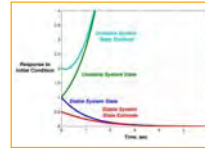
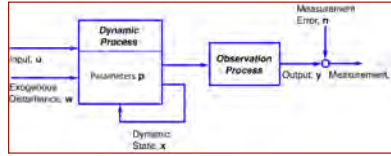


Linear-Optimal Estimation

Robert Stengel
Optimal Control and Estimation MAE 546
Princeton University, 2015



Linear-optimal Gaussian estimator for discrete-time system (**Kalman filter**)

2nd-order example

Alternative forms of the Kalman filter equations

M conditioning

Correlated inputs and measurement noise

Time-correlated measurement noise

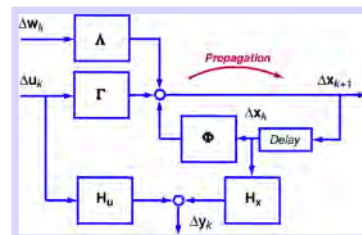
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<http://www.princeton.edu/~stengel/MAE546.html>
<http://www.princeton.edu/~stengel/OptConEst.html>

1

Uncertain Linear, Time-Varying (LTV) Dynamic Model

- Discrete-time LTV model with known coefficients



$$\mathbf{x}_k = \Phi_{k-1} \mathbf{x}_{k-1} + \Gamma_{k-1} \mathbf{u}_{k-1} + \Lambda_{k-1} \mathbf{w}_{k-1}$$

$$\mathbf{z}_k = \mathbf{H}_k \mathbf{x}_k + \mathbf{n}_k$$

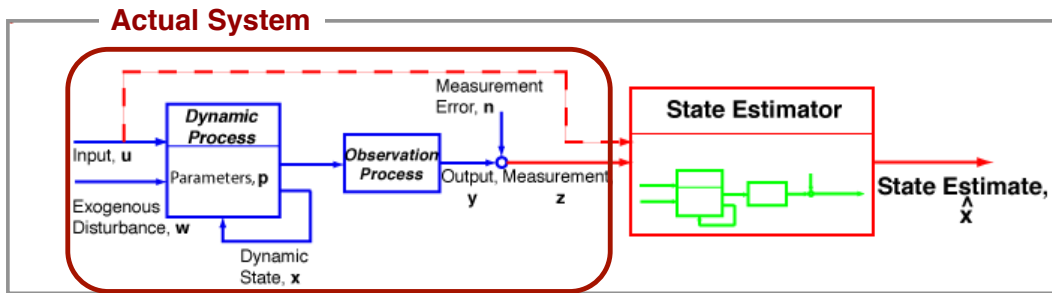
$$\dim(\mathbf{w}_k) = s \times 1$$

$$\dim(\mathbf{z}_k) = r \times 1$$

- Initial condition and disturbance inputs are not known precisely
- Measurement of state is transformed and is subject to error

2

State Estimation



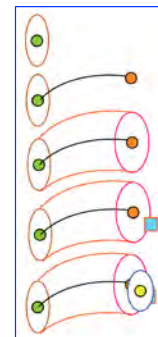
- **Goals**
 - Minimize effects of measurement error on knowledge of the state
 - Reconstruct full state from reduced measurement set ($r \leq n$)
 - Average redundant measurements ($r \geq n$) to estimate the full state
- **Method**
 - Provide optimal balance between measurements and estimates based on the dynamic model alone

3



Linear-Optimal State Estimation

- **Kalman filter** is the optimal estimator for discrete-time linear systems with Gaussian uncertainty
- It has five equations
 - 1) State estimate extrapolation
 - 2) Covariance estimate extrapolation
 - 3) Filter gain computation
 - 4) State estimate update
 - 5) Covariance estimate “update”
- **Notation**



$\hat{\mathbf{x}}_k(-)$: Estimate at k^{th} instant **before** measurement update
 $\hat{\mathbf{x}}_k(+)$: Estimate at k^{th} instant **after** measurement update

4

Equations of the Kalman Filter

1) State estimate extrapolation (or propagation)

$$\hat{\mathbf{x}}_k(-) = \Phi_{k-1} \hat{\mathbf{x}}_{k-1}(+) + \Gamma_{k-1} \mathbf{u}_{k-1}$$

2) Covariance estimate extrapolation (or propagation)

$$\mathbf{P}_k(-) = \Phi_{k-1} \mathbf{P}_{k-1}(+) \Phi_{k-1}^T + \mathbf{Q}_{k-1}$$

5

Equations of the Kalman Filter

3) Filter gain computation

$$\mathbf{K}_k = \mathbf{P}_k(-) \mathbf{H}_k^T \left[\mathbf{H}_k \mathbf{P}_k(-) \mathbf{H}_k^T + \mathbf{R}_k \right]^{-1}$$

4) State estimate update

$$\hat{\mathbf{x}}_k(+) = \hat{\mathbf{x}}_k(-) + \mathbf{K}_k \left[\mathbf{z}_k - \mathbf{H}_k \hat{\mathbf{x}}_k(-) \right]$$

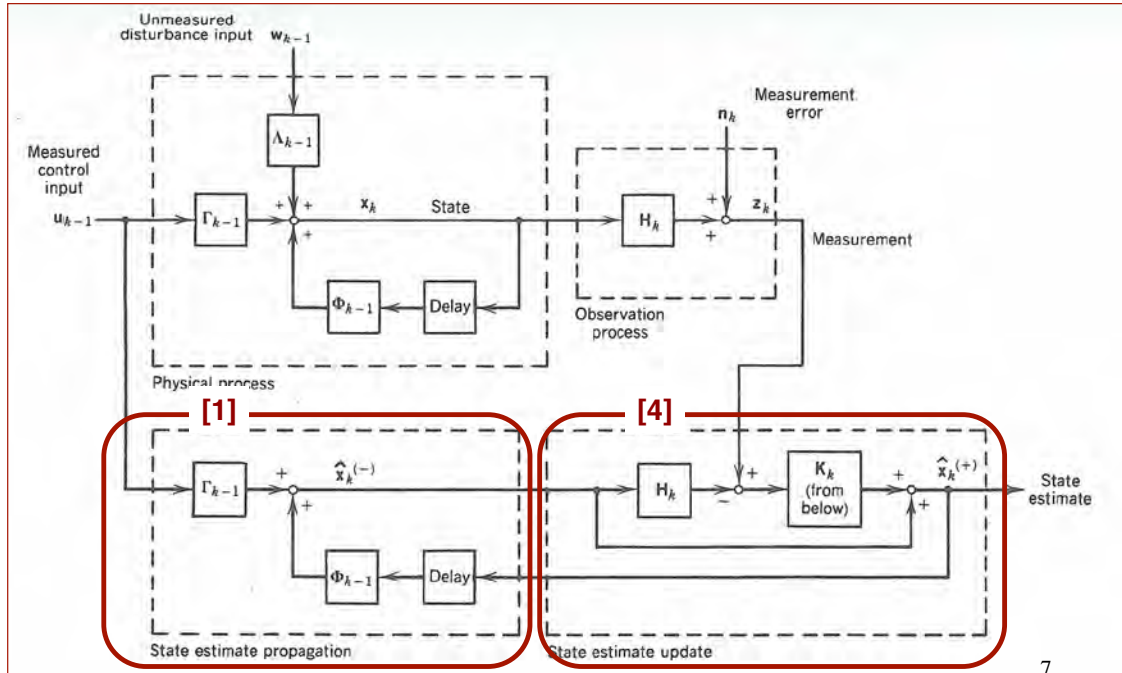
5) Covariance estimate “update”

