

ECE 147C Final Project

Max Crisafulli & Eric Hsieh

ECE 147C

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Rotational Inverted Pendulum

Why Use This Hardware?

- Quanser Hardware allows for easy interaction with MATLAB/SIMULINK via the Quarc package
- Classic (and fun) toy example of LQR control of an unstable system.



Figure 1: The Quanser SRV02 Rotational Pendulum

Equations of Motion - Conventions

Lagrangian EoM Method

$$\mathcal{L} = T^* - V$$

$$\frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \dot{\xi}_i} - \frac{\partial \mathcal{L}}{\partial \xi_i} = \Xi_i$$

$$\xi_i = \begin{bmatrix} \theta \\ \alpha \end{bmatrix}$$

$$\Xi_i = \begin{bmatrix} \tau - B_r \dot{\theta} \\ -B_p \dot{\alpha} \end{bmatrix}$$

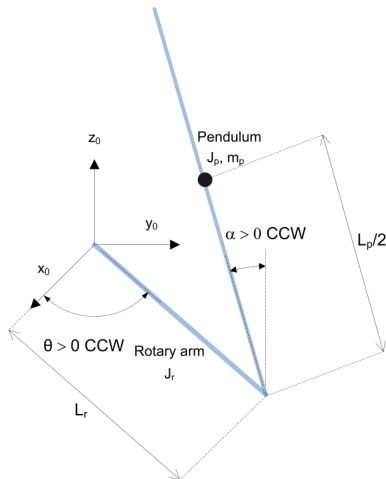


Figure 2: Inverted Pendulum Diagram

Equations of Motion - Full Nonlinear

$$\text{EoM \# 1 - } \xi_1 = \theta : \frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \dot{\theta}} - \frac{\partial \mathcal{L}}{\partial \theta} = \tau - B_r \dot{\theta}$$

$$\left(m_p L_r^2 + \frac{1}{4} m_p L_p^2 - \frac{1}{4} m_p L_p \cos(\alpha)^2 + J_r \right) \ddot{\theta} - \left(\frac{1}{2} m_p L_p L_r \cos(\alpha) \right) \ddot{\alpha} \\ + \left(\frac{1}{2} m_p L_p^2 \sin(\alpha) \cos(\alpha) \right) \dot{\theta} \dot{\alpha} + \left(\frac{1}{2} m_p L_p L_r \sin(\alpha) \right) \dot{\alpha}^2 = \tau - B_r \dot{\theta}$$

$$\text{EoM \# 2 - } \xi_2 = \alpha : \frac{d}{dt} \frac{\partial \mathcal{L}}{\partial \dot{\alpha}} - \frac{\partial \mathcal{L}}{\partial \alpha} = -B_p \dot{\alpha}$$

$$-\frac{1}{2} m_p L_p L_r \cos(\alpha) \ddot{\theta} + \left(J_p + \frac{1}{4} m_p L_p^2 \right) \ddot{\alpha} - \frac{1}{4} m_p L_p^2 \cos(\alpha) \sin(\alpha) \dot{\theta}^2 \\ - \frac{1}{2} m_p L_p g \sin(\alpha) = -B_p \dot{\alpha}$$

$$\text{Torque as a Function of Voltage: } \tau = \frac{\eta_g K_g \eta_m k_t (V_m - K_g k_m \dot{\theta})}{R_m}$$

Equations of Motion - Linearization

With a non-linear $f(z)$, you can linearize around the upright position as follows:

$$z^T = [\theta, \alpha, \dot{\theta}, \dot{\alpha}, \ddot{\theta}, \ddot{\alpha}] \quad \& \quad z_0^T = [0, 0, 0, 0, 0, 0] \quad (\text{Upright Equilibrium})$$

$$\textbf{General Linearization Form: } f_{\text{lin}}(z) = f(z_0) + \sum_{\forall z} \left(\frac{\partial f(z)}{\partial z_i} \bigg|_{z=z_0} \cdot z_i \right)$$

$$\textbf{Linearized EoM \#1: } (m_p L_r^2 + J_r) \ddot{\theta} - \frac{1}{2} m_p L_p L_r \ddot{\alpha} = \tau - B_r \dot{\theta}$$

$$\textbf{Linearized EoM \#2: } -\frac{1}{2} m_p L_p L_r \ddot{\theta} + \left(J_p + \frac{1}{4} m_p L_p^2 \right) \ddot{\alpha} - \frac{1}{2} m_p L_p g \alpha = -B_p \dot{\alpha}$$

