

Preliminaries

- Office Hours
 - Tues Thurs, 1:30-3pm
- Assistants in Instruction:
 - Will Coogan
 - Yibin Zhang
 - Office hours: TBD
 - Precepts, tutorials: TBD

GRADING

- Class participation: 10%
- Assignments: 45%
- Term Paper: 45%
- MATLAB/SimuLink, STK-AGI, CREO
- Course Home Page, Syllabus, and Links
 - www.princeton.edu/~stengel/MAE342.html
- Wednesday afternoon "Lab Sessions" following regular class: TBD

Text and References

- Principal textbook:
- Spacecraft Systems Engineering, Fortescue, Stark, and Swinerd, J. Wiley & Sons, 2011
- Supplemental references
 - Space Mission Analysis and Design, Wertz et al, Microcosm Press, 2011
 - Fundamentals of Space Systems, V. L. Pisacane, Oxford University Press, 2005
 - Various technical reports and papers (e.g., NASA and AIAA pubs)
 - Books on reserve at Engineering Library (paper and on-line)
 - Web pages
 - http://blackboard.princeton.edu/
 - http://www.princeton.edu/~stengel/MAE342.html

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First Half of the Term

- Overview and Preliminaries
- Orbital Mechanics
- Planetary Defense
- Spacecraft Guidance
- Spacecraft Environment
- Chemical/Nuclear Propulsion Systems
- Electric Propulsion Systems
- Launch Vehicles
- Spacecraft Structures
- Spacecraft Configurations
- Spacecraft Dynamics
- Spacecraft Control



Second Half of the Term

- System Engineering & Integration
- Sensors & Actuators
- Electrical Power Systems
- Thermal Control
- Telecommunications
- Telemetry, Command, Data Handling & Processing
- Spacecraft Mechanisms
- Electromagnetic Compatibility
- Space Robotics
- Human Factors of Spaceflight
- Product Assurance
- Ground Segment





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Electronic Devices in Class

- Silence all cellphones and computer alarms
- Don't check e-mail or send text, tweets, etc.
- If you <u>must</u> make a call or send a message, you may leave the room to do so
- Tablets/laptops for class-related material ONLY

Collaborative Learning

- Significant student participation in most classes, Q&A
- Slides will be available before each class
- Discussion of slides by students
- Randomly assigned teams for assignments during first half
- Project-oriented teams during second half
- Single grade for each team

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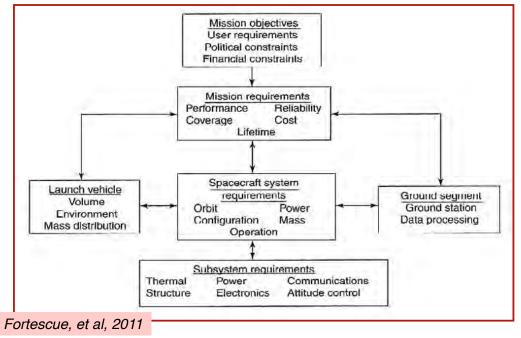
Written Assignment Reporting Format

- Assignments evolve toward Technical Reports
 - Abstract
 - Introduction
 - Methods and Results
 - Conclusion
 - References
- Write-ups should present explanations, not just numbers, graphs, or computer code
- Orderliness and neatness count
- Don't forget your name, date, and assignment title and number

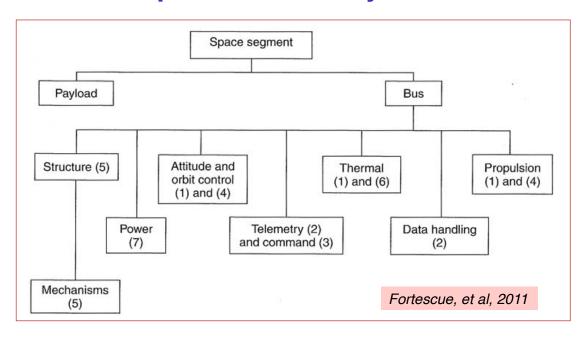
Overview

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Spacecraft Mission Objectives and Requirements



Spacecraft Subsystems



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Functional Requirements of Spacecraft Subsystems

- 1. Desired dynamic state (attitude, position, velocity) must be maintained
- 2. Desired orbits for the mission must be maintained
- 3. Payload must be operable
- 4. Payload must be held together and mounted on the spacecraft structure
- 5. Payload must operate reliably over some specified period
- 6. Adequate power must be provided
- 7. Operation should be largely autonomous, but ...
 - a) Data must be communicated to the ground
 - b) Ground control must be maintained

Brief History of Spaceflight

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Early History of Space Systems