$$\mathbf{P}_k(+) = \left[\mathbf{P}_k^{-1}(-) + \mathbf{H}_k^T \mathbf{R}_k^{-1} \mathbf{H}_k \right]^{-1}$$

6

Diagram of State Estimate

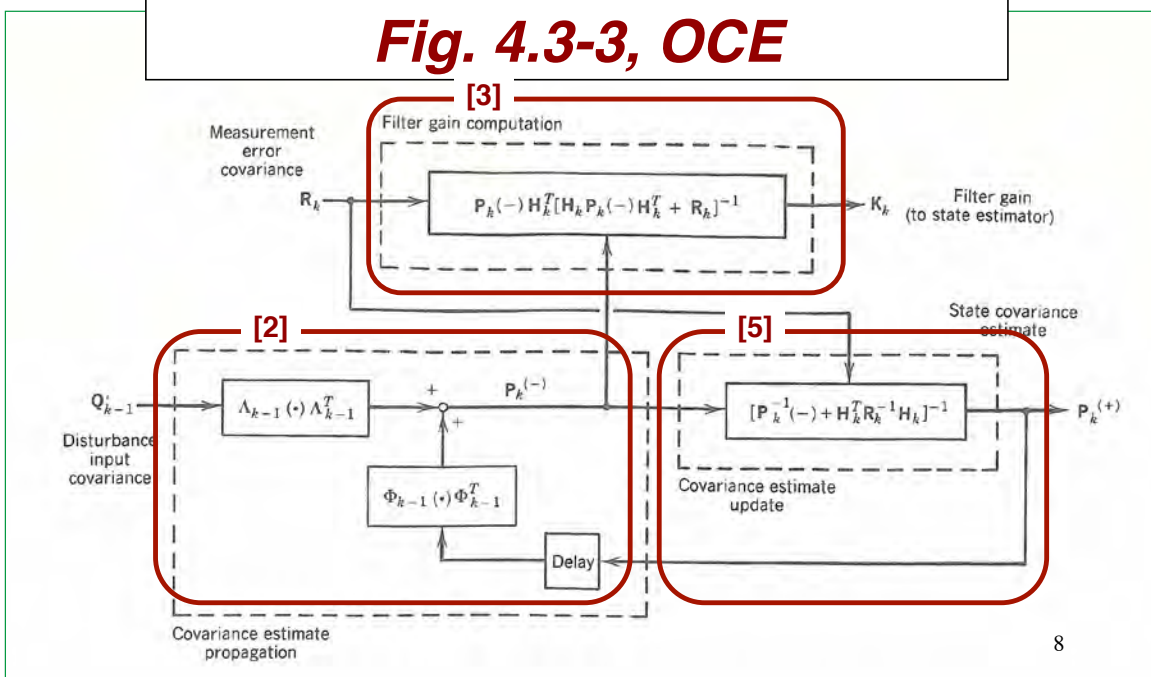
Fig. 4.3-2, OCE



7

Covariance Estimate and Filter Gain Matrix

Fig. 4.3-3, OCE



8

Alternative Expressions for \mathbf{K}_k

$$\begin{aligned} \mathbf{K}_k &= \mathbf{P}_k(-) \mathbf{H}_k^T \left[\mathbf{H}_k \mathbf{P}_k(-) \mathbf{H}_k^T + \mathbf{R}_k \right]^{-1} \\ \mathbf{I}_r &= \mathbf{R}_k^{-1} \mathbf{R}_k \\ \mathbf{K}_k &= \mathbf{P}_k(-) \mathbf{H}_k^T \mathbf{R}_k^{-1} \mathbf{R}_k \left[\mathbf{H}_k \mathbf{P}_k(-) \mathbf{H}_k^T + \mathbf{R}_k \right]^{-1} \\ \text{Matrix Identity} \quad \mathbf{K}_k &= \mathbf{P}_k(-) \mathbf{H}_k^T \mathbf{R}_k^{-1} \left[\mathbf{H}_k \mathbf{P}_k(-) \mathbf{H}_k^T \mathbf{R}_k^{-1} + \mathbf{I}_r \right]^{-1} \\ \mathbf{K}_k \left[\mathbf{I}_r + \mathbf{H}_k \mathbf{P}_k(-) \mathbf{H}_k^T \mathbf{R}_k^{-1} \right] &= \mathbf{P}_k(-) \mathbf{H}_k^T \mathbf{R}_k^{-1} \\ \text{Multiply by inverse} \\ \mathbf{K}_k &= \mathbf{P}_k(-) \mathbf{H}_k^T \mathbf{R}_k^{-1} - \mathbf{K}_k \mathbf{H}_k \mathbf{P}_k(-) \mathbf{H}_k^T \mathbf{R}_k^{-1} \\ \text{Subtract from both sides} \\ \mathbf{K}_k &= (\mathbf{I}_n - \mathbf{K}_k \mathbf{H}_k) \mathbf{P}_k(-) \mathbf{H}_k^T \mathbf{R}_k^{-1} \\ \text{Collect terms} \end{aligned}$$

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Alternative Expressions for \mathbf{K}_k and $\mathbf{P}_k(+)$

- From matrix inversion lemma

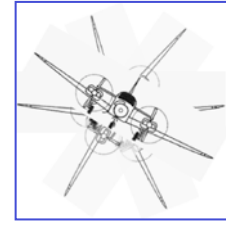
$$\begin{aligned} \mathbf{P}_k(+) &= \left[\mathbf{P}_k^{-1}(-) + \mathbf{H}_k^T \mathbf{R}_k^{-1} \mathbf{H}_k \right]^{-1} \\ &= \mathbf{P}_k(-) - \mathbf{P}_k(-) \mathbf{H}_k^T \left[\mathbf{H}_k \mathbf{P}_k(-) \mathbf{H}_k^T + \mathbf{R}_k \right]^{-1} \mathbf{H}_k \mathbf{P}_k(-) \\ &= (\mathbf{I}_n - \mathbf{K}_k \mathbf{H}_k) \mathbf{P}_k(-) \end{aligned}$$