Organizing into matrix EoM form $M(q)\ddot{q} + B(q, \dot{q})\dot{q} + G(q) = \tau$, $q^T = [\theta, \alpha]$

$$\begin{bmatrix} m_p L_r^2 + J_r & -\frac{1}{2} m_p L_p L_r \\ -\frac{1}{2} m_p L_p L_r & J_p + \frac{1}{4} m_p L_p^2 \end{bmatrix} \begin{bmatrix} \ddot{\theta} \\ \ddot{\alpha} \end{bmatrix} + \begin{bmatrix} B_r & 0 \\ 0 & B_p \end{bmatrix} \begin{bmatrix} \dot{\theta} \\ \dot{\alpha} \end{bmatrix} + \begin{bmatrix} 0 \\ -\frac{1}{2} m_p L_p g \alpha \end{bmatrix} = \begin{bmatrix} \tau \\ 0 \end{bmatrix}$$

$$\textbf{Reorganizing: } \begin{bmatrix} \ddot{\theta} \\ \ddot{\alpha} \end{bmatrix} = \frac{1}{J_T} \begin{bmatrix} J_p + \frac{1}{4} m_p L_p^2 & \frac{1}{2} m_p L_p L_r \\ \frac{1}{2} m_p L_p L_r & J_r + m_p L_r^2 \end{bmatrix} \begin{bmatrix} \tau - B_r \dot{\theta} \\ \frac{1}{2} m_p L_p g \alpha - B_p \dot{\alpha} \end{bmatrix}$$

Where $J_T = \det(M(q))$

Linearized Matrices (Continuous State Space Model)

$$\ddot{\theta} = \frac{1}{J_T} \left(- \left(J_p + \frac{1}{4} m_p L_p^2 \right) B_r \dot{\theta} - \frac{1}{2} m_p L_p L_r B_p \dot{\alpha} + \frac{1}{4} m_p^2 L_p^2 L_r g \alpha + \left(J_p + \frac{1}{4} m_p L_p^2 \right) \tau \right)$$

$$\ddot{\alpha} = \frac{1}{J_T} \left(\frac{1}{2} m_p L_p L_r B_r \dot{\theta} - (J_r + m_p L_r^2) B_p \dot{\alpha} + \frac{1}{2} m_p L_p g (J_r + m_p L_r^2) \alpha + \frac{1}{2} m_p L_p L_r \tau \right)$$

$$A = \frac{1}{J_T} \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & \frac{1}{4} m_p^2 L_p^2 L_r g & -(J_p + \frac{1}{4} m_p L_p^2) B_r & -\frac{1}{2} m_p L_p L_r B_p \\ 0 & \frac{1}{2} m_p L_p g (J_r + m_p L_r^2) & \frac{1}{2} m_p L_p L_r B_r & -(J_r + m_p L_r^2) B_p \end{bmatrix}$$

$$B = \frac{K_g k_t}{R_m J_T} \begin{bmatrix} 0 \\ 0 \\ J_p + \frac{1}{4} m_p L_p^2 \\ \frac{1}{2} m_p L_p L_r \end{bmatrix} \quad C = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \end{bmatrix} \quad x = \begin{bmatrix} \theta \\ \alpha \\ \dot{\theta} \\ \dot{\alpha} \end{bmatrix}$$

$$\begin{bmatrix} \dot{\theta} \\ \dot{\alpha} \\ \ddot{\theta} \\ \ddot{\alpha} \end{bmatrix} = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \\ 0 & 81.40 & -10.25 & -0.93 \\ 0 & 122.05 & -10.33 & -1.39 \end{bmatrix} \begin{bmatrix} \theta \\ \alpha \\ \dot{\theta} \\ \dot{\alpha} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \\ 83.47 \\ -80.32 \end{bmatrix} V_m$$

Energy-Based Swing Up Control

NOTE: The α used in this context is $\alpha = 0$ being the DOWNWARD position.

$$J_p \ddot{\alpha} + \frac{1}{2} m_p g L_p \sin(\alpha) = \frac{1}{2} m_p L_p u \cos(\alpha)$$

$$E_p = \frac{1}{2} m_p g L_p (1 - \cos(\alpha)), \quad E_k = \frac{1}{2} J_p \dot{\alpha}^2, \quad E = E_p + E_k$$

Using the idea of nonlinear control:

$$\tau = \frac{\eta_g K_g \eta_m k_t (u - K_g k_m \dot{\theta})}{R_m}$$

$$u = \frac{m_r L_r R_m}{\eta_g K_g \eta_m k_t} \cdot \mu \cdot \text{sign}\{\cos(\alpha) \dot{\alpha}\} \cdot (E - E_r) + K_g k_m \dot{\theta}$$

Where μ is the swing-up gain and E_r is the potential energy of the upright pendulum ($\alpha = \pi$), in this case $m_p g L_p = 0.42\text{J}$.