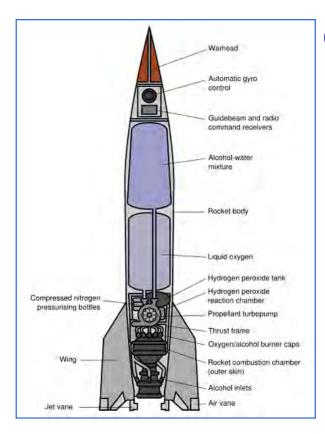
- 1926-1945: Goddard rockets; V-2 and its precursors
 - -Development of rocket technology
 - -Development of guidance and control systems
- 1945-1949: Learning from the V-2; Altitude sounding
 - -V-2/WAC-Corporal to 250-mi altitude (Project Bumper)
 - -Development of military missiles





German A-3





German A-4 (V-2) Rocket



- Liquid-fuel rocket
- 6,084 built; 1000+ test flights; 3,225 launched in combat
- Gyroscopes and accelerometer for guidance
- Air and jet vanes for pitch, yaw, and roll control torques
- Aft tail for aerodynamic stability

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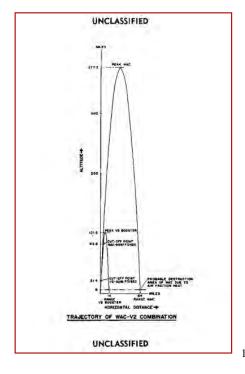
Project Bumper (V-2/WAC Corporal)







- 8 flights, 4 failures; Mach 9, 400-km apogee
- Engineering development
- High-altitude photography
- Atmospheric temperature profile
- Cosmic radiation



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Post-WWII History of Space Systems

- 1945-1957: Payload design; animals in space
 - -Sounding rockets
 - –Aerobee, Viking → Vanguard
 - -IRBMs and ICBMs received major emphasis in US and USSR
- 1957-1961: Unmanned satellites; animals in orbit; manned spaceflight about the Earth

-1957: Sputnik 1

−1958: Van Allen belts (Explorer 1); NACA -> NASA

-1959: Luna 1-3

-1961: Gagarin orbit; Ham and Shepard sub-orbit; Enos orbit

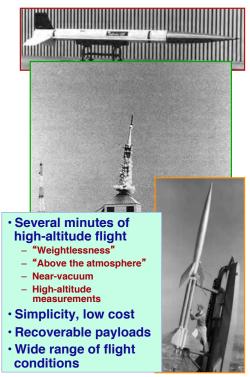


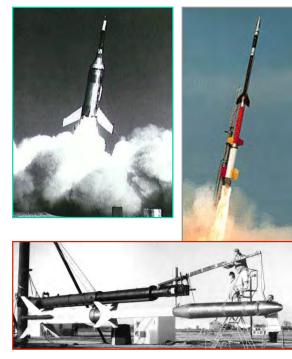




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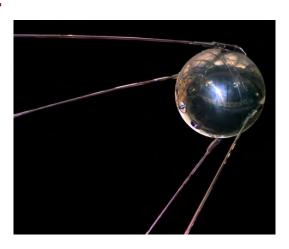
Sounding Rockets





Sputnik 1 (October 4, 1957)

- 84 kg, 58-cm diameter
- 96-min, elliptical orbit
- 1,440 orbits
- Measurements
 - Gravity
 - lonospheric effects
 - Internal temperature and pressure
 - Micrometeoroid detection



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R-7 (Semyorka) Launch Vehicle (October 4, 1957)



- 1-1/2-stage ICBM
- 4 strap-on booster rockets
- 1 core-stage rocket
- Liquid oxygen and kerosene
- Lift-off thrust: 3.9 MN
- Gross weight: 267 metric tons





Project Vanguard (1957-1959)



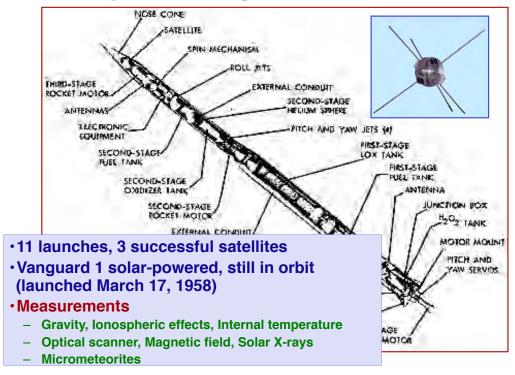


3 stages

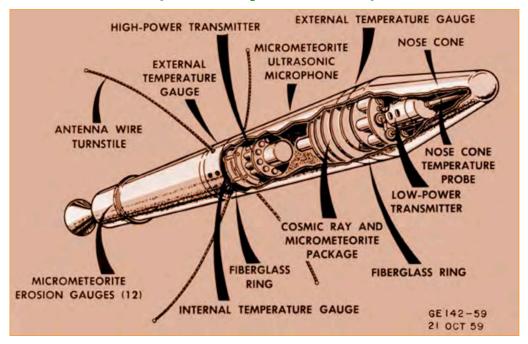
- 1st-stage based on Viking; gimballed motor for control
- 2nd-stage based on Aerobee; reaction-control thrusters
- Solid-fuel 3rd stage; spin stabilized

