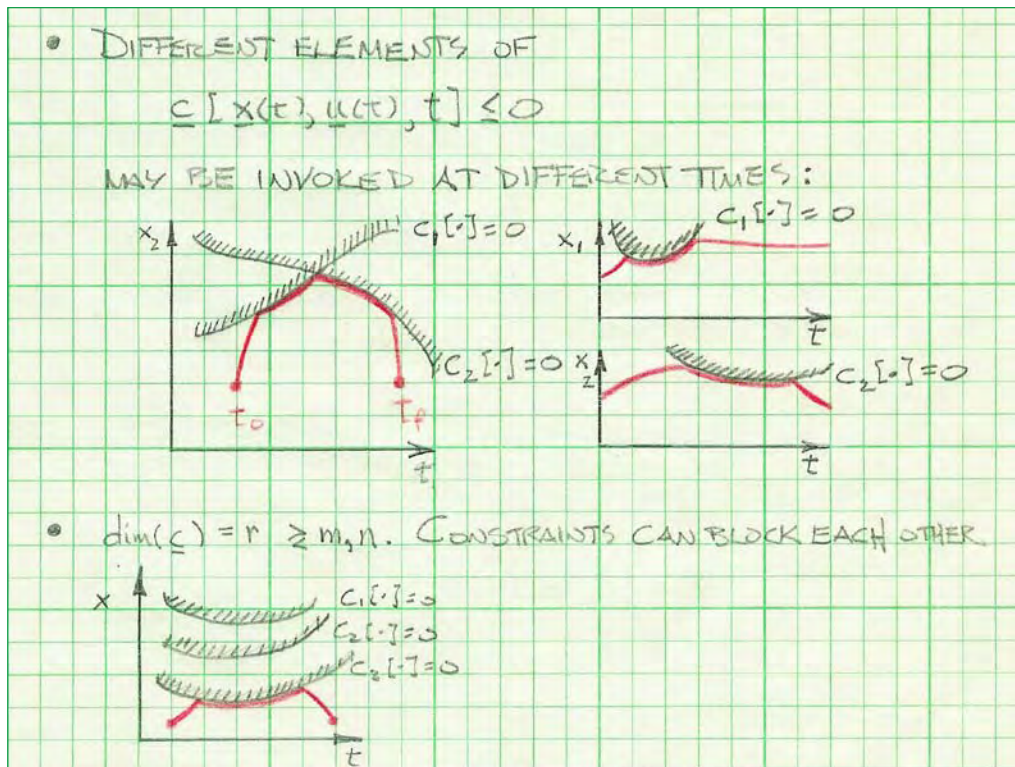
- PRINCIPAL DIFFERENCE FROM EQUALITY CONSTRAINT:

INEQUALITY CONSTRAINT NORMALLY NOT INVOKED

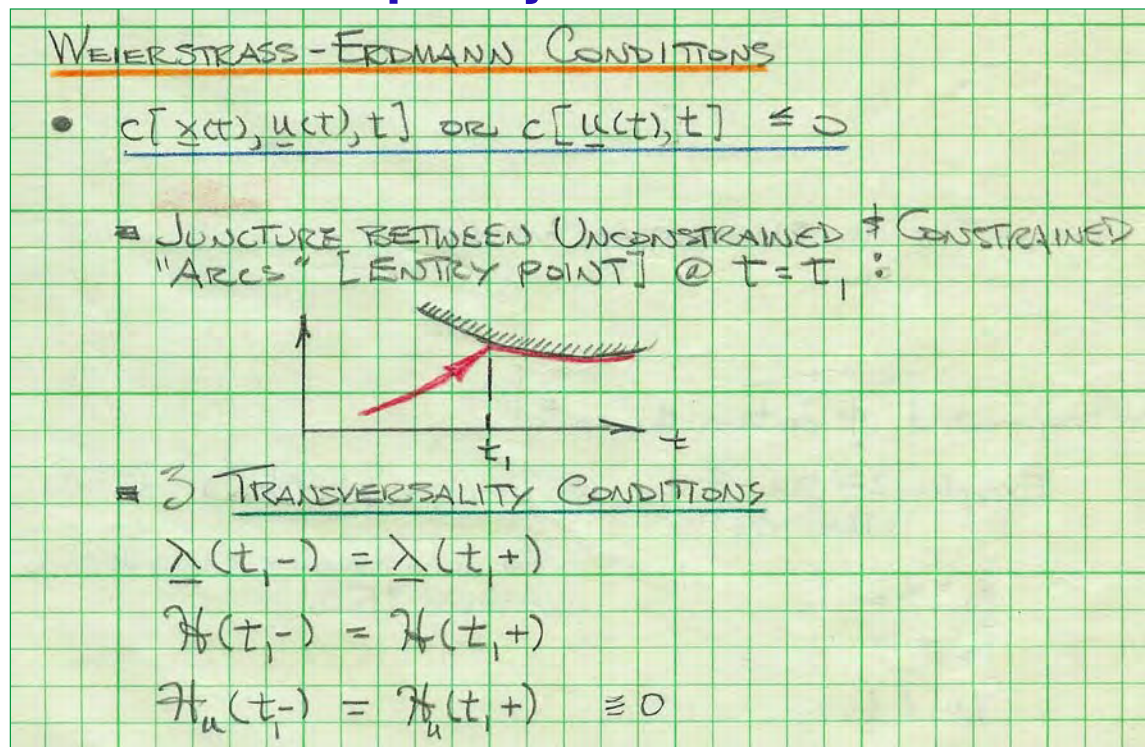
THROUGHOUT INTERVAL (t_0, t_f) :



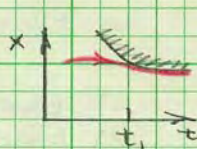
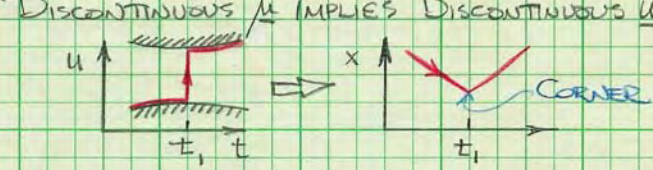
Inequality Constraints



