

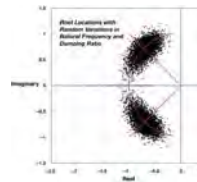
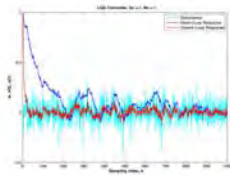
# Stochastic Control

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Robotics and Intelligent Systems, MAE 345, Princeton University, 2015

## Learning Objectives

- Overview of the Linear-Quadratic-Gaussian (LQG) Regulator
- Introduction to Stochastic Robust Control Laws



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<http://www.princeton.edu/~stengel/MAE345.html>

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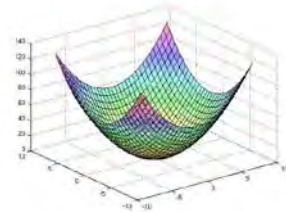
## *Stochastic Optimal Control*

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# Deterministic vs. Stochastic Optimal Control

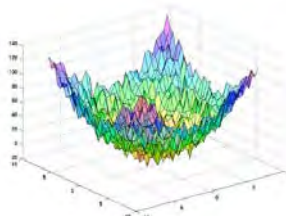
## • Deterministic control

- Known dynamic process
  - precise input
  - precise initial condition
  - precise measurement
- Optimal control minimizes  $J^* = J(x^*, u^*)$



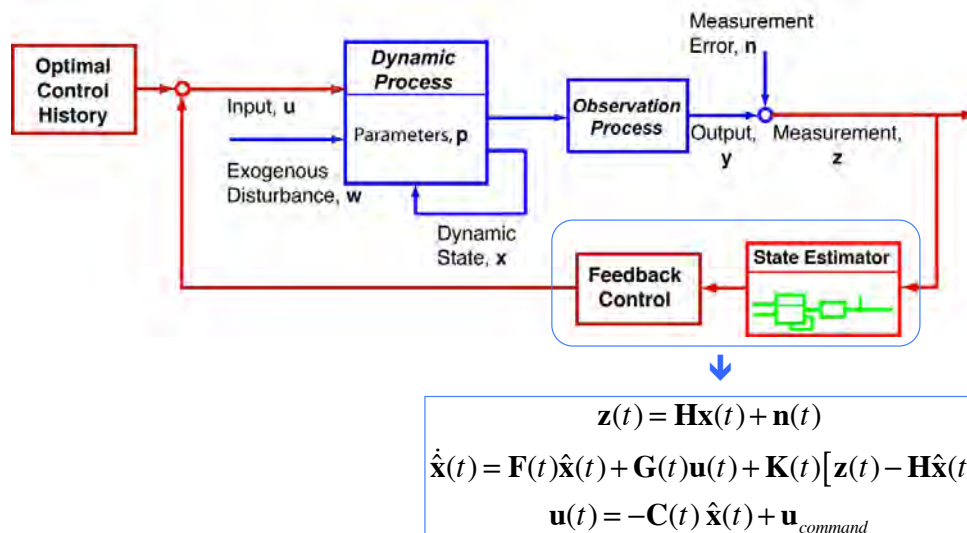
## • Stochastic control

- Known dynamic process
  - unknown input
  - imprecise initial condition
  - imprecise or incomplete measurement
- Optimal control minimizes  $E\{J[x^*, u^*]\}$



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# Linear-Quadratic-Gaussian (LQG) Control of a Dynamic Process



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# Linear-Quadratic (**LQ**) Control Equations (Continuous-Time Model)

Open-Loop System State Dynamics

$$\dot{\mathbf{x}}(t) = \mathbf{F}\mathbf{x}(t) + \mathbf{G}\mathbf{u}(t)$$

Control Law

$$\mathbf{u}(t) = -\mathbf{C}\mathbf{x}(t) + \mathbf{u}_{command}$$

Closed-Loop System State Dynamics

$$\dot{\mathbf{x}}(t) = (\mathbf{F} - \mathbf{G}\mathbf{C})\mathbf{x}(t) + \mathbf{G}\mathbf{C}\mathbf{u}_{command}$$

Characteristic Equation

$$|s\mathbf{I} - (\mathbf{F} - \mathbf{G}\mathbf{C})| = 0$$

How many eigenvalues?

*n*

Stable or unstable?

**Stable, with correct  
design criteria, **F**, and **G****

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# Linear-Quadratic-Gaussian (**LQG**) Control Equations (Continuous-Time Model)

Open-Loop System State Dynamics and Measurement

$$\dot{\mathbf{x}}(t) = \mathbf{F}\mathbf{x}(t) + \mathbf{G}\mathbf{u}(t) + \mathbf{L}\mathbf{w}(t)$$

$$\mathbf{z}(t) = \mathbf{H}\mathbf{x}(t) + \mathbf{n}(t)$$

State Estimate

$$\dot{\hat{\mathbf{x}}}(t) = \mathbf{F}\hat{\mathbf{x}}(t) + \mathbf{G}\mathbf{u}(t) + \mathbf{K}[\mathbf{z}(t) - \mathbf{H}\hat{\mathbf{x}}(t)]$$

Control Law

$$\mathbf{u}(t) = -\mathbf{C}\hat{\mathbf{x}}(t) + \mathbf{u}_{command}$$

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# Linear-Quadratic-Gaussian (LQG) Control Equations (Continuous-Time Model)