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Project Vanguard (1957-1959)



Explorer 1 (January 31, 1958)

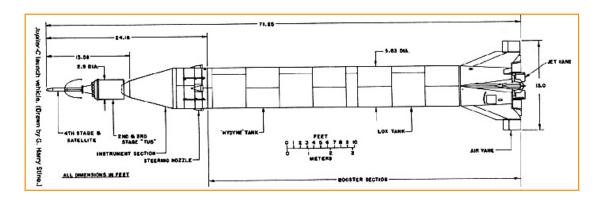


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Juno 1 Launched Explorer 1 (January 31, 1958)

Juno lineage from V-2

- Jupiter
- Redstone



Earth Satellite "Firsts"

Communications satellites

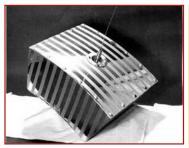
- 1960: Echo 1

1961: First amateur radio satellite (OSCAR 1)

- 1962: Telstar 1

1963: Geosynchronous satellite (Syncom 1)









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Earth Satellite "Firsts"

• 1960: Weather satellite: TIROS-1

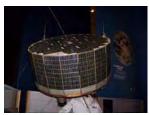
• 1972: Earth observation satellite: Landsat 1 (ERTS 1)

• 1962: Navigation satellite: Transit

1962: Astronomical satellite: Ariel (UK/US)









Lunar Probe "Firsts"

·1959:

- -Lunar flyby (Luna 1, Pioneer 4)
- -Lunar impact (Luna 2)
- -Pictures of "The Far Side" (Luna 3)

·1966:

- -Lunar soft landing(Luna 9, Surveyor 1)
- -Lunar orbit (Lunar Orbiter 1)













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Typical Lunar Probe Instrumentation, 1959

Radios for telemetry and command Cosmic ray counter Magnetometers Temperature Pressure Micrometeorite sensors Sodium cloud release

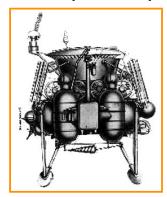


Lunar Probe "Firsts"



- · 1967:
- High-resolution photos (Ranger 7)
- · 1969: Apollo 11
- · 1970:
- Robotic sample return (Luna 16)
- Robotic lunar rover (Luna 17)





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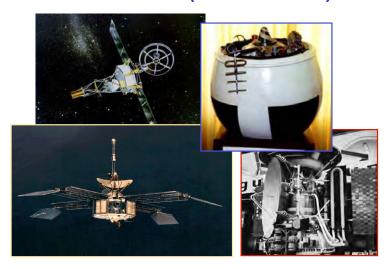
Inner-Planet Probe "Firsts"

• 1962: Venus flyby (Mariner 2)

• 1964: Mars flyby (Mariner 4)

• 1970: Venus lander (Venera 8)

• 1971: Mars orbit (Mars 2 Orbiter)



Inner-Planet Probe "Firsts"

•1973: Mercury flyby (Mariner 10)

•1975: Mars landing (Viking 1)

•1978: Venus orbit (Pioneer Venus 1)

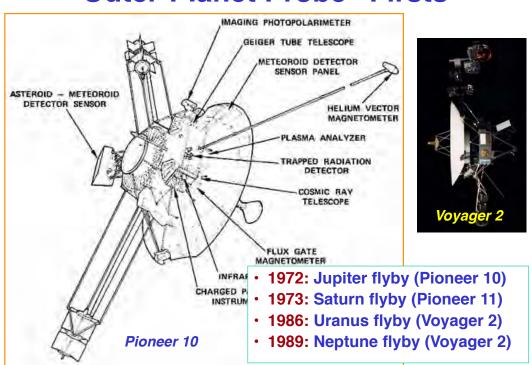


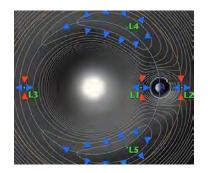




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Outer-Planet Probe "Firsts"

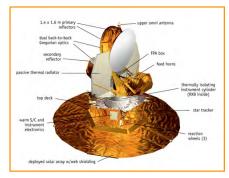




Lagrange-Point "Firsts"

- 5 equilibrium points in a rotating 2-body system, e.g.,
 - Sun-Earth
 - Earth-Moon
- 1978: Solar observatory at L1 (ISEE-3); later rendezvoused with a comet as ICE (1983)
- 2001: Astronomical observatory at L2 (Wilkinson Microwave Anisotropy Probe)