Inequality Constraints



Inequality Constraints

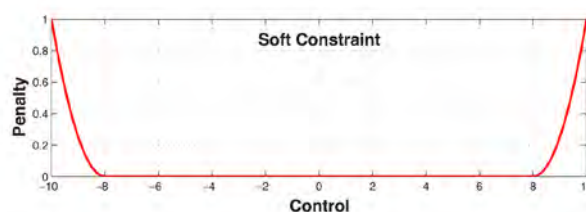
- CORNERS
 CONTINUOUS u IMPLIES CONTINUOUS \dot{x}
 $u(t,-) = u(t,+) \Rightarrow \dot{x}(t,-) = \dot{x}(t,+)$
 NOT A CORNER

- DISCONTINUOUS u IMPLIES DISCONTINUOUS \dot{x}

- SIMPLE DISCONTINUITY
 $\dot{x}(t,-) = \dot{x}(t,+)$
- HIGHER-ORDER DISCONTINUITY
 IMPULSIVE CONTROL
 STATE DISCONTINUITY

“Soft” Control Inequality Constraint

$$L \triangleq L_{\text{original}} + \epsilon \mathbf{c}^T \mathbf{c} \quad \epsilon : \text{Scalar penalty weight}$$

Scalar Example

$$c[u(t)] = \begin{cases} (u - u_{\max})^2 & , \quad u \geq u_{\max} \\ 0 & , \quad u_{\min} < u < u_{\max} \\ (u - u_{\min})^2 & , \quad u \leq u_{\min} \end{cases}$$



Numerical Optimization

Numerical Optimization Methods

	Optimality of Solution	Solution Method			Iteration Variables	Order of ODE ^a Solution
		$\mathbf{x}(t)$	$\boldsymbol{\lambda}(t)$	$\mathbf{u}(t)$		
Parametric	approximate	ODE ^a	–	I^b	$\mathbf{u}(\mathbf{k}_u, t)$	n
Penalty function	approximate	I	–	I	$\mathbf{x}(\mathbf{k}_x, t), \mathbf{u}(\mathbf{k}_u, t)$	none
Dynamic programming	exact	ODE	PDE ^c	I	$\mathbf{u}(t)$	n
Neighboring extremal	exact	ODE	ODE	$\mathcal{H}_{\mathbf{u}} = \mathbf{0}$	$\boldsymbol{\lambda}(t_0)$	$2n$
Quasilinearization	exact	I	I	$\mathcal{H}_{\mathbf{u}} = \mathbf{0}$	$\mathbf{x}(t), \boldsymbol{\lambda}(t)$	$2n^d$
Gradient	exact	ODE	ODE	I	$\mathbf{u}(t)$	$2n$

^aODE: ordinary differential equation.

^bIteration.

^cPDE: Partial differential equation; HJB equation; one dependent variable (V), $(n+1)$ independent variables (\mathbf{x}, t) , $\partial V / \partial \mathbf{x}$ corresponds to $\boldsymbol{\lambda}^T$.

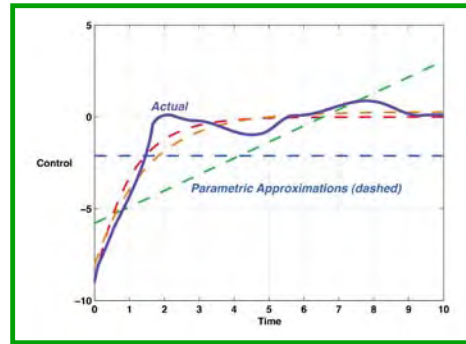
^dPerturbation equation for $\Delta \mathbf{x}(t)$ and $\Delta \boldsymbol{\lambda}(t)$.

Parametric Optimization

$$\min_{\mathbf{u}(t)} J = \phi[\mathbf{x}(t_f)] + \int_{t_o}^{t_f} L[\mathbf{x}(t), \mathbf{u}(t)] dt$$

subject to

$$\dot{\mathbf{x}}(t) = \mathbf{f}[\mathbf{x}(t), \mathbf{u}(t)], \quad \mathbf{x}(t_o) \text{ given}$$



Control specified by a parameter vector, \mathbf{k}
No adjoint equations

Examples

$$u(t) = k$$

$$u(t) = k_0 + k_1 t + k_2 t^2 + \dots, \quad \mathbf{k} = \begin{bmatrix} k_0 & k_1 & \dots & k_m \end{bmatrix}^T$$

$$\mathbf{u}(t) = \mathbf{k}$$

$$\mathbf{u}(t) = \mathbf{k}_0 + \mathbf{k}_1 t + \mathbf{k}_2 t^2 + \dots, \quad \mathbf{k} = \begin{bmatrix} \mathbf{k}_0 & \mathbf{k}_1 & \dots & \mathbf{k}_m \end{bmatrix}$$

$$\mathbf{u}(t) = \mathbf{k}_0 + \mathbf{k}_1 \sin\left(\frac{\pi t}{t_f - t_o}\right) + \mathbf{k}_2 \cos\left(\frac{\pi t}{t_f - t_o}\right), \quad \mathbf{k} = \begin{bmatrix} \mathbf{k}_0 & \mathbf{k}_1 & \mathbf{k}_2 \end{bmatrix}$$

Parametric Optimization

$$\min_{\mathbf{u}(t)} J = \phi[\mathbf{x}(t_f)] + \int_{t_o}^{t_f} L[\mathbf{x}(t), \mathbf{u}(t)] dt$$

subject to

$$\dot{\mathbf{x}}(t) = \mathbf{f}[\mathbf{x}(t), \mathbf{u}(t)], \quad \mathbf{x}(t_o) \text{ given}$$