- This covariance update does not inherently preserve symmetry
- With $\mathbf{P}_k(+)$ known, estimation gain is

$$\mathbf{K}_k = (\mathbf{I}_n - \mathbf{K}_k \mathbf{H}_k) \mathbf{P}_k(-) \mathbf{H}_k^T \mathbf{R}_k^{-1} = \mathbf{P}_k(+) \mathbf{H}_k^T \mathbf{R}_k^{-1}$$

10

Second-Order Example of Kalman Filter



- Rolling motion of an airplane, continuous-time

$$\begin{bmatrix} \Delta \dot{p} \\ \Delta \dot{\phi} \end{bmatrix} = \begin{bmatrix} L_p & 0 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} \Delta p \\ \Delta \phi \end{bmatrix} + \begin{bmatrix} L_{\delta A} \\ 0 \end{bmatrix} \Delta \delta A + \begin{bmatrix} L_p \\ 0 \end{bmatrix} \Delta p_w$$

$$\begin{bmatrix} \Delta p \\ \Delta \phi \end{bmatrix} = \begin{bmatrix} \text{Roll rate, rad/s} \\ \text{Roll angle, rad} \end{bmatrix}$$

$\Delta \delta A = \text{Aileron deflection, rad}$
 $\Delta p_w = \text{Turbulence disturbance, rad/s}$

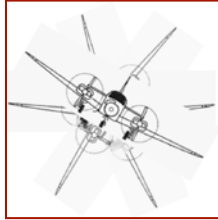
- Rolling motion of an airplane, discrete-time

$$\begin{bmatrix} \Delta p_k \\ \Delta \phi_k \end{bmatrix} = \begin{bmatrix} e^{L_p T} & 0 \\ \frac{(e^{L_p T} - 1)}{L_p} & 1 \end{bmatrix} \begin{bmatrix} \Delta p_{k-1} \\ \Delta \phi_{k-1} \end{bmatrix} + \begin{bmatrix} \sim L_{\delta A} T \\ 0 \end{bmatrix} \Delta \delta A_{k-1} + \begin{bmatrix} \sim L_p T \\ 0 \end{bmatrix} \Delta p_{w_{k-1}}$$

$$= \begin{bmatrix} \varphi_{11} & 0 \\ \varphi_{21} & 1 \end{bmatrix} \begin{bmatrix} \Delta p_{k-1} \\ \Delta \phi_{k-1} \end{bmatrix} + \begin{bmatrix} \gamma_1 \\ 0 \end{bmatrix} \Delta \delta A_{k-1} + \begin{bmatrix} \lambda_1 \\ 0 \end{bmatrix} \Delta p_{w_{k-1}}$$

$T = \text{sampling interval, s}$

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Second-Order Example of Kalman Filter

Rate and Angle Measurement

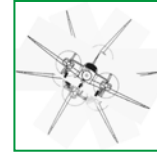
$$\begin{bmatrix} \Delta p_M \\ \Delta \phi_M \end{bmatrix}_k = \begin{bmatrix} \Delta p + \Delta n_p \\ \Delta \phi + \Delta n_\phi \end{bmatrix}_k = \mathbf{I} \Delta \mathbf{x}_k + \Delta \mathbf{n}_k$$

1) State Estimate Extrapolation

$$\begin{bmatrix} \Delta \hat{p}_k(-) \\ \Delta \hat{\phi}_k(-) \end{bmatrix} = \begin{bmatrix} \varphi_{11} & 0 \\ \varphi_{21} & 1 \end{bmatrix} \begin{bmatrix} \Delta \hat{p}_{k-1}(+) \\ \Delta \hat{\phi}_{k-1}(+) \end{bmatrix} + \begin{bmatrix} \gamma_1 \\ 0 \end{bmatrix} \Delta \delta A_{k-1}$$

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Second-Order Example of Kalman Filter



2) Covariance Extrapolation

$$\begin{bmatrix} p_{11}(-) & p_{12}(-) \\ p_{21}(-) & p_{22}(-) \end{bmatrix}_k = \begin{bmatrix} \varphi_{11} & 0 \\ \varphi_{21} & 1 \end{bmatrix} \begin{bmatrix} p_{11}(+) & p_{12}(+) \\ p_{21}(+) & p_{22}(+) \end{bmatrix}_{k-1} \begin{bmatrix} \varphi_{11} & \varphi_{21} \\ 0 & 1 \end{bmatrix} + \begin{bmatrix} \sigma_p^2 & 0 \\ 0 & 0 \end{bmatrix}$$

$$\mathbf{Q}_{k-1} \approx \begin{bmatrix} L_p \\ 0 \end{bmatrix} \mathbf{Q}'_C \begin{bmatrix} L_p & 0 \end{bmatrix}^T = \begin{bmatrix} \sigma_p^2 & 0 \\ 0 & 0 \end{bmatrix}$$

3) Gain Computation

$$\begin{bmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{bmatrix}_k = \begin{bmatrix} p_{11}(-) & p_{12}(-) \\ p_{21}(-) & p_{22}(-) \end{bmatrix}_k \left\{ \begin{bmatrix} p_{11}(-) & p_{12}(-) \\ p_{21}(-) & p_{22}(-) \end{bmatrix}_k + \begin{bmatrix} \sigma_{p_M}^2 & 0 \\ 0 & \sigma_{\phi_M}^2 \end{bmatrix} \right\}^{-1}$$

$$\mathbf{R}_k \delta_{jk} = \begin{bmatrix} \sigma_{p_M}^2 & 0 \\ 0 & \sigma_{\phi_M}^2 \end{bmatrix}_k$$

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Second-Order Example of Kalman Filter