LQR/LQG Balancing Control

LQR Balancing Control was designed using Bryson's Rule where:

$$Q_{ii} = \frac{1}{(x_{i,max})^2}, \text{ and } R_{ii} = \frac{1}{(u_{max})^2}$$
$$x_{max} = \begin{bmatrix} \theta_{max} \\ \alpha_{max} \\ \dot{\theta}_{max} \\ \dot{\alpha}_{max} \end{bmatrix} = \begin{bmatrix} 4^\circ \cdot \frac{\pi}{180^\circ} \text{ rad} \\ 1.5^\circ \cdot \frac{\pi}{180^\circ} \text{ rad} \\ 100 \frac{\text{rad}}{\text{sec}} \\ 1 \frac{\text{rad}}{\text{sec}} \end{bmatrix} \quad \& \quad u_{max} = 5V$$

The states θ and α were measured and used to predict ALL the states using the linearized system model in a Kalman Filter (LQG) state estimator.

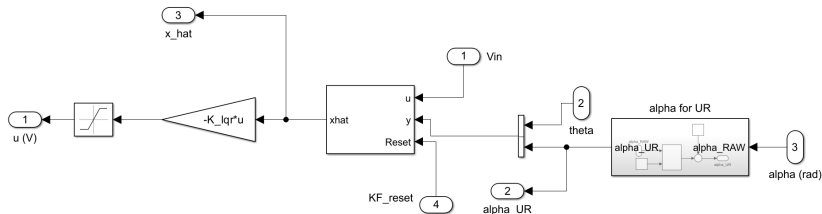


Figure 3: LQR Kalman Filter Feedback

Kalman Filter State Estimation

For a discrete-time linear system described by the following form. Our discrete SS model was generated via the `c2d` command in MATLAB from our continuous model. The A, B, C matrices in this case are the outputs of the `c2d` function.

$$x_{k+1} = A_k x_k + B_k u_k + w_k, \quad w_k \sim \mathcal{N}(0, Q_k)$$

$$y_k = C_k x_k + v_k, \quad v_k \sim \mathcal{N}(0, R_k)$$

The Kalman Filter Equations are defined as follows.

Predict State: $\hat{x}_{k|k-1} = A_k \hat{x}_{k-1|k-1} + B_k u_k$

Predict Covariance: $\hat{P}_{k|k-1} = A_k P_{k-1|k-1} A_k^T + Q_k$

Kalman Gain: $K_k = \hat{P}_{k|k-1} C_k^T (C_k \hat{P}_{k|k-1} C_k^T + R_k)^{-1}$

Update State: $x_{k|k} = \hat{x}_{k|k-1} + K_k (y_k - C_k \hat{x}_{k|k-1})$

Update Covariance: $P_{k|k} = (I - K_k C_k) \hat{P}_{k|k-1}$

Switching Logic & Upright Angle

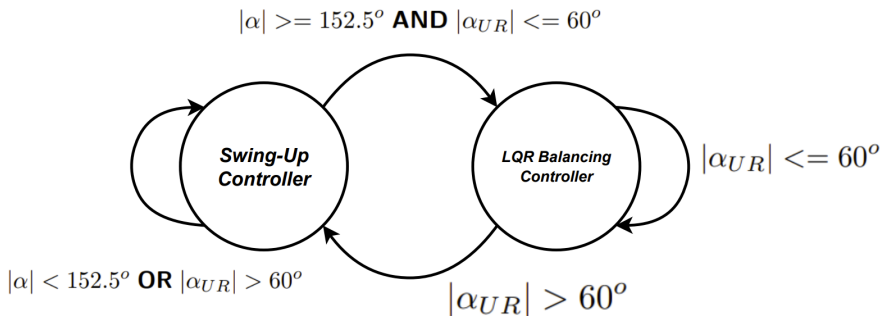


Figure 4: Controller Switching Logic

NOTE: In this case α is the ABSOLUTE sensor measurement where $\alpha = 0$ is the downward position. The upright alpha α_{UR} is the augmented α for use with the linearized LQR controller and Kalman Filter state estimator, defined as follows:

$$\alpha_{UR} = \text{mod}(\alpha, 2\pi) - \pi$$

Implementation Results

Please turn your attention to the inverted pendulum for a demonstration...