- **Necessary and sufficient conditions for a minimum**
- **Use static search algorithm to find minimizing control parameter, \mathbf{k}**

$$\frac{\partial J}{\partial \mathbf{k}} = 0$$

$$\frac{\partial^2 J}{\partial \mathbf{k}^2} > 0$$

Parametric Optimization Example

$$\min_{\mathbf{u}(t)} J = \phi[\mathbf{x}(t_f)] + \int_{t_o}^{t_f} L[\mathbf{x}(t), \mathbf{u}(t)] dt$$

subject to

$$\dot{\mathbf{x}}(t) = \mathbf{f}[\mathbf{x}(t), \mathbf{u}(t)], \quad \mathbf{x}(t_o) \text{ given}$$

$$u(t) \triangleq (k_0 + k_1 t + k_2 t^2)$$

$$\dot{V} = \left(T_{\max} - C_D \frac{1}{2} \rho V^2 S \right) / m - g \sin \gamma$$

$$\dot{\gamma} = \frac{1}{V} \left\{ \left[C_L (k_0 + k_1 t + k_2 t^2) \frac{1}{2} \rho V^2 S \right] / m - g \cos \gamma \right\}$$

$$\dot{h} = V \sin \gamma$$

$$\dot{r} = V \cos \gamma$$

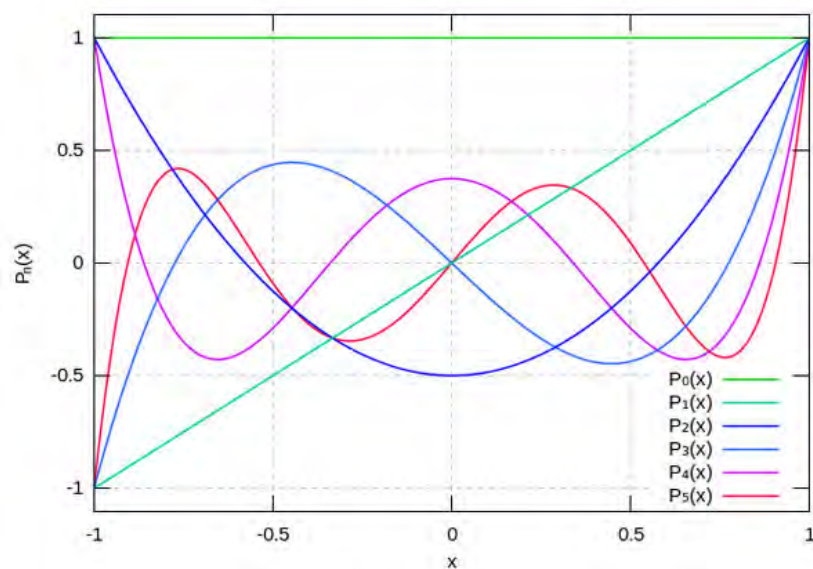
$$\dot{m} = -(SFC)(T)$$

$$\mathbf{a} = \begin{bmatrix} k_0 \\ k_1 \\ k_2 \end{bmatrix}$$

$$\frac{\partial J}{\partial \mathbf{k}} = 0$$

$$\frac{\partial^2 J}{\partial \mathbf{k}^2} > 0$$

Legendre Polynomials



Solutions to Legendre's differential equation

Legendre Polynomials

Polynomials can be generated by Rodriques's formula

$$P_n(x) = \frac{1}{2^n n!} \frac{d^n}{dx^n} \left[(x^2 - 1)^n \right]$$

**Optimizing Control:
Find minimizing values of k_n**

$$u(x) = k_0 P_0(x) + k_1 P_1(x) + k_2 P_2(x) + k_3 P_3(x) + k_4 P_4(x) + k_5 P_5(x) + \dots$$

$$x \triangleq \frac{t}{t_f - t_o}$$

$$P_0(x) = 1$$

$$P_1(x) = x$$

$$P_2(x) = \frac{1}{2}(3x^2 - 1)$$

$$P_3(x) = \frac{1}{2}(5x^3 - 3x)$$

$$P_4(x) = \frac{1}{8}(35x^4 - 30x^2 + 3)$$

$$P_5(x) = \frac{1}{8}(63x^5 - 70x^3 + 15x)$$

Control History Optimized with Legendre Polynomials Could be expressed as a Simple Power Series

$$u^*(x) = k_0^* P_0(x) + k_1^* P_1(x) + k_2^* P_2(x) + k_3^* P_3(x) + k_4^* P_4(x) + k_5^* P_5(x) + \dots$$

$$u^*(x) = a_0^* + a_1^* x + a_2^* x^2 + a_3^* x^3 + a_4^* x^4 + a_5^* x^5 + \dots$$

$$a_0^* = k_0^* - k_2^* \left(\frac{1}{2} \right) + k_4^* \left(\frac{3}{8} \right) + \dots$$

$$a_1^* = k_1^* - k_3^* \left(\frac{3}{2} \right) + k_5^* \left(\frac{15}{8} \right) + \dots$$

$$a_2^* = k_2^* \left(\frac{3}{2} \right) - k_4^* \left(\frac{30}{8} \right) + \dots$$

...

$$P_0(x) = 1$$

$$P_1(x) = x$$

$$P_2(x) = \frac{1}{2}(3x^2 - 1)$$

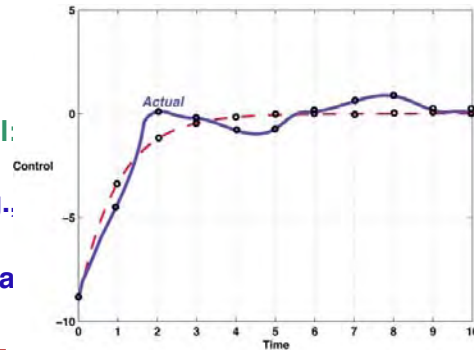
$$P_3(x) = \frac{1}{2}(5x^3 - 3x)$$

$$P_4(x) = \frac{1}{8}(35x^4 - 30x^2 + 3)$$

$$P_5(x) = \frac{1}{8}(63x^5 - 70x^3 + 15x)$$

Parametric Optimization: Collocation

- Admissible controls occur at discrete times, k
- Cost and dynamic constraint are discretized
- “Pseudospectral” Optimal Control: State and adjoint points may be connected by basis functions, e.g., Legendre polynomials
- Continuous solution approached as time interval decreased



$$\min_{\mathbf{u}_k} J = \phi[\mathbf{x}_{k_f}] + \sum_{k=0}^{k_f-1} L[\mathbf{x}_k, \mathbf{u}_k]$$

subject to

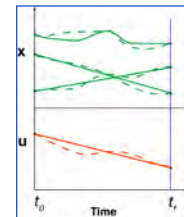
$$\mathbf{x}_{k+1} = \mathbf{f}_k[\mathbf{x}_k, \mathbf{u}_k], \quad \mathbf{x}_0 \text{ given}$$

http://en.wikipedia.org/wiki/Collocation_method

http://en.wikipedia.org/wiki/Legendre_polynomials

http://en.wikipedia.org/wiki/Pseudospectral_optimal_control

Penalty Function Method Balakrishnan's “Epsilon” Technique



- No integration of the dynamic equation
- Parametric optimization of the state and control history

$$\begin{aligned} \mathbf{x}(t) &\equiv \mathbf{x}(\mathbf{k}_x, t) \\ \mathbf{u}(t) &\equiv \mathbf{u}(\mathbf{k}_u, t) \end{aligned}$$

$$\begin{aligned} \dim(\mathbf{k}_x) &\geq n \\ \dim(\mathbf{k}_u) &\geq m \end{aligned}$$