4) State Estimate Update

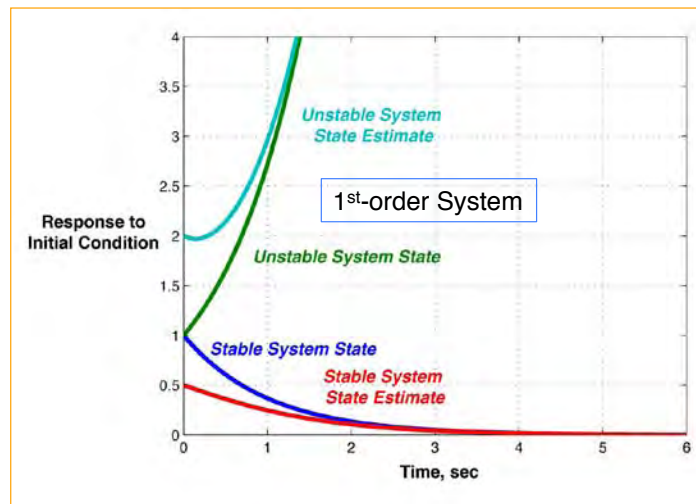
$$\begin{bmatrix} \Delta \hat{p}_k(+) \\ \Delta \hat{\phi}_k(+) \end{bmatrix} = \begin{bmatrix} \Delta \hat{p}_k(-) \\ \Delta \hat{\phi}_k(-) \end{bmatrix} + \begin{bmatrix} k_{11} & k_{12} \\ k_{21} & k_{22} \end{bmatrix}_k \left\{ \begin{bmatrix} \Delta p_{M_k} \\ \Delta \phi_{M_k} \end{bmatrix} - \begin{bmatrix} \Delta \hat{p}_k(-) \\ \Delta \hat{\phi}_k(-) \end{bmatrix} \right\}$$

5) Covariance "Update"

$$\begin{bmatrix} p_{11}(+) & p_{12}(+) \\ p_{21}(+) & p_{22}(+) \end{bmatrix}_k = \left\{ \begin{bmatrix} p_{11}(-) & p_{12}(-) \\ p_{21}(-) & p_{22}(-) \end{bmatrix}_k + \begin{bmatrix} \frac{1}{\sigma_{p_M}^2} & 0 \\ 0 & \frac{1}{\sigma_{\phi_M}^2} \end{bmatrix}_k \right\}^{-1}$$

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Example of Stability of Kalman Filter Estimate



Estimate is stable because it converges to actual state

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Example: Propagating a Scalar Probability Density Function

Scalar LTI system with zero-mean Gaussian random input

$$x_{k+1} = bx_k + \sqrt{1-b^2}u_k + \sqrt{1-b^2}w_k, \quad x_0 \text{ given}$$

Propagation of the mean value

$$\bar{x}_{i+1} = b\bar{x}_i + \sqrt{1-b^2}\bar{u}_i, \quad \bar{x}_0 \text{ given}$$

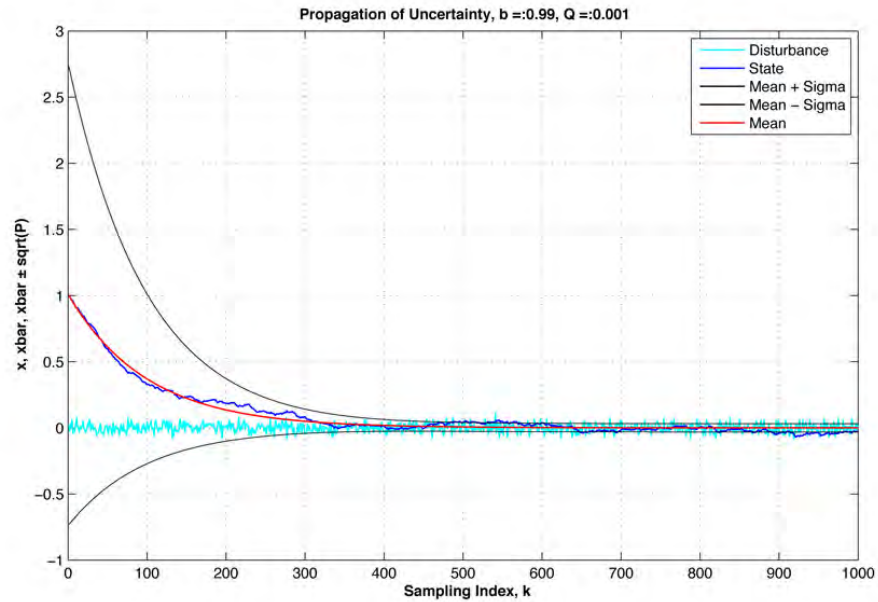
Propagation of the variance

$$P_{k+1} = b^2P_k + (1-b^2)Q_k, \quad P_0 \text{ given}$$

$$Q_k = E(w_k^2)$$

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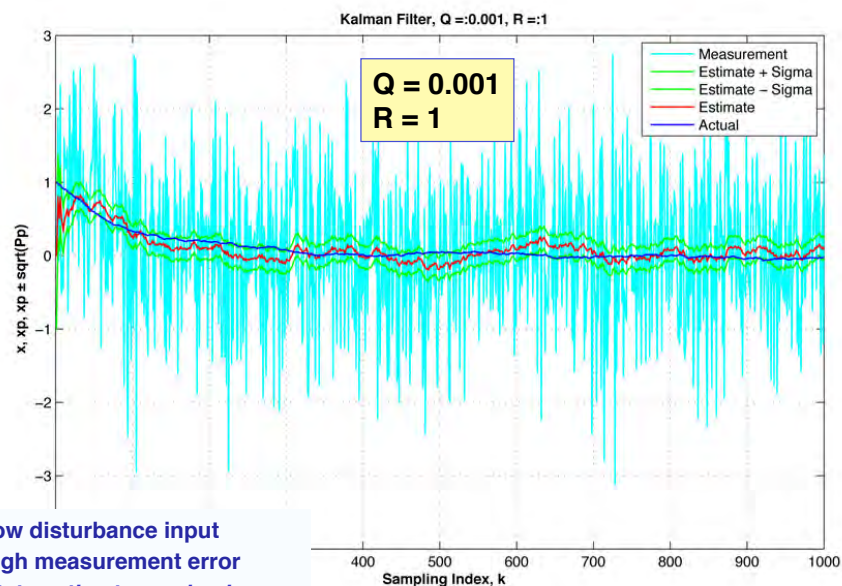
Example: System Response to Small Disturbance