Closed-Loop System State and Estimate Dynamics  
(neglect command)

$$\dot{\mathbf{x}}(t) = \mathbf{F}\mathbf{x}(t) + \mathbf{G}[-\mathbf{C}\hat{\mathbf{x}}(t)] + \mathbf{L}\mathbf{w}(t)$$

$$\dot{\hat{\mathbf{x}}}(t) = \mathbf{F}\hat{\mathbf{x}}(t) + \mathbf{G}[-\mathbf{C}\hat{\mathbf{x}}(t)] + \mathbf{K}[\mathbf{z}(t) - \mathbf{H}\hat{\mathbf{x}}(t)]$$

How many eigenvalues? **2n**

Stable or unstable? **TBD**

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## LQG Separation Property

Optimal estimation algorithm does not depend on the  
optimal control algorithm

$$\begin{aligned}\mathbf{K}(t) &= \mathbf{P}(t)\mathbf{H}^T\mathbf{N}^{-1}(t) \\ \dot{\mathbf{P}}(t) &= \mathbf{F}(t)\mathbf{P}(t) + \mathbf{P}(t)\mathbf{F}^T(t) + \mathbf{L}(t)\mathbf{W}(t)\mathbf{L}^T(t) - \mathbf{P}(t)\mathbf{H}^T\mathbf{N}^{-1}(t)\mathbf{H}\mathbf{P}(t)\end{aligned}$$

Optimal control algorithm does not depend on the  
optimal estimation algorithm

$$\begin{aligned}\mathbf{C}(t) &= \mathbf{R}^{-1}(t)\mathbf{G}^T(t)\mathbf{S}(t) \\ \dot{\mathbf{S}}(t) &= -\mathbf{Q}(t) - \mathbf{F}(t)^T\mathbf{S}(t) - \mathbf{S}(t)\mathbf{F}(t) + \mathbf{S}(t)\mathbf{G}(t)\mathbf{R}^{-1}(t)\mathbf{G}^T(t)\mathbf{S}(t)\end{aligned}$$

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# LQG Certainty Equivalence

Stochastic feedback control is computed  
from optimal estimate of the state

$$\mathbf{u}^*(t) = -\mathbf{R}^{-1}\mathbf{G}^T(t)\mathbf{S}(t)\hat{\mathbf{x}}(t) = -\mathbf{C}(t)\hat{\mathbf{x}}(t)$$

Stochastic feedback control law is the **same**  
**as** the deterministic control law

$$\mathbf{u}^*(t) = -\mathbf{R}^{-1}\mathbf{G}^T(t)\mathbf{S}(t)\mathbf{x}(t) = -\mathbf{C}(t)\mathbf{x}(t)$$

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*Asymptotic Stability of the  
LQG Regulator  
(with no parameter uncertainty)*

# System Equations with LQG Control

With perfect knowledge of the system

$$\dot{\mathbf{x}}(t) = \mathbf{F}\mathbf{x}(t) + \mathbf{G}[-\mathbf{C}\hat{\mathbf{x}}(t)] + \mathbf{L}\mathbf{w}(t)$$

$$\dot{\hat{\mathbf{x}}}(t) = \mathbf{F}\hat{\mathbf{x}}(t) + \mathbf{G}[-\mathbf{C}\hat{\mathbf{x}}(t)] + \mathbf{K}[\mathbf{z}(t) - \mathbf{H}\hat{\mathbf{x}}(t)]$$

**State estimate error**

$$\boldsymbol{\varepsilon}(t) \triangleq \mathbf{x}(t) - \hat{\mathbf{x}}(t)$$

**State estimate error dynamics**

$$\dot{\boldsymbol{\varepsilon}}(t) = (\mathbf{F} - \mathbf{K}\mathbf{H})\boldsymbol{\varepsilon}(t) + \mathbf{L}\mathbf{w}(t) - \mathbf{K}\mathbf{n}(t)$$

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## Control-Loop and Estimator Eigenvalues are Uncoupled

$$\begin{bmatrix} \dot{\mathbf{x}}(t) \\ \dot{\boldsymbol{\varepsilon}}(t) \end{bmatrix} = \begin{bmatrix} (\mathbf{F} - \mathbf{G}\mathbf{C}) & \mathbf{G}\mathbf{C} \\ \mathbf{0} & (\mathbf{F} - \mathbf{K}\mathbf{H}) \end{bmatrix} \begin{bmatrix} \mathbf{x}(t) \\ \boldsymbol{\varepsilon}(t) \end{bmatrix} + \begin{bmatrix} \mathbf{L} & \mathbf{0} \\ \mathbf{L} & -\mathbf{K} \end{bmatrix} \begin{bmatrix} \mathbf{w}(t) \\ \mathbf{n}(t) \end{bmatrix}$$

**Upper-block-triangular stability matrix**

**LQG system is stable because**

$(\mathbf{F} - \mathbf{G}\mathbf{C})$  is stable

$(\mathbf{F} - \mathbf{K}\mathbf{H})$  is stable

**Estimate error affects state response**

$$\dot{\mathbf{x}}(t) = (\mathbf{F} - \mathbf{G}\mathbf{C})\mathbf{x}(t) + \mathbf{G}\mathbf{C}\boldsymbol{\varepsilon}(t) + \mathbf{L}\mathbf{w}(t)$$