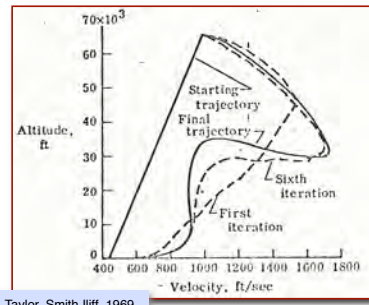
- Augment the integral cost function by the dynamic equation error

$$\min_{\mathbf{u}(t), \mathbf{x}(t)} J = \phi[\mathbf{x}(t_f), t_f] + \int_{t_0}^{t_f} \left\{ L[\mathbf{x}(t), \mathbf{u}(t), t] + \left(\frac{1}{\epsilon} \right) \left(\mathbf{f}[\mathbf{x}(t), \mathbf{u}(t), t] - \dot{\mathbf{x}}(t) \right)^T \{ \bullet \} \right\} dt$$

$1/\epsilon$ is the penalty for not satisfying the dynamic constraint

Penalty Function Method

- Choose reasonable starting values of state and control parameters
 - e.g., state and control satisfy boundary conditions
- Evaluate cost function



Taylor, Smith liff, 1969

$$J_0 = \varphi[\mathbf{x}_0(t_f)] + \int_{t_0}^{t_f} \left\{ L[\mathbf{x}_0(t), \mathbf{u}(t)] + \left(\frac{1}{\varepsilon} \right) \left(\{ \mathbf{f}[\mathbf{x}_0(t), \mathbf{u}_0(t)] - \dot{\mathbf{x}}_0(t) \}^T \{ \bullet \} \right) \right\} dt$$

Update state and control parameters (e.g., steepest descent)

$$\mathbf{k}_{\mathbf{x}_{i+1}} = \mathbf{k}_{\mathbf{x}_i} - \alpha \left[\frac{\partial J}{\partial \mathbf{k}_{\mathbf{x}}} \Big|_{\mathbf{k}_{\mathbf{x}} = \mathbf{k}_{\mathbf{x}_i}} \right]^T$$

$$\mathbf{k}_{\mathbf{u}_{i+1}} = \mathbf{k}_{\mathbf{u}_i} - \alpha \left[\frac{\partial J}{\partial \mathbf{k}_{\mathbf{u}}} \Big|_{\mathbf{k}_{\mathbf{u}} = \mathbf{k}_{\mathbf{u}_i}} \right]^T$$

$$\begin{aligned} \mathbf{x}_{i+1}(t) &\equiv \mathbf{x}(\mathbf{k}_{\mathbf{x}_{i+1}}, t) \\ \mathbf{u}_{i+1}(t) &\equiv \mathbf{u}(\mathbf{k}_{\mathbf{u}_{i+1}}, t) \end{aligned}$$

Re-evaluate cost with higher penalty

Repeat to convergence

$$J_i \rightarrow J_{i+1} \rightarrow J^*, \quad \varepsilon \rightarrow 0, \quad \mathbf{f}[\mathbf{x}(t), \mathbf{u}(t), t] \rightarrow \dot{\mathbf{x}}(t)$$

Neighboring Extremal Method

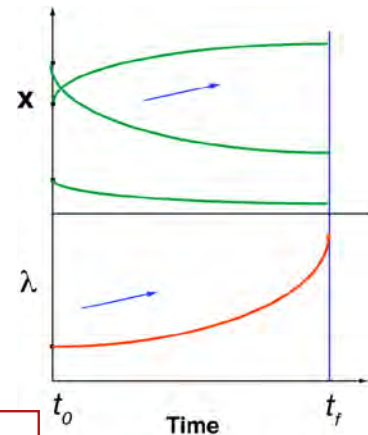
“Shooting Method”: Integrate both state and adjoint vector forward in time

$$\begin{aligned} \dot{\mathbf{x}}_{k+1}(t) &= \mathbf{f}[\mathbf{x}_{k+1}(t), \mathbf{u}_k(t)], \\ \mathbf{x}_0(t_0) &\text{ given, initial guess for } \mathbf{u}_0(t) \end{aligned}$$

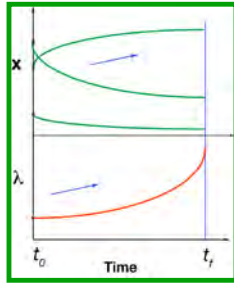
$$\dot{\lambda}_{k+1}(t) = - \left[\frac{\partial L}{\partial \mathbf{x}} \Big|_k(t) + \lambda_{k+1}^T(t) \mathbf{F}_k(t) \right]^T, \quad \lambda_k(t_0) \text{ given}$$

with

$$\mathbf{u}_{k+1}(t) \text{ defined by } \frac{\partial H[\mathbf{x}_k(t), \mathbf{u}_k(t), \lambda_{k+1}(t), t]}{\partial \mathbf{u}} = \left[L_{\mathbf{u}_k}(t) + \lambda_{k+1}^T(t) \mathbf{G}_k(t) \right] = \mathbf{0}$$



... but how do you know the initial value of the adjoint vector?