Large estimate of initial uncertainty

17

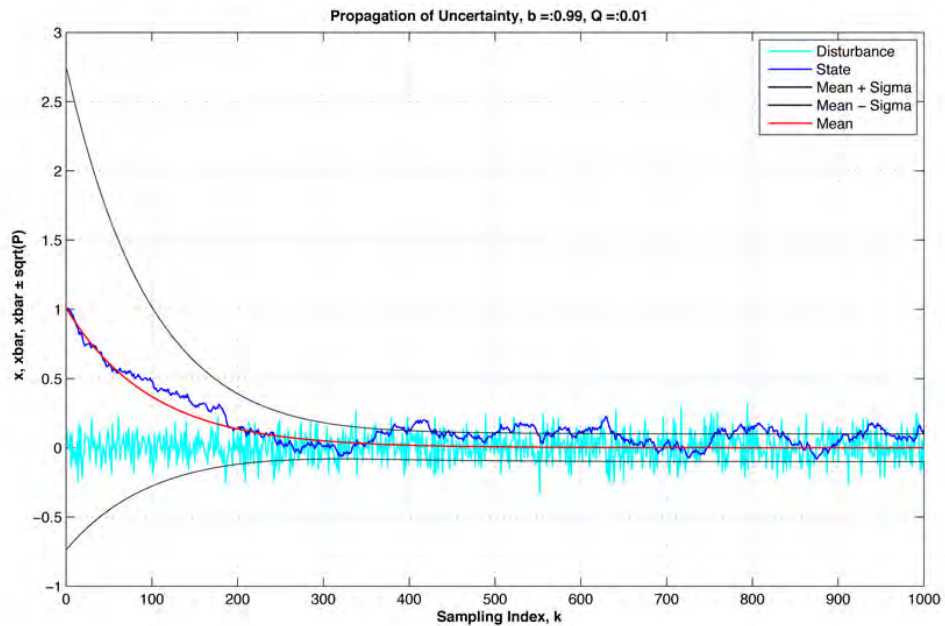
Example of Kalman Filter Estimation



- Low disturbance input
- High measurement error
- *State estimate emphasizes system model*

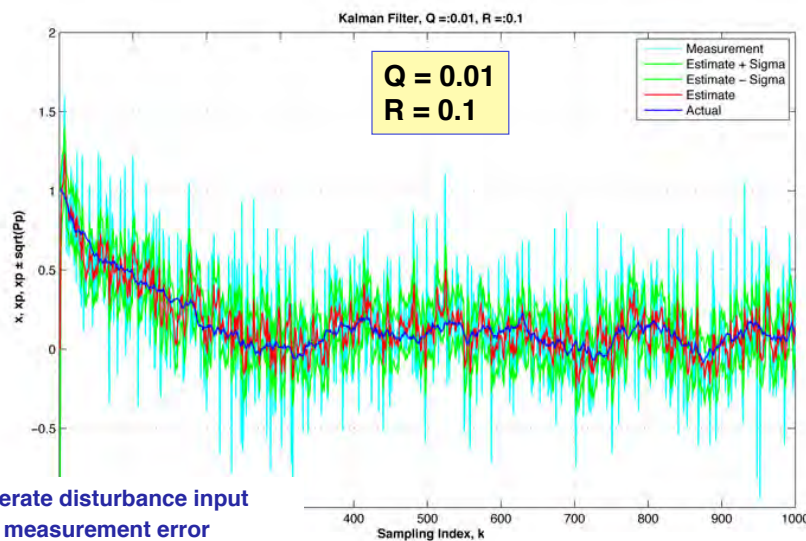
18

Example: System Response to Larger Disturbance



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Example of Kalman Filter Estimation

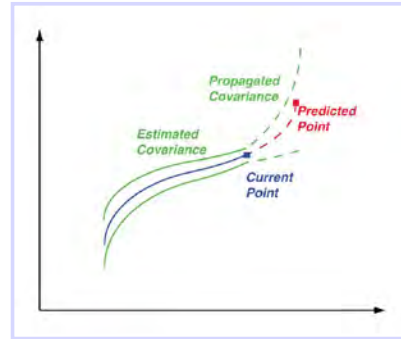


- Moderate disturbance input
- Low measurement error
- *State estimate emphasizes measurement*

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Linear-Optimal Predictor

t_k : Current time, sec
 t_K : Future time, sec



- State estimate extrapolation (or propagation) from last estimate

$$\hat{\mathbf{x}}_K = \Phi(t_K - t_k) \hat{\mathbf{x}}_k(+) + \Gamma(t_K - t_k) \mathbf{u}_k$$

- Covariance estimate extrapolation (or propagation)

$$\mathbf{P}_K = \Phi(t_K - t_k) \mathbf{P}_k(+) \Phi^T(t_K - t_k) + \mathbf{Q}_k(t_K - t_k)$$

- Predictor analogous to Kalman filter without measurement

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Alternative Forms of the Kalman Filter

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Simplifying the Gain Calculation

Filter gain computation for r measurements

$$\mathbf{K}_k = \mathbf{P}_k(-) \mathbf{H}_k^T \left[\mathbf{H}_k \mathbf{P}_k(-) \mathbf{H}_k^T + \mathbf{R}_k \right]^{-1}$$

$r \times r$ inversion required

For $r = 1$

$$\mathbf{K}_k = \frac{\mathbf{P}_k(-) \mathbf{H}_k^T}{\mathbf{H}_k \mathbf{P}_k(-) \mathbf{H}_k^T + r_k}$$

scalar division

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Consider One Measurement at Each Sampling Interval

- With r measurements and diagonal \mathbf{R} , consider just one measurement at a time

$$\mathbf{z} = \begin{bmatrix} z_1 \\ z_2 \\ \dots \\ z_r \end{bmatrix}; \quad \mathbf{R} = \begin{bmatrix} r_{11} & 0 & \dots & 0 \\ 0 & r_{22} & \dots & 0 \\ \dots & \dots & \dots & \dots \\ 0 & 0 & \dots & r_{rr} \end{bmatrix}; \quad \mathbf{H} = \begin{bmatrix} \mathbf{H}_1 \\ \mathbf{H}_2 \\ \dots \\ \mathbf{H}_r \end{bmatrix}$$

$$\begin{aligned} \mathbf{z}_{k+1} &= z_1 \\ \mathbf{z}_{k+2} &= z_2 \\ &\dots \\ \mathbf{z}_{k+r} &= z_r \end{aligned}$$

Scalar Update

$$\begin{aligned} \hat{\mathbf{x}}_k(-) &= \Phi_{k-1} \hat{\mathbf{x}}_{k-1}(+) + \Gamma_{k-1} \mathbf{u}_{k-1} \\ \hat{\mathbf{x}}_k(+) &= \hat{\mathbf{x}}_k(-) + \mathbf{K}_{i_k} [\mathbf{z}_{i_k} - \mathbf{H}_{i_k} \hat{\mathbf{x}}_k(-)] \end{aligned}$$

$$\mathbf{K}_{i_k} = \frac{\mathbf{P}_k(-) \mathbf{H}_{i_k}^T}{\mathbf{H}_{i_k} \mathbf{P}_k(-) \mathbf{H}_{i_k}^T + r_{i_k}}$$

- Cycle through all measurements varying \mathbf{H} , and repeat cycle
- Wasteful, as it does not use all available information

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Sequential Processing of Measurements at Each Sampling Interval