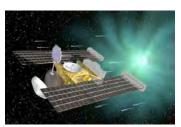
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Comet and Asteroid Rendezvous "Firsts"

1999: Comet sample return (Stardust)

• 2005: Asteroid landing (Muses-C/Hayabusa)

• 2014: Comet 67P Flyby/Landing (Rosetta/Philae)









Manned Spacecraft Launch Vehicles

1961: Gagarin orbit (Vostok); Shepard sub-orbit (Mercury/Redstone)

•1962: Glenn orbit (Mercury/ Atlas)

•1964: USSR 3-person crew in orbit (Voshkod)

•1965: US 2-person crew in orbit (Gemini/Titan II)









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Manned Spacecraft

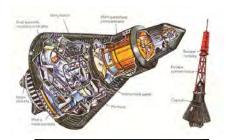
•1961: Gagarin orbit (Vostok); Shepard sub-orbit (Mercury/Redstone)

•1962: Glenn orbit (Mercury/Atlas)

1963: X-15 rocket plane reaches 100-km altitude
1964: USSR 3-person crew in orbit (Voshkod)
1965: US 2-person crew in orbit (Gemini/Titan II)









Manned Flight to the Moon

• 1961-1972: Apollo Program

- 6 lunar landings and returns



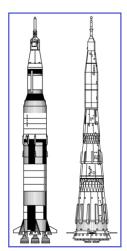


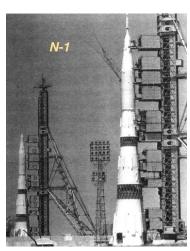


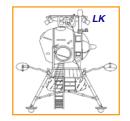
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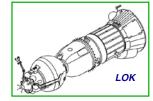
Manned Flight to the Moon

- 1961-1974: Soviet Lunar Program
 - 4 launches (unmanned), none successful









Space Stations

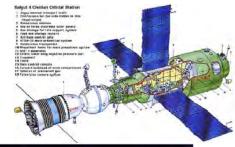
•1971-1982: Salyut •1973-1974: Skylab

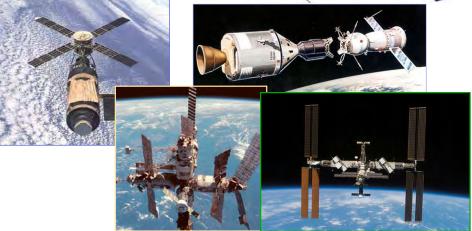
•1975: Soyuz-Apollo docking

•1986-2001: Mir

•1998-present: International

Space Station





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Space Shuttle



1981-2011:

- 5-7 astronauts
- 50,000-lb payload
- 135 missions flown
- 5 operational vehicles; 2 destroyed

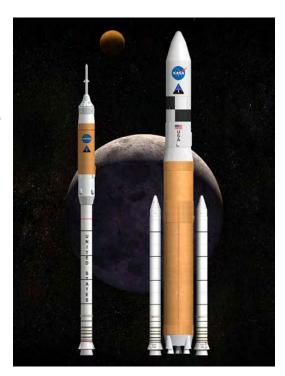
• 1986: Challenger accident

· 2003: Columbia accident



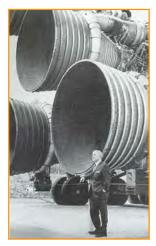
Project Constellation

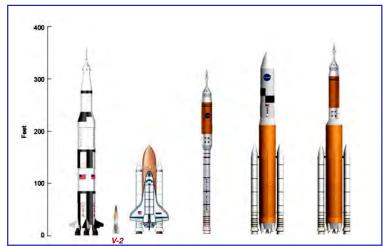
- Orion: Crew Spacecraft
- Ares 1: Crew Launch Vehicle
 - First (only) unmanned launch: 2009
 - 5-segment Shuttle Solid-Rocket Booster
- Ares 5: Cargo Launch Vehicle
- Manned return to the Moon
- Project cancelled in 2011
- Development of Orion continues
- Ares 5 morphed into the Space Launch System (SLS)



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Saturn V - Space Shuttle – Ares - SLS Size Comparison





Planetary Defense Term Project



- Design of spacecraft to protect against asteroid/ comet impact that would extinguish life on Earth
- Detection, characterization, intercept, and deflection of a "Doomsday Rock"
- Design Teams to address distant and near-Earth intercepts
- Single final report written "with one voice" by the class as a Working Group

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Assignment #1 Report on the Book, Project Icarus

- Teams will discuss segments of the 1979 book during Lab Session, Feb 10th, including the following:
 - Overview
 - Main points
 - Conclusions to be drawn
- Team members TBD

Next Time: Orbital Mechanics

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Supplemental Material

Math Review

- Scalars and Vectors
- Sums and Multiplication
- Inner Product
- Derivatives and Integrals