**Actual state does not affect error response**

**Disturbance affects both equally**

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# Discrete-Time LQG Controller

Kalman filter produces state estimate

$$\hat{\mathbf{x}}_k(-) = \Phi \hat{\mathbf{x}}_{k-1}(+) - \Gamma \mathbf{C}_{k-1} \hat{\mathbf{x}}_{k-1}(+)$$

$$\hat{\mathbf{x}}_k(+) = \hat{\mathbf{x}}_k(-) + \mathbf{K}_k [\mathbf{z}_k - \mathbf{H} \hat{\mathbf{x}}_k(-)]$$

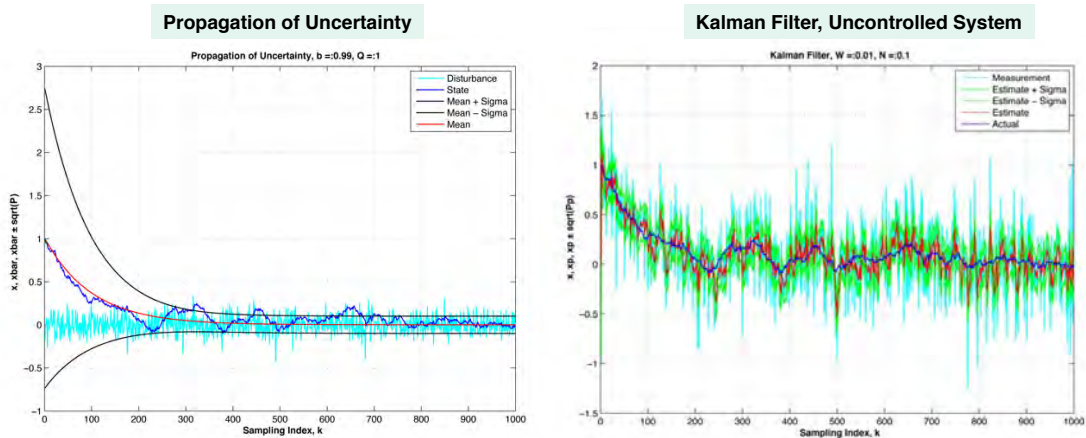
Closed-loop system uses state estimate for feedback control ( $\mathbf{u}_{command} = 0$ )

$$\mathbf{u}_k = -\mathbf{C}_k \hat{\mathbf{x}}_k(+)$$

$$\mathbf{x}_{k+1}(-) = \Phi \mathbf{x}_k(-) - \Gamma \mathbf{C}_k \hat{\mathbf{x}}_k(+)$$

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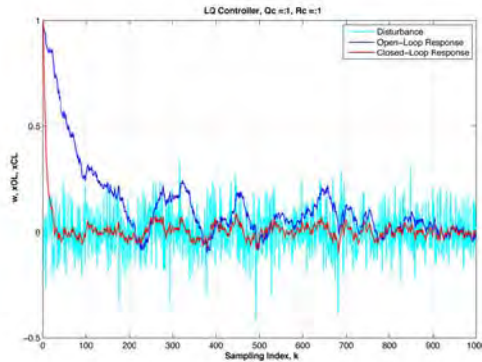
## Response of Discrete-Time 1<sup>st</sup>-Order System to Disturbance, and Kalman Filter Estimate from Noisy Measurement



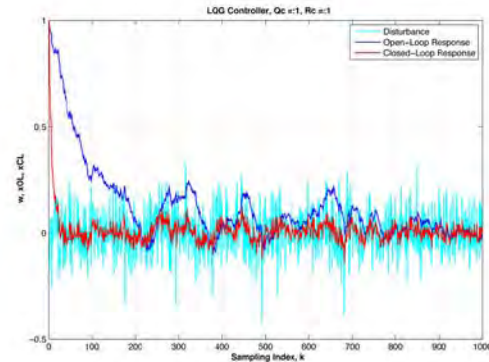
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# Comparison of 1<sup>st</sup>-Order Discrete-Time LQ and LQG Control Response

Linear-Quadratic Control with Noise-free Measurement

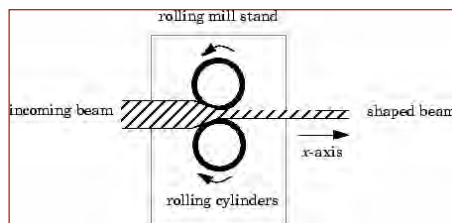


Linear-Quadratic-Gaussian Control with Noisy Measurement



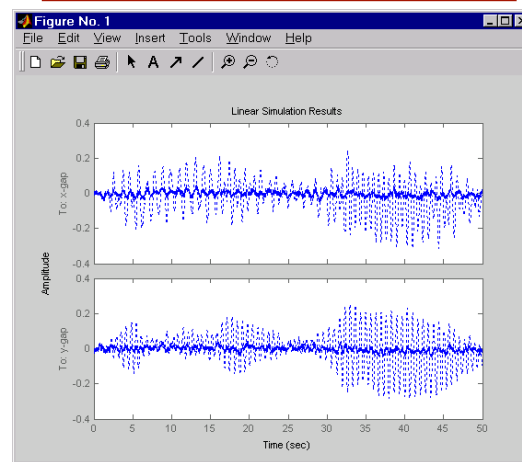
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## MATLAB Demo: LQG Rolling Mill Control System Design Example