- With r measurements, form an inner loop of calculations, processing one sample at a time

for $i = 1, r$

$$\mathbf{K}_{i_k} = \mathbf{P}_{i-1_k} (+) \mathbf{H}_{i_k}^T / [\mathbf{H}_{i_k} \mathbf{P}_{i-1_k} (+) \mathbf{H}_{i_k}^T + r_{i_k}]$$

$$\mathbf{P}_{i_k} (+) = (\mathbf{I}_n - \mathbf{K}_{i_k} \mathbf{H}_{i_k}) \mathbf{P}_{i-1_k} (+), \quad \mathbf{P}_{0_k} (+) = \mathbf{P}_k (-)$$

$$\hat{\mathbf{x}}_{i_k} (+) = \hat{\mathbf{x}}_{i-1_k} (+) + \mathbf{K}_{i_k} [\mathbf{z}_{i_k} - \mathbf{H}_{i_k} \hat{\mathbf{x}}_{i-1_k} (+)]$$

end

$$\hat{\mathbf{x}}_k (+) = \hat{\mathbf{x}}_{r_k} (+)$$

$$\mathbf{P}_k (+) = \mathbf{P}_{r_k} (+)$$

$$\mathbf{z}_{1_k} = \mathbf{z}_1$$

$$\mathbf{z}_{2_k} = \mathbf{z}_2$$

...

$$\mathbf{z}_{r_k} = \mathbf{z}_r$$

$$\mathbf{H}_{1_k} = \mathbf{H}_1$$

$$\mathbf{H}_{2_k} = \mathbf{H}_2$$

...

$$\mathbf{H}_{r_k} = \mathbf{H}_{r_k}$$

Joseph Form of the Filter (“Stabilized Kalman Filter”)

Guaranteed to retain positive-definiteness and symmetry

State update is

$$\hat{\mathbf{x}}_k (+) = [\mathbf{I}_n - \mathbf{K}_k \mathbf{H}_k] \hat{\mathbf{x}}_k (-) + \mathbf{K}_k \mathbf{z}_k$$

Pre- and post-update measurement errors

$$\boldsymbol{\varepsilon}_k (-) = \mathbf{x}_k - \hat{\mathbf{x}}_k (-); \quad \boldsymbol{\varepsilon}_k (+) = \mathbf{x}_k - \hat{\mathbf{x}}_k (+)$$

Measurement error is updated by

$$\boldsymbol{\varepsilon}_k (+) = [\mathbf{I}_n - \mathbf{K}_k \mathbf{H}_k] \boldsymbol{\varepsilon}_k (-) + \mathbf{K}_k \mathbf{n}_k$$

Joseph Form of the Filter ("Stabilized Kalman Filter")

$$\boldsymbol{\varepsilon}_k(+)=\left[\mathbf{I}_n-\mathbf{K}_k\mathbf{H}_k\right]\boldsymbol{\varepsilon}_k(-)+\mathbf{K}_k\mathbf{n}_k$$

Definitions

$$E\left(\boldsymbol{\varepsilon}_k\boldsymbol{\varepsilon}_k^T\right)=\mathbf{P}_k;\quad E\left(\mathbf{n}_k\mathbf{n}_k^T\right)=\mathbf{R}_k;\quad E\left(\boldsymbol{\varepsilon}_k\mathbf{n}_k^T\right)=\mathbf{0}$$

Then, covariance update is the outer product of the expected error

$$\mathbf{P}_k(+)=\left[\mathbf{I}_n-\mathbf{K}_k\mathbf{H}_k\right]\mathbf{P}_k(-)\left[\mathbf{I}_n-\mathbf{K}_k\mathbf{H}_k\right]^T+\mathbf{K}_k\mathbf{R}_k\mathbf{K}_k^T$$

- Update equation is symmetric
- Equation updates covariance whether or not **K** is optimal
 - Evaluate error covariance of a sub-optimal filter
 - Design and evaluate a low-order filter for a high-order system
- Does not require an ($n \times n$) inversion

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Information Matrix Form of the Kalman Filter

- Filter based on the inverse of **P**

P: State error covariance (small is good)

J = **P**⁻¹: Information matrix (large is good)

- Information filter equations replace large inversions by small inversions
- See *OCE*

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M Conditioning of the Filter Computation

- Calculations may be inaccurate if system contains
 - very fast and slow modes
 - very noisy and near-perfect measurements
 - very large and small disturbance inputs
- Condition number of a matrix, \mathbf{P} , is the ratio of singular values

$$k(\mathbf{P}) = \left[\frac{\lambda_{\max}(\mathbf{P}^T \mathbf{P})}{\lambda_{\min}(\mathbf{P}^T \mathbf{P})} \right]^{1/2} = \frac{\sigma_{\max}(\mathbf{P})}{\sigma_{\min}(\mathbf{P})} = \frac{\bar{\sigma}(\mathbf{P})}{\underline{\sigma}(\mathbf{P})}$$

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Solutions to *M* Conditioning

- Double, triple, ..., precision arithmetic, or
- Formulate the equations to solve for the square root of \mathbf{P}
- Define

$$\mathbf{P} \triangleq \mathbf{S}\mathbf{S}^T \text{ or } \mathbf{S}^T \mathbf{S}$$

then $k(\mathbf{P}) = k(\mathbf{S}\mathbf{S}^T) = k^2(\mathbf{S})$, which is order 10^x

$$k(\mathbf{S}) = \sqrt{k(\mathbf{P})}, \text{ which is order } 10^{x/2}$$

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“U-D” Formulation of the Kalman Filter

Factorization of P

$$\mathbf{P} = \mathbf{U}\mathbf{D}\mathbf{U}^T \sim (\mathbf{U}\mathbf{D}^{1/2})(\mathbf{U}\mathbf{D}^{1/2})^T \sim \mathbf{S}\mathbf{S}^T$$

where

\mathbf{U} : Unit upper triangular matrix:
$$\begin{bmatrix} 1 & \cdot & \cdot & \cdot \\ 0 & 1 & \cdot & \cdot \\ 0 & 0 & 1 & \cdot \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

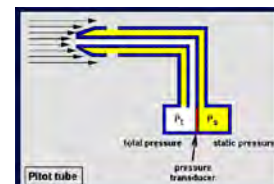
\mathbf{D} : Diagonal matrix:
$$\begin{bmatrix} \cdot & 0 & 0 & 0 \\ 0 & \cdot & 0 & 0 \\ 0 & 0 & \cdot & 0 \\ 0 & 0 & 0 & \cdot \end{bmatrix}$$

- No square roots in the factorization, but formulation has square-root conditioning
- Covariance update uses sequential processing
- Algorithm originally expressed in pseudo-code (Bierman and Thornton, 1977)
- See OCE (pp. 357-360) for equations and pseudo-code

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Kalman Filters for More Complex Systems*