Neighboring Extremal Method

All trajectories are optimal (i.e., “extremals”) for some cost function because

$$\frac{\partial H}{\partial \mathbf{u}} = H_{\mathbf{u}} = [L_{\mathbf{u}} + \boldsymbol{\lambda}^T \mathbf{G}] = \mathbf{0}$$

Integrating state equation computes a value for $\phi[\mathbf{x}(t_f)]$

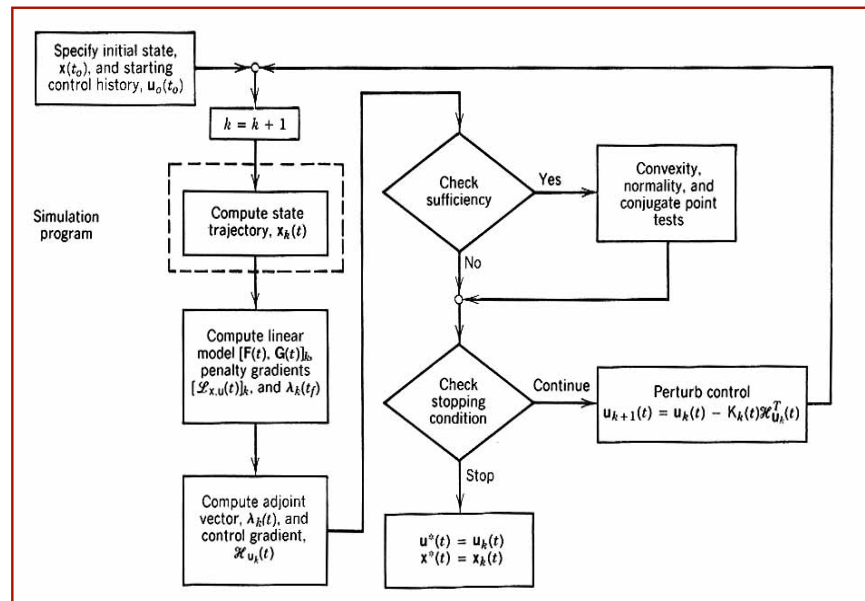
$$\mathbf{x}(t_f) = \mathbf{x}(t_0) + \int_{t_0}^{t_f} \mathbf{f}[\mathbf{x}_{k+1}(t), \mathbf{u}_k(t)]; \quad \phi[\mathbf{x}(t_f)] \rightarrow \frac{\partial \phi[\mathbf{x}(t_f)]}{\partial \mathbf{x}} = \boldsymbol{\lambda}^T(t_f)$$

Use a **learning rule** to estimate the initial value of the adjoint vector, e.g.,

$$\boldsymbol{\lambda}_{k+1}^T(t_0) = \boldsymbol{\lambda}_k^T(t_0) - \alpha [\boldsymbol{\lambda}_k^T(t_f) - \boldsymbol{\lambda}_{desired}^T]^T$$

Gradient-Based Methods

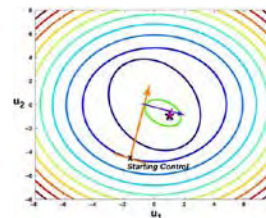
Gradient-Based Search Algorithms



Gradient-Based Search Algorithms

Steepest Descent

$$\mathbf{u}_{k+1}(t) = \mathbf{u}_k(t) - \varepsilon_k \left[\frac{\partial H}{\partial \mathbf{u}}(t) \right]_k^T$$



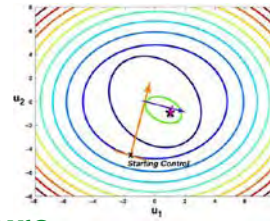
Newton Raphson

$$\mathbf{u}_{k+1}(t) = \mathbf{u}_k(t) - \left[\frac{\partial^2 H}{\partial \mathbf{u}^2}(t) \right]_k^{-1} \left[\frac{\partial H}{\partial \mathbf{u}}(t) \right]_k^T$$

Generalized Direct Search

$$\mathbf{u}_{k+1}(t) = \mathbf{u}_k(t) - \mathbf{K}_k \left[\frac{\partial H}{\partial \mathbf{u}}(t) \right]_k^T$$

Numerical Optimization Using Steepest-Descent Algorithm



Iterative bidirectional procedure

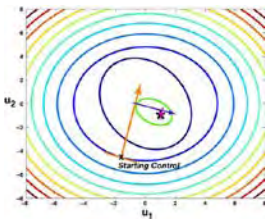
Forward solution to find the state, $\mathbf{x}(t)$

Backward solution to find the adjoint vector, $\boldsymbol{\lambda}(t)$

Steepest-descent adjustment of control history, $\mathbf{u}(t)$

$$\dot{\mathbf{x}}_k(t) = \mathbf{f}[\mathbf{x}_k(t), \mathbf{u}_{k-1}(t)], \quad \mathbf{x}(t_o) \text{ given}$$

Use educated guess for $\mathbf{u}_0(t)$ on first iteration

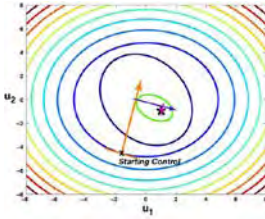


Numerical Optimization Using Steepest-Descent Algorithm

$$\dot{\boldsymbol{\lambda}}_k(t) = - \left[\frac{\partial H}{\partial \mathbf{x}} \right]_k^T = - \left[L_{\mathbf{x}}(t) + \boldsymbol{\lambda}^T(t) \mathbf{F}(t) \right]_k^T,$$

$$\boldsymbol{\lambda}(t_f) = \left\{ \frac{\partial \phi[\mathbf{x}(t_f)]}{\partial \mathbf{x}} \right\}^T \quad [E - L \text{ \#2 and \#1}]$$

Use $\mathbf{x}_{k-1}(t)$ and $\mathbf{u}_{k-1}(t)$ from previous step



Numerical Optimization Using Steepest-Descent Algorithm

$$\left(\frac{\partial H}{\partial \mathbf{u}} \right)_k = \left[L_{\mathbf{u}}(t) + \boldsymbol{\lambda}^T(t) \mathbf{G}(t) \right]_k \quad [E - L \#3]$$

$$\begin{aligned} \mathbf{u}_{k+1}(t) &= \mathbf{u}_k(t) - \varepsilon \left[\frac{\partial H}{\partial \mathbf{u}} \Big|_{\mathbf{u}(t)=\mathbf{u}_k(t)} \right]^T \\ &= \mathbf{u}_k(t) - \varepsilon \left[L_{\mathbf{u}} + \boldsymbol{\lambda}^T(t) \mathbf{G}(t) \right]_k^T \end{aligned}$$