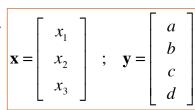
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Scalars and Vectors

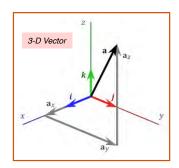
- Scalar: usually lower case: a, b, c, ..., x, y, z
- Vector: usually bold or with underbar: x or x
 - · Ordered set
 - Column of scalars
 - Dimension = n x 1

Transpose: interchange rows and columns

$$\mathbf{x}^T = \begin{bmatrix} x_1 & x_2 & x_3 \end{bmatrix}$$



3 x 1 4 x 1



Multiplication of Vector by Scalar

Multiplication of vector by scalar is associative, commutative, and distributive

$$a\mathbf{x} = \mathbf{x}a = \begin{bmatrix} ax_1 \\ ax_2 \\ ax_3 \end{bmatrix} \qquad a(\mathbf{x} + \mathbf{y}) = (\mathbf{x} + \mathbf{y})a = (a\mathbf{x} + a\mathbf{y})$$

$$\frac{\dim(\mathbf{x}) = \dim(\mathbf{y})}{\dim(\mathbf{y})}$$

$$\left| a\mathbf{x}^T = \left[\begin{array}{ccc} ax_1 & ax_2 & ax_3 \end{array} \right] \right|$$

• Could we add $(\mathbf{x} + a)$? • Only if $\dim(\mathbf{x}) = (1 \times 1)$

MATLAB allows it as an "overloaded function" https:// en.wikipedia.org/wiki/Function overloading

Addition

Conformable vectors and matrices are added term by term

$$\mathbf{x} = \left[\begin{array}{c} a \\ b \end{array} \right] \quad ; \quad \mathbf{z} = \left[\begin{array}{c} c \\ d \end{array} \right]$$

$$\mathbf{x} + \mathbf{z} = \begin{bmatrix} a+c \\ b+d \end{bmatrix}$$

Inner (Dot) Product

Inner (dot) product of vectors produces a scalar result

$$\mathbf{x}^{T}\mathbf{x} = \mathbf{x} \bullet \mathbf{x} = \begin{bmatrix} x_1 & x_2 & x_3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}$$

$$(1 \times m)(m \times 1) = (1 \times 1)$$

$$= (x_1^2 + x_2^2 + x_3^2)$$

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Derivatives and Integrals of Vectors

Derivatives and integrals of vectors are vectors of derivatives and integrals

$$\frac{d\mathbf{x}}{dt} = \begin{bmatrix} \frac{dx_1}{dt} \\ \frac{dx_2}{dt} \\ \frac{dx_3}{dt} \end{bmatrix}$$

$$\frac{d\mathbf{x}}{dt} = \begin{bmatrix} dx_1/dt \\ dt_2/dt \\ dx_3/dt \end{bmatrix} \qquad \int \mathbf{x} dt = \begin{bmatrix} \int x_1 dt \\ \int x_2 dt \\ \int x_3 dt \end{bmatrix}$$

$$\mathbf{x}(t) = \begin{bmatrix} 7 \\ 8t \\ 9t^2 \end{bmatrix}; \quad \frac{d\mathbf{x}(t)}{dt} = \begin{bmatrix} 0 \\ 8 \\ 18t \end{bmatrix}$$

$$\mathbf{x}(t) = \begin{bmatrix} 7 \\ 8t \\ 9t^2 \end{bmatrix}; \quad \int \mathbf{x}(t)dt = \begin{bmatrix} 7t + x_1(0) \\ 8t^2/2 + x_2(0) \\ 9t^3/3 + x_3(0) \end{bmatrix}$$

$$\mathbf{x}(t) = \begin{bmatrix} 7 \\ 8t \\ 9t^2 \end{bmatrix}; \quad \int \mathbf{x}(t)dt = \begin{bmatrix} 7t + x_1(0) \\ 8t^2/2 + x_2(0) \\ 9t^3/3 + x_3(0) \end{bmatrix}$$

MATLAB Code for Math Review

MATLAB Code for Math Review

```
Vector Addition
   zz = [8; 9; 10]

u = x + zz
% Inner (Dot) Product
    ZZZ = X' * X
  Symbolic Toolbox
    disp(' ')
    disp('Symbolic Toolbox')
    disp(' ')
    syms x y z z1 z2 z3 z4
   y = x * x
z = diff(y)
z1 = int(y)
                       %
                           Define Function
                       %
                           Differentiate Function
                     % Integrate Function
    z2 = [x; y; z] % Column Vector
    z3 = diff(z2)
                       % Derivative of Column Vector
    z4 = int(z2) % Integral of Column Vector
```

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MATLAB Command Window Output for Math Review