- **Correlated Disturbance Input and Measurement Error**
 - Measurements may be corrupted by the same processes that force the system (e.g., turbulence)
 - Consider estimation-error dynamics, as in Joseph form
 - Stepwise minimization of estimation cost function w.r.t. filter gain matrix
 - See OCE
- **Time-Correlated (“Colored”) Measurement Error**
 - Augment system with measurement error dynamics
 - Use measurement differencing
 - Two-step state and covariance estimates
 - See OCE



* Arthur Bryson and students

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*Next Time:
Kalman-Bucy Filters for
Continuous-Time Systems*

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***Supplemental
Material***

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Descriptions of Random Variables

$$E(\mathbf{x}_0) = \hat{\mathbf{x}}_o; \quad E[(\mathbf{x}_0 - \hat{\mathbf{x}}_o)(\mathbf{x}_0 - \hat{\mathbf{x}}_o)^T] = \mathbf{P}_0$$

$$E(\mathbf{w}_k) = \mathbf{0}; \quad E(\mathbf{w}_j \mathbf{w}_k^T) = \mathbf{Q}'_k \delta_{jk}$$

$$E[\mathbf{u}_k] = \bar{\mathbf{u}}_k; \quad E\{[\mathbf{u}_k - \bar{\mathbf{u}}_k][\mathbf{u}_k - \bar{\mathbf{u}}_k]^T\} = \mathbf{0}$$

$$E(\mathbf{n}_k) = \mathbf{0}; \quad E(\mathbf{n}_j \mathbf{n}_k^T) = \mathbf{R}_k \delta_{jk}$$

$$E(\mathbf{w}_j \mathbf{n}_k^T) = \mathbf{0}$$

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Program for Example of Kalman Filter Estimate Error Stability

```
% Kalman-Bucy Filter Estimate Stability
% Copyright by Robert Stengel. All rights reserved.
% 4/11/2010

clear

% First-Order Stable and Unstable Systems

StableSys = ss(-1,1,1,0);
UnstableSys = ss(1,1,1,0);

[Kstable,Lstable,Pstable] = kalman(StableSys,1,1,0);
[Kunstable,Lunstable,Punstable] = kalman(UnstableSys,1,1,0);

StableSysEst = ss((-1-Lstable),1,1,0);
UnstableSysEst = ss((1-Lunstable),1,1,0);

% Response

t = [0:0.01:6];
[y1,t1,x1] = initial(StableSys,1,t);
[y2,t2,x2] = initial(UnstableSys,1,t);
[y3,t3,x3] = initial(StableSysEst,0.5,t);
[y4,t4,x4] = initial(UnstableSysEst,-1,t);

figure
y3 = interp1(t3,y3,t1);
y4 = interp1(t4,y4,t2);
plot(t1,y1,t2,y2,t1,(y1-y3),t2,(y2-y4)),grid, axis([0 6 0 4]) ax
```

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Program for Covariance Propagation and Kalman Filter Estimation

```
% Scalar Propagation of Mean and Variance
% Kalman Filter
% Copyright by Robert Stengel. All rights reserved.
% 4/12/2011

clear
'=====
'Propagation of Uncertainty'
date

b      = 0.99
b2     = b*b
bb     = 1 - b^2
sqrbb  = sqrt(bb)
w      = [];
x      = [];
xbar   = [];

x(1)   = 1
xbar(1) = x(1)
P(1)   = 3
Q      = 0.01
u      = 0

for k = 1:999
    w(k) = sqrt(Q)*randn(1);
    x(k+1) = b*x(k) + sqrbb*u + sqrbb*w(k);
    xbar(k+1) = b*xbar(k) + sqrbb*u;
    P(k+1) = b2*P(k) + bb*Q;
end

k      = [1:1000];
w(1000) = sqrt(Q)*randn(1);

figure
plot(k,w,'c',k,x,'b',k,(xbar + sqrt(P)), 'k',k,(xbar -
sqrt(P)), 'k',k,xbar,'r'),grid
title(['Propagation of Uncertainty, b =', num2str(b), ', Q =', num2str(Q)], xlabel('Sampling Index, k'), ylabel('x, xbar, xbar ± sqrt(P)'))
legend('Disturbance', 'State', 'Mean + Sigma', 'Mean - Sigma', 'Mean')
```

```
% Kalman Filter
'=====
'Kalman Filter'

xm(1) = 0
xp(1) = 0
Pm(1) = 1
Pp(1) = 1
z(1) = xm(1)
Q      = 0
R      = 0.1
u      = 0

for k = 1:999
    n(k) = sqrt(R)*randn(1);
    xm(k+1) = b*xp(k) + sqrbb*u;
    Pm(k+1) = b2*Pp(k) + Q;
    K      = Pm(k+1) / (Pm(k+1) + R);
    z(k+1) = x(k+1) + n(k);
    xp(k+1) = xm(k+1) + K*(z(k+1) - xm(k+1));
    Pp(k+1) = 1 / ((1/Pm(k+1)) + 1/R);
end

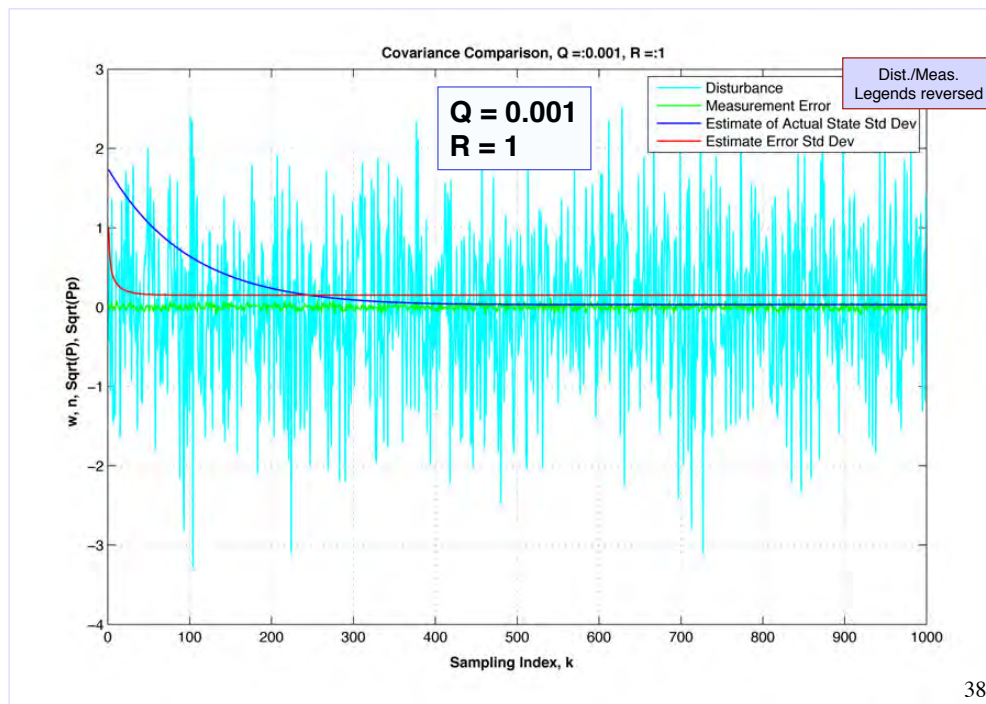
k      = [1:1000];
n(1000) = sqrt(R)*randn(1);
figure
plot(k,z,'c',k,(xp + sqrt(Pp)), 'g',k,(xp -
sqrt(Pp)), 'g',k,xp,'r',k,x,'b'),grid
title(['Kalman Filter, Q =', num2str(Q), ', R =', num2str(R)]),
xlabel('Sampling Index, k'), ylabel('x, xp, xp ± sqrt(Pp)')
legend('Measurement', 'Estimate + Sigma', 'Estimate - Sigma', 'Estimate', 'Actual')
```

```
% Covariance Comparison

figure
SP = sqrt(P);
SPp = sqrt(Pp);
plot(k,n,'c',k,w,'g',k,SP,'b',k,SPp,'r'), grid
title(['Covariance Comparison, Q =', num2str(Q), ', R =', num2str(R)]), xlabel('Sampling Index, k'), ylabel('w, n, Sqrt(P), Sqrt(Pp)')
legend('Measurement Error', 'Disturbance', 'Estimate of Actual State Std Dev', 'Estimate Error Std Dev')
```