Use $\mathbf{x}(t)$, $\boldsymbol{\lambda}(t)$, and $\mathbf{u}(t)$ from previous step

Finding the Best Steepest-Descent Gain

$J_{0_k} [u_k(t), 0 < t < t_f]$: Best solution from the previous iteration

Calculate the gradient, $\frac{\partial H_k}{\partial u}(t)$, in $0 < t < t_f$

$J_{1_k} \left[u_k(t) - \varepsilon_k \frac{\partial H_k}{\partial u}(t), 0 < t < t_f \right]$: Steepest - descent calculation of cost (1)

$J_{2_k} \left[u_k(t) - 2\varepsilon_k \frac{\partial H_k}{\partial u}(t), 0 < t < t_f \right]$: Steepest - descent calculation of cost (2)

$$J(\varepsilon) = a_0 + a_1 \varepsilon + a_2 \varepsilon^2$$

$$\begin{bmatrix} J_{0_k} \\ J_{1_k} \\ J_{2_k} \end{bmatrix} = \begin{bmatrix} a_0 + a_1(0) + a_2(0)^2 \\ a_0 + a_1(\varepsilon_k) + a_2(\varepsilon_k)^2 \\ a_0 + a_1(2\varepsilon_k) + a_2(2\varepsilon_k)^2 \end{bmatrix} = \begin{bmatrix} 1 & 0 & 0 \\ 1 & (\varepsilon_k) & (\varepsilon_k)^2 \\ 1 & (2\varepsilon_k) & (2\varepsilon_k)^2 \end{bmatrix} \begin{bmatrix} a_0 \\ a_1 \\ a_2 \end{bmatrix}$$

Solve for a_0 , a_1 , and a_2

Find ε^* that minimizes $J(\varepsilon)$

$J_{k+1} \left[u_{k+1}(t) = u_k(t) - \varepsilon^* \frac{\partial H_k}{\partial u}(t), 0 < t < t_f \right]$: Best steepest - descent calculation of cost

Go to next iteration

Steepest-Descent Algorithm for Problem with Terminal Constraint

$$\min_{\mathbf{u}(t)} J = \phi[\mathbf{x}(t_f)] + \int_{t_o}^{t_f} L[\mathbf{x}(t), \mathbf{u}(t)] dt$$

$$\psi[\mathbf{x}(t_f)] \equiv 0 \text{ (scalar)}$$

$$\frac{\partial H_c}{\partial \mathbf{u}} = \left[\frac{\partial H_0}{\partial \mathbf{u}} - \left(\frac{a}{b} \right) \frac{\partial H_1}{\partial \mathbf{u}} \right] = 0$$

see Lecture 3 for *a* and *b* definitions

Chose $\mathbf{u}_{k+1}(t)$ such that

$$\begin{aligned} \mathbf{u}_{k+1}(t) &= \mathbf{u}_k(t) - \varepsilon \left[\frac{\partial H_c}{\partial \mathbf{u}} \Big|_{\mathbf{u}(t)=\mathbf{u}_k(t)} \right]^T \\ &= \mathbf{u}_k(t) - \varepsilon \left[L_{\mathbf{u}}^T + \mathbf{G}^T(t) \lambda_0(t) \right]_k - \frac{1}{b_k} \mathbf{G}_k^T(t) \lambda_1(t) \psi_k[\mathbf{x}(t_f)] \end{aligned}$$

Zero Gradient Algorithm for Quadratic Control Cost

$$\min_{\mathbf{u}(t)} J = \phi[\mathbf{x}(t_f)] + \int_{t_o}^{t_f} \left\{ L[\mathbf{x}(t)] + \frac{1}{2} \mathbf{u}^T(t) \mathbf{R} \mathbf{u}(t) \right\} dt$$

$$H[\mathbf{x}(t), \mathbf{u}(t), \lambda(t)] = \left\{ L[\mathbf{x}(t)] + \frac{1}{2} \mathbf{u}^T(t) \mathbf{R} \mathbf{u}(t) \right\} + \lambda^T(t) \mathbf{f}[\mathbf{x}(t), \mathbf{u}(t)]$$

Optimality condition:

$$\frac{\partial H}{\partial \mathbf{u}}(t) = H_{\mathbf{u}}(t) = \left[\mathbf{u}^T(t) \mathbf{R} + \lambda^T(t) \mathbf{G}(t) \right] \equiv \mathbf{0}$$

Zero Gradient Algorithm for Quadratic Control Cost

$$\frac{\partial H}{\partial \mathbf{u}}(t) = H_{\mathbf{u}}(t) = [\mathbf{u}^T(t)\mathbf{R} + \boldsymbol{\lambda}^T(t)\mathbf{G}(t)] \equiv 0$$

Optimal control, $\mathbf{u}^*(t)$

$$\begin{aligned}\mathbf{u}^{*T}(t)\mathbf{R} &= -\boldsymbol{\lambda}^{*T}(t)\mathbf{G}^*(t) \\ \mathbf{u}^*(t) &= -\mathbf{R}^{-1}\mathbf{G}^{*T}(t)\boldsymbol{\lambda}^*(t)\end{aligned}$$

But $\mathbf{G}_k(t)$ and $\boldsymbol{\lambda}_k(t)$ are sub-optimal before convergence, and optimal control cannot be computed in single step

\therefore Chose $\mathbf{u}_{k+1}(t)$ such that

$$\begin{aligned}\mathbf{u}_{k+1}(t) &= (1 - \varepsilon)\mathbf{u}_k(t) - \varepsilon[\mathbf{R}^{-1}\mathbf{G}_k^T(t)\boldsymbol{\lambda}_k(t)] \\ \varepsilon &\triangleq \text{Relaxation parameter} < 1\end{aligned}$$

Stopping Conditions for Numerical Optimization

- Computed total cost, J , reaches a theoretical minimum, e.g., zero
- Convergence of J is essentially complete
- Control gradient, $H_{\mathbf{u}}(t)$, is essentially zero throughout $[t_o, t_f]$
- Terminal cost/constraint is satisfied, and integral cost is “good enough”