- **Maintain desired thickness of shaped beam**
- **Account for random**
  - variations in thickness/hardness of incoming beam
  - eccentricity in rolling cylinders
  - measurement errors

Open- and Closed-Loop Response





# *Robust Stochastic Control*

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## Stochastic, Robust, and Adaptive Control

- Stochastic controller
  - minimize response to random initial conditions, disturbances, and measurement errors
  - perfect knowledge of the plant
- **Robust controller**
  - fixed gains and structure
  - minimize likelihood of instability or unsatisfactory performance due to parameter uncertainty in the plant
- Adaptive controller
  - variable gains and/or structure
  - minimize likelihood of instability or unsatisfactory performance due to plant parameter uncertainty, disturbances, and measurement errors

*Practical controller may have elements of all three*

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# Robust Control System Design

- Make closed-loop response insensitive to plant parameter variations
- **Robust controller**
  - Fixed gains and structure
  - Minimize likelihood of instability
  - Minimize likelihood of unsatisfactory performance

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## Probabilistic Robust Control Design



- Design a fixed-parameter controller for **stochastic robustness**
- **Monte Carlo Evaluation** of competing designs
- **Genetic Algorithm or Simulated Annealing** search for best design

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# Representations of Uncertainty

## Characteristic equation of the uncontrolled system

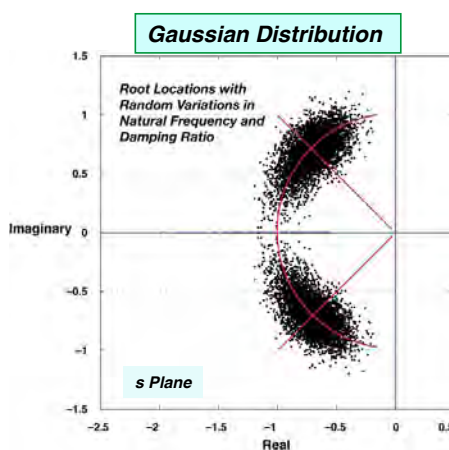
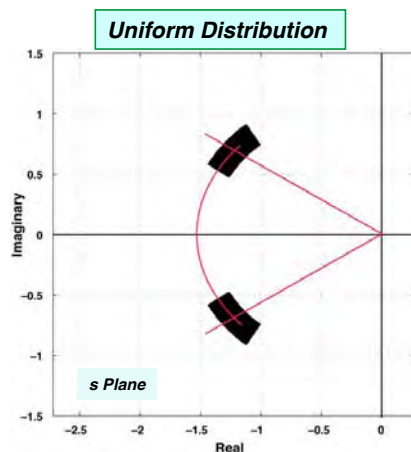
$$\begin{aligned} |s\mathbf{I} - \mathbf{F}| &= \det(s\mathbf{I} - \mathbf{F}) \triangleq \\ \Delta(s) &= s^n + a_{n-1}s^{n-1} + \dots + a_1s + a_0 \\ &= (s - \lambda_1)(s - \lambda_2)(\dots)(s - \lambda_n) = 0 \end{aligned}$$

- Uncertainty can be expressed in
  - Elements of  $\mathbf{F}$
  - Coefficients of  $\Delta(s)$
  - Eigenvalues of  $\mathbf{F}$

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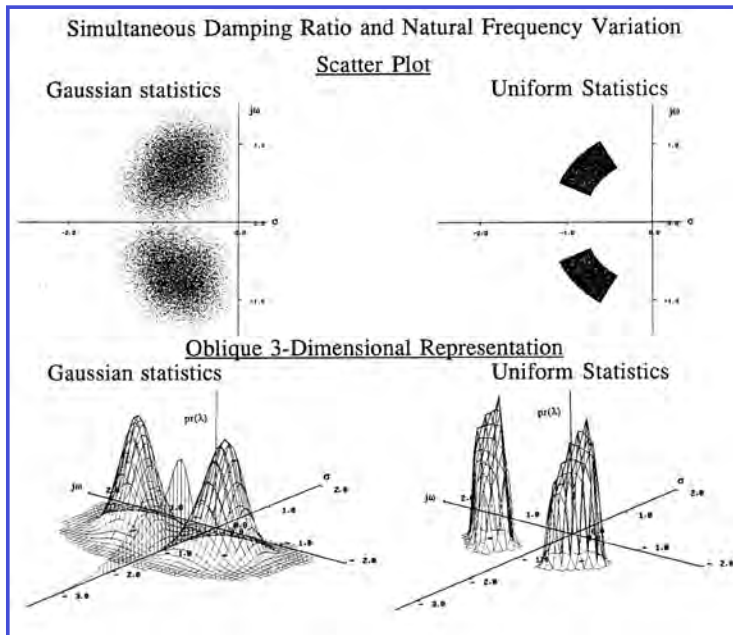
## Root Locations for an Uncertain 2<sup>nd</sup>-Order System

- Variation may be represented by
  - Worst-case, e.g., Upper/lower bounds of uniform distribution
  - Probability, e.g., Gaussian distribution



