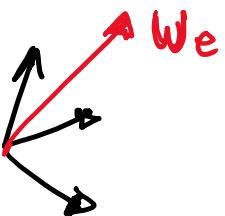


# **H1- Orientation Control in robot manipulators**

- until now we focused on Cartesian control (3D) we want now to incorporate tracking of the orientation of a frame (e.g. attached at the end-effector).

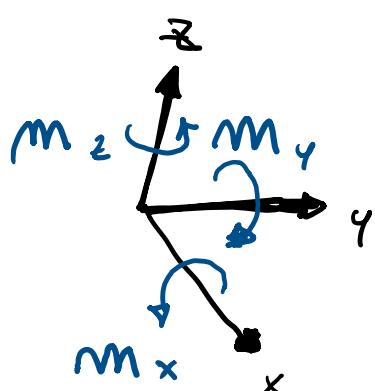
## ORIENTATION JACOBIAN

  $w_e \in \mathbb{R}^3$

End-effector  
FRAME

$$J_o \quad \dot{q} \in \mathbb{R}^m$$

NOTE: This is the angular part of the geometric Jacobian



$$m = [m_x, m_y, m_z] \in \mathbb{R}^3$$

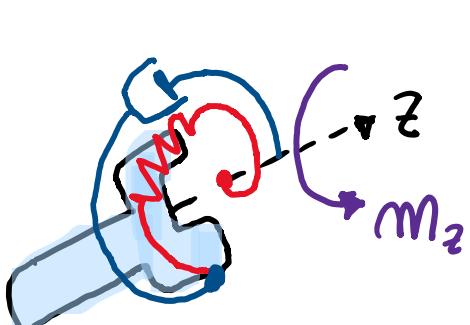
$$J_o^T \quad \dot{\theta} \in \mathbb{R}^m$$

## PD FOR ORIENTATION CONTROL

$$u = J_o^T (K_o e_o + D_o (\omega^d - J_o \dot{q}))$$

orientation error  $\dot{e}_o$  already in euclidean space

- This is equivalent to have torsional springs and dampers in the 3 directions



$$m_z = K_{o,zz} e_{o,z} \in \mathbb{R}$$

$$m = K_o e_o \in \mathbb{R}^3$$

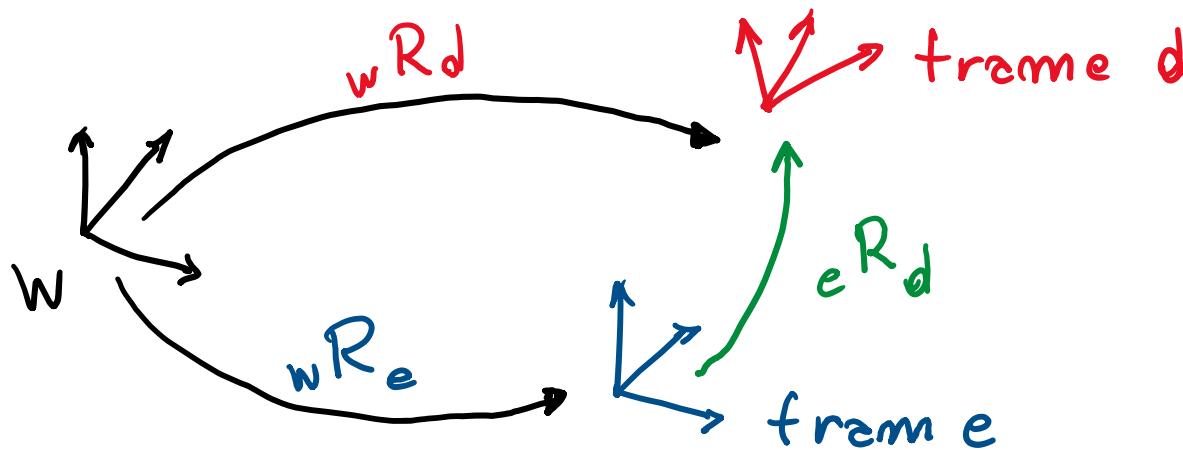
$$\hookrightarrow K_o = \begin{bmatrix} K_{oxx} & 0 & 0 \\ 0 & K_{oyy} & 0 \\ 0 & 0 & K_{ozz} \end{bmatrix} \in \mathbb{R}^{3 \times 3}$$

- To control the orientation we need to define an error variable that indicates the deviation between the actual and the desired orientation

- The position error  $e_p$  is simply defined as  

$$e_p = p^d - p$$
- for the orientation error  $e_o$  difficulty arise due to the fact that we need to consider the algebra of the rotation group
- its expression depends on the particular representation that we chose for the orientation
  - rotation matrix
  - angle-axis
  - euler angles
  - quaternions

- let us consider the rotation matrix representing the desired  $wR_d$  and the actual  $wR_e$  end-effector orientations