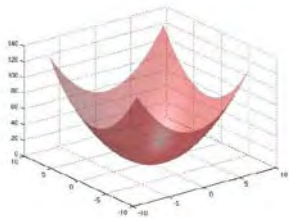
$$J_{k+1} = 0 + \varepsilon$$

$$J_{k+1} > J_k - \varepsilon$$

$$\begin{aligned} |H_{\mathbf{u}_{k+1}}(t)| &= 0 \pm \varepsilon \text{ in } [t_o, t_f] \\ \text{or} \\ \int_{t_o}^{t_f} H_{\mathbf{u}_{k+1}}^T(t) H_{\mathbf{u}_{k+1}}(t) dt &= 0 + \varepsilon \end{aligned}$$

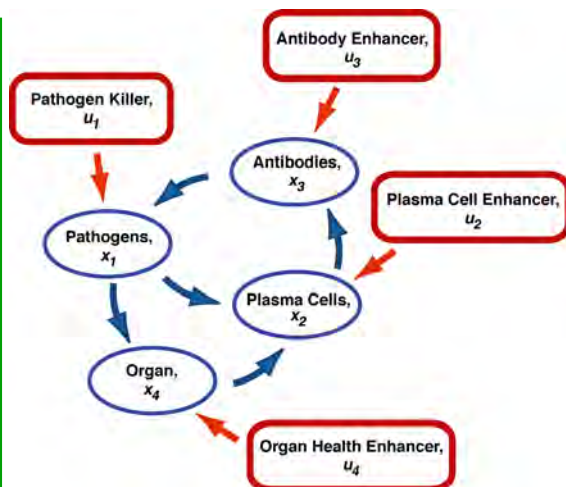
$$\varphi_{k+1}(t_f) = 0 + \varepsilon, \text{ or } \psi_{k+1}(t_f) = 0 \pm \varepsilon, \text{ and } \int_{t_o}^{t_f} L[\mathbf{x}(t), \mathbf{u}(t)] dt < \delta$$

Optimal Treatment of an Infection



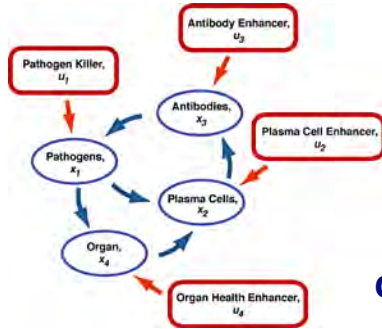
Model of Infection and Immune Response

- x_1 = Concentration of a **pathogen**, which displays antigen
- x_2 = Concentration of **plasma cells**, which are carriers and producers of antibodies
- x_3 = Concentration of **antibodies**, which recognize antigen and kill pathogen
- x_4 = Relative characteristic of a **damaged organ** [0 = healthy, 1 = dead]



Infection Dynamics

Fourth-order ordinary differential equation, including effects of therapy (control)



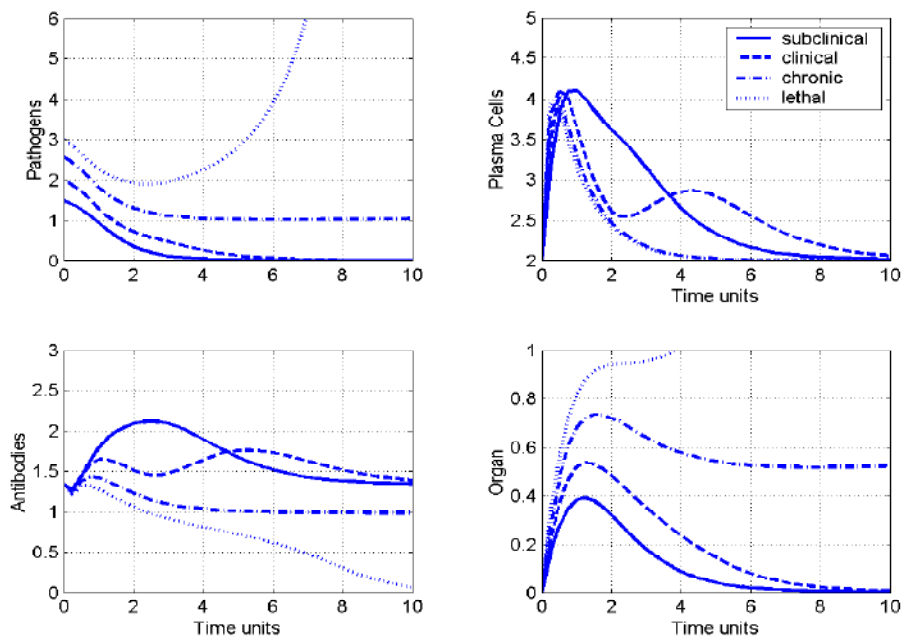
$$\dot{x}_1 = (a_{11} - a_{12}x_3)x_1 + b_1u_1 + w_1$$

$$\dot{x}_2 = a_{21}(x_4)a_{22}x_1x_3 - a_{23}(x_2 - x_2^*) + b_2u_2 + w_2$$

$$\dot{x}_3 = a_{31}x_2 - (a_{32} + a_{33}x_1)x_3 + b_3u_3 + w_3$$

$$\dot{x}_4 = a_{41}x_1 - a_{42}x_4 + b_4u_4 + w_4$$

Uncontrolled Response to Infection



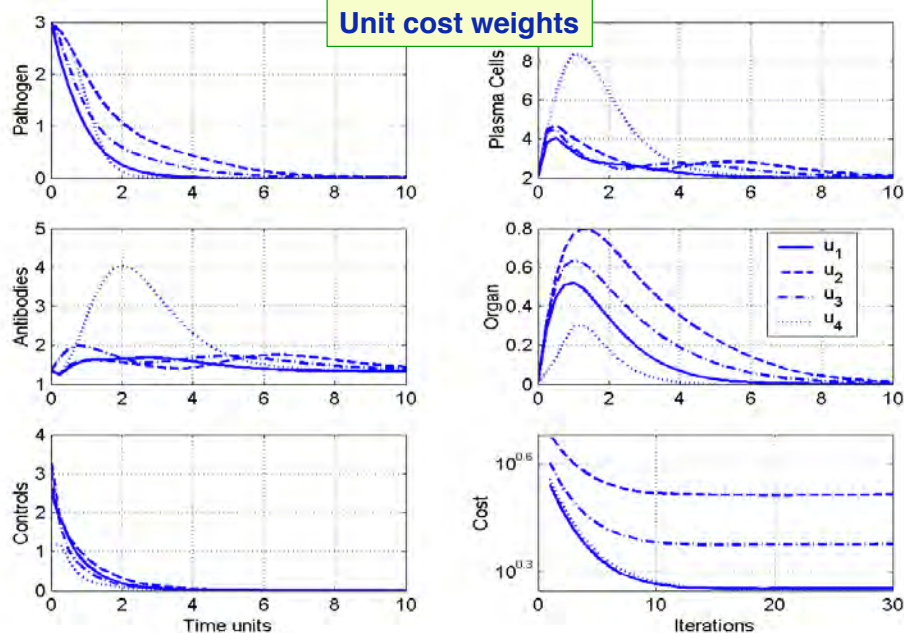
Cost Function to be Minimized by Optimal Therapy