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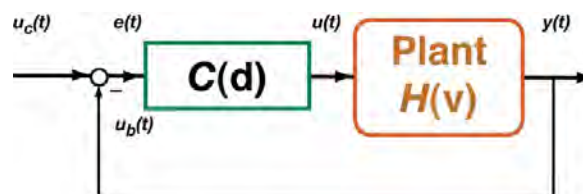
# “3-D” Stochastic Root Loci for 2<sup>nd</sup>-Order Example



- Root distributions are nonlinear functions of parameter distributions
- Unbounded parameter distributions always lead to non-zero probability of instability
- Bounded distributions may be guaranteed to be stable

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## Probability of Satisfying a Design Metric



$$Pr(\mathbf{d}, \mathbf{v}) \approx \frac{1}{N} \sum_{i=1}^N e[C(\mathbf{d}), H(\mathbf{v})]$$

- Probability of satisfying a design metric
  - **d**: Control design parameter vector [e.g., SA, GA, ...]
  - **v**: Uncertain plant parameter vector [e.g., RNG]
  - **e**: Binary indicator, e.g.,  
0: satisfactory 1: unsatisfactory
  - **H(v)**: Plant
  - **C(d)**: Controller (Compensator)

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# Design Control System to Minimize Probability of Instability

- Characteristic equation of the closed-loop system

$$\Delta_{closed-loop}(s) = \left| s\mathbf{I} - [\mathbf{F}(\mathbf{v}) - \mathbf{G}(\mathbf{v})\mathbf{C}(\mathbf{d})] \right|$$

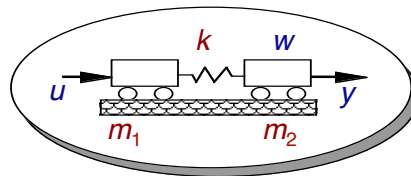
$$= \left[ (s - \lambda_1)(s - \lambda_2)(\dots)(s - \lambda_n) \right]_{closed-loop} = 0$$

- Monte Carlo evaluation of probability of instability with uncertain plant parameters
- Minimize probability of instability using numerical search of control parameters

$$\min_{\mathbf{d}} \left\{ \Pr \left[ \operatorname{Re}(\lambda_i, i = 1, n) > 0 \right] \right\}$$

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## Control Design Example\*



- **Challenge:** Design a feedback compensator for a 4<sup>th</sup>-order spring-mass system (“the plant”) whose parameters are bounded but unknown
  - Minimize the likelihood of instability
  - Satisfy a settling time requirement
  - Don’t use too much control

\* 1990 American Control Conference Robust Control Benchmark Problem

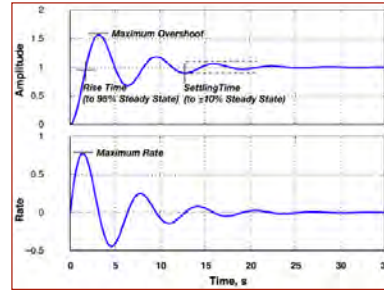
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# Design Cost Function

- Probability of Instability,  $Pr_i$ 
  - $e_i = 1$  (unstable) or 0 (stable)
- Probability of Settling Time Exceedance,  $Pr_{ts}$ 
  - $e_{ts} = 1$  (exceeded) or 0 (not exceeded)
- Probability of Control Limit Exceedance,  $Pr_u$ 
  - $e_u = 1$  (exceeded) or 0 (not exceeded)
- Each metric has a binomial distribution

$$\text{pr}(x) = \frac{n!}{k!(n-k)!} p(x)^k [1-p(x)]^{n-k} \triangleq \binom{n}{k} p(x)^k [1-p(x)]^{n-k}$$

= probability of exactly  $k$  successes in  $n$  trials, in  $(0,1)$   
 $\sim$  normal distribution for large  $n$



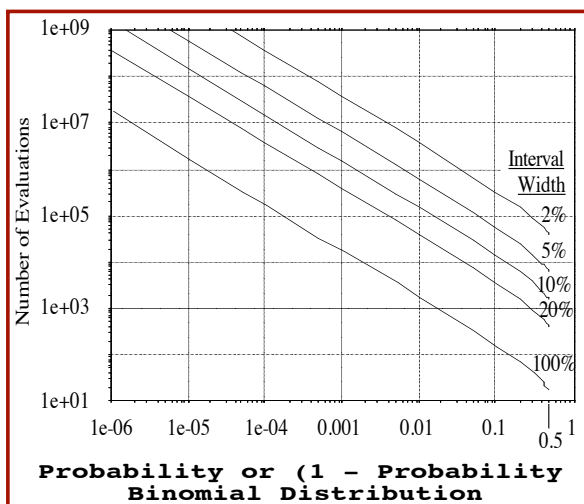
- Design Cost Function
  - High probabilities weighted more than low probabilities
  - $J = aPr_i^2 + bPr_{ts}^2 + cPr_u^2$
  - $a = 1$
  - $b = c = 0.01$

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## Monte Carlo Evaluation of Probability of Satisfying a Design Metric

$$Pr_k(\mathbf{d}, \mathbf{v}) \approx \frac{1}{N} \sum_{i=1}^N e_k [C(\mathbf{d}), H(\mathbf{v})], \quad k = 1, 3$$