- $wR_e$  is computed from  $q$  through direct kinematics
- The relative orientation can be obtained by composing subsequent rotations

$${}^w R_d = {}^w R_e \cdot {}^e R_d$$

left-multiply by  ${}^w R_e^T$

$$\boxed{{}^e R_d = {}^w R_e^T {}^w R_d}$$

orientation of frame d  
with respect to frame e

- Note The orientation error can be defined in 2 ways according to which coordinate frame is used as reference (we use frame e)
  - ⊖ Using directly  $e R_d$  for orientation control is limited, due to the difficulty of handling the 9 elements of the matrix
- ⇒ better to express  $e R_d$  in terms of angle-axis

## ORIENTATION ERROR WITH ANGLE-AXIS

- let's define the orientation error  $e_o$  as:

$$e_o = \hat{r} \sin(\Delta\theta)$$

[LUH 1980]

- using this definition it is possible to proof  
That we can link  $e_o$  to the elements of  $eR_d = R$

$$e_o = \frac{1}{2} \begin{bmatrix} R_{32} - R_{23} \\ R_{13} - R_{31} \\ R_{21} - R_{12} \end{bmatrix}$$

- otherwise if we define it as  $e_o = \hat{r} \Delta\theta$  we use the formulas to compute separately  $\hat{r}, \Delta\theta$  of Lecture E02

- note That  $e_0$  is expressed in the frame  $e$  and Therefore it should be mapped in the same frame of  $S$  (i.e  $w$ ) before using for the computation of the restoring moment  $M$ :

$$w e_0 = w R_e e_0$$

## ORIENTATION ERROR WITH EULER ANGLES

- we can define the orientation error analogously to the position error:

$$e_o = \Phi_d - \Phi_e(q)$$

- where  $\Phi_d$  and  $\Phi_e$  can be extracted from the rotation matrix  ${}^w R_d$  and  ${}^e R_d$
- if we compute the moment  $m_\phi = k_o e_o$  will be defined in a frame with non-orthogonal axes
- let us use the analytic jacobian  $J_A$  instead of the geometric one to map to torques:



$$u = J_{A,0}^T K_0 e_0 + J_0^T D_0 (\omega^d - \omega)$$

or if we went to use  $\dot{\phi}$  for damping

$$u = J_{A,0}^T [K_0 e_0 + D_0 (\dot{\phi}^d - \dot{\phi})]$$

$\hookrightarrow J_{A,0} = T_w(\phi) J_0 \rightarrow$  angular part of geometric jacobian

- ⊖ There are representation singularities in  $J_A$  due to  $T_w(\phi)$
- ⊖ Need to derive  $\dot{\phi}_e$  from  $wR_e(q)$  with  $\approx 2\pi m/2 \rightarrow$  (complex) because we have no closed form expression from  $q$  to  $\dot{\phi}_e$

## ORIENTATION ERROR WITH QUATERNIONS

① compute the quaternions  $Q_e$  and  $Q_d$  associated to  $wR_e$  and  $wR_d$

$$Q = \frac{1}{2} \begin{bmatrix} A \\ (R_{23} - R_{32}) / A \\ (R_{31} - R_{13}) / A \\ (R_{12} - R_{21}) / A \end{bmatrix}, \quad A = (1 + R_{11} + R_{22} + R_{33})^{1/2}$$

② composition of quaternions is analogous to composition of rotation matrix and is used. The quaternion multiplication  $\otimes$  formula

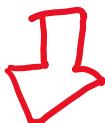
$$R_3 = R_1 \circ R_2$$



$$Q_3 = Q_1 \otimes Q_2 = \begin{bmatrix} n_1 n_2 - \varepsilon_1^T \varepsilon_2 \\ n_1 \varepsilon_2 + n_2 \varepsilon_1 + \varepsilon_1 \times \varepsilon_2 \end{bmatrix}$$

in our case The quaternion that represents The relative orientation of frame d with respect to frame e is:

$${}^e R_d = {}_w R_e^T {}_w R_d$$



$$\Delta Q = \bar{Q}_e \otimes Q_d = \begin{bmatrix} n_e \\ -\epsilon_e \end{bmatrix} \otimes \begin{bmatrix} n_d \\ \epsilon_d \end{bmatrix}$$

↓

quaternion  
inverse

$$= \begin{bmatrix} n_e n_d + \epsilon_e^T \epsilon_d \\ n_e \epsilon_d - n_d \epsilon_e - \epsilon_e \times \epsilon_d \end{bmatrix}$$

desired

$\epsilon_e$  : vector part  
of  $\Delta Q$

- remember That  $e_o$  is expressed in e frame and need To be mapped into w frame

- ⊕ no singularity
- ⊖ requires intermediate computation of  $Q_e$  from  $wR_e(q)$  because there is no direct mapping from  $q$  to  $Q_e$

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- Quaternion kinematics for the error-state Kalman filter, Joan Solà, 2017.