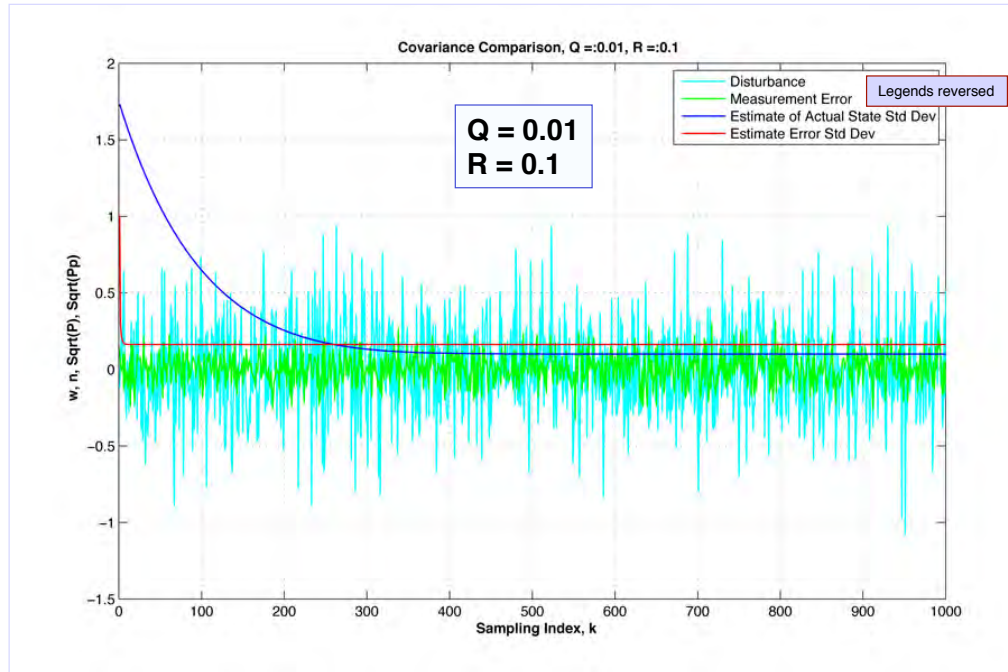
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Example of Kalman Filter Estimation



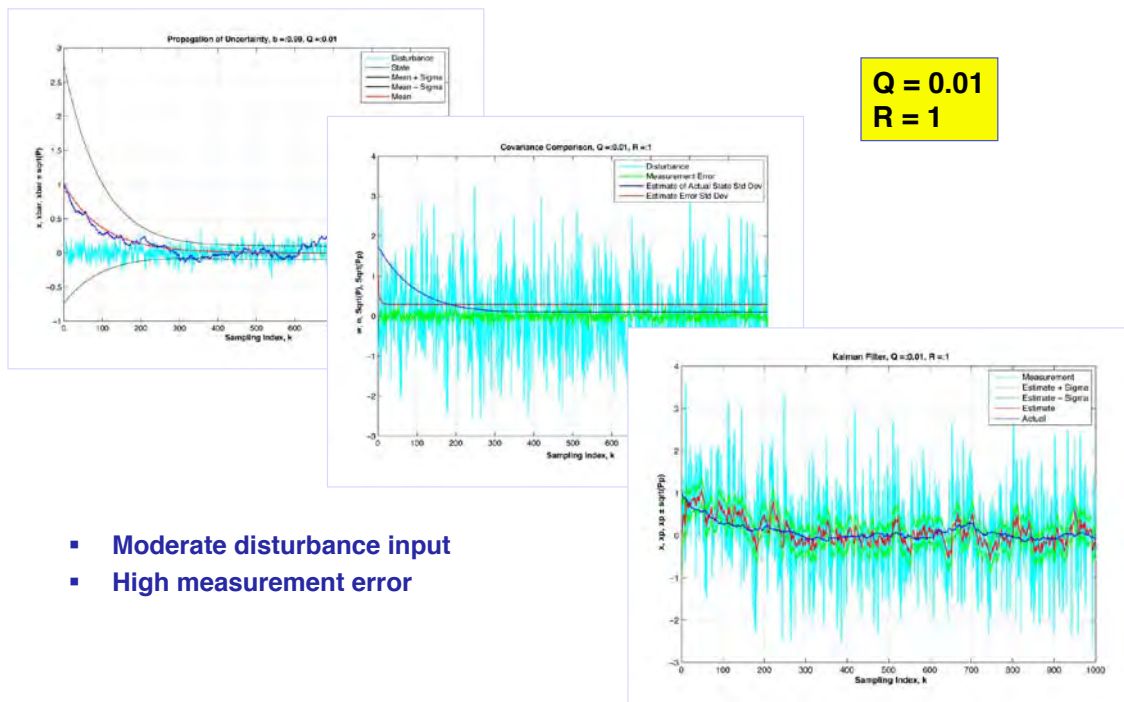
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Example of Kalman Filter Estimation



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Example of Kalman Filter Estimation



- Moderate disturbance input
- High measurement error

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