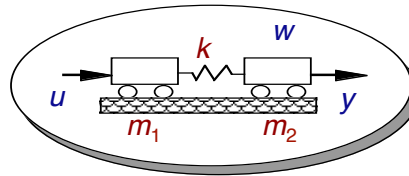
$$J = aPr_i^2(\mathbf{d}, \mathbf{v}) + bPr_{ts}^2(\mathbf{d}, \mathbf{v}) + cPr_u^2(\mathbf{d}, \mathbf{v})$$



- Compute  $\mathbf{v}$  using random number generators over  $N$  trials
  - Required number of trials depends on outcome probability and desired confidence interval
- Search for best  $\mathbf{d}$  using a genetic algorithm to minimize  $J$

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# Uncertain Plant\*



$$y = x_2 + n$$

## Plant dynamic equation

$$\begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \dot{x}_3 \\ \dot{x}_4 \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ -k/m_1 & k/m_1 & 0 & 0 \\ k/m_2 & -k/m_2 & 0 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ x_3 \\ x_4 \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 1/m_1 \\ 0 \end{bmatrix} u + \begin{bmatrix} 0 \\ 0 \\ 0 \\ 1/m_2 \end{bmatrix} w$$

## 4<sup>th</sup>-Order Plant characteristic equation

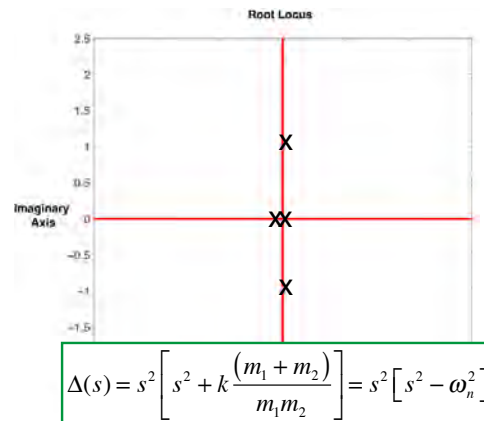
$$\Delta(s) = s^2 \left[ s^2 + k \frac{(m_1 + m_2)}{m_1 m_2} \right] = s^2 [s^2 - \omega_n^2]$$

\* 1990 American Control Conference Robust Control Benchmark Problem

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# Parameter Uncertainties, Root Locus, and Control Law

- Parameters of mass-spring system
  - Uniform probability density functions for
    - $0.5 < m_1, m_2 < 1.5$
    - $0.5 < -k < 2$
- Effects of parameters on root locations (right)



$$\Delta(s) = s^2 \left[ s^2 + k \frac{(m_1 + m_2)}{m_1 m_2} \right] = s^2 [s^2 - \omega_n^2]$$

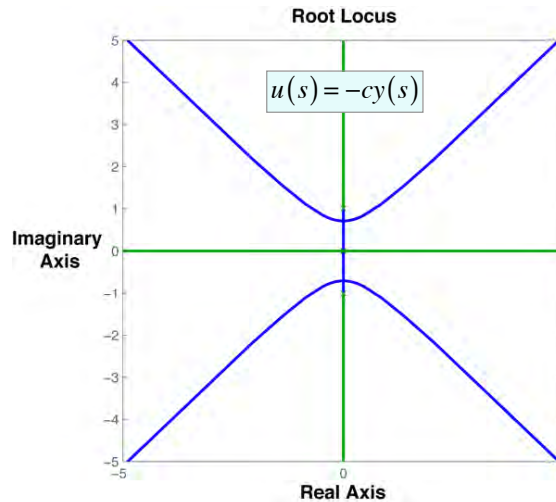
- Single-input/single-output feedback control law

$$u(s) = -C(s)y(s)$$

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# Mass-Spring-Mass Stabilization Requires Compensation

- Proportional feedback alone **cannot** stabilize the system
- Feedback of **either sign** drives at least one root into the right half plane



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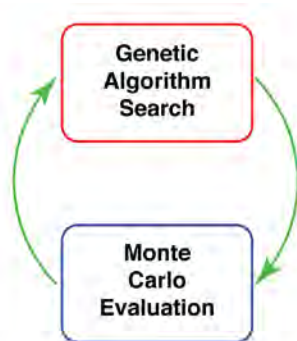
## Search-and-Sweep Design of Family of Robust Feedback Compensators

Begin with lowest-order feedback compensator

$$C_{12}(s) = \frac{a_0 + a_1 s}{b_0 + b_1 s + b_2 s^2} \equiv C(\mathbf{d})$$

Arrange parameters as binary design vector

$$\mathbf{d} = \{a_0, a_1, b_0, b_1, b_2\}$$



$$\mathbf{d}^* = \{a_0^*, a_1^*, b_0^*, b_1^*, b_2^*\}$$

Search for design vector,  $\mathbf{d}$ , that minimizes  $J$

$$\begin{aligned} m_1 &= rand(1) + 0.5 \\ m_2 &= rand(1) + 0.5 \\ k &= -1.5 * rand(1) + 0.5 \end{aligned}$$

Monte Carlo evaluation with uncertain parameters,  $\mathbf{v}$

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# Search-and-Sweep Design of Family of Robust Feedback Compensators