>>>M/	AE 34	15 Le	ctur	e 1	Math	Rev	iew<<<	
Date	and	Time	are	24	-May-	2013	12:31	:13
a =	4							
X =	1 2 3							
y =	4 5 6 7							
xT =	1	2	3	3				
yT =	4	5	(6	7			

```
W =

4
8
12

V =

4
8
12

WT = 4
8
12

ZZ =

8
9
10

U =

9
11
13

ZZZ = 14
```

Symbolic Toolbox
y = x^2
z = 2*x
$z1 = x^3/3$
z2 = x x^2 2*x
z3 = 1 2*x 2
z4 = x^2/2 x^3/3 x^2

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Math Review

- Matrix and Transpose
- Sums and Multiplication
- Matrix Products
- Identity Matrix
- Matrix Inverse
- Transformations

Matrix and Transpose

- Matrix:
 - Usually bold capital or capital: F or F
 - Dimension = $(m \times n)$

$$\mathbf{A} = \begin{bmatrix} a & b & c \\ d & e & f \\ g & h & k \\ l & m & n \end{bmatrix}$$

Transpose:

Interchange rows and columns

$$\mathbf{A} = \begin{bmatrix} a & b & c \\ d & e & f \\ g & h & k \\ l & m & n \end{bmatrix} \qquad \mathbf{A}^T = \begin{bmatrix} a & d & g & l \\ b & e & h & m \\ c & f & k & n \end{bmatrix}$$

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Matrix Products

Matrix-vector product transforms one vector into another

$$\mathbf{y} = A\mathbf{x} = \begin{bmatrix} a & b & c \\ d & e & f \\ g & h & k \\ l & m & n \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix} = \begin{bmatrix} ax_1 + bx_2 + cx_3 \\ dx_1 + ex_2 + fx_3 \\ gx_1 + hx_2 + kx_3 \\ lx_1 + mx_2 + nx_3 \end{bmatrix}$$

$$(n \times 1) = (n \times m)(m \times 1)$$

Matrix-matrix product produces a new matrix

$$\mathbf{A} = \begin{bmatrix} a_1 & a_2 \\ a_3 & a_4 \end{bmatrix}; \quad \mathbf{B} = \begin{bmatrix} b_1 & b_2 \\ b_3 & b_4 \end{bmatrix}; \quad \mathbf{AB} = \begin{bmatrix} (a_1b_1 + a_2b_3) & (a_1b_2 + a_2b_4) \\ (a_3b_1 + a_4b_3) & (a_3b_2 + a_4b_4) \end{bmatrix}$$

$$(n \times m) = (n \times l)(l \times m)$$

Numerical Example 1

$$\mathbf{y} = \mathbf{A}\mathbf{x} = \begin{bmatrix} 2 & 4 & 6 \\ 3 & -5 & 7 \\ 4 & 1 & 8 \\ -9 & -6 & -3 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \end{bmatrix}$$

$$(n \times 1) = (n \times m)(m \times 1)$$

$$= \begin{bmatrix} (2x_1 + 4x_2 + 6x_3) \\ (3x_1 - 5x_2 + 7x_3) \\ (4x_1 + x_2 + 8x_3) \\ (-9x_1 - 6x_2 - 3x_3) \end{bmatrix} = \begin{bmatrix} y_1 \\ y_2 \\ y_3 \\ y_4 \end{bmatrix}$$

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Numerical Example 2

$$\mathbf{A} = \begin{bmatrix} 1 & 2 \\ 3 & 4 \end{bmatrix} \; ; \quad \mathbf{B} = \begin{bmatrix} 5 & 6 \\ 7 & 8 \end{bmatrix} \quad \mathbf{AB} = \begin{bmatrix} (5+14) & (6+16) \\ (15+28) & (18+32) \end{bmatrix} = \begin{bmatrix} 19 & 22 \\ 43 & 50 \end{bmatrix}$$

$$\mathbf{x}_{A} = \mathbf{A}\mathbf{x}_{B} \quad ; \quad \begin{bmatrix} x_{1} \\ x_{2} \end{bmatrix}_{A} = \begin{bmatrix} 1 & 2 \\ 3 & 4 \end{bmatrix} \begin{bmatrix} x_{1} \\ x_{2} \end{bmatrix}_{B}$$

$$\mathbf{x}_{B} = \mathbf{B}\mathbf{x}_{o} \quad ; \quad \begin{bmatrix} x_{1} \\ x_{2} \end{bmatrix}_{R} = \begin{bmatrix} 5 & 6 \\ 7 & 8 \end{bmatrix} \begin{bmatrix} x_{1} \\ x_{2} \end{bmatrix}_{o}$$

$$\mathbf{x}_{A} = \mathbf{A}\mathbf{x}_{B} = \mathbf{A}\mathbf{B}\mathbf{x}_{o} \quad ; \quad \begin{bmatrix} x_{1} \\ x_{2} \end{bmatrix}_{A} = \begin{bmatrix} 19 & 22 \\ 43 & 50 \end{bmatrix} \begin{bmatrix} x_{1} \\ x_{2} \end{bmatrix}_{o}$$