$$J = \frac{1}{2} \left(p_{11} x_{1_f}^2 + p_{44} x_{4_f}^2 \right) + \frac{1}{2} \int_{t_o}^{t_f} \left(q_{11} x_1^2 + q_{44} x_4^2 + r u^2 \right) dt$$

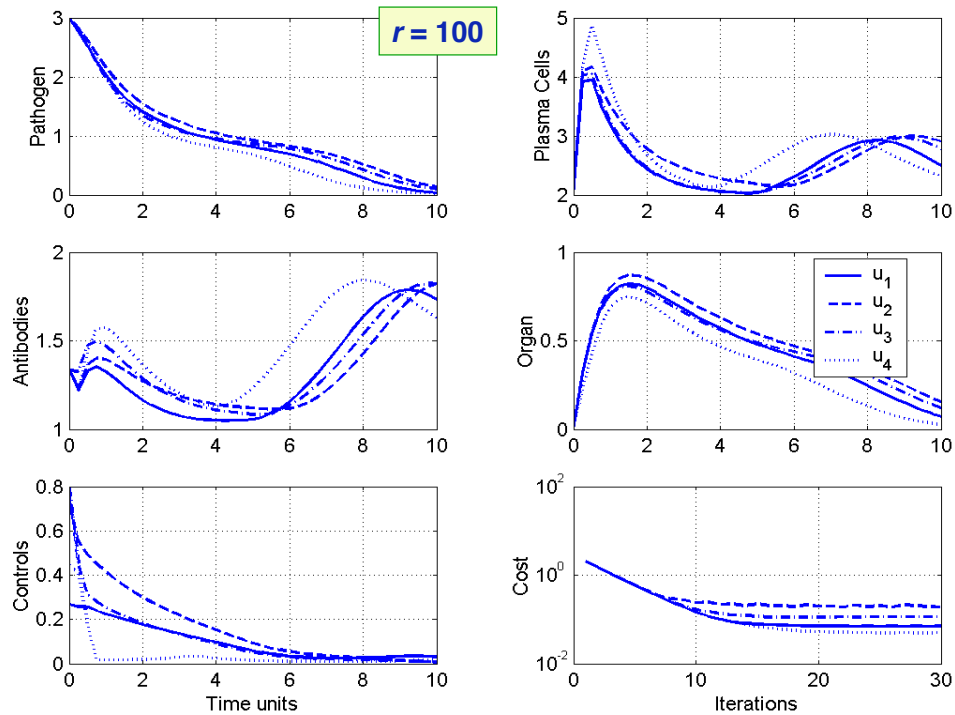
- Tradeoffs between final values, integral values over a fixed time interval, state, and control
- Cost function includes weighted square values of
 - Final concentration of the pathogen
 - Final health of the damaged organ (0 is good, 1 is bad)
 - Integral of pathogen concentration
 - Integral health of the damaged organ (0 is good, 1 is bad)
 - Integral of drug usage
- Drug cost may reflect physiological cost (side effects) or financial cost

Examples of Optimal Therapy

- u_1 = Pathogen killer
- u_2 = Plasma cell enhancer
- u_3 = Antibody enhancer
- u_4 = Organ health enhancer



Effects of Increased Drug “Cost”

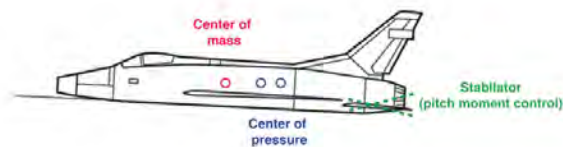


***Next Time:
Minimum-Time and -Fuel
Problems***

***Reading
OCE: Section 3.5, 3.6***

Supplemental Material

Examples of Equality Constraints



$$\mathbf{c}[\mathbf{x}(t), \mathbf{u}(t)] \equiv 0$$

Pitch Moment = 0 = fcn(Mach Number, Stabilator Trim Angle)

$$\mathbf{c}[\mathbf{u}(t)] \equiv 0$$

Stabilator Trim Angle – constant = 0

$$\mathbf{c}[\mathbf{x}(t)] \equiv 0$$

Altitude – constant = 0

Minimum-Error-Norm Solution

$$\begin{aligned}\dim(\mathbf{x}) &= r \times 1 \\ \dim(\mathbf{y}) &= m \times 1 \\ r &> m\end{aligned}$$

- Euclidean error norm for linear equation

$$\|\mathbf{Ax} - \mathbf{y}\|_2^2 = [\mathbf{Ax} - \mathbf{y}]^T [\mathbf{Ax} - \mathbf{y}]$$

- Necessary condition for minimum error

$$\frac{\partial}{\partial \mathbf{x}} \|\mathbf{Ax} - \mathbf{y}\|_2^2 = 2[\mathbf{Ax} - \mathbf{y}]^T = 0$$

- Express \mathbf{x} as right pseudoinverse

$$\begin{aligned}2[\mathbf{Ax} - \mathbf{y}]^T &= 2\left\{\mathbf{A}\left[\mathbf{A}^T(\mathbf{AA}^T)^{-1}\mathbf{y}\right] - \mathbf{y}\right\}^T = 2\left\{(\mathbf{AA}^T)(\mathbf{AA}^T)^{-1}\mathbf{y} - \mathbf{y}\right\}^T \\ &= 2[\mathbf{y} - \mathbf{y}]^T = 0\end{aligned}$$

- Therefore, \mathbf{x} is the minimizing solution, as long as \mathbf{AA}^T is non-singular