## 1) Define next higher-order compensator

$$C_{22}(s) = \frac{a_0 + a_1s + a_2s^2}{b_0 + b_1s + b_2s^2}$$

## 2) Optimize over all parameters, including optimal coefficients in starting population

$$\mathbf{d} = \{a_0^*, a_1^*, a_2^*, b_0^*, b_1^*, b_2^*\} \Rightarrow \mathbf{d}^{**} = \{a_0^{**}, a_1^{**}, a_2^{**}, b_0^{**}, b_1^{**}, b_2^{**}\}$$

## 3) Sweep to satisfactory design or no further improvement

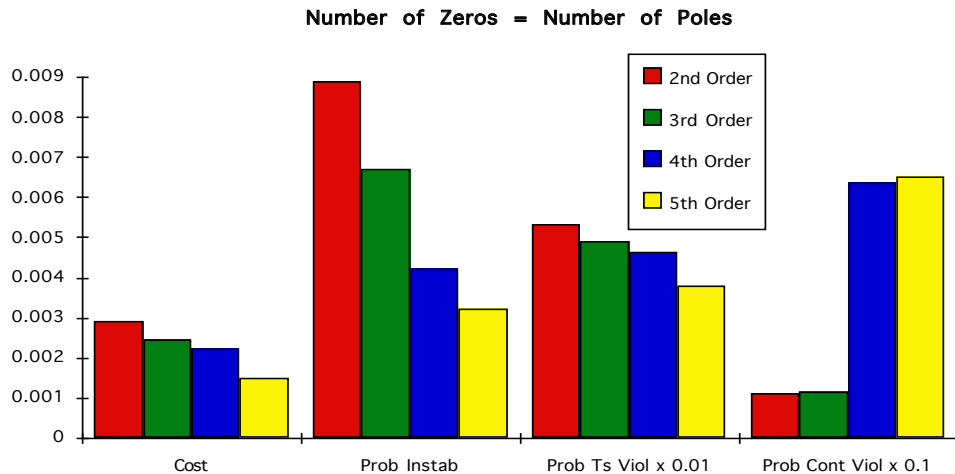
$$C_{23}(s) = \frac{a_0 + a_1s + a_2s^2}{b_0 + b_1s + b_2s^2 + b_3s^3}$$

$$C_{33}(s) = \frac{a_0 + a_1s + a_2s^2 + a_3s^3}{b_0 + b_1s + b_2s^2 + b_3s^3}$$

$$C_{34}(s) = \frac{a_0 + a_1s + a_2s^2 + a_3s^3}{b_0 + b_1s + b_2s^2 + b_3s^3 + b_4s^4} \dots$$

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# Design Cost and Probabilities for Optimal 2<sup>nd</sup>- to 5<sup>th</sup>-Order Compensators



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***Next Time:  
Parameter Estimation and  
Adaptive Control***

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***Supplemental Material***

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# Linear-Quadratic Gaussian Optimal Control Law

- Minimize expected value of cost, subject to uncertainty\*

$$\min_u E(J) \approx E(J^*)$$

- Stochastic optimal feedback control law combines the linear-optimal control law with a linear-optimal state estimate

$$\mathbf{u}^*(t) = -\mathbf{R}^{-1}\mathbf{G}^T(t)\mathbf{S}(t)\hat{\mathbf{x}}(t) = -\mathbf{C}(t)\hat{\mathbf{x}}(t)$$

- where  $\hat{\mathbf{x}}(t)$  is an optimal estimate of the state perturbation

\* See reading for details

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## Example: Probability of Stable Control of an Unstable Plant



- Longitudinal dynamics for a Forward-Swept-Wing Demonstrator

$$\mathbf{F} = \begin{bmatrix} -2gf_{11}/V & \rho V^2 f_{12}/2 & \rho V f_{13} & -g \\ -45/V^2 & \rho V f_{22}/2 & 1 & 0 \\ 0 & \rho V^2 f_{32}/2 & \rho V f_{33} & 0 \\ 0 & 0 & 1 & 0 \end{bmatrix}; \quad \mathbf{G} = \begin{bmatrix} g_{11} & g_{12} \\ 0 & 0 \\ g_{31} & g_{32} \\ 0 & 0 \end{bmatrix}; \quad \mathbf{x} = \begin{bmatrix} V \\ \alpha \\ q \\ \theta \end{bmatrix}$$

- Nominal eigenvalues (one unstable)

$$\lambda_{1-4} = -0.1 \pm 0.057j, \quad -5.15, \quad 3.35$$

Air density and airspeed,  $\rho$  and  $V$ , have uniform distributions ( $\pm 30\%$ )

10 coefficients have Gaussian distributions ( $\sigma = 30\%$ )

$$\mathbf{p} = \begin{bmatrix} \rho & V & f_{11} & f_{12} & f_{13} & f_{22} & f_{32} & f_{33} & g_{11} & g_{12} & g_{31} & g_{32} \end{bmatrix}^T$$

Environment

Uncontrolled Dynamics

Control Effect

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# LQ Regulators for the Example



- Three stabilizing feedback control laws

- **Case a)** LQR with low control weighting

$$\mathbf{Q} = \text{diag}(1,1,1,0); \quad \mathbf{R} = (1,1); \quad \lambda_{1-4, \text{nominal}} = -35, -5.1, -3.3, -.02$$

$$\mathbf{C} = \begin{bmatrix} 0.17 & 130 & 33 & 0.36 \\ 0.98 & -11 & -3 & -1.1 \end{bmatrix}$$