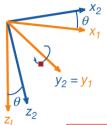
Square Matrix Identity and Inverse

- Identity matrix: no change when it multiplies a conformable vector or matrix

A non-singular square matrix multiplied by its inverse forms an identity matrix

$$\mathbf{A}\mathbf{A}^{-1} = \mathbf{A}^{-1}\mathbf{A} = \mathbf{I}$$

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Matrix Inverse Example

Transformation $|\mathbf{x}_2 = \mathbf{A}\mathbf{x}_1|$

$$\mathbf{x}_2 = \mathbf{A}\mathbf{x}_1$$

$$\begin{bmatrix} x \\ y \\ z \end{bmatrix}_{2} = \begin{bmatrix} \cos\theta & 0 & -\sin\theta \\ 0 & 1 & 0 \\ \sin\theta & 0 & \cos\theta \end{bmatrix} \begin{bmatrix} x \\ y \\ z \end{bmatrix}_{1}$$

Inverse Transformation $\mathbf{x}_1 = \mathbf{A}^{-1}\mathbf{x}_2$

$$\begin{bmatrix} x \\ y \\ z \end{bmatrix}_{1} = \begin{bmatrix} \cos\theta & 0 & \sin\theta \\ 0 & 1 & 0 \\ -\sin\theta & 0 & \cos\theta \end{bmatrix} \begin{bmatrix} x \\ y \\ z \end{bmatrix}_{2}$$

Consequently, ...

$$\mathbf{A}\mathbf{A}^{-1} = \mathbf{A}^{-1}\mathbf{A} = \mathbf{I}$$

$$\mathbf{A}\mathbf{A}^{-1} = \begin{bmatrix} \cos\theta & 0 & -\sin\theta \\ 0 & 1 & 0 \\ \sin\theta & 0 & \cos\theta \end{bmatrix} \begin{bmatrix} \cos\theta & 0 & -\sin\theta \\ 0 & 1 & 0 \\ \sin\theta & 0 & \cos\theta \end{bmatrix}^{-1}$$

$$= \begin{bmatrix} \cos\theta & 0 & -\sin\theta \\ 0 & 1 & 0 \\ \sin\theta & 0 & \cos\theta \end{bmatrix} \begin{bmatrix} \cos\theta & 0 & \sin\theta \\ 0 & 1 & 0 \\ -\sin\theta & 0 & \cos\theta \end{bmatrix}$$

$$\mathbf{x}_{2} = \mathbf{A}\mathbf{x}_{1} = \mathbf{A}\mathbf{A}^{-1}\mathbf{x}_{2} = \mathbf{x}_{2}$$

$$= \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 0 \\ 0 & 0 & 1 \end{bmatrix}$$

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Computation of (n x n) Matrix Inverse

$$\mathbf{y} = \mathbf{A}\mathbf{x}; \quad \mathbf{x} = \mathbf{A}^{-1}\mathbf{y}$$

 $\dim(\mathbf{x}) = \dim(\mathbf{y}) = (n \times 1); \quad \dim(\mathbf{A}) = (n \times n)$

$$[\mathbf{A}]^{-1} = \frac{\operatorname{Adj}(\mathbf{A})}{|\mathbf{A}|} = \frac{\operatorname{Adj}(\mathbf{A})}{\det \mathbf{A}} \quad \frac{(n \times n)}{(1 \times 1)}$$
$$= \frac{\mathbf{C}^{T}}{\det \mathbf{A}}; \quad \mathbf{C} = matrix \ of \ cofactors$$

Cofactors are signed minors of A

ijth minor of **A** is the determinant of **A** with the ith row and ith column removed

MATLAB Code for Math Review Use of Symbolic Variables

```
% MAE 345 Lecture 2 Math Review
Rob Stengel

clear
disp(' ')
disp('========')
disp('>>>MAE 345 Lecture 2 Math Review<<<')
disp('========')
disp(' ')
disp(['Date and Time are ', num2str(datestr(now))]);
disp(' ')

% Matrix
syms A AT a b c d e f g h k l m n
A = [a b c;d e f;g h k;l m n] % Matrix
AT = A' % Matrix Transpose

% Matrix-Vector Product
syms x x1 x2 x3 y1 y2 y3 y4
x = [x1;x2;x3]
y = [y1;y2;y3;y4]
y = A * x</pre>
```