- **Case b)** LQR with high control weighting

$$\mathbf{Q} = \text{diag}(1,1,1,0); \quad \mathbf{R} = (1000,1000); \quad \lambda_{1-4, \text{nominal}} = -5.2, -3.4, -1.1, -.02$$

$$\mathbf{C} = \begin{bmatrix} 0.03 & 83 & 21 & -0.06 \\ 0.01 & -63 & -16 & -1.9 \end{bmatrix}$$

- **Case c)** Case b with gains multiplied by 5 for bandwidth (loop-transfer) recovery

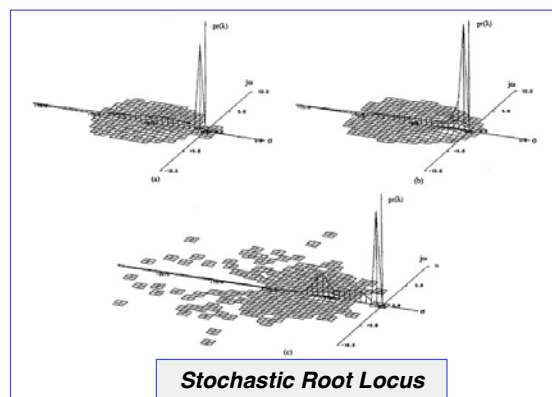
$$\lambda_{1-4, \text{nominal}} = -32, -5.2, -3.4, -0.01$$

$$\mathbf{C} = \begin{bmatrix} 0.13 & 413 & 105 & -0.32 \\ 0.05 & -313 & -81 & -1.1 - 9.5 \end{bmatrix}$$

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## Stochastic Robustness (Ray, Stengel, 1991)

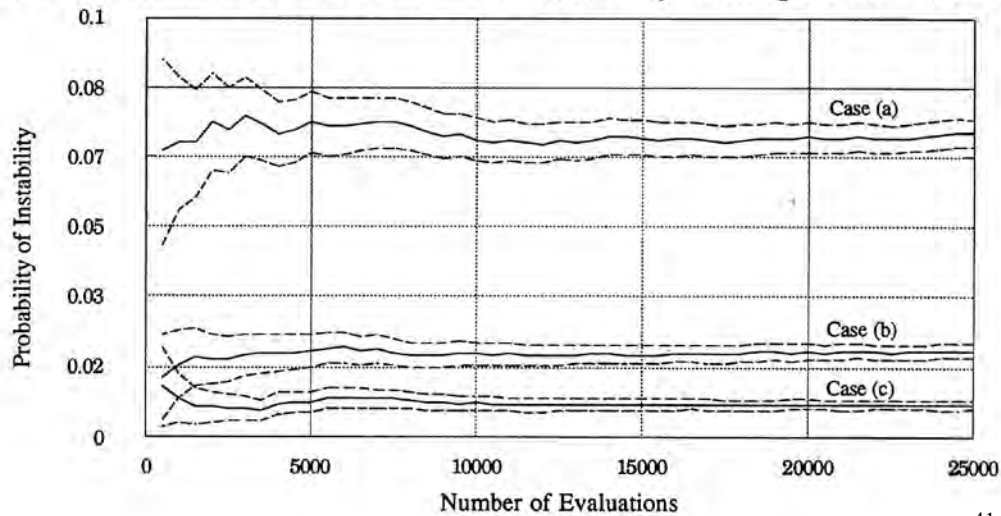
- Distribution of closed-loop roots with
  - Gaussian uncertainty in 10 parameters
  - Uniform uncertainty in velocity and air density
- 25,000 Monte Carlo evaluations
- **Probability of instability**
  - a)  $\text{Pr} = 0.072$
  - b)  $\text{Pr} = 0.021$
  - c)  $\text{Pr} = 0.0076$



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# Probabilities of Instability for the Three Cases

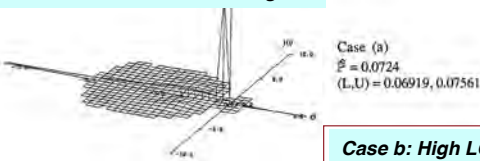
95% CONFIDENCE INTERVALS (with dynamic pressure effects)



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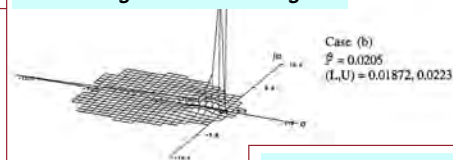
# Stochastic Root Loci for the Three Cases

Case a: Low LQ Control Weights

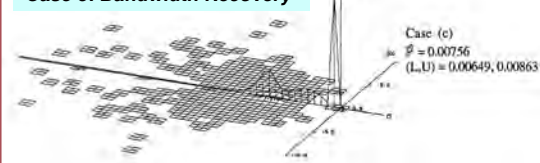


with Gaussian Aerodynamic Uncertainty

Case b: High LQ Control Weights



Case c: Bandwidth Recovery



- Probabilities of instability with 30% uniform aerodynamic uncertainty
  - Case a:  $3.4 \times 10^{-4}$
  - Case b: 0
  - Case c: 0

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