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MATLAB Code for Math Review

```
% Matrix-Matrix Product
syms A a1 a2 a3 a4 B b1 b2 b3 b4 AB
A = [a1 a2;a3 a4]
B = [b1 b2;b3 b4]
AB = A * B

% Example 1
syms A
A = [2 4 6;3 -5 7;4 1 8;-9 -6 -3]
y = A * x

% Example 2
A = [1 2;3 4]
B = [5 6;7 8]
AB = A * B

syms xA xB x0
x0 = [x1;x2]
xA = A * xB
xB = B * x0
xA = A * B * x0
```

MATLAB Code for Math Review

```
Matrix Identity and Inverse
I3
            eye(3)
        =
            I3 * x
Х
syms A Ainv
            [a b c;d e f;g h k]
            inv(A)
Ainv
            simplify(A * Ainv)
I3
13
            simplify(Ainv * A)
Matrix Inverse Example
syms A Th cTh sTh Ainv
        = [cTh 0 sTh; 0 1 0; -sTh 0 cTh]
Ainv
             inv(A)
detA
             det(A)
cTh
            cos(Th)
        =
sTh
             sin(Th)
Th
             pi / 4
syms A Ainv
             [cos(Th) 0 sin(Th); 0 1 0; -sin(Th) 0 cos(Th)]
Ainv
Consequently, ...
I3 = A * Ainv
Computation of (n \times n) Inverse
       = det(A)
= Ainv * detA
detA
AdjA
```

MATLAB Command Window Output for Math Review

```
>>>MAE 345 Lecture 2 Math Review<<<pre>Date and Time are 03-Sep-2013 13:49:40

A =
    [ a, b, c]
    [ d, e, f]
    [ g, h, k]
    [ l, m, n]

AT =
    [ conj(a), conj(d), conj(g), conj(l)]
    [ conj(b), conj(e), conj(h), conj(m)]
    [ conj(c), conj(f), conj(k), conj(n)]

x =
    x1
    x2
    x3

y =
    y1
    y2
    y3
    y4

y =
    a*x1 + b*x2 + c*x3
    d*x1 + e*x2 + f*x3
    g*x1 + h*x2 + k*x3
    l*x1 + m*x2 + n*x3
```

```
x0 =
x1
x2

xA =
[ xB, 2*xB]
[ 3*xB, 4*xB]

xB =
5*x1 + 6*x2
7*x1 + 8*x2

xA =
19*x1 + 22*x2
43*x1 + 50*x2

I3 =

1 0 0
0 1 0
0 0 1

x =
x1
x2
x3
```

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MATLAB Command Window Output for Math Review

```
A = [ a, b, c] [ d, e, f] [ g, h, k]

Ainv = [ (f*h - e*k)/(a*f*h - b*f*g - c*d*h + c*e*g - a*e*k + b*d*k), -(c*h - b*k)/(a*f*h - b*f*g - c*d*h + c*e*g - a*e*k + b*d*k), -(b*f - c*e)/(a*f*h - b*f*g - c*d*h + c*e*g - a*e*k + b*d*k)] [ -(f*g - d*k)/(a*f*h - b*f*g - c*d*h + c*e*g - a*e*k + b*d*k)] [ -(f*g - a*k)/(a*f*h - b*f*g - c*d*h + c*e*g - a*e*k + b*d*k), (a*f - c*d)/(a*f*h - b*f*g - c*d*h + c*e*g - a*e*k + b*d*k), (a*f - c*d)/(a*f*h - b*f*g - c*d*h + c*e*g - a*e*k + b*d*k), (a*f - c*d)/(a*f*h - b*f*g - c*d*h + c*e*g - a*e*k + b*d*k), (a*h - b*g)/(a*f*h - b*f*g - c*d*h + c*e*g - a*e*k + b*d*k), -(a*e - b*d)/(a*f*h - b*f*g - c*d*h + c*e*g - a*e*k + b*d*k)]

I3 = [ 1, 0, 0] [ 0, 1, 0] [ 0, 0, 1]

I3 = [ 1, 0, 0] [ 0, 0, 1]
```

```
A = [ cTh, 0, sTh]
        [ 0, 1, 0]
        [ -sTh, 0, cTh]
Ainv =
This is a standard for the standard form of the sta
detA = cTh^2 + sTh^2
cTh = cos(Th)
sTh = sin(Th)
Th = 0.7854
A = 0.7071
                                                                                                                                                                              0.7071
                                                                                                         1.0000
                                                                                                                                                                                      0.7071
                 -0.7071
Ainv = 0.7071
                                                                                                                                                                    0 -0.7071
                                                                                                                          1.0000
                                                  0.7071
                                                                                                                                                                   0 0.7071
I3 = 1
                                                                            0
                                     0
detA = 1
AdjA = 0.7071
                                                                                                                                                                                               -0.7071
                                                                                                                               1.0000
                                                                                                                                                                                                                                                                                                          69
                                                   0.7071
                                                                                                                                                                                                     0